

# Detailed Member Calculations

**Units: N&mm**

**Regulation: ASCE 41-17**

## Calculation No. 1

column C1, Floor 1

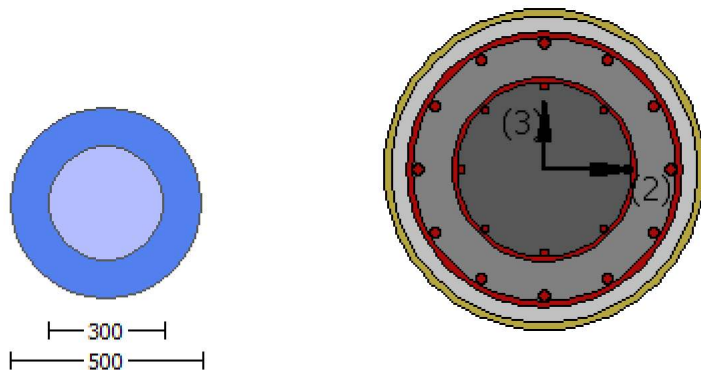
Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Shear capacity  $VR_d$

Edge: Start

Local Axis: (2)



Start Of Calculation of Shear Capacity for element: column JCC1 of floor 1

At local axis: 2

Integration Section: (a)

Section Type: rcjcs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE 41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Jacket

New material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 25.00$

New material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 500.00$

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Concrete Elasticity, Ec = 26999.444
Steel Elasticity, Es = 200000.00
Existing Column
New material of Primary Member: Concrete Strength, fc = fc_lower_bound = 25.00
New material of Primary Member: Steel Strength, fs = fs_lower_bound = 500.00
Concrete Elasticity, Ec = 26999.444
Steel Elasticity, Es = 200000.00
#####
Note: Especially for the calculation of  $\gamma$  for displacement ductility demand,
the expected (mean value) strengths are used (7.5.1.3, ASCE 41-17) because bending is considered as
Deformation-Controlled Action (Table C7-1, ASCE 41-17).
Jacket
New material: Concrete Strength, fc = fcm = 33.00
New material: Steel Strength, fs = fsm = 555.56
Existing Column
New material: Concrete Strength, fc = fcm = 33.00
New material: Steel Strength, fs = fsm = 555.56
#####
External Diameter, D = 500.00
Internal Diameter, D = 300.00
Cover Thickness, c = 25.00
Element Length, L = 3000.00
Primary Member
Ribbed Bars
Ductile Steel
Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)
Longitudinal Bars With Ends Lapped Starting at the End Sections
Inadequate Lap Length with  $l_o/l_{ou,min} = l_b/l_d = 0.30$ 
FRP Wrapping Data
Type: Carbon
Dry properties (design values)
Thickness, t = 1.00
Tensile Strength,  $f_{fu} = 840.00$ 
Tensile Modulus,  $E_f = 82000.00$ 
Elongation,  $e_{fu} = 0.009$ 
Number of directions, NoDir = 1
Fiber orientations,  $b_i = 0.00^\circ$ 
Number of layers, NL = 1
Radius of rounding corners, R = 40.00
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Stepwise Properties
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EDGE -A-
Bending Moment,  $M_a = -1.4341E+007$ 
Shear Force,  $V_a = -4779.203$ 
EDGE -B-
Bending Moment,  $M_b = 0.07981876$ 
Shear Force,  $V_b = 4779.203$ 
BOTH EDGES
Axial Force,  $F = -7387.302$ 
Longitudinal Reinforcement Area Distribution (in 2 divisions)
-Tension:  $As_t = 0.00$ 
-Compression:  $As_c = 3053.628$ 
Longitudinal Reinforcement Area Distribution (in 3 divisions)
-Tension:  $As_{t,ten} = 1017.876$ 
-Compression:  $As_{l,com} = 1017.876$ 
-Middle:  $As_{l,mid} = 1017.876$ 
Mean Diameter of Tension Reinforcement,  $Db_{L,ten} = 18.00$ 
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New component: From table 7-7, ASCE 41_17: Final Shear Capacity  $V_R = 1.0 * V_n = 515957.182$ 
 $V_n$  ((10.3), ASCE 41-17) =  $knI * V_{CoI0} = 515957.182$ 
 $V_{CoI} = 515957.182$ 
 $knI = 1.00$ 

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displacement\_ductility\_demand = 0.01760997

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

Mean concrete strength:  $f_c' = (f_c'_{\text{jacket}} \cdot \text{Area}_{\text{jacket}} + f_c'_{\text{core}} \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 25.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 4.00$

$\mu_u = 1.4341 \times 10^7$

$V_u = 4779.203$

$d = 0.8 \cdot D = 400.00$

$N_u = 7387.302$

$A_g = 196349.541$

From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 246740.11$

$V_{s1} = 246740.11$  is calculated for jacket, with:

$A_v = \pi/2 \cdot A_{\text{stirrup}} = 123370.055$

$f_y = 500.00$

$s = 100.00$

$V_{s1}$  is multiplied by  $\text{Col1} = 1.00$

$s/d = 0.25$

$V_{s2} = 0.00$  is calculated for core, with:

$A_v = \pi/2 \cdot A_{\text{stirrup}} = 78956.835$

$f_y = 500.00$

$s = 250.00$

$V_{s2}$  is multiplied by  $\text{Col2} = 0.00$

$s/d = 1.04167$

$V_f$  ((11-3)-(11.4), ACI 440) = 308320.00

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression,

where  $a$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\alpha = 45^\circ$  and  $\alpha = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\alpha_1 = \alpha_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \alpha_1)|, |V_f(-45, \alpha_1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{\text{Dir}} = 1.00$

$d_{fv} = d$  (figure 11.2, ACI 440) = 470.00

$f_{fe}$  ((11-5), ACI 440) = 328.00

$E_f = 82000.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.009$

From (11-11), ACI 440:  $V_s + V_f \leq 417394.406$

$b_w \cdot d = \pi \cdot d^2 / 4 = 125663.706$

displacement\_ductility\_demand is calculated as  $\delta / y$

- Calculation of  $\delta / y$  for END A -  
for rotation axis 3 and integ. section (a)

From analysis, chord rotation  $\theta = 0.00015961$

$y = (M_y \cdot L_s / 3) / E_{\text{eff}} = 0.00906346$  ((4.29), Biskinis Phd))

$M_y = 2.2517 \times 10^8$

$L_s = M/V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) = 3000.736

From table 10.5, ASCE 41\_17:  $E_{\text{eff}} = \text{factor} \cdot E_c \cdot I_g = 2.4850 \times 10^{13}$

factor = 0.30

$A_g = 196349.541$

Mean concrete strength:  $f_c' = (f_c'_{\text{jacket}} \cdot \text{Area}_{\text{jacket}} + f_c'_{\text{core}} \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 33.00$

$N = 7387.302$

$E_c \cdot I_g = E_c \cdot I_{g_{\text{jacket}}} + E_c \cdot I_{g_{\text{core}}} = 8.2833 \times 10^{13}$

Calculation of Yielding Moment  $M_y$

Calculation of  $y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 2.2517\text{E}+008$

$y = 6.4424840\text{E}-006$

$M_{y\_ten} (8c) = 2.2517\text{E}+008$

$_{y\_ten} (7c) = 61.8137$

error of function (7c) = 0.00026387

$M_{y\_com} (8d) = 8.1922\text{E}+008$

$_{y\_com} (7d) = 63.67468$

error of function (7d) = -0.0087488

with ((10.1), ASCE 41-17)  $e_y = \text{Min}(e_y, 1.25 \cdot e_y \cdot (l_b/l_d)^{2/3}) = 0.0027778$

$e_{co} = 0.002$

$a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)

$d_1 = 44.00$

$R = 250.00$

$v = 0.00101374$

$N = 7387.302$

$A_c = 196349.541$

((10.1), ASCE 41-17)  $= \text{Min}( , 1.25 \cdot \cdot (l_b/l_d)^{2/3}) = 0.23280363$

with  $f_c^*$  ((12.3), ACI 440) = 37.11312

$f_c = 33.00$

$f_l = 1.312$

$k = 1$

Effective FRP thickness,  $t_f = N L \cdot t \cdot \cos(b_1) = 1.00$

$e_{fe}$  ((12.5) and (12.7)) = 0.004

$f_u = 0.009$

$E_f = 82000.00$

Calculation of ratio  $l_b/l_d$

Inadequate Lap Length with  $l_b/l_d = 0.30$

End Of Calculation of Shear Capacity for element: column JCC1 of floor 1

At local axis: 2

Integration Section: (a)

## Calculation No. 2

column C1, Floor 1

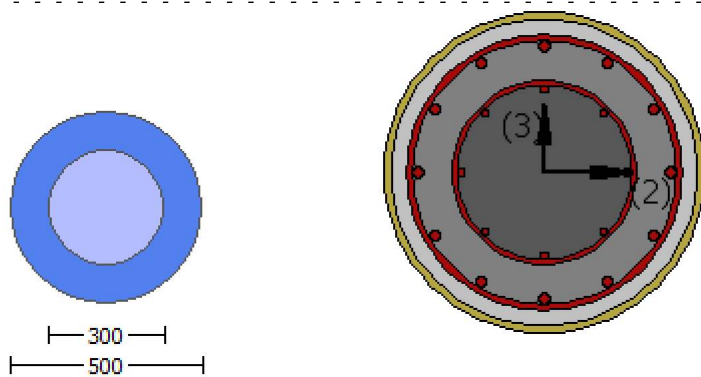
Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity (  $\mu$  )

Edge: Start

Local Axis: (2)



Start Of Calculation of Shear Capacity ratio for element: column JCC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rcjcs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Mean strength values are used for both shear and moment calculations.

Consequently:

Jacket

New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

Existing Column

New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Jacket

New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 694.45$

Existing Column

New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 694.45$

#####

External Diameter,  $D = 500.00$

Internal Diameter,  $D = 300.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.51481

Element Length,  $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Inadequate Lap Length with  $l_o/l_{ou,min} = 0.30$

FRP Wrapping Data

Type: Carbon

Dry properties (design values)

Thickness,  $t = 1.00$

Tensile Strength,  $f_{fu} = 840.00$

Tensile Modulus,  $E_f = 82000.00$

Elongation,  $ef_u = 0.009$

Number of directions,  $NoDir = 1$

Fiber orientations,  $bi: 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

## Stepwise Properties

At local axis: 3

EDGE -A-

Shear Force,  $V_a = 3.4807814E-030$

EDGE -B-

Shear Force,  $V_b = -3.4807814E-030$

BOTH EDGES

Axial Force,  $F = -7389.214$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 0.00$

-Compression:  $As_c = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t,ten} = 1017.876$

-Compression:  $As_{l,com} = 1017.876$

-Middle:  $As_{l,mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.23017425$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 162459.651$

with

$M_{pr1} = \text{Max}(\mu_{1+}, \mu_{1-}) = 2.4369E+008$

$\mu_{1+} = 2.4369E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$\mu_{1-} = 2.4369E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(\mu_{2+}, \mu_{2-}) = 2.4369E+008$

$\mu_{2+} = 2.4369E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination

$\mu_{2-} = 2.4369E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

Calculation of  $\mu_{1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$

$\mu_u = 2.4369E+008$

$= 0.80285146$

$' = 0.71403327$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TDY:  $f_{cc} = f_c^* \quad c = 49.98874$

conf. factor  $c = 1.51481$

$f_c = 33.00$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 389.0139$

$l_b/l_d = 0.30$

$d_1 = 44.00$

$R = 250.00$

$v = 0.00100813$

$N = 7389.214$

$A_c = 196349.541$

$= \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.1620683$

Calculation of ratio  $l_b/l_d$

Inadequate Lap Length with  $l_b/l_d = 0.30$

Calculation of  $\mu_{1-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.4369E+008

= 0.80285146  
' = 0.71403327  
error of function (3.68), Biskinis Phd = 27987.158  
From 5A.2, TDY: fcc = fc\* c = 49.98874  
conf. factor c = 1.51481  
fc = 33.00  
From 10.3.5, ASCE 41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 389.0139  
lb/d = 0.30  
d1 = 44.00  
R = 250.00  
v = 0.00100813  
N = 7389.214  
Ac = 196349.541  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.1620683

Calculation of ratio lb/d

Inadequate Lap Length with lb/d = 0.30

Calculation of Mu2+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.4369E+008

= 0.80285146  
' = 0.71403327  
error of function (3.68), Biskinis Phd = 27987.158  
From 5A.2, TDY: fcc = fc\* c = 49.98874  
conf. factor c = 1.51481  
fc = 33.00  
From 10.3.5, ASCE 41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 389.0139  
lb/d = 0.30  
d1 = 44.00  
R = 250.00  
v = 0.00100813  
N = 7389.214  
Ac = 196349.541  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.1620683

Calculation of ratio lb/d

Inadequate Lap Length with lb/d = 0.30

Calculation of Mu2-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.4369E+008

= 0.80285146  
' = 0.71403327  
error of function (3.68), Biskinis Phd = 27987.158  
From 5A.2, TDY: fcc = fc\* c = 49.98874

conf. factor  $c = 1.51481$   
 $f_c = 33.00$   
 From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 389.0139$   
 $l_b/l_d = 0.30$   
 $d_1 = 44.00$   
 $R = 250.00$   
 $v = 0.00100813$   
 $N = 7389.214$   
 $A_c = 196349.541$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.1620683$

Calculation of ratio  $l_b/l_d$

Inadequate Lap Length with  $l_b/l_d = 0.30$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 705811.584$

Calculation of Shear Strength at edge 1,  $V_{r1} = 705811.584$

$V_{r1} = V_{\text{Col}} ((10.3), \text{ASCE 41-17}) = k_{nl} \cdot V_{\text{Col}0}$

$V_{\text{Col}0} = 705811.584$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

Mean concrete strength:  $f'_c = (f'_c_{\text{jacket}} \cdot \text{Area}_{\text{jacket}} + f'_c_{\text{core}} \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 33.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.2438899\text{E-}011$

$\nu_u = 3.4807814\text{E-}030$

$d = 0.8 \cdot D = 400.00$

$N_u = 7389.214$

$A_g = 196349.541$

From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 274157.871$

$V_{s1} = 274157.871$  is calculated for jacket, with:

$A_v = \cdot /2 \cdot A_{\text{stirrup}} = 123370.055$

$f_y = 555.56$

$s = 100.00$

$V_{s1}$  is multiplied by  $\text{Col}1 = 1.00$

$s/d = 0.25$

$V_{s2} = 0.00$  is calculated for core, with:

$A_v = \cdot /2 \cdot A_{\text{stirrup}} = 78956.835$

$f_y = 555.56$

$s = 250.00$

$V_{s2}$  is multiplied by  $\text{Col}2 = 0.00$

$s/d = 1.04167$

$V_f ((11-3)-(11.4), \text{ACI 440}) = 308320.00$

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression, where  $\alpha$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\alpha = 45^\circ$  and  $\alpha = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\alpha_1 = b_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \alpha_1)|, |V_f(-45, \alpha_1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{\text{Dir}} = 1.00$

$d_{fv} = d$  (figure 11.2, ACI 440) = 470.00

$f_{fe} ((11-5), \text{ACI 440}) = 328.00$

$E_f = 82000.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.009$

From (11-11), ACI 440:  $V_s + V_f \leq 479549.663$



$$b_w \cdot d = \frac{b \cdot d^2}{4} = 125663.706$$

Calculation of Shear Strength at edge 2,  $V_{r2} = 705811.584$

$$V_{r2} = V_{Col} \text{ ((10.3), ASCE 41-17)} = k_{nl} \cdot V_{Col0}$$

$$V_{Col0} = 705811.584$$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

Mean concrete strength:  $f_c' = (f_c'_{jacket} \cdot Area_{jacket} + f_c'_{core} \cdot Area_{core}) / Area_{section} = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$$M / Vd = 2.00$$

$$\mu_u = 1.2438899E-011$$

$$\nu_u = 3.4807814E-030$$

$$d = 0.8 \cdot D = 400.00$$

$$N_u = 7389.214$$

$$A_g = 196349.541$$

From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 274157.871$

$V_{s1} = 274157.871$  is calculated for jacket, with:

$$A_v = \frac{1}{2} \cdot A_{stirrup} = 123370.055$$

$$f_y = 555.56$$

$$s = 100.00$$

$V_{s1}$  is multiplied by  $Col1 = 1.00$

$$s/d = 0.25$$

$V_{s2} = 0.00$  is calculated for core, with:

$$A_v = \frac{1}{2} \cdot A_{stirrup} = 78956.835$$

$$f_y = 555.56$$

$$s = 250.00$$

$V_{s2}$  is multiplied by  $Col2 = 0.00$

$$s/d = 1.04167$$

$$V_f \text{ ((11-3)-(11.4), ACI 440)} = 308320.00$$

$f = 0.95$ , for fully-wrapped sections

$w_f / s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression, where  $a$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = b1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, a1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{Dir} = 1.00$

$$d_{fv} = d \text{ (figure 11.2, ACI 440)} = 470.00$$

$$f_{fe} \text{ ((11-5), ACI 440)} = 328.00$$

$$E_f = 82000.00$$

$$f_e = 0.004, \text{ from (11.6a), ACI 440}$$

$$\text{with } f_u = 0.009$$

From (11-11), ACI 440:  $V_s + V_f \leq 479549.663$

$$b_w \cdot d = \frac{b \cdot d^2}{4} = 125663.706$$

End Of Calculation of Shear Capacity ratio for element: column JCC1 of floor 1

At local axis: 3

Start Of Calculation of Shear Capacity ratio for element: column JCC1 of floor 1

At Shear local axis: 2

(Bending local axis: 3)

Section Type: rcjcs

Constant Properties

Knowledge Factor,  $\phi = 1.00$

Mean strength values are used for both shear and moment calculations.

```

Consequently:
Jacket
New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$ 
New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$ 
Concrete Elasticity,  $E_c = 26999.444$ 
Steel Elasticity,  $E_s = 200000.00$ 
Existing Column
New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$ 
New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$ 
Concrete Elasticity,  $E_c = 26999.444$ 
Steel Elasticity,  $E_s = 200000.00$ 
#####
Note: Especially for the calculation of moment strengths,
the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14
Jacket
New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 694.45$ 
Existing Column
New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 694.45$ 
#####
External Diameter,  $D = 500.00$ 
Internal Diameter,  $D = 300.00$ 
Cover Thickness,  $c = 25.00$ 
Mean Confinement Factor overall section = 1.51481
Element Length,  $L = 3000.00$ 
Primary Member
Ribbed Bars
Ductile Steel
Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)
Longitudinal Bars With Ends Lapped Starting at the End Sections
Inadequate Lap Length with  $l_o/l_{ou,min} = 0.30$ 
FRP Wrapping Data
Type: Carbon
Dry properties (design values)
Thickness,  $t = 1.00$ 
Tensile Strength,  $f_{fu} = 840.00$ 
Tensile Modulus,  $E_f = 82000.00$ 
Elongation,  $ef_u = 0.009$ 
Number of directions,  $NoDir = 1$ 
Fiber orientations,  $bi: 0.00^\circ$ 
Number of layers,  $NL = 1$ 
Radius of rounding corners,  $R = 40.00$ 
-----

Stepwise Properties
-----
At local axis: 2
EDGE -A-
Shear Force,  $V_a = -2.1312935E-046$ 
EDGE -B-
Shear Force,  $V_b = 2.1312935E-046$ 
BOTH EDGES
Axial Force,  $F = -7389.214$ 
Longitudinal Reinforcement Area Distribution (in 2 divisions)
-Tension:  $As_t = 0.00$ 
-Compression:  $As_c = 3053.628$ 
Longitudinal Reinforcement Area Distribution (in 3 divisions)
-Tension:  $As_{t,ten} = 1017.876$ 
-Compression:  $As_{l,com} = 1017.876$ 
-Middle:  $As_{l,mid} = 1017.876$ 
-----
-----

Calculation of Shear Capacity ratio ,  $V_e/V_r = 0.23017425$ 
Member Controlled by Flexure ( $V_e/V_r < 1$ )
Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 162459.651$ 
with

```

$$Mpr1 = \text{Max}(Mu1+, Mu1-) = 2.4369E+008$$

Mu1+ = 2.4369E+008, is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

Mu1- = 2.4369E+008, is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$$Mpr2 = \text{Max}(Mu2+, Mu2-) = 2.4369E+008$$

Mu2+ = 2.4369E+008, is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

Mu2- = 2.4369E+008, is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

Calculation of Mu1+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.4369E+008

$$= 0.80285146$$

$$' = 0.71403327$$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 49.98874$

conf. factor  $c = 1.51481$

$$f_c = 33.00$$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 389.0139$

$$l_b/d = 0.30$$

$$d1 = 44.00$$

$$R = 250.00$$

$$v = 0.00100813$$

$$N = 7389.214$$

$$Ac = 196349.541$$

$$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.1620683$$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of Mu1-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.4369E+008

$$= 0.80285146$$

$$' = 0.71403327$$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 49.98874$

conf. factor  $c = 1.51481$

$$f_c = 33.00$$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 389.0139$

$$l_b/d = 0.30$$

$$d1 = 44.00$$

$$R = 250.00$$

$$v = 0.00100813$$

$$N = 7389.214$$

$$Ac = 196349.541$$

$$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.1620683$$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of Mu2+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.4369E+008

$$= 0.80285146$$

$$' = 0.71403327$$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TBDY: fcc = fc\* c = 49.98874

conf. factor c = 1.51481

fc = 33.00

From 10.3.5, ASCE 41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 389.0139

$$lb/d = 0.30$$

$$d1 = 44.00$$

$$R = 250.00$$

$$v = 0.00100813$$

$$N = 7389.214$$

$$Ac = 196349.541$$

$$= *Min(1,1.25*(lb/d)^ 2/3) = 0.1620683$$

Calculation of ratio lb/d

Inadequate Lap Length with lb/d = 0.30

Calculation of Mu2-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.4369E+008

$$= 0.80285146$$

$$' = 0.71403327$$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TBDY: fcc = fc\* c = 49.98874

conf. factor c = 1.51481

fc = 33.00

From 10.3.5, ASCE 41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 389.0139

$$lb/d = 0.30$$

$$d1 = 44.00$$

$$R = 250.00$$

$$v = 0.00100813$$

$$N = 7389.214$$

$$Ac = 196349.541$$

$$= *Min(1,1.25*(lb/d)^ 2/3) = 0.1620683$$

Calculation of ratio lb/d

Inadequate Lap Length with lb/d = 0.30

Calculation of Shear Strength Vr = Min(Vr1,Vr2) = 705811.584

Calculation of Shear Strength at edge 1, Vr1 = 705811.584

Vr1 = VCol ((10.3), ASCE 41-17) = knl\*VColO

$$VColO = 705811.584$$

knl = 1 (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

Mean concrete strength:  $f_c' = (f_c'_{\text{jacket}} \cdot \text{Area}_{\text{jacket}} + f_c'_{\text{core}} \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.5940012\text{E-}011$

$V_u = 2.1312935\text{E-}046$

$d = 0.8 \cdot D = 400.00$

$N_u = 7389.214$

$A_g = 196349.541$

From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 274157.871$

$V_{s1} = 274157.871$  is calculated for jacket, with:

$A_v = /2 \cdot A_{\text{stirrup}} = 123370.055$

$f_y = 555.56$

$s = 100.00$

$V_{s1}$  is multiplied by  $\text{Col1} = 1.00$

$s/d = 0.25$

$V_{s2} = 0.00$  is calculated for core, with:

$A_v = /2 \cdot A_{\text{stirrup}} = 78956.835$

$f_y = 555.56$

$s = 250.00$

$V_{s2}$  is multiplied by  $\text{Col2} = 0.00$

$s/d = 1.04167$

$V_f$  ((11-3)-(11.4), ACI 440) = 308320.00

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression,

where  $a$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(, )$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $a = 45^\circ$  and  $a = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta_1 = b_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, a_1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / \text{NoDir} = 1.00$

$d_{fv} = d$  (figure 11.2, ACI 440) = 470.00

$f_{fe}$  ((11-5), ACI 440) = 328.00

$E_f = 82000.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.009$

From (11-11), ACI 440:  $V_s + V_f \leq 479549.663$

$b_w \cdot d = \cdot d \cdot d / 4 = 125663.706$

Calculation of Shear Strength at edge 2,  $V_{r2} = 705811.584$

$V_{r2} = V_{\text{Col}} ((10.3), \text{ASCE } 41-17) = \text{knl} \cdot V_{\text{Col0}}$

$V_{\text{Col0}} = 705811.584$

knl = 1 (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

Mean concrete strength:  $f_c' = (f_c'_{\text{jacket}} \cdot \text{Area}_{\text{jacket}} + f_c'_{\text{core}} \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.5940012\text{E-}011$

$V_u = 2.1312935\text{E-}046$

$d = 0.8 \cdot D = 400.00$

$N_u = 7389.214$

$A_g = 196349.541$

From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 274157.871$

$V_{s1} = 274157.871$  is calculated for jacket, with:

$A_v = /2 \cdot A_{\text{stirrup}} = 123370.055$

$f_y = 555.56$

$s = 100.00$   
 $V_{s1}$  is multiplied by  $Col1 = 1.00$   
 $s/d = 0.25$   
 $V_{s2} = 0.00$  is calculated for core, with:  
 $A_v = \sqrt{2} \cdot A_{stirrup} = 78956.835$   
 $f_y = 555.56$   
 $s = 250.00$   
 $V_{s2}$  is multiplied by  $Col2 = 0.00$   
 $s/d = 1.04167$   
 $V_f ((11-3)-(11.4), ACI 440) = 308320.00$   
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta_1 = \theta_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, a_1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{oDir} = 1.00$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 470.00  
 $f_{fe} ((11-5), ACI 440) = 328.00$   
 $E_f = 82000.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.009$   
 From (11-11), ACI 440:  $V_s + V_f \leq 479549.663$   
 $b_w \cdot d = \sqrt{3} \cdot d^2 / 4 = 125663.706$

-----  
 End Of Calculation of Shear Capacity ratio for element: column JCC1 of floor 1  
 At local axis: 2  
 -----

Start Of Calculation of Chord Rotation Capacity for element: column JCC1 of floor 1  
 At local axis: 2  
 Integration Section: (a)  
 Section Type: rcjcs

Constant Properties

-----  
 Knowledge Factor,  $K = 1.00$   
 Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE 41-17.  
 Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17  
 Consequently:  
 Jacket  
 New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$   
 New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$   
 Concrete Elasticity,  $E_c = 26999.444$   
 Steel Elasticity,  $E_s = 200000.00$   
 Existing Column  
 New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$   
 New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$   
 Concrete Elasticity,  $E_c = 26999.444$   
 Steel Elasticity,  $E_s = 200000.00$   
 External Diameter,  $D = 500.00$   
 Internal Diameter,  $D = 300.00$   
 Cover Thickness,  $c = 25.00$   
 Element Length,  $L = 3000.00$   
 Primary Member  
 Ribbed Bars  
 Ductile Steel  
 Without Detailing for Earthquake Resistance (including stirrups not closed at  $135^\circ$ )  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Inadequate Lap Length with  $l_b/l_d = 0.30$   
 FRP Wrapping Data  
 Type: Carbon

Dry properties (design values)  
 Thickness,  $t = 1.00$   
 Tensile Strength,  $f_{fu} = 840.00$   
 Tensile Modulus,  $E_f = 82000.00$   
 Elongation,  $\epsilon_{fu} = 0.009$   
 Number of directions,  $N_{oDir} = 1$   
 Fiber orientations,  $b_i = 0.00^\circ$   
 Number of layers,  $N_L = 1$   
 Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

Bending Moment,  $M = 7.3474831E-010$   
 Shear Force,  $V_2 = -4779.203$   
 Shear Force,  $V_3 = -2.1566928E-013$   
 Axial Force,  $F = -7387.302$   
 Longitudinal Reinforcement Area Distribution (in 2 divisions)  
   -Tension:  $A_{sl,t} = 0.00$   
   -Compression:  $A_{sl,c} = 3053.628$   
 Longitudinal Reinforcement Area Distribution (in 3 divisions)  
   -Tension:  $A_{sl,ten} = 1017.876$   
   -Compression:  $A_{sl,com} = 1017.876$   
   -Middle:  $A_{sl,mid} = 1017.876$   
 Mean Diameter of Tension Reinforcement,  $D_{bL} = 18.00$

New component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_R = 1.0^*$   $u = 0.00453062$   
 $u = y + p = 0.00453062$

- Calculation of  $y$  -

$y = (M_y * L_s / 3) / E_{eff} = 0.00453062$  ((4.29), Biskinis Phd))  
 $M_y = 2.2517E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 1500.00  
 From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 2.4850E+013$   
 $factor = 0.30$   
 $A_g = 196349.541$   
 Mean concrete strength:  $f'_c = (f'_{c,jacket} * Area_{jacket} + f'_{c,core} * Area_{core}) / Area_{section} = 33.00$   
 $N = 7387.302$   
 $E_c * I_g = E_{c,jacket} * I_{g,jacket} + E_{c,core} * I_{g,core} = 8.2833E+013$

#### Calculation of Yielding Moment $M_y$

Calculation of  $y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \min(M_{y,ten}, M_{y,com}) = 2.2517E+008$   
 $y = 6.4424840E-006$   
 $M_{y,ten} (8c) = 2.2517E+008$   
 $_{ten} (7c) = 61.8137$   
 error of function (7c) = 0.00026387  
 $M_{y,com} (8d) = 8.1922E+008$   
 $_{com} (7d) = 63.67468$   
 error of function (7d) = -0.0087488  
 with ((10.1), ASCE 41-17)  $e_y = \min(e_y, 1.25 * e_y * (I_b / I_d)^{2/3}) = 0.0027778$   
 $e_{co} = 0.002$   
 $a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 250.00$   
 $v = 0.00101374$   
 $N = 7387.302$   
 $A_c = 196349.541$

$((10.1), \text{ASCE 41-17}) = \text{Min}( , 1.25 * (lb/ld)^{2/3} ) = 0.23280363$   
 with  $f_c^* ((12.3), \text{ACI 440}) = 37.11312$   
 $f_c = 33.00$   
 $f_l = 1.312$   
 $k = 1$   
 Effective FRP thickness,  $t_f = NL * t * \cos(b1) = 1.00$   
 $e_{fe} ((12.5) \text{ and } (12.7)) = 0.004$   
 $f_u = 0.009$   
 $E_f = 82000.00$

Calculation of ratio  $lb/ld$

Inadequate Lap Length with  $lb/ld = 0.30$

- Calculation of  $p$  -

From table 10-9:  $p = 0.00$

with:

- Columns controlled by inadequate development or splicing along the clear height because  $lb/ld < 1$

shear control ratio  $V_y E / V_{col} E = 0.23017425$

$d = d_{\text{external}} = 0.00$

$s = s_{\text{external}} = 0.00$

$t = s1 + s2 + 2 * t_f / bw * (f_{fe} / f_s) = 0.00795744$

jacket:  $s1 = A_{v1} * (Dc1 / 2) / (s1 * A_g) = 0.0027646$

$A_{v1} = 78.53982$ , is the area of stirrup

$Dc1 = D_{\text{ext}} - 2 * \text{cover} - \text{External Hoop Diameter} = 440.00$ , is the total Length of all stirrups parallel to loading (shear) direction

$s1 = 100.00$

core:  $s2 = A_{v2} * (Dc2 / 2) / (s2 * A_g) = 0.00046968$

$A_{v2} = 50.26548$ , is the area of stirrup

$Dc2 = D_{\text{int}} - \text{Internal Hoop Diameter} = 292.00$ , is the total Length of all stirrups parallel to loading (shear) direction

$s2 = 250.00$

The term  $2 * t_f / bw * (f_{fe} / f_s)$  is implemented to account for FRP contribution

where  $f = 2 * t_f / bw$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength

All these variables have already been given in Shear control ratio calculation.

For the normalisation  $f_s$  of jacket is used.

$N_{UD} = 7387.302$

$A_g = 196349.541$

$f_{cE} = (f_{c\_jacket} * \text{Area\_jacket} + f_{c\_core} * \text{Area\_core}) / \text{section\_area} = 33.00$

$f_{yIE} = (f_{y\_ext\_Long\_Reinf} * \text{Area\_ext\_Long\_Reinf} + f_{y\_int\_Long\_Reinf} * \text{Area\_int\_Long\_Reinf}) / \text{Area\_Tot\_Long\_Rein} = 21219958E-314$

$f_{yTE} = (f_{y\_ext\_Trans\_Reinf} * \text{Area\_ext\_Trans\_Reinf} + f_{y\_int\_Trans\_Reinf} * \text{Area\_int\_Trans\_Reinf}) / \text{Area\_Tot\_Trans\_Rein} = 555.56$

$p_l = \text{Area\_Tot\_Long\_Rein} / (A_g) = 0.015552$

$f_{cE} = 33.00$

End Of Calculation of Chord Rotation Capacity for element: column JCC1 of floor 1

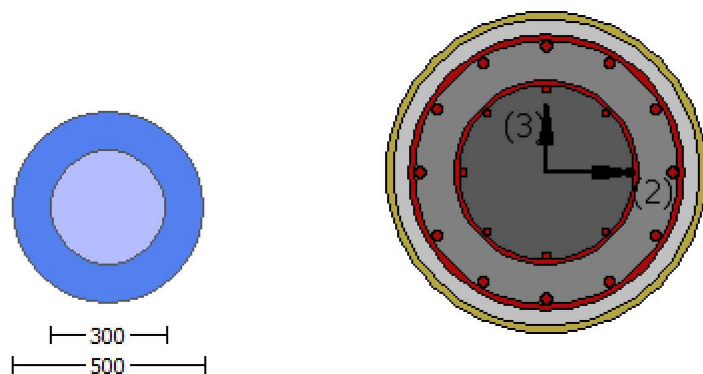
At local axis: 2

Integration Section: (a)

### Calculation No. 3



column C1, Floor 1  
 Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)  
 Analysis: Uniform +X  
 Check: Shear capacity VRd  
 Edge: Start  
 Local Axis: (3)



Start Of Calculation of Shear Capacity for element: column JCC1 of floor 1  
 At local axis: 3  
 Integration Section: (a)  
 Section Type: rcjcs

#### Constant Properties

Knowledge Factor,  $\gamma = 1.00$   
 Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE 41-17.  
 Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17  
 Consequently:

#### Jacket

New material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 25.00$   
 New material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 500.00$   
 Concrete Elasticity,  $E_c = 26999.444$   
 Steel Elasticity,  $E_s = 200000.00$

#### Existing Column

New material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 25.00$   
 New material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 500.00$   
 Concrete Elasticity,  $E_c = 26999.444$   
 Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of  $\gamma$  for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE 41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE 41-17).

#### Jacket

New material: Concrete Strength,  $f_c = f_{cm} = 33.00$   
 New material: Steel Strength,  $f_s = f_{sm} = 555.56$

#### Existing Column

New material: Concrete Strength,  $f_c = f_{cm} = 33.00$   
 New material: Steel Strength,  $f_s = f_{sm} = 555.56$

#####

External Diameter,  $D = 500.00$

Internal Diameter,  $D = 300.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Inadequate Lap Length with  $l_o/l_{ou,min} = l_b/l_d = 0.30$   
 FRP Wrapping Data  
 Type: Carbon  
 Dry properties (design values)  
 Thickness,  $t = 1.00$   
 Tensile Strength,  $f_{fu} = 840.00$   
 Tensile Modulus,  $E_f = 82000.00$   
 Elongation,  $e_{fu} = 0.009$   
 Number of directions,  $NoDir = 1$   
 Fiber orientations,  $bi: 0.00^\circ$   
 Number of layers,  $NL = 1$   
 Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

EDGE -A-  
 Bending Moment,  $M_a = 7.3474831E-010$   
 Shear Force,  $V_a = -2.1566928E-013$   
 EDGE -B-  
 Bending Moment,  $M_b = -8.7529383E-011$   
 Shear Force,  $V_b = 2.1566928E-013$   
 BOTH EDGES  
 Axial Force,  $F = -7387.302$   
 Longitudinal Reinforcement Area Distribution (in 2 divisions)  
   -Tension:  $As_t = 0.00$   
   -Compression:  $As_c = 3053.628$   
 Longitudinal Reinforcement Area Distribution (in 3 divisions)  
   -Tension:  $As_{t,ten} = 1017.876$   
   -Compression:  $As_{c,com} = 1017.876$   
   -Middle:  $As_{mid} = 1017.876$   
 Mean Diameter of Tension Reinforcement,  $Db_{L,ten} = 18.00$

New component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $VR = 1.0 \cdot V_n = 614519.957$   
 $V_n ((10.3), ASCE 41-17) = knl \cdot V_{Col0} = 614519.957$   
 $V_{Col} = 614519.957$   
 $knl = 1.00$   
 $displacement\_ductility\_demand = 0.00$

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_{s+} + f \cdot V_f$ '  
 where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)  
 Mean concrete strength:  $fc' = (fc'_{jacket} \cdot Area_{jacket} + fc'_{core} \cdot Area_{core}) / Area_{section} = 25.00$ , but  $fc'^{0.5} \leq 8.3$   
 MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $M_u = 7.3474831E-010$   
 $V_u = 2.1566928E-013$   
 $d = 0.8 \cdot D = 400.00$   
 $N_u = 7387.302$   
 $A_g = 196349.541$   
 From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 246740.11$   
 $V_{s1} = 246740.11$  is calculated for jacket, with:  
 $A_v = /2 \cdot A_{stirrup} = 123370.055$   
 $f_y = 500.00$   
 $s = 100.00$   
 $V_{s1}$  is multiplied by  $Col1 = 1.00$   
 $s/d = 0.25$   
 $V_{s2} = 0.00$  is calculated for core, with:  
 $A_v = /2 \cdot A_{stirrup} = 78956.835$   
 $f_y = 500.00$   
 $s = 250.00$

$V_{s2}$  is multiplied by  $Col2 = 0.00$   
 $s/d = 1.04167$   
 $V_f((11-3)-(11.4), ACI 440) = 308320.00$   
 $f = 0.95$ , for fully-wrapped sections  
 $wf/sf = 1$  (FRP strips adjacent to one another).  
 $\ln(11.3) \sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(, )$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $a = 45^\circ$  and  $a = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $b1 = 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, a1)|)$ , with:  
 total thickness per orientation,  $tf1 = NL * t / NoDir = 1.00$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 470.00  
 $f_{fe}((11-5), ACI 440) = 328.00$   
 $E_f = 82000.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.009$   
 From (11-11), ACI 440:  $V_s + V_f \leq 417394.406$   
 $bw * d = *d * d / 4 = 125663.706$

displacement\_ductility\_demand is calculated as  $\delta / y$

- Calculation of  $\delta / y$  for END A -  
 for rotation axis 2 and integ. section (a)

From analysis, chord rotation  $\theta = 9.5640318E-021$   
 $y = (M_y * L_s / 3) / E_{eff} = 0.00453062$  ((4.29), Biskinis Phd))  
 $M_y = 2.2517E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 1500.00  
 From table 10.5, ASCE 41-17:  $E_{eff} = \text{factor} * E_c * I_g = 2.4850E+013$   
 $\text{factor} = 0.30$   
 $A_g = 196349.541$   
 Mean concrete strength:  $f'_c = (f'_c_{\text{jacket}} * \text{Area}_{\text{jacket}} + f'_c_{\text{core}} * \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 33.00$   
 $N = 7387.302$   
 $E_c * I_g = E_{c_{\text{jacket}}} * I_{g_{\text{jacket}}} + E_{c_{\text{core}}} * I_{g_{\text{core}}} = 8.2833E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\delta$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y_{\text{ten}}}, M_{y_{\text{com}}}) = 2.2517E+008$   
 $y = 6.4424840E-006$   
 $M_{y_{\text{ten}}} (8c) = 2.2517E+008$   
 $\delta_{\text{ten}} (7c) = 61.8137$   
 error of function (7c) = 0.00026387  
 $M_{y_{\text{com}}} (8d) = 8.1922E+008$   
 $\delta_{\text{com}} (7d) = 63.67468$   
 error of function (7d) = -0.0087488  
 with ((10.1), ASCE 41-17)  $e_y = \text{Min}(e_y, 1.25 * e_y * (I_b / I_d)^{2/3}) = 0.0027778$   
 $e_{co} = 0.002$   
 $a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d1 = 44.00$   
 $R = 250.00$   
 $v = 0.00101374$   
 $N = 7387.302$   
 $A_c = 196349.541$   
 ((10.1), ASCE 41-17)  $\theta = \text{Min}(\theta, 1.25 * \theta * (I_b / I_d)^{2/3}) = 0.23280363$   
 with  $f'_c$  ((12.3), ACI 440) = 37.11312  
 $f_c = 33.00$   
 $f_l = 1.312$   
 $k = 1$   
 Effective FRP thickness,  $tf = NL * t * \cos(b1) = 1.00$   
 $e_{fe}((12.5) \text{ and } (12.7)) = 0.004$

$f_u = 0.009$   
 $E_f = 82000.00$

Calculation of ratio  $I_b/I_d$

Inadequate Lap Length with  $I_b/I_d = 0.30$

End Of Calculation of Shear Capacity for element: column JCC1 of floor 1

At local axis: 3

Integration Section: (a)

## Calculation No. 4

column C1, Floor 1

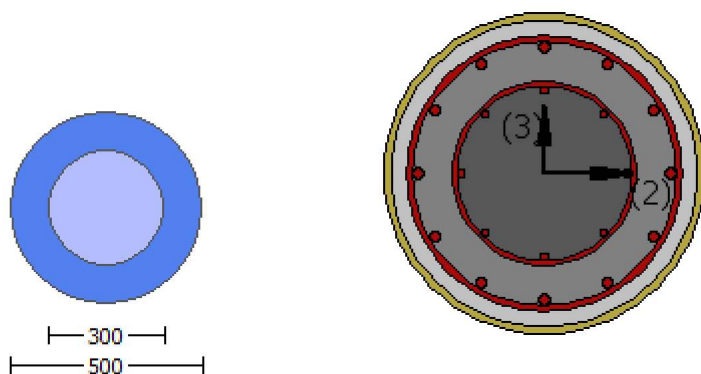
Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity (  $\phi$  )

Edge: Start

Local Axis: (3)



Start Of Calculation of Shear Capacity ratio for element: column JCC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rcjcs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Mean strength values are used for both shear and moment calculations.

Consequently:

Jacket

New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

Existing Column  
 New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$   
 New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$   
 Concrete Elasticity,  $E_c = 26999.444$   
 Steel Elasticity,  $E_s = 200000.00$   
 #####  
 Note: Especially for the calculation of moment strengths,  
 the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14  
 Jacket  
 New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 694.45$   
 Existing Column  
 New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 694.45$   
 #####  
 External Diameter,  $D = 500.00$   
 Internal Diameter,  $D = 300.00$   
 Cover Thickness,  $c = 25.00$   
 Mean Confinement Factor overall section = 1.51481  
 Element Length,  $L = 3000.00$   
 Primary Member  
 Ribbed Bars  
 Ductile Steel  
 Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Inadequate Lap Length with  $l_o/l_{ou,min} = 0.30$   
 FRP Wrapping Data  
 Type: Carbon  
 Dry properties (design values)  
 Thickness,  $t = 1.00$   
 Tensile Strength,  $f_{fu} = 840.00$   
 Tensile Modulus,  $E_f = 82000.00$   
 Elongation,  $\epsilon_{fu} = 0.009$   
 Number of directions,  $N_{oDir} = 1$   
 Fiber orientations,  $b_i: 0.00^\circ$   
 Number of layers,  $N_L = 1$   
 Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

At local axis: 3  
 EDGE -A-  
 Shear Force,  $V_a = 3.4807814E-030$   
 EDGE -B-  
 Shear Force,  $V_b = -3.4807814E-030$   
 BOTH EDGES  
 Axial Force,  $F = -7389.214$   
 Longitudinal Reinforcement Area Distribution (in 2 divisions)  
 -Tension:  $A_{sl,t} = 0.00$   
 -Compression:  $A_{sl,c} = 3053.628$   
 Longitudinal Reinforcement Area Distribution (in 3 divisions)  
 -Tension:  $A_{sl,ten} = 1017.876$   
 -Compression:  $A_{sl,com} = 1017.876$   
 -Middle:  $A_{sl,mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.23017425$   
 Member Controlled by Flexure ( $V_e/V_r < 1$ )  
 Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 162459.651$   
 with  
 $M_{pr1} = \max(\mu_{u1+}, \mu_{u1-}) = 2.4369E+008$   
 $\mu_{u1+} = 2.4369E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction  
 which is defined for the static loading combination  
 $\mu_{u1-} = 2.4369E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment  
 direction which is defined for the static loading combination  
 $M_{pr2} = \max(\mu_{u2+}, \mu_{u2-}) = 2.4369E+008$

Mu2+ = 2.4369E+008, is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

Mu2- = 2.4369E+008, is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

-----  
Calculation of Mu1+  
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.4369E+008  
-----

= 0.80285146  
' = 0.71403327  
error of function (3.68), Biskinis Phd = 27987.158  
From 5A.2, TDY: fcc = fc\* c = 49.98874  
conf. factor c = 1.51481  
fc = 33.00  
From 10.3.5, ASCE 41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 389.0139  
lb/d = 0.30  
d1 = 44.00  
R = 250.00  
v = 0.00100813  
N = 7389.214  
Ac = 196349.541  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.1620683  
-----

Calculation of ratio lb/d  
-----

Inadequate Lap Length with lb/d = 0.30  
-----  
-----  
-----

Calculation of Mu1-  
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.4369E+008  
-----

= 0.80285146  
' = 0.71403327  
error of function (3.68), Biskinis Phd = 27987.158  
From 5A.2, TDY: fcc = fc\* c = 49.98874  
conf. factor c = 1.51481  
fc = 33.00  
From 10.3.5, ASCE 41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 389.0139  
lb/d = 0.30  
d1 = 44.00  
R = 250.00  
v = 0.00100813  
N = 7389.214  
Ac = 196349.541  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.1620683  
-----

Calculation of ratio lb/d  
-----

Inadequate Lap Length with lb/d = 0.30  
-----  
-----  
-----

Calculation of Mu2+  
-----

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.4369E+008

$$= 0.80285146$$

$$' = 0.71403327$$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TBDY: fcc = fc\* c = 49.98874

conf. factor c = 1.51481

fc = 33.00

From 10.3.5, ASCE 41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 389.0139

lb/d = 0.30

d1 = 44.00

R = 250.00

v = 0.00100813

N = 7389.214

Ac = 196349.541

$$= *Min(1,1.25*(lb/d)^ 2/3) = 0.1620683$$

Calculation of ratio lb/d

Inadequate Lap Length with lb/d = 0.30

Calculation of Mu2-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.4369E+008

$$= 0.80285146$$

$$' = 0.71403327$$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TBDY: fcc = fc\* c = 49.98874

conf. factor c = 1.51481

fc = 33.00

From 10.3.5, ASCE 41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 389.0139

lb/d = 0.30

d1 = 44.00

R = 250.00

v = 0.00100813

N = 7389.214

Ac = 196349.541

$$= *Min(1,1.25*(lb/d)^ 2/3) = 0.1620683$$

Calculation of ratio lb/d

Inadequate Lap Length with lb/d = 0.30

Calculation of Shear Strength Vr = Min(Vr1,Vr2) = 705811.584

Calculation of Shear Strength at edge 1, Vr1 = 705811.584

Vr1 = VCol ((10.3), ASCE 41-17) = knl\*VColO

VColO = 705811.584

knl = 1 (zero step-static loading)

NOTE: In expression (10-3) 'Vs = Av\*fy\*d/s' is replaced by 'Vs+ f\*VF'  
where Vf is the contribution of FRPs (11.3), ACI 440).

$$= 1 \text{ (normal-weight concrete)}$$

Mean concrete strength:  $f_c' = (f_c'_{\text{jacket}} \cdot \text{Area}_{\text{jacket}} + f_c'_{\text{core}} \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.2438899E-011$

$\nu_u = 3.4807814E-030$

$d = 0.8 \cdot D = 400.00$

$N_u = 7389.214$

$A_g = 196349.541$

From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 274157.871$

$V_{s1} = 274157.871$  is calculated for jacket, with:

$A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 123370.055$

$f_y = 555.56$

$s = 100.00$

$V_{s1}$  is multiplied by  $\text{Col1} = 1.00$

$s/d = 0.25$

$V_{s2} = 0.00$  is calculated for core, with:

$A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 78956.835$

$f_y = 555.56$

$s = 250.00$

$V_{s2}$  is multiplied by  $\text{Col2} = 0.00$

$s/d = 1.04167$

$V_f ((11-3)-(11.4), \text{ACI } 440) = 308320.00$

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot \alpha) \sin \alpha$  which is more a generalised expression, where  $\alpha$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\alpha = 45^\circ$  and  $\alpha = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\alpha = b1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, a1)|)$ , with:

total thickness per orientation,  $t_f1 = N_L \cdot t / N_{\text{Dir}} = 1.00$

$d_{fv} = d$  (figure 11.2, ACI 440) = 470.00

$f_{fe} ((11-5), \text{ACI } 440) = 328.00$

$E_f = 82000.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.009$

From (11-11), ACI 440:  $V_s + V_f \leq 479549.663$

$b_w \cdot d = \sqrt{d} \cdot d / 4 = 125663.706$

Calculation of Shear Strength at edge 2,  $V_{r2} = 705811.584$

$V_{r2} = V_{\text{Col}} ((10.3), \text{ASCE } 41-17) = k_{nl} \cdot V_{\text{ColO}}$

$V_{\text{ColO}} = 705811.584$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\mu_u = 1$  (normal-weight concrete)

Mean concrete strength:  $f_c' = (f_c'_{\text{jacket}} \cdot \text{Area}_{\text{jacket}} + f_c'_{\text{core}} \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.2438899E-011$

$\nu_u = 3.4807814E-030$

$d = 0.8 \cdot D = 400.00$

$N_u = 7389.214$

$A_g = 196349.541$

From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 274157.871$

$V_{s1} = 274157.871$  is calculated for jacket, with:

$A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 123370.055$

$f_y = 555.56$

$s = 100.00$

$V_{s1}$  is multiplied by  $\text{Col1} = 1.00$

$s/d = 0.25$

$V_{s2} = 0.00$  is calculated for core, with:

$A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 78956.835$

$f_y = 555.56$



$s = 250.00$   
 $V_{s2}$  is multiplied by  $Col2 = 0.00$   
 $s/d = 1.04167$   
 $V_f((11-3)-(11.4), ACI 440) = 308320.00$   
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a)\sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f( , )$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $a = 45^\circ$  and  $a = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta_1 = \theta_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, a_1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = NL * t / NoDir = 1.00$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 470.00  
 $f_{fe}((11-5), ACI 440) = 328.00$   
 $E_f = 82000.00$   
 $f_{e} = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.009$   
 From (11-11), ACI 440:  $V_s + V_f \leq 479549.663$   
 $b_w * d = \rho * d^2 / 4 = 125663.706$

-----  
 End Of Calculation of Shear Capacity ratio for element: column JCC1 of floor 1  
 At local axis: 3  
 -----

-----  
 Start Of Calculation of Shear Capacity ratio for element: column JCC1 of floor 1  
 At Shear local axis: 2  
 (Bending local axis: 3)  
 Section Type: rcjcs

Constant Properties

-----  
 Knowledge Factor,  $\phi = 1.00$   
 Mean strength values are used for both shear and moment calculations.  
 Consequently:  
 Jacket  
 New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$   
 New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$   
 Concrete Elasticity,  $E_c = 26999.444$   
 Steel Elasticity,  $E_s = 200000.00$   
 Existing Column  
 New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$   
 New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$   
 Concrete Elasticity,  $E_c = 26999.444$   
 Steel Elasticity,  $E_s = 200000.00$   
 #####  
 Note: Especially for the calculation of moment strengths,  
 the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14  
 Jacket  
 New material: Steel Strength,  $f_s = 1.25 * f_{sm} = 694.45$   
 Existing Column  
 New material: Steel Strength,  $f_s = 1.25 * f_{sm} = 694.45$   
 #####  
 External Diameter,  $D = 500.00$   
 Internal Diameter,  $D = 300.00$   
 Cover Thickness,  $c = 25.00$   
 Mean Confinement Factor overall section = 1.51481  
 Element Length,  $L = 3000.00$   
 Primary Member  
 Ribbed Bars  
 Ductile Steel  
 Without Detailing for Earthquake Resistance (including stirrups not closed at  $135^\circ$ )  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Inadequate Lap Length with  $l_o/l_{ou, min} = 0.30$

#### FRP Wrapping Data

Type: Carbon

Dry properties (design values)

Thickness,  $t = 1.00$

Tensile Strength,  $f_{fu} = 840.00$

Tensile Modulus,  $E_f = 82000.00$

Elongation,  $\epsilon_{fu} = 0.009$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

At local axis: 2

EDGE -A-

Shear Force,  $V_a = -2.1312935E-046$

EDGE -B-

Shear Force,  $V_b = 2.1312935E-046$

BOTH EDGES

Axial Force,  $F = -7389.214$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{sl,t} = 0.00$

-Compression:  $A_{sl,c} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{sl,ten} = 1017.876$

-Compression:  $A_{sl,com} = 1017.876$

-Middle:  $A_{sl,mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.23017425$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 162459.651$  with

$M_{pr1} = \max(\mu_{u1+}, \mu_{u1-}) = 2.4369E+008$

$\mu_{u1+} = 2.4369E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$\mu_{u1-} = 2.4369E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \max(\mu_{u2+}, \mu_{u2-}) = 2.4369E+008$

$\mu_{u2+} = 2.4369E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination

$\mu_{u2-} = 2.4369E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

#### Calculation of $\mu_{u1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$

$\mu_u = 2.4369E+008$

$\beta = 0.80285146$

$\beta' = 0.71403327$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 49.98874$

conf. factor  $c = 1.51481$

$f_c = 33.00$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \min(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 389.0139$

$l_b/l_d = 0.30$

$d_1 = 44.00$

$R = 250.00$

$v = 0.00100813$

$$N = 7389.214$$

$$Ac = 196349.541$$

$$= *Min(1, 1.25 * (lb/d)^{2/3}) = 0.1620683$$

Calculation of ratio lb/d

Inadequate Lap Length with lb/d = 0.30

Calculation of Mu1-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.4369E+008

$$= 0.80285146$$

$$' = 0.71403327$$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TBDY: fcc = fc\* c = 49.98874

conf. factor c = 1.51481

fc = 33.00

From 10.3.5, ASCE 41-17, Final value of fy: fy\*Min(1, 1.25\*(lb/d)^{2/3}) = 389.0139

lb/d = 0.30

d1 = 44.00

R = 250.00

v = 0.00100813

N = 7389.214

Ac = 196349.541

$$= *Min(1, 1.25 * (lb/d)^{2/3}) = 0.1620683$$

Calculation of ratio lb/d

Inadequate Lap Length with lb/d = 0.30

Calculation of Mu2+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.4369E+008

$$= 0.80285146$$

$$' = 0.71403327$$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TBDY: fcc = fc\* c = 49.98874

conf. factor c = 1.51481

fc = 33.00

From 10.3.5, ASCE 41-17, Final value of fy: fy\*Min(1, 1.25\*(lb/d)^{2/3}) = 389.0139

lb/d = 0.30

d1 = 44.00

R = 250.00

v = 0.00100813

N = 7389.214

Ac = 196349.541

$$= *Min(1, 1.25 * (lb/d)^{2/3}) = 0.1620683$$

Calculation of ratio lb/d

Inadequate Lap Length with lb/d = 0.30

## Calculation of Mu2-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.4369E+008

$$= 0.80285146$$

$$\rho = 0.71403327$$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 49.98874$

$$\text{conf. factor } c = 1.51481$$

$$f_c = 33.00$$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 389.0139$

$$l_b/d = 0.30$$

$$d_1 = 44.00$$

$$R = 250.00$$

$$v = 0.00100813$$

$$N = 7389.214$$

$$A_c = 196349.541$$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.1620683$$

## Calculation of ratio $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 705811.584$

Calculation of Shear Strength at edge 1,  $V_{r1} = 705811.584$

$V_{r1} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{Col0}$

$$V_{Col0} = 705811.584$$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

Mean concrete strength:  $f'_c = (f'_c \cdot \text{jacket} \cdot \text{Area}_{\text{jacket}} + f'_c \cdot \text{core} \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 33.00$ , but  $f'_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$$M/Vd = 2.00$$

$$Mu = 1.5940012E-011$$

$$Vu = 2.1312935E-046$$

$$d = 0.8 \cdot D = 400.00$$

$$Nu = 7389.214$$

$$Ag = 196349.541$$

From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 274157.871$

$V_{s1} = 274157.871$  is calculated for jacket, with:

$$A_v = \sqrt{2} \cdot A_{\text{stirup}} = 123370.055$$

$$f_y = 555.56$$

$$s = 100.00$$

$V_{s1}$  is multiplied by  $\text{Col1} = 1.00$

$$s/d = 0.25$$

$V_{s2} = 0.00$  is calculated for core, with:

$$A_v = \sqrt{2} \cdot A_{\text{stirup}} = 78956.835$$

$$f_y = 555.56$$

$$s = 250.00$$

$V_{s2}$  is multiplied by  $\text{Col2} = 0.00$

$$s/d = 1.04167$$

$V_f$  ((11-3)-(11.4), ACI 440) = 308320.00

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \cot \alpha) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta_1 = \theta_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, \alpha_1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{\text{Dir}} = 1.00$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 470.00  
 $f_{fe}((11-5), \text{ACI 440}) = 328.00$   
 $E_f = 82000.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.009$   
 From (11-11), ACI 440:  $V_s + V_f \leq 479549.663$   
 $b_w \cdot d = \frac{A_s \cdot d}{4} = 125663.706$

Calculation of Shear Strength at edge 2,  $V_{r2} = 705811.584$   
 $V_{r2} = V_{\text{Col}}((10.3), \text{ASCE 41-17}) = k_n l \cdot V_{\text{Col}0}$   
 $V_{\text{Col}0} = 705811.584$   
 $k_n l = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\lambda = 1$  (normal-weight concrete)  
 Mean concrete strength:  $f'_c = (f'_c \cdot \text{jacket} \cdot \text{Area}_{\text{jacket}} + f'_c \cdot \text{core} \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 33.00$ , but  $f'_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M / V d = 2.00$   
 $\mu_u = 1.5940012 \text{E-}011$   
 $\nu_u = 2.1312935 \text{E-}046$   
 $d = 0.8 \cdot D = 400.00$   
 $N_u = 7389.214$   
 $A_g = 196349.541$   
 From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 274157.871$   
 $V_{s1} = 274157.871$  is calculated for jacket, with:  
 $A_v = \frac{A_s}{2} = 123370.055$   
 $f_y = 555.56$   
 $s = 100.00$   
 $V_{s1}$  is multiplied by  $\text{Col}1 = 1.00$   
 $s/d = 0.25$   
 $V_{s2} = 0.00$  is calculated for core, with:  
 $A_v = \frac{A_s}{2} = 78956.835$   
 $f_y = 555.56$   
 $s = 250.00$   
 $V_{s2}$  is multiplied by  $\text{Col}2 = 0.00$   
 $s/d = 1.04167$   
 $V_f((11-3)-(11.4), \text{ACI 440}) = 308320.00$   
 $f = 0.95$ , for fully-wrapped sections  
 $w_f / s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \cot \alpha) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta_1 = \theta_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, \alpha_1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{\text{Dir}} = 1.00$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 470.00  
 $f_{fe}((11-5), \text{ACI 440}) = 328.00$   
 $E_f = 82000.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.009$   
 From (11-11), ACI 440:  $V_s + V_f \leq 479549.663$   
 $b_w \cdot d = \frac{A_s \cdot d}{4} = 125663.706$

End Of Calculation of Shear Capacity ratio for element: column JCC1 of floor 1  
 At local axis: 2

Start Of Calculation of Chord Rotation Capacity for element: column JCC1 of floor 1

At local axis: 3

Integration Section: (a)

Section Type: rcjcs

#### Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE 41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Jacket

New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

Existing Column

New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

External Diameter,  $D = 500.00$

Internal Diameter,  $D = 300.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Inadequate Lap Length with  $l_b/l_d = 0.30$

FRP Wrapping Data

Type: Carbon

Dry properties (design values)

Thickness,  $t = 1.00$

Tensile Strength,  $f_{fu} = 840.00$

Tensile Modulus,  $E_f = 82000.00$

Elongation,  $ε_{fu} = 0.009$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i: 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

Bending Moment,  $M = -1.4341E+007$

Shear Force,  $V_2 = -4779.203$

Shear Force,  $V_3 = -2.1566928E-013$

Axial Force,  $F = -7387.302$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{slt} = 0.00$

-Compression:  $A_{slc} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{sl,ten} = 1017.876$

-Compression:  $A_{sl,com} = 1017.876$

-Middle:  $A_{sl,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $DbL = 18.00$

New component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_{,R} = 1.0^*$   $u = 0.00906346$

$$u = y + p = 0.00906346$$

- Calculation of  $y$  -

$$y = (M_y * L_s / 3) / E_{eff} = 0.00906346 \text{ ((4.29), Biskinis Phd)}$$

$$M_y = 2.2517E+008$$

$$L_s = M/V \text{ (with } L_s > 0.1 * L \text{ and } L_s < 2 * L) = 3000.736$$

$$\text{From table 10.5, ASCE 41-17: } E_{eff} = \text{factor} * E_c * I_g = 2.4850E+013$$

$$\text{factor} = 0.30$$

$$A_g = 196349.541$$

$$\text{Mean concrete strength: } f'_c = (f'_{c\_jacket} * A_{jacket} + f'_{c\_core} * A_{core}) / A_{section} = 33.00$$

$$N = 7387.302$$

$$E_c * I_g = E_{c\_jacket} * I_{g\_jacket} + E_{c\_core} * I_{g\_core} = 8.2833E+013$$

Calculation of Yielding Moment  $M_y$

Calculation of  $y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$$M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 2.2517E+008$$

$$y = 6.4424840E-006$$

$$M_{y\_ten} (8c) = 2.2517E+008$$

$$y_{ten} (7c) = 61.8137$$

$$\text{error of function (7c)} = 0.00026387$$

$$M_{y\_com} (8d) = 8.1922E+008$$

$$y_{com} (7d) = 63.67468$$

$$\text{error of function (7d)} = -0.0087488$$

$$\text{with ((10.1), ASCE 41-17) } e_y = \text{Min}(e_y, 1.25 * e_y * (I_b / I_d)^{2/3}) = 0.0027778$$

$$e_{co} = 0.002$$

$$a_{pl} = 0.45 \text{ ((9c) in Biskinis and Fardis for FRP Wrap)}$$

$$d_1 = 44.00$$

$$R = 250.00$$

$$v = 0.00101374$$

$$N = 7387.302$$

$$A_c = 196349.541$$

$$\text{((10.1), ASCE 41-17) } = \text{Min}( , 1.25 * (I_b / I_d)^{2/3}) = 0.23280363$$

$$\text{with } f'_c \text{ ((12.3), ACI 440) } = 37.11312$$

$$f'_c = 33.00$$

$$f_l = 1.312$$

$$k = 1$$

$$\text{Effective FRP thickness, } t_f = N L * t * \cos(b_1) = 1.00$$

$$e_{fe} \text{ ((12.5) and (12.7)) } = 0.004$$

$$f_u = 0.009$$

$$E_f = 82000.00$$

Calculation of ratio  $I_b / I_d$

Inadequate Lap Length with  $I_b / I_d = 0.30$

- Calculation of  $p$  -

From table 10-9:  $p = 0.00$

with:

- Columns controlled by inadequate development or splicing along the clear height because  $I_b / I_d < 1$

$$\text{shear control ratio } V_y E / V_{col} E = 0.23017425$$

$$d = d_{external} = 0.00$$

$$s = s_{external} = 0.00$$

$$t = s_1 + s_2 + 2 * t_f / b_w * (f_{fe} / f_s) = 0.00795744$$

$$\text{jacket: } s_1 = A_{v1} * (D_{c1} / 2) / (s_1 * A_g) = 0.0027646$$

$$A_{v1} = 78.53982, \text{ is the area of stirrup}$$

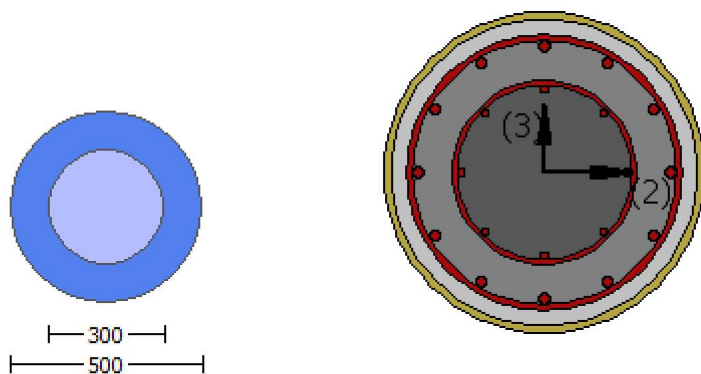
$$D_{c1} = D_{ext} - 2 * \text{cover} - \text{External Hoop Diameter} = 440.00, \text{ is the total Length of all stirrups parallel to loading (shear) direction}$$

$s1 = 100.00$   
 core:  $s2 = Av2 * (\pi Dc2/2) / (s2 * Ag) = 0.00046968$   
 $Av2 = 50.26548$ , is the area of stirrup  
 $Dc2 = Dint - \text{Internal Hoop Diameter} = 292.00$ , is the total Length of all stirrups parallel to loading (shear) direction  
 $s2 = 250.00$   
 The term  $2*tf/bw*(ffe/fs)$  is implemented to account for FRP contribution where  $f = 2*tf/bw$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $ffe/fs$  normalises  $f$  to steel strength  
 All these variables have already been given in Shear control ratio calculation.  
 For the normalisation  $fs$  of jacket is used.  
 $NUD = 7387.302$   
 $Ag = 196349.541$   
 $f_{cE} = (f_{c\_jacket} * Area\_jacket + f_{c\_core} * Area\_core) / section\_area = 33.00$   
 $f_{yIE} = (f_{y\_ext\_Long\_Reinf} * Area\_ext\_Long\_Reinf + f_{y\_int\_Long\_Reinf} * Area\_int\_Long\_Reinf) / Area\_Tot\_Long\_Rein = 2.1219958E-314$   
 $f_{yTE} = (f_{y\_ext\_Trans\_Reinf} * Area\_ext\_Trans\_Reinf + f_{y\_int\_Trans\_Reinf} * Area\_int\_Trans\_Reinf) / Area\_Tot\_Trans\_Rein = 555.56$   
 $pI = Area\_Tot\_Long\_Rein / (Ag) = 0.015552$   
 $f_{cE} = 33.00$

-----  
 End Of Calculation of Chord Rotation Capacity for element: column JCC1 of floor 1  
 At local axis: 3  
 Integration Section: (a)

## Calculation No. 5

column C1, Floor 1  
 Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)  
 Analysis: Uniform +X  
 Check: Shear capacity  $VRd$   
 Edge: End  
 Local Axis: (2)



Start Of Calculation of Shear Capacity for element: column JCC1 of floor 1  
 At local axis: 2  
 Integration Section: (b)



Section Type: rcjcs

## Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE 41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Jacket

New material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 25.00$

New material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 500.00$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

Existing Column

New material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 25.00$

New material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 500.00$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of  $\gamma$  for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE 41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE 41-17).

Jacket

New material: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material: Steel Strength,  $f_s = f_{sm} = 555.56$

Existing Column

New material: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material: Steel Strength,  $f_s = f_{sm} = 555.56$

#####

External Diameter,  $D = 500.00$

Internal Diameter,  $D = 300.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Inadequate Lap Length with  $l_o/l_{ou,min} = l_b/l_d = 0.30$

FRP Wrapping Data

Type: Carbon

Dry properties (design values)

Thickness,  $t = 1.00$

Tensile Strength,  $f_{fu} = 840.00$

Tensile Modulus,  $E_f = 82000.00$

Elongation,  $\epsilon_{fu} = 0.009$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i: 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

## Stepwise Properties

EDGE -A-

Bending Moment,  $M_a = -1.4341E+007$

Shear Force,  $V_a = -4779.203$

EDGE -B-

Bending Moment,  $M_b = 0.07981876$

Shear Force,  $V_b = 4779.203$

BOTH EDGES

Axial Force,  $F = -7387.302$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 0.00$

-Compression:  $As_c = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{l,ten} = 1017.876$

-Compression:  $A_{sl,com} = 1017.876$

-Middle:  $A_{sl,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $Db_{L,ten} = 18.00$

New component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = 1.0 \cdot V_n = 614519.957$

$V_n$  ((10.3), ASCE 41-17) =  $k_n \cdot V_{Col0} = 614519.957$

$V_{Col} = 614519.957$

$k_n = 1.00$

$displacement\_ductility\_demand = 0.09552499$

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ '

where  $V_f$  is the contribution of FRPs ((11.3), ACI 440).

= 1 (normal-weight concrete)

Mean concrete strength:  $f'_c = (f'_c\_jacket \cdot Area\_jacket + f'_c\_core \cdot Area\_core) / Area\_section = 25.00$ , but  $f'_c^{0.5} \leq 8.3$  MPa ((22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 0.07981876$

$V_u = 4779.203$

$d = 0.8 \cdot D = 400.00$

$N_u = 7387.302$

$A_g = 196349.541$

From ((11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 246740.11$

$V_{s1} = 246740.11$  is calculated for jacket, with:

$A_v = \sqrt{2} \cdot A_{stirrup} = 123370.055$

$f_y = 500.00$

$s = 100.00$

$V_{s1}$  is multiplied by  $Col1 = 1.00$

$s/d = 0.25$

$V_{s2} = 0.00$  is calculated for core, with:

$A_v = \sqrt{2} \cdot A_{stirrup} = 78956.835$

$f_y = 500.00$

$s = 250.00$

$V_{s2}$  is multiplied by  $Col2 = 0.00$

$s/d = 1.04167$

$V_f$  ((11-3)-(11.4), ACI 440) =  $308320.00$

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In ((11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression, where  $a$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(, )$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $a = 45^\circ$  and  $a = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $a_1 = b_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, a_1)|, |V_f(-45, a_1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{oDir} = 1.00$

$d_{fv} = d$  (figure 11.2, ACI 440) =  $470.00$

$f_{fe}$  ((11-5), ACI 440) =  $328.00$

$E_f = 82000.00$

$f_e = 0.004$ , from ((11.6a), ACI 440

with  $f_u = 0.009$

From ((11-11), ACI 440:  $V_s + V_f \leq 417394.406$

$b_w \cdot d = \sqrt{4} \cdot d^2 / 4 = 125663.706$

$displacement\_ductility\_demand$  is calculated as  $\delta / y$

- Calculation of  $\delta / y$  for END B -

for rotation axis 3 and integ. section (b)

From analysis, chord rotation  $\theta = 8.6557465E-005$

$y = (M_y \cdot L_s / 3) / E_{eff} = 0.00090612$  ((4.29), Biskinis Phd))

$M_y = 2.2517E+008$

$L_s = M/V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) =  $300.00$

From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 2.4850E+013$   
 $factor = 0.30$   
 $A_g = 196349.541$   
Mean concrete strength:  $f_c' = (f_c'_{jacket} * Area_{jacket} + f_c'_{core} * Area_{core}) / Area_{section} = 33.00$   
 $N = 7387.302$   
 $E_c * I_g = E_{c\_jacket} * I_{g\_jacket} + E_{c\_core} * I_{g\_core} = 8.2833E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\gamma$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \min(M_{y\_ten}, M_{y\_com}) = 2.2517E+008$   
 $\gamma = 6.4424840E-006$   
 $M_{y\_ten} (8c) = 2.2517E+008$   
 $\gamma_{ten} (7c) = 61.8137$   
error of function (7c) = 0.00026387  
 $M_{y\_com} (8d) = 8.1922E+008$   
 $\gamma_{com} (7d) = 63.67468$   
error of function (7d) = -0.0087488  
with ((10.1), ASCE 41-17)  $\gamma_y = \min(\gamma_y, 1.25 * \gamma_y * (I_b / I_d)^{2/3}) = 0.0027778$   
 $e_{co} = 0.002$   
 $a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 250.00$   
 $v = 0.00101374$   
 $N = 7387.302$   
 $A_c = 196349.541$   
((10.1), ASCE 41-17)  $\gamma = \min(\gamma, 1.25 * \gamma * (I_b / I_d)^{2/3}) = 0.23280363$   
with  $f_c' ((12.3), ACI 440) = 37.11312$   
 $f_c = 33.00$   
 $f_l = 1.312$   
 $k = 1$   
Effective FRP thickness,  $t_f = NL * t * \cos(b_1) = 1.00$   
 $e_{fe} ((12.5) \text{ and } (12.7)) = 0.004$   
 $f_u = 0.009$   
 $E_f = 82000.00$

Calculation of ratio  $I_b / I_d$

Inadequate Lap Length with  $I_b / I_d = 0.30$

End Of Calculation of Shear Capacity for element: column JCC1 of floor 1

At local axis: 2

Integration Section: (b)

## Calculation No. 6

column C1, Floor 1

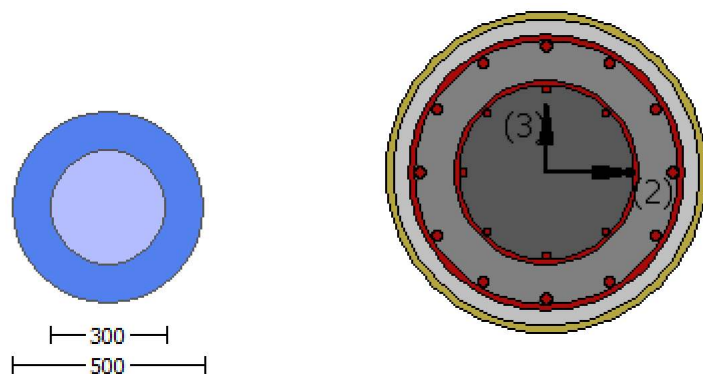
Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity ( $\phi$ )

Edge: End

Local Axis: (2)



Start Of Calculation of Shear Capacity ratio for element: column JCC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rcjcs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Mean strength values are used for both shear and moment calculations.

Consequently:

Jacket

New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

Existing Column

New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Jacket

New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 694.45$

Existing Column

New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 694.45$

#####

External Diameter,  $D = 500.00$

Internal Diameter,  $D = 300.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.51481

Element Length,  $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Inadequate Lap Length with  $l_o/l_{ou,min} = 0.30$

#### FRP Wrapping Data

Type: Carbon

Dry properties (design values)

Thickness,  $t = 1.00$

Tensile Strength,  $f_{fu} = 840.00$

Tensile Modulus,  $E_f = 82000.00$

Elongation,  $\epsilon_{fu} = 0.009$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

At local axis: 3

EDGE -A-

Shear Force,  $V_a = 3.4807814E-030$

EDGE -B-

Shear Force,  $V_b = -3.4807814E-030$

BOTH EDGES

Axial Force,  $F = -7389.214$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{sl,t} = 0.00$

-Compression:  $A_{sl,c} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{sl,ten} = 1017.876$

-Compression:  $A_{sl,com} = 1017.876$

-Middle:  $A_{sl,mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.23017425$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 162459.651$

with

$M_{pr1} = \max(\mu_{u1+}, \mu_{u1-}) = 2.4369E+008$

$\mu_{u1+} = 2.4369E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$\mu_{u1-} = 2.4369E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \max(\mu_{u2+}, \mu_{u2-}) = 2.4369E+008$

$\mu_{u2+} = 2.4369E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination

$\mu_{u2-} = 2.4369E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

#### Calculation of $\mu_{u1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$

$\mu_u = 2.4369E+008$

$\beta_1 = 0.80285146$

$\beta_2 = 0.71403327$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 49.98874$

conf. factor  $c = 1.51481$

$f_c = 33.00$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \min(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 389.0139$

$l_b/l_d = 0.30$

$d_1 = 44.00$

$R = 250.00$

$v = 0.00100813$

$$N = 7389.214$$

$$Ac = 196349.541$$

$$= *Min(1, 1.25*(lb/d)^{2/3}) = 0.1620683$$

Calculation of ratio lb/d

Inadequate Lap Length with lb/d = 0.30

Calculation of Mu1-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.4369E+008

$$= 0.80285146$$

$$' = 0.71403327$$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TBDY: fcc = fc\* c = 49.98874

conf. factor c = 1.51481

fc = 33.00

From 10.3.5, ASCE 41-17, Final value of fy: fy\*Min(1, 1.25\*(lb/d)^{2/3}) = 389.0139

lb/d = 0.30

d1 = 44.00

R = 250.00

v = 0.00100813

N = 7389.214

Ac = 196349.541

$$= *Min(1, 1.25*(lb/d)^{2/3}) = 0.1620683$$

Calculation of ratio lb/d

Inadequate Lap Length with lb/d = 0.30

Calculation of Mu2+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.4369E+008

$$= 0.80285146$$

$$' = 0.71403327$$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TBDY: fcc = fc\* c = 49.98874

conf. factor c = 1.51481

fc = 33.00

From 10.3.5, ASCE 41-17, Final value of fy: fy\*Min(1, 1.25\*(lb/d)^{2/3}) = 389.0139

lb/d = 0.30

d1 = 44.00

R = 250.00

v = 0.00100813

N = 7389.214

Ac = 196349.541

$$= *Min(1, 1.25*(lb/d)^{2/3}) = 0.1620683$$

Calculation of ratio lb/d

Inadequate Lap Length with lb/d = 0.30

## Calculation of Mu2-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
 $\text{Mu} = 2.4369\text{E}+008$

$$= 0.80285146$$

$$\rho = 0.71403327$$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 49.98874$

$$\text{conf. factor } c = 1.51481$$

$$f_c = 33.00$$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 389.0139$

$$l_b/d = 0.30$$

$$d_1 = 44.00$$

$$R = 250.00$$

$$v = 0.00100813$$

$$N = 7389.214$$

$$A_c = 196349.541$$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.1620683$$

## Calculation of ratio $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 705811.584$

Calculation of Shear Strength at edge 1,  $V_{r1} = 705811.584$

$V_{r1} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{Col0}$

$$V_{Col0} = 705811.584$$

$$k_{nl} = 1 \text{ (zero step-static loading)}$$

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$$= 1 \text{ (normal-weight concrete)}$$

Mean concrete strength:  $f_c' = (f_c'_{\text{jacket}} \cdot \text{Area}_{\text{jacket}} + f_c'_{\text{core}} \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$$M/Vd = 2.00$$

$$\text{Mu} = 1.2438899\text{E}-011$$

$$\text{Vu} = 3.4807814\text{E}-030$$

$$d = 0.8 \cdot D = 400.00$$

$$N_u = 7389.214$$

$$A_g = 196349.541$$

From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 274157.871$

$V_{s1} = 274157.871$  is calculated for jacket, with:

$$A_v = \rho_s \cdot A_{\text{stirup}} = 123370.055$$

$$f_y = 555.56$$

$$s = 100.00$$

$V_{s1}$  is multiplied by  $\text{Col1} = 1.00$

$$s/d = 0.25$$

$V_{s2} = 0.00$  is calculated for core, with:

$$A_v = \rho_s \cdot A_{\text{stirup}} = 78956.835$$

$$f_y = 555.56$$

$$s = 250.00$$

$V_{s2}$  is multiplied by  $\text{Col2} = 0.00$

$$s/d = 1.04167$$

$V_f$  ((11-3)-(11.4), ACI 440) = 308320.00

$$f = 0.95, \text{ for fully-wrapped sections}$$

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \cot \alpha) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta_1 = \theta_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, \alpha_1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = N_L \cdot t / \text{NoDir} = 1.00$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 470.00  
 $f_{fe}((11-5), \text{ACI 440}) = 328.00$   
 $E_f = 82000.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.009$   
 From (11-11), ACI 440:  $V_s + V_f \leq 479549.663$   
 $b_w \cdot d = \frac{A_s \cdot d}{4} = 125663.706$

Calculation of Shear Strength at edge 2,  $V_{r2} = 705811.584$   
 $V_{r2} = V_{Col}((10.3), \text{ASCE 41-17}) = k_n l \cdot V_{Col0}$   
 $V_{Col0} = 705811.584$   
 $k_n l = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\lambda = 1$  (normal-weight concrete)  
 Mean concrete strength:  $f'_c = (f'_c \cdot \text{jacket} \cdot \text{Area}_{\text{jacket}} + f'_c \cdot \text{core} \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 33.00$ , but  $f'_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M / Vd = 2.00$   
 $\mu_u = 1.2438899 \text{E-}011$   
 $\nu_u = 3.4807814 \text{E-}030$   
 $d = 0.8 \cdot D = 400.00$   
 $N_u = 7389.214$   
 $A_g = 196349.541$   
 From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 274157.871$   
 $V_{s1} = 274157.871$  is calculated for jacket, with:  
 $A_v = \frac{A_s}{2} = 123370.055$   
 $f_y = 555.56$   
 $s = 100.00$   
 $V_{s1}$  is multiplied by  $\text{Col1} = 1.00$   
 $s/d = 0.25$   
 $V_{s2} = 0.00$  is calculated for core, with:  
 $A_v = \frac{A_s}{2} = 78956.835$   
 $f_y = 555.56$   
 $s = 250.00$   
 $V_{s2}$  is multiplied by  $\text{Col2} = 0.00$   
 $s/d = 1.04167$   
 $V_f((11-3)-(11.4), \text{ACI 440}) = 308320.00$   
 $f = 0.95$ , for fully-wrapped sections  
 $w_f / s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \cot \alpha) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta_1 = \theta_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, \alpha_1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = N_L \cdot t / \text{NoDir} = 1.00$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 470.00  
 $f_{fe}((11-5), \text{ACI 440}) = 328.00$   
 $E_f = 82000.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.009$   
 From (11-11), ACI 440:  $V_s + V_f \leq 479549.663$   
 $b_w \cdot d = \frac{A_s \cdot d}{4} = 125663.706$

End Of Calculation of Shear Capacity ratio for element: column JCC1 of floor 1  
 At local axis: 3



Start Of Calculation of Shear Capacity ratio for element: column JCC1 of floor 1

At Shear local axis: 2

(Bending local axis: 3)

Section Type: rcjcs

#### Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Mean strength values are used for both shear and moment calculations.

Consequently:

Jacket

New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

Existing Column

New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Jacket

New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 694.45$

Existing Column

New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 694.45$

#####

External Diameter,  $D = 500.00$

Internal Diameter,  $D = 300.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.51481

Element Length,  $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Inadequate Lap Length with  $l_o/l_{ou,min} = 0.30$

FRP Wrapping Data

Type: Carbon

Dry properties (design values)

Thickness,  $t = 1.00$

Tensile Strength,  $f_{fu} = 840.00$

Tensile Modulus,  $E_f = 82000.00$

Elongation,  $\epsilon_{fu} = 0.009$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i: 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

At local axis: 2

EDGE -A-

Shear Force,  $V_a = -2.1312935E-046$

EDGE -B-

Shear Force,  $V_b = 2.1312935E-046$

BOTH EDGES

Axial Force,  $F = -7389.214$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{st} = 0.00$

-Compression:  $As_c = 3053.628$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $As_{ten} = 1017.876$   
-Compression:  $As_{com} = 1017.876$   
-Middle:  $As_{mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.23017425$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 162459.651$   
with

$M_{pr1} = \text{Max}(\mu_{u1+}, \mu_{u1-}) = 2.4369E+008$

$\mu_{u1+} = 2.4369E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$\mu_{u1-} = 2.4369E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(\mu_{u2+}, \mu_{u2-}) = 2.4369E+008$

$\mu_{u2+} = 2.4369E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination

$\mu_{u2-} = 2.4369E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

Calculation of  $\mu_{u1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$   
 $\mu_u = 2.4369E+008$

$\phi = 0.80285146$

$\lambda = 0.71403327$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TB DY:  $f_{cc} = f_c^* \quad c = 49.98874$

conf. factor  $c = 1.51481$

$f_c = 33.00$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 389.0139$

$l_b/d = 0.30$

$d_1 = 44.00$

$R = 250.00$

$v = 0.00100813$

$N = 7389.214$

$A_c = 196349.541$

$= \phi \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.1620683$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of  $\mu_{u1-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$   
 $\mu_u = 2.4369E+008$

$\phi = 0.80285146$

$\lambda = 0.71403327$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TB DY:  $f_{cc} = f_c^* \quad c = 49.98874$

conf. factor  $c = 1.51481$

$f_c = 33.00$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 389.0139$

$l_b/d = 0.30$

$d_1 = 44.00$

$R = 250.00$

$v = 0.00100813$

$N = 7389.214$

$A_c = 196349.541$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.1620683$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of  $\mu_{2+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$\mu = 2.4369\text{E}+008$

$= 0.80285146$

$' = 0.71403327$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 49.98874$

conf. factor  $c = 1.51481$

$f_c = 33.00$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 389.0139$

$l_b/d = 0.30$

$d_1 = 44.00$

$R = 250.00$

$v = 0.00100813$

$N = 7389.214$

$A_c = 196349.541$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.1620683$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of  $\mu_{2-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$\mu = 2.4369\text{E}+008$

$= 0.80285146$

$' = 0.71403327$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 49.98874$

conf. factor  $c = 1.51481$

$f_c = 33.00$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 389.0139$

$l_b/d = 0.30$

$d_1 = 44.00$

$R = 250.00$

$v = 0.00100813$

$N = 7389.214$

$A_c = 196349.541$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.1620683$

Calculation of ratio  $l_b/l_d$

Inadequate Lap Length with  $l_b/l_d = 0.30$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 705811.584$

Calculation of Shear Strength at edge 1,  $V_{r1} = 705811.584$

$V_{r1} = V_{Col} \text{ ((10.3), ASCE 41-17)} = k_n l^* V_{ColO}$

$V_{ColO} = 705811.584$

$k_n l = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f^* V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

Mean concrete strength:  $f_c' = (f_c'_{\text{jacket}} \cdot \text{Area}_{\text{jacket}} + f_c'_{\text{core}} \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/V_d = 2.00$

$\mu_u = 1.5940012E-011$

$\nu_u = 2.1312935E-046$

$d = 0.8 \cdot D = 400.00$

$N_u = 7389.214$

$A_g = 196349.541$

From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 274157.871$

$V_{s1} = 274157.871$  is calculated for jacket, with:

$A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 123370.055$

$f_y = 555.56$

$s = 100.00$

$V_{s1}$  is multiplied by  $\text{Col1} = 1.00$

$s/d = 0.25$

$V_{s2} = 0.00$  is calculated for core, with:

$A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 78956.835$

$f_y = 555.56$

$s = 250.00$

$V_{s2}$  is multiplied by  $\text{Col2} = 0.00$

$s/d = 1.04167$

$V_f \text{ ((11-3)-(11.4), ACI 440)} = 308320.00$

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \alpha + \cos \alpha$  is replaced with  $(\cot \alpha + \cot \alpha) \sin \alpha$  which is more a generalised expression,

where  $\alpha$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\alpha = 45^\circ$  and  $\alpha = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\alpha_1 = b_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \alpha_1)|, |V_f(-45, \alpha_1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{\text{Dir}} = 1.00$

$d_{fv} = d$  (figure 11.2, ACI 440) = 470.00

$f_{fe} \text{ ((11-5), ACI 440)} = 328.00$

$E_f = 82000.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.009$

From (11-11), ACI 440:  $V_s + V_f \leq 479549.663$

$b_w \cdot d = \sqrt{2} \cdot d \cdot d / 4 = 125663.706$

Calculation of Shear Strength at edge 2,  $V_{r2} = 705811.584$

$V_{r2} = V_{Col} \text{ ((10.3), ASCE 41-17)} = k_n l^* V_{ColO}$

$V_{ColO} = 705811.584$

$k_n l = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f^* V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\lambda = 1$  (normal-weight concrete)  
Mean concrete strength:  $f_c' = (f_c'_{\text{jacket}} \cdot \text{Area}_{\text{jacket}} + f_c'_{\text{core}} \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $\mu_u = 1.5940012\text{E-}011$   
 $\nu_u = 2.1312935\text{E-}046$   
 $d = 0.8 \cdot D = 400.00$   
 $N_u = 7389.214$   
 $A_g = 196349.541$   
From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 274157.871$   
 $V_{s1} = 274157.871$  is calculated for jacket, with:  
 $A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 123370.055$   
 $f_y = 555.56$   
 $s = 100.00$   
 $V_{s1}$  is multiplied by  $\text{Col1} = 1.00$   
 $s/d = 0.25$   
 $V_{s2} = 0.00$  is calculated for core, with:  
 $A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 78956.835$   
 $f_y = 555.56$   
 $s = 250.00$   
 $V_{s2}$  is multiplied by  $\text{Col2} = 0.00$   
 $s/d = 1.04167$   
 $V_f((11-3)-(11.4), \text{ACI } 440) = 308320.00$   
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot_a) \sin \alpha$  which is more a generalised expression, where  $\alpha$  is the angle of the crack direction (see KANEPE).  
This later relation, considered as a function  $V_f(\alpha, \theta)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
orientation 1:  $\theta = \theta_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, \theta_1)|)$ , with:  
total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{\text{Dir}} = 1.00$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 470.00  
 $f_{fe}((11-5), \text{ACI } 440) = 328.00$   
 $E_f = 82000.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
with  $f_u = 0.009$   
From (11-11), ACI 440:  $V_s + V_f \leq 479549.663$   
 $b_w \cdot d = \lambda \cdot d^2 / 4 = 125663.706$

End Of Calculation of Shear Capacity ratio for element: column JCC1 of floor 1  
At local axis: 2

Start Of Calculation of Chord Rotation Capacity for element: column JCC1 of floor 1  
At local axis: 2  
Integration Section: (b)  
Section Type: rcjcs

Constant Properties

Knowledge Factor,  $\lambda = 1.00$   
Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE 41-17.  
Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17  
Consequently:  
Jacket  
New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$   
New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$   
Concrete Elasticity,  $E_c = 26999.444$   
Steel Elasticity,  $E_s = 200000.00$   
Existing Column  
New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$   
New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$   
Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$   
 External Diameter,  $D = 500.00$   
 Internal Diameter,  $D = 300.00$   
 Cover Thickness,  $c = 25.00$   
 Element Length,  $L = 3000.00$   
 Primary Member  
 Ribbed Bars  
 Ductile Steel  
 Without Detailing for Earthquake Resistance (including stirrups not closed at  $135^\circ$ )  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Inadequate Lap Length with  $l_b/l_d = 0.30$   
 FRP Wrapping Data  
 Type: Carbon  
 Dry properties (design values)  
 Thickness,  $t = 1.00$   
 Tensile Strength,  $f_{fu} = 840.00$   
 Tensile Modulus,  $E_f = 82000.00$   
 Elongation,  $ef_u = 0.009$   
 Number of directions,  $NoDir = 1$   
 Fiber orientations,  $bi: 0.00^\circ$   
 Number of layers,  $NL = 1$   
 Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

Bending Moment,  $M = -8.7529383E-011$   
 Shear Force,  $V_2 = 4779.203$   
 Shear Force,  $V_3 = 2.1566928E-013$   
 Axial Force,  $F = -7387.302$   
 Longitudinal Reinforcement Area Distribution (in 2 divisions)  
     -Tension:  $As_t = 0.00$   
     -Compression:  $As_c = 3053.628$   
 Longitudinal Reinforcement Area Distribution (in 3 divisions)  
     -Tension:  $As_{t,ten} = 1017.876$   
     -Compression:  $As_{c,com} = 1017.876$   
     -Middle:  $As_{c,mid} = 1017.876$   
 Mean Diameter of Tension Reinforcement,  $Db_L = 18.00$

New component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_R = 1.0^*$   $u = 0.00453062$   
 $u = y + p = 0.00453062$

- Calculation of  $y$  -

$y = (M_y * L_s / 3) / E_{eff} = 0.00453062$  ((4.29), Biskinis Phd))  
 $M_y = 2.2517E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) =  $1500.00$   
 From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 2.4850E+013$   
 $factor = 0.30$   
 $A_g = 196349.541$   
 Mean concrete strength:  $f'_c = (f'_{c,jacket} * Area_{jacket} + f'_{c,core} * Area_{core}) / Area_{section} = 33.00$   
 $N = 7387.302$   
 $E_c * I_g = E_{c,jacket} * I_{g,jacket} + E_{c,core} * I_{g,core} = 8.2833E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \min(M_{y,ten}, M_{y,com}) = 2.2517E+008$   
 $y = 6.4424840E-006$   
 $M_{y,ten} (8c) = 2.2517E+008$

$\rho_{ten}(7c) = 61.8137$   
error of function (7c) = 0.00026387  
My\_com (8d) = 8.1922E+008  
 $\rho_{com}(7d) = 63.67468$   
error of function (7d) = -0.0087488  
with ((10.1), ASCE 41-17)  $\rho_y = \text{Min}(\rho_y, 1.25 \cdot \rho_y \cdot (l_b/l_d)^{2/3}) = 0.0027778$   
 $\rho_{co} = 0.002$   
 $\rho_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 250.00$   
 $v = 0.00101374$   
 $N = 7387.302$   
 $A_c = 196349.541$   
((10.1), ASCE 41-17)  $\rho = \text{Min}(\rho, 1.25 \cdot \rho \cdot (l_b/l_d)^{2/3}) = 0.23280363$   
with  $f_c^*$  ((12.3), ACI 440) = 37.11312  
 $f_c = 33.00$   
 $f_l = 1.312$   
 $k = 1$   
Effective FRP thickness,  $t_f = N \cdot t \cdot \cos(b_1) = 1.00$   
 $\rho_{fe}$  ((12.5) and (12.7)) = 0.004  
 $f_u = 0.009$   
 $E_f = 82000.00$

Calculation of ratio  $l_b/l_d$

Inadequate Lap Length with  $l_b/l_d = 0.30$

- Calculation of  $\rho_p$  -

From table 10-9:  $\rho_p = 0.00$

with:

- Columns controlled by inadequate development or splicing along the clear height because  $l_b/l_d < 1$   
shear control ratio  $V_y E / V_{co} I_{OE} = 0.23017425$   
 $d = d_{external} = 0.00$   
 $s = s_{external} = 0.00$   
 $t = s_1 + s_2 + 2 \cdot t_f / b_w \cdot (f_{fe} / f_s) = 0.00795744$   
jacket:  $s_1 = A_{v1} \cdot (D_{c1} / 2) / (s_1 \cdot A_g) = 0.0027646$   
 $A_{v1} = 78.53982$ , is the area of stirrup  
 $D_{c1} = D_{ext} - 2 \cdot \text{cover} - \text{External Hoop Diameter} = 440.00$ , is the total Length of all stirrups parallel to loading (shear) direction  
 $s_1 = 100.00$   
core:  $s_2 = A_{v2} \cdot (D_{c2} / 2) / (s_2 \cdot A_g) = 0.00046968$   
 $A_{v2} = 50.26548$ , is the area of stirrup  
 $D_{c2} = D_{int} - \text{Internal Hoop Diameter} = 292.00$ , is the total Length of all stirrups parallel to loading (shear) direction  
 $s_2 = 250.00$

The term  $2 \cdot t_f / b_w \cdot (f_{fe} / f_s)$  is implemented to account for FRP contribution  
where  $f = 2 \cdot t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength  
All these variables have already been given in Shear control ratio calculation.  
For the normalisation  $f_s$  of jacket is used.

$N_{UD} = 7387.302$   
 $A_g = 196349.541$   
 $f_{cE} = (f_{c,jacket} \cdot \text{Area}_{jacket} + f_{c,core} \cdot \text{Area}_{core}) / \text{section\_area} = 33.00$   
 $f_{yE} = (f_{y,ext\_Long\_Reinf} \cdot \text{Area}_{ext\_Long\_Reinf} + f_{y,int\_Long\_Reinf} \cdot \text{Area}_{int\_Long\_Reinf}) / \text{Area}_{Tot\_Long\_Rein} = 21219958E-314$   
 $f_{yE} = (f_{y,ext\_Trans\_Reinf} \cdot \text{Area}_{ext\_Trans\_Reinf} + f_{y,int\_Trans\_Reinf} \cdot \text{Area}_{int\_Trans\_Reinf}) / \text{Area}_{Tot\_Trans\_Rein} = 555.56$   
 $\rho_l = \text{Area}_{Tot\_Long\_Rein} / (A_g) = 0.015552$   
 $f_{cE} = 33.00$

End Of Calculation of Chord Rotation Capacity for element: column JCC1 of floor 1

At local axis: 2

Integration Section: (b)

## Calculation No. 7

column C1, Floor 1

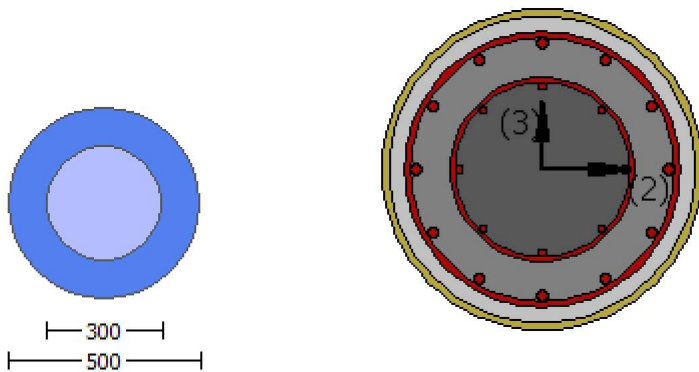
Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Shear capacity  $V_{Rd}$

Edge: End

Local Axis: (3)



Start Of Calculation of Shear Capacity for element: column JCC1 of floor 1

At local axis: 3

Integration Section: (b)

Section Type: rcjcs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE 41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Jacket

New material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 25.00$

New material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 500.00$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

Existing Column

New material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 25.00$

New material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 500.00$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of  $\gamma$  for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE 41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE 41-17).

Jacket

New material: Concrete Strength,  $f_c = f_{cm} = 33.00$



```

New material: Steel Strength,  $f_s = f_{sm} = 555.56$ 
Existing Column
New material: Concrete Strength,  $f_c = f_{cm} = 33.00$ 
New material: Steel Strength,  $f_s = f_{sm} = 555.56$ 
#####
External Diameter,  $D = 500.00$ 
Internal Diameter,  $D = 300.00$ 
Cover Thickness,  $c = 25.00$ 
Element Length,  $L = 3000.00$ 
Primary Member
Ribbed Bars
Ductile Steel
Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)
Longitudinal Bars With Ends Lapped Starting at the End Sections
Inadequate Lap Length with  $l_o/l_{ou,min} = l_b/l_d = 0.30$ 
FRP Wrapping Data
Type: Carbon
Dry properties (design values)
Thickness,  $t = 1.00$ 
Tensile Strength,  $f_{fu} = 840.00$ 
Tensile Modulus,  $E_f = 82000.00$ 
Elongation,  $ε_{fu} = 0.009$ 
Number of directions,  $N_{oDir} = 1$ 
Fiber orientations,  $bi: 0.00^\circ$ 
Number of layers,  $N_L = 1$ 
Radius of rounding corners,  $R = 40.00$ 
-----

Stepwise Properties
-----
EDGE -A-
Bending Moment,  $M_a = 7.3474831E-010$ 
Shear Force,  $V_a = -2.1566928E-013$ 
EDGE -B-
Bending Moment,  $M_b = -8.7529383E-011$ 
Shear Force,  $V_b = 2.1566928E-013$ 
BOTH EDGES
Axial Force,  $F = -7387.302$ 
Longitudinal Reinforcement Area Distribution (in 2 divisions)
  -Tension:  $A_{slt} = 0.00$ 
  -Compression:  $A_{slc} = 3053.628$ 
Longitudinal Reinforcement Area Distribution (in 3 divisions)
  -Tension:  $A_{sl,ten} = 1017.876$ 
  -Compression:  $A_{sl,com} = 1017.876$ 
  -Middle:  $A_{sl,mid} = 1017.876$ 
Mean Diameter of Tension Reinforcement,  $D_{bL,ten} = 18.00$ 
-----
-----

New component: From table 7-7, ASCE 41_17: Final Shear Capacity  $V_R = 1.0 \cdot V_n = 614519.957$ 
 $V_n ((10.3), ASCE 41-17) = knl \cdot V_{CoIO} = 614519.957$ 
 $V_{CoI} = 614519.957$ 
 $knl = 1.00$ 
displacement_ductility_demand = 0.00
-----

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_{s+} = f \cdot V_f$ '
where  $V_f$  is the contribution of FRPs (11.3), ACI 440).
-----
= 1 (normal-weight concrete)
Mean concrete strength:  $f'_c = (f'_c_{jacket} \cdot Area_{jacket} + f'_c_{core} \cdot Area_{core}) / Area_{section} = 25.00$ , but  $f'_c^{0.5} \leq 8.3$ 
MPa (22.5.3.1, ACI 318-14)
 $M/Vd = 2.00$ 
 $M_u = 8.7529383E-011$ 
 $V_u = 2.1566928E-013$ 
 $d = 0.8 \cdot D = 400.00$ 
 $N_u = 7387.302$ 

```

$A_g = 196349.541$   
 From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 246740.11$   
 $V_{s1} = 246740.11$  is calculated for jacket, with:  
 $A_v = \frac{1}{2} A_{stirrup} = 123370.055$   
 $f_y = 500.00$   
 $s = 100.00$   
 $V_{s1}$  is multiplied by  $Col1 = 1.00$   
 $s/d = 0.25$   
 $V_{s2} = 0.00$  is calculated for core, with:  
 $A_v = \frac{1}{2} A_{stirrup} = 78956.835$   
 $f_y = 500.00$   
 $s = 250.00$   
 $V_{s2}$  is multiplied by  $Col2 = 0.00$   
 $s/d = 1.04167$   
 $V_f ((11-3)-(11.4), ACI 440) = 308320.00$   
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f( , )$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $a = 45^\circ$  and  $a = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $a = b1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, a1)|)$ , with:  
 total thickness per orientation,  $tf1 = NL * t / NoDir = 1.00$   
 $dfv = d$  (figure 11.2, ACI 440) = 470.00  
 $f_{fe} ((11-5), ACI 440) = 328.00$   
 $E_f = 82000.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.009$   
 From (11-11), ACI 440:  $V_s + V_f \leq 417394.406$   
 $bw * d = \frac{1}{4} * d * d = 125663.706$

displacement\_ductility\_demand is calculated as  $\Delta / y$

- Calculation of  $\Delta / y$  for END B -  
for rotation axis 2 and integ. section (b)

From analysis, chord rotation  $\theta = 5.1499725E-021$   
 $y = (M_y * L_s / 3) / E_{eff} = 0.00453062$  ((4.29), Biskinis Phd))  
 $M_y = 2.2517E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 1500.00  
 From table 10.5, ASCE 41\_17:  $E_{eff} = \text{factor} * E_c * I_g = 2.4850E+013$   
 factor = 0.30  
 $A_g = 196349.541$   
 Mean concrete strength:  $f_c' = (f_c'_{jacket} * Area_{jacket} + f_c'_{core} * Area_{core}) / Area_{section} = 33.00$   
 $N = 7387.302$   
 $E_c * I_g = E_{c,jacket} * I_{g,jacket} + E_{c,core} * I_{g,core} = 8.2833E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\Delta$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y,ten}, M_{y,com}) = 2.2517E+008$   
 $y = 6.4424840E-006$   
 $M_{y,ten} (8c) = 2.2517E+008$   
 $\Delta_{ten} (7c) = 61.8137$   
 error of function (7c) = 0.00026387  
 $M_{y,com} (8d) = 8.1922E+008$   
 $\Delta_{com} (7d) = 63.67468$   
 error of function (7d) = -0.0087488  
 with ((10.1), ASCE 41-17)  $e_y = \text{Min}(e_y, 1.25 * e_y * (I_b / I_d)^{2/3}) = 0.0027778$   
 $e_{co} = 0.002$   
 $\alpha_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)

$d1 = 44.00$   
 $R = 250.00$   
 $v = 0.00101374$   
 $N = 7387.302$   
 $Ac = 196349.541$   
 $((10.1), ASCE 41-17) = \text{Min}( , 1.25 * (lb/d)^{2/3} ) = 0.23280363$   
 with  $fc^* ((12.3), ACI 440) = 37.11312$   
 $fc = 33.00$   
 $fl = 1.312$   
 $k = 1$   
 Effective FRP thickness,  $tf = NL * t * \cos(b1) = 1.00$   
 $efe ((12.5) \text{ and } (12.7)) = 0.004$   
 $fu = 0.009$   
 $Ef = 82000.00$

Calculation of ratio  $lb/d$

Inadequate Lap Length with  $lb/d = 0.30$

End Of Calculation of Shear Capacity for element: column JCC1 of floor 1

At local axis: 3

Integration Section: (b)

## Calculation No. 8

column C1, Floor 1

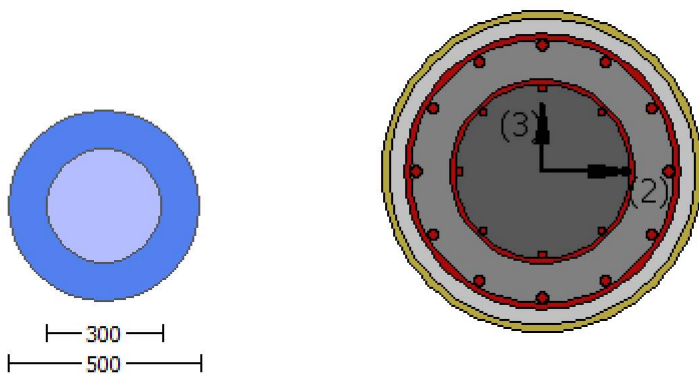
Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity (  $\phi$  )

Edge: End

Local Axis: (3)



Start Of Calculation of Shear Capacity ratio for element: column JCC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rcjcs

## Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Mean strength values are used for both shear and moment calculations.

Consequently:

Jacket

New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

Existing Column

New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Jacket

New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 694.45$

Existing Column

New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 694.45$

#####

External Diameter,  $D = 500.00$

Internal Diameter,  $D = 300.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.51481

Element Length,  $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Inadequate Lap Length with  $l_o/l_{ou,min} = 0.30$

FRP Wrapping Data

Type: Carbon

Dry properties (design values)

Thickness,  $t = 1.00$

Tensile Strength,  $f_{fu} = 840.00$

Tensile Modulus,  $E_f = 82000.00$

Elongation,  $\epsilon_{fu} = 0.009$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i: 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

## Stepwise Properties

At local axis: 3

EDGE -A-

Shear Force,  $V_a = 3.4807814E-030$

EDGE -B-

Shear Force,  $V_b = -3.4807814E-030$

BOTH EDGES

Axial Force,  $F = -7389.214$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 0.00$

-Compression:  $As_c = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t,ten} = 1017.876$

-Compression:  $As_{l,com} = 1017.876$

-Middle:  $As_{l,mid} = 1017.876$

Calculation of Shear Capacity ratio ,  $V_e/V_r = 0.23017425$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 162459.651$   
with

$M_{pr1} = \text{Max}(\mu_{1+}, \mu_{1-}) = 2.4369\text{E}+008$

$\mu_{1+} = 2.4369\text{E}+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$\mu_{1-} = 2.4369\text{E}+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(\mu_{2+}, \mu_{2-}) = 2.4369\text{E}+008$

$\mu_{2+} = 2.4369\text{E}+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

$\mu_{2-} = 2.4369\text{E}+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

-----  
Calculation of  $\mu_{1+}$   
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$   
 $\mu_u = 2.4369\text{E}+008$   
-----

$\phi = 0.80285146$

$\phi' = 0.71403327$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TB DY:  $f_{cc} = f_c^* \quad c = 49.98874$

conf. factor  $c = 1.51481$

$f_c = 33.00$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 389.0139$

$l_b/d = 0.30$

$d_1 = 44.00$

$R = 250.00$

$v = 0.00100813$

$N = 7389.214$

$A_c = 196349.541$

$= \phi \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.1620683$   
-----

Calculation of ratio  $l_b/d$   
-----

Inadequate Lap Length with  $l_b/d = 0.30$   
-----  
-----  
-----

Calculation of  $\mu_{1-}$   
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$   
 $\mu_u = 2.4369\text{E}+008$   
-----

$\phi = 0.80285146$

$\phi' = 0.71403327$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TB DY:  $f_{cc} = f_c^* \quad c = 49.98874$

conf. factor  $c = 1.51481$

$f_c = 33.00$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 389.0139$

$l_b/d = 0.30$

$d_1 = 44.00$

$R = 250.00$

$v = 0.00100813$

$N = 7389.214$

$A_c = 196349.541$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.1620683$$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of  $\mu_{2+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 2.4369 \times 10^8$

$$= 0.80285146$$

$$\lambda = 0.71403327$$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 49.98874$

conf. factor  $c = 1.51481$

$f_c = 33.00$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 389.0139$

$l_b/d = 0.30$

$d_1 = 44.00$

$R = 250.00$

$v = 0.00100813$

$N = 7389.214$

$A_c = 196349.541$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.1620683$$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of  $\mu_{2-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 2.4369 \times 10^8$

$$= 0.80285146$$

$$\lambda = 0.71403327$$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 49.98874$

conf. factor  $c = 1.51481$

$f_c = 33.00$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 389.0139$

$l_b/d = 0.30$

$d_1 = 44.00$

$R = 250.00$

$v = 0.00100813$

$N = 7389.214$

$A_c = 196349.541$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.1620683$$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of Shear Strength  $V_r = \min(V_{r1}, V_{r2}) = 705811.584$

Calculation of Shear Strength at edge 1,  $V_{r1} = 705811.584$

$V_{r1} = V_{Col} \text{ ((10.3), ASCE 41-17)} = k_{nl} * V_{Col0}$

$V_{Col0} = 705811.584$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

Mean concrete strength:  $f_c' = (f_c'_{jacket} * Area_{jacket} + f_c'_{core} * Area_{core}) / Area_{section} = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M / Vd = 2.00$

$\mu_u = 1.2438899E-011$

$\nu_u = 3.4807814E-030$

$d = 0.8 * D = 400.00$

$N_u = 7389.214$

$A_g = 196349.541$

From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 274157.871$

$V_{s1} = 274157.871$  is calculated for jacket, with:

$A_v = \frac{1}{2} * A_{stirrup} = 123370.055$

$f_y = 555.56$

$s = 100.00$

$V_{s1}$  is multiplied by  $Col1 = 1.00$

$s/d = 0.25$

$V_{s2} = 0.00$  is calculated for core, with:

$A_v = \frac{1}{2} * A_{stirrup} = 78956.835$

$f_y = 555.56$

$s = 250.00$

$V_{s2}$  is multiplied by  $Col2 = 0.00$

$s/d = 1.04167$

$V_f \text{ ((11-3)-(11.4), ACI 440)} = 308320.00$

$f = 0.95$ , for fully-wrapped sections

$w_f / s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression, where  $a$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = b1 + 90^\circ = 90.00$

$V_f = \min(|V_f(45, \theta)|, |V_f(-45, a1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L * t / N_{Dir} = 1.00$

$d_{fv} = d$  (figure 11.2, ACI 440) = 470.00

$f_{fe} \text{ ((11-5), ACI 440)} = 328.00$

$E_f = 82000.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.009$

From (11-11), ACI 440:  $V_s + V_f \leq 479549.663$

$b_w * d = \frac{1}{4} * d * d = 125663.706$

Calculation of Shear Strength at edge 2,  $V_{r2} = 705811.584$

$V_{r2} = V_{Col} \text{ ((10.3), ASCE 41-17)} = k_{nl} * V_{Col0}$

$V_{Col0} = 705811.584$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

Mean concrete strength:  $f_c' = (f_c'_{jacket} * Area_{jacket} + f_c'_{core} * Area_{core}) / Area_{section} = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M / Vd = 2.00$

$\mu_u = 1.2438899E-011$

$\nu_u = 3.4807814E-030$

$d = 0.8 * D = 400.00$

$N_u = 7389.214$   
 $A_g = 196349.541$   
 From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 274157.871$   
 $V_{s1} = 274157.871$  is calculated for jacket, with:  
 $A_v = \frac{1}{2} A_{stirrup} = 123370.055$   
 $f_y = 555.56$   
 $s = 100.00$   
 $V_{s1}$  is multiplied by  $Col1 = 1.00$   
 $s/d = 0.25$   
 $V_{s2} = 0.00$  is calculated for core, with:  
 $A_v = \frac{1}{2} A_{stirrup} = 78956.835$   
 $f_y = 555.56$   
 $s = 250.00$   
 $V_{s2}$  is multiplied by  $Col2 = 0.00$   
 $s/d = 1.04167$   
 $V_f ((11-3)-(11.4), ACI 440) = 308320.00$   
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f( , )$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $a = 45^\circ$  and  $a = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $a_1 = b_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, a_1)|, |V_f(-45, a_1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{oDir} = 1.00$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 470.00  
 $f_{fe} ((11-5), ACI 440) = 328.00$   
 $E_f = 82000.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.009$   
 From (11-11), ACI 440:  $V_s + V_f \leq 479549.663$   
 $b_w \cdot d = \frac{1}{4} \cdot d \cdot d = 125663.706$

-----  
 End Of Calculation of Shear Capacity ratio for element: column JCC1 of floor 1  
 At local axis: 3  
 -----

-----  
 Start Of Calculation of Shear Capacity ratio for element: column JCC1 of floor 1  
 At Shear local axis: 2  
 (Bending local axis: 3)  
 Section Type: rcjcs

Constant Properties

-----  
 Knowledge Factor,  $\phi = 1.00$   
 Mean strength values are used for both shear and moment calculations.  
 Consequently:  
 Jacket  
 New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$   
 New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$   
 Concrete Elasticity,  $E_c = 26999.444$   
 Steel Elasticity,  $E_s = 200000.00$   
 Existing Column  
 New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$   
 New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$   
 Concrete Elasticity,  $E_c = 26999.444$   
 Steel Elasticity,  $E_s = 200000.00$   
 #####  
 Note: Especially for the calculation of moment strengths,  
 the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14  
 Jacket  
 New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 694.45$   
 Existing Column  
 New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 694.45$



#####

External Diameter, D = 500.00  
Internal Diameter, D = 300.00  
Cover Thickness, c = 25.00  
Mean Confinement Factor overall section = 1.51481  
Element Length, L = 3000.00  
Primary Member  
Ribbed Bars  
Ductile Steel  
Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Inadequate Lap Length with  $l_o/l_{ou,min} = 0.30$   
FRP Wrapping Data  
Type: Carbon  
Dry properties (design values)  
Thickness, t = 1.00  
Tensile Strength,  $f_{fu} = 840.00$   
Tensile Modulus,  $E_f = 82000.00$   
Elongation,  $ε_{fu} = 0.009$   
Number of directions, NoDir = 1  
Fiber orientations, bi: 0.00°  
Number of layers, NL = 1  
Radius of rounding corners, R = 40.00

#### Stepwise Properties

At local axis: 2  
EDGE -A-  
Shear Force,  $V_a = -2.1312935E-046$   
EDGE -B-  
Shear Force,  $V_b = 2.1312935E-046$   
BOTH EDGES  
Axial Force, F = -7389.214  
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $A_{sl,t} = 0.00$   
-Compression:  $A_{sl,c} = 3053.628$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $A_{sl,ten} = 1017.876$   
-Compression:  $A_{sl,com} = 1017.876$   
-Middle:  $A_{sl,mid} = 1017.876$

Calculation of Shear Capacity ratio ,  $V_e/V_r = 0.23017425$   
Member Controlled by Flexure ( $V_e/V_r < 1$ )  
Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 162459.651$   
with  
 $M_{pr1} = \text{Max}(Mu_{1+}, Mu_{1-}) = 2.4369E+008$   
 $Mu_{1+} = 2.4369E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction  
which is defined for the static loading combination  
 $Mu_{1-} = 2.4369E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment  
direction which is defined for the static loading combination  
 $M_{pr2} = \text{Max}(Mu_{2+}, Mu_{2-}) = 2.4369E+008$   
 $Mu_{2+} = 2.4369E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction  
which is defined for the the static loading combination  
 $Mu_{2-} = 2.4369E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment  
direction which is defined for the the static loading combination

#### Calculation of $Mu_{1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $Mu$   
 $Mu = 2.4369E+008$

= 0.80285146  
 ' = 0.71403327  
 error of function (3.68), Biskinis Phd = 27987.158  
 From 5A.2, TBDY:  $f_{cc} = f_c^* \quad c = 49.98874$   
 conf. factor  $c = 1.51481$   
 $f_c = 33.00$   
 From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 389.0139$   
 $l_b/d = 0.30$   
 $d1 = 44.00$   
 $R = 250.00$   
 $v = 0.00100813$   
 $N = 7389.214$   
 $Ac = 196349.541$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.1620683$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of  $\mu_1$ -

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 2.4369E+008$

= 0.80285146  
 ' = 0.71403327  
 error of function (3.68), Biskinis Phd = 27987.158  
 From 5A.2, TBDY:  $f_{cc} = f_c^* \quad c = 49.98874$   
 conf. factor  $c = 1.51481$   
 $f_c = 33.00$   
 From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 389.0139$   
 $l_b/d = 0.30$   
 $d1 = 44.00$   
 $R = 250.00$   
 $v = 0.00100813$   
 $N = 7389.214$   
 $Ac = 196349.541$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.1620683$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of  $\mu_2$ +

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 2.4369E+008$

= 0.80285146  
 ' = 0.71403327  
 error of function (3.68), Biskinis Phd = 27987.158  
 From 5A.2, TBDY:  $f_{cc} = f_c^* \quad c = 49.98874$   
 conf. factor  $c = 1.51481$   
 $f_c = 33.00$   
 From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 389.0139$   
 $l_b/d = 0.30$

$d1 = 44.00$   
 $R = 250.00$   
 $v = 0.00100813$   
 $N = 7389.214$   
 $Ac = 196349.541$   
 $= *Min(1, 1.25*(lb/d)^{2/3}) = 0.1620683$

Calculation of ratio  $lb/d$

Inadequate Lap Length with  $lb/d = 0.30$

Calculation of  $Mu2$ -

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $Mu$   
 $Mu = 2.4369E+008$

$= 0.80285146$   
 $' = 0.71403327$   
 error of function (3.68), Biskinis Phd = 27987.158  
 From 5A.2, TDY:  $fcc = fc * c = 49.98874$   
 conf. factor  $c = 1.51481$   
 $fc = 33.00$   
 From 10.3.5, ASCE 41-17, Final value of  $fy$ :  $fy * Min(1, 1.25*(lb/d)^{2/3}) = 389.0139$   
 $lb/d = 0.30$   
 $d1 = 44.00$   
 $R = 250.00$   
 $v = 0.00100813$   
 $N = 7389.214$   
 $Ac = 196349.541$   
 $= *Min(1, 1.25*(lb/d)^{2/3}) = 0.1620683$

Calculation of ratio  $lb/d$

Inadequate Lap Length with  $lb/d = 0.30$

Calculation of Shear Strength  $Vr = Min(Vr1, Vr2) = 705811.584$

Calculation of Shear Strength at edge 1,  $Vr1 = 705811.584$   
 $Vr1 = VCol$  ((10.3), ASCE 41-17) =  $knl * VCol0$   
 $VCol0 = 705811.584$   
 $knl = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $Vs = Av * fy * d / s$ ' is replaced by ' $Vs + f * Vf$ ' where  $Vf$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)  
 Mean concrete strength:  $fc' = (fc'_{jacket} * Area_{jacket} + fc'_{core} * Area_{core}) / Area_{section} = 33.00$ , but  $fc'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $Mu = 1.5940012E-011$   
 $Vu = 2.1312935E-046$   
 $d = 0.8 * D = 400.00$   
 $Nu = 7389.214$   
 $Ag = 196349.541$   
 From (11.5.4.8), ACI 318-14:  $Vs = Vs1 + Vs2 = 274157.871$   
 $Vs1 = 274157.871$  is calculated for jacket, with:  
 $Av = /2 * A_{stirrup} = 123370.055$   
 $fy = 555.56$

$s = 100.00$   
 $V_{s1}$  is multiplied by  $Col1 = 1.00$   
 $s/d = 0.25$   
 $V_{s2} = 0.00$  is calculated for core, with:  
 $A_v = \frac{1}{2} A_{stirrup} = 78956.835$   
 $f_y = 555.56$   
 $s = 250.00$   
 $V_{s2}$  is multiplied by  $Col2 = 0.00$   
 $s/d = 1.04167$   
 $V_f ((11-3)-(11.4), ACI 440) = 308320.00$   
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = b1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, a1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = N_L * t / N_{oDir} = 1.00$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 470.00  
 $f_{fe} ((11-5), ACI 440) = 328.00$   
 $E_f = 82000.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.009$   
 From (11-11), ACI 440:  $V_s + V_f \leq 479549.663$   
 $b_w * d = \frac{V_s * d}{4} = 125663.706$

Calculation of Shear Strength at edge 2,  $V_{r2} = 705811.584$   
 $V_{r2} = V_{Col} ((10.3), ASCE 41-17) = k_{nl} * V_{ColO}$   
 $V_{ColO} = 705811.584$   
 $k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ '  
 where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\gamma_c = 1$  (normal-weight concrete)  
 Mean concrete strength:  $f'_c = (f'_{c\_jacket} * Area_{jacket} + f'_{c\_core} * Area_{core}) / Area_{section} = 33.00$ , but  $f'_c^{0.5} \leq 8.3$   
 MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $M_u = 1.5940012E-011$   
 $V_u = 2.1312935E-046$   
 $d = 0.8 * D = 400.00$   
 $N_u = 7389.214$   
 $A_g = 196349.541$   
 From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 274157.871$   
 $V_{s1} = 274157.871$  is calculated for jacket, with:  
 $A_v = \frac{1}{2} A_{stirrup} = 123370.055$   
 $f_y = 555.56$   
 $s = 100.00$   
 $V_{s1}$  is multiplied by  $Col1 = 1.00$   
 $s/d = 0.25$   
 $V_{s2} = 0.00$  is calculated for core, with:  
 $A_v = \frac{1}{2} A_{stirrup} = 78956.835$   
 $f_y = 555.56$   
 $s = 250.00$   
 $V_{s2}$  is multiplied by  $Col2 = 0.00$   
 $s/d = 1.04167$   
 $V_f ((11-3)-(11.4), ACI 440) = 308320.00$   
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = b1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, a1)|)$ , with:

total thickness per orientation,  $tf1 = NL * t / NoDir = 1.00$   
 $dfv = d$  (figure 11.2, ACI 440) = 470.00  
 $ffe$  ((11-5), ACI 440) = 328.00  
 $Ef = 82000.00$   
 $fe = 0.004$ , from (11.6a), ACI 440  
with  $fu = 0.009$   
From (11-11), ACI 440:  $Vs + Vf <= 479549.663$   
 $bw * d = *d * d / 4 = 125663.706$

End Of Calculation of Shear Capacity ratio for element: column JCC1 of floor 1  
At local axis: 2

Start Of Calculation of Chord Rotation Capacity for element: column JCC1 of floor 1

At local axis: 3  
Integration Section: (b)  
Section Type: rcjcs

Constant Properties

Knowledge Factor,  $= 1.00$   
Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE 41-17.  
Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17  
Consequently:  
Jacket  
New material of Primary Member: Concrete Strength,  $fc = fcm = 33.00$   
New material of Primary Member: Steel Strength,  $fs = fsm = 555.56$   
Concrete Elasticity,  $Ec = 26999.444$   
Steel Elasticity,  $Es = 200000.00$   
Existing Column  
New material of Primary Member: Concrete Strength,  $fc = fcm = 33.00$   
New material of Primary Member: Steel Strength,  $fs = fsm = 555.56$   
Concrete Elasticity,  $Ec = 26999.444$   
Steel Elasticity,  $Es = 200000.00$   
External Diameter,  $D = 500.00$   
Internal Diameter,  $D = 300.00$   
Cover Thickness,  $c = 25.00$   
Element Length,  $L = 3000.00$   
Primary Member  
Ribbed Bars  
Ductile Steel  
Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Inadequate Lap Length with  $lb / ld = 0.30$   
FRP Wrapping Data  
Type: Carbon  
Dry properties (design values)  
Thickness,  $t = 1.00$   
Tensile Strength,  $ffu = 840.00$   
Tensile Modulus,  $Ef = 82000.00$   
Elongation,  $efu = 0.009$   
Number of directions,  $NoDir = 1$   
Fiber orientations,  $bi: 0.00^\circ$   
Number of layers,  $NL = 1$   
Radius of rounding corners,  $R = 40.00$

Stepwise Properties

Bending Moment,  $M = 0.07981876$   
Shear Force,  $V2 = 4779.203$   
Shear Force,  $V3 = 2.1566928E-013$   
Axial Force,  $F = -7387.302$   
Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 0.00$   
 -Compression:  $As_c = 3053.628$   
 Longitudinal Reinforcement Area Distribution (in 3 divisions)  
 -Tension:  $As_{ten} = 1017.876$   
 -Compression:  $As_{com} = 1017.876$   
 -Middle:  $As_{mid} = 1017.876$   
 Mean Diameter of Tension Reinforcement,  $Db_L = 18.00$

New component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_R = 1.0^*$   $u = 0.00090612$   
 $u = y + p = 0.00090612$

- Calculation of  $y$  -

$y = (My * L_s / 3) / E_{eff} = 0.00090612$  ((4.29), Biskinis Phd))  
 $My = 2.2517E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 300.00  
 From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 2.4850E+013$   
 $factor = 0.30$   
 $A_g = 196349.541$   
 Mean concrete strength:  $fc' = (fc'_{jacket} * Area_{jacket} + fc'_{core} * Area_{core}) / Area_{section} = 33.00$   
 $N = 7387.302$   
 $E_c * I_g = E_{c\_jacket} * I_{g\_jacket} + E_{c\_core} * I_{g\_core} = 8.2833E+013$

Calculation of Yielding Moment  $My$

Calculation of  $y$  and  $My$  according to (7) - (8) in Biskinis and Fardis

$My = \min(My_{ten}, My_{com}) = 2.2517E+008$   
 $y = 6.4424840E-006$   
 $My_{ten}$  (8c) = 2.2517E+008  
 $_{ten}$  (7c) = 61.8137  
 error of function (7c) = 0.00026387  
 $My_{com}$  (8d) = 8.1922E+008  
 $_{com}$  (7d) = 63.67468  
 error of function (7d) = -0.0087488  
 with ((10.1), ASCE 41-17)  $ey = \min(ey, 1.25 * ey * (I_b / I_d)^{2/3}) = 0.0027778$   
 $eco = 0.002$   
 $apl = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d1 = 44.00$   
 $R = 250.00$   
 $v = 0.00101374$   
 $N = 7387.302$   
 $Ac = 196349.541$   
 ((10.1), ASCE 41-17)  $= \min( , 1.25 * (I_b / I_d)^{2/3}) = 0.23280363$   
 with  $fc^*$  ((12.3), ACI 440) = 37.11312  
 $fc = 33.00$   
 $f_l = 1.312$   
 $k = 1$   
 Effective FRP thickness,  $tf = NL * t * \cos(b1) = 1.00$   
 $efe$  ((12.5) and (12.7)) = 0.004  
 $fu = 0.009$   
 $Ef = 82000.00$

Calculation of ratio  $I_b / I_d$

Inadequate Lap Length with  $I_b / I_d = 0.30$

- Calculation of  $p$  -

From table 10-9:  $p = 0.00$

with:

- Columns controlled by inadequate development or splicing along the clear height because  $l_b/l_d < 1$

shear control ratio  $V_{yE}/V_{CoIE} = 0.23017425$

$d = d_{\text{external}} = 0.00$

$s = s_{\text{external}} = 0.00$

$t = s_1 + s_2 + 2 \cdot t_f/bw \cdot (f_{fe}/f_s) = 0.00795744$

jacket:  $s_1 = A_{v1} \cdot (D_{c1}/2)/(s_1 \cdot A_g) = 0.0027646$

$A_{v1} = 78.53982$ , is the area of stirrup

$D_{c1} = D_{\text{ext}} - 2 \cdot \text{cover} - \text{External Hoop Diameter} = 440.00$ , is the total Length of all stirrups parallel to loading

(shear) direction

$s_1 = 100.00$

core:  $s_2 = A_{v2} \cdot (D_{c2}/2)/(s_2 \cdot A_g) = 0.00046968$

$A_{v2} = 50.26548$ , is the area of stirrup

$D_{c2} = D_{\text{int}} - \text{Internal Hoop Diameter} = 292.00$ , is the total Length of all stirrups parallel to loading (shear)

direction

$s_2 = 250.00$

The term  $2 \cdot t_f/bw \cdot (f_{fe}/f_s)$  is implemented to account for FRP contribution

where  $f = 2 \cdot t_f/bw$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe}/f_s$  normalises  $f$  to steel strength

All these variables have already been given in Shear control ratio calculation.

For the normalisation  $f_s$  of jacket is used.

$NUD = 7387.302$

$A_g = 196349.541$

$f_{cE} = (f_{c\_jacket} \cdot \text{Area\_jacket} + f_{c\_core} \cdot \text{Area\_core})/\text{section\_area} = 33.00$

$f_{yIE} = (f_{y\_ext\_Long\_Reinf} \cdot \text{Area\_ext\_Long\_Reinf} + f_{y\_int\_Long\_Reinf} \cdot \text{Area\_int\_Long\_Reinf})/\text{Area\_Tot\_Long\_Rein} = 21219958E-314$

$f_{yTE} = (f_{y\_ext\_Trans\_Reinf} \cdot \text{Area\_ext\_Trans\_Reinf} + f_{y\_int\_Trans\_Reinf} \cdot \text{Area\_int\_Trans\_Reinf})/\text{Area\_Tot\_Trans\_Rein} = 555.56$

$p_l = \text{Area\_Tot\_Long\_Rein}/(A_g) = 0.015552$

$f_{cE} = 33.00$

End Of Calculation of Chord Rotation Capacity for element: column JCC1 of floor 1

At local axis: 3

Integration Section: (b)

## Calculation No. 9

column C1, Floor 1

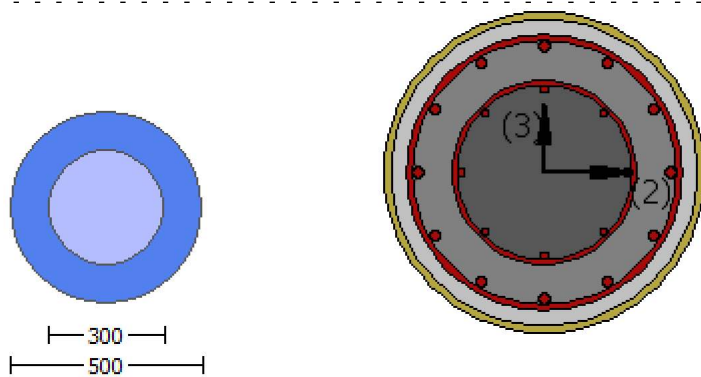
Limit State: Life Safety (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Shear capacity  $V_{Rd}$

Edge: Start

Local Axis: (2)



Start Of Calculation of Shear Capacity for element: column JCC1 of floor 1  
 At local axis: 2  
 Integration Section: (a)  
 Section Type: rcjcs

Constant Properties

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Knowledge Factor,  $\gamma = 1.00$   
 Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE 41-17.  
 Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17  
 Consequently:  
 Jacket  
 New material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 25.00$   
 New material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 500.00$   
 Concrete Elasticity,  $E_c = 26999.444$   
 Steel Elasticity,  $E_s = 200000.00$   
 Existing Column  
 New material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 25.00$   
 New material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 500.00$   
 Concrete Elasticity,  $E_c = 26999.444$   
 Steel Elasticity,  $E_s = 200000.00$   
 #####  
 Note: Especially for the calculation of  $\gamma$  for displacement ductility demand,  
 the expected (mean value) strengths are used (7.5.1.3, ASCE 41-17) because bending is considered as  
 Deformation-Controlled Action (Table C7-1, ASCE 41-17).  
 Jacket  
 New material: Concrete Strength,  $f_c = f_{cm} = 33.00$   
 New material: Steel Strength,  $f_s = f_{sm} = 555.56$   
 Existing Column  
 New material: Concrete Strength,  $f_c = f_{cm} = 33.00$   
 New material: Steel Strength,  $f_s = f_{sm} = 555.56$   
 #####  
 External Diameter,  $D = 500.00$   
 Internal Diameter,  $D = 300.00$   
 Cover Thickness,  $c = 25.00$   
 Element Length,  $L = 3000.00$   
 Primary Member  
 Ribbed Bars  
 Ductile Steel  
 Without Detailing for Earthquake Resistance (including stirrups not closed at  $135^\circ$ )  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Inadequate Lap Length with  $l_o/l_{ou,min} = l_b/l_d = 0.30$   
 FRP Wrapping Data  
 Type: Carbon  
 Dry properties (design values)  
 Thickness,  $t = 1.00$   
 Tensile Strength,  $f_{fu} = 840.00$   
 Tensile Modulus,  $E_f = 82000.00$   
 Elongation,  $e_{fu} = 0.009$   
 Number of directions,  $NoDir = 1$   
 Fiber orientations,  $bi: 0.00^\circ$   
 Number of layers,  $NL = 1$



Radius of rounding corners, R = 40.00

### Stepwise Properties

EDGE -A-

Bending Moment, Ma = -2.2653E+007

Shear Force, Va = -7549.016

EDGE -B-

Bending Moment, Mb = 0.12607815

Shear Force, Vb = 7549.016

BOTH EDGES

Axial Force, F = -7386.194

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension: Aslt = 1272.345

-Compression: Aslc = 1781.283

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension: Asl,ten = 1017.876

-Compression: Asl,com = 1017.876

-Middle: Asl,mid = 1017.876

Mean Diameter of Tension Reinforcement, DbL,ten = 18.00

New component: From table 7-7, ASCE 41\_17: Final Shear Capacity VR = 1.0\*Vn = 515957.072

Vn ((10.3), ASCE 41-17) = knl\*VCol0 = 515957.072

VCol = 515957.072

knl = 1.00

displacement\_ductility\_demand = 0.02781594

NOTE: In expression (10-3) 'Vs = Av\*fy\*d/s' is replaced by 'Vs+ f\*Vf'  
where Vf is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

Mean concrete strength:  $fc' = (fc'_{jacket} \cdot Area_{jacket} + fc'_{core} \cdot Area_{core}) / Area_{section} = 25.00$ , but  $fc'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

M/Vd = 4.00

Mu = 2.2653E+007

Vu = 7549.016

d = 0.8\*D = 400.00

Nu = 7386.194

Ag = 196349.541

From (11.5.4.8), ACI 318-14: Vs = Vs1 + Vs2 = 246740.11

Vs1 = 246740.11 is calculated for jacket, with:

$Av = \frac{1}{2} \cdot A_{stirrup} = 123370.055$

fy = 500.00

s = 100.00

Vs1 is multiplied by Col1 = 1.00

s/d = 0.25

Vs2 = 0.00 is calculated for core, with:

$Av = \frac{1}{2} \cdot A_{stirrup} = 78956.835$

fy = 500.00

s = 250.00

Vs2 is multiplied by Col2 = 0.00

s/d = 1.04167

Vf ((11-3)-(11.4), ACI 440) = 308320.00

f = 0.95, for fully-wrapped sections

wf/sf = 1 (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression, where  $\alpha$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function Vf(  $\alpha$  ), is implemented for every different fiber orientation ai, as well as for 2 crack directions,  $\alpha = 45^\circ$  and  $\alpha = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\alpha_1 = \alpha_1 + 90^\circ = 90.00$

$Vf = \min(|Vf(45, \alpha_1)|, |Vf(-45, \alpha_1)|)$ , with:

total thickness per orientation, tf1 = NL\*t/NoDir = 1.00

dfv = d (figure 11.2, ACI 440) = 470.00

$f_{fe} ((11-5), ACI 440) = 328.00$   
 $E_f = 82000.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.009$   
 From (11-11), ACI 440:  $V_s + V_f \leq 417394.406$   
 $b_w \cdot d = \frac{d^2}{4} = 125663.706$

displacement ductility demand is calculated as  $\frac{1}{y}$

- Calculation of  $\frac{1}{y}$  for END A -  
 for rotation axis 3 and integ. section (a)

From analysis, chord rotation  $\theta = 0.00025211$   
 $y = (M_y \cdot L_s / 3) / E_{eff} = 0.00906345$  ((4.29), Biskinis Phd))  
 $M_y = 2.2517E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) = 3000.736  
 From table 10.5, ASCE 41-17:  $E_{eff} = \text{factor} \cdot E_c \cdot I_g = 2.4850E+013$   
 $\text{factor} = 0.30$   
 $A_g = 196349.541$   
 Mean concrete strength:  $f'_c = (f'_{c\_jacket} \cdot \text{Area}_{jacket} + f'_{c\_core} \cdot \text{Area}_{core}) / \text{Area}_{section} = 33.00$   
 $N = 7386.194$   
 $E_c \cdot I_g = E_{c\_jacket} \cdot I_{g\_jacket} + E_{c\_core} \cdot I_{g\_core} = 8.2833E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\frac{1}{y}$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \min(M_{y\_ten}, M_{y\_com}) = 2.2517E+008$   
 $y = 6.4424821E-006$   
 $M_{y\_ten} (8c) = 2.2517E+008$   
 $\frac{1}{y_{ten}} (7c) = 61.81367$   
 error of function (7c) = 0.00026388  
 $M_{y\_com} (8d) = 8.1922E+008$   
 $\frac{1}{y_{com}} (7d) = 63.67467$   
 error of function (7d) = -0.00874883  
 with ((10.1), ASCE 41-17)  $e_y = \min(e_y, 1.25 \cdot e_y \cdot (I_b/I_d)^{2/3}) = 0.0027778$   
 $e_{co} = 0.002$   
 $a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 250.00$   
 $v = 0.00101359$   
 $N = 7386.194$   
 $A_c = 196349.541$   
 ((10.1), ASCE 41-17)  $\frac{1}{y} = \min(\frac{1}{y}, 1.25 \cdot \frac{1}{y} \cdot (I_b/I_d)^{2/3}) = 0.23280363$   
 with  $f'_c$  ((12.3), ACI 440) = 37.11312  
 $f_c = 33.00$   
 $f_l = 1.312$   
 $k = 1$   
 Effective FRP thickness,  $t_f = N \cdot t \cdot \cos(b_1) = 1.00$   
 $e_f$  ((12.5) and (12.7)) = 0.004  
 $f_u = 0.009$   
 $E_f = 82000.00$

Calculation of ratio  $I_b/I_d$

Inadequate Lap Length with  $I_b/I_d = 0.30$

End Of Calculation of Shear Capacity for element: column JCC1 of floor 1  
 At local axis: 2  
 Integration Section: (a)

## Calculation No. 10

column C1, Floor 1

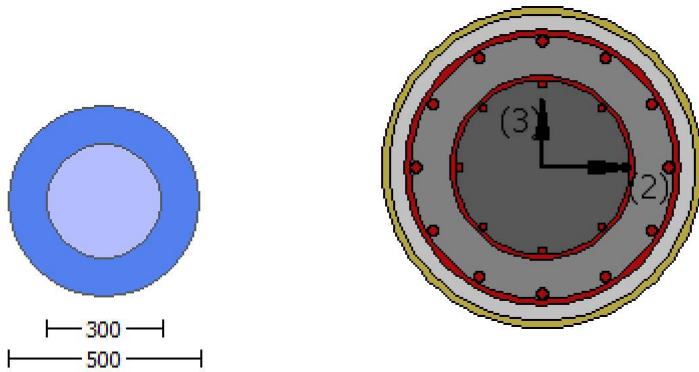
Limit State: Life Safety (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity (  $\phi$  )

Edge: Start

Local Axis: (2)



Start Of Calculation of Shear Capacity ratio for element: column JCC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rcjcs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Mean strength values are used for both shear and moment calculations.

Consequently:

Jacket

New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

Existing Column

New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Jacket

New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 694.45$

Existing Column

New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 694.45$

#####

|   |
|---|
| External Diameter, D = 500.00   |
| Internal Diameter, D = 300.00   |
| Cover Thickness, c = 25.00  |
| Mean Confinement Factor overall section = 1.51481   |
| Element Length, L = 3000.00   |
| Primary Member  |
| Ribbed Bars   |
| Ductile Steel   |
| Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)   |
| Longitudinal Bars With Ends Lapped Starting at the End Sections   |
| Inadequate Lap Length with $l_o/l_{o,min}$ = 0.30   |
| FRP Wrapping Data   |
| Type: Carbon  |
| Dry properties (design values)  |
| Thickness, t = 1.00   |
| Tensile Strength, $f_{fu}$ = 840.00   |
| Tensile Modulus, $E_f$ = 82000.00   |
| Elongation, $ε_{fu}$ = 0.009  |
| Number of directions, NoDir = 1   |
| Fiber orientations, $b_i$ : 0.00°   |
| Number of layers, NL = 1  |
| Radius of rounding corners, R = 40.00   |
| -----   |
| Stepwise Properties   |
| -----   |
| At local axis: 3  |
| EDGE -A-  |
| Shear Force, $V_a$ = 3.4807814E-030   |
| EDGE -B-  |
| Shear Force, $V_b$ = -3.4807814E-030  |
| BOTH EDGES  |
| Axial Force, F = -7389.214  |
| Longitudinal Reinforcement Area Distribution (in 2 divisions)   |
| -Tension: $A_{sl,t}$ = 0.00   |
| -Compression: $A_{sl,c}$ = 3053.628   |
| Longitudinal Reinforcement Area Distribution (in 3 divisions)   |
| -Tension: $A_{sl,ten}$ = 1017.876   |
| -Compression: $A_{sl,com}$ = 1017.876   |
| -Middle: $A_{sl,mid}$ = 1017.876  |
| -----   |
| -----   |
| Calculation of Shear Capacity ratio , $V_e/V_r$ = 0.23017425  |
| Member Controlled by Flexure ( $V_e/V_r < 1$ )  |
| Calculation of Shear Demand from fig. R18.6.5, ACI 318-14 $V_e = (M_{pr1} + M_{pr2})/l_n = 162459.651$                        |
| with  |
| $M_{pr1} = \text{Max}(\mu_{u1+} , \mu_{u1-}) = 2.4369\text{E}+008$  |
| $\mu_{u1+} = 2.4369\text{E}+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction |
| which is defined for the static loading combination   |
| $\mu_{u1-} = 2.4369\text{E}+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment         |
| direction which is defined for the static loading combination   |
| $M_{pr2} = \text{Max}(\mu_{u2+} , \mu_{u2-}) = 2.4369\text{E}+008$  |
| $\mu_{u2+} = 2.4369\text{E}+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction |
| which is defined for the the static loading combination   |
| $\mu_{u2-} = 2.4369\text{E}+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment         |
| direction which is defined for the the static loading combination   |
| -----   |
| Calculation of $\mu_{u1+}$  |
| -----   |
| -----   |
| Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), $\mu_u$   |
| $\mu_u = 2.4369\text{E}+008$  |
| -----   |

= 0.80285146  
 ' = 0.71403327  
 error of function (3.68), Biskinis Phd = 27987.158  
 From 5A.2, TBDY:  $f_{cc} = f_c^* \quad c = 49.98874$   
 conf. factor  $c = 1.51481$   
 $f_c = 33.00$   
 From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 389.0139$   
 $l_b/l_d = 0.30$   
 $d_1 = 44.00$   
 $R = 250.00$   
 $v = 0.00100813$   
 $N = 7389.214$   
 $Ac = 196349.541$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.1620683$

Calculation of ratio  $l_b/l_d$

Inadequate Lap Length with  $l_b/l_d = 0.30$

Calculation of  $\mu_1$ -

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 2.4369\text{E}+008$

= 0.80285146  
 ' = 0.71403327  
 error of function (3.68), Biskinis Phd = 27987.158  
 From 5A.2, TBDY:  $f_{cc} = f_c^* \quad c = 49.98874$   
 conf. factor  $c = 1.51481$   
 $f_c = 33.00$   
 From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 389.0139$   
 $l_b/l_d = 0.30$   
 $d_1 = 44.00$   
 $R = 250.00$   
 $v = 0.00100813$   
 $N = 7389.214$   
 $Ac = 196349.541$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.1620683$

Calculation of ratio  $l_b/l_d$

Inadequate Lap Length with  $l_b/l_d = 0.30$

Calculation of  $\mu_2$ +

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 2.4369\text{E}+008$

= 0.80285146  
 ' = 0.71403327  
 error of function (3.68), Biskinis Phd = 27987.158  
 From 5A.2, TBDY:  $f_{cc} = f_c^* \quad c = 49.98874$   
 conf. factor  $c = 1.51481$   
 $f_c = 33.00$   
 From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 389.0139$   
 $l_b/l_d = 0.30$   
 $d_1 = 44.00$

$$\begin{aligned}
 R &= 250.00 \\
 v &= 0.00100813 \\
 N &= 7389.214 \\
 A_c &= 196349.541 \\
 &= * \text{Min}(1, 1.25 * (l_b / d)^{2/3}) = 0.1620683
 \end{aligned}$$

Calculation of ratio  $l_b / d$

Inadequate Lap Length with  $l_b / d = 0.30$

Calculation of  $\mu_2$ -

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 2.4369 \text{E}+008$

$$\begin{aligned}
 &= 0.80285146 \\
 ' &= 0.71403327 \\
 \text{error of function (3.68), Biskinis Phd} &= 27987.158 \\
 \text{From 5A.2, TBDY: } f_{cc} &= f_c' \cdot c = 49.98874 \\
 \text{conf. factor } c &= 1.51481 \\
 f_c &= 33.00 \\
 \text{From 10.3.5, ASCE 41-17, Final value of } f_y &: f_y * \text{Min}(1, 1.25 * (l_b / d)^{2/3}) = 389.0139 \\
 l_b / d &= 0.30 \\
 d_1 &= 44.00 \\
 R &= 250.00 \\
 v &= 0.00100813 \\
 N &= 7389.214 \\
 A_c &= 196349.541 \\
 &= * \text{Min}(1, 1.25 * (l_b / d)^{2/3}) = 0.1620683
 \end{aligned}$$

Calculation of ratio  $l_b / d$

Inadequate Lap Length with  $l_b / d = 0.30$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 705811.584$

Calculation of Shear Strength at edge 1,  $V_{r1} = 705811.584$

$V_{r1} = V_{c0} \text{ ((10.3), ASCE 41-17)} = k_{nl} * V_{c0}$

$V_{c0} = 705811.584$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ '  
 where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$$\begin{aligned}
 &= 1 \text{ (normal-weight concrete)} \\
 \text{Mean concrete strength: } f_c' &= (f_c'_{\text{jacket}} * \text{Area}_{\text{jacket}} + f_c'_{\text{core}} * \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 33.00, \text{ but } f_c'^{0.5} \leq 8.3 \\
 \text{MPa (22.5.3.1, ACI 318-14)} \\
 M / V_d &= 2.00 \\
 \mu &= 1.2438899 \text{E-011} \\
 V_u &= 3.4807814 \text{E-030} \\
 d &= 0.8 * D = 400.00 \\
 N_u &= 7389.214 \\
 A_g &= 196349.541 \\
 \text{From (11.5.4.8), ACI 318-14: } V_s &= V_{s1} + V_{s2} = 274157.871 \\
 V_{s1} &= 274157.871 \text{ is calculated for jacket, with:} \\
 A_v &= /2 * A_{\text{stirrup}} = 123370.055 \\
 f_y &= 555.56 \\
 s &= 100.00
 \end{aligned}$$

Vs1 is multiplied by Col1 = 1.00  
s/d = 0.25

Vs2 = 0.00 is calculated for core, with:

$A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 78956.835$

$f_y = 555.56$

s = 250.00

Vs2 is multiplied by Col2 = 0.00

s/d = 1.04167

Vf ((11-3)-(11.4), ACI 440) = 308320.00

f = 0.95, for fully-wrapped sections

wf/sf = 1 (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression,  
where  $a$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function Vf(  $a$  ), is implemented for every different fiber orientation  $a_i$ ,  
as well as for 2 crack directions,  $a = 45^\circ$  and  $a = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta_1 = b_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, a_1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{\text{Dir}} = 1.00$

dfv = d (figure 11.2, ACI 440) = 470.00

ffe ((11-5), ACI 440) = 328.00

$E_f = 82000.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.009$

From (11-11), ACI 440:  $V_s + V_f \leq 479549.663$

$b_w \cdot d = \sqrt{3} \cdot d \cdot d / 4 = 125663.706$

Calculation of Shear Strength at edge 2,  $V_{r2} = 705811.584$

$V_{r2} = V_{\text{Col}} ((10.3), \text{ASCE 41-17}) = k_{nl} \cdot V_{\text{ColO}}$

$V_{\text{ColO}} = 705811.584$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ '  
where Vf is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

Mean concrete strength:  $f'_c = (f'_c_{\text{jacket}} \cdot \text{Area}_{\text{jacket}} + f'_c_{\text{core}} \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 33.00$ , but  $f'_c^{0.5} \leq 8.3$   
MPa (22.5.3.1, ACI 318-14)

$M / Vd = 2.00$

$\mu_u = 1.2438899\text{E-}011$

$\nu_u = 3.4807814\text{E-}030$

d = 0.8 · D = 400.00

$N_u = 7389.214$

$A_g = 196349.541$

From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 274157.871$

Vs1 = 274157.871 is calculated for jacket, with:

$A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 123370.055$

$f_y = 555.56$

s = 100.00

Vs1 is multiplied by Col1 = 1.00

s/d = 0.25

Vs2 = 0.00 is calculated for core, with:

$A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 78956.835$

$f_y = 555.56$

s = 250.00

Vs2 is multiplied by Col2 = 0.00

s/d = 1.04167

Vf ((11-3)-(11.4), ACI 440) = 308320.00

f = 0.95, for fully-wrapped sections

wf/sf = 1 (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression,  
where  $a$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function Vf(  $a$  ), is implemented for every different fiber orientation  $a_i$ ,  
as well as for 2 crack directions,  $a = 45^\circ$  and  $a = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta_1 = b_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, a_1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{\text{Dir}} = 1.00$

```

dfv = d (figure 11.2, ACI 440) = 470.00
ffe ((11-5), ACI 440) = 328.00
Ef = 82000.00
fe = 0.004, from (11.6a), ACI 440
with fu = 0.009
From (11-11), ACI 440: Vs + Vf <= 479549.663
bw*d = *d*d/4 = 125663.706
-----

End Of Calculation of Shear Capacity ratio for element: column JCC1 of floor 1
At local axis: 3
-----

Start Of Calculation of Shear Capacity ratio for element: column JCC1 of floor 1
At Shear local axis: 2
(Bending local axis: 3)
Section Type: rcjcs

Constant Properties
-----
Knowledge Factor, = 1.00
Mean strength values are used for both shear and moment calculations.
Consequently:
Jacket
New material of Primary Member: Concrete Strength, fc = fcm = 33.00
New material of Primary Member: Steel Strength, fs = fsm = 555.56
Concrete Elasticity, Ec = 26999.444
Steel Elasticity, Es = 200000.00
Existing Column
New material of Primary Member: Concrete Strength, fc = fcm = 33.00
New material of Primary Member: Steel Strength, fs = fsm = 555.56
Concrete Elasticity, Ec = 26999.444
Steel Elasticity, Es = 200000.00
#####
Note: Especially for the calculation of moment strengths,
the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14
Jacket
New material: Steel Strength, fs = 1.25*fsm = 694.45
Existing Column
New material: Steel Strength, fs = 1.25*fsm = 694.45
#####
External Diameter, D = 500.00
Internal Diameter, D = 300.00
Cover Thickness, c = 25.00
Mean Confinement Factor overall section = 1.51481
Element Length, L = 3000.00
Primary Member
Ribbed Bars
Ductile Steel
Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)
Longitudinal Bars With Ends Lapped Starting at the End Sections
Inadequate Lap Length with lo/lo,min = 0.30
FRP Wrapping Data
Type: Carbon
Dry properties (design values)
Thickness, t = 1.00
Tensile Strength, ffu = 840.00
Tensile Modulus, Ef = 82000.00
Elongation, efu = 0.009
Number of directions, NoDir = 1
Fiber orientations, bi: 0.00°
Number of layers, NL = 1
Radius of rounding corners, R = 40.00
-----

```



## Stepwise Properties

At local axis: 2

EDGE -A-

Shear Force,  $V_a = -2.1312935E-046$

EDGE -B-

Shear Force,  $V_b = 2.1312935E-046$

BOTH EDGES

Axial Force,  $F = -7389.214$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 0.00$

-Compression:  $As_c = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t,ten} = 1017.876$

-Compression:  $As_{l,com} = 1017.876$

-Middle:  $As_{l,mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.23017425$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 162459.651$   
with

$M_{pr1} = \text{Max}(Mu_{1+}, Mu_{1-}) = 2.4369E+008$

$Mu_{1+} = 2.4369E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$Mu_{1-} = 2.4369E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(Mu_{2+}, Mu_{2-}) = 2.4369E+008$

$Mu_{2+} = 2.4369E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination

$Mu_{2-} = 2.4369E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

Calculation of  $Mu_{1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$

$M_u = 2.4369E+008$

$= 0.80285146$

$' = 0.71403327$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TDY:  $f_{cc} = f_c^* \quad c = 49.98874$

conf. factor  $c = 1.51481$

$f_c = 33.00$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 389.0139$

$l_b/l_d = 0.30$

$d_1 = 44.00$

$R = 250.00$

$v = 0.00100813$

$N = 7389.214$

$Ac = 196349.541$

$= \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.1620683$

Calculation of ratio  $l_b/l_d$

Inadequate Lap Length with  $l_b/l_d = 0.30$

Calculation of  $Mu_{1-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.4369E+008

= 0.80285146  
' = 0.71403327  
error of function (3.68), Biskinis Phd = 27987.158  
From 5A.2, TDY: fcc = fc\* c = 49.98874  
conf. factor c = 1.51481  
fc = 33.00  
From 10.3.5, ASCE 41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 389.0139  
lb/d = 0.30  
d1 = 44.00  
R = 250.00  
v = 0.00100813  
N = 7389.214  
Ac = 196349.541  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.1620683

Calculation of ratio lb/d

Inadequate Lap Length with lb/d = 0.30

Calculation of Mu2+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.4369E+008

= 0.80285146  
' = 0.71403327  
error of function (3.68), Biskinis Phd = 27987.158  
From 5A.2, TDY: fcc = fc\* c = 49.98874  
conf. factor c = 1.51481  
fc = 33.00  
From 10.3.5, ASCE 41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 389.0139  
lb/d = 0.30  
d1 = 44.00  
R = 250.00  
v = 0.00100813  
N = 7389.214  
Ac = 196349.541  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.1620683

Calculation of ratio lb/d

Inadequate Lap Length with lb/d = 0.30

Calculation of Mu2-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.4369E+008

= 0.80285146  
' = 0.71403327  
error of function (3.68), Biskinis Phd = 27987.158  
From 5A.2, TDY: fcc = fc\* c = 49.98874

conf. factor  $c = 1.51481$   
 $f_c = 33.00$   
 From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 389.0139$   
 $l_b/l_d = 0.30$   
 $d_1 = 44.00$   
 $R = 250.00$   
 $v = 0.00100813$   
 $N = 7389.214$   
 $A_c = 196349.541$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.1620683$

Calculation of ratio  $l_b/l_d$

Inadequate Lap Length with  $l_b/l_d = 0.30$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 705811.584$

Calculation of Shear Strength at edge 1,  $V_{r1} = 705811.584$

$V_{r1} = V_{\text{Col}} ((10.3), \text{ASCE 41-17}) = k_{nl} \cdot V_{\text{Col}0}$

$V_{\text{Col}0} = 705811.584$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

Mean concrete strength:  $f'_c = (f'_c_{\text{jacket}} \cdot \text{Area}_{\text{jacket}} + f'_c_{\text{core}} \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 33.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.5940012\text{E-}011$

$\nu_u = 2.1312935\text{E-}046$

$d = 0.8 \cdot D = 400.00$

$N_u = 7389.214$

$A_g = 196349.541$

From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 274157.871$

$V_{s1} = 274157.871$  is calculated for jacket, with:

$A_v = \cdot /2 \cdot A_{\text{stirrup}} = 123370.055$

$f_y = 555.56$

$s = 100.00$

$V_{s1}$  is multiplied by  $\text{Col}1 = 1.00$

$s/d = 0.25$

$V_{s2} = 0.00$  is calculated for core, with:

$A_v = \cdot /2 \cdot A_{\text{stirrup}} = 78956.835$

$f_y = 555.56$

$s = 250.00$

$V_{s2}$  is multiplied by  $\text{Col}2 = 0.00$

$s/d = 1.04167$

$V_f ((11-3)-(11.4), \text{ACI 440}) = 308320.00$

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression, where  $\alpha$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\alpha, \theta)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta_1 = b_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, \theta_1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{\text{Dir}} = 1.00$

$d_{fv} = d$  (figure 11.2, ACI 440) = 470.00

$f_{fe} ((11-5), \text{ACI 440}) = 328.00$

$E_f = 82000.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.009$

From (11-11), ACI 440:  $V_s + V_f \leq 479549.663$

$$b_w \cdot d = \frac{1}{4} \cdot d \cdot d = 125663.706$$

Calculation of Shear Strength at edge 2,  $V_{r2} = 705811.584$

$$V_{r2} = V_{Col} \text{ ((10.3), ASCE 41-17)} = k_{nl} \cdot V_{Col0}$$

$$V_{Col0} = 705811.584$$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

Mean concrete strength:  $f_c' = (f_c'_{jacket} \cdot Area_{jacket} + f_c'_{core} \cdot Area_{core}) / Area_{section} = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$$M / Vd = 2.00$$

$$\mu_u = 1.5940012E-011$$

$$\nu_u = 2.1312935E-046$$

$$d = 0.8 \cdot D = 400.00$$

$$N_u = 7389.214$$

$$A_g = 196349.541$$

From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 274157.871$

$V_{s1} = 274157.871$  is calculated for jacket, with:

$$A_v = \frac{1}{2} \cdot A_{stirrup} = 123370.055$$

$$f_y = 555.56$$

$$s = 100.00$$

$V_{s1}$  is multiplied by  $Col1 = 1.00$

$$s/d = 0.25$$

$V_{s2} = 0.00$  is calculated for core, with:

$$A_v = \frac{1}{2} \cdot A_{stirrup} = 78956.835$$

$$f_y = 555.56$$

$$s = 250.00$$

$V_{s2}$  is multiplied by  $Col2 = 0.00$

$$s/d = 1.04167$$

$$V_f \text{ ((11-3)-(11.4), ACI 440)} = 308320.00$$

$f = 0.95$ , for fully-wrapped sections

$w_f / s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression, where  $a$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = b1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, a1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{oDir} = 1.00$

$$d_{fv} = d \text{ (figure 11.2, ACI 440)} = 470.00$$

$$f_{fe} \text{ ((11-5), ACI 440)} = 328.00$$

$$E_f = 82000.00$$

$$f_e = 0.004, \text{ from (11.6a), ACI 440}$$

$$\text{with } f_u = 0.009$$

From (11-11), ACI 440:  $V_s + V_f \leq 479549.663$

$$b_w \cdot d = \frac{1}{4} \cdot d \cdot d = 125663.706$$

End Of Calculation of Shear Capacity ratio for element: column JCC1 of floor 1

At local axis: 2

Start Of Calculation of Chord Rotation Capacity for element: column JCC1 of floor 1

At local axis: 2

Integration Section: (a)

Section Type: rcjcs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE 41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Jacket

New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

Existing Column

New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

External Diameter,  $D = 500.00$

Internal Diameter,  $D = 300.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Inadequate Lap Length with  $l_b/l_d = 0.30$

FRP Wrapping Data

Type: Carbon

Dry properties (design values)

Thickness,  $t = 1.00$

Tensile Strength,  $f_{fu} = 840.00$

Tensile Modulus,  $E_f = 82000.00$

Elongation,  $ε_{fu} = 0.009$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i: 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

Stepwise Properties

Bending Moment,  $M = 1.1533666E-009$

Shear Force,  $V_2 = -7549.016$

Shear Force,  $V_3 = -3.4066156E-013$

Axial Force,  $F = -7386.194$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 1272.345$

-Compression:  $As_c = 1781.283$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t,ten} = 1017.876$

-Compression:  $As_{l,com} = 1017.876$

-Middle:  $As_{l,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $Db_L = 18.00$

New component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_R = 1.0^*$   $u = 0.03453062$

$u = y + p = 0.03453062$

- Calculation of  $y$  -

$y = (M_y * L_s / 3) / E_{eff} = 0.00453062$  ((4.29), Biskinis Phd))

$M_y = 2.2517E+008$

$L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 1500.00

From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 2.4850E+013$

factor = 0.30

$A_g = 196349.541$

Mean concrete strength:  $f_c' = (f_c'_{jacket} * Area_{jacket} + f_c'_{core} * Area_{core}) / Area_{section} = 33.00$

$N = 7386.194$

$$E_c I_g = E_{c\_jacket} I_{g\_jacket} + E_{c\_core} I_{g\_core} = 8.2833E+013$$

#### Calculation of Yielding Moment $M_y$

Calculation of  $\rho_y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$$M_y = \min(M_{y\_ten}, M_{y\_com}) = 2.2517E+008$$

$$\rho_y = 6.4424821E-006$$

$$M_{y\_ten} (8c) = 2.2517E+008$$

$$\rho_{y\_ten} (7c) = 61.81367$$

$$\text{error of function (7c)} = 0.00026388$$

$$M_{y\_com} (8d) = 8.1922E+008$$

$$\rho_{y\_com} (7d) = 63.67467$$

$$\text{error of function (7d)} = -0.00874883$$

$$\text{with } ((10.1), \text{ASCE 41-17}) \rho_y = \min(\rho_y, 1.25 \rho_y (I_b/I_d)^{2/3}) = 0.0027778$$

$$\rho_{eco} = 0.002$$

$$\rho_{apl} = 0.45 ((9c) \text{ in Biskinis and Fardis for FRP Wrap})$$

$$d_1 = 44.00$$

$$R = 250.00$$

$$v = 0.00101359$$

$$N = 7386.194$$

$$A_c = 196349.541$$

$$((10.1), \text{ASCE 41-17}) \rho_y = \min(\rho_y, 1.25 \rho_y (I_b/I_d)^{2/3}) = 0.23280363$$

$$\text{with } f_c^* ((12.3), \text{ACI 440}) = 37.11312$$

$$f_c = 33.00$$

$$f_l = 1.312$$

$$k = 1$$

$$\text{Effective FRP thickness, } t_f = N L^* t \cos(b_1) = 1.00$$

$$e_{fe} ((12.5) \text{ and } (12.7)) = 0.004$$

$$f_u = 0.009$$

$$E_f = 82000.00$$

#### Calculation of ratio $I_b/I_d$

Inadequate Lap Length with  $I_b/I_d = 0.30$

- Calculation of  $\rho_p$  -

From table 10-9:  $\rho_p = 0.03$

with:

- Columns controlled by inadequate development or splicing along the clear height because  $I_b/I_d < 1$

$$\text{shear control ratio } V_y E / V_{col} O E = 0.23017425$$

$$d = d_{\text{external}} = 0.00$$

$$s = s_{\text{external}} = 0.00$$

$$t = s_1 + s_2 + 2 t_f / b_w (f_{fe} / f_s) = 0.00795744$$

jacket:  $s_1 = A_{v1} (D_{c1} / 2) / (s_1 A_g) = 0.0027646$

$$A_{v1} = 78.53982, \text{ is the area of stirrup}$$

$$D_{c1} = D_{\text{ext}} - 2 \cdot \text{cover} - \text{External Hoop Diameter} = 440.00, \text{ is the total Length of all stirrups parallel to loading (shear) direction}$$

$$s_1 = 100.00$$

core:  $s_2 = A_{v2} (D_{c2} / 2) / (s_2 A_g) = 0.00046968$

$$A_{v2} = 50.26548, \text{ is the area of stirrup}$$

$$D_{c2} = D_{\text{int}} - \text{Internal Hoop Diameter} = 292.00, \text{ is the total Length of all stirrups parallel to loading (shear) direction}$$

$$s_2 = 250.00$$

The term  $2 t_f / b_w (f_{fe} / f_s)$  is implemented to account for FRP contribution where  $f = 2 t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength. All these variables have already been given in Shear control ratio calculation. For the normalisation  $f_s$  of jacket is used.

$$N U D = 7386.194$$

$$A_g = 196349.541$$

$$f_{cE} = (f_{c\_jacket} \cdot \text{Area}_{\text{jacket}} + f_{c\_core} \cdot \text{Area}_{\text{core}}) / \text{section\_area} = 33.00$$

$f_{yE} = (f_{y\_ext\_Long\_Reinf} \cdot Area\_ext\_Long\_Reinf + f_{y\_int\_Long\_Reinf} \cdot Area\_int\_Long\_Reinf) / Area\_Tot\_Long\_Rein = 21219958E-314$   
 $f_{tE} = (f_{y\_ext\_Trans\_Reinf} \cdot Area\_ext\_Trans\_Reinf + f_{y\_int\_Trans\_Reinf} \cdot Area\_int\_Trans\_Reinf) / Area\_Tot\_Trans\_Rein = 555.56$   
 $p_l = Area\_Tot\_Long\_Rein / (A_g) = 0.015552$   
 $f_{cE} = 33.00$

End Of Calculation of Chord Rotation Capacity for element: column JCC1 of floor 1

At local axis: 2

Integration Section: (a)

## Calculation No. 11

column C1, Floor 1

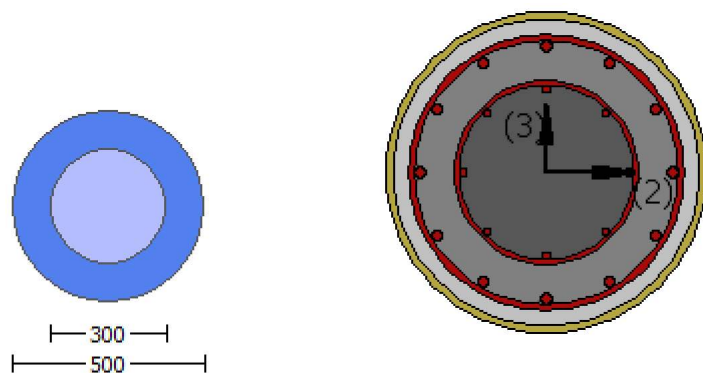
Limit State: Life Safety (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Shear capacity VRd

Edge: Start

Local Axis: (3)



Start Of Calculation of Shear Capacity for element: column JCC1 of floor 1

At local axis: 3

Integration Section: (a)

Section Type: rcjcs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE 41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Jacket

New material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 25.00$

New material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 500.00$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

```

Existing Column
New material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 25.00$ 
New material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 500.00$ 
Concrete Elasticity,  $E_c = 26999.444$ 
Steel Elasticity,  $E_s = 200000.00$ 
#####
Note: Especially for the calculation of  $\mu$  for displacement ductility demand,
the expected (mean value) strengths are used (7.5.1.3, ASCE 41-17) because bending is considered as
Deformation-Controlled Action (Table C7-1, ASCE 41-17).
Jacket
New material: Concrete Strength,  $f_c = f_{cm} = 33.00$ 
New material: Steel Strength,  $f_s = f_{sm} = 555.56$ 
Existing Column
New material: Concrete Strength,  $f_c = f_{cm} = 33.00$ 
New material: Steel Strength,  $f_s = f_{sm} = 555.56$ 
#####
External Diameter,  $D = 500.00$ 
Internal Diameter,  $D = 300.00$ 
Cover Thickness,  $c = 25.00$ 
Element Length,  $L = 3000.00$ 
Primary Member
Ribbed Bars
Ductile Steel
Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)
Longitudinal Bars With Ends Lapped Starting at the End Sections
Inadequate Lap Length with  $l_o/l_{ou,min} = l_b/l_d = 0.30$ 
FRP Wrapping Data
Type: Carbon
Dry properties (design values)
Thickness,  $t = 1.00$ 
Tensile Strength,  $f_{fu} = 840.00$ 
Tensile Modulus,  $E_f = 82000.00$ 
Elongation,  $e_{fu} = 0.009$ 
Number of directions,  $NoDir = 1$ 
Fiber orientations,  $bi: 0.00^\circ$ 
Number of layers,  $NL = 1$ 
Radius of rounding corners,  $R = 40.00$ 
-----

Stepwise Properties
-----
EDGE -A-
Bending Moment,  $M_a = 1.1533666E-009$ 
Shear Force,  $V_a = -3.4066156E-013$ 
EDGE -B-
Bending Moment,  $M_b = -1.3104847E-010$ 
Shear Force,  $V_b = 3.4066156E-013$ 
BOTH EDGES
Axial Force,  $F = -7386.194$ 
Longitudinal Reinforcement Area Distribution (in 2 divisions)
-Tension:  $As_t = 1272.345$ 
-Compression:  $As_c = 1781.283$ 
Longitudinal Reinforcement Area Distribution (in 3 divisions)
-Tension:  $As_{l,ten} = 1017.876$ 
-Compression:  $As_{l,com} = 1017.876$ 
-Middle:  $As_{l,mid} = 1017.876$ 
Mean Diameter of Tension Reinforcement,  $Db_{L,ten} = 18.00$ 
-----
-----

New component: From table 7-7, ASCE 41_17: Final Shear Capacity  $V_R = 1.0 \cdot V_n = 614519.737$ 
 $V_n ((10.3), ASCE 41-17) = knl \cdot V_{CoIO} = 614519.737$ 
 $V_{CoI} = 614519.737$ 
 $knl = 1.00$ 
displacement_ductility_demand = 0.00
-----

```



NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

Mean concrete strength:  $f'_c = (f'_c \cdot \text{Area}_{\text{jacket}} + f'_c \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 25.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.1533666E-009$

$V_u = 3.4066156E-013$

$d = 0.8 \cdot D = 400.00$

$N_u = 7386.194$

$A_g = 196349.541$

From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 246740.11$

$V_{s1} = 246740.11$  is calculated for jacket, with:

$A_v = \pi/2 \cdot A_{\text{stirrup}} = 123370.055$

$f_y = 500.00$

$s = 100.00$

$V_{s1}$  is multiplied by  $\text{Col1} = 1.00$

$s/d = 0.25$

$V_{s2} = 0.00$  is calculated for core, with:

$A_v = \pi/2 \cdot A_{\text{stirrup}} = 78956.835$

$f_y = 500.00$

$s = 250.00$

$V_{s2}$  is multiplied by  $\text{Col2} = 0.00$

$s/d = 1.04167$

$V_f$  ((11-3)-(11.4), ACI 440) = 308320.00

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression, where  $a$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta)$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = b1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, a1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{\text{Dir}} = 1.00$

$d_{fv} = d$  (figure 11.2, ACI 440) = 470.00

$f_{fe}$  ((11-5), ACI 440) = 328.00

$E_f = 82000.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.009$

From (11-11), ACI 440:  $V_s + V_f \leq 417394.406$

$b_w \cdot d = \pi \cdot d^2 / 4 = 125663.706$

displacement ductility demand is calculated as  $\delta / y$

- Calculation of  $\delta / y$  for END A -

for rotation axis 2 and integ. section (a)

From analysis, chord rotation  $\theta = 1.5106917E-020$

$y = (M_y \cdot L_s / 3) / E_{\text{eff}} = 0.00453062$  ((4.29), Biskinis Phd))

$M_y = 2.2517E+008$

$L_s = M/V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) = 1500.00

From table 10.5, ASCE 41\_17:  $E_{\text{eff}} = \text{factor} \cdot E_c \cdot I_g = 2.4850E+013$

factor = 0.30

$A_g = 196349.541$

Mean concrete strength:  $f'_c = (f'_c \cdot \text{Area}_{\text{jacket}} + f'_c \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 33.00$

$N = 7386.194$

$E_c \cdot I_g = E_c \cdot \text{Area}_{\text{jacket}} \cdot I_{g_{\text{jacket}}} + E_c \cdot \text{Area}_{\text{core}} \cdot I_{g_{\text{core}}} = 8.2833E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\delta$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 2.2517\text{E}+008$   
 $y = 6.4424821\text{E}-006$   
 $M_{y\_ten} (8c) = 2.2517\text{E}+008$   
 $_{ten} (7c) = 61.81367$   
error of function (7c) = 0.00026388  
 $M_{y\_com} (8d) = 8.1922\text{E}+008$   
 $_{com} (7d) = 63.67467$   
error of function (7d) = -0.00874883  
with ((10.1), ASCE 41-17)  $e_y = \text{Min}(e_y, 1.25 * e_y * (l_b / l_d)^{2/3}) = 0.0027778$   
 $e_{co} = 0.002$   
 $a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 250.00$   
 $v = 0.00101359$   
 $N = 7386.194$   
 $A_c = 196349.541$   
((10.1), ASCE 41-17)  $= \text{Min}( , 1.25 * (l_b / l_d)^{2/3}) = 0.23280363$   
with  $f_c^*$  ((12.3), ACI 440) = 37.11312  
 $f_c = 33.00$   
 $f_l = 1.312$   
 $k = 1$   
Effective FRP thickness,  $t_f = N L * t * \text{Cos}(b_1) = 1.00$   
 $e_{fe}$  ((12.5) and (12.7)) = 0.004  
 $f_u = 0.009$   
 $E_f = 82000.00$

Calculation of ratio  $l_b / l_d$

Inadequate Lap Length with  $l_b / l_d = 0.30$

End Of Calculation of Shear Capacity for element: column JCC1 of floor 1

At local axis: 3

Integration Section: (a)

## Calculation No. 12

column C1, Floor 1

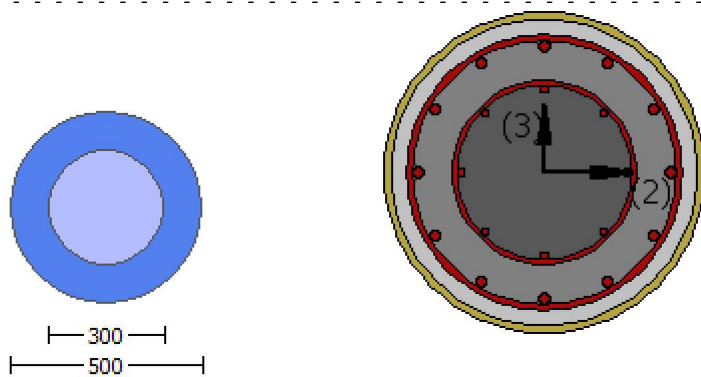
Limit State: Life Safety (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity (  $\theta_u$  )

Edge: Start

Local Axis: (3)



Start Of Calculation of Shear Capacity ratio for element: column JCC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rcjcs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Mean strength values are used for both shear and moment calculations.

Consequently:

Jacket

New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

Existing Column

New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Jacket

New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 694.45$

Existing Column

New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 694.45$

#####

External Diameter,  $D = 500.00$

Internal Diameter,  $D = 300.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.51481

Element Length,  $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Inadequate Lap Length with  $l_o/l_{ou,min} = 0.30$

FRP Wrapping Data

Type: Carbon

Dry properties (design values)

Thickness,  $t = 1.00$

Tensile Strength,  $f_{fu} = 840.00$

Tensile Modulus,  $E_f = 82000.00$

Elongation,  $ef_u = 0.009$

Number of directions,  $NoDir = 1$

Fiber orientations,  $bi: 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

## Stepwise Properties

At local axis: 3

EDGE -A-

Shear Force,  $V_a = 3.4807814E-030$

EDGE -B-

Shear Force,  $V_b = -3.4807814E-030$

BOTH EDGES

Axial Force,  $F = -7389.214$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 0.00$

-Compression:  $As_c = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t,ten} = 1017.876$

-Compression:  $As_{l,com} = 1017.876$

-Middle:  $As_{l,mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.23017425$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 162459.651$   
with

$M_{pr1} = \text{Max}(\mu_{1+}, \mu_{1-}) = 2.4369E+008$

$\mu_{1+} = 2.4369E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$\mu_{1-} = 2.4369E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(\mu_{2+}, \mu_{2-}) = 2.4369E+008$

$\mu_{2+} = 2.4369E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination

$\mu_{2-} = 2.4369E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

Calculation of  $\mu_{1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$

$\mu_u = 2.4369E+008$

$= 0.80285146$

$' = 0.71403327$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TDY:  $f_{cc} = f_c^* \quad c = 49.98874$

conf. factor  $c = 1.51481$

$f_c = 33.00$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 389.0139$

$l_b/l_d = 0.30$

$d_1 = 44.00$

$R = 250.00$

$v = 0.00100813$

$N = 7389.214$

$A_c = 196349.541$

$= \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.1620683$

Calculation of ratio  $l_b/l_d$

Inadequate Lap Length with  $l_b/l_d = 0.30$

Calculation of  $\mu_{1-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.4369E+008

= 0.80285146  
' = 0.71403327  
error of function (3.68), Biskinis Phd = 27987.158  
From 5A.2, TDY: fcc = fc\* c = 49.98874  
conf. factor c = 1.51481  
fc = 33.00  
From 10.3.5, ASCE 41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 389.0139  
lb/d = 0.30  
d1 = 44.00  
R = 250.00  
v = 0.00100813  
N = 7389.214  
Ac = 196349.541  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.1620683

Calculation of ratio lb/d

Inadequate Lap Length with lb/d = 0.30

Calculation of Mu2+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.4369E+008

= 0.80285146  
' = 0.71403327  
error of function (3.68), Biskinis Phd = 27987.158  
From 5A.2, TDY: fcc = fc\* c = 49.98874  
conf. factor c = 1.51481  
fc = 33.00  
From 10.3.5, ASCE 41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 389.0139  
lb/d = 0.30  
d1 = 44.00  
R = 250.00  
v = 0.00100813  
N = 7389.214  
Ac = 196349.541  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.1620683

Calculation of ratio lb/d

Inadequate Lap Length with lb/d = 0.30

Calculation of Mu2-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.4369E+008

= 0.80285146  
' = 0.71403327  
error of function (3.68), Biskinis Phd = 27987.158  
From 5A.2, TDY: fcc = fc\* c = 49.98874

conf. factor  $c = 1.51481$   
 $f_c = 33.00$   
 From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 389.0139$   
 $l_b/l_d = 0.30$   
 $d_1 = 44.00$   
 $R = 250.00$   
 $v = 0.00100813$   
 $N = 7389.214$   
 $A_c = 196349.541$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.1620683$

Calculation of ratio  $l_b/l_d$

Inadequate Lap Length with  $l_b/l_d = 0.30$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 705811.584$

Calculation of Shear Strength at edge 1,  $V_{r1} = 705811.584$

$V_{r1} = V_{\text{Col}} ((10.3), \text{ASCE 41-17}) = k_{nl} \cdot V_{\text{Col}0}$

$V_{\text{Col}0} = 705811.584$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

Mean concrete strength:  $f'_c = (f'_c_{\text{jacket}} \cdot \text{Area}_{\text{jacket}} + f'_c_{\text{core}} \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 33.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.2438899\text{E-}011$

$\nu_u = 3.4807814\text{E-}030$

$d = 0.8 \cdot D = 400.00$

$N_u = 7389.214$

$A_g = 196349.541$

From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 274157.871$

$V_{s1} = 274157.871$  is calculated for jacket, with:

$A_v = \cdot /2 \cdot A_{\text{stirrup}} = 123370.055$

$f_y = 555.56$

$s = 100.00$

$V_{s1}$  is multiplied by  $\text{Col}1 = 1.00$

$s/d = 0.25$

$V_{s2} = 0.00$  is calculated for core, with:

$A_v = \cdot /2 \cdot A_{\text{stirrup}} = 78956.835$

$f_y = 555.56$

$s = 250.00$

$V_{s2}$  is multiplied by  $\text{Col}2 = 0.00$

$s/d = 1.04167$

$V_f ((11-3)-(11.4), \text{ACI 440}) = 308320.00$

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression, where  $\alpha$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\alpha = 45^\circ$  and  $\alpha = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\alpha_1 = b_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \alpha_1)|, |V_f(-45, \alpha_1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{\text{Dir}} = 1.00$

$d_{fv} = d$  (figure 11.2, ACI 440) = 470.00

$f_{fe} ((11-5), \text{ACI 440}) = 328.00$

$E_f = 82000.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.009$

From (11-11), ACI 440:  $V_s + V_f \leq 479549.663$

$$b_w*d = *d*d/4 = 125663.706$$

Calculation of Shear Strength at edge 2,  $V_{r2} = 705811.584$

$$V_{r2} = V_{Col} \text{ ((10.3), ASCE 41-17)} = k_{nl}*V_{Col0}$$

$$V_{Col0} = 705811.584$$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v*f_y*d/s$ ' is replaced by ' $V_s + f*V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

Mean concrete strength:  $f_c' = (f_c'_{jacket}*Area_{jacket} + f_c'_{core}*Area_{core})/Area_{section} = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$$M/Vd = 2.00$$

$$\mu_u = 1.2438899E-011$$

$$\nu_u = 3.4807814E-030$$

$$d = 0.8*D = 400.00$$

$$N_u = 7389.214$$

$$A_g = 196349.541$$

From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 274157.871$

$V_{s1} = 274157.871$  is calculated for jacket, with:

$$A_v = /2*A_{stirrup} = 123370.055$$

$$f_y = 555.56$$

$$s = 100.00$$

$V_{s1}$  is multiplied by  $Col1 = 1.00$

$$s/d = 0.25$$

$V_{s2} = 0.00$  is calculated for core, with:

$$A_v = /2*A_{stirrup} = 78956.835$$

$$f_y = 555.56$$

$$s = 250.00$$

$V_{s2}$  is multiplied by  $Col2 = 0.00$

$$s/d = 1.04167$$

$$V_f \text{ ((11-3)-(11.4), ACI 440)} = 308320.00$$

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a)\sin a$  which is more a generalised expression, where  $a$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(, )$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $a = 45^\circ$  and  $a = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $a_1 = b_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, a_1)|, |V_f(-45, a_1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L*t/NoDir = 1.00$

$$d_{fv} = d \text{ (figure 11.2, ACI 440)} = 470.00$$

$$f_{fe} \text{ ((11-5), ACI 440)} = 328.00$$

$$E_f = 82000.00$$

$$f_e = 0.004, \text{ from (11.6a), ACI 440}$$

$$\text{with } f_u = 0.009$$

From (11-11), ACI 440:  $V_s + V_f \leq 479549.663$

$$b_w*d = *d*d/4 = 125663.706$$

End Of Calculation of Shear Capacity ratio for element: column JCC1 of floor 1

At local axis: 3

Start Of Calculation of Shear Capacity ratio for element: column JCC1 of floor 1

At Shear local axis: 2

(Bending local axis: 3)

Section Type: rcjcs

Constant Properties

Knowledge Factor,  $= 1.00$

Mean strength values are used for both shear and moment calculations.

```

Consequently:
Jacket
New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$ 
New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$ 
Concrete Elasticity,  $E_c = 26999.444$ 
Steel Elasticity,  $E_s = 200000.00$ 
Existing Column
New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$ 
New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$ 
Concrete Elasticity,  $E_c = 26999.444$ 
Steel Elasticity,  $E_s = 200000.00$ 
#####
Note: Especially for the calculation of moment strengths,
the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14
Jacket
New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 694.45$ 
Existing Column
New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 694.45$ 
#####
External Diameter,  $D = 500.00$ 
Internal Diameter,  $D = 300.00$ 
Cover Thickness,  $c = 25.00$ 
Mean Confinement Factor overall section = 1.51481
Element Length,  $L = 3000.00$ 
Primary Member
Ribbed Bars
Ductile Steel
Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)
Longitudinal Bars With Ends Lapped Starting at the End Sections
Inadequate Lap Length with  $l_o/l_{ou,min} = 0.30$ 
FRP Wrapping Data
Type: Carbon
Dry properties (design values)
Thickness,  $t = 1.00$ 
Tensile Strength,  $f_{fu} = 840.00$ 
Tensile Modulus,  $E_f = 82000.00$ 
Elongation,  $e_{fu} = 0.009$ 
Number of directions,  $NoDir = 1$ 
Fiber orientations,  $bi: 0.00^\circ$ 
Number of layers,  $NL = 1$ 
Radius of rounding corners,  $R = 40.00$ 
-----

Stepwise Properties
-----
At local axis: 2
EDGE -A-
Shear Force,  $V_a = -2.1312935E-046$ 
EDGE -B-
Shear Force,  $V_b = 2.1312935E-046$ 
BOTH EDGES
Axial Force,  $F = -7389.214$ 
Longitudinal Reinforcement Area Distribution (in 2 divisions)
-Tension:  $As_t = 0.00$ 
-Compression:  $As_c = 3053.628$ 
Longitudinal Reinforcement Area Distribution (in 3 divisions)
-Tension:  $As_{t,ten} = 1017.876$ 
-Compression:  $As_{l,com} = 1017.876$ 
-Middle:  $As_{l,mid} = 1017.876$ 
-----
-----

Calculation of Shear Capacity ratio ,  $V_e/V_r = 0.23017425$ 
Member Controlled by Flexure ( $V_e/V_r < 1$ )
Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 162459.651$ 
with

```



$$M_{pr1} = \text{Max}(M_{u1+}, M_{u1-}) = 2.4369\text{E}+008$$

$M_{u1+} = 2.4369\text{E}+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$M_{u1-} = 2.4369\text{E}+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$$M_{pr2} = \text{Max}(M_{u2+}, M_{u2-}) = 2.4369\text{E}+008$$

$M_{u2+} = 2.4369\text{E}+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination

$M_{u2-} = 2.4369\text{E}+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

Calculation of  $M_{u1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$   
 $M_u = 2.4369\text{E}+008$

$$= 0.80285146$$

$$' = 0.71403327$$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TDY:  $f_{cc} = f_c \cdot c = 49.98874$

conf. factor  $c = 1.51481$

$$f_c = 33.00$$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 389.0139$

$$l_b/d = 0.30$$

$$d_1 = 44.00$$

$$R = 250.00$$

$$v = 0.00100813$$

$$N = 7389.214$$

$$A_c = 196349.541$$

$$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.1620683$$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of  $M_{u1-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$   
 $M_u = 2.4369\text{E}+008$

$$= 0.80285146$$

$$' = 0.71403327$$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TDY:  $f_{cc} = f_c \cdot c = 49.98874$

conf. factor  $c = 1.51481$

$$f_c = 33.00$$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 389.0139$

$$l_b/d = 0.30$$

$$d_1 = 44.00$$

$$R = 250.00$$

$$v = 0.00100813$$

$$N = 7389.214$$

$$A_c = 196349.541$$

$$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.1620683$$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of Mu2+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.4369E+008

= 0.80285146  
' = 0.71403327  
error of function (3.68), Biskinis Phd = 27987.158  
From 5A.2, TBDY: fcc = fc\* c = 49.98874  
conf. factor c = 1.51481  
fc = 33.00  
From 10.3.5, ASCE 41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 389.0139  
lb/d = 0.30  
d1 = 44.00  
R = 250.00  
v = 0.00100813  
N = 7389.214  
Ac = 196349.541  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.1620683

Calculation of ratio lb/d

Inadequate Lap Length with lb/d = 0.30

Calculation of Mu2-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.4369E+008

= 0.80285146  
' = 0.71403327  
error of function (3.68), Biskinis Phd = 27987.158  
From 5A.2, TBDY: fcc = fc\* c = 49.98874  
conf. factor c = 1.51481  
fc = 33.00  
From 10.3.5, ASCE 41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 389.0139  
lb/d = 0.30  
d1 = 44.00  
R = 250.00  
v = 0.00100813  
N = 7389.214  
Ac = 196349.541  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.1620683

Calculation of ratio lb/d

Inadequate Lap Length with lb/d = 0.30

Calculation of Shear Strength Vr = Min(Vr1,Vr2) = 705811.584

Calculation of Shear Strength at edge 1, Vr1 = 705811.584  
Vr1 = VCol ((10.3), ASCE 41-17) = knl\*VColO  
VColO = 705811.584

knl = 1 (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

Mean concrete strength:  $f_c' = (f_c'_{\text{jacket}} \cdot \text{Area}_{\text{jacket}} + f_c'_{\text{core}} \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.5940012\text{E}-011$

$\nu_u = 2.1312935\text{E}-046$

$d = 0.8 \cdot D = 400.00$

$N_u = 7389.214$

$A_g = 196349.541$

From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 274157.871$

$V_{s1} = 274157.871$  is calculated for jacket, with:

$A_v = \frac{1}{2} \cdot A_{\text{stirrup}} = 123370.055$

$f_y = 555.56$

$s = 100.00$

$V_{s1}$  is multiplied by  $\text{Col1} = 1.00$

$s/d = 0.25$

$V_{s2} = 0.00$  is calculated for core, with:

$A_v = \frac{1}{2} \cdot A_{\text{stirrup}} = 78956.835$

$f_y = 555.56$

$s = 250.00$

$V_{s2}$  is multiplied by  $\text{Col2} = 0.00$

$s/d = 1.04167$

$V_f$  ((11-3)-(11.4), ACI 440) = 308320.00

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression, where  $a$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta)$ , is implemented for every different fiber orientation  $\theta_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta_1 = \theta_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, \theta_1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{\text{Dir}} = 1.00$

$d_{fv} = d$  (figure 11.2, ACI 440) = 470.00

$f_{fe}$  ((11-5), ACI 440) = 328.00

$E_f = 82000.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.009$

From (11-11), ACI 440:  $V_s + V_f \leq 479549.663$

$b_w \cdot d = \frac{1}{4} \cdot d \cdot d = 125663.706$

Calculation of Shear Strength at edge 2,  $V_{r2} = 705811.584$

$V_{r2} = V_{\text{Col}}((10.3), \text{ASCE 41-17}) = \text{knl} \cdot V_{\text{Col0}}$

$V_{\text{Col0}} = 705811.584$

knl = 1 (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

Mean concrete strength:  $f_c' = (f_c'_{\text{jacket}} \cdot \text{Area}_{\text{jacket}} + f_c'_{\text{core}} \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.5940012\text{E}-011$

$\nu_u = 2.1312935\text{E}-046$

$d = 0.8 \cdot D = 400.00$

$N_u = 7389.214$

$A_g = 196349.541$

From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 274157.871$

$V_{s1} = 274157.871$  is calculated for jacket, with:

$A_v = \frac{1}{2} \cdot A_{\text{stirrup}} = 123370.055$

$f_y = 555.56$

$s = 100.00$   
 $V_{s1}$  is multiplied by  $Col1 = 1.00$   
 $s/d = 0.25$   
 $V_{s2} = 0.00$  is calculated for core, with:  
 $A_v = \sqrt{2} \cdot A_{stirrup} = 78956.835$   
 $f_y = 555.56$   
 $s = 250.00$   
 $V_{s2}$  is multiplied by  $Col2 = 0.00$   
 $s/d = 1.04167$   
 $V_f ((11-3)-(11.4), ACI 440) = 308320.00$   
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = b1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, a1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{oDir} = 1.00$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 470.00  
 $f_{fe} ((11-5), ACI 440) = 328.00$   
 $E_f = 82000.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.009$   
 From (11-11), ACI 440:  $V_s + V_f \leq 479549.663$   
 $b_w \cdot d = \sqrt{V_s + V_f} \cdot d / 4 = 125663.706$

-----  
 End Of Calculation of Shear Capacity ratio for element: column JCC1 of floor 1  
 At local axis: 2  
 -----

Start Of Calculation of Chord Rotation Capacity for element: column JCC1 of floor 1  
 At local axis: 3  
 Integration Section: (a)  
 Section Type: rcjcs

Constant Properties

-----  
 Knowledge Factor,  $\phi = 1.00$   
 Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE 41-17.  
 Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17  
 Consequently:  
 Jacket  
 New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$   
 New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$   
 Concrete Elasticity,  $E_c = 26999.444$   
 Steel Elasticity,  $E_s = 200000.00$   
 Existing Column  
 New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$   
 New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$   
 Concrete Elasticity,  $E_c = 26999.444$   
 Steel Elasticity,  $E_s = 200000.00$   
 External Diameter,  $D = 500.00$   
 Internal Diameter,  $D = 300.00$   
 Cover Thickness,  $c = 25.00$   
 Element Length,  $L = 3000.00$   
 Primary Member  
 Ribbed Bars  
 Ductile Steel  
 Without Detailing for Earthquake Resistance (including stirrups not closed at  $135^\circ$ )  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Inadequate Lap Length with  $l_b/l_d = 0.30$   
 FRP Wrapping Data  
 Type: Carbon

Dry properties (design values)  
 Thickness,  $t = 1.00$   
 Tensile Strength,  $f_{fu} = 840.00$   
 Tensile Modulus,  $E_f = 82000.00$   
 Elongation,  $\epsilon_{fu} = 0.009$   
 Number of directions,  $N_{oDir} = 1$   
 Fiber orientations,  $b_i = 0.00^\circ$   
 Number of layers,  $N_L = 1$   
 Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

Bending Moment,  $M = -2.2653E+007$   
 Shear Force,  $V_2 = -7549.016$   
 Shear Force,  $V_3 = -3.4066156E-013$   
 Axial Force,  $F = -7386.194$   
 Longitudinal Reinforcement Area Distribution (in 2 divisions)  
   -Tension:  $A_{sl,t} = 1272.345$   
   -Compression:  $A_{sl,c} = 1781.283$   
 Longitudinal Reinforcement Area Distribution (in 3 divisions)  
   -Tension:  $A_{sl,ten} = 1017.876$   
   -Compression:  $A_{sl,com} = 1017.876$   
   -Middle:  $A_{sl,mid} = 1017.876$   
 Mean Diameter of Tension Reinforcement,  $D_{bL} = 18.00$

New component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $\phi_{u,R} = 1.0^*$   $\phi_u = 0.03906345$   
 $\phi_u = \phi_y + \phi_p = 0.03906345$

- Calculation of  $\phi_y$  -

$\phi_y = (M_y * L_s / 3) / E_{eff} = 0.00906345$  ((4.29), Biskinis Phd))  
 $M_y = 2.2517E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) =  $3000.736$   
 From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 2.4850E+013$   
 $factor = 0.30$   
 $A_g = 196349.541$   
 Mean concrete strength:  $f'_c = (f'_{c,jacket} * Area_{jacket} + f'_{c,core} * Area_{core}) / Area_{section} = 33.00$   
 $N = 7386.194$   
 $E_c * I_g = E_{c,jacket} * I_{g,jacket} + E_{c,core} * I_{g,core} = 8.2833E+013$

#### Calculation of Yielding Moment $M_y$

Calculation of  $\phi_y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \min(M_{y,ten}, M_{y,com}) = 2.2517E+008$   
 $\phi_y = 6.4424821E-006$   
 $M_{y,ten} (8c) = 2.2517E+008$   
 $\phi_{y,ten} (7c) = 61.81367$   
 error of function (7c) =  $0.00026388$   
 $M_{y,com} (8d) = 8.1922E+008$   
 $\phi_{y,com} (7d) = 63.67467$   
 error of function (7d) =  $-0.00874883$   
 with ((10.1), ASCE 41-17)  $\phi_y = \min(\phi_y, 1.25 * \phi_y * (I_b / I_d)^{2/3}) = 0.0027778$   
 $\epsilon_{co} = 0.002$   
 $\alpha_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 250.00$   
 $v = 0.00101359$   
 $N = 7386.194$   
 $A_c = 196349.541$

$((10.1), ASCE\ 41-17) = \text{Min}( , 1.25 * (lb/ld)^{2/3} ) = 0.23280363$   
 with  $f_c^* ((12.3), ACI\ 440) = 37.11312$   
 $f_c = 33.00$   
 $f_l = 1.312$   
 $k = 1$   
 Effective FRP thickness,  $t_f = NL * t * \cos(b_1) = 1.00$   
 $e_{fe} ((12.5) \text{ and } (12.7)) = 0.004$   
 $f_u = 0.009$   
 $E_f = 82000.00$

Calculation of ratio  $lb/ld$

Inadequate Lap Length with  $lb/ld = 0.30$

- Calculation of  $p$  -

From table 10-9:  $p = 0.03$

with:

- Columns controlled by inadequate development or splicing along the clear height because  $lb/ld < 1$

shear control ratio  $V_y E / V_{col} O E = 0.23017425$

$d = d_{\text{external}} = 0.00$

$s = s_{\text{external}} = 0.00$

$t = s_1 + s_2 + 2 * t_f / b_w * (f_{fe} / f_s) = 0.00795744$

jacket:  $s_1 = A_{v1} * (Dc_1 / 2) / (s_1 * A_g) = 0.0027646$

$A_{v1} = 78.53982$ , is the area of stirrup

$Dc_1 = D_{\text{ext}} - 2 * \text{cover} - \text{External Hoop Diameter} = 440.00$ , is the total Length of all stirrups parallel to loading (shear) direction

$s_1 = 100.00$

core:  $s_2 = A_{v2} * (Dc_2 / 2) / (s_2 * A_g) = 0.00046968$

$A_{v2} = 50.26548$ , is the area of stirrup

$Dc_2 = D_{\text{int}} - \text{Internal Hoop Diameter} = 292.00$ , is the total Length of all stirrups parallel to loading (shear) direction

$s_2 = 250.00$

The term  $2 * t_f / b_w * (f_{fe} / f_s)$  is implemented to account for FRP contribution

where  $f = 2 * t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength

All these variables have already been given in Shear control ratio calculation.

For the normalisation  $f_s$  of jacket is used.

$NUD = 7386.194$

$A_g = 196349.541$

$f_{cE} = (f_{c\_jacket} * \text{Area\_jacket} + f_{c\_core} * \text{Area\_core}) / \text{section\_area} = 33.00$

$f_{yIE} = (f_{y\_ext\_Long\_Reinf} * \text{Area\_ext\_Long\_Reinf} + f_{y\_int\_Long\_Reinf} * \text{Area\_int\_Long\_Reinf}) / \text{Area\_Tot\_Long\_Rein} = 21219958E-314$

$f_{yTE} = (f_{y\_ext\_Trans\_Reinf} * \text{Area\_ext\_Trans\_Reinf} + f_{y\_int\_Trans\_Reinf} * \text{Area\_int\_Trans\_Reinf}) / \text{Area\_Tot\_Trans\_Rein} = 555.56$

$p_l = \text{Area\_Tot\_Long\_Rein} / (A_g) = 0.015552$

$f_{cE} = 33.00$

End Of Calculation of Chord Rotation Capacity for element: column JCC1 of floor 1

At local axis: 3

Integration Section: (a)

## Calculation No. 13

column C1, Floor 1

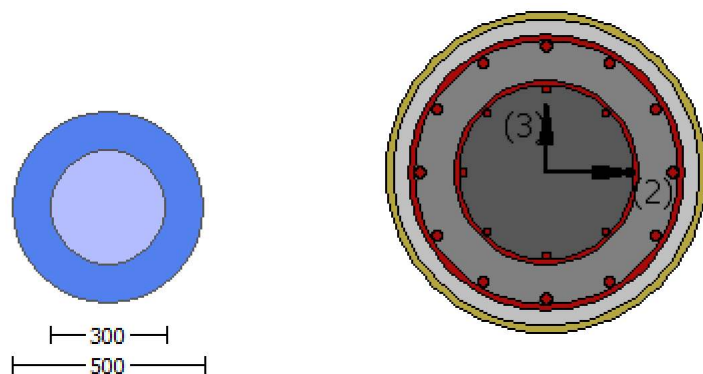
Limit State: Life Safety (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Shear capacity VRd

Edge: End

Local Axis: (2)



Start Of Calculation of Shear Capacity for element: column JCC1 of floor 1

At local axis: 2

Integration Section: (b)

Section Type: rcjcs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE 41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Jacket

New material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 25.00$

New material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 500.00$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

Existing Column

New material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 25.00$

New material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 500.00$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of  $\gamma$  for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE 41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE 41-17).

Jacket

New material: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material: Steel Strength,  $f_s = f_{sm} = 555.56$

Existing Column

New material: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material: Steel Strength,  $f_s = f_{sm} = 555.56$

#####

External Diameter,  $D = 500.00$

Internal Diameter,  $D = 300.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Inadequate Lap Length with  $l_o/l_{ou,min} = l_b/l_d = 0.30$   
 FRP Wrapping Data  
 Type: Carbon  
 Dry properties (design values)  
 Thickness,  $t = 1.00$   
 Tensile Strength,  $f_{fu} = 840.00$   
 Tensile Modulus,  $E_f = 82000.00$   
 Elongation,  $e_{fu} = 0.009$   
 Number of directions,  $NoDir = 1$   
 Fiber orientations,  $b_i: 0.00^\circ$   
 Number of layers,  $NL = 1$   
 Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

EDGE -A-  
 Bending Moment,  $M_a = -2.2653E+007$   
 Shear Force,  $V_a = -7549.016$   
 EDGE -B-  
 Bending Moment,  $M_b = 0.12607815$   
 Shear Force,  $V_b = 7549.016$   
 BOTH EDGES  
 Axial Force,  $F = -7386.194$   
 Longitudinal Reinforcement Area Distribution (in 2 divisions)  
   -Tension:  $As_t = 0.00$   
   -Compression:  $As_c = 3053.628$   
 Longitudinal Reinforcement Area Distribution (in 3 divisions)  
   -Tension:  $As_{t,ten} = 1017.876$   
   -Compression:  $As_{c,com} = 1017.876$   
   -Middle:  $As_{mid} = 1017.876$   
 Mean Diameter of Tension Reinforcement,  $Db_{L,ten} = 18.00$

New component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = 1.0 \cdot V_n = 614519.737$   
 $V_n ((10.3), ASCE 41-17) = knl \cdot V_{Col0} = 614519.737$   
 $V_{Col} = 614519.737$   
 $knl = 1.00$   
 $displacement\_ductility\_demand = 0.15088713$

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ '  
 where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)  
 Mean concrete strength:  $fc' = (fc'_{jacket} \cdot Area_{jacket} + fc'_{core} \cdot Area_{core}) / Area_{section} = 25.00$ , but  $fc'^{0.5} \leq 8.3$   
 MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $M_u = 0.12607815$   
 $V_u = 7549.016$   
 $d = 0.8 \cdot D = 400.00$   
 $N_u = 7386.194$   
 $Ag = 196349.541$   
 From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 246740.11$   
 $V_{s1} = 246740.11$  is calculated for jacket, with:  
 $A_v = /2 \cdot A_{stirrup} = 123370.055$   
 $f_y = 500.00$   
 $s = 100.00$   
 $V_{s1}$  is multiplied by  $Col1 = 1.00$   
 $s/d = 0.25$   
 $V_{s2} = 0.00$  is calculated for core, with:  
 $A_v = /2 \cdot A_{stirrup} = 78956.835$   
 $f_y = 500.00$   
 $s = 250.00$



$V_{s2}$  is multiplied by  $Col2 = 0.00$   
 $s/d = 1.04167$   
 $V_f((11-3)-(11.4), ACI 440) = 308320.00$   
 $f = 0.95$ , for fully-wrapped sections  
 $wf/sf = 1$  (FRP strips adjacent to one another).  
 $\ln(11.3) \sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(, )$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $= 45^\circ$  and  $= -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $b_1 = b_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, a_1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = NL * t / NoDir = 1.00$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 470.00  
 $f_{fe}((11-5), ACI 440) = 328.00$   
 $E_f = 82000.00$   
 $f_{e} = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.009$   
 From (11-11), ACI 440:  $V_s + V_f \leq 417394.406$   
 $b_w * d = *d * d / 4 = 125663.706$

displacement\_ductility\_demand is calculated as  $\delta / y$

- Calculation of  $\delta / y$  for END B -  
 for rotation axis 3 and integ. section (b)

From analysis, chord rotation  $= 0.00013672$   
 $y = (M_y * L_s / 3) / E_{eff} = 0.00090612$  ((4.29), Biskinis Phd))  
 $M_y = 2.2517E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 300.00  
 From table 10.5, ASCE 41\_17:  $E_{eff} = \text{factor} * E_c * I_g = 2.4850E+013$   
 $\text{factor} = 0.30$   
 $A_g = 196349.541$   
 Mean concrete strength:  $f'_c = (f'_c_{\text{jacket}} * \text{Area}_{\text{jacket}} + f'_c_{\text{core}} * \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 33.00$   
 $N = 7386.194$   
 $E_c * I_g = E_{c_{\text{jacket}}} * I_{g_{\text{jacket}}} + E_{c_{\text{core}}} * I_{g_{\text{core}}} = 8.2833E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\delta$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y_{\text{ten}}}, M_{y_{\text{com}}}) = 2.2517E+008$   
 $y = 6.4424821E-006$   
 $M_{y_{\text{ten}}} (8c) = 2.2517E+008$   
 $_{\text{ten}} (7c) = 61.81367$   
 error of function (7c) = 0.00026388  
 $M_{y_{\text{com}}} (8d) = 8.1922E+008$   
 $_{\text{com}} (7d) = 63.67467$   
 error of function (7d) = -0.00874883  
 with ((10.1), ASCE 41-17)  $e_y = \text{Min}(e_y, 1.25 * e_y * (I_b / I_d)^{2/3}) = 0.0027778$   
 $e_{co} = 0.002$   
 $a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 250.00$   
 $v = 0.00101359$   
 $N = 7386.194$   
 $A_c = 196349.541$   
 ((10.1), ASCE 41-17)  $= \text{Min}(, 1.25 * * (I_b / I_d)^{2/3}) = 0.23280363$   
 with  $f'_c$  ((12.3), ACI 440) = 37.11312  
 $f_c = 33.00$   
 $f_l = 1.312$   
 $k = 1$   
 Effective FRP thickness,  $t_f = NL * t * \cos(b_1) = 1.00$   
 $e_{fe}((12.5) \text{ and } (12.7)) = 0.004$

$f_u = 0.009$   
 $E_f = 82000.00$

Calculation of ratio  $I_b/I_d$

Inadequate Lap Length with  $I_b/I_d = 0.30$

End Of Calculation of Shear Capacity for element: column JCC1 of floor 1

At local axis: 2

Integration Section: (b)

## Calculation No. 14

column C1, Floor 1

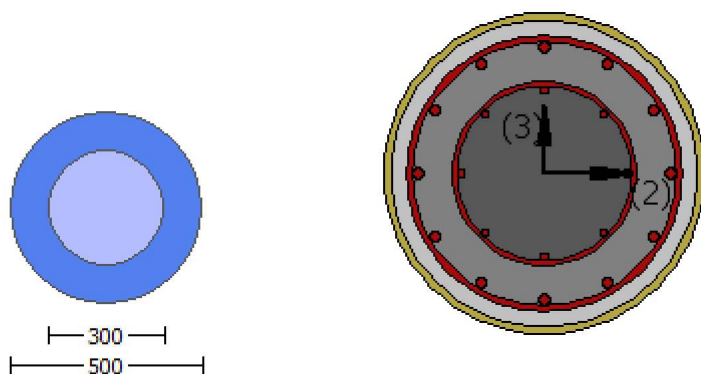
Limit State: Life Safety (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity (  $\phi$  )

Edge: End

Local Axis: (2)



Start Of Calculation of Shear Capacity ratio for element: column JCC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rcjcs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Mean strength values are used for both shear and moment calculations.

Consequently:

Jacket

New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

Existing Column  
 New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$   
 New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$   
 Concrete Elasticity,  $E_c = 26999.444$   
 Steel Elasticity,  $E_s = 200000.00$   
 #####  
 Note: Especially for the calculation of moment strengths,  
 the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14  
 Jacket  
 New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 694.45$   
 Existing Column  
 New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 694.45$   
 #####  
 External Diameter,  $D = 500.00$   
 Internal Diameter,  $D = 300.00$   
 Cover Thickness,  $c = 25.00$   
 Mean Confinement Factor overall section = 1.51481  
 Element Length,  $L = 3000.00$   
 Primary Member  
 Ribbed Bars  
 Ductile Steel  
 Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Inadequate Lap Length with  $l_o/l_{ou,min} = 0.30$   
 FRP Wrapping Data  
 Type: Carbon  
 Dry properties (design values)  
 Thickness,  $t = 1.00$   
 Tensile Strength,  $f_{fu} = 840.00$   
 Tensile Modulus,  $E_f = 82000.00$   
 Elongation,  $\epsilon_{fu} = 0.009$   
 Number of directions,  $N_{oDir} = 1$   
 Fiber orientations,  $b_i: 0.00^\circ$   
 Number of layers,  $N_L = 1$   
 Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

At local axis: 3  
 EDGE -A-  
 Shear Force,  $V_a = 3.4807814E-030$   
 EDGE -B-  
 Shear Force,  $V_b = -3.4807814E-030$   
 BOTH EDGES  
 Axial Force,  $F = -7389.214$   
 Longitudinal Reinforcement Area Distribution (in 2 divisions)  
 -Tension:  $A_{sl,t} = 0.00$   
 -Compression:  $A_{sl,c} = 3053.628$   
 Longitudinal Reinforcement Area Distribution (in 3 divisions)  
 -Tension:  $A_{sl,ten} = 1017.876$   
 -Compression:  $A_{sl,com} = 1017.876$   
 -Middle:  $A_{sl,mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.23017425$   
 Member Controlled by Flexure ( $V_e/V_r < 1$ )  
 Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 162459.651$   
 with  
 $M_{pr1} = \text{Max}(M_{u1+}, M_{u1-}) = 2.4369E+008$   
 $M_{u1+} = 2.4369E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction  
 which is defined for the static loading combination  
 $M_{u1-} = 2.4369E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment  
 direction which is defined for the static loading combination  
 $M_{pr2} = \text{Max}(M_{u2+}, M_{u2-}) = 2.4369E+008$

Mu2+ = 2.4369E+008, is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

Mu2- = 2.4369E+008, is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

-----  
Calculation of Mu1+  
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.4369E+008  
-----

= 0.80285146  
' = 0.71403327  
error of function (3.68), Biskinis Phd = 27987.158  
From 5A.2, TDY: fcc = fc\* c = 49.98874  
conf. factor c = 1.51481  
fc = 33.00  
From 10.3.5, ASCE 41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 389.0139  
lb/d = 0.30  
d1 = 44.00  
R = 250.00  
v = 0.00100813  
N = 7389.214  
Ac = 196349.541  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.1620683  
-----

Calculation of ratio lb/d  
-----

Inadequate Lap Length with lb/d = 0.30  
-----  
-----  
-----

Calculation of Mu1-  
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.4369E+008  
-----

= 0.80285146  
' = 0.71403327  
error of function (3.68), Biskinis Phd = 27987.158  
From 5A.2, TDY: fcc = fc\* c = 49.98874  
conf. factor c = 1.51481  
fc = 33.00  
From 10.3.5, ASCE 41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 389.0139  
lb/d = 0.30  
d1 = 44.00  
R = 250.00  
v = 0.00100813  
N = 7389.214  
Ac = 196349.541  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.1620683  
-----

Calculation of ratio lb/d  
-----

Inadequate Lap Length with lb/d = 0.30  
-----  
-----  
-----

Calculation of Mu2+  
-----

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.4369E+008

$$= 0.80285146$$

$$' = 0.71403327$$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TBDY: fcc = fc\* c = 49.98874

conf. factor c = 1.51481

$$fc = 33.00$$

From 10.3.5, ASCE 41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 389.0139

$$lb/d = 0.30$$

$$d1 = 44.00$$

$$R = 250.00$$

$$v = 0.00100813$$

$$N = 7389.214$$

$$Ac = 196349.541$$

$$= *Min(1,1.25*(lb/d)^ 2/3) = 0.1620683$$

Calculation of ratio lb/d

Inadequate Lap Length with lb/d = 0.30

Calculation of Mu2-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.4369E+008

$$= 0.80285146$$

$$' = 0.71403327$$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TBDY: fcc = fc\* c = 49.98874

conf. factor c = 1.51481

$$fc = 33.00$$

From 10.3.5, ASCE 41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 389.0139

$$lb/d = 0.30$$

$$d1 = 44.00$$

$$R = 250.00$$

$$v = 0.00100813$$

$$N = 7389.214$$

$$Ac = 196349.541$$

$$= *Min(1,1.25*(lb/d)^ 2/3) = 0.1620683$$

Calculation of ratio lb/d

Inadequate Lap Length with lb/d = 0.30

Calculation of Shear Strength Vr = Min(Vr1,Vr2) = 705811.584

Calculation of Shear Strength at edge 1, Vr1 = 705811.584

Vr1 = VCol ((10.3), ASCE 41-17) = knl\*VColO

$$VColO = 705811.584$$

knl = 1 (zero step-static loading)

NOTE: In expression (10-3) 'Vs = Av\*fy\*d/s' is replaced by 'Vs+ f\*VF'  
where Vf is the contribution of FRPs (11.3), ACI 440).

$$= 1 \text{ (normal-weight concrete)}$$

Mean concrete strength:  $f_c' = (f_c'_{\text{jacket}} \cdot \text{Area}_{\text{jacket}} + f_c'_{\text{core}} \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.2438899\text{E-}011$

$\nu_u = 3.4807814\text{E-}030$

$d = 0.8 \cdot D = 400.00$

$N_u = 7389.214$

$A_g = 196349.541$

From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 274157.871$

$V_{s1} = 274157.871$  is calculated for jacket, with:

$A_v = \frac{1}{2} \cdot A_{\text{stirrup}} = 123370.055$

$f_y = 555.56$

$s = 100.00$

$V_{s1}$  is multiplied by  $\text{Col1} = 1.00$

$s/d = 0.25$

$V_{s2} = 0.00$  is calculated for core, with:

$A_v = \frac{1}{2} \cdot A_{\text{stirrup}} = 78956.835$

$f_y = 555.56$

$s = 250.00$

$V_{s2}$  is multiplied by  $\text{Col2} = 0.00$

$s/d = 1.04167$

$V_f ((11-3)-(11.4), \text{ACI } 440) = 308320.00$

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression, where  $a$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta)$ , is implemented for every different fiber orientation  $\theta_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = \theta_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \theta)|)$ , with:

total thickness per orientation,  $t_f = N_L \cdot t / N_{\text{Dir}} = 1.00$

$d_{fv} = d$  (figure 11.2, ACI 440) = 470.00

$f_{fe} ((11-5), \text{ACI } 440) = 328.00$

$E_f = 82000.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.009$

From (11-11), ACI 440:  $V_s + V_f \leq 479549.663$

$b_w \cdot d = \frac{1}{4} \cdot d \cdot d = 125663.706$

Calculation of Shear Strength at edge 2,  $V_{r2} = 705811.584$

$V_{r2} = V_{\text{Col}} ((10.3), \text{ASCE } 41-17) = k_n l \cdot V_{\text{ColO}}$

$V_{\text{ColO}} = 705811.584$

$k_n l = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\mu_u = 1$  (normal-weight concrete)

Mean concrete strength:  $f_c' = (f_c'_{\text{jacket}} \cdot \text{Area}_{\text{jacket}} + f_c'_{\text{core}} \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.2438899\text{E-}011$

$\nu_u = 3.4807814\text{E-}030$

$d = 0.8 \cdot D = 400.00$

$N_u = 7389.214$

$A_g = 196349.541$

From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 274157.871$

$V_{s1} = 274157.871$  is calculated for jacket, with:

$A_v = \frac{1}{2} \cdot A_{\text{stirrup}} = 123370.055$

$f_y = 555.56$

$s = 100.00$

$V_{s1}$  is multiplied by  $\text{Col1} = 1.00$

$s/d = 0.25$

$V_{s2} = 0.00$  is calculated for core, with:

$A_v = \frac{1}{2} \cdot A_{\text{stirrup}} = 78956.835$

$f_y = 555.56$

$s = 250.00$   
 $V_{s2}$  is multiplied by  $Col2 = 0.00$   
 $s/d = 1.04167$   
 $V_f((11-3)-(11.4), ACI 440) = 308320.00$   
 $f = 0.95$ , for fully-wrapped sections  
 $wf/sf = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a)\sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f( , )$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $a = 45^\circ$  and  $a = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta_1 = \theta_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, a_1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = NL * t / NoDir = 1.00$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 470.00  
 $f_{fe}((11-5), ACI 440) = 328.00$   
 $E_f = 82000.00$   
 $f_{e1} = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.009$   
 From (11-11), ACI 440:  $V_s + V_f \leq 479549.663$   
 $b_w * d = \rho * d^2 / 4 = 125663.706$

-----  
 End Of Calculation of Shear Capacity ratio for element: column JCC1 of floor 1  
 At local axis: 3  
 -----

-----  
 Start Of Calculation of Shear Capacity ratio for element: column JCC1 of floor 1  
 At Shear local axis: 2  
 (Bending local axis: 3)  
 Section Type: rcjcs

Constant Properties

-----  
 Knowledge Factor,  $\phi = 1.00$   
 Mean strength values are used for both shear and moment calculations.  
 Consequently:  
 Jacket  
 New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$   
 New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$   
 Concrete Elasticity,  $E_c = 26999.444$   
 Steel Elasticity,  $E_s = 200000.00$   
 Existing Column  
 New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$   
 New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$   
 Concrete Elasticity,  $E_c = 26999.444$   
 Steel Elasticity,  $E_s = 200000.00$   
 #####  
 Note: Especially for the calculation of moment strengths,  
 the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14  
 Jacket  
 New material: Steel Strength,  $f_s = 1.25 * f_{sm} = 694.45$   
 Existing Column  
 New material: Steel Strength,  $f_s = 1.25 * f_{sm} = 694.45$   
 #####  
 External Diameter,  $D = 500.00$   
 Internal Diameter,  $D = 300.00$   
 Cover Thickness,  $c = 25.00$   
 Mean Confinement Factor overall section = 1.51481  
 Element Length,  $L = 3000.00$   
 Primary Member  
 Ribbed Bars  
 Ductile Steel  
 Without Detailing for Earthquake Resistance (including stirrups not closed at  $135^\circ$ )  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Inadequate Lap Length with  $l_o/l_{ou, min} = 0.30$

#### FRP Wrapping Data

Type: Carbon

Dry properties (design values)

Thickness,  $t = 1.00$

Tensile Strength,  $f_{fu} = 840.00$

Tensile Modulus,  $E_f = 82000.00$

Elongation,  $\epsilon_{fu} = 0.009$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

At local axis: 2

EDGE -A-

Shear Force,  $V_a = -2.1312935E-046$

EDGE -B-

Shear Force,  $V_b = 2.1312935E-046$

BOTH EDGES

Axial Force,  $F = -7389.214$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{sl,t} = 0.00$

-Compression:  $A_{sl,c} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{sl,ten} = 1017.876$

-Compression:  $A_{sl,com} = 1017.876$

-Middle:  $A_{sl,mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.23017425$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 162459.651$  with

$M_{pr1} = \max(\mu_{u1+}, \mu_{u1-}) = 2.4369E+008$

$\mu_{u1+} = 2.4369E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$\mu_{u1-} = 2.4369E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \max(\mu_{u2+}, \mu_{u2-}) = 2.4369E+008$

$\mu_{u2+} = 2.4369E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination

$\mu_{u2-} = 2.4369E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

#### Calculation of $\mu_{u1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$

$\mu_u = 2.4369E+008$

$\beta_1 = 0.80285146$

$\beta_2 = 0.71403327$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 49.98874$

conf. factor  $c = 1.51481$

$f_c = 33.00$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \min(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 389.0139$

$l_b/l_d = 0.30$

$d_1 = 44.00$

$R = 250.00$

$v = 0.00100813$



$$N = 7389.214$$

$$Ac = 196349.541$$

$$= *Min(1, 1.25 * (lb/d)^{2/3}) = 0.1620683$$

Calculation of ratio lb/d

Inadequate Lap Length with lb/d = 0.30

Calculation of Mu1-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.4369E+008

$$= 0.80285146$$

$$' = 0.71403327$$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TBDY: fcc = fc\* c = 49.98874

conf. factor c = 1.51481

fc = 33.00

From 10.3.5, ASCE 41-17, Final value of fy: fy\*Min(1, 1.25\*(lb/d)^{2/3}) = 389.0139

lb/d = 0.30

d1 = 44.00

R = 250.00

v = 0.00100813

N = 7389.214

Ac = 196349.541

$$= *Min(1, 1.25 * (lb/d)^{2/3}) = 0.1620683$$

Calculation of ratio lb/d

Inadequate Lap Length with lb/d = 0.30

Calculation of Mu2+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.4369E+008

$$= 0.80285146$$

$$' = 0.71403327$$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TBDY: fcc = fc\* c = 49.98874

conf. factor c = 1.51481

fc = 33.00

From 10.3.5, ASCE 41-17, Final value of fy: fy\*Min(1, 1.25\*(lb/d)^{2/3}) = 389.0139

lb/d = 0.30

d1 = 44.00

R = 250.00

v = 0.00100813

N = 7389.214

Ac = 196349.541

$$= *Min(1, 1.25 * (lb/d)^{2/3}) = 0.1620683$$

Calculation of ratio lb/d

Inadequate Lap Length with lb/d = 0.30

## Calculation of Mu2-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.4369E+008

$$= 0.80285146$$

$$\rho = 0.71403327$$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 49.98874$

conf. factor  $c = 1.51481$

$$f_c = 33.00$$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 389.0139$

$$l_b/d = 0.30$$

$$d_1 = 44.00$$

$$R = 250.00$$

$$v = 0.00100813$$

$$N = 7389.214$$

$$A_c = 196349.541$$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.1620683$$

## Calculation of ratio lb/d

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 705811.584$

Calculation of Shear Strength at edge 1,  $V_{r1} = 705811.584$

$V_{r1} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{Col0}$

$$V_{Col0} = 705811.584$$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

Mean concrete strength:  $f_c' = (f_c'_{\text{jacket}} \cdot \text{Area}_{\text{jacket}} + f_c'_{\text{core}} \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$$M/Vd = 2.00$$

$$Mu = 1.5940012E-011$$

$$Vu = 2.1312935E-046$$

$$d = 0.8 \cdot D = 400.00$$

$$Nu = 7389.214$$

$$Ag = 196349.541$$

From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 274157.871$

$V_{s1} = 274157.871$  is calculated for jacket, with:

$$A_v = \frac{1}{2} \cdot A_{\text{stirup}} = 123370.055$$

$$f_y = 555.56$$

$$s = 100.00$$

$V_{s1}$  is multiplied by  $Col1 = 1.00$

$$s/d = 0.25$$

$V_{s2} = 0.00$  is calculated for core, with:

$$A_v = \frac{1}{2} \cdot A_{\text{stirup}} = 78956.835$$

$$f_y = 555.56$$

$$s = 250.00$$

$V_{s2}$  is multiplied by  $Col2 = 0.00$

$$s/d = 1.04167$$

$V_f$  ((11-3)-(11.4), ACI 440) = 308320.00

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \cot \alpha) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE).  
This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
orientation 1:  $\theta_1 = \theta_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, \alpha_1)|)$ , with:  
total thickness per orientation,  $t_{f1} = N_L \cdot t / \text{NoDir} = 1.00$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 470.00  
 $f_{fe}((11-5), \text{ACI 440}) = 328.00$   
 $E_f = 82000.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
with  $f_u = 0.009$   
From (11-11), ACI 440:  $V_s + V_f \leq 479549.663$   
 $b_w \cdot d = \frac{A_s \cdot d}{4} = 125663.706$

Calculation of Shear Strength at edge 2,  $V_{r2} = 705811.584$   
 $V_{r2} = V_{Col}((10.3), \text{ASCE 41-17}) = k_n l \cdot V_{Col0}$   
 $V_{Col0} = 705811.584$   
 $k_n l = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\lambda = 1$  (normal-weight concrete)  
Mean concrete strength:  $f'_c = (f'_c \cdot \text{jacket} \cdot \text{Area}_{\text{jacket}} + f'_c \cdot \text{core} \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 33.00$ , but  $f'_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $\mu_u = 1.5940012E-011$   
 $\nu_u = 2.1312935E-046$   
 $d = 0.8 \cdot D = 400.00$   
 $N_u = 7389.214$   
 $A_g = 196349.541$   
From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 274157.871$   
 $V_{s1} = 274157.871$  is calculated for jacket, with:  
 $A_v = \frac{A_s}{2} = 123370.055$   
 $f_y = 555.56$   
 $s = 100.00$   
 $V_{s1}$  is multiplied by  $\text{Col1} = 1.00$   
 $s/d = 0.25$   
 $V_{s2} = 0.00$  is calculated for core, with:  
 $A_v = \frac{A_s}{2} = 78956.835$   
 $f_y = 555.56$   
 $s = 250.00$   
 $V_{s2}$  is multiplied by  $\text{Col2} = 0.00$   
 $s/d = 1.04167$   
 $V_f((11-3)-(11.4), \text{ACI 440}) = 308320.00$   
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \cot \alpha) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE).  
This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
orientation 1:  $\theta_1 = \theta_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, \alpha_1)|)$ , with:  
total thickness per orientation,  $t_{f1} = N_L \cdot t / \text{NoDir} = 1.00$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 470.00  
 $f_{fe}((11-5), \text{ACI 440}) = 328.00$   
 $E_f = 82000.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
with  $f_u = 0.009$   
From (11-11), ACI 440:  $V_s + V_f \leq 479549.663$   
 $b_w \cdot d = \frac{A_s \cdot d}{4} = 125663.706$

End Of Calculation of Shear Capacity ratio for element: column JCC1 of floor 1  
At local axis: 2

Start Of Calculation of Chord Rotation Capacity for element: column JCC1 of floor 1

At local axis: 2

Integration Section: (b)

Section Type: rcjcs

#### Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE 41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Jacket

New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

Existing Column

New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

External Diameter,  $D = 500.00$

Internal Diameter,  $D = 300.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Inadequate Lap Length with  $l_b/l_d = 0.30$

FRP Wrapping Data

Type: Carbon

Dry properties (design values)

Thickness,  $t = 1.00$

Tensile Strength,  $f_{fu} = 840.00$

Tensile Modulus,  $E_f = 82000.00$

Elongation,  $ε_{fu} = 0.009$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $bi = 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

Bending Moment,  $M = -1.3104847E-010$

Shear Force,  $V_2 = 7549.016$

Shear Force,  $V_3 = 3.4066156E-013$

Axial Force,  $F = -7386.194$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 0.00$

-Compression:  $As_c = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{l,ten} = 1017.876$

-Compression:  $As_{l,com} = 1017.876$

-Middle:  $As_{l,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $Db_L = 18.00$

New component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_{,R} = 1.0^* u = 0.03453062$

$$u = y + p = 0.03453062$$

- Calculation of  $y$  -

$$y = (M_y \cdot L_s / 3) / E_{\text{eff}} = 0.00453062 \text{ ((4.29), Biskinis Phd)}$$

$$M_y = 2.2517\text{E}+008$$

$$L_s = M/V \text{ (with } L_s > 0.1 \cdot L \text{ and } L_s < 2 \cdot L) = 1500.00$$

$$\text{From table 10.5, ASCE 41-17: } E_{\text{eff}} = \text{factor} \cdot E_c \cdot I_g = 2.4850\text{E}+013$$

$$\text{factor} = 0.30$$

$$A_g = 196349.541$$

$$\text{Mean concrete strength: } f_c' = (f_c'_{\text{jacket}} \cdot \text{Area}_{\text{jacket}} + f_c'_{\text{core}} \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 33.00$$

$$N = 7386.194$$

$$E_c \cdot I_g = E_{c_{\text{jacket}}} \cdot I_{g_{\text{jacket}}} + E_{c_{\text{core}}} \cdot I_{g_{\text{core}}} = 8.2833\text{E}+013$$

Calculation of Yielding Moment  $M_y$

Calculation of  $y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$$M_y = \text{Min}(M_{y_{\text{ten}}}, M_{y_{\text{com}}}) = 2.2517\text{E}+008$$

$$y = 6.4424821\text{E}-006$$

$$M_{y_{\text{ten}}} (8c) = 2.2517\text{E}+008$$

$$_{\text{ten}} (7c) = 61.81367$$

$$\text{error of function (7c)} = 0.00026388$$

$$M_{y_{\text{com}}} (8d) = 8.1922\text{E}+008$$

$$_{\text{com}} (7d) = 63.67467$$

$$\text{error of function (7d)} = -0.00874883$$

$$\text{with ((10.1), ASCE 41-17) } e_y = \text{Min}(e_y, 1.25 \cdot e_y \cdot (I_b / I_d)^{2/3}) = 0.0027778$$

$$e_{co} = 0.002$$

$$a_{pl} = 0.45 \text{ ((9c) in Biskinis and Fardis for FRP Wrap)}$$

$$d_1 = 44.00$$

$$R = 250.00$$

$$v = 0.00101359$$

$$N = 7386.194$$

$$A_c = 196349.541$$

$$((10.1), \text{ASCE 41-17}) = \text{Min}( , 1.25 \cdot (I_b / I_d)^{2/3}) = 0.23280363$$

$$\text{with } f_c' \text{ ((12.3), ACI 440)} = 37.11312$$

$$f_c = 33.00$$

$$f_l = 1.312$$

$$k = 1$$

$$\text{Effective FRP thickness, } t_f = N L \cdot t \cdot \cos(b_1) = 1.00$$

$$e_{fe} \text{ ((12.5) and (12.7))} = 0.004$$

$$f_u = 0.009$$

$$E_f = 82000.00$$

Calculation of ratio  $I_b / I_d$

Inadequate Lap Length with  $I_b / I_d = 0.30$

- Calculation of  $p$  -

From table 10-9:  $p = 0.03$

with:

- Columns controlled by inadequate development or splicing along the clear height because  $I_b / I_d < 1$

$$\text{shear control ratio } V_y E / V_{co} I_{OE} = 0.23017425$$

$$d = d_{\text{external}} = 0.00$$

$$s = s_{\text{external}} = 0.00$$

$$t = s_1 + s_2 + 2 \cdot t_f / b_w \cdot (f_{fe} / f_s) = 0.00795744$$

$$\text{jacket: } s_1 = A_{v1} \cdot (D_{c1} / 2) / (s_1 \cdot A_g) = 0.0027646$$

$$A_{v1} = 78.53982, \text{ is the area of stirrup}$$

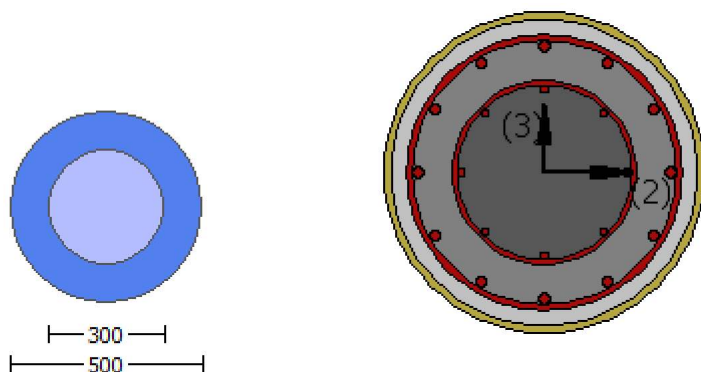
$$D_{c1} = D_{\text{ext}} - 2 \cdot \text{cover} - \text{External Hoop Diameter} = 440.00, \text{ is the total Length of all stirrups parallel to loading (shear) direction}$$

$s1 = 100.00$   
 core:  $s2 = Av2 * (Dc2/2) / (s2 * Ag) = 0.00046968$   
 $Av2 = 50.26548$ , is the area of stirrup  
 $Dc2 = Dint - Internal\ Hoop\ Diameter = 292.00$ , is the total Length of all stirrups parallel to loading (shear) direction  
 $s2 = 250.00$   
 The term  $2*tf/bw*(ffe/fs)$  is implemented to account for FRP contribution  
 where  $f = 2*tf/bw$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $ffe/fs$  normalises  $f$  to steel strength  
 All these variables have already been given in Shear control ratio calculation.  
 For the normalisation  $fs$  of jacket is used.  
 $NUD = 7386.194$   
 $Ag = 196349.541$   
 $f_{cE} = (f_{c,jacket} * Area_{jacket} + f_{c,core} * Area_{core}) / section\_area = 33.00$   
 $f_{yIE} = (f_{y,ext\_Long\_Reinf} * Area_{ext\_Long\_Reinf} + f_{y,int\_Long\_Reinf} * Area_{int\_Long\_Reinf}) / Area_{Tot\_Long\_Rein} = 2.1219958E-314$   
 $f_{yTE} = (f_{y,ext\_Trans\_Reinf} * Area_{ext\_Trans\_Reinf} + f_{y,int\_Trans\_Reinf} * Area_{int\_Trans\_Reinf}) / Area_{Tot\_Trans\_Rein} = 555.56$   
 $\rho_l = Area_{Tot\_Long\_Rein} / (Ag) = 0.015552$   
 $f_{cE} = 33.00$

-----  
 End Of Calculation of Chord Rotation Capacity for element: column JCC1 of floor 1  
 At local axis: 2  
 Integration Section: (b)

## Calculation No. 15

column C1, Floor 1  
 Limit State: Life Safety (data interpolation between analysis steps 1 and 2)  
 Analysis: Uniform +X  
 Check: Shear capacity  $VRd$   
 Edge: End  
 Local Axis: (3)



Start Of Calculation of Shear Capacity for element: column JCC1 of floor 1  
 At local axis: 3  
 Integration Section: (b)

Section Type: rcjcs

## Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE 41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Jacket

New material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 25.00$

New material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 500.00$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

Existing Column

New material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 25.00$

New material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 500.00$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of  $\gamma$  for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE 41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE 41-17).

Jacket

New material: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material: Steel Strength,  $f_s = f_{sm} = 555.56$

Existing Column

New material: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material: Steel Strength,  $f_s = f_{sm} = 555.56$

#####

External Diameter,  $D = 500.00$

Internal Diameter,  $D = 300.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Inadequate Lap Length with  $l_o/l_{ou,min} = l_b/l_d = 0.30$

FRP Wrapping Data

Type: Carbon

Dry properties (design values)

Thickness,  $t = 1.00$

Tensile Strength,  $f_{fu} = 840.00$

Tensile Modulus,  $E_f = 82000.00$

Elongation,  $\epsilon_{fu} = 0.009$

Number of directions,  $NoDir = 1$

Fiber orientations,  $bi: 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

## Stepwise Properties

EDGE -A-

Bending Moment,  $M_a = 1.1533666E-009$

Shear Force,  $V_a = -3.4066156E-013$

EDGE -B-

Bending Moment,  $M_b = -1.3104847E-010$

Shear Force,  $V_b = 3.4066156E-013$

BOTH EDGES

Axial Force,  $F = -7386.194$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 0.00$

-Compression:  $As_c = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{l,ten} = 1017.876$

-Compression:  $A_{sl,com} = 1017.876$

-Middle:  $A_{sl,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $Db_{L,ten} = 18.00$

New component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = 1.0 \cdot V_n = 614519.737$

$V_n$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{Col0} = 614519.737$

$V_{Col} = 614519.737$

$k_{nl} = 1.00$

displacement\_ductility\_demand = 0.00

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ '

where  $V_f$  is the contribution of FRPs ((11.3), ACI 440).

= 1 (normal-weight concrete)

Mean concrete strength:  $f'_c = (f'_c_{jacket} \cdot Area_{jacket} + f'_c_{core} \cdot Area_{core}) / Area_{section} = 25.00$ , but  $f'_c^{0.5} \leq 8.3$  MPa ((22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.3104847E-010$

$\nu_u = 3.4066156E-013$

$d = 0.8 \cdot D = 400.00$

$N_u = 7386.194$

$A_g = 196349.541$

From ((11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 246740.11$

$V_{s1} = 246740.11$  is calculated for jacket, with:

$A_v = \sqrt{2} \cdot A_{stirrup} = 123370.055$

$f_y = 500.00$

$s = 100.00$

$V_{s1}$  is multiplied by  $Col1 = 1.00$

$s/d = 0.25$

$V_{s2} = 0.00$  is calculated for core, with:

$A_v = \sqrt{2} \cdot A_{stirrup} = 78956.835$

$f_y = 500.00$

$s = 250.00$

$V_{s2}$  is multiplied by  $Col2 = 0.00$

$s/d = 1.04167$

$V_f$  ((11-3)-(11.4), ACI 440) = 308320.00

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In ((11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression, where  $a$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\alpha = 45^\circ$  and  $\alpha = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\alpha_1 = \alpha_1 + 90^\circ = 90.00$

$V_f = \min(|V_f(45, \alpha_1)|, |V_f(-45, \alpha_1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{oDir} = 1.00$

$d_{fv} = d$  (figure 11.2, ACI 440) = 470.00

$f_{fe}$  ((11-5), ACI 440) = 328.00

$E_f = 82000.00$

$f_e = 0.004$ , from ((11.6a), ACI 440

with  $f_u = 0.009$

From ((11-11), ACI 440:  $V_s + V_f \leq 417394.406$

$b_w \cdot d = \sqrt{2} \cdot d / 4 = 125663.706$

displacement\_ductility\_demand is calculated as  $\delta_u / y$

- Calculation of  $\delta_u / y$  for END B -

for rotation axis 2 and integ. section (b)

From analysis, chord rotation  $\theta = 8.1346664E-021$

$y = (M_y \cdot L_s / 3) / E_{eff} = 0.00453062$  ((4.29), Biskinis Phd))

$M_y = 2.2517E+008$

$L_s = M/V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) = 1500.00



From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 2.4850E+013$   
 $factor = 0.30$   
 $A_g = 196349.541$   
Mean concrete strength:  $f_c' = (f_c'_{jacket} * Area_{jacket} + f_c'_{core} * Area_{core}) / Area_{section} = 33.00$   
 $N = 7386.194$   
 $E_c * I_g = E_{c\_jacket} * I_{g\_jacket} + E_{c\_core} * I_{g\_core} = 8.2833E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\phi_y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \min(M_{y\_ten}, M_{y\_com}) = 2.2517E+008$   
 $y = 6.4424821E-006$   
 $M_{y\_ten} (8c) = 2.2517E+008$   
 $\phi_{y\_ten} (7c) = 61.81367$   
error of function (7c) = 0.00026388  
 $M_{y\_com} (8d) = 8.1922E+008$   
 $\phi_{y\_com} (7d) = 63.67467$   
error of function (7d) = -0.00874883  
with ((10.1), ASCE 41-17)  $\phi_y = \min(\phi_y, 1.25 * \phi_y * (I_b / I_d)^{2/3}) = 0.0027778$   
 $\phi_{co} = 0.002$   
 $\phi_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 250.00$   
 $v = 0.00101359$   
 $N = 7386.194$   
 $A_c = 196349.541$   
((10.1), ASCE 41-17)  $\phi_y = \min(\phi_y, 1.25 * \phi_y * (I_b / I_d)^{2/3}) = 0.23280363$   
with  $f_c' ((12.3), ACI 440) = 37.11312$   
 $f_c = 33.00$   
 $f_l = 1.312$   
 $k = 1$   
Effective FRP thickness,  $t_f = NL * t * \cos(b_1) = 1.00$   
 $\phi_{fe} ((12.5) \text{ and } (12.7)) = 0.004$   
 $f_u = 0.009$   
 $E_f = 82000.00$

Calculation of ratio  $I_b / I_d$

Inadequate Lap Length with  $I_b / I_d = 0.30$

End Of Calculation of Shear Capacity for element: column JCC1 of floor 1

At local axis: 3

Integration Section: (b)

## Calculation No. 16

column C1, Floor 1

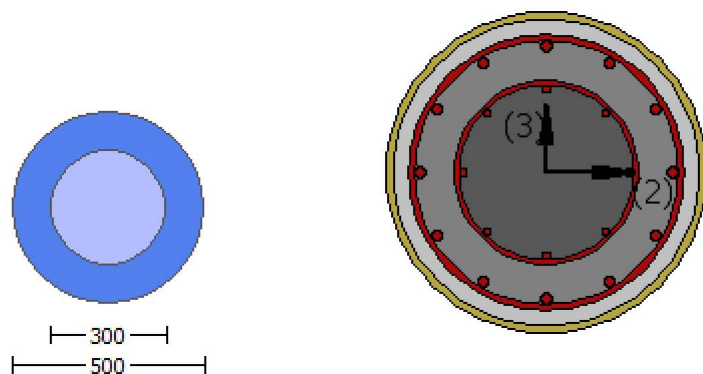
Limit State: Life Safety (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity ( $\phi$ )

Edge: End

Local Axis: (3)



Start Of Calculation of Shear Capacity ratio for element: column JCC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rcjcs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Mean strength values are used for both shear and moment calculations.

Consequently:

Jacket

New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

Existing Column

New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Jacket

New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 694.45$

Existing Column

New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 694.45$

#####

External Diameter,  $D = 500.00$

Internal Diameter,  $D = 300.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.51481

Element Length,  $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Inadequate Lap Length with  $l_o/l_{ou,min} = 0.30$

#### FRP Wrapping Data

Type: Carbon

Dry properties (design values)

Thickness,  $t = 1.00$

Tensile Strength,  $f_{fu} = 840.00$

Tensile Modulus,  $E_f = 82000.00$

Elongation,  $\epsilon_{fu} = 0.009$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

At local axis: 3

EDGE -A-

Shear Force,  $V_a = 3.4807814E-030$

EDGE -B-

Shear Force,  $V_b = -3.4807814E-030$

BOTH EDGES

Axial Force,  $F = -7389.214$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{sl,t} = 0.00$

-Compression:  $A_{sl,c} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{sl,ten} = 1017.876$

-Compression:  $A_{sl,com} = 1017.876$

-Middle:  $A_{sl,mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.23017425$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 162459.651$  with

$M_{pr1} = \max(\mu_{u1+}, \mu_{u1-}) = 2.4369E+008$

$\mu_{u1+} = 2.4369E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$\mu_{u1-} = 2.4369E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \max(\mu_{u2+}, \mu_{u2-}) = 2.4369E+008$

$\mu_{u2+} = 2.4369E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination

$\mu_{u2-} = 2.4369E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

#### Calculation of $\mu_{u1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$

$\mu_u = 2.4369E+008$

$\beta_1 = 0.80285146$

$\beta_2 = 0.71403327$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 49.98874$

conf. factor  $c = 1.51481$

$f_c = 33.00$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \min(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 389.0139$

$l_b/l_d = 0.30$

$d_1 = 44.00$

$R = 250.00$

$v = 0.00100813$

$$N = 7389.214$$

$$Ac = 196349.541$$

$$= *Min(1, 1.25 * (lb/d)^{2/3}) = 0.1620683$$

Calculation of ratio lb/d

Inadequate Lap Length with lb/d = 0.30

Calculation of Mu1-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.4369E+008

$$= 0.80285146$$

$$' = 0.71403327$$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TDY: fcc = fc\* c = 49.98874

conf. factor c = 1.51481

fc = 33.00

From 10.3.5, ASCE 41-17, Final value of fy: fy\*Min(1, 1.25\*(lb/d)^{2/3}) = 389.0139

lb/d = 0.30

d1 = 44.00

R = 250.00

v = 0.00100813

N = 7389.214

Ac = 196349.541

$$= *Min(1, 1.25 * (lb/d)^{2/3}) = 0.1620683$$

Calculation of ratio lb/d

Inadequate Lap Length with lb/d = 0.30

Calculation of Mu2+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.4369E+008

$$= 0.80285146$$

$$' = 0.71403327$$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TDY: fcc = fc\* c = 49.98874

conf. factor c = 1.51481

fc = 33.00

From 10.3.5, ASCE 41-17, Final value of fy: fy\*Min(1, 1.25\*(lb/d)^{2/3}) = 389.0139

lb/d = 0.30

d1 = 44.00

R = 250.00

v = 0.00100813

N = 7389.214

Ac = 196349.541

$$= *Min(1, 1.25 * (lb/d)^{2/3}) = 0.1620683$$

Calculation of ratio lb/d

Inadequate Lap Length with lb/d = 0.30

## Calculation of Mu2-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
 $\text{Mu} = 2.4369\text{E}+008$

$$= 0.80285146$$

$$\rho = 0.71403327$$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 49.98874$

conf. factor  $c = 1.51481$

$$f_c = 33.00$$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 389.0139$

$$l_b/d = 0.30$$

$$d_1 = 44.00$$

$$R = 250.00$$

$$v = 0.00100813$$

$$N = 7389.214$$

$$A_c = 196349.541$$

$$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.1620683$$

## Calculation of ratio $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 705811.584$

Calculation of Shear Strength at edge 1,  $V_{r1} = 705811.584$

$V_{r1} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_n l \cdot V_{Col0}$

$$V_{Col0} = 705811.584$$

$k_n l = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

Mean concrete strength:  $f_c' = (f_c'_{\text{jacket}} \cdot \text{Area}_{\text{jacket}} + f_c'_{\text{core}} \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$$M/Vd = 2.00$$

$$\text{Mu} = 1.2438899\text{E}-011$$

$$V_u = 3.4807814\text{E}-030$$

$$d = 0.8 \cdot D = 400.00$$

$$N_u = 7389.214$$

$$A_g = 196349.541$$

From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 274157.871$

$V_{s1} = 274157.871$  is calculated for jacket, with:

$$A_v = \sqrt{2} \cdot A_{\text{stirup}} = 123370.055$$

$$f_y = 555.56$$

$$s = 100.00$$

$V_{s1}$  is multiplied by  $\text{Col1} = 1.00$

$$s/d = 0.25$$

$V_{s2} = 0.00$  is calculated for core, with:

$$A_v = \sqrt{2} \cdot A_{\text{stirup}} = 78956.835$$

$$f_y = 555.56$$

$$s = 250.00$$

$V_{s2}$  is multiplied by  $\text{Col2} = 0.00$

$$s/d = 1.04167$$

$V_f$  ((11-3)-(11.4), ACI 440) = 308320.00

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a)\sin a$  which is more a generalised expression, where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta)$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta_1 = \theta_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, a_1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = N_L \cdot t / \text{NoDir} = 1.00$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 470.00  
 $f_{fe}((11-5), \text{ACI 440}) = 328.00$   
 $E_f = 82000.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.009$   
 From (11-11), ACI 440:  $V_s + V_f \leq 479549.663$   
 $b_w \cdot d = \sqrt[4]{d \cdot d} = 125663.706$

Calculation of Shear Strength at edge 2,  $V_{r2} = 705811.584$   
 $V_{r2} = V_{Col}((10.3), \text{ASCE 41-17}) = k_n l \cdot V_{Col0}$   
 $V_{Col0} = 705811.584$   
 $k_n l = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\lambda = 1$  (normal-weight concrete)  
 Mean concrete strength:  $f'_c = (f'_c \cdot \text{jacket} \cdot \text{Area}_{\text{jacket}} + f'_c \cdot \text{core} \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 33.00$ , but  $f'_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $\mu_u = 1.2438899\text{E-}011$   
 $\nu_u = 3.4807814\text{E-}030$   
 $d = 0.8 \cdot D = 400.00$   
 $N_u = 7389.214$   
 $A_g = 196349.541$   
 From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 274157.871$   
 $V_{s1} = 274157.871$  is calculated for jacket, with:  
 $A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 123370.055$   
 $f_y = 555.56$   
 $s = 100.00$   
 $V_{s1}$  is multiplied by  $\text{Col1} = 1.00$   
 $s/d = 0.25$   
 $V_{s2} = 0.00$  is calculated for core, with:  
 $A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 78956.835$   
 $f_y = 555.56$   
 $s = 250.00$   
 $V_{s2}$  is multiplied by  $\text{Col2} = 0.00$   
 $s/d = 1.04167$   
 $V_f((11-3)-(11.4), \text{ACI 440}) = 308320.00$   
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a)\sin a$  which is more a generalised expression, where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta)$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta_1 = \theta_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, a_1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = N_L \cdot t / \text{NoDir} = 1.00$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 470.00  
 $f_{fe}((11-5), \text{ACI 440}) = 328.00$   
 $E_f = 82000.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.009$   
 From (11-11), ACI 440:  $V_s + V_f \leq 479549.663$   
 $b_w \cdot d = \sqrt[4]{d \cdot d} = 125663.706$

End Of Calculation of Shear Capacity ratio for element: column JCC1 of floor 1  
 At local axis: 3

Start Of Calculation of Shear Capacity ratio for element: column JCC1 of floor 1

At Shear local axis: 2

(Bending local axis: 3)

Section Type: rcjcs

#### Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Mean strength values are used for both shear and moment calculations.

Consequently:

Jacket

New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

Existing Column

New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Jacket

New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 694.45$

Existing Column

New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 694.45$

#####

External Diameter,  $D = 500.00$

Internal Diameter,  $D = 300.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.51481

Element Length,  $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Inadequate Lap Length with  $l_o/l_{ou,min} = 0.30$

FRP Wrapping Data

Type: Carbon

Dry properties (design values)

Thickness,  $t = 1.00$

Tensile Strength,  $f_{fu} = 840.00$

Tensile Modulus,  $E_f = 82000.00$

Elongation,  $\epsilon_{fu} = 0.009$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i: 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

At local axis: 2

EDGE -A-

Shear Force,  $V_a = -2.1312935E-046$

EDGE -B-

Shear Force,  $V_b = 2.1312935E-046$

BOTH EDGES

Axial Force,  $F = -7389.214$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{st} = 0.00$

-Compression:  $As_c = 3053.628$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $As_{ten} = 1017.876$   
-Compression:  $As_{com} = 1017.876$   
-Middle:  $As_{mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.23017425$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 162459.651$   
with

$M_{pr1} = \text{Max}(\mu_{u1+}, \mu_{u1-}) = 2.4369E+008$

$\mu_{u1+} = 2.4369E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$\mu_{u1-} = 2.4369E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(\mu_{u2+}, \mu_{u2-}) = 2.4369E+008$

$\mu_{u2+} = 2.4369E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination

$\mu_{u2-} = 2.4369E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

Calculation of  $\mu_{u1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$   
 $\mu_u = 2.4369E+008$

$\phi = 0.80285146$

$\phi' = 0.71403327$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TB DY:  $f_{cc} = f_c^* \quad c = 49.98874$

conf. factor  $c = 1.51481$

$f_c = 33.00$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 389.0139$

$l_b/d = 0.30$

$d_1 = 44.00$

$R = 250.00$

$v = 0.00100813$

$N = 7389.214$

$A_c = 196349.541$

$\phi = \phi' \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.1620683$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of  $\mu_{u1-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$   
 $\mu_u = 2.4369E+008$

$\phi = 0.80285146$

$\phi' = 0.71403327$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TB DY:  $f_{cc} = f_c^* \quad c = 49.98874$

conf. factor  $c = 1.51481$

$f_c = 33.00$



From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 389.0139$

$l_b/d = 0.30$

$d_1 = 44.00$

$R = 250.00$

$v = 0.00100813$

$N = 7389.214$

$A_c = 196349.541$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.1620683$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of  $\mu_{2+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$\mu = 2.4369\text{E}+008$

$= 0.80285146$

$' = 0.71403327$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 49.98874$

conf. factor  $c = 1.51481$

$f_c = 33.00$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 389.0139$

$l_b/d = 0.30$

$d_1 = 44.00$

$R = 250.00$

$v = 0.00100813$

$N = 7389.214$

$A_c = 196349.541$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.1620683$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of  $\mu_{2-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$\mu = 2.4369\text{E}+008$

$= 0.80285146$

$' = 0.71403327$

error of function (3.68), Biskinis Phd = 27987.158

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 49.98874$

conf. factor  $c = 1.51481$

$f_c = 33.00$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 389.0139$

$l_b/d = 0.30$

$d_1 = 44.00$

$R = 250.00$

$v = 0.00100813$

$N = 7389.214$

$A_c = 196349.541$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.1620683$

Calculation of ratio  $l_b/l_d$

Inadequate Lap Length with  $l_b/l_d = 0.30$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 705811.584$

Calculation of Shear Strength at edge 1,  $V_{r1} = 705811.584$

$V_{r1} = V_{Col} \text{ ((10.3), ASCE 41-17)} = k_n l^* V_{ColO}$

$V_{ColO} = 705811.584$

$k_n l = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f^* V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

Mean concrete strength:  $f_c' = (f_c'_{\text{jacket}} \cdot \text{Area}_{\text{jacket}} + f_c'_{\text{core}} \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/V_d = 2.00$

$\mu_u = 1.5940012E-011$

$\nu_u = 2.1312935E-046$

$d = 0.8 \cdot D = 400.00$

$N_u = 7389.214$

$A_g = 196349.541$

From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 274157.871$

$V_{s1} = 274157.871$  is calculated for jacket, with:

$A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 123370.055$

$f_y = 555.56$

$s = 100.00$

$V_{s1}$  is multiplied by  $\text{Col1} = 1.00$

$s/d = 0.25$

$V_{s2} = 0.00$  is calculated for core, with:

$A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 78956.835$

$f_y = 555.56$

$s = 250.00$

$V_{s2}$  is multiplied by  $\text{Col2} = 0.00$

$s/d = 1.04167$

$V_f \text{ ((11-3)-(11.4), ACI 440)} = 308320.00$

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \alpha + \cos \alpha$  is replaced with  $(\cot \alpha + \cot \alpha) \sin \alpha$  which is more a generalised expression,

where  $\alpha$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\alpha = 45^\circ$  and  $\alpha = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\alpha_1 = b_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \alpha_1)|, |V_f(-45, \alpha_1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{oDir} = 1.00$

$d_{fv} = d$  (figure 11.2, ACI 440) = 470.00

$f_{fe} \text{ ((11-5), ACI 440)} = 328.00$

$E_f = 82000.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.009$

From (11-11), ACI 440:  $V_s + V_f \leq 479549.663$

$b_w \cdot d = \sqrt{2} \cdot d \cdot d / 4 = 125663.706$

Calculation of Shear Strength at edge 2,  $V_{r2} = 705811.584$

$V_{r2} = V_{Col} \text{ ((10.3), ASCE 41-17)} = k_n l^* V_{ColO}$

$V_{ColO} = 705811.584$

$k_n l = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f^* V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\lambda = 1$  (normal-weight concrete)  
Mean concrete strength:  $f_c' = (f_c'_{\text{jacket}} \cdot \text{Area}_{\text{jacket}} + f_c'_{\text{core}} \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $\mu_u = 1.5940012\text{E-}011$   
 $\nu_u = 2.1312935\text{E-}046$   
 $d = 0.8 \cdot D = 400.00$   
 $N_u = 7389.214$   
 $A_g = 196349.541$   
From (11.5.4.8), ACI 318-14:  $V_s = V_{s1} + V_{s2} = 274157.871$   
 $V_{s1} = 274157.871$  is calculated for jacket, with:  
 $A_v = \rho_s \cdot A_{\text{stirrup}} = 123370.055$   
 $f_y = 555.56$   
 $s = 100.00$   
 $V_{s1}$  is multiplied by  $\text{Col1} = 1.00$   
 $s/d = 0.25$   
 $V_{s2} = 0.00$  is calculated for core, with:  
 $A_v = \rho_s \cdot A_{\text{stirrup}} = 78956.835$   
 $f_y = 555.56$   
 $s = 250.00$   
 $V_{s2}$  is multiplied by  $\text{Col2} = 0.00$   
 $s/d = 1.04167$   
 $V_f((11-3)-(11.4), \text{ACI } 440) = 308320.00$   
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \cot \alpha) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE).  
This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
orientation 1:  $\theta = \theta_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, \theta_1)|)$ , with:  
total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{\text{Dir}} = 1.00$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 470.00  
 $f_{fe}((11-5), \text{ACI } 440) = 328.00$   
 $E_f = 82000.00$   
 $f_{fe} = 0.004$ , from (11.6a), ACI 440  
with  $f_u = 0.009$   
From (11-11), ACI 440:  $V_s + V_f \leq 479549.663$   
 $b_w \cdot d = \rho_s \cdot d^2 / 4 = 125663.706$

End Of Calculation of Shear Capacity ratio for element: column JCC1 of floor 1  
At local axis: 2

Start Of Calculation of Chord Rotation Capacity for element: column JCC1 of floor 1  
At local axis: 3  
Integration Section: (b)  
Section Type: rcjcs

#### Constant Properties

Knowledge Factor,  $\lambda = 1.00$   
Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE 41-17.  
Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17  
Consequently:  
Jacket  
New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$   
New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$   
Concrete Elasticity,  $E_c = 26999.444$   
Steel Elasticity,  $E_s = 200000.00$   
Existing Column  
New material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$   
New material of Primary Member: Steel Strength,  $f_s = f_{sm} = 555.56$   
Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$   
 External Diameter,  $D = 500.00$   
 Internal Diameter,  $D = 300.00$   
 Cover Thickness,  $c = 25.00$   
 Element Length,  $L = 3000.00$   
 Primary Member  
 Ribbed Bars  
 Ductile Steel  
 Without Detailing for Earthquake Resistance (including stirrups not closed at  $135^\circ$ )  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Inadequate Lap Length with  $l_b/l_d = 0.30$   
 FRP Wrapping Data  
 Type: Carbon  
 Dry properties (design values)  
 Thickness,  $t = 1.00$   
 Tensile Strength,  $f_{fu} = 840.00$   
 Tensile Modulus,  $E_f = 82000.00$   
 Elongation,  $ε_{fu} = 0.009$   
 Number of directions,  $N_{Dir} = 1$   
 Fiber orientations,  $β_i = 0.00^\circ$   
 Number of layers,  $N_L = 1$   
 Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

Bending Moment,  $M = 0.12607815$   
 Shear Force,  $V_2 = 7549.016$   
 Shear Force,  $V_3 = 3.4066156E-013$   
 Axial Force,  $F = -7386.194$   
 Longitudinal Reinforcement Area Distribution (in 2 divisions)  
   -Tension:  $A_{st} = 0.00$   
   -Compression:  $A_{sc} = 3053.628$   
 Longitudinal Reinforcement Area Distribution (in 3 divisions)  
   -Tension:  $A_{st,ten} = 1017.876$   
   -Compression:  $A_{st,com} = 1017.876$   
   -Middle:  $A_{st,mid} = 1017.876$   
 Mean Diameter of Tension Reinforcement,  $D_{bL} = 18.00$

New component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_R = 1.0^*$   $u = 0.03090612$   
 $u = y + p = 0.03090612$

- Calculation of  $y$  -

$y = (M_y * L_s / 3) / E_{eff} = 0.00090612$  ((4.29), Biskinis Phd))  
 $M_y = 2.2517E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) =  $300.00$   
 From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 2.4850E+013$   
 $factor = 0.30$   
 $A_g = 196349.541$   
 Mean concrete strength:  $f'_c = (f'_{c,jacket} * Area_{jacket} + f'_{c,core} * Area_{core}) / Area_{section} = 33.00$   
 $N = 7386.194$   
 $E_c * I_g = E_{c,jacket} * I_{g,jacket} + E_{c,core} * I_{g,core} = 8.2833E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \min(M_{y,ten}, M_{y,com}) = 2.2517E+008$   
 $y = 6.4424821E-006$   
 $M_{y,ten} (8c) = 2.2517E+008$

$\rho_{ten}(7c) = 61.81367$   
error of function (7c) = 0.00026388  
My\_com (8d) = 8.1922E+008  
 $\rho_{com}(7d) = 63.67467$   
error of function (7d) = -0.00874883  
with ((10.1), ASCE 41-17)  $\rho_y = \text{Min}(\rho_y, 1.25 \cdot \rho_y \cdot (l_b/l_d)^{2/3}) = 0.0027778$   
 $\rho_{co} = 0.002$   
 $\rho_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 250.00$   
 $v = 0.00101359$   
 $N = 7386.194$   
 $A_c = 196349.541$   
((10.1), ASCE 41-17)  $\rho = \text{Min}(\rho, 1.25 \cdot \rho \cdot (l_b/l_d)^{2/3}) = 0.23280363$   
with  $f_c^*$  ((12.3), ACI 440) = 37.11312  
 $f_c = 33.00$   
 $f_l = 1.312$   
 $k = 1$   
Effective FRP thickness,  $t_f = N \cdot t \cdot \cos(b_1) = 1.00$   
 $\rho_{fe}$  ((12.5) and (12.7)) = 0.004  
 $\rho_u = 0.009$   
 $E_f = 82000.00$

Calculation of ratio  $l_b/l_d$

Inadequate Lap Length with  $l_b/l_d = 0.30$

- Calculation of  $\rho_p$  -

From table 10-9:  $\rho_p = 0.03$

with:

- Columns controlled by inadequate development or splicing along the clear height because  $l_b/l_d < 1$   
shear control ratio  $V_y E / V_{co} I_{OE} = 0.23017425$   
 $d = d_{external} = 0.00$   
 $s = s_{external} = 0.00$   
 $t = s_1 + s_2 + 2 \cdot t_f / b_w \cdot (f_{fe} / f_s) = 0.00795744$   
jacket:  $s_1 = A_{v1} \cdot (D_{c1} / 2) / (s_1 \cdot A_g) = 0.0027646$   
 $A_{v1} = 78.53982$ , is the area of stirrup  
 $D_{c1} = D_{ext} - 2 \cdot \text{cover} - \text{External Hoop Diameter} = 440.00$ , is the total Length of all stirrups parallel to loading (shear) direction  
 $s_1 = 100.00$   
core:  $s_2 = A_{v2} \cdot (D_{c2} / 2) / (s_2 \cdot A_g) = 0.00046968$   
 $A_{v2} = 50.26548$ , is the area of stirrup  
 $D_{c2} = D_{int} - \text{Internal Hoop Diameter} = 292.00$ , is the total Length of all stirrups parallel to loading (shear) direction  
 $s_2 = 250.00$

The term  $2 \cdot t_f / b_w \cdot (f_{fe} / f_s)$  is implemented to account for FRP contribution  
where  $f = 2 \cdot t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength  
All these variables have already been given in Shear control ratio calculation.  
For the normalisation  $f_s$  of jacket is used.

$N_{UD} = 7386.194$   
 $A_g = 196349.541$   
 $f_{cE} = (f_{c,jacket} \cdot \text{Area}_{jacket} + f_{c,core} \cdot \text{Area}_{core}) / \text{section\_area} = 33.00$   
 $f_{yLE} = (f_{y,ext\_Long\_Reinf} \cdot \text{Area}_{ext\_Long\_Reinf} + f_{y,int\_Long\_Reinf} \cdot \text{Area}_{int\_Long\_Reinf}) / \text{Area}_{Tot\_Long\_Rein} = 21219958E-314$   
 $f_{yTE} = (f_{y,ext\_Trans\_Reinf} \cdot \text{Area}_{ext\_Trans\_Reinf} + f_{y,int\_Trans\_Reinf} \cdot \text{Area}_{int\_Trans\_Reinf}) / \text{Area}_{Tot\_Trans\_Rein} = 555.56$   
 $\rho_l = \text{Area}_{Tot\_Long\_Rein} / (A_g) = 0.015552$   
 $f_{cE} = 33.00$

End Of Calculation of Chord Rotation Capacity for element: column JCC1 of floor 1

At local axis: 3

Integration Section: (b)

