

Detailed Member Calculations

Units: N&mm

Regulation: ASCE 41-17

Calculation No. 1

column C1, Floor 1

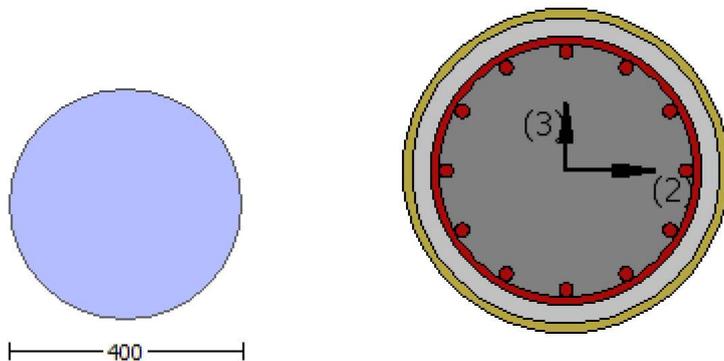
Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Shear capacity V_{Rd}

Edge: Start

Local Axis: (2)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (a)

Section Type: rccs

Constant Properties

Knowledge Factor, $\gamma = 1.00$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE 41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength, $f_c = f_{c_lower_bound} = 16.00$

Existing material of Primary Member: Steel Strength, $f_s = f_{s_lower_bound} = 400.00$

Concrete Elasticity, $E_c = 21019.039$

Steel Elasticity, $E_s = 200000.00$

#####

Note: Especially for the calculation of γ for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE 41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE 41-17).

Existing material: Concrete Strength, $f_c = f_{cm} = 20.00$

Existing material: Steel Strength, $f_s = f_{sm} = 444.44$

#####

Diameter, $D = 400.00$

Cover Thickness, $c = 25.00$

Element Length, $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length $l_o = l_b = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness, $t = 1.016$

Tensile Strength, $f_{fu} = 1055.00$

Tensile Modulus, $E_f = 64828.00$

Elongation, $e_{fu} = 0.01$

Number of directions, $NoDir = 1$

Fiber orientations, $b_i = 0.00^\circ$

Number of layers, $NL = 1$

Radius of rounding corners, $R = 40.00$

Stepwise Properties

EDGE -A-

Bending Moment, $M_a = -1.0423E+007$

Shear Force, $V_a = -3472.528$

EDGE -B-

Bending Moment, $M_b = 0.09265189$

Shear Force, $V_b = 3472.528$

BOTH EDGES

Axial Force, $F = -4769.844$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension: $A_{sl,t} = 1272.345$

-Compression: $A_{sl,c} = 1781.283$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension: $A_{sl,ten} = 1017.876$

-Compression: $A_{sl,com} = 1017.876$

-Middle: $A_{sl,mid} = 1017.876$

Mean Diameter of Tension Reinforcement, $Db_{L,ten} = 18.00$

Existing component: From table 7-7, ASCE 41_17: Final Shear Capacity $V_R = \phi V_n = 235775.141$

V_n ((10.3), ASCE 41-17) = $k_n l \cdot V_{CoI0} = 235775.141$

$V_{CoI} = 235775.141$

$k_n l = 1.00$

$displacement_ductility_demand = 0.01890466$

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + \phi \cdot V_f$ ' where V_f is the contribution of FRPs (11.3), ACI 440).

 $\gamma = 1$ (normal-weight concrete)

$f_c' = 16.00$, but $f_c^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$M / Vd = 4.00$

$M_u = 1.0423E+007$

$V_u = 3472.528$

$d = 0.8 \cdot D = 320.00$

$N_u = 4769.844$
 $A_g = 125663.706$
 From (11.5.4.8), ACI 318-14: $V_s = 0.00$
 $A_v = \sqrt{2} \cdot A_{stirrup} = 123370.055$
 $f_y = 400.00$
 $s = 360.00$
 V_s is multiplied by $\text{Col} = 0.00$
 $s/d = 1.125$
 V_f ((11-3)-(11.4), ACI 440) = 194961.134
 $f = 0.95$, for fully-wrapped sections
 $w_f/s_f = 1$ (FRP strips adjacent to one another).
 In (11.3) $\sin + \cos$ is replaced with $(\cot + \cot a) \sin a$ which is more a generalised expression,
 where a is the angle of the crack direction (see KANEPE).
 This later relation, considered as a function $V_f(\theta, a)$, is implemented for every different fiber orientation a_i ,
 as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.
 orientation 1: $\theta = b1 + 90^\circ = 90.00$
 $V_f = \text{Min}(|V_f(45, a1)|, |V_f(-45, a1)|)$, with:
 total thickness per orientation, $t_{f1} = N_L \cdot t / \text{NoDir} = 1.016$
 $d_{fv} = d$ (figure 11.2, ACI 440) = 370.00
 f_{fe} ((11-5), ACI 440) = 259.312
 $E_f = 64828.00$
 $f_e = 0.004$, from (11.6a), ACI 440
 with $f_u = 0.01$
 From (11-11), ACI 440: $V_s + V_f \leq 213705.936$
 $b_w \cdot d = \sqrt{2} \cdot d^2 / 4 = 80424.772$

 displacement ductility demand is calculated as δ / y

- Calculation of δ / y for END A -
 for rotation axis 3 and integ. section (a)

 From analysis, chord rotation $\theta = 0.00035245$
 $y = (M_y \cdot L_s / 3) / E_{eff} = 0.01864342$ ((4.29), Biskinis Phd))
 $M_y = 1.4766E+008$
 $L_s = M/V$ (with $L_s > 0.1 \cdot L$ and $L_s < 2 \cdot L$) = 3001.447
 From table 10.5, ASCE 41_17: $E_{eff} = \text{factor} \cdot E_c \cdot I_g = 7.9240E+012$
 factor = 0.30
 $A_g = 125663.706$
 $f_c' = 20.00$
 $N = 4769.844$
 $E_c \cdot I_g = 2.6413E+013$

 Calculation of Yielding Moment M_y

 Calculation of δ / y and M_y according to (7) - (8) in Biskinis and Fardis

 $M_y = \text{Min}(M_{y_ten}, M_{y_com}) = 1.4766E+008$
 $y = 8.1688007E-006$
 M_{y_ten} (8c) = 1.4766E+008
 δ_{ten} (7c) = 72.40642
 error of function (7c) = 0.00117228
 M_{y_com} (8d) = 3.7493E+008
 δ_{com} (7d) = 69.91126
 error of function (7d) = 0.00301342
 with ((10.1), ASCE 41-17) $e_y = \text{Min}(e_y, 1.25 \cdot e_y \cdot (l_b / l_d)^{2/3}) = 0.0022222$
 $e_{co} = 0.002$
 $a_{pl} = 0.45$ ((9c) in Biskinis and Fardis for FRP Wrap)
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00157305$
 $N = 4769.844$
 $A_c = 125663.706$
 ((10.1), ASCE 41-17) $e_y = \text{Min}(e_y, 1.25 \cdot e_y \cdot (l_b / l_d)^{2/3}) = 0.44757577$

with f_c^* ((12.3), ACI 440) = 24.12975
 $f_c = 20.00$
 $f_l = 1.3173$
 $k = 1$
Effective FRP thickness, $t_f = NL * t * \cos(b_1) = 1.016$
 e_{fe} ((12.5) and (12.7)) = 0.004
 $f_u = 0.01$
 $E_f = 64828.00$

Calculation of ratio l_b/l_d

Lap Length: $l_d/l_{d,min} = 0.33490748$

$l_b = 300.00$

$l_d = 895.7698$

Calculation of I according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

= 1

$d_b = 18.00$

Mean strength value of all re-bars: $f_y = 444.44$

$f_c' = 20.00$, but $f_c^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \sqrt{2} * \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

End Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (a)

Calculation No. 2

column C1, Floor 1

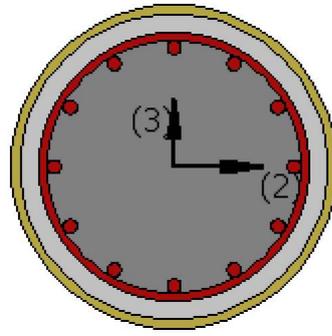
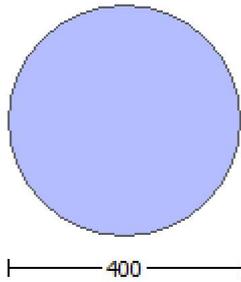
Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity (θ)

Edge: Start

Local Axis: (2)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 3
 (Bending local axis: 2)
 Section Type: rccs

Constant Properties

Knowledge Factor, $\gamma = 1.00$
 Mean strength values are used for both shear and moment calculations.
 Consequently:
 Existing material of Primary Member: Concrete Strength, $f_c = f_{cm} = 20.00$
 Existing material of Primary Member: Steel Strength, $f_s = f_{sm} = 444.44$
 Concrete Elasticity, $E_c = 21019.039$
 Steel Elasticity, $E_s = 200000.00$

 Note: Especially for the calculation of moment strengths,
 the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14
 Existing material: Steel Strength, $f_s = 1.25 \cdot f_{sm} = 555.55$

 Diameter, $D = 400.00$
 Cover Thickness, $c = 25.00$
 Mean Confinement Factor overall section = 1.84055
 Element Length, $L = 3000.00$
 Primary Member
 Ribbed Bars
 Ductile Steel
 Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)
 Longitudinal Bars With Ends Lapped Starting at the End Sections
 Lap Length $l_o = 300.00$
 FRP Wrapping Data
 Type: Carbon
 Cured laminate properties (design values)
 Thickness, $t = 1.016$
 Tensile Strength, $f_{fu} = 1055.00$
 Tensile Modulus, $E_f = 64828.00$
 Elongation, $e_{fu} = 0.01$
 Number of directions, $NoDir = 1$
 Fiber orientations, $b_i: 0.00^\circ$
 Number of layers, $NL = 1$
 Radius of rounding corners, $R = 40.00$

Stepwise Properties

At local axis: 3
 EDGE -A-
 Shear Force, $V_a = 1.2107607E-031$
 EDGE -B-
 Shear Force, $V_b = -1.2107607E-031$
 BOTH EDGES
 Axial Force, $F = -4771.233$
 Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension: Aslt = 0.00

-Compression: Aslc = 3053.628

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension: Asl,ten = 1017.876

-Compression: Asl,com = 1017.876

-Middle: Asl,mid = 1017.876

Calculation of Shear Capacity ratio , $V_e/V_r = 0.30252729$

Member Controlled by Flexure ($V_e/V_r < 1$)

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14 $V_e = (M_{pr1} + M_{pr2})/l_n = 90202.132$

with

$M_{pr1} = \text{Max}(M_{u1+} , M_{u1-}) = 1.3530E+008$

$M_{u1+} = 1.3530E+008$, is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$M_{u1-} = 1.3530E+008$, is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(M_{u2+} , M_{u2-}) = 1.3530E+008$

$M_{u2+} = 1.3530E+008$, is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

$M_{u2-} = 1.3530E+008$, is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

Calculation of M_{u1+}

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), M_u

$M_u = 1.3530E+008$

$\phi = 0.90757121$

$\phi' = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$

conf. factor $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$\phi = \phi' \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of l_b, min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$\beta = 1$

$d_b = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3 \text{ MPa}$ (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of Mu1-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu
Mu = 1.3530E+008

= 0.90757121
' = 0.80580716
error of function (3.68), Biskinis Phd = 28928.286
From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$
conf. factor $c = 1.84055$
 $f_c = 20.00$
From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$
 $l_b/l_d = 0.26792599$
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00155946$
 $N = 4771.233$
 $A_c = 125663.706$
= $\cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$
 $l_b = 300.00$
 $l_d = 1119.712$
Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
= 1
 $db = 18.00$
Mean strength value of all re-bars: $f_y = 555.55$
 $f_c' = 20.00$, but $f_c^{0.5} \leq 8.3 \text{ MPa}$ (22.5.3.1, ACI 318-14)
 $t = 1.00$
 $s = 0.80$
 $e = 1.00$
 $cb = 25.00$
 $K_{tr} = 1.14232$
 $A_{tr} = \sqrt{2} \cdot \text{Area of stirrup} = 123.3701$
 $s = 360.00$
 $n = 12.00$

Calculation of Mu2+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu
Mu = 1.3530E+008

= 0.90757121
' = 0.80580716
error of function (3.68), Biskinis Phd = 28928.286
From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$
conf. factor $c = 1.84055$
 $f_c = 20.00$
From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$
 $l_b/l_d = 0.26792599$
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00155946$
 $N = 4771.233$
 $A_c = 125663.706$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.28805051$$

Calculation of ratio l_b/d

Lap Length: $l_b/d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of l_b, min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$d_b = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f'_c = 20.00$, but $f_c^{0.5} \leq 8.3 \text{ MPa}$ (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of μ_2

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ

$\mu = 1.3530 \times 10^8$

$= 0.90757121$

$' = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY: $f_{cc} = f'_c \cdot c = 36.81095$

conf. factor $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 288.6089$

$l_b/d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/d

Lap Length: $l_b/d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of l_b, min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$d_b = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f'_c = 20.00$, but $f_c^{0.5} \leq 8.3 \text{ MPa}$ (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of Shear Strength $V_r = \text{Min}(V_{r1}, V_{r2}) = 298161.965$

Calculation of Shear Strength at edge 1, $V_{r1} = 298161.965$

$V_{r1} = V_{Col}$ ((10.3), ASCE 41-17) = $k_{nl} \cdot V_{Col0}$

$V_{Col0} = 298161.965$

$k_{nl} = 1$ (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ '
where V_f is the contribution of FRPs (11.3), ACI 440).

 $k_c = 1$ (normal-weight concrete)

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.2899604E-011$

$\nu_u = 1.2107607E-031$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14: $V_s = 0.00$

$A_v = \frac{1}{2} \cdot A_{\text{stirrup}} = 123370.055$

$f_y = 444.44$

$s = 360.00$

V_s is multiplied by $\text{Col} = 0.00$

$s/d = 1.125$

V_f ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$, for fully-wrapped sections

$w_f/s_f = 1$ (FRP strips adjacent to one another).

In (11.3) $\sin \theta + \cos \theta$ is replaced with $(\cot \theta + \csc \theta) \sin \alpha$ which is more a generalised expression,
where θ is the angle of the crack direction (see KANEPE).

This later relation, considered as a function $V_f(\theta, \alpha)$, is implemented for every different fiber orientation α_i ,
as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.

orientation 1: $\theta = b_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \alpha)|)$, with:

total thickness per orientation, $t_{f1} = N_L \cdot t / N_{\text{Dir}} = 1.016$

$d_{fv} = d$ (figure 11.2, ACI 440) = 370.00

f_{fe} ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$, from (11.6a), ACI 440

with $f_u = 0.01$

From (11-11), ACI 440: $V_s + V_f \leq 238930.50$

$b_w \cdot d = \frac{1}{4} \cdot d^2 = 80424.772$

Calculation of Shear Strength at edge 2, $V_{r2} = 298161.965$

$V_{r2} = V_{Col}$ ((10.3), ASCE 41-17) = $k_{nl} \cdot V_{Col0}$

$V_{Col0} = 298161.965$

$k_{nl} = 1$ (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ '
where V_f is the contribution of FRPs (11.3), ACI 440).

 $k_c = 1$ (normal-weight concrete)

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.2899604E-011$

$\nu_u = 1.2107607E-031$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14: $V_s = 0.00$

$A_v = \frac{1}{2} \cdot A_{\text{stirrup}} = 123370.055$

$f_y = 444.44$

$s = 360.00$

V_s is multiplied by $\text{Col} = 0.00$

$s/d = 1.125$

Vf ((11-3)-(11.4), ACI 440) = 194961.134

f = 0.95, for fully-wrapped sections

wf/sf = 1 (FRP strips adjacent to one another).

In (11.3) $\sin \theta + \cos \theta$ is replaced with $(\cot \theta + \cot \alpha) \sin \alpha$ which is more a generalised expression, where θ is the angle of the crack direction (see KANEPE).

This later relation, considered as a function Vf(θ), is implemented for every different fiber orientation ai, as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.

orientation 1: $\theta_1 = b_1 + 90^\circ = 90.00$

Vf = Min(|Vf(45, 1)|, |Vf(-45, a1)|), with:

total thickness per orientation, tf1 = NL*t/NoDir = 1.016

dfv = d (figure 11.2, ACI 440) = 370.00

ffe ((11-5), ACI 440) = 259.312

Ef = 64828.00

fe = 0.004, from (11.6a), ACI 440

with fu = 0.01

From (11-11), ACI 440: Vs + Vf <= 238930.50

bw*d = $\frac{V_s + V_f \cdot d}{4} = 80424.772$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1
At local axis: 3

Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1
At Shear local axis: 2
(Bending local axis: 3)
Section Type: rccs

Constant Properties

Knowledge Factor, $\phi = 1.00$

Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength, $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength, $f_s = f_{sm} = 444.44$

Concrete Elasticity, $E_c = 21019.039$

Steel Elasticity, $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength, $f_s = 1.25 \cdot f_{sm} = 555.55$

#####

Diameter, D = 400.00

Cover Thickness, c = 25.00

Mean Confinement Factor overall section = 1.84055

Element Length, L = 3000.00

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length lo = 300.00

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness, t = 1.016

Tensile Strength, $f_{fu} = 1055.00$

Tensile Modulus, Ef = 64828.00

Elongation, $e_{fu} = 0.01$

Number of directions, NoDir = 1

Fiber orientations, bi: 0.00°

Number of layers, NL = 1

Radius of rounding corners, R = 40.00

Stepwise Properties

At local axis: 2

EDGE -A-

Shear Force, $V_a = -7.4135260E-048$

EDGE -B-

Shear Force, $V_b = 7.4135260E-048$

BOTH EDGES

Axial Force, $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension: $A_{sl,t} = 0.00$

-Compression: $A_{sl,c} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension: $A_{sl,ten} = 1017.876$

-Compression: $A_{sl,com} = 1017.876$

-Middle: $A_{sl,mid} = 1017.876$

Calculation of Shear Capacity ratio, $V_e/V_r = 0.30252729$

Member Controlled by Flexure ($V_e/V_r < 1$)

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14 $V_e = (M_{pr1} + M_{pr2})/l_n = 90202.132$

with

$M_{pr1} = \text{Max}(M_{u1+}, M_{u1-}) = 1.3530E+008$

$M_{u1+} = 1.3530E+008$, is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$M_{u1-} = 1.3530E+008$, is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(M_{u2+}, M_{u2-}) = 1.3530E+008$

$M_{u2+} = 1.3530E+008$, is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

$M_{u2-} = 1.3530E+008$, is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

Calculation of M_{u1+}

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), M_u

$M_u = 1.3530E+008$

$\phi = 0.90757121$

$\phi' = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$

conf. factor $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$\phi' = \phi \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of l_b, min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$\phi = 1$

$d_b = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$
 $f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)
 $t = 1.00$
 $s = 0.80$
 $e = 1.00$
 $cb = 25.00$
 $K_{tr} = 1.14232$
 $A_{tr} = \frac{1}{2} * \text{Area of stirrup} = 123.3701$
 $s = 360.00$
 $n = 12.00$

Calculation of μ_{1-}

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ
 $\mu = 1.3530E+008$

 $\phi = 0.90757121$
 $\lambda = 0.80580716$
error of function (3.68), Biskinis Phd = 28928.286
From 5A.2, TBDY: $f_{cc} = f_c' * c = 36.81095$
conf. factor $c = 1.84055$
 $f_c = 20.00$
From 10.3.5, ASCE 41-17, Final value of f_y : $f_y * \text{Min}(1, 1.25 * (l_b/d)^{2/3}) = 288.6089$
 $l_b/d = 0.26792599$
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00155946$
 $N = 4771.233$
 $A_c = 125663.706$
 $\phi * \text{Min}(1, 1.25 * (l_b/d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/d

Lap Length: $l_b/d = 0.26792599$
 $l_b = 300.00$
 $l_d = 1119.712$
Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
 $\phi = 1$

$db = 18.00$
Mean strength value of all re-bars: $f_y = 555.55$
 $f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)
 $t = 1.00$
 $s = 0.80$
 $e = 1.00$
 $cb = 25.00$
 $K_{tr} = 1.14232$
 $A_{tr} = \frac{1}{2} * \text{Area of stirrup} = 123.3701$
 $s = 360.00$
 $n = 12.00$

Calculation of μ_{2+}

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ
 $\mu = 1.3530E+008$

 $\phi = 0.90757121$
 $\lambda = 0.80580716$
error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$

conf. factor $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$db = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$cb = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of μ_2

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ

$\mu = 1.3530E+008$

$= 0.90757121$

$' = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$

conf. factor $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$db = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

t = 1.00
s = 0.80
e = 1.00
cb = 25.00
Ktr = 1.14232
Atr = /2 * Area of stirrup = 123.3701
s = 360.00
n = 12.00

Calculation of Shear Strength $V_r = \text{Min}(V_{r1}, V_{r2}) = 298161.965$

Calculation of Shear Strength at edge 1, $V_{r1} = 298161.965$

$V_{r1} = V_{Col}$ ((10.3), ASCE 41-17) = knl*VCol0

VCol0 = 298161.965

knl = 1 (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ '
where V_f is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)
 $f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

M/Vd = 2.00

$\mu_u = 5.9321525E-012$

$\nu_u = 7.4135260E-048$

d = 0.8*D = 320.00

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14: $V_s = 0.00$

$A_v = /2 * A_{stirrup} = 123370.055$

$f_y = 444.44$

s = 360.00

V_s is multiplied by Col = 0.00

s/d = 1.125

V_f ((11-3)-(11.4), ACI 440) = 194961.134

f = 0.95, for fully-wrapped sections

wf/sf = 1 (FRP strips adjacent to one another).

In (11.3) $\sin + \cos$ is replaced with $(\cot + \cot) \sin \alpha$ which is more a generalised expression,
where α is the angle of the crack direction (see KANEPE).

This later relation, considered as a function $V_f(\alpha)$, is implemented for every different fiber orientation α_i ,
as well as for 2 crack directions, $\alpha = 45^\circ$ and $\alpha = -45^\circ$ to take into consideration the cyclic seismic loading.

orientation 1: $\alpha = 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \alpha)|, |V_f(-45, \alpha)|)$, with:

total thickness per orientation, $t_{f1} = NL * t / N_{Dir} = 1.016$

$d_{fv} = d$ (figure 11.2, ACI 440) = 370.00

f_{fe} ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_{e} = 0.004$, from (11.6a), ACI 440

with $f_u = 0.01$

From (11-11), ACI 440: $V_s + V_f \leq 238930.50$

$b_w * d = *d * d / 4 = 80424.772$

Calculation of Shear Strength at edge 2, $V_{r2} = 298161.965$

$V_{r2} = V_{Col}$ ((10.3), ASCE 41-17) = knl*VCol0

VCol0 = 298161.965

knl = 1 (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ '
where V_f is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

M/Vd = 2.00

$\mu_u = 5.9321525E-012$

$V_u = 7.4135260E-048$
 $d = 0.8 \cdot D = 320.00$
 $Nu = 4771.233$
 $Ag = 125663.706$
 From (11.5.4.8), ACI 318-14: $V_s = 0.00$
 $A_v = \frac{1}{2} \cdot A_{stirrup} = 123370.055$
 $f_y = 444.44$
 $s = 360.00$
 V_s is multiplied by $Col = 0.00$
 $s/d = 1.125$
 V_f ((11-3)-(11.4), ACI 440) = 194961.134
 $f = 0.95$, for fully-wrapped sections
 $w_f/s_f = 1$ (FRP strips adjacent to one another).
 In (11.3) $\sin \theta + \cos \theta$ is replaced with $(\cot \theta + \cot \alpha) \sin \alpha$ which is more a generalised expression,
 where θ is the angle of the crack direction (see KANEPE).
 This later relation, considered as a function $V_f(\theta, \alpha)$, is implemented for every different cyclic fiber orientation α_i ,
 as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.
 orientation 1: $\theta = b1 + 90^\circ = 90.00$
 $V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \theta)|)$, with:
 total thickness per orientation, $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$
 $d_{fv} = d$ (figure 11.2, ACI 440) = 370.00
 f_{fe} ((11-5), ACI 440) = 259.312
 $E_f = 64828.00$
 $f_e = 0.004$, from (11.6a), ACI 440
 with $f_u = 0.01$
 From (11-11), ACI 440: $V_s + V_f \leq 238930.50$
 $b_w \cdot d = \frac{1}{4} \cdot d \cdot d = 80424.772$

 End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1
 At local axis: 2

 Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1
 At local axis: 2
 Integration Section: (a)
 Section Type: rccs

Constant Properties

 Knowledge Factor, $\gamma = 1.00$
 Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE 41-17.
 Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17
 Consequently:
 Existing material of Primary Member: Concrete Strength, $f_c = f_{cm} = 20.00$
 Existing material of Primary Member: Steel Strength, $f_s = f_{sm} = 444.44$
 Concrete Elasticity, $E_c = 21019.039$
 Steel Elasticity, $E_s = 200000.00$
 Diameter, $D = 400.00$
 Cover Thickness, $c = 25.00$
 Element Length, $L = 3000.00$
 Primary Member
 Ribbed Bars
 Ductile Steel
 Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)
 Longitudinal Bars With Ends Lapped Starting at the End Sections
 Lap Length $l_b = 300.00$
 FRP Wrapping Data
 Type: Carbon
 Cured laminate properties (design values)
 Thickness, $t = 1.016$
 Tensile Strength, $f_{fu} = 1055.00$
 Tensile Modulus, $E_f = 64828.00$
 Elongation, $e_{fu} = 0.01$
 Number of directions, $\text{NoDir} = 1$

Fiber orientations, θ_i : 0.00°
Number of layers, NL = 1
Radius of rounding corners, R = 40.00

Stepwise Properties

Bending Moment, M = 4.7721810E-010
Shear Force, V2 = -3472.528
Shear Force, V3 = -1.2919419E-013
Axial Force, F = -4769.844
Longitudinal Reinforcement Area Distribution (in 2 divisions)
-Tension: A_{st} = 1272.345
-Compression: A_{sc} = 1781.283
Longitudinal Reinforcement Area Distribution (in 3 divisions)
-Tension: $A_{st,ten}$ = 1017.876
-Compression: $A_{sc,com}$ = 1017.876
-Middle: $A_{st,mid}$ = 1017.876
Mean Diameter of Tension Reinforcement, D_{bL} = 18.00

Existing component: From table 7-7, ASCE 41_17: Final chord rotation Capacity $\phi_{u,R} = \phi_u = 0.00931721$
 $\phi_u = \phi_y + \phi_p = 0.00931721$

- Calculation of ϕ_y -

$\phi_y = (M_y * L_s / 3) / E_{eff} = 0.00931721$ ((4.29), Biskinis Phd)
 $M_y = 1.4766E+008$
 $L_s = M/V$ (with $L_s > 0.1 * L$ and $L_s < 2 * L$) = 1500.00
From table 10.5, ASCE 41_17: $E_{eff} = factor * E_c * I_g = 7.9240E+012$
factor = 0.30
 $A_g = 125663.706$
 $f_c' = 20.00$
N = 4769.844
 $E_c * I_g = 2.6413E+013$

Calculation of Yielding Moment M_y

Calculation of ϕ_y and M_y according to (7) - (8) in Biskinis and Fardis

$M_y = \min(M_{y,ten}, M_{y,com}) = 1.4766E+008$
 $\phi_y = 8.1688007E-006$
 $M_{y,ten}$ (8c) = 1.4766E+008
 $\phi_{y,ten}$ (7c) = 72.40642
error of function (7c) = 0.00117228
 $M_{y,com}$ (8d) = 3.7493E+008
 $\phi_{y,com}$ (7d) = 69.91126
error of function (7d) = 0.00301342
with ((10.1), ASCE 41-17) $\phi_y = \min(\phi_y, 1.25 * \phi_y * (I_b / I_d)^{2/3}) = 0.0022222$
 $\phi_{y,eco} = 0.002$
apl = 0.45 ((9c) in Biskinis and Fardis for FRP Wrap)
d1 = 44.00
R = 200.00
v = 0.00157305
N = 4769.844
Ac = 125663.706
((10.1), ASCE 41-17) $\phi_y = \min(\phi_y, 1.25 * \phi_y * (I_b / I_d)^{2/3}) = 0.44757577$
with f_c^* ((12.3), ACI 440) = 24.12975
 $f_c = 20.00$
fl = 1.3173
k = 1
Effective FRP thickness, $t_f = NL * t * \cos(\theta_1) = 1.016$

$\epsilon_{fe}((12.5) \text{ and } (12.7)) = 0.004$
 $f_u = 0.01$
 $E_f = 64828.00$

Calculation of ratio l_b/l_d

Lap Length: $l_d/l_{d,min} = 0.33490748$

$l_b = 300.00$

$l_d = 895.7698$

Calculation of l according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
 $= 1$

$d_b = 18.00$

Mean strength value of all re-bars: $f_y = 444.44$

$f'_c = 20.00$, but $f'_c^{0.5} \leq 8.3 \text{ MPa}$ (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

- Calculation of ρ -

From table 10-9: $\rho = 0.00$

with:

- Columns controlled by inadequate development or splicing along the clear height because $l_b/l_d < 1$

shear control ratio $V_y E / V_{col} O E = 0.30252729$

$d = 0.00$

$s = 0.00$

$t = \frac{2 \cdot A_v}{d_c \cdot s} + \frac{4 \cdot t_f}{D} \cdot \left(\frac{f_{fe}}{f_s} \right) = 0.00721126$

$A_v = 78.53982$, is the area of the circular stirrup

$d_c = D - 2 \cdot \text{cover} - \text{Hoop Diameter} = 340.00$

The term $\frac{2 \cdot t_f}{b_w} \cdot \left(\frac{f_{fe}}{f_s} \right)$ is implemented to account for FRP contribution

where $f = \frac{2 \cdot t_f}{b_w}$ is FRP ratio (EC8 - 3, A.4.4.3(6)) and f_{fe}/f_s normalises f to steel strength

All these variables have already been given in Shear control ratio calculation.

$N_{UD} = 4769.844$

$A_g = 125663.706$

$f'_c E = 20.00$

$f_{yt} E = f_{yl} E = 444.44$

$\rho_l = \frac{\text{Area}_{Tot_Long_Rein}}{A_g} = 0.0243$

$f'_c E = 20.00$

End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (a)

Calculation No. 3

column C1, Floor 1

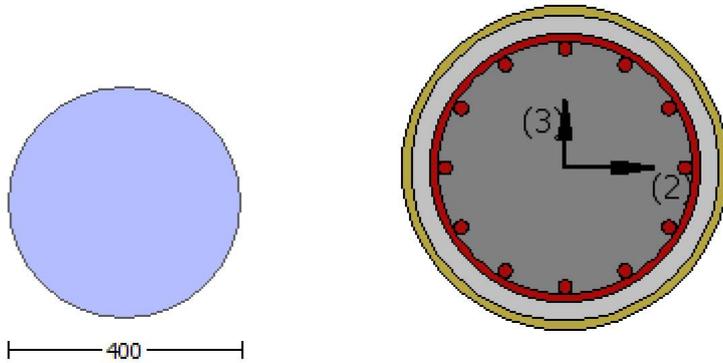
Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Shear capacity VRd

Edge: Start

Local Axis: (3)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (a)

Section Type: rccs

Constant Properties

Knowledge Factor, $\gamma = 1.00$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE 41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength, $f_c = f_{c_lower_bound} = 16.00$

Existing material of Primary Member: Steel Strength, $f_s = f_{s_lower_bound} = 400.00$

Concrete Elasticity, $E_c = 21019.039$

Steel Elasticity, $E_s = 200000.00$

#####

Note: Especially for the calculation of γ for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE 41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE 41-17).

Existing material: Concrete Strength, $f_c = f_{cm} = 20.00$

Existing material: Steel Strength, $f_s = f_{sm} = 444.44$

#####

Diameter, $D = 400.00$

Cover Thickness, $c = 25.00$

Element Length, $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length $l_o = l_b = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness, $t = 1.016$

Tensile Strength, $f_{fu} = 1055.00$

Tensile Modulus, $E_f = 64828.00$

Elongation, $e_{fu} = 0.01$

Number of directions, $NoDir = 1$

Fiber orientations, θ_i : 0.00°
Number of layers, NL = 1
Radius of rounding corners, R = 40.00

Stepwise Properties

EDGE -A-
Bending Moment, M_a = 4.7721810E-010
Shear Force, V_a = -1.2919419E-013
EDGE -B-
Bending Moment, M_b = -8.9325839E-011
Shear Force, V_b = 1.2919419E-013
BOTH EDGES
Axial Force, F = -4769.844
Longitudinal Reinforcement Area Distribution (in 2 divisions)
-Tension: A_{st} = 1272.345
-Compression: A_{sc} = 1781.283
Longitudinal Reinforcement Area Distribution (in 3 divisions)
-Tension: $A_{st,ten}$ = 1017.876
-Compression: $A_{sc,com}$ = 1017.876
-Middle: $A_{st,mid}$ = 1017.876
Mean Diameter of Tension Reinforcement, $D_{bL,ten}$ = 18.00

Existing component: From table 7-7, ASCE 41_17: Final Shear Capacity $V_R = \phi V_n = 286337.204$
 V_n ((10.3), ASCE 41-17) = $k_n \phi V_{CoI} = 286337.204$
 $V_{CoI} = 286337.204$
 $k_n = 1.00$
displacement_ductility_demand = 0.00

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + \phi V_f$ '
where V_f is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)
 $f_c' = 16.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)
 $M / Vd = 2.00$
 $M_u = 4.7721810E-010$
 $V_u = 1.2919419E-013$
 $d = 0.8 \cdot D = 320.00$
 $N_u = 4769.844$
 $A_g = 125663.706$
From (11.5.4.8), ACI 318-14: $V_s = 0.00$
 $A_v = \phi / 2 \cdot A_{stirrup} = 123370.055$
 $f_y = 400.00$
 $s = 360.00$
 V_s is multiplied by $\phi_{CoI} = 0.00$
 $s/d = 1.125$
 V_f ((11-3)-(11.4), ACI 440) = 194961.134
 $\phi = 0.95$, for fully-wrapped sections
 $w_f / s_f = 1$ (FRP strips adjacent to one another).
In (11.3) $\sin \theta + \cos \theta$ is replaced with $(\cot \theta + \csc \theta) \sin \alpha$ which is more a generalised expression,
where θ is the angle of the crack direction (see KANEPE).
This later relation, considered as a function $V_f(\theta, \alpha)$, is implemented for every different fiber orientation α_i ,
as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.
orientation 1: $\theta = \theta_1 + 90^\circ = 90.00$
 $V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, \theta_1)|)$, with:
total thickness per orientation, $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$
 $d_{fv} = d$ (figure 11.2, ACI 440) = 370.00
 f_{fe} ((11-5), ACI 440) = 259.312
 $E_f = 64828.00$
 $f_e = 0.004$, from (11.6a), ACI 440
with $f_u = 0.01$
From (11-11), ACI 440: $V_s + V_f \leq 213705.936$
 $b_w \cdot d = \phi \cdot d^2 / 4 = 80424.772$

displacement_ductility_demand is calculated as δ / y

- Calculation of δ / y for END A -
for rotation axis 2 and integ. section (a)

From analysis, chord rotation $\theta = 2.1684378E-020$
 $y = (M_y * L_s / 3) / E_{eff} = 0.00931721$ ((4.29), Biskinis Phd))
 $M_y = 1.4766E+008$
 $L_s = M/V$ (with $L_s > 0.1 * L$ and $L_s < 2 * L$) = 1500.00
From table 10.5, ASCE 41_17: $E_{eff} = factor * E_c * I_g = 7.9240E+012$
factor = 0.30
Ag = 125663.706
fc' = 20.00
N = 4769.844
 $E_c * I_g = 2.6413E+013$

Calculation of Yielding Moment M_y

Calculation of δ / y and M_y according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y_ten}, M_{y_com}) = 1.4766E+008$
 $y = 8.1688007E-006$
 M_{y_ten} (8c) = 1.4766E+008
 y_{ten} (7c) = 72.40642
error of function (7c) = 0.00117228
 M_{y_com} (8d) = 3.7493E+008
 y_{com} (7d) = 69.91126
error of function (7d) = 0.00301342
with ((10.1), ASCE 41-17) $e_y = \text{Min}(e_y, 1.25 * e_y * (l_b / l_d)^{2/3}) = 0.0022222$
eco = 0.002
apl = 0.45 ((9c) in Biskinis and Fardis for FRP Wrap)
d1 = 44.00
R = 200.00
v = 0.00157305
N = 4769.844
Ac = 125663.706
((10.1), ASCE 41-17) $e = \text{Min}(e, 1.25 * e * (l_b / l_d)^{2/3}) = 0.44757577$
with fc' ((12.3), ACI 440) = 24.12975
fc = 20.00
fl = 1.3173
k = 1
Effective FRP thickness, tf = NL * t * Cos(b1) = 1.016
efe ((12.5) and (12.7)) = 0.004
fu = 0.01
Ef = 64828.00

Calculation of ratio l_b / l_d

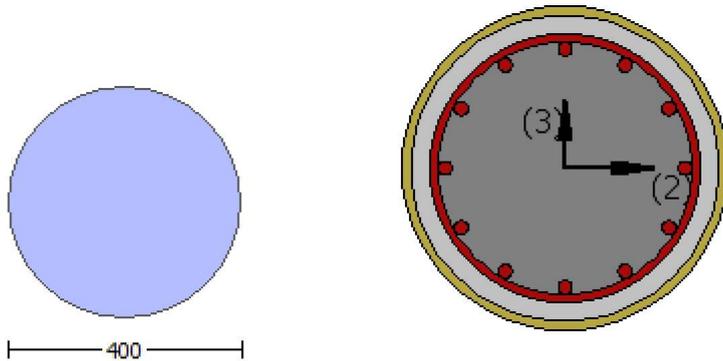
Lap Length: $l_d / l_{d,min} = 0.33490748$
 $l_b = 300.00$
 $l_d = 895.7698$
Calculation of I according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
= 1
db = 18.00
Mean strength value of all re-bars: $f_y = 444.44$
fc' = 20.00, but $fc'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)
t = 1.00
s = 0.80
e = 1.00
cb = 25.00

$K_{tr} = 1.14232$
 $A_{tr} = \sqrt{2} * \text{Area of stirrup} = 123.3701$
 $s = 360.00$
 $n = 12.00$

 End Of Calculation of Shear Capacity for element: column CC1 of floor 1
 At local axis: 3
 Integration Section: (a)

Calculation No. 4

column C1, Floor 1
 Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)
 Analysis: Uniform +X
 Check: Chord rotation capacity (θ)
 Edge: Start
 Local Axis: (3)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1
 At Shear local axis: 3
 (Bending local axis: 2)
 Section Type: rccs

Constant Properties

 Knowledge Factor, $\gamma = 1.00$
 Mean strength values are used for both shear and moment calculations.
 Consequently:
 Existing material of Primary Member: Concrete Strength, $f_c = f_{cm} = 20.00$
 Existing material of Primary Member: Steel Strength, $f_s = f_{sm} = 444.44$
 Concrete Elasticity, $E_c = 21019.039$
 Steel Elasticity, $E_s = 200000.00$

 Note: Especially for the calculation of moment strengths,
 the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14
 Existing material: Steel Strength, $f_s = 1.25 * f_{sm} = 555.55$

#####

Diameter, D = 400.00
 Cover Thickness, c = 25.00
 Mean Confinement Factor overall section = 1.84055
 Element Length, L = 3000.00
 Primary Member
 Ribbed Bars
 Ductile Steel
 Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)
 Longitudinal Bars With Ends Lapped Starting at the End Sections
 Lap Length l_o = 300.00
 FRP Wrapping Data
 Type: Carbon
 Cured laminate properties (design values)
 Thickness, t = 1.016
 Tensile Strength, f_{fu} = 1055.00
 Tensile Modulus, E_f = 64828.00
 Elongation, e_{fu} = 0.01
 Number of directions, NoDir = 1
 Fiber orientations, b_i : 0.00°
 Number of layers, NL = 1
 Radius of rounding corners, R = 40.00

 Stepwise Properties

At local axis: 3
 EDGE -A-
 Shear Force, V_a = 1.2107607E-031
 EDGE -B-
 Shear Force, V_b = -1.2107607E-031
 BOTH EDGES
 Axial Force, F = -4771.233
 Longitudinal Reinforcement Area Distribution (in 2 divisions)
 -Tension: A_{sl} = 0.00
 -Compression: A_{slc} = 3053.628
 Longitudinal Reinforcement Area Distribution (in 3 divisions)
 -Tension: $A_{sl,ten}$ = 1017.876
 -Compression: $A_{sl,com}$ = 1017.876
 -Middle: $A_{sl,mid}$ = 1017.876

 Calculation of Shear Capacity ratio , V_e/V_r = 0.30252729
 Member Controlled by Flexure ($V_e/V_r < 1$)
 Calculation of Shear Demand from fig. R18.6.5, ACI 318-14 $V_e = (M_{pr1} + M_{pr2})/l_n = 90202.132$
 with
 $M_{pr1} = \text{Max}(Mu_{1+} , Mu_{1-}) = 1.3530E+008$
 $Mu_{1+} = 1.3530E+008$, is the ultimate moment strength at the edge 1 of the member in the actual moment direction
 which is defined for the static loading combination
 $Mu_{1-} = 1.3530E+008$, is the ultimate moment strength at the edge 1 of the member in the opposite moment
 direction which is defined for the static loading combination
 $M_{pr2} = \text{Max}(Mu_{2+} , Mu_{2-}) = 1.3530E+008$
 $Mu_{2+} = 1.3530E+008$, is the ultimate moment strength at the edge 2 of the member in the actual moment direction
 which is defined for the the static loading combination
 $Mu_{2-} = 1.3530E+008$, is the ultimate moment strength at the edge 2 of the member in the opposite moment
 direction which is defined for the the static loading combination

 Calculation of Mu_{1+}

 Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu
 $Mu = 1.3530E+008$

 = 0.90757121

$\rho = 0.80580716$
 error of function (3.68), Biskinis Phd = 28928.286
 From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$
 conf. factor $c = 1.84055$
 $f_c = 20.00$
 From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$
 $l_b/l_d = 0.26792599$
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00155946$
 $N = 4771.233$
 $A_c = 125663.706$
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$
 $l_b = 300.00$
 $l_d = 1119.712$
 Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
 $= 1$
 $d_b = 18.00$
 Mean strength value of all re-bars: $f_y = 555.55$
 $f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)
 $t = 1.00$
 $s = 0.80$
 $e = 1.00$
 $cb = 25.00$
 $K_{tr} = 1.14232$
 $A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$
 $s = 360.00$
 $n = 12.00$

Calculation of μ_1

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ
 $\mu = 1.3530E+008$

$= 0.90757121$
 $\rho = 0.80580716$
 error of function (3.68), Biskinis Phd = 28928.286
 From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$
 conf. factor $c = 1.84055$
 $f_c = 20.00$
 From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$
 $l_b/l_d = 0.26792599$
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00155946$
 $N = 4771.233$
 $A_c = 125663.706$
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$
 $l_b = 300.00$
 $l_d = 1119.712$
 Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
 $= 1$
 $d_b = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$
 $f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)
 $t = 1.00$
 $s = 0.80$
 $e = 1.00$
 $cb = 25.00$
 $K_{tr} = 1.14232$
 $A_{tr} = \frac{1}{2} * \text{Area of stirrup} = 123.3701$
 $s = 360.00$
 $n = 12.00$

Calculation of μ_{2+}

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ
 $\mu = 1.3530E+008$

 $\phi = 0.90757121$
 $\lambda = 0.80580716$
error of function (3.68), Biskinis Phd = 28928.286
From 5A.2, TBDY: $f_{cc} = f_c' * c = 36.81095$
conf. factor $c = 1.84055$
 $f_c = 20.00$
From 10.3.5, ASCE 41-17, Final value of f_y : $f_y * \text{Min}(1, 1.25 * (l_b / l_d)^{2/3}) = 288.6089$
 $l_b / l_d = 0.26792599$
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00155946$
 $N = 4771.233$
 $A_c = 125663.706$
 $\phi * \text{Min}(1, 1.25 * (l_b / l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b / l_d

Lap Length: $l_b / l_d = 0.26792599$
 $l_b = 300.00$
 $l_d = 1119.712$
Calculation of l_b, min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
 $\phi = 1$

$db = 18.00$
Mean strength value of all re-bars: $f_y = 555.55$
 $f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)
 $t = 1.00$
 $s = 0.80$
 $e = 1.00$
 $cb = 25.00$
 $K_{tr} = 1.14232$
 $A_{tr} = \frac{1}{2} * \text{Area of stirrup} = 123.3701$
 $s = 360.00$
 $n = 12.00$

Calculation of μ_{2-}

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ
 $\mu = 1.3530E+008$

 $\phi = 0.90757121$
 $\lambda = 0.80580716$
error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$
conf. factor $c = 1.84055$
 $f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$
 $l_b/l_d = 0.26792599$
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00155946$
 $N = 4771.233$
 $A_c = 125663.706$
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$d_b = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of Shear Strength $V_r = \text{Min}(V_{r1}, V_{r2}) = 298161.965$

Calculation of Shear Strength at edge 1, $V_{r1} = 298161.965$

$V_{r1} = V_{Col}$ ((10.3), ASCE 41-17) = $k_{nl} \cdot V_{Col0}$

$V_{Col0} = 298161.965$

$k_{nl} = 1$ (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where V_f is the contribution of FRPs (11.3), ACI 440).

$= 1$ (normal-weight concrete)

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.2899604E-011$

$\nu_u = 1.2107607E-031$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14: $V_s = 0.00$

$A_v = \frac{1}{2} \cdot A_{stirrup} = 123370.055$

$f_y = 444.44$

$s = 360.00$

V_s is multiplied by $Col = 0.00$

$s/d = 1.125$

V_f ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$, for fully-wrapped sections

$w_f/s_f = 1$ (FRP strips adjacent to one another).

In (11.3) $\sin \theta + \cos \theta$ is replaced with $(\cot \theta + \csc \theta) \sin \alpha$ which is more a generalised expression, where θ is the angle of the crack direction (see KANEPE).

This later relation, considered as a function $V_f(\theta, \alpha)$, is implemented for every different fiber orientation α_i , as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.

orientation 1: $\theta = 45^\circ + 90^\circ = 135^\circ$

$V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, a1)|)$, with:
 total thickness per orientation, $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$
 $d_{fv} = d$ (figure 11.2, ACI 440) = 370.00
 f_{fe} ((11-5), ACI 440) = 259.312
 $E_f = 64828.00$
 $f_e = 0.004$, from (11.6a), ACI 440
 with $f_u = 0.01$
 From (11-11), ACI 440: $V_s + V_f \leq 238930.50$
 $b_w \cdot d = \sqrt[4]{d^3} = 80424.772$

Calculation of Shear Strength at edge 2, $V_{r2} = 298161.965$
 $V_{r2} = V_{Col}$ ((10.3), ASCE 41-17) = $k_{nl} \cdot V_{Col0}$
 $V_{Col0} = 298161.965$
 $k_{nl} = 1$ (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ '
 where V_f is the contribution of FRPs (11.3), ACI 440).

$\beta = 1$ (normal-weight concrete)
 $f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)
 $M / Vd = 2.00$
 $\mu_u = 1.2899604E-011$
 $\nu_u = 1.2107607E-031$
 $d = 0.8 \cdot D = 320.00$
 $N_u = 4771.233$
 $A_g = 125663.706$
 From (11.5.4.8), ACI 318-14: $V_s = 0.00$
 $A_v = \sqrt{2} \cdot A_{stirrup} = 123370.055$
 $f_y = 444.44$
 $s = 360.00$

V_s is multiplied by $\beta_{Col} = 0.00$
 $s/d = 1.125$
 V_f ((11-3)-(11.4), ACI 440) = 194961.134
 $f = 0.95$, for fully-wrapped sections
 $w_f / s_f = 1$ (FRP strips adjacent to one another).

In (11.3) $\sin \theta + \cos \theta$ is replaced with $(\cot \theta + \csc \theta) \sin \alpha$ which is more a generalised expression,
 where θ is the angle of the crack direction (see KANEPE).
 This later relation, considered as a function $V_f(\theta, \alpha)$, is implemented for every different fiber orientation α_i ,
 as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.

orientation 1: $\theta = b1 + 90^\circ = 90.00$
 $V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, a1)|)$, with:
 total thickness per orientation, $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$
 $d_{fv} = d$ (figure 11.2, ACI 440) = 370.00
 f_{fe} ((11-5), ACI 440) = 259.312
 $E_f = 64828.00$
 $f_e = 0.004$, from (11.6a), ACI 440
 with $f_u = 0.01$
 From (11-11), ACI 440: $V_s + V_f \leq 238930.50$
 $b_w \cdot d = \sqrt[4]{d^3} = 80424.772$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1
 At local axis: 3

Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1
 At Shear local axis: 2
 (Bending local axis: 3)
 Section Type: rccs

Constant Properties

Knowledge Factor, $\beta = 1.00$
 Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength, $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength, $f_s = f_{sm} = 444.44$

Concrete Elasticity, $E_c = 21019.039$

Steel Elasticity, $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength, $f_s = 1.25 \cdot f_{sm} = 555.55$

#####

Diameter, $D = 400.00$

Cover Thickness, $c = 25.00$

Mean Confinement Factor overall section = 1.84055

Element Length, $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length $l_o = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness, $t = 1.016$

Tensile Strength, $f_{fu} = 1055.00$

Tensile Modulus, $E_f = 64828.00$

Elongation, $e_{fu} = 0.01$

Number of directions, $N_{oDir} = 1$

Fiber orientations, $b_i = 0.00^\circ$

Number of layers, $N_L = 1$

Radius of rounding corners, $R = 40.00$

Stepwise Properties

At local axis: 2

EDGE -A-

Shear Force, $V_a = -7.4135260E-048$

EDGE -B-

Shear Force, $V_b = 7.4135260E-048$

BOTH EDGES

Axial Force, $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension: $A_{sl,t} = 0.00$

-Compression: $A_{sl,c} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension: $A_{sl,ten} = 1017.876$

-Compression: $A_{sl,com} = 1017.876$

-Middle: $A_{sl,mid} = 1017.876$

Calculation of Shear Capacity ratio, $V_e/V_r = 0.30252729$

Member Controlled by Flexure ($V_e/V_r < 1$)

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14 $V_e = (M_{pr1} + M_{pr2})/l_n = 90202.132$

with

$M_{pr1} = \text{Max}(M_{u1+}, M_{u1-}) = 1.3530E+008$

$M_{u1+} = 1.3530E+008$, is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$M_{u1-} = 1.3530E+008$, is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(M_{u2+}, M_{u2-}) = 1.3530E+008$

$M_{u2+} = 1.3530E+008$, is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination

$M_{u2-} = 1.3530E+008$, is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

Calculation of Mu1+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu
Mu = 1.3530E+008

= 0.90757121
' = 0.80580716
error of function (3.68), Biskinis Phd = 28928.286
From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$
conf. factor $c = 1.84055$
 $f_c = 20.00$
From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 288.6089$
 $l_b/d = 0.26792599$
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00155946$
 $N = 4771.233$
 $A_c = 125663.706$
= $\cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/d

Lap Length: $l_b/d = 0.26792599$
 $l_b = 300.00$
 $l_d = 1119.712$
Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
= 1
 $d_b = 18.00$
Mean strength value of all re-bars: $f_y = 555.55$
 $f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3 \text{ MPa}$ (22.5.3.1, ACI 318-14)
 $t = 1.00$
 $s = 0.80$
 $e = 1.00$
 $cb = 25.00$
 $K_{tr} = 1.14232$
 $A_{tr} = \sqrt{2} \cdot \text{Area of stirrup} = 123.3701$
 $s = 360.00$
 $n = 12.00$

Calculation of Mu1-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu
Mu = 1.3530E+008

= 0.90757121
' = 0.80580716
error of function (3.68), Biskinis Phd = 28928.286
From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$
conf. factor $c = 1.84055$
 $f_c = 20.00$
From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 288.6089$
 $l_b/d = 0.26792599$
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00155946$
 $N = 4771.233$
 $A_c = 125663.706$
= $\cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of l_b, \min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$d_b = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f'_c = 20.00$, but $f_c^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of μ_{2+}

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ

$\mu = 1.3530E+008$

$= 0.90757121$

$' = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY: $f_{cc} = f'_c \cdot c = 36.81095$

conf. factor $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of l_b, \min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$d_b = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f'_c = 20.00$, but $f_c^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of μ_2 -

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ

$\mu = 1.3530E+008$

$\mu = 0.90757121$

$\mu = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$

conf. factor $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$d_b = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f_c' = 20.00$, but $f_c^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of Shear Strength $V_r = \text{Min}(V_{r1}, V_{r2}) = 298161.965$

Calculation of Shear Strength at edge 1, $V_{r1} = 298161.965$

$V_{r1} = V_{Co1}$ ((10.3), ASCE 41-17) = $k_{nl} \cdot V_{Co10}$

$V_{Co10} = 298161.965$

$k_{nl} = 1$ (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where V_f is the contribution of FRPs (11.3), ACI 440).

$= 1$ (normal-weight concrete)

$f_c' = 20.00$, but $f_c^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 5.9321525E-012$

$V_u = 7.4135260E-048$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14: $V_s = 0.00$

$A_v = \frac{1}{2} \cdot A_{\text{stirrup}} = 123370.055$

$f_y = 444.44$

$s = 360.00$

Vs is multiplied by Col = 0.00

$$s/d = 1.125$$

$$V_f \text{ ((11-3)-(11.4), ACI 440) } = 194961.134$$

f = 0.95, for fully-wrapped sections

wf/sf = 1 (FRP strips adjacent to one another).

In (11.3) $\sin \theta + \cos \theta$ is replaced with $(\cot \theta + \csc \theta) \sin \alpha$ which is more a generalised expression, where θ is the angle of the crack direction (see KANEPE).

This later relation, considered as a function $V_f(\theta, \alpha)$, is implemented for every different fiber orientation α_i , as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.

$$\text{orientation 1: } \theta = b1 + 90^\circ = 90.00$$

$$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \theta)|), \text{ with:}$$

$$\text{total thickness per orientation, } t_{f1} = NL * t / \text{NoDir} = 1.016$$

$$d_{fv} = d \text{ (figure 11.2, ACI 440) } = 370.00$$

$$f_{fe} \text{ ((11-5), ACI 440) } = 259.312$$

$$E_f = 64828.00$$

$$f_e = 0.004, \text{ from (11.6a), ACI 440}$$

$$\text{with } f_u = 0.01$$

$$\text{From (11-11), ACI 440: } V_s + V_f \leq 238930.50$$

$$b_w * d = \rho * d^2 / 4 = 80424.772$$

Calculation of Shear Strength at edge 2, $V_{r2} = 298161.965$

$$V_{r2} = V_{Col} \text{ ((10.3), ASCE 41-17) } = k_{nl} * V_{Col0}$$

$$V_{Col0} = 298161.965$$

$$k_{nl} = 1 \text{ (zero step-static loading)}$$

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ ' where V_f is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

$$f_c' = 20.00, \text{ but } f_c'^{0.5} \leq 8.3 \text{ MPa (22.5.3.1, ACI 318-14)}$$

$$M / Vd = 2.00$$

$$M_u = 5.9321525E-012$$

$$V_u = 7.4135260E-048$$

$$d = 0.8 * D = 320.00$$

$$N_u = 4771.233$$

$$A_g = 125663.706$$

$$\text{From (11.5.4.8), ACI 318-14: } V_s = 0.00$$

$$A_v = \rho / 2 * A_{stirup} = 123370.055$$

$$f_y = 444.44$$

$$s = 360.00$$

Vs is multiplied by Col = 0.00

$$s/d = 1.125$$

$$V_f \text{ ((11-3)-(11.4), ACI 440) } = 194961.134$$

f = 0.95, for fully-wrapped sections

wf/sf = 1 (FRP strips adjacent to one another).

In (11.3) $\sin \theta + \cos \theta$ is replaced with $(\cot \theta + \csc \theta) \sin \alpha$ which is more a generalised expression, where θ is the angle of the crack direction (see KANEPE).

This later relation, considered as a function $V_f(\theta, \alpha)$, is implemented for every different fiber orientation α_i , as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.

$$\text{orientation 1: } \theta = b1 + 90^\circ = 90.00$$

$$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \theta)|), \text{ with:}$$

$$\text{total thickness per orientation, } t_{f1} = NL * t / \text{NoDir} = 1.016$$

$$d_{fv} = d \text{ (figure 11.2, ACI 440) } = 370.00$$

$$f_{fe} \text{ ((11-5), ACI 440) } = 259.312$$

$$E_f = 64828.00$$

$$f_e = 0.004, \text{ from (11.6a), ACI 440}$$

$$\text{with } f_u = 0.01$$

$$\text{From (11-11), ACI 440: } V_s + V_f \leq 238930.50$$

$$b_w * d = \rho * d^2 / 4 = 80424.772$$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At local axis: 2

Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (a)

Section Type: rccs

Constant Properties

Knowledge Factor, $k = 1.00$

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE 41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength, $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength, $f_s = f_{sm} = 444.44$

Concrete Elasticity, $E_c = 21019.039$

Steel Elasticity, $E_s = 200000.00$

Diameter, $D = 400.00$

Cover Thickness, $c = 25.00$

Element Length, $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length $l_b = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness, $t = 1.016$

Tensile Strength, $f_{fu} = 1055.00$

Tensile Modulus, $E_f = 64828.00$

Elongation, $e_{fu} = 0.01$

Number of directions, $N_{oDir} = 1$

Fiber orientations, $b_i = 0.00^\circ$

Number of layers, $N_L = 1$

Radius of rounding corners, $R = 40.00$

Stepwise Properties

Bending Moment, $M = -1.0423E+007$

Shear Force, $V_2 = -3472.528$

Shear Force, $V_3 = -1.2919419E-013$

Axial Force, $F = -4769.844$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension: $A_{st} = 1272.345$

-Compression: $A_{sc} = 1781.283$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension: $A_{st,ten} = 1017.876$

-Compression: $A_{st,com} = 1017.876$

-Middle: $A_{st,mid} = 1017.876$

Mean Diameter of Tension Reinforcement, $DbL = 18.00$

Existing component: From table 7-7, ASCE 41_17: Final chord rotation Capacity $u_{R} = \gamma \cdot u = 0.01864342$

$u = \gamma \cdot u + p = 0.01864342$

- Calculation of γ -

$\gamma = (M \cdot L_s / 3) / E_{eff} = 0.01864342$ ((4.29), Biskinis Phd))

$M_y = 1.4766E+008$

$L_s = M/V$ (with $L_s > 0.1 \cdot L$ and $L_s < 2 \cdot L$) = 3001.447

From table 10.5, ASCE 41_17: $E_{eff} = \text{factor} \cdot E_c \cdot I_g = 7.9240E+012$

factor = 0.30

Ag = 125663.706
fc' = 20.00
N = 4769.844
Ec*Ig = 2.6413E+013

Calculation of Yielding Moment My

Calculation of ρ_y and My according to (7) - (8) in Biskinis and Fardis

My = Min(My_ten, My_com) = 1.4766E+008
y = 8.1688007E-006
My_ten (8c) = 1.4766E+008
 _ten (7c) = 72.40642
error of function (7c) = 0.00117228
My_com (8d) = 3.7493E+008
 _com (7d) = 69.91126
error of function (7d) = 0.00301342
with ((10.1), ASCE 41-17) $e_y = \text{Min}(e_y, 1.25 * e_y * (l_b/l_d)^{2/3}) = 0.0022222$
 eco = 0.002
 apl = 0.45 ((9c) in Biskinis and Fardis for FRP Wrap)
 d1 = 44.00
 R = 200.00
 v = 0.00157305
 N = 4769.844
 Ac = 125663.706
 ((10.1), ASCE 41-17) $= \text{Min}(, 1.25 * * (l_b/l_d)^{2/3}) = 0.44757577$
with fc' ((12.3), ACI 440) = 24.12975
 fc = 20.00
 fl = 1.3173
 k = 1
 Effective FRP thickness, $t_f = NL * t * \text{Cos}(b1) = 1.016$
 efe ((12.5) and (12.7)) = 0.004
 fu = 0.01
 Ef = 64828.00

Calculation of ratio lb/l_d

Lap Length: $l_d/l_{d,min} = 0.33490748$
lb = 300.00
ld = 895.7698
Calculation of l according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
 = 1
 db = 18.00
 Mean strength value of all re-bars: fy = 444.44
 fc' = 20.00, but $fc'^{0.5} \leq 8.3 \text{ MPa}$ (22.5.3.1, ACI 318-14)
 t = 1.00
 s = 0.80
 e = 1.00
 cb = 25.00
 Ktr = 1.14232
 Atr = $\sqrt{2} * \text{Area of stirrup} = 123.3701$
 s = 360.00
 n = 12.00

- Calculation of ρ_p -

From table 10-9: $\rho_p = 0.00$

with:

- Columns controlled by inadequate development or splicing along the clear height because $l_b/l_d < 1$
shear control ratio $V_y E / V_{CoI} E = 0.30252729$
d = 0.00
s = 0.00

$$t = 2 \cdot A_v / (d_c \cdot s) + 4 \cdot t_f / D \cdot (f_{fe} / f_s) = 0.00721126$$

$A_v = 78.53982$, is the area of the circular stirrup

$d_c = D - 2 \cdot \text{cover} - \text{Hoop Diameter} = 340.00$

The term $2 \cdot t_f / b_w \cdot (f_{fe} / f_s)$ is implemented to account for FRP contribution

where $f = 2 \cdot t_f / b_w$ is FRP ratio (EC8 - 3, A.4.4.3(6)) and f_{fe} / f_s normalises f to steel strength

All these variables have already been given in Shear control ratio calculation.

$$N_{UD} = 4769.844$$

$$A_g = 125663.706$$

$$f_{cE} = 20.00$$

$$f_{yE} = f_{yI} = 444.44$$

$$p_l = \text{Area_Tot_Long_Rein} / (A_g) = 0.0243$$

$$f_{cE} = 20.00$$

End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (a)

Calculation No. 5

column C1, Floor 1

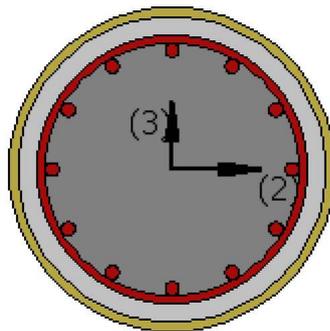
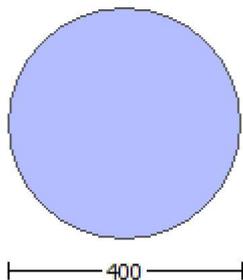
Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Shear capacity V_{Rd}

Edge: End

Local Axis: (2)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (b)

Section Type: rccs

Constant Properties

Knowledge Factor, $\gamma = 1.00$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE 41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength, $f_c = f_{c_lower_bound} = 16.00$

Existing material of Primary Member: Steel Strength, $f_s = f_{s_lower_bound} = 400.00$

Concrete Elasticity, $E_c = 21019.039$

Steel Elasticity, $E_s = 200000.00$

#####

Note: Especially for the calculation of γ for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE 41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE 41-17).

Existing material: Concrete Strength, $f_c = f_{cm} = 20.00$

Existing material: Steel Strength, $f_s = f_{sm} = 444.44$

#####

Diameter, $D = 400.00$

Cover Thickness, $c = 25.00$

Element Length, $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length $l_o = l_b = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness, $t = 1.016$

Tensile Strength, $f_{fu} = 1055.00$

Tensile Modulus, $E_f = 64828.00$

Elongation, $e_{fu} = 0.01$

Number of directions, $NoDir = 1$

Fiber orientations, $b_i = 0.00^\circ$

Number of layers, $NL = 1$

Radius of rounding corners, $R = 40.00$

Stepwise Properties

EDGE -A-

Bending Moment, $M_a = -1.0423E+007$

Shear Force, $V_a = -3472.528$

EDGE -B-

Bending Moment, $M_b = 0.09265189$

Shear Force, $V_b = 3472.528$

BOTH EDGES

Axial Force, $F = -4769.844$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension: $A_{sl,t} = 0.00$

-Compression: $A_{sl,c} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension: $A_{sl,ten} = 1017.876$

-Compression: $A_{sl,com} = 1017.876$

-Middle: $A_{sl,mid} = 1017.876$

Mean Diameter of Tension Reinforcement, $Db_{L,ten} = 18.00$

Existing component: From table 7-7, ASCE 41_17: Final Shear Capacity $V_R = \phi V_n = 286337.204$

V_n ((10.3), ASCE 41-17) = $k_n \phi V_{CoI} = 286337.204$

$V_{CoI} = 286337.204$

$k_n = 1.00$

displacement_ductility_demand = 0.10586773

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_{s+} = \phi \cdot V_f$ ' where V_f is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

$f'_c = 16.00$, but $f'_c \cdot 0.5 \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$
 $\mu_u = 0.09265189$
 $V_u = 3472.528$
 $d = 0.8 \cdot D = 320.00$
 $N_u = 4769.844$
 $A_g = 125663.706$
 From (11.5.4.8), ACI 318-14: $V_s = 0.00$
 $A_v = \sqrt{2} \cdot A_{stirrup} = 123370.055$
 $f_y = 400.00$
 $s = 360.00$
 V_s is multiplied by $\text{Col} = 0.00$
 $s/d = 1.125$
 V_f ((11-3)-(11.4), ACI 440) = 194961.134
 $f = 0.95$, for fully-wrapped sections
 $w_f/s_f = 1$ (FRP strips adjacent to one another).
 In (11.3) $\sin + \cos$ is replaced with $(\cot + \cot a) \sin a$ which is more a generalised expression,
 where a is the angle of the crack direction (see KANEPE).
 This later relation, considered as a function $V_f(\theta, a)$, is implemented for every different fiber orientation a_i ,
 as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.
 orientation 1: $\theta = 45^\circ + 90^\circ = 135^\circ$
 $V_f = \text{Min}(|V_f(45, a_1)|, |V_f(-45, a_1)|)$, with:
 total thickness per orientation, $t_{f1} = N_L \cdot t / N_{Dir} = 1.016$
 $d_{fv} = d$ (figure 11.2, ACI 440) = 370.00
 f_{fe} ((11-5), ACI 440) = 259.312
 $E_f = 64828.00$
 $f_{e} = 0.004$, from (11.6a), ACI 440
 with $f_u = 0.01$
 From (11-11), ACI 440: $V_s + V_f \leq 213705.936$
 $b_w \cdot d = N \cdot d^2 / 4 = 80424.772$

displacement ductility demand is calculated as δ / y

- Calculation of δ / y for END B -
for rotation axis 3 and integ. section (b)

From analysis, chord rotation $\theta = 0.00019728$
 $y = (M_y \cdot L_s / 3) / E_{eff} = 0.00186344$ ((4.29), Biskinis Phd)
 $M_y = 1.4766E+008$
 $L_s = M/V$ (with $L_s > 0.1 \cdot L$ and $L_s < 2 \cdot L$) = 300.00
 From table 10.5, ASCE 41_17: $E_{eff} = \text{factor} \cdot E_c \cdot I_g = 7.9240E+012$
 $\text{factor} = 0.30$
 $A_g = 125663.706$
 $f_c' = 20.00$
 $N = 4769.844$
 $E_c \cdot I_g = 2.6413E+013$

Calculation of Yielding Moment M_y

Calculation of δ / y and M_y according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y_ten}, M_{y_com}) = 1.4766E+008$
 $y = 8.1688007E-006$
 M_{y_ten} (8c) = 1.4766E+008
 δ_{ten} (7c) = 72.40642
 error of function (7c) = 0.00117228
 M_{y_com} (8d) = 3.7493E+008
 δ_{com} (7d) = 69.91126
 error of function (7d) = 0.00301342
 with ((10.1), ASCE 41-17) $e_y = \text{Min}(e_y, 1.25 \cdot e_y \cdot (I_b / I_d)^{2/3}) = 0.0022222$
 $e_{co} = 0.002$
 $a_{pl} = 0.45$ ((9c) in Biskinis and Fardis for FRP Wrap)
 $d_1 = 44.00$
 $R = 200.00$

$v = 0.00157305$
 $N = 4769.844$
 $A_c = 125663.706$
 $((10.1), ASCE 41-17) = \text{Min}(, 1.25 * (lb/ld)^{2/3}) = 0.44757577$
with f_c^* ((12.3), ACI 440) = 24.12975
 $f_c = 20.00$
 $f_l = 1.3173$
 $k = 1$
Effective FRP thickness, $t_f = NL * t * \text{Cos}(b1) = 1.016$
 e_{fe} ((12.5) and (12.7)) = 0.004
 $f_u = 0.01$
 $E_f = 64828.00$

Calculation of ratio lb/ld

Lap Length: $ld/ld, \text{min} = 0.33490748$

$lb = 300.00$

$ld = 895.7698$

Calculation of I according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$db = 18.00$

Mean strength value of all re-bars: $f_y = 444.44$

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3 \text{ MPa}$ (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$cb = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = /2 * \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

End Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (b)

Calculation No. 6

column C1, Floor 1

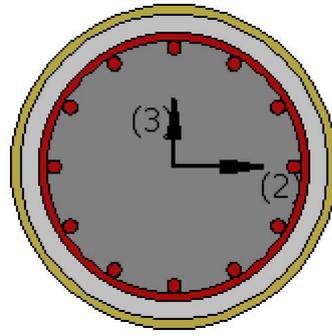
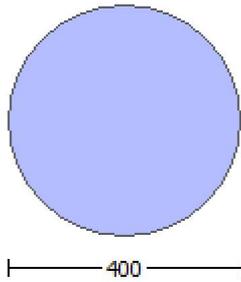
Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity (θ)

Edge: End

Local Axis: (2)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 3
 (Bending local axis: 2)
 Section Type: rccs

Constant Properties

Knowledge Factor, $\gamma = 1.00$
 Mean strength values are used for both shear and moment calculations.
 Consequently:
 Existing material of Primary Member: Concrete Strength, $f_c = f_{cm} = 20.00$
 Existing material of Primary Member: Steel Strength, $f_s = f_{sm} = 444.44$
 Concrete Elasticity, $E_c = 21019.039$
 Steel Elasticity, $E_s = 200000.00$

 Note: Especially for the calculation of moment strengths,
 the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14
 Existing material: Steel Strength, $f_s = 1.25 \cdot f_{sm} = 555.55$

 Diameter, $D = 400.00$
 Cover Thickness, $c = 25.00$
 Mean Confinement Factor overall section = 1.84055
 Element Length, $L = 3000.00$
 Primary Member
 Ribbed Bars
 Ductile Steel
 Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)
 Longitudinal Bars With Ends Lapped Starting at the End Sections
 Lap Length $l_o = 300.00$
 FRP Wrapping Data
 Type: Carbon
 Cured laminate properties (design values)
 Thickness, $t = 1.016$
 Tensile Strength, $f_{fu} = 1055.00$
 Tensile Modulus, $E_f = 64828.00$
 Elongation, $e_{fu} = 0.01$
 Number of directions, $NoDir = 1$
 Fiber orientations, $b_i: 0.00^\circ$
 Number of layers, $NL = 1$
 Radius of rounding corners, $R = 40.00$

Stepwise Properties

At local axis: 3
 EDGE -A-
 Shear Force, $V_a = 1.2107607E-031$
 EDGE -B-
 Shear Force, $V_b = -1.2107607E-031$
 BOTH EDGES
 Axial Force, $F = -4771.233$
 Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension: Aslt = 0.00

-Compression: Aslc = 3053.628

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension: Asl,ten = 1017.876

-Compression: Asl,com = 1017.876

-Middle: Asl,mid = 1017.876

Calculation of Shear Capacity ratio , $V_e/V_r = 0.30252729$

Member Controlled by Flexure ($V_e/V_r < 1$)

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14 $V_e = (M_{pr1} + M_{pr2})/l_n = 90202.132$

with

$M_{pr1} = \text{Max}(M_{u1+} , M_{u1-}) = 1.3530E+008$

$M_{u1+} = 1.3530E+008$, is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$M_{u1-} = 1.3530E+008$, is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(M_{u2+} , M_{u2-}) = 1.3530E+008$

$M_{u2+} = 1.3530E+008$, is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

$M_{u2-} = 1.3530E+008$, is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

Calculation of M_{u1+}

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), M_u

$M_u = 1.3530E+008$

$\phi = 0.90757121$

$\phi' = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$

conf. factor $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$\phi = \phi' \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of l_b, min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$\phi = 1$

$d_b = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3 \text{ MPa}$ (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of Mu1-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu
Mu = 1.3530E+008

= 0.90757121
' = 0.80580716
error of function (3.68), Biskinis Phd = 28928.286
From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$
conf. factor $c = 1.84055$
 $f_c = 20.00$
From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$
 $l_b/l_d = 0.26792599$
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00155946$
 $N = 4771.233$
 $Ac = 125663.706$
= $\cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$
 $l_b = 300.00$
 $l_d = 1119.712$
Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
= 1
 $db = 18.00$
Mean strength value of all re-bars: $f_y = 555.55$
 $f_c' = 20.00$, but $f_c^{0.5} \leq 8.3 \text{ MPa}$ (22.5.3.1, ACI 318-14)
 $t = 1.00$
 $s = 0.80$
 $e = 1.00$
 $cb = 25.00$
 $K_{tr} = 1.14232$
 $A_{tr} = \sqrt{2} \cdot \text{Area of stirrup} = 123.3701$
 $s = 360.00$
 $n = 12.00$

Calculation of Mu2+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu
Mu = 1.3530E+008

= 0.90757121
' = 0.80580716
error of function (3.68), Biskinis Phd = 28928.286
From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$
conf. factor $c = 1.84055$
 $f_c = 20.00$
From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$
 $l_b/l_d = 0.26792599$
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00155946$
 $N = 4771.233$
 $Ac = 125663.706$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.28805051$$

Calculation of ratio l_b/d

Lap Length: $l_b/d = 0.26792599$

$l_b = 300.00$

$d = 1119.712$

Calculation of l_b, min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$d_b = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f'_c = 20.00$, but $f_c^{0.5} \leq 8.3 \text{ MPa}$ (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of μ_2

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ

$\mu = 1.3530 \times 10^8$

$= 0.90757121$

$' = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY: $f_{cc} = f'_c \cdot c = 36.81095$

conf. factor $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 288.6089$

$l_b/d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/d

Lap Length: $l_b/d = 0.26792599$

$l_b = 300.00$

$d = 1119.712$

Calculation of l_b, min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$d_b = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f'_c = 20.00$, but $f_c^{0.5} \leq 8.3 \text{ MPa}$ (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of Shear Strength $V_r = \text{Min}(V_{r1}, V_{r2}) = 298161.965$

Calculation of Shear Strength at edge 1, $V_{r1} = 298161.965$

$V_{r1} = V_{Col}$ ((10.3), ASCE 41-17) = $k_{nl} * V_{Col0}$

$V_{Col0} = 298161.965$

$k_{nl} = 1$ (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ '
where V_f is the contribution of FRPs (11.3), ACI 440).

 $k_c = 1$ (normal-weight concrete)

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.2899604E-011$

$\mu_v = 1.2107607E-031$

$d = 0.8 * D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14: $V_s = 0.00$

$A_v = \frac{1}{2} * A_{stirrup} = 123370.055$

$f_y = 444.44$

$s = 360.00$

V_s is multiplied by $\phi_{Col} = 0.00$

$s/d = 1.125$

V_f ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$, for fully-wrapped sections

$w_f/s_f = 1$ (FRP strips adjacent to one another).

In (11.3) $\sin \theta + \cos \theta$ is replaced with $(\cot \theta + \csc \theta) \sin \alpha$ which is more a generalised expression,
where θ is the angle of the crack direction (see KANEPE).

This later relation, considered as a function $V_f(\theta, \alpha)$, is implemented for every different fiber orientation α_i ,
as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.

orientation 1: $\theta = b_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \alpha)|)$, with:

total thickness per orientation, $t_{f1} = N_L * t / N_{oDir} = 1.016$

$d_{fv} = d$ (figure 11.2, ACI 440) = 370.00

f_{fe} ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$, from (11.6a), ACI 440

with $f_u = 0.01$

From (11-11), ACI 440: $V_s + V_f \leq 238930.50$

$b_w * d = \frac{1}{4} * d * d = 80424.772$

Calculation of Shear Strength at edge 2, $V_{r2} = 298161.965$

$V_{r2} = V_{Col}$ ((10.3), ASCE 41-17) = $k_{nl} * V_{Col0}$

$V_{Col0} = 298161.965$

$k_{nl} = 1$ (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ '
where V_f is the contribution of FRPs (11.3), ACI 440).

 $k_c = 1$ (normal-weight concrete)

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.2899604E-011$

$\mu_v = 1.2107607E-031$

$d = 0.8 * D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14: $V_s = 0.00$

$A_v = \frac{1}{2} * A_{stirrup} = 123370.055$

$f_y = 444.44$

$s = 360.00$

V_s is multiplied by $\phi_{Col} = 0.00$

$s/d = 1.125$

Vf ((11-3)-(11.4), ACI 440) = 194961.134

f = 0.95, for fully-wrapped sections

wf/sf = 1 (FRP strips adjacent to one another).

In (11.3) $\sin \theta + \cos \theta$ is replaced with $(\cot \theta + \cot \alpha) \sin \alpha$ which is more a generalised expression, where θ is the angle of the crack direction (see KANEPE).

This later relation, considered as a function Vf(θ), is implemented for every different fiber orientation ai, as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.

orientation 1: $\theta_1 = b_1 + 90^\circ = 90.00$

Vf = Min(|Vf(45, θ_1)|, |Vf(-45, θ_1)|), with:

total thickness per orientation, $t_{f1} = NL * t / NoDir = 1.016$

dfv = d (figure 11.2, ACI 440) = 370.00

ffe ((11-5), ACI 440) = 259.312

Ef = 64828.00

fe = 0.004, from (11.6a), ACI 440

with fu = 0.01

From (11-11), ACI 440: $V_s + V_f \leq 238930.50$

$b_w * d = \frac{V_s * d}{4} = 80424.772$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At local axis: 3

Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 2

(Bending local axis: 3)

Section Type: rccs

Constant Properties

Knowledge Factor, $\phi = 1.00$

Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength, $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength, $f_s = f_{sm} = 444.44$

Concrete Elasticity, $E_c = 21019.039$

Steel Elasticity, $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength, $f_s = 1.25 * f_{sm} = 555.55$

#####

Diameter, $D = 400.00$

Cover Thickness, $c = 25.00$

Mean Confinement Factor overall section = 1.84055

Element Length, $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length $l_o = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness, $t = 1.016$

Tensile Strength, $f_{fu} = 1055.00$

Tensile Modulus, $E_f = 64828.00$

Elongation, $e_{fu} = 0.01$

Number of directions, $NoDir = 1$

Fiber orientations, $b_i: 0.00^\circ$

Number of layers, $NL = 1$

Radius of rounding corners, $R = 40.00$

Stepwise Properties

At local axis: 2

EDGE -A-

Shear Force, $V_a = -7.4135260E-048$

EDGE -B-

Shear Force, $V_b = 7.4135260E-048$

BOTH EDGES

Axial Force, $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension: $As_t = 0.00$

-Compression: $As_c = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension: $As_{t,ten} = 1017.876$

-Compression: $As_{c,com} = 1017.876$

-Middle: $As_{mid} = 1017.876$

Calculation of Shear Capacity ratio, $V_e/V_r = 0.30252729$

Member Controlled by Flexure ($V_e/V_r < 1$)

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14 $V_e = (M_{pr1} + M_{pr2})/l_n = 90202.132$

with

$M_{pr1} = \text{Max}(Mu_{1+}, Mu_{1-}) = 1.3530E+008$

$Mu_{1+} = 1.3530E+008$, is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$Mu_{1-} = 1.3530E+008$, is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(Mu_{2+}, Mu_{2-}) = 1.3530E+008$

$Mu_{2+} = 1.3530E+008$, is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination

$Mu_{2-} = 1.3530E+008$, is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

Calculation of Mu_{1+}

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), M_u

$M_u = 1.3530E+008$

$= 0.90757121$

$' = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY: $f_{cc} = f_c^* c = 36.81095$

conf. factor $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of f_y : $f_y * \text{Min}(1, 1.25 * (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$Ac = 125663.706$

$= * \text{Min}(1, 1.25 * (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of l_b, min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$db = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$
 $f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)
 $t = 1.00$
 $s = 0.80$
 $e = 1.00$
 $cb = 25.00$
 $K_{tr} = 1.14232$
 $A_{tr} = \frac{1}{2} * \text{Area of stirrup} = 123.3701$
 $s = 360.00$
 $n = 12.00$

Calculation of μ_{1-}

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ
 $\mu = 1.3530E+008$

 $\mu = 0.90757121$
 $\mu' = 0.80580716$
error of function (3.68), Biskinis Phd = 28928.286
From 5A.2, TBDY: $f_{cc} = f_c' * c = 36.81095$
conf. factor $c = 1.84055$
 $f_c = 20.00$
From 10.3.5, ASCE 41-17, Final value of f_y : $f_y * \text{Min}(1, 1.25 * (l_b/d)^{2/3}) = 288.6089$
 $l_b/d = 0.26792599$
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00155946$
 $N = 4771.233$
 $A_c = 125663.706$
 $\mu = \mu' * \text{Min}(1, 1.25 * (l_b/d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/d

Lap Length: $l_b/d = 0.26792599$
 $l_b = 300.00$
 $l_d = 1119.712$
Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
 $\mu = 1$

$db = 18.00$
Mean strength value of all re-bars: $f_y = 555.55$
 $f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)
 $t = 1.00$
 $s = 0.80$
 $e = 1.00$
 $cb = 25.00$
 $K_{tr} = 1.14232$
 $A_{tr} = \frac{1}{2} * \text{Area of stirrup} = 123.3701$
 $s = 360.00$
 $n = 12.00$

Calculation of μ_{2+}

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ
 $\mu = 1.3530E+008$

 $\mu = 0.90757121$
 $\mu' = 0.80580716$
error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$

conf. factor $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$db = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$cb = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of μ_2

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ

$\mu = 1.3530E+008$

$= 0.90757121$

$' = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$

conf. factor $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$db = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

t = 1.00
s = 0.80
e = 1.00
cb = 25.00
Ktr = 1.14232
Atr = /2 * Area of stirrup = 123.3701
s = 360.00
n = 12.00

Calculation of Shear Strength $V_r = \text{Min}(V_{r1}, V_{r2}) = 298161.965$

Calculation of Shear Strength at edge 1, $V_{r1} = 298161.965$

$V_{r1} = V_{Col}$ ((10.3), ASCE 41-17) = $k_{nl} * V_{Col0}$

$V_{Col0} = 298161.965$

$k_{nl} = 1$ (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ '
where V_f is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$M / Vd = 2.00$

$\mu_u = 5.9321525E-012$

$\nu_u = 7.4135260E-048$

$d = 0.8 * D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14: $V_s = 0.00$

$A_v = /2 * A_{stirrup} = 123370.055$

$f_y = 444.44$

$s = 360.00$

V_s is multiplied by $\phi_{Col} = 0.00$

$s/d = 1.125$

V_f ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$, for fully-wrapped sections

$w_f / s_f = 1$ (FRP strips adjacent to one another).

In (11.3) $\sin \theta + \cos \theta$ is replaced with $(\cot \theta + \csc \theta) \sin \alpha$ which is more a generalised expression,
where θ is the angle of the crack direction (see KANEPE).

This later relation, considered as a function $V_f(\theta, \alpha)$, is implemented for every different fiber orientation α_i ,
as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.

orientation 1: $\theta = 45^\circ$ and $\alpha = 90^\circ$

$V_f = \text{Min}(|V_f(45, 90)|, |V_f(-45, 90)|)$, with:

total thickness per orientation, $t_{f1} = N_L * t / N_{oDir} = 1.016$

$d_{fv} = d$ (figure 11.2, ACI 440) = 370.00

f_{fe} ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_{e} = 0.004$, from (11.6a), ACI 440

with $f_u = 0.01$

From (11-11), ACI 440: $V_s + V_f \leq 238930.50$

$b_w * d = f_{fe} * d^2 / 4 = 80424.772$

Calculation of Shear Strength at edge 2, $V_{r2} = 298161.965$

$V_{r2} = V_{Col}$ ((10.3), ASCE 41-17) = $k_{nl} * V_{Col0}$

$V_{Col0} = 298161.965$

$k_{nl} = 1$ (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ '
where V_f is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$M / Vd = 2.00$

$\mu_u = 5.9321525E-012$

$V_u = 7.4135260E-048$
 $d = 0.8 \cdot D = 320.00$
 $Nu = 4771.233$
 $Ag = 125663.706$
 From (11.5.4.8), ACI 318-14: $V_s = 0.00$
 $A_v = \frac{1}{2} \cdot A_{stirrup} = 123370.055$
 $f_y = 444.44$
 $s = 360.00$
 V_s is multiplied by $Col = 0.00$
 $s/d = 1.125$
 V_f ((11-3)-(11.4), ACI 440) = 194961.134
 $f = 0.95$, for fully-wrapped sections
 $w_f/s_f = 1$ (FRP strips adjacent to one another).
 In (11.3) $\sin \theta + \cos \theta$ is replaced with $(\cot \theta + \cot \alpha) \sin \alpha$ which is more a generalised expression,
 where θ is the angle of the crack direction (see KANEPE).
 This later relation, considered as a function $V_f(\theta, \alpha)$, is implemented for every different cyclic fiber orientation α_i ,
 as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.
 orientation 1: $\theta = b_1 + 90^\circ = 90.00$
 $V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \theta)|)$, with:
 total thickness per orientation, $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$
 $d_{fv} = d$ (figure 11.2, ACI 440) = 370.00
 f_{fe} ((11-5), ACI 440) = 259.312
 $E_f = 64828.00$
 $f_e = 0.004$, from (11.6a), ACI 440
 with $f_u = 0.01$
 From (11-11), ACI 440: $V_s + V_f \leq 238930.50$
 $b_w \cdot d = \frac{1}{4} \cdot d \cdot d = 80424.772$

 End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1
 At local axis: 2

 Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1
 At local axis: 2
 Integration Section: (b)
 Section Type: rccs

Constant Properties

 Knowledge Factor, $\gamma = 1.00$
 Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE 41-17.
 Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17
 Consequently:
 Existing material of Primary Member: Concrete Strength, $f_c = f_{cm} = 20.00$
 Existing material of Primary Member: Steel Strength, $f_s = f_{sm} = 444.44$
 Concrete Elasticity, $E_c = 21019.039$
 Steel Elasticity, $E_s = 200000.00$
 Diameter, $D = 400.00$
 Cover Thickness, $c = 25.00$
 Element Length, $L = 3000.00$
 Primary Member
 Ribbed Bars
 Ductile Steel
 Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)
 Longitudinal Bars With Ends Lapped Starting at the End Sections
 Lap Length $l_b = 300.00$
 FRP Wrapping Data
 Type: Carbon
 Cured laminate properties (design values)
 Thickness, $t = 1.016$
 Tensile Strength, $f_{fu} = 1055.00$
 Tensile Modulus, $E_f = 64828.00$
 Elongation, $e_{fu} = 0.01$
 Number of directions, $\text{NoDir} = 1$

Fiber orientations, β_i : 0.00°
Number of layers, NL = 1
Radius of rounding corners, R = 40.00

Stepwise Properties

Bending Moment, M = -8.9325839E-011
Shear Force, V2 = 3472.528
Shear Force, V3 = 1.2919419E-013
Axial Force, F = -4769.844
Longitudinal Reinforcement Area Distribution (in 2 divisions)
-Tension: A_{st} = 0.00
-Compression: A_{sc} = 3053.628
Longitudinal Reinforcement Area Distribution (in 3 divisions)
-Tension: $A_{st,ten}$ = 1017.876
-Compression: $A_{sc,com}$ = 1017.876
-Middle: $A_{st,mid}$ = 1017.876
Mean Diameter of Tension Reinforcement, D_{bL} = 18.00

Existing component: From table 7-7, ASCE 41_17: Final chord rotation Capacity $u_{,R} = u = 0.00931721$
 $u = y + p = 0.00931721$

- Calculation of y -

$y = (M_y * L_s / 3) / E_{eff} = 0.00931721$ ((4.29), Biskinis Phd)
 $M_y = 1.4766E+008$
 $L_s = M/V$ (with $L_s > 0.1 * L$ and $L_s < 2 * L$) = 1500.00
From table 10.5, ASCE 41_17: $E_{eff} = factor * E_c * I_g = 7.9240E+012$
factor = 0.30
 $A_g = 125663.706$
 $f_c' = 20.00$
 $N = 4769.844$
 $E_c * I_g = 2.6413E+013$

Calculation of Yielding Moment M_y

Calculation of y and M_y according to (7) - (8) in Biskinis and Fardis

$M_y = \min(M_{y,ten}, M_{y,com}) = 1.4766E+008$
 $y = 8.1688007E-006$
 $M_{y,ten}$ (8c) = 1.4766E+008
 y_{ten} (7c) = 72.40642
error of function (7c) = 0.00117228
 $M_{y,com}$ (8d) = 3.7493E+008
 y_{com} (7d) = 69.91126
error of function (7d) = 0.00301342
with ((10.1), ASCE 41-17) $e_y = \min(e_y, 1.25 * e_y * (I_b / I_d)^{2/3}) = 0.0022222$
 $e_{co} = 0.002$
 $a_{pl} = 0.45$ ((9c) in Biskinis and Fardis for FRP Wrap)
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00157305$
 $N = 4769.844$
 $A_c = 125663.706$
((10.1), ASCE 41-17) $e_y = \min(e_y, 1.25 * e_y * (I_b / I_d)^{2/3}) = 0.44757577$
with f_c^* ((12.3), ACI 440) = 24.12975
 $f_c = 20.00$
 $f_l = 1.3173$
 $k = 1$
Effective FRP thickness, $t_f = NL * t * \cos(\beta_1) = 1.016$

$\epsilon_{fe} ((12.5) \text{ and } (12.7)) = 0.004$
 $f_u = 0.01$
 $E_f = 64828.00$

Calculation of ratio l_b/l_d

Lap Length: $l_d/l_{d,min} = 0.33490748$

$l_b = 300.00$

$l_d = 895.7698$

Calculation of λ according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
 $\lambda = 1$

$d_b = 18.00$

Mean strength value of all re-bars: $f_y = 444.44$

$f'_c = 20.00$, but $f'_c^{0.5} \leq 8.3 \text{ MPa}$ (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} * \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

- Calculation of ρ -

From table 10-9: $\rho = 0.00$

with:

- Columns controlled by inadequate development or splicing along the clear height because $l_b/l_d < 1$

shear control ratio $V_y E / V_{col} O E = 0.30252729$

$d = 0.00$

$s = 0.00$

$t = \frac{2 * A_v}{d_c * s} + \frac{4 * t_f}{D} * \frac{f_{fe}}{f_s} = 0.00721126$

$A_v = 78.53982$, is the area of the circular stirrup

$d_c = D - 2 * \text{cover} - \text{Hoop Diameter} = 340.00$

The term $\frac{2 * t_f}{b_w} * \frac{f_{fe}}{f_s}$ is implemented to account for FRP contribution

where $f = \frac{2 * t_f}{b_w}$ is FRP ratio (EC8 - 3, A.4.4.3(6)) and f_{fe}/f_s normalises f to steel strength

All these variables have already been given in Shear control ratio calculation.

$N_{UD} = 4769.844$

$A_g = 125663.706$

$f'_c E = 20.00$

$f_{yt} E = f_{yl} E = 444.44$

$\rho_l = \frac{\text{Area}_{Tot_Long_Rein}}{A_g} = 0.0243$

$f'_c E = 20.00$

End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (b)

Calculation No. 7

column C1, Floor 1

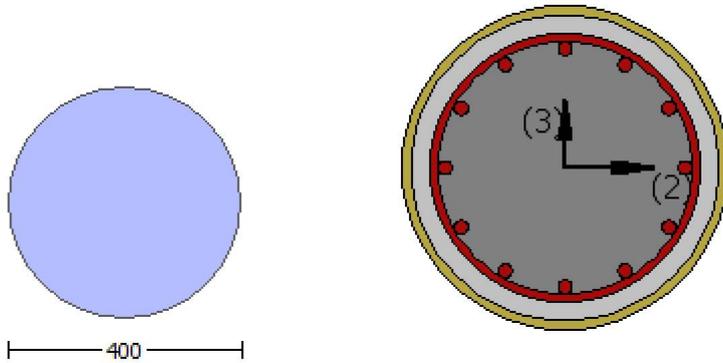
Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Shear capacity VRd

Edge: End

Local Axis: (3)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (b)

Section Type: rccs

Constant Properties

Knowledge Factor, $\gamma = 1.00$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE 41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength, $f_c = f_{c_lower_bound} = 16.00$

Existing material of Primary Member: Steel Strength, $f_s = f_{s_lower_bound} = 400.00$

Concrete Elasticity, $E_c = 21019.039$

Steel Elasticity, $E_s = 200000.00$

#####

Note: Especially for the calculation of γ for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE 41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE 41-17).

Existing material: Concrete Strength, $f_c = f_{cm} = 20.00$

Existing material: Steel Strength, $f_s = f_{sm} = 444.44$

#####

Diameter, $D = 400.00$

Cover Thickness, $c = 25.00$

Element Length, $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length $l_o = l_b = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness, $t = 1.016$

Tensile Strength, $f_{fu} = 1055.00$

Tensile Modulus, $E_f = 64828.00$

Elongation, $e_{fu} = 0.01$

Number of directions, $NoDir = 1$

Fiber orientations, b_i : 0.00°
Number of layers, NL = 1
Radius of rounding corners, R = 40.00

Stepwise Properties

EDGE -A-
Bending Moment, M_a = 4.7721810E-010
Shear Force, V_a = -1.2919419E-013
EDGE -B-
Bending Moment, M_b = -8.9325839E-011
Shear Force, V_b = 1.2919419E-013
BOTH EDGES
Axial Force, F = -4769.844
Longitudinal Reinforcement Area Distribution (in 2 divisions)
-Tension: A_{st} = 0.00
-Compression: A_{sc} = 3053.628
Longitudinal Reinforcement Area Distribution (in 3 divisions)
-Tension: $A_{st,ten}$ = 1017.876
-Compression: $A_{sc,com}$ = 1017.876
-Middle: $A_{st,mid}$ = 1017.876
Mean Diameter of Tension Reinforcement, $D_{bL,ten}$ = 18.00

Existing component: From table 7-7, ASCE 41_17: Final Shear Capacity $V_R = \phi V_n = 286337.204$
 V_n ((10.3), ASCE 41-17) = $k_n \phi V_{CoI} = 286337.204$
 $V_{CoI} = 286337.204$
 $k_n = 1.00$
displacement_ductility_demand = 0.00

NOTE: In expression (10-3) ' $V_s = A_v \phi_f y d / s$ ' is replaced by ' $V_s + \phi_f V_f$ '
where V_f is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)
 $f_c' = 16.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)
 $M/Vd = 2.00$
 $M_u = 8.9325839E-011$
 $V_u = 1.2919419E-013$
 $d = 0.8 \cdot D = 320.00$
 $N_u = 4769.844$
 $A_g = 125663.706$
From (11.5.4.8), ACI 318-14: $V_s = 0.00$
 $A_v = \phi / 2 \cdot A_{stirrup} = 123370.055$
 $f_y = 400.00$
 $s = 360.00$
 V_s is multiplied by $\phi_{CoI} = 0.00$
 $s/d = 1.125$
 V_f ((11-3)-(11.4), ACI 440) = 194961.134
 $\phi = 0.95$, for fully-wrapped sections
 $w_f/s_f = 1$ (FRP strips adjacent to one another).
In (11.3) $\sin \theta + \cos \theta$ is replaced with $(\cot \theta + \csc \theta) \sin \alpha$ which is more a generalised expression,
where θ is the angle of the crack direction (see KANEPE).
This later relation, considered as a function $V_f(\theta, \alpha)$, is implemented for every different fiber orientation α_i ,
as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.
orientation 1: $\theta = b_1 + 90^\circ = 90.00$
 $V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \theta)|)$, with:
total thickness per orientation, $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$
 $d_{fv} = d$ (figure 11.2, ACI 440) = 370.00
 f_{fe} ((11-5), ACI 440) = 259.312
 $E_f = 64828.00$
 $f_e = 0.004$, from (11.6a), ACI 440
with $f_u = 0.01$
From (11-11), ACI 440: $V_s + V_f \leq 213705.936$
 $b_w \cdot d = \phi \cdot d^2 / 4 = 80424.772$

displacement_ductility_demand is calculated as δ / y

- Calculation of δ / y for END B -
for rotation axis 2 and integ. section (b)

From analysis, chord rotation $\theta = 1.1597918E-020$
 $y = (M_y * L_s / 3) / E_{eff} = 0.00931721$ ((4.29), Biskinis Phd)
 $M_y = 1.4766E+008$
 $L_s = M/V$ (with $L_s > 0.1 * L$ and $L_s < 2 * L$) = 1500.00
From table 10.5, ASCE 41_17: $E_{eff} = factor * E_c * I_g = 7.9240E+012$
factor = 0.30
Ag = 125663.706
fc' = 20.00
N = 4769.844
 $E_c * I_g = 2.6413E+013$

Calculation of Yielding Moment M_y

Calculation of δ / y and M_y according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y_ten}, M_{y_com}) = 1.4766E+008$
 $y = 8.1688007E-006$
 M_{y_ten} (8c) = 1.4766E+008
 δ / y (7c) = 72.40642
error of function (7c) = 0.00117228
 M_{y_com} (8d) = 3.7493E+008
 δ / y (7d) = 69.91126
error of function (7d) = 0.00301342
with ((10.1), ASCE 41-17) $e_y = \text{Min}(e_y, 1.25 * e_y * (l_b / l_d)^{2/3}) = 0.0022222$
 $e_{co} = 0.002$
apl = 0.45 ((9c) in Biskinis and Fardis for FRP Wrap)
d1 = 44.00
R = 200.00
v = 0.00157305
N = 4769.844
Ac = 125663.706
((10.1), ASCE 41-17) $e_s = \text{Min}(e_s, 1.25 * e_s * (l_b / l_d)^{2/3}) = 0.44757577$
with fc' ((12.3), ACI 440) = 24.12975
fc = 20.00
fl = 1.3173
k = 1
Effective FRP thickness, $t_f = N * L * \text{Cos}(b1) = 1.016$
efe ((12.5) and (12.7)) = 0.004
fu = 0.01
Ef = 64828.00

Calculation of ratio l_b / l_d

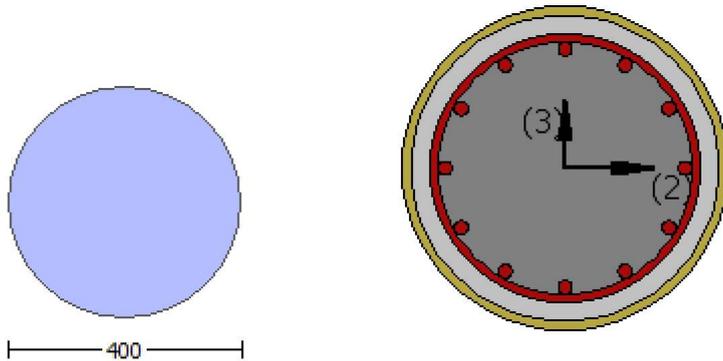
Lap Length: $l_d / l_{d,min} = 0.33490748$
 $l_b = 300.00$
 $l_d = 895.7698$
Calculation of l according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
 $l = 1$
db = 18.00
Mean strength value of all re-bars: $f_y = 444.44$
fc' = 20.00, but $fc'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)
t = 1.00
s = 0.80
e = 1.00
cb = 25.00

$K_{tr} = 1.14232$
 $A_{tr} = \sqrt{2} * \text{Area of stirrup} = 123.3701$
 $s = 360.00$
 $n = 12.00$

 End Of Calculation of Shear Capacity for element: column CC1 of floor 1
 At local axis: 3
 Integration Section: (b)

Calculation No. 8

column C1, Floor 1
 Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)
 Analysis: Uniform +X
 Check: Chord rotation capacity (θ)
 Edge: End
 Local Axis: (3)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1
 At Shear local axis: 3
 (Bending local axis: 2)
 Section Type: rccs

Constant Properties

 Knowledge Factor, $\gamma = 1.00$
 Mean strength values are used for both shear and moment calculations.
 Consequently:
 Existing material of Primary Member: Concrete Strength, $f_c = f_{cm} = 20.00$
 Existing material of Primary Member: Steel Strength, $f_s = f_{sm} = 444.44$
 Concrete Elasticity, $E_c = 21019.039$
 Steel Elasticity, $E_s = 200000.00$

 Note: Especially for the calculation of moment strengths,
 the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14
 Existing material: Steel Strength, $f_s = 1.25 * f_{sm} = 555.55$
 #####

Diameter, $D = 400.00$
Cover Thickness, $c = 25.00$
Mean Confinement Factor overall section = 1.84055
Element Length, $L = 3000.00$
Primary Member
Ribbed Bars
Ductile Steel
Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)
Longitudinal Bars With Ends Lapped Starting at the End Sections
Lap Length $l_o = 300.00$
FRP Wrapping Data
Type: Carbon
Cured laminate properties (design values)
Thickness, $t = 1.016$
Tensile Strength, $f_{fu} = 1055.00$
Tensile Modulus, $E_f = 64828.00$
Elongation, $e_{fu} = 0.01$
Number of directions, $NoDir = 1$
Fiber orientations, $bi: 0.00^\circ$
Number of layers, $NL = 1$
Radius of rounding corners, $R = 40.00$

Stepwise Properties

At local axis: 3
EDGE -A-
Shear Force, $V_a = 1.2107607E-031$
EDGE -B-
Shear Force, $V_b = -1.2107607E-031$
BOTH EDGES
Axial Force, $F = -4771.233$
Longitudinal Reinforcement Area Distribution (in 2 divisions)
-Tension: $As_t = 0.00$
-Compression: $As_c = 3053.628$
Longitudinal Reinforcement Area Distribution (in 3 divisions)
-Tension: $As_{t,ten} = 1017.876$
-Compression: $As_{c,com} = 1017.876$
-Middle: $As_{c,mid} = 1017.876$

Calculation of Shear Capacity ratio , $V_e/V_r = 0.30252729$
Member Controlled by Flexure ($V_e/V_r < 1$)
Calculation of Shear Demand from fig. R18.6.5, ACI 318-14 $V_e = (M_{pr1} + M_{pr2})/l_n = 90202.132$
with
 $M_{pr1} = \text{Max}(Mu_{1+} , Mu_{1-}) = 1.3530E+008$
 $Mu_{1+} = 1.3530E+008$, is the ultimate moment strength at the edge 1 of the member in the actual moment direction
which is defined for the static loading combination
 $Mu_{1-} = 1.3530E+008$, is the ultimate moment strength at the edge 1 of the member in the opposite moment
direction which is defined for the static loading combination
 $M_{pr2} = \text{Max}(Mu_{2+} , Mu_{2-}) = 1.3530E+008$
 $Mu_{2+} = 1.3530E+008$, is the ultimate moment strength at the edge 2 of the member in the actual moment direction
which is defined for the the static loading combination
 $Mu_{2-} = 1.3530E+008$, is the ultimate moment strength at the edge 2 of the member in the opposite moment
direction which is defined for the the static loading combination

Calculation of Mu_{1+}

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu
 $Mu = 1.3530E+008$

= 0.90757121

$\rho = 0.80580716$
 error of function (3.68), Biskinis Phd = 28928.286
 From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$
 conf. factor $c = 1.84055$
 $f_c = 20.00$
 From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$
 $l_b/l_d = 0.26792599$
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00155946$
 $N = 4771.233$
 $A_c = 125663.706$
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$
 $l_b = 300.00$
 $l_d = 1119.712$
 Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
 $= 1$
 $d_b = 18.00$
 Mean strength value of all re-bars: $f_y = 555.55$
 $f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)
 $t = 1.00$
 $s = 0.80$
 $e = 1.00$
 $cb = 25.00$
 $K_{tr} = 1.14232$
 $A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$
 $s = 360.00$
 $n = 12.00$

Calculation of μ_1

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ
 $\mu = 1.3530E+008$

$\rho = 0.90757121$
 $\rho = 0.80580716$
 error of function (3.68), Biskinis Phd = 28928.286
 From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$
 conf. factor $c = 1.84055$
 $f_c = 20.00$
 From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$
 $l_b/l_d = 0.26792599$
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00155946$
 $N = 4771.233$
 $A_c = 125663.706$
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$
 $l_b = 300.00$
 $l_d = 1119.712$
 Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
 $= 1$
 $d_b = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$
 $f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)
 $t = 1.00$
 $s = 0.80$
 $e = 1.00$
 $cb = 25.00$
 $K_{tr} = 1.14232$
 $A_{tr} = \frac{1}{2} * \text{Area of stirrup} = 123.3701$
 $s = 360.00$
 $n = 12.00$

Calculation of μ_{2+}

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ
 $\mu = 1.3530E+008$

 $\phi = 0.90757121$
 $\lambda = 0.80580716$
error of function (3.68), Biskinis Phd = 28928.286
From 5A.2, TBDY: $f_{cc} = f_c' * c = 36.81095$
conf. factor $c = 1.84055$
 $f_c = 20.00$
From 10.3.5, ASCE 41-17, Final value of f_y : $f_y * \text{Min}(1, 1.25 * (l_b / l_d)^{2/3}) = 288.6089$
 $l_b / l_d = 0.26792599$
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00155946$
 $N = 4771.233$
 $A_c = 125663.706$
 $\phi * \text{Min}(1, 1.25 * (l_b / l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b / l_d

Lap Length: $l_b / l_d = 0.26792599$
 $l_b = 300.00$
 $l_d = 1119.712$
Calculation of l_b, min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
 $\phi = 1$

$db = 18.00$
Mean strength value of all re-bars: $f_y = 555.55$
 $f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)
 $t = 1.00$
 $s = 0.80$
 $e = 1.00$
 $cb = 25.00$
 $K_{tr} = 1.14232$
 $A_{tr} = \frac{1}{2} * \text{Area of stirrup} = 123.3701$
 $s = 360.00$
 $n = 12.00$

Calculation of μ_{2-}

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ
 $\mu = 1.3530E+008$

 $\phi = 0.90757121$
 $\lambda = 0.80580716$
error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$

conf. factor $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$d_b = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of Shear Strength $V_r = \text{Min}(V_{r1}, V_{r2}) = 298161.965$

Calculation of Shear Strength at edge 1, $V_{r1} = 298161.965$

$V_{r1} = V_{Col}$ ((10.3), ASCE 41-17) = $k_{nl} \cdot V_{Col0}$

$V_{Col0} = 298161.965$

$k_{nl} = 1$ (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where V_f is the contribution of FRPs (11.3), ACI 440).

$= 1$ (normal-weight concrete)

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.2899604E-011$

$\nu_u = 1.2107607E-031$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14: $V_s = 0.00$

$A_v = \frac{1}{2} \cdot A_{stirrup} = 123370.055$

$f_y = 444.44$

$s = 360.00$

V_s is multiplied by $\text{Col} = 0.00$

$s/d = 1.125$

V_f ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$, for fully-wrapped sections

$w_f/s_f = 1$ (FRP strips adjacent to one another).

In (11.3) $\sin \theta + \cos \theta$ is replaced with $(\cot \theta + \csc \theta) \sin \alpha$ which is more a generalised expression, where θ is the angle of the crack direction (see KANEPE).

This later relation, considered as a function $V_f(\theta, \alpha)$, is implemented for every different fiber orientation α_i , as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.

orientation 1: $\theta = 45^\circ + 90^\circ = 135^\circ$

$V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, a1)|)$, with:
 total thickness per orientation, $t_{f1} = NL * t / \text{NoDir} = 1.016$
 $d_{fv} = d$ (figure 11.2, ACI 440) = 370.00
 f_{fe} ((11-5), ACI 440) = 259.312
 $E_f = 64828.00$
 $f_e = 0.004$, from (11.6a), ACI 440
 with $f_u = 0.01$
 From (11-11), ACI 440: $V_s + V_f \leq 238930.50$
 $b_w * d = *d * d / 4 = 80424.772$

Calculation of Shear Strength at edge 2, $V_{r2} = 298161.965$
 $V_{r2} = V_{Col}$ ((10.3), ASCE 41-17) = $k_{nl} * V_{Col0}$
 $V_{Col0} = 298161.965$
 $k_{nl} = 1$ (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ '
 where V_f is the contribution of FRPs (11.3), ACI 440).

$= 1$ (normal-weight concrete)
 $f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)
 $M / Vd = 2.00$
 $\mu_u = 1.2899604E-011$
 $\nu_u = 1.2107607E-031$
 $d = 0.8 * D = 320.00$
 $N_u = 4771.233$
 $A_g = 125663.706$
 From (11.5.4.8), ACI 318-14: $V_s = 0.00$
 $A_v = /2 * A_{stirrup} = 123370.055$
 $f_y = 444.44$
 $s = 360.00$

V_s is multiplied by $Col = 0.00$
 $s/d = 1.125$
 V_f ((11-3)-(11.4), ACI 440) = 194961.134
 $f = 0.95$, for fully-wrapped sections
 $w_f / s_f = 1$ (FRP strips adjacent to one another).

In (11.3) $\sin + \cos$ is replaced with $(\cot + \cot a) \sin a$ which is more a generalised expression,
 where a is the angle of the crack direction (see KANEPE).
 This later relation, considered as a function $V_f(,)$, is implemented for every different fiber orientation a_i ,
 as well as for 2 crack directions, $a = 45^\circ$ and $a = -45^\circ$ to take into consideration the cyclic seismic loading.

orientation 1: $\theta = b1 + 90^\circ = 90.00$
 $V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, a1)|)$, with:
 total thickness per orientation, $t_{f1} = NL * t / \text{NoDir} = 1.016$
 $d_{fv} = d$ (figure 11.2, ACI 440) = 370.00
 f_{fe} ((11-5), ACI 440) = 259.312
 $E_f = 64828.00$
 $f_e = 0.004$, from (11.6a), ACI 440
 with $f_u = 0.01$
 From (11-11), ACI 440: $V_s + V_f \leq 238930.50$
 $b_w * d = *d * d / 4 = 80424.772$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1
 At local axis: 3

Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1
 At Shear local axis: 2
 (Bending local axis: 3)
 Section Type: rccs

Constant Properties

Knowledge Factor, $= 1.00$
 Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength, $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength, $f_s = f_{sm} = 444.44$

Concrete Elasticity, $E_c = 21019.039$

Steel Elasticity, $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength, $f_s = 1.25 \cdot f_{sm} = 555.55$

#####

Diameter, $D = 400.00$

Cover Thickness, $c = 25.00$

Mean Confinement Factor overall section = 1.84055

Element Length, $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length $l_o = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness, $t = 1.016$

Tensile Strength, $f_{fu} = 1055.00$

Tensile Modulus, $E_f = 64828.00$

Elongation, $e_{fu} = 0.01$

Number of directions, $N_{oDir} = 1$

Fiber orientations, $b_i = 0.00^\circ$

Number of layers, $N_L = 1$

Radius of rounding corners, $R = 40.00$

Stepwise Properties

At local axis: 2

EDGE -A-

Shear Force, $V_a = -7.4135260E-048$

EDGE -B-

Shear Force, $V_b = 7.4135260E-048$

BOTH EDGES

Axial Force, $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension: $A_{sl,t} = 0.00$

-Compression: $A_{sl,c} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension: $A_{sl,ten} = 1017.876$

-Compression: $A_{sl,com} = 1017.876$

-Middle: $A_{sl,mid} = 1017.876$

Calculation of Shear Capacity ratio, $V_e/V_r = 0.30252729$

Member Controlled by Flexure ($V_e/V_r < 1$)

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14 $V_e = (M_{pr1} + M_{pr2})/l_n = 90202.132$

with

$M_{pr1} = \text{Max}(Mu_{1+}, Mu_{1-}) = 1.3530E+008$

$Mu_{1+} = 1.3530E+008$, is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$Mu_{1-} = 1.3530E+008$, is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(Mu_{2+}, Mu_{2-}) = 1.3530E+008$

$Mu_{2+} = 1.3530E+008$, is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination

$Mu_{2-} = 1.3530E+008$, is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

Calculation of Mu1+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu
Mu = 1.3530E+008

= 0.90757121
' = 0.80580716
error of function (3.68), Biskinis Phd = 28928.286
From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$
conf. factor $c = 1.84055$
 $f_c = 20.00$
From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 288.6089$
 $l_b/d = 0.26792599$
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00155946$
 $N = 4771.233$
 $A_c = 125663.706$
= $\cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/d

Lap Length: $l_b/d = 0.26792599$
 $l_b = 300.00$
 $l_d = 1119.712$
Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
= 1
 $d_b = 18.00$
Mean strength value of all re-bars: $f_y = 555.55$
 $f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3 \text{ MPa}$ (22.5.3.1, ACI 318-14)
 $t = 1.00$
 $s = 0.80$
 $e = 1.00$
 $cb = 25.00$
 $K_{tr} = 1.14232$
 $A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$
 $s = 360.00$
 $n = 12.00$

Calculation of Mu1-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu
Mu = 1.3530E+008

= 0.90757121
' = 0.80580716
error of function (3.68), Biskinis Phd = 28928.286
From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$
conf. factor $c = 1.84055$
 $f_c = 20.00$
From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 288.6089$
 $l_b/d = 0.26792599$
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00155946$
 $N = 4771.233$
 $A_c = 125663.706$
= $\cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of l_b, \min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$d_b = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f'_c = 20.00$, but $f_c^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} * \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of μ_{2+}

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ

$\mu = 1.3530E+008$

$= 0.90757121$

$' = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY: $f_{cc} = f'_c * c = 36.81095$

conf. factor $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of f_y : $f_y * \text{Min}(1, 1.25 * (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= * \text{Min}(1, 1.25 * (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of l_b, \min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$d_b = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f'_c = 20.00$, but $f_c^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} * \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of μ_2 -

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ

$\mu = 1.3530E+008$

$\mu = 0.90757121$

$\mu = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$

conf. factor $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$d_b = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of Shear Strength $V_r = \text{Min}(V_{r1}, V_{r2}) = 298161.965$

Calculation of Shear Strength at edge 1, $V_{r1} = 298161.965$

$V_{r1} = V_{Co1}$ ((10.3), ASCE 41-17) = $k_{nl} \cdot V_{Co10}$

$V_{Co10} = 298161.965$

$k_{nl} = 1$ (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where V_f is the contribution of FRPs (11.3), ACI 440).

$= 1$ (normal-weight concrete)

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 5.9321525E-012$

$V_u = 7.4135260E-048$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14: $V_s = 0.00$

$A_v = \frac{1}{2} \cdot A_{\text{stirrup}} = 123370.055$

$f_y = 444.44$

$s = 360.00$

Vs is multiplied by Col = 0.00

$$s/d = 1.125$$

$$V_f \text{ ((11-3)-(11.4), ACI 440) } = 194961.134$$

f = 0.95, for fully-wrapped sections

wf/sf = 1 (FRP strips adjacent to one another).

In (11.3) $\sin \theta + \cos \theta$ is replaced with $(\cot \theta + \csc \theta) \sin \alpha$ which is more a generalised expression, where θ is the angle of the crack direction (see KANEPE).

This later relation, considered as a function $V_f(\theta, \alpha)$, is implemented for every different fiber orientation α_i , as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.

$$\text{orientation 1: } \theta = b1 + 90^\circ = 90.00$$

$$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \theta)|), \text{ with:}$$

$$\text{total thickness per orientation, } t_{f1} = NL * t / \text{NoDir} = 1.016$$

$$d_{fv} = d \text{ (figure 11.2, ACI 440) } = 370.00$$

$$f_{fe} \text{ ((11-5), ACI 440) } = 259.312$$

$$E_f = 64828.00$$

$$f_e = 0.004, \text{ from (11.6a), ACI 440}$$

$$\text{with } f_u = 0.01$$

$$\text{From (11-11), ACI 440: } V_s + V_f \leq 238930.50$$

$$b_w * d = \rho * d^2 / 4 = 80424.772$$

Calculation of Shear Strength at edge 2, $V_{r2} = 298161.965$

$$V_{r2} = V_{Col} \text{ ((10.3), ASCE 41-17) } = k_{nl} * V_{Col0}$$

$$V_{Col0} = 298161.965$$

$k_{nl} = 1$ (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ ' where V_f is the contribution of FRPs (11.3), ACI 440).

$\rho = 1$ (normal-weight concrete)

$$f_c' = 20.00, \text{ but } f_c'^{0.5} \leq 8.3 \text{ MPa (22.5.3.1, ACI 318-14)}$$

$$M / Vd = 2.00$$

$$\mu_u = 5.9321525E-012$$

$$V_u = 7.4135260E-048$$

$$d = 0.8 * D = 320.00$$

$$N_u = 4771.233$$

$$A_g = 125663.706$$

From (11.5.4.8), ACI 318-14: $V_s = 0.00$

$$A_v = \rho / 2 * A_{stirrup} = 123370.055$$

$$f_y = 444.44$$

$$s = 360.00$$

Vs is multiplied by Col = 0.00

$$s/d = 1.125$$

$$V_f \text{ ((11-3)-(11.4), ACI 440) } = 194961.134$$

f = 0.95, for fully-wrapped sections

wf/sf = 1 (FRP strips adjacent to one another).

In (11.3) $\sin \theta + \cos \theta$ is replaced with $(\cot \theta + \csc \theta) \sin \alpha$ which is more a generalised expression, where θ is the angle of the crack direction (see KANEPE).

This later relation, considered as a function $V_f(\theta, \alpha)$, is implemented for every different fiber orientation α_i , as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.

$$\text{orientation 1: } \theta = b1 + 90^\circ = 90.00$$

$$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \theta)|), \text{ with:}$$

$$\text{total thickness per orientation, } t_{f1} = NL * t / \text{NoDir} = 1.016$$

$$d_{fv} = d \text{ (figure 11.2, ACI 440) } = 370.00$$

$$f_{fe} \text{ ((11-5), ACI 440) } = 259.312$$

$$E_f = 64828.00$$

$$f_e = 0.004, \text{ from (11.6a), ACI 440}$$

$$\text{with } f_u = 0.01$$

$$\text{From (11-11), ACI 440: } V_s + V_f \leq 238930.50$$

$$b_w * d = \rho * d^2 / 4 = 80424.772$$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At local axis: 2

Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (b)

Section Type: rccs

Constant Properties

Knowledge Factor, $k = 1.00$

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE 41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength, $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength, $f_s = f_{sm} = 444.44$

Concrete Elasticity, $E_c = 21019.039$

Steel Elasticity, $E_s = 200000.00$

Diameter, $D = 400.00$

Cover Thickness, $c = 25.00$

Element Length, $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length $l_b = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness, $t = 1.016$

Tensile Strength, $f_{fu} = 1055.00$

Tensile Modulus, $E_f = 64828.00$

Elongation, $e_{fu} = 0.01$

Number of directions, $N_{oDir} = 1$

Fiber orientations, $b_i = 0.00^\circ$

Number of layers, $N_L = 1$

Radius of rounding corners, $R = 40.00$

Stepwise Properties

Bending Moment, $M = 0.09265189$

Shear Force, $V_2 = 3472.528$

Shear Force, $V_3 = 1.2919419E-013$

Axial Force, $F = -4769.844$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension: $A_{st} = 0.00$

-Compression: $A_{sc} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension: $A_{st,ten} = 1017.876$

-Compression: $A_{st,com} = 1017.876$

-Middle: $A_{st,mid} = 1017.876$

Mean Diameter of Tension Reinforcement, $D_bL = 18.00$

Existing component: From table 7-7, ASCE 41_17: Final chord rotation Capacity $u_{,R} = \gamma + p = 0.00186344$

$u = \gamma + p = 0.00186344$

- Calculation of γ -

$\gamma = (M \gamma L_s / 3) / E_{eff} = 0.00186344$ ((4.29), Biskinis Phd))

$M \gamma = 1.4766E+008$

$L_s = M / V$ (with $L_s > 0.1 * L$ and $L_s < 2 * L$) = 300.00

From table 10.5, ASCE 41_17: $E_{eff} = factor * E_c * I_g = 7.9240E+012$

factor = 0.30

Ag = 125663.706
fc' = 20.00
N = 4769.844
Ec*Ig = 2.6413E+013

Calculation of Yielding Moment My

Calculation of ϕ_y and My according to (7) - (8) in Biskinis and Fardis

My = Min(My_ten, My_com) = 1.4766E+008
y = 8.1688007E-006
My_ten (8c) = 1.4766E+008
 _ten (7c) = 72.40642
error of function (7c) = 0.00117228
My_com (8d) = 3.7493E+008
 _com (7d) = 69.91126
error of function (7d) = 0.00301342
with ((10.1), ASCE 41-17) $\phi_y = \text{Min}(\phi_y, 1.25 * \phi_y * (l_b/l_d)^{2/3}) = 0.0022222$
 eco = 0.002
 apl = 0.45 ((9c) in Biskinis and Fardis for FRP Wrap)
 d1 = 44.00
 R = 200.00
 v = 0.00157305
 N = 4769.844
 Ac = 125663.706
 ((10.1), ASCE 41-17) $\phi_y = \text{Min}(\phi_y, 1.25 * \phi_y * (l_b/l_d)^{2/3}) = 0.44757577$
with fc' ((12.3), ACI 440) = 24.12975
 fc = 20.00
 fl = 1.3173
 k = 1
 Effective FRP thickness, tf = NL*t*cos(b1) = 1.016
 efe ((12.5) and (12.7)) = 0.004
 fu = 0.01
 Ef = 64828.00

Calculation of ratio lb/l_d

Lap Length: $l_d/l_{d,min} = 0.33490748$
lb = 300.00
ld = 895.7698
Calculation of l according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
 = 1
 db = 18.00
Mean strength value of all re-bars: fy = 444.44
fc' = 20.00, but $fc'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)
 t = 1.00
 s = 0.80
 e = 1.00
cb = 25.00
Ktr = 1.14232
Atr = $\sqrt{2} * \text{Area of stirrup} = 123.3701$
 s = 360.00
 n = 12.00

- Calculation of ϕ_p -

From table 10-9: $\phi_p = 0.00$

with:

- Columns controlled by inadequate development or splicing along the clear height because $l_b/l_d < 1$
shear control ratio $V_y/E/V_{CoI} = 0.30252729$
d = 0.00
s = 0.00

$$t = 2 \cdot A_v / (d_c \cdot s) + 4 \cdot t_f / D \cdot (f_{fe} / f_s) = 0.00721126$$

$A_v = 78.53982$, is the area of the circular stirrup

$d_c = D - 2 \cdot \text{cover} - \text{Hoop Diameter} = 340.00$

The term $2 \cdot t_f / b_w \cdot (f_{fe} / f_s)$ is implemented to account for FRP contribution

where $f = 2 \cdot t_f / b_w$ is FRP ratio (EC8 - 3, A.4.4.3(6)) and f_{fe} / f_s normalises f to steel strength

All these variables have already been given in Shear control ratio calculation.

$$N_{UD} = 4769.844$$

$$A_g = 125663.706$$

$$f_{cE} = 20.00$$

$$f_{tE} = f_{yE} = 444.44$$

$$p_l = \text{Area_Tot_Long_Rein} / (A_g) = 0.0243$$

$$f_{cE} = 20.00$$

End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (b)

Calculation No. 9

column C1, Floor 1

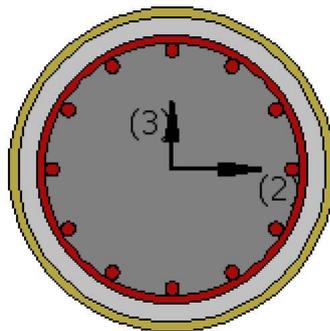
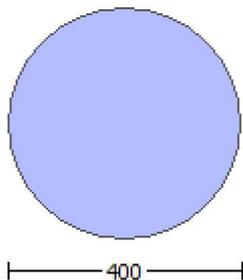
Limit State: Collapse Prevention (data interpolation between analysis steps 2 and 3)

Analysis: Uniform +X

Check: Shear capacity V_{Rd}

Edge: Start

Local Axis: (2)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (a)

Section Type: rccs

Constant Properties

Knowledge Factor, $\gamma = 1.00$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE 41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength, $f_c = f_{c_lower_bound} = 16.00$

Existing material of Primary Member: Steel Strength, $f_s = f_{s_lower_bound} = 400.00$

Concrete Elasticity, $E_c = 21019.039$

Steel Elasticity, $E_s = 200000.00$

#####

Note: Especially for the calculation of γ for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE 41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE 41-17).

Existing material: Concrete Strength, $f_c = f_{cm} = 20.00$

Existing material: Steel Strength, $f_s = f_{sm} = 444.44$

#####

Diameter, $D = 400.00$

Cover Thickness, $c = 25.00$

Element Length, $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length $l_o = l_b = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness, $t = 1.016$

Tensile Strength, $f_{fu} = 1055.00$

Tensile Modulus, $E_f = 64828.00$

Elongation, $e_{fu} = 0.01$

Number of directions, $N_{oDir} = 1$

Fiber orientations, $b_i = 0.00^\circ$

Number of layers, $N_L = 1$

Radius of rounding corners, $R = 40.00$

Stepwise Properties

EDGE -A-

Bending Moment, $M_a = -1.3471E+007$

Shear Force, $V_a = -4486.792$

EDGE -B-

Bending Moment, $M_b = 2684.487$

Shear Force, $V_b = 4486.792$

BOTH EDGES

Axial Force, $F = -4783.229$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension: $A_{sl} = 1272.345$

-Compression: $A_{sc} = 1781.283$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension: $A_{sl,ten} = 1017.876$

-Compression: $A_{sl,com} = 1017.876$

-Middle: $A_{sl,mid} = 1017.876$

Mean Diameter of Tension Reinforcement, $D_{bL,ten} = 18.00$

Existing component: From table 7-7, ASCE 41_17: Final Shear Capacity $V_R = \phi V_n = 235776.467$

V_n ((10.3), ASCE 41-17) = $k_n \phi V_{CoI} = 235776.467$

$V_{CoI} = 235776.467$

$k_n = 1.00$

$displacement_ductility_demand = 0.02901189$

NOTE: In expression (10-3) ' $V_s = A_v \phi f_y d/s$ ' is replaced by ' $V_{s+} = \phi V_f$ ' where V_f is the contribution of FRPs (11.3), ACI 440).

 $\phi = 1$ (normal-weight concrete)

$f_c' = 16.00$, but $f_c^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$M/Vd = 4.00$
 $M_u = 1.3471E+007$
 $V_u = 4486.792$
 $d = 0.8 \cdot D = 320.00$
 $N_u = 4783.229$
 $A_g = 125663.706$
 From (11.5.4.8), ACI 318-14: $V_s = 0.00$
 $A_v = \sqrt{2} \cdot A_{stirrup} = 123370.055$
 $f_y = 400.00$
 $s = 360.00$
 V_s is multiplied by $\text{Col} = 0.00$
 $s/d = 1.125$
 V_f ((11-3)-(11.4), ACI 440) = 194961.134
 $f = 0.95$, for fully-wrapped sections
 $w_f/s_f = 1$ (FRP strips adjacent to one another).
 In (11.3) $\sin + \cos$ is replaced with $(\cot + \cot a) \sin a$ which is more a generalised expression,
 where a is the angle of the crack direction (see KANEPE).
 This later relation, considered as a function $V_f(\theta, a)$, is implemented for every different fiber orientation a_i ,
 as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.
 orientation 1: $\theta = b_1 + 90^\circ = 90.00$
 $V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, a_1)|)$, with:
 total thickness per orientation, $t_{f1} = N_L \cdot t / \text{NoDir} = 1.016$
 $d_{fv} = d$ (figure 11.2, ACI 440) = 370.00
 f_{fe} ((11-5), ACI 440) = 259.312
 $E_f = 64828.00$
 $f_e = 0.004$, from (11.6a), ACI 440
 with $f_u = 0.01$
 From (11-11), ACI 440: $V_s + V_f \leq 213705.936$
 $b_w \cdot d = \sqrt{V_s \cdot V_f} = 80424.772$

displacement_ductility_demand is calculated as δ / y

- Calculation of δ / y for END A -
for rotation axis 3 and integ. section (a)

From analysis, chord rotation $\theta = 0.00054105$
 $y = (M_y \cdot L_s / 3) / E_{eff} = 0.01864917$ ((4.29), Biskinis Phd)
 $M_y = 1.4766E+008$
 $L_s = M/V$ (with $L_s > 0.1 \cdot L$ and $L_s < 2 \cdot L$) = 3002.334
 From table 10.5, ASCE 41_17: $E_{eff} = \text{factor} \cdot E_c \cdot I_g = 7.9240E+012$
 $\text{factor} = 0.30$
 $A_g = 125663.706$
 $f_c' = 20.00$
 $N = 4783.229$
 $E_c \cdot I_g = 2.6413E+013$

Calculation of Yielding Moment M_y

Calculation of δ / y and M_y according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y_ten}, M_{y_com}) = 1.4766E+008$
 $y = 8.1688588E-006$
 M_{y_ten} (8c) = 1.4766E+008
 δ_{ten} (7c) = 72.4069
 error of function (7c) = 0.00117279
 M_{y_com} (8d) = 3.7493E+008
 δ_{com} (7d) = 69.91139
 error of function (7d) = 0.00300901
 with ((10.1), ASCE 41-17) $e_y = \text{Min}(e_y, 1.25 \cdot e_y \cdot (I_b / I_d)^{2/3}) = 0.0022222$
 $e_{co} = 0.002$
 $a_{pl} = 0.45$ ((9c) in Biskinis and Fardis for FRP Wrap)
 $d_1 = 44.00$
 $R = 200.00$

$v = 0.00157746$
 $N = 4783.229$
 $A_c = 125663.706$
 $((10.1), ASCE 41-17) = \text{Min}(, 1.25 * (lb/ld)^{2/3}) = 0.44757577$
with f_c^* ((12.3), ACI 440) = 24.12975
 $f_c = 20.00$
 $f_l = 1.3173$
 $k = 1$
Effective FRP thickness, $t_f = NL * t * \text{Cos}(b1) = 1.016$
 e_{fe} ((12.5) and (12.7)) = 0.004
 $f_u = 0.01$
 $E_f = 64828.00$

Calculation of ratio lb/ld

Lap Length: $ld/ld, \text{min} = 0.33490748$

$lb = 300.00$

$ld = 895.7698$

Calculation of I according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$db = 18.00$

Mean strength value of all re-bars: $f_y = 444.44$

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3 \text{ MPa}$ (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$cb = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = /2 * \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

End Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (a)

Calculation No. 10

column C1, Floor 1

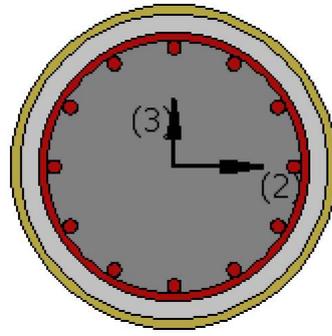
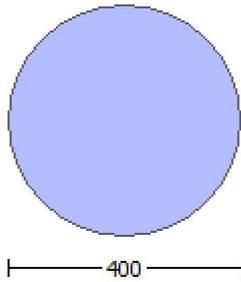
Limit State: Collapse Prevention (data interpolation between analysis steps 2 and 3)

Analysis: Uniform +X

Check: Chord rotation capacity (θ)

Edge: Start

Local Axis: (2)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 3
 (Bending local axis: 2)
 Section Type: rccs

Constant Properties

Knowledge Factor, $\gamma = 1.00$
 Mean strength values are used for both shear and moment calculations.
 Consequently:
 Existing material of Primary Member: Concrete Strength, $f_c = f_{cm} = 20.00$
 Existing material of Primary Member: Steel Strength, $f_s = f_{sm} = 444.44$
 Concrete Elasticity, $E_c = 21019.039$
 Steel Elasticity, $E_s = 200000.00$

 Note: Especially for the calculation of moment strengths,
 the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14
 Existing material: Steel Strength, $f_s = 1.25 \cdot f_{sm} = 555.55$

 Diameter, $D = 400.00$
 Cover Thickness, $c = 25.00$
 Mean Confinement Factor overall section = 1.84055
 Element Length, $L = 3000.00$
 Primary Member
 Ribbed Bars
 Ductile Steel
 Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)
 Longitudinal Bars With Ends Lapped Starting at the End Sections
 Lap Length $l_o = 300.00$
 FRP Wrapping Data
 Type: Carbon
 Cured laminate properties (design values)
 Thickness, $t = 1.016$
 Tensile Strength, $f_{fu} = 1055.00$
 Tensile Modulus, $E_f = 64828.00$
 Elongation, $e_{fu} = 0.01$
 Number of directions, $NoDir = 1$
 Fiber orientations, $b_i: 0.00^\circ$
 Number of layers, $NL = 1$
 Radius of rounding corners, $R = 40.00$

Stepwise Properties

At local axis: 3
 EDGE -A-
 Shear Force, $V_a = 1.2107607E-031$
 EDGE -B-
 Shear Force, $V_b = -1.2107607E-031$
 BOTH EDGES
 Axial Force, $F = -4771.233$
 Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension: Aslt = 0.00

-Compression: Aslc = 3053.628

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension: Asl,ten = 1017.876

-Compression: Asl,com = 1017.876

-Middle: Asl,mid = 1017.876

Calculation of Shear Capacity ratio , $V_e/V_r = 0.30252729$

Member Controlled by Flexure ($V_e/V_r < 1$)

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14 $V_e = (M_{pr1} + M_{pr2})/l_n = 90202.132$

with

$M_{pr1} = \text{Max}(M_{u1+} , M_{u1-}) = 1.3530E+008$

$M_{u1+} = 1.3530E+008$, is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$M_{u1-} = 1.3530E+008$, is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(M_{u2+} , M_{u2-}) = 1.3530E+008$

$M_{u2+} = 1.3530E+008$, is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

$M_{u2-} = 1.3530E+008$, is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

Calculation of M_{u1+}

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), M_u

$M_u = 1.3530E+008$

$\phi = 0.90757121$

$\phi' = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$

conf. factor $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$\phi = \phi' \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of l_b, min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$\beta = 1$

$d_b = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3 \text{ MPa}$ (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of Mu1-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu
Mu = 1.3530E+008

= 0.90757121
' = 0.80580716
error of function (3.68), Biskinis Phd = 28928.286
From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$
conf. factor $c = 1.84055$
 $f_c = 20.00$
From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$
 $l_b/l_d = 0.26792599$
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00155946$
 $N = 4771.233$
 $Ac = 125663.706$
= $\cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$
 $l_b = 300.00$
 $l_d = 1119.712$
Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
= 1
 $db = 18.00$
Mean strength value of all re-bars: $f_y = 555.55$
 $f_c' = 20.00$, but $f_c^{0.5} \leq 8.3 \text{ MPa}$ (22.5.3.1, ACI 318-14)
 $t = 1.00$
 $s = 0.80$
 $e = 1.00$
 $cb = 25.00$
 $K_{tr} = 1.14232$
 $A_{tr} = \sqrt{2} \cdot \text{Area of stirrup} = 123.3701$
 $s = 360.00$
 $n = 12.00$

Calculation of Mu2+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu
Mu = 1.3530E+008

= 0.90757121
' = 0.80580716
error of function (3.68), Biskinis Phd = 28928.286
From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$
conf. factor $c = 1.84055$
 $f_c = 20.00$
From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$
 $l_b/l_d = 0.26792599$
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00155946$
 $N = 4771.233$
 $Ac = 125663.706$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.28805051$$

Calculation of ratio l_b/d

Lap Length: $l_b/d = 0.26792599$

$l_b = 300.00$

$d = 1119.712$

Calculation of l_b, min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$\beta = 1$

$d_b = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f'_c = 20.00$, but $f_c^{0.5} \leq 8.3 \text{ MPa}$ (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \pi/4 \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of μ_2

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ

$\mu = 1.3530 \times 10^8$

$\beta = 0.90757121$

$\beta' = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY: $f_{cc} = f'_c \cdot c = 36.81095$

conf. factor $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 288.6089$

$l_b/d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$\beta = \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/d

Lap Length: $l_b/d = 0.26792599$

$l_b = 300.00$

$d = 1119.712$

Calculation of l_b, min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$\beta = 1$

$d_b = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f'_c = 20.00$, but $f_c^{0.5} \leq 8.3 \text{ MPa}$ (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \pi/4 \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of Shear Strength $V_r = \text{Min}(V_{r1}, V_{r2}) = 298161.965$

Calculation of Shear Strength at edge 1, $V_{r1} = 298161.965$

$V_{r1} = V_{\text{Col}} ((10.3), \text{ASCE 41-17}) = k_{nl} * V_{\text{Col}0}$

$V_{\text{Col}0} = 298161.965$

$k_{nl} = 1$ (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ '
where V_f is the contribution of FRPs (11.3), ACI 440).

$k = 1$ (normal-weight concrete)

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.2899604E-011$

$\nu_u = 1.2107607E-031$

$d = 0.8 * D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14: $V_s = 0.00$

$A_v = \frac{1}{2} * A_{\text{stirrup}} = 123370.055$

$f_y = 444.44$

$s = 360.00$

V_s is multiplied by $\text{Col} = 0.00$

$s/d = 1.125$

$V_f ((11-3)-(11.4), \text{ACI 440}) = 194961.134$

$f = 0.95$, for fully-wrapped sections

$w_f/s_f = 1$ (FRP strips adjacent to one another).

In (11.3) $\sin \theta + \cos \theta$ is replaced with $(\cot \theta + \csc \theta) \sin \alpha$ which is more a generalised expression,
where θ is the angle of the crack direction (see KANEPE).

This later relation, considered as a function $V_f(\theta, \alpha)$, is implemented for every different fiber orientation α_i ,
as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.

orientation 1: $\theta = b_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \alpha)|)$, with:

total thickness per orientation, $t_{f1} = N_L * t / N_{\text{Dir}} = 1.016$

$d_{fv} = d$ (figure 11.2, ACI 440) = 370.00

$f_{fe} ((11-5), \text{ACI 440}) = 259.312$

$E_f = 64828.00$

$f_e = 0.004$, from (11.6a), ACI 440

with $f_u = 0.01$

From (11-11), ACI 440: $V_s + V_f \leq 238930.50$

$b_w * d = \frac{1}{4} * d * d = 80424.772$

Calculation of Shear Strength at edge 2, $V_{r2} = 298161.965$

$V_{r2} = V_{\text{Col}} ((10.3), \text{ASCE 41-17}) = k_{nl} * V_{\text{Col}0}$

$V_{\text{Col}0} = 298161.965$

$k_{nl} = 1$ (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ '
where V_f is the contribution of FRPs (11.3), ACI 440).

$k = 1$ (normal-weight concrete)

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.2899604E-011$

$\nu_u = 1.2107607E-031$

$d = 0.8 * D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14: $V_s = 0.00$

$A_v = \frac{1}{2} * A_{\text{stirrup}} = 123370.055$

$f_y = 444.44$

$s = 360.00$

V_s is multiplied by $\text{Col} = 0.00$

$s/d = 1.125$

Vf ((11-3)-(11.4), ACI 440) = 194961.134

f = 0.95, for fully-wrapped sections

wf/sf = 1 (FRP strips adjacent to one another).

In (11.3) $\sin \theta + \cos \theta$ is replaced with $(\cot \theta + \cot \alpha) \sin \alpha$ which is more a generalised expression, where θ is the angle of the crack direction (see KANEPE).

This later relation, considered as a function Vf(θ), is implemented for every different fiber orientation ai, as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.

orientation 1: $\theta_1 = b_1 + 90^\circ = 90.00$

Vf = Min(|Vf(45, 1)|, |Vf(-45, a1)|), with:

total thickness per orientation, tf1 = NL*t/NoDir = 1.016

dfv = d (figure 11.2, ACI 440) = 370.00

ffe ((11-5), ACI 440) = 259.312

Ef = 64828.00

fe = 0.004, from (11.6a), ACI 440

with fu = 0.01

From (11-11), ACI 440: Vs + Vf <= 238930.50

bw*d = $\frac{V_s + V_f \cdot d}{4} = 80424.772$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1
At local axis: 3

Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1
At Shear local axis: 2
(Bending local axis: 3)
Section Type: rccs

Constant Properties

Knowledge Factor, $\gamma = 1.00$

Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength, $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength, $f_s = f_{sm} = 444.44$

Concrete Elasticity, $E_c = 21019.039$

Steel Elasticity, $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength, $f_s = 1.25 \cdot f_{sm} = 555.55$

#####

Diameter, D = 400.00

Cover Thickness, c = 25.00

Mean Confinement Factor overall section = 1.84055

Element Length, L = 3000.00

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length lo = 300.00

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness, t = 1.016

Tensile Strength, $f_{fu} = 1055.00$

Tensile Modulus, Ef = 64828.00

Elongation, $e_{fu} = 0.01$

Number of directions, NoDir = 1

Fiber orientations, bi: 0.00°

Number of layers, NL = 1

Radius of rounding corners, R = 40.00

Stepwise Properties

At local axis: 2

EDGE -A-

Shear Force, $V_a = -7.4135260E-048$

EDGE -B-

Shear Force, $V_b = 7.4135260E-048$

BOTH EDGES

Axial Force, $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension: $A_{sl,t} = 0.00$

-Compression: $A_{sl,c} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension: $A_{sl,ten} = 1017.876$

-Compression: $A_{sl,com} = 1017.876$

-Middle: $A_{sl,mid} = 1017.876$

Calculation of Shear Capacity ratio, $V_e/V_r = 0.30252729$

Member Controlled by Flexure ($V_e/V_r < 1$)

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14 $V_e = (M_{pr1} + M_{pr2})/l_n = 90202.132$

with

$M_{pr1} = \text{Max}(M_{u1+}, M_{u1-}) = 1.3530E+008$

$M_{u1+} = 1.3530E+008$, is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$M_{u1-} = 1.3530E+008$, is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(M_{u2+}, M_{u2-}) = 1.3530E+008$

$M_{u2+} = 1.3530E+008$, is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination

$M_{u2-} = 1.3530E+008$, is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

Calculation of M_{u1+}

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), M_u

$M_u = 1.3530E+008$

$\phi = 0.90757121$

$\phi' = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$

conf. factor $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$\phi' = \phi \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of l_b, min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$\phi = 1$

$d_b = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$
 $f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)
 $t = 1.00$
 $s = 0.80$
 $e = 1.00$
 $cb = 25.00$
 $K_{tr} = 1.14232$
 $A_{tr} = \frac{1}{2} * \text{Area of stirrup} = 123.3701$
 $s = 360.00$
 $n = 12.00$

Calculation of μ_{1-}

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ
 $\mu = 1.3530E+008$

 $\phi = 0.90757121$
 $\lambda = 0.80580716$
error of function (3.68), Biskinis Phd = 28928.286
From 5A.2, TBDY: $f_{cc} = f_c' * c = 36.81095$
conf. factor $c = 1.84055$
 $f_c = 20.00$
From 10.3.5, ASCE 41-17, Final value of f_y : $f_y * \text{Min}(1, 1.25 * (l_b / d)^{2/3}) = 288.6089$
 $l_b / d = 0.26792599$
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00155946$
 $N = 4771.233$
 $A_c = 125663.706$
 $\phi * \text{Min}(1, 1.25 * (l_b / d)^{2/3}) = 0.28805051$

Calculation of ratio l_b / d

Lap Length: $l_b / d = 0.26792599$
 $l_b = 300.00$
 $d = 1119.712$
Calculation of l_b, min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
 $\phi = 1$

$db = 18.00$
Mean strength value of all re-bars: $f_y = 555.55$
 $f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)
 $t = 1.00$
 $s = 0.80$
 $e = 1.00$
 $cb = 25.00$
 $K_{tr} = 1.14232$
 $A_{tr} = \frac{1}{2} * \text{Area of stirrup} = 123.3701$
 $s = 360.00$
 $n = 12.00$

Calculation of μ_{2+}

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ
 $\mu = 1.3530E+008$

 $\phi = 0.90757121$
 $\lambda = 0.80580716$
error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$

conf. factor $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$db = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$cb = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of μ_2

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ

$\mu = 1.3530E+008$

$= 0.90757121$

$' = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$

conf. factor $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$db = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

t = 1.00
s = 0.80
e = 1.00
cb = 25.00
Ktr = 1.14232
Atr = /2 * Area of stirrup = 123.3701
s = 360.00
n = 12.00

Calculation of Shear Strength $V_r = \text{Min}(V_{r1}, V_{r2}) = 298161.965$

Calculation of Shear Strength at edge 1, $V_{r1} = 298161.965$

$V_{r1} = V_{Col}$ ((10.3), ASCE 41-17) = knl*VCol0

VCol0 = 298161.965

knl = 1 (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ '
where V_f is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)
 $f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

M/Vd = 2.00

$\mu_u = 5.9321525E-012$

$\nu_u = 7.4135260E-048$

d = 0.8*D = 320.00

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14: $V_s = 0.00$

$A_v = /2 * A_{stirrup} = 123370.055$

$f_y = 444.44$

s = 360.00

V_s is multiplied by Col = 0.00

s/d = 1.125

V_f ((11-3)-(11.4), ACI 440) = 194961.134

f = 0.95, for fully-wrapped sections

wf/sf = 1 (FRP strips adjacent to one another).

In (11.3) $\sin + \cos$ is replaced with $(\cot + \cot_a) \sin a$ which is more a generalised expression,
where a is the angle of the crack direction (see KANEPE).

This later relation, considered as a function $V_f(\theta, a)$, is implemented for every different fiber orientation a_i ,
as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.

orientation 1: $\theta = b1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, a1)|)$, with:

total thickness per orientation, $t_{f1} = NL * t / N_{oDir} = 1.016$

$d_{fv} = d$ (figure 11.2, ACI 440) = 370.00

f_{fe} ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_{e} = 0.004$, from (11.6a), ACI 440

with $f_u = 0.01$

From (11-11), ACI 440: $V_s + V_f \leq 238930.50$

$b_w * d = *d * d / 4 = 80424.772$

Calculation of Shear Strength at edge 2, $V_{r2} = 298161.965$

$V_{r2} = V_{Col}$ ((10.3), ASCE 41-17) = knl*VCol0

VCol0 = 298161.965

knl = 1 (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ '
where V_f is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

M/Vd = 2.00

$\mu_u = 5.9321525E-012$

$V_u = 7.4135260E-048$
 $d = 0.8 \cdot D = 320.00$
 $Nu = 4771.233$
 $Ag = 125663.706$
 From (11.5.4.8), ACI 318-14: $V_s = 0.00$
 $A_v = \frac{1}{2} \cdot A_{stirrup} = 123370.055$
 $f_y = 444.44$
 $s = 360.00$
 V_s is multiplied by $Col = 0.00$
 $s/d = 1.125$
 V_f ((11-3)-(11.4), ACI 440) = 194961.134
 $f = 0.95$, for fully-wrapped sections
 $w_f/s_f = 1$ (FRP strips adjacent to one another).
 In (11.3) $\sin \theta + \cos \theta$ is replaced with $(\cot \theta + \cot \alpha) \sin \alpha$ which is more a generalised expression,
 where θ is the angle of the crack direction (see KANEPE).
 This later relation, considered as a function $V_f(\theta, \alpha)$, is implemented for every different cyclic fiber orientation α_i ,
 as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.
 orientation 1: $\theta = b1 + 90^\circ = 90.00$
 $V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \theta)|)$, with:
 total thickness per orientation, $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$
 $d_{fv} = d$ (figure 11.2, ACI 440) = 370.00
 f_{fe} ((11-5), ACI 440) = 259.312
 $E_f = 64828.00$
 $f_e = 0.004$, from (11.6a), ACI 440
 with $f_u = 0.01$
 From (11-11), ACI 440: $V_s + V_f \leq 238930.50$
 $b_w \cdot d = \frac{1}{4} \cdot d \cdot d = 80424.772$

 End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1
 At local axis: 2

 Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1
 At local axis: 2
 Integration Section: (a)
 Section Type: rccs

Constant Properties

 Knowledge Factor, $\gamma = 1.00$
 Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE 41-17.
 Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17
 Consequently:
 Existing material of Primary Member: Concrete Strength, $f_c = f_{cm} = 20.00$
 Existing material of Primary Member: Steel Strength, $f_s = f_{sm} = 444.44$
 Concrete Elasticity, $E_c = 21019.039$
 Steel Elasticity, $E_s = 200000.00$
 Diameter, $D = 400.00$
 Cover Thickness, $c = 25.00$
 Element Length, $L = 3000.00$
 Primary Member
 Ribbed Bars
 Ductile Steel
 Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)
 Longitudinal Bars With Ends Lapped Starting at the End Sections
 Lap Length $l_b = 300.00$
 FRP Wrapping Data
 Type: Carbon
 Cured laminate properties (design values)
 Thickness, $t = 1.016$
 Tensile Strength, $f_{fu} = 1055.00$
 Tensile Modulus, $E_f = 64828.00$
 Elongation, $e_{fu} = 0.01$
 Number of directions, $\text{NoDir} = 1$

Fiber orientations, β_i : 0.00°
Number of layers, NL = 1
Radius of rounding corners, R = 40.00

Stepwise Properties

Bending Moment, M = 6.2300792E-010
Shear Force, V2 = -4486.792
Shear Force, V3 = -1.7064698E-013
Axial Force, F = -4783.229
Longitudinal Reinforcement Area Distribution (in 2 divisions)
-Tension: A_{st} = 1272.345
-Compression: A_{sc} = 1781.283
Longitudinal Reinforcement Area Distribution (in 3 divisions)
-Tension: $A_{st,ten}$ = 1017.876
-Compression: $A_{sc,com}$ = 1017.876
-Middle: $A_{st,mid}$ = 1017.876
Mean Diameter of Tension Reinforcement, D_bL = 18.00

Existing component: From table 7-7, ASCE 41_17: Final chord rotation Capacity $u_R = \phi u = 0.05131733$
 $u = \phi u + p = 0.05131733$

- Calculation of ϕ_y -

$y = (M_y * L_s / 3) / E_{eff} = 0.00931733$ ((4.29), Biskinis Phd)
 $M_y = 1.4766E+008$
 $L_s = M/V$ (with $L_s > 0.1 * L$ and $L_s < 2 * L$) = 1500.00
From table 10.5, ASCE 41_17: $E_{eff} = factor * E_c * I_g = 7.9240E+012$
factor = 0.30
 $A_g = 125663.706$
 $f_c' = 20.00$
N = 4783.229
 $E_c * I_g = 2.6413E+013$

Calculation of Yielding Moment M_y

Calculation of ϕ_y and M_y according to (7) - (8) in Biskinis and Fardis

$M_y = \min(M_{y,ten}, M_{y,com}) = 1.4766E+008$
 $y = 8.1688588E-006$
 $M_{y,ten}$ (8c) = 1.4766E+008
 ϕ_{ten} (7c) = 72.4069
error of function (7c) = 0.00117279
 $M_{y,com}$ (8d) = 3.7493E+008
 ϕ_{com} (7d) = 69.91139
error of function (7d) = 0.00300901
with ((10.1), ASCE 41-17) $e_y = \min(e_y, 1.25 * e_y * (I_b / I_d)^{2/3}) = 0.0022222$
 $e_{co} = 0.002$
 $a_{pl} = 0.45$ ((9c) in Biskinis and Fardis for FRP Wrap)
 $d_1 = 44.00$
R = 200.00
 $v = 0.00157746$
N = 4783.229
 $A_c = 125663.706$
((10.1), ASCE 41-17) $\phi = \min(\phi, 1.25 * \phi * (I_b / I_d)^{2/3}) = 0.44757577$
with f_c^* ((12.3), ACI 440) = 24.12975
 $f_l = 20.00$
 $\beta_1 = 1.3173$
k = 1
Effective FRP thickness, $t_f = NL * t * \cos(\beta_1) = 1.016$

ϵ_{fe} ((12.5) and (12.7)) = 0.004
 $f_u = 0.01$
 $E_f = 64828.00$

Calculation of ratio l_b/l_d

Lap Length: $l_d/l_{d,min} = 0.33490748$

$l_b = 300.00$

$l_d = 895.7698$

Calculation of l according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
 $= 1$

$d_b = 18.00$

Mean strength value of all re-bars: $f_y = 444.44$

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} * \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

- Calculation of p -

From table 10-9: $p = 0.042$

with:

- Columns controlled by inadequate development or splicing along the clear height because $l_b/l_d < 1$

shear control ratio $V_y E / V_{CoI} E = 0.30252729$

$d = 0.00$

$s = 0.00$

$t = 2 * A_v / (d_c * s) + 4 * t_f / D * (f_{fe} / f_s) = 0.00721126$

$A_v = 78.53982$, is the area of the circular stirrup

$d_c = D - 2 * \text{cover} - \text{Hoop Diameter} = 340.00$

The term $2 * t_f / b_w * (f_{fe} / f_s)$ is implemented to account for FRP contribution

where $f = 2 * t_f / b_w$ is FRP ratio (EC8 - 3, A.4.4.3(6)) and f_{fe} / f_s normalises f to steel strength

All these variables have already been given in Shear control ratio calculation.

$N_{UD} = 4783.229$

$A_g = 125663.706$

$f_c E = 20.00$

$f_y E = f_{yI} E = 444.44$

$p_l = \text{Area}_{Tot_Long_Rein} / (A_g) = 0.0243$

$f_c E = 20.00$

End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (a)

Calculation No. 11

column C1, Floor 1

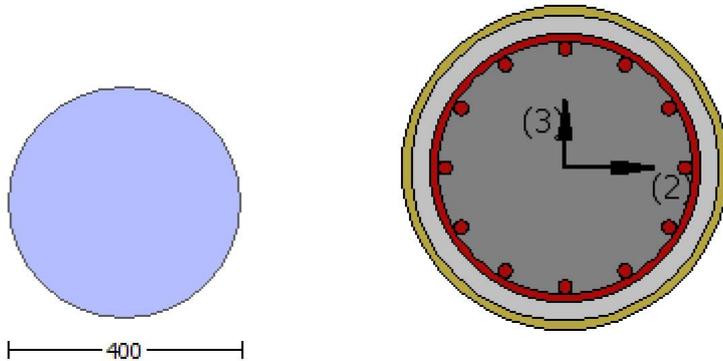
Limit State: Collapse Prevention (data interpolation between analysis steps 2 and 3)

Analysis: Uniform +X

Check: Shear capacity VRd

Edge: Start

Local Axis: (3)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (a)

Section Type: rccs

Constant Properties

Knowledge Factor, $\gamma = 1.00$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE 41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength, $f_c = f_{c_lower_bound} = 16.00$

Existing material of Primary Member: Steel Strength, $f_s = f_{s_lower_bound} = 400.00$

Concrete Elasticity, $E_c = 21019.039$

Steel Elasticity, $E_s = 200000.00$

#####

Note: Especially for the calculation of γ for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE 41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE 41-17).

Existing material: Concrete Strength, $f_c = f_{cm} = 20.00$

Existing material: Steel Strength, $f_s = f_{sm} = 444.44$

#####

Diameter, $D = 400.00$

Cover Thickness, $c = 25.00$

Element Length, $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length $l_o = l_b = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness, $t = 1.016$

Tensile Strength, $f_{fu} = 1055.00$

Tensile Modulus, $E_f = 64828.00$

Elongation, $e_{fu} = 0.01$

Number of directions, $NoDir = 1$

Fiber orientations, θ_i : 0.00°
Number of layers, NL = 1
Radius of rounding corners, R = 40.00

Stepwise Properties

EDGE -A-
Bending Moment, M_a = 6.2300792E-010
Shear Force, V_a = -1.7064698E-013
EDGE -B-
Bending Moment, M_b = -1.1086227E-010
Shear Force, V_b = 1.7064698E-013
BOTH EDGES
Axial Force, F = -4783.229
Longitudinal Reinforcement Area Distribution (in 2 divisions)
-Tension: A_{st} = 1272.345
-Compression: A_{sc} = 1781.283
Longitudinal Reinforcement Area Distribution (in 3 divisions)
-Tension: $A_{st,ten}$ = 1017.876
-Compression: $A_{sc,com}$ = 1017.876
-Middle: $A_{st,mid}$ = 1017.876
Mean Diameter of Tension Reinforcement, $D_{bL,ten}$ = 18.00

Existing component: From table 7-7, ASCE 41_17: Final Shear Capacity $V_R = \phi V_n = 286339.856$
 V_n ((10.3), ASCE 41-17) = $k_n V_{CoI} = 286339.856$
 $V_{CoI} = 286339.856$
 $k_n = 1.00$
displacement_ductility_demand = 0.00

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + \phi V_f$ ' where V_f is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)
 $f_c' = 16.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)
 $M/Vd = 2.00$
 $M_u = 6.2300792E-010$
 $V_u = 1.7064698E-013$
 $d = 0.8 \cdot D = 320.00$
 $N_u = 4783.229$
 $A_g = 125663.706$
From (11.5.4.8), ACI 318-14: $V_s = 0.00$
 $A_v = \phi / 2 \cdot A_{stirrup} = 123370.055$
 $f_y = 400.00$
 $s = 360.00$
 V_s is multiplied by $\phi_{CoI} = 0.00$
 $s/d = 1.125$
 V_f ((11-3)-(11.4), ACI 440) = 194961.134
 $\phi = 0.95$, for fully-wrapped sections
 $w_f/s_f = 1$ (FRP strips adjacent to one another).
In (11.3) $\sin \theta + \cos \theta$ is replaced with $(\cot \theta + \csc \theta) \sin \alpha$ which is more a generalised expression, where θ is the angle of the crack direction (see KANEPE).
This later relation, considered as a function $V_f(\theta, \alpha)$, is implemented for every different fiber orientation α_i , as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.
orientation 1: $\theta = \theta_1 + 90^\circ = 90.00$
 $V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, \theta_1)|)$, with:
total thickness per orientation, $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$
 $d_{fv} = d$ (figure 11.2, ACI 440) = 370.00
 f_{fe} ((11-5), ACI 440) = 259.312
 $E_f = 64828.00$
 $f_e = 0.004$, from (11.6a), ACI 440
with $f_u = 0.01$
From (11-11), ACI 440: $V_s + V_f \leq 213705.936$
 $b_w \cdot d = \phi \cdot d^2 / 4 = 80424.772$

displacement_ductility_demand is calculated as δ / y

- Calculation of δ / y for END A -
for rotation axis 2 and integ. section (a)

From analysis, chord rotation $\theta = 1.4283643E-020$
 $y = (M_y * L_s / 3) / E_{eff} = 0.00931733$ ((4.29), Biskinis Phd)
 $M_y = 1.4766E+008$
 $L_s = M/V$ (with $L_s > 0.1 * L$ and $L_s < 2 * L$) = 1500.00
From table 10.5, ASCE 41_17: $E_{eff} = factor * E_c * I_g = 7.9240E+012$
factor = 0.30
Ag = 125663.706
fc' = 20.00
N = 4783.229
 $E_c * I_g = 2.6413E+013$

Calculation of Yielding Moment M_y

Calculation of δ / y and M_y according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y_ten}, M_{y_com}) = 1.4766E+008$
 $y = 8.1688588E-006$
 M_{y_ten} (8c) = 1.4766E+008
 δ / y (7c) = 72.4069
error of function (7c) = 0.00117279
 M_{y_com} (8d) = 3.7493E+008
 δ / y (7d) = 69.91139
error of function (7d) = 0.00300901
with ((10.1), ASCE 41-17) $e_y = \text{Min}(e_y, 1.25 * e_y * (l_b / l_d)^{2/3}) = 0.0022222$
eco = 0.002
apl = 0.45 ((9c) in Biskinis and Fardis for FRP Wrap)
d1 = 44.00
R = 200.00
v = 0.00157746
N = 4783.229
Ac = 125663.706
((10.1), ASCE 41-17) $e_y = \text{Min}(e_y, 1.25 * e_y * (l_b / l_d)^{2/3}) = 0.44757577$
with fc' ((12.3), ACI 440) = 24.12975
fc = 20.00
fl = 1.3173
k = 1
Effective FRP thickness, tf = NL * t * Cos(b1) = 1.016
efe ((12.5) and (12.7)) = 0.004
fu = 0.01
Ef = 64828.00

Calculation of ratio l_b / l_d

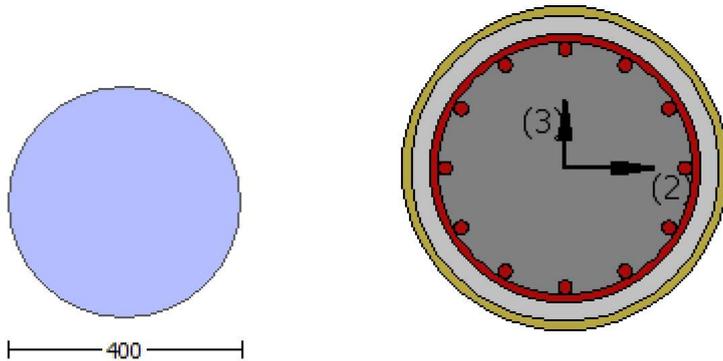
Lap Length: $l_d / l_{d,min} = 0.33490748$
 $l_b = 300.00$
 $l_d = 895.7698$
Calculation of I according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
= 1
db = 18.00
Mean strength value of all re-bars: fy = 444.44
fc' = 20.00, but $fc'^{0.5} < 8.3$ MPa (22.5.3.1, ACI 318-14)
t = 1.00
s = 0.80
e = 1.00
cb = 25.00

$K_{tr} = 1.14232$
 $A_{tr} = \sqrt{2} * \text{Area of stirrup} = 123.3701$
 $s = 360.00$
 $n = 12.00$

 End Of Calculation of Shear Capacity for element: column CC1 of floor 1
 At local axis: 3
 Integration Section: (a)

Calculation No. 12

column C1, Floor 1
 Limit State: Collapse Prevention (data interpolation between analysis steps 2 and 3)
 Analysis: Uniform +X
 Check: Chord rotation capacity (θ)
 Edge: Start
 Local Axis: (3)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1
 At Shear local axis: 3
 (Bending local axis: 2)
 Section Type: rccs

Constant Properties

 Knowledge Factor, $\gamma = 1.00$
 Mean strength values are used for both shear and moment calculations.
 Consequently:
 Existing material of Primary Member: Concrete Strength, $f_c = f_{cm} = 20.00$
 Existing material of Primary Member: Steel Strength, $f_s = f_{sm} = 444.44$
 Concrete Elasticity, $E_c = 21019.039$
 Steel Elasticity, $E_s = 200000.00$

 Note: Especially for the calculation of moment strengths,
 the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14
 Existing material: Steel Strength, $f_s = 1.25 * f_{sm} = 555.55$
 #####

Diameter, D = 400.00
 Cover Thickness, c = 25.00
 Mean Confinement Factor overall section = 1.84055
 Element Length, L = 3000.00
 Primary Member
 Ribbed Bars
 Ductile Steel
 Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)
 Longitudinal Bars With Ends Lapped Starting at the End Sections
 Lap Length l_o = 300.00
 FRP Wrapping Data
 Type: Carbon
 Cured laminate properties (design values)
 Thickness, t = 1.016
 Tensile Strength, f_{fu} = 1055.00
 Tensile Modulus, E_f = 64828.00
 Elongation, e_{fu} = 0.01
 Number of directions, NoDir = 1
 Fiber orientations, b_i : 0.00°
 Number of layers, NL = 1
 Radius of rounding corners, R = 40.00

 Stepwise Properties

At local axis: 3
 EDGE -A-
 Shear Force, V_a = 1.2107607E-031
 EDGE -B-
 Shear Force, V_b = -1.2107607E-031
 BOTH EDGES
 Axial Force, F = -4771.233
 Longitudinal Reinforcement Area Distribution (in 2 divisions)
 -Tension: A_{sl} = 0.00
 -Compression: A_{sc} = 3053.628
 Longitudinal Reinforcement Area Distribution (in 3 divisions)
 -Tension: $A_{sl,ten}$ = 1017.876
 -Compression: $A_{sl,com}$ = 1017.876
 -Middle: $A_{sl,mid}$ = 1017.876

 Calculation of Shear Capacity ratio , V_e/V_r = 0.30252729
 Member Controlled by Flexure ($V_e/V_r < 1$)
 Calculation of Shear Demand from fig. R18.6.5, ACI 318-14 $V_e = (M_{pr1} + M_{pr2})/l_n = 90202.132$
 with
 $M_{pr1} = \text{Max}(Mu_{1+} , Mu_{1-}) = 1.3530E+008$
 $Mu_{1+} = 1.3530E+008$, is the ultimate moment strength at the edge 1 of the member in the actual moment direction
 which is defined for the static loading combination
 $Mu_{1-} = 1.3530E+008$, is the ultimate moment strength at the edge 1 of the member in the opposite moment
 direction which is defined for the static loading combination
 $M_{pr2} = \text{Max}(Mu_{2+} , Mu_{2-}) = 1.3530E+008$
 $Mu_{2+} = 1.3530E+008$, is the ultimate moment strength at the edge 2 of the member in the actual moment direction
 which is defined for the the static loading combination
 $Mu_{2-} = 1.3530E+008$, is the ultimate moment strength at the edge 2 of the member in the opposite moment
 direction which is defined for the the static loading combination

 Calculation of Mu_{1+}

 Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), M_u
 $M_u = 1.3530E+008$

 = 0.90757121

$\lambda = 0.80580716$
 error of function (3.68), Biskinis Phd = 28928.286
 From 5A.2, TBDY: $f_{cc} = f_c \cdot \lambda = 36.81095$
 conf. factor $\lambda = 1.84055$
 $f_c = 20.00$
 From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$
 $l_b/l_d = 0.26792599$
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00155946$
 $N = 4771.233$
 $A_c = 125663.706$
 $= \lambda \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$
 $l_b = 300.00$
 $l_d = 1119.712$
 Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
 $\lambda = 1$
 $d_b = 18.00$
 Mean strength value of all re-bars: $f_y = 555.55$
 $f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)
 $t = 1.00$
 $s = 0.80$
 $e = 1.00$
 $c_b = 25.00$
 $K_{tr} = 1.14232$
 $A_{tr} = \lambda / 2 \cdot \text{Area of stirrup} = 123.3701$
 $s = 360.00$
 $n = 12.00$

Calculation of μ_1

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ
 $\mu = 1.3530E+008$

$\lambda = 0.90757121$
 $\lambda = 0.80580716$
 error of function (3.68), Biskinis Phd = 28928.286
 From 5A.2, TBDY: $f_{cc} = f_c \cdot \lambda = 36.81095$
 conf. factor $\lambda = 1.84055$
 $f_c = 20.00$
 From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$
 $l_b/l_d = 0.26792599$
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00155946$
 $N = 4771.233$
 $A_c = 125663.706$
 $= \lambda \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$
 $l_b = 300.00$
 $l_d = 1119.712$
 Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
 $\lambda = 1$
 $d_b = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$
 $f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)
 $t = 1.00$
 $s = 0.80$
 $e = 1.00$
 $cb = 25.00$
 $K_{tr} = 1.14232$
 $A_{tr} = \frac{1}{2} * \text{Area of stirrup} = 123.3701$
 $s = 360.00$
 $n = 12.00$

Calculation of μ_{2+}

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ
 $\mu = 1.3530E+008$

 $\phi = 0.90757121$
 $\lambda = 0.80580716$
error of function (3.68), Biskinis Phd = 28928.286
From 5A.2, TBDY: $f_{cc} = f_c' * c = 36.81095$
conf. factor $c = 1.84055$
 $f_c = 20.00$
From 10.3.5, ASCE 41-17, Final value of f_y : $f_y * \text{Min}(1, 1.25 * (l_b/d)^{2/3}) = 288.6089$
 $l_b/d = 0.26792599$
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00155946$
 $N = 4771.233$
 $A_c = 125663.706$
 $\phi * \text{Min}(1, 1.25 * (l_b/d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/d

Lap Length: $l_b/d = 0.26792599$
 $l_b = 300.00$
 $l_d = 1119.712$
Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
 $\phi = 1$

$db = 18.00$
Mean strength value of all re-bars: $f_y = 555.55$
 $f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)
 $t = 1.00$
 $s = 0.80$
 $e = 1.00$
 $cb = 25.00$
 $K_{tr} = 1.14232$
 $A_{tr} = \frac{1}{2} * \text{Area of stirrup} = 123.3701$
 $s = 360.00$
 $n = 12.00$

Calculation of μ_{2-}

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ
 $\mu = 1.3530E+008$

 $\phi = 0.90757121$
 $\lambda = 0.80580716$
error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$
conf. factor $c = 1.84055$
 $f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$
 $l_b/l_d = 0.26792599$
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00155946$
 $N = 4771.233$
 $A_c = 125663.706$
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$d_b = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of Shear Strength $V_r = \text{Min}(V_{r1}, V_{r2}) = 298161.965$

Calculation of Shear Strength at edge 1, $V_{r1} = 298161.965$

$V_{r1} = V_{Col}$ ((10.3), ASCE 41-17) = $k_{nl} \cdot V_{Col0}$

$V_{Col0} = 298161.965$

$k_{nl} = 1$ (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where V_f is the contribution of FRPs (11.3), ACI 440).

$= 1$ (normal-weight concrete)

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.2899604E-011$

$\nu_u = 1.2107607E-031$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14: $V_s = 0.00$

$A_v = \frac{1}{2} \cdot A_{stirrup} = 123370.055$

$f_y = 444.44$

$s = 360.00$

V_s is multiplied by $\text{Col} = 0.00$

$s/d = 1.125$

V_f ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$, for fully-wrapped sections

$w_f/s_f = 1$ (FRP strips adjacent to one another).

In (11.3) $\sin \theta + \cos \theta$ is replaced with $(\cot \theta + \csc \theta) \sin \alpha$ which is more a generalised expression, where θ is the angle of the crack direction (see KANEPE).

This later relation, considered as a function $V_f(\theta, \alpha)$, is implemented for every different fiber orientation α_i , as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.

orientation 1: $\theta = 45^\circ + 90^\circ = 135^\circ$

$V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, a1)|)$, with:
total thickness per orientation, $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$
 $d_{fv} = d$ (figure 11.2, ACI 440) = 370.00
 f_{fe} ((11-5), ACI 440) = 259.312
 $E_f = 64828.00$
 $f_e = 0.004$, from (11.6a), ACI 440
with $f_u = 0.01$
From (11-11), ACI 440: $V_s + V_f \leq 238930.50$
 $b_w \cdot d = \frac{1}{4} \cdot d \cdot d = 80424.772$

Calculation of Shear Strength at edge 2, $V_{r2} = 298161.965$
 $V_{r2} = V_{Col}$ ((10.3), ASCE 41-17) = $k_{nl} \cdot V_{Col0}$
 $V_{Col0} = 298161.965$
 $k_{nl} = 1$ (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ '
where V_f is the contribution of FRPs (11.3), ACI 440).

 $\lambda = 1$ (normal-weight concrete)
 $f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)
 $M / Vd = 2.00$
 $\mu_u = 1.2899604E-011$
 $\nu_u = 1.2107607E-031$
 $d = 0.8 \cdot D = 320.00$
 $N_u = 4771.233$
 $A_g = 125663.706$
From (11.5.4.8), ACI 318-14: $V_s = 0.00$
 $A_v = \frac{1}{2} \cdot A_{stirrup} = 123370.055$
 $f_y = 444.44$
 $s = 360.00$

V_s is multiplied by $\lambda = 1.00$
 $s/d = 1.125$

V_f ((11-3)-(11.4), ACI 440) = 194961.134
 $f = 0.95$, for fully-wrapped sections

$w_f / s_f = 1$ (FRP strips adjacent to one another).

In (11.3) $\sin \theta + \cos \theta$ is replaced with $(\cot \theta + \csc \theta) \sin \alpha$ which is more a generalised expression,
where θ is the angle of the crack direction (see KANEPE).

This later relation, considered as a function $V_f(\theta, \alpha)$, is implemented for every different fiber orientation α_i ,
as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.

orientation 1: $\theta = 45^\circ + 90^\circ = 135^\circ$

$V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, a1)|)$, with:
total thickness per orientation, $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$
 $d_{fv} = d$ (figure 11.2, ACI 440) = 370.00
 f_{fe} ((11-5), ACI 440) = 259.312
 $E_f = 64828.00$
 $f_e = 0.004$, from (11.6a), ACI 440
with $f_u = 0.01$
From (11-11), ACI 440: $V_s + V_f \leq 238930.50$
 $b_w \cdot d = \frac{1}{4} \cdot d \cdot d = 80424.772$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1
At local axis: 3

Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1
At Shear local axis: 2
(Bending local axis: 3)
Section Type: rccs

Constant Properties

Knowledge Factor, $\lambda = 1.00$
Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength, $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength, $f_s = f_{sm} = 444.44$

Concrete Elasticity, $E_c = 21019.039$

Steel Elasticity, $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength, $f_s = 1.25 \cdot f_{sm} = 555.55$

#####

Diameter, $D = 400.00$

Cover Thickness, $c = 25.00$

Mean Confinement Factor overall section = 1.84055

Element Length, $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length $l_o = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness, $t = 1.016$

Tensile Strength, $f_{fu} = 1055.00$

Tensile Modulus, $E_f = 64828.00$

Elongation, $e_{fu} = 0.01$

Number of directions, $N_{oDir} = 1$

Fiber orientations, $b_i = 0.00^\circ$

Number of layers, $N_L = 1$

Radius of rounding corners, $R = 40.00$

Stepwise Properties

At local axis: 2

EDGE -A-

Shear Force, $V_a = -7.4135260E-048$

EDGE -B-

Shear Force, $V_b = 7.4135260E-048$

BOTH EDGES

Axial Force, $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension: $A_{sl,t} = 0.00$

-Compression: $A_{sl,c} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension: $A_{sl,ten} = 1017.876$

-Compression: $A_{sl,com} = 1017.876$

-Middle: $A_{sl,mid} = 1017.876$

Calculation of Shear Capacity ratio, $V_e/V_r = 0.30252729$

Member Controlled by Flexure ($V_e/V_r < 1$)

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14 $V_e = (M_{pr1} + M_{pr2})/l_n = 90202.132$

with

$M_{pr1} = \text{Max}(M_{u1+}, M_{u1-}) = 1.3530E+008$

$M_{u1+} = 1.3530E+008$, is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$M_{u1-} = 1.3530E+008$, is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(M_{u2+}, M_{u2-}) = 1.3530E+008$

$M_{u2+} = 1.3530E+008$, is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination

$M_{u2-} = 1.3530E+008$, is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

Calculation of Mu1+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu
Mu = 1.3530E+008

= 0.90757121
' = 0.80580716
error of function (3.68), Biskinis Phd = 28928.286
From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$
conf. factor $c = 1.84055$
 $f_c = 20.00$
From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 288.6089$
 $l_b/d = 0.26792599$
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00155946$
 $N = 4771.233$
 $A_c = 125663.706$
= $\cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/d

Lap Length: $l_b/d = 0.26792599$
 $l_b = 300.00$
 $l_d = 1119.712$
Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
= 1
 $d_b = 18.00$
Mean strength value of all re-bars: $f_y = 555.55$
 $f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3 \text{ MPa}$ (22.5.3.1, ACI 318-14)
 $t = 1.00$
 $s = 0.80$
 $e = 1.00$
 $cb = 25.00$
 $K_{tr} = 1.14232$
 $A_{tr} = \sqrt{2} \cdot \text{Area of stirrup} = 123.3701$
 $s = 360.00$
 $n = 12.00$

Calculation of Mu1-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu
Mu = 1.3530E+008

= 0.90757121
' = 0.80580716
error of function (3.68), Biskinis Phd = 28928.286
From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$
conf. factor $c = 1.84055$
 $f_c = 20.00$
From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 288.6089$
 $l_b/d = 0.26792599$
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00155946$
 $N = 4771.233$
 $A_c = 125663.706$
= $\cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of l_b, \min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$db = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f'_c = 20.00$, but $f_c^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$cb = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of μ_{2+}

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ

$\mu = 1.3530E+008$

$= 0.90757121$

$' = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY: $f_{cc} = f'_c \cdot c = 36.81095$

conf. factor $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of l_b, \min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$db = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f'_c = 20.00$, but $f_c^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$cb = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of μ_2 -

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ

$\mu = 1.3530E+008$

$\mu = 0.90757121$

$\mu = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$

conf. factor $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$d_b = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of Shear Strength $V_r = \text{Min}(V_{r1}, V_{r2}) = 298161.965$

Calculation of Shear Strength at edge 1, $V_{r1} = 298161.965$

$V_{r1} = V_{Co1}$ ((10.3), ASCE 41-17) = $k_{nl} \cdot V_{Co10}$

$V_{Co10} = 298161.965$

$k_{nl} = 1$ (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where V_f is the contribution of FRPs (11.3), ACI 440).

$= 1$ (normal-weight concrete)

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu = 5.9321525E-012$

$V_u = 7.4135260E-048$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14: $V_s = 0.00$

$A_v = \frac{1}{2} \cdot A_{\text{stirrup}} = 123370.055$

$f_y = 444.44$

$s = 360.00$

Vs is multiplied by Col = 0.00

$$s/d = 1.125$$

$$V_f \text{ ((11-3)-(11.4), ACI 440) } = 194961.134$$

f = 0.95, for fully-wrapped sections

wf/sf = 1 (FRP strips adjacent to one another).

In (11.3) $\sin \theta + \cos \theta$ is replaced with $(\cot \theta + \csc \theta) \sin \alpha$ which is more a generalised expression, where θ is the angle of the crack direction (see KANEPE).

This later relation, considered as a function $V_f(\theta, \alpha)$, is implemented for every different fiber orientation α_i , as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.

$$\text{orientation 1: } \theta = b1 + 90^\circ = 90.00$$

$$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \theta)|), \text{ with:}$$

$$\text{total thickness per orientation, } t_{f1} = NL * t / \text{NoDir} = 1.016$$

$$d_{fv} = d \text{ (figure 11.2, ACI 440) } = 370.00$$

$$f_{fe} \text{ ((11-5), ACI 440) } = 259.312$$

$$E_f = 64828.00$$

$$f_e = 0.004, \text{ from (11.6a), ACI 440}$$

$$\text{with } f_u = 0.01$$

$$\text{From (11-11), ACI 440: } V_s + V_f \leq 238930.50$$

$$b_w * d = \rho * d^2 / 4 = 80424.772$$

Calculation of Shear Strength at edge 2, $V_{r2} = 298161.965$

$$V_{r2} = V_{Col} \text{ ((10.3), ASCE 41-17) } = k_{nl} * V_{Col0}$$

$$V_{Col0} = 298161.965$$

$$k_{nl} = 1 \text{ (zero step-static loading)}$$

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ ' where V_f is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

$$f_c' = 20.00, \text{ but } f_c^{0.5} \leq 8.3 \text{ MPa (22.5.3.1, ACI 318-14)}$$

$$M / Vd = 2.00$$

$$\mu_u = 5.9321525E-012$$

$$V_u = 7.4135260E-048$$

$$d = 0.8 * D = 320.00$$

$$N_u = 4771.233$$

$$A_g = 125663.706$$

$$\text{From (11.5.4.8), ACI 318-14: } V_s = 0.00$$

$$A_v = \rho / 2 * A_{stirup} = 123370.055$$

$$f_y = 444.44$$

$$s = 360.00$$

Vs is multiplied by Col = 0.00

$$s/d = 1.125$$

$$V_f \text{ ((11-3)-(11.4), ACI 440) } = 194961.134$$

f = 0.95, for fully-wrapped sections

wf/sf = 1 (FRP strips adjacent to one another).

In (11.3) $\sin \theta + \cos \theta$ is replaced with $(\cot \theta + \csc \theta) \sin \alpha$ which is more a generalised expression, where θ is the angle of the crack direction (see KANEPE).

This later relation, considered as a function $V_f(\theta, \alpha)$, is implemented for every different fiber orientation α_i , as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.

$$\text{orientation 1: } \theta = b1 + 90^\circ = 90.00$$

$$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \theta)|), \text{ with:}$$

$$\text{total thickness per orientation, } t_{f1} = NL * t / \text{NoDir} = 1.016$$

$$d_{fv} = d \text{ (figure 11.2, ACI 440) } = 370.00$$

$$f_{fe} \text{ ((11-5), ACI 440) } = 259.312$$

$$E_f = 64828.00$$

$$f_e = 0.004, \text{ from (11.6a), ACI 440}$$

$$\text{with } f_u = 0.01$$

$$\text{From (11-11), ACI 440: } V_s + V_f \leq 238930.50$$

$$b_w * d = \rho * d^2 / 4 = 80424.772$$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At local axis: 2

Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (a)

Section Type: rccs

Constant Properties

Knowledge Factor, $\gamma = 1.00$

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE 41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength, $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength, $f_s = f_{sm} = 444.44$

Concrete Elasticity, $E_c = 21019.039$

Steel Elasticity, $E_s = 200000.00$

Diameter, $D = 400.00$

Cover Thickness, $c = 25.00$

Element Length, $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length $l_b = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness, $t = 1.016$

Tensile Strength, $f_{fu} = 1055.00$

Tensile Modulus, $E_f = 64828.00$

Elongation, $e_{fu} = 0.01$

Number of directions, $N_{oDir} = 1$

Fiber orientations, $b_i = 0.00^\circ$

Number of layers, $N_L = 1$

Radius of rounding corners, $R = 40.00$

Stepwise Properties

Bending Moment, $M = -1.3471E+007$

Shear Force, $V_2 = -4486.792$

Shear Force, $V_3 = -1.7064698E-013$

Axial Force, $F = -4783.229$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension: $A_{st} = 1272.345$

-Compression: $A_{sc} = 1781.283$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension: $A_{st,ten} = 1017.876$

-Compression: $A_{st,com} = 1017.876$

-Middle: $A_{st,mid} = 1017.876$

Mean Diameter of Tension Reinforcement, $DbL = 18.00$

Existing component: From table 7-7, ASCE 41_17: Final chord rotation Capacity $u_{R} = \gamma \cdot u = 0.06064917$

$u = \gamma \cdot u = 0.06064917$

- Calculation of γ -

$\gamma = (M \cdot L_s / 3) / E_{eff} = 0.01864917$ ((4.29), Biskinis Phd))

$M_y = 1.4766E+008$

$L_s = M/V$ (with $L_s > 0.1 \cdot L$ and $L_s < 2 \cdot L$) = 3002.334

From table 10.5, ASCE 41_17: $E_{eff} = factor \cdot E_c \cdot I_g = 7.9240E+012$

factor = 0.30

Ag = 125663.706
fc' = 20.00
N = 4783.229
Ec*Ig = 2.6413E+013

Calculation of Yielding Moment My

Calculation of ϕ_y and My according to (7) - (8) in Biskinis and Fardis

My = Min(My_ten, My_com) = 1.4766E+008
y = 8.1688588E-006
My_ten (8c) = 1.4766E+008
_ten (7c) = 72.4069
error of function (7c) = 0.00117279
My_com (8d) = 3.7493E+008
_com (7d) = 69.91139
error of function (7d) = 0.00300901
with ((10.1), ASCE 41-17) $\phi_y = \text{Min}(\phi_y, 1.25 * \phi_y * (l_b/l_d)^{2/3}) = 0.0022222$
eco = 0.002
apl = 0.45 ((9c) in Biskinis and Fardis for FRP Wrap)
d1 = 44.00
R = 200.00
v = 0.00157746
N = 4783.229
Ac = 125663.706
((10.1), ASCE 41-17) $\phi_y = \text{Min}(\phi_y, 1.25 * \phi_y * (l_b/l_d)^{2/3}) = 0.44757577$
with fc' ((12.3), ACI 440) = 24.12975
fc = 20.00
fl = 1.3173
k = 1
Effective FRP thickness, tf = NL*t*cos(b1) = 1.016
efe ((12.5) and (12.7)) = 0.004
fu = 0.01
Ef = 64828.00

Calculation of ratio lb/l_d

Lap Length: $l_d/l_{d,min} = 0.33490748$
lb = 300.00
ld = 895.7698
Calculation of l according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
= 1
db = 18.00
Mean strength value of all re-bars: fy = 444.44
fc' = 20.00, but $fc'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)
t = 1.00
s = 0.80
e = 1.00
cb = 25.00
Ktr = 1.14232
Atr = $\sqrt{2} * \text{Area of stirrup} = 123.3701$
s = 360.00
n = 12.00

- Calculation of ρ_p -

From table 10-9: $\rho_p = 0.042$

with:

- Columns controlled by inadequate development or splicing along the clear height because $l_b/l_d < 1$
shear control ratio $V_y/E/ColOE = 0.30252729$
d = 0.00
s = 0.00

$$t = 2 \cdot A_v / (d_c \cdot s) + 4 \cdot t_f / D \cdot (f_{fe} / f_s) = 0.00721126$$

$A_v = 78.53982$, is the area of the circular stirrup

$d_c = D - 2 \cdot \text{cover} - \text{Hoop Diameter} = 340.00$

The term $2 \cdot t_f / b_w \cdot (f_{fe} / f_s)$ is implemented to account for FRP contribution

where $f = 2 \cdot t_f / b_w$ is FRP ratio (EC8 - 3, A.4.4.3(6)) and f_{fe} / f_s normalises f to steel strength

All these variables have already been given in Shear control ratio calculation.

$$N_{UD} = 4783.229$$

$$A_g = 125663.706$$

$$f_{cE} = 20.00$$

$$f_{tE} = f_{yE} = 444.44$$

$$p_l = \text{Area_Tot_Long_Rein} / (A_g) = 0.0243$$

$$f_{cE} = 20.00$$

End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (a)

Calculation No. 13

column C1, Floor 1

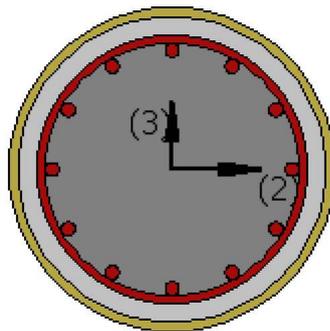
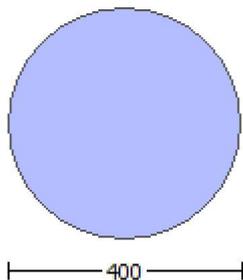
Limit State: Collapse Prevention (data interpolation between analysis steps 2 and 3)

Analysis: Uniform +X

Check: Shear capacity V_{Rd}

Edge: End

Local Axis: (2)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (b)

Section Type: rccs

Constant Properties

Knowledge Factor, $\gamma = 1.00$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE 41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength, $f_c = f_{c_lower_bound} = 16.00$

Existing material of Primary Member: Steel Strength, $f_s = f_{s_lower_bound} = 400.00$

Concrete Elasticity, $E_c = 21019.039$

Steel Elasticity, $E_s = 200000.00$

#####

Note: Especially for the calculation of γ for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE 41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE 41-17).

Existing material: Concrete Strength, $f_c = f_{cm} = 20.00$

Existing material: Steel Strength, $f_s = f_{sm} = 444.44$

#####

Diameter, $D = 400.00$

Cover Thickness, $c = 25.00$

Element Length, $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length $l_o = l_b = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness, $t = 1.016$

Tensile Strength, $f_{fu} = 1055.00$

Tensile Modulus, $E_f = 64828.00$

Elongation, $e_{fu} = 0.01$

Number of directions, $NoDir = 1$

Fiber orientations, $b_i = 0.00^\circ$

Number of layers, $NL = 1$

Radius of rounding corners, $R = 40.00$

Stepwise Properties

EDGE -A-

Bending Moment, $M_a = -1.3471E+007$

Shear Force, $V_a = -4486.792$

EDGE -B-

Bending Moment, $M_b = 2684.487$

Shear Force, $V_b = 4486.792$

BOTH EDGES

Axial Force, $F = -4783.229$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension: $A_{sl,t} = 0.00$

-Compression: $A_{sl,c} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension: $A_{sl,ten} = 1017.876$

-Compression: $A_{sl,com} = 1017.876$

-Middle: $A_{sl,mid} = 1017.876$

Mean Diameter of Tension Reinforcement, $Db_{L,ten} = 18.00$

Existing component: From table 7-7, ASCE 41_17: Final Shear Capacity $V_R = \phi V_n = 286339.856$

V_n ((10.3), ASCE 41-17) = $k_n \phi V_{CoI} = 286339.856$

$V_{CoI} = 286339.856$

$k_n = 1.00$

displacement_ductility_demand = 0.13687722

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_{s+} = \phi \cdot V_f$ ' where V_f is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

$f'_c = 16.00$, but $f'_c \cdot 0.5 \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$
 $\mu_u = 2684.487$
 $V_u = 4486.792$
 $d = 0.8 \cdot D = 320.00$
 $N_u = 4783.229$
 $A_g = 125663.706$
 From (11.5.4.8), ACI 318-14: $V_s = 0.00$
 $A_v = \sqrt{2} \cdot A_{stirrup} = 123370.055$
 $f_y = 400.00$
 $s = 360.00$
 V_s is multiplied by $\text{Col} = 0.00$
 $s/d = 1.125$
 V_f ((11-3)-(11.4), ACI 440) = 194961.134
 $f = 0.95$, for fully-wrapped sections
 $w_f/s_f = 1$ (FRP strips adjacent to one another).
 In (11.3) $\sin + \cos$ is replaced with $(\cot + \cot a) \sin a$ which is more a generalised expression,
 where a is the angle of the crack direction (see KANEPE).
 This later relation, considered as a function $V_f(\theta, a)$, is implemented for every different fiber orientation a_i ,
 as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.
 orientation 1: $\theta = 45^\circ + 90^\circ = 135^\circ$
 $V_f = \text{Min}(|V_f(45, a_1)|, |V_f(-45, a_1)|)$, with:
 total thickness per orientation, $t_{f1} = N_L \cdot t / N_{Dir} = 1.016$
 $d_{fv} = d$ (figure 11.2, ACI 440) = 370.00
 f_{fe} ((11-5), ACI 440) = 259.312
 $E_f = 64828.00$
 $f_{e} = 0.004$, from (11.6a), ACI 440
 with $f_u = 0.01$
 From (11-11), ACI 440: $V_s + V_f \leq 213705.936$
 $b_w \cdot d = N_u \cdot d / 4 = 80424.772$

displacement ductility demand is calculated as δ / y

- Calculation of δ / y for END B -
 for rotation axis 3 and integ. section (b)

From analysis, chord rotation $\theta = 0.00025507$
 $y = (M_y \cdot L_s / 3) / E_{eff} = 0.00186347$ ((4.29), Biskinis Phd)
 $M_y = 1.4766E+008$
 $L_s = M/V$ (with $L_s > 0.1 \cdot L$ and $L_s < 2 \cdot L$) = 300.00
 From table 10.5, ASCE 41_17: $E_{eff} = \text{factor} \cdot E_c \cdot I_g = 7.9240E+012$
 $\text{factor} = 0.30$
 $A_g = 125663.706$
 $f_c' = 20.00$
 $N = 4783.229$
 $E_c \cdot I_g = 2.6413E+013$

Calculation of Yielding Moment M_y

Calculation of δ / y and M_y according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y_ten}, M_{y_com}) = 1.4766E+008$
 $y = 8.1688588E-006$
 M_{y_ten} (8c) = 1.4766E+008
 δ_{ten} (7c) = 72.4069
 error of function (7c) = 0.00117279
 M_{y_com} (8d) = 3.7493E+008
 δ_{com} (7d) = 69.91139
 error of function (7d) = 0.00300901
 with ((10.1), ASCE 41-17) $e_y = \text{Min}(e_y, 1.25 \cdot e_y \cdot (I_b / I_d)^{2/3}) = 0.0022222$
 $e_{co} = 0.002$
 $a_{pl} = 0.45$ ((9c) in Biskinis and Fardis for FRP Wrap)
 $d_1 = 44.00$
 $R = 200.00$

$v = 0.00157746$
 $N = 4783.229$
 $A_c = 125663.706$
 $((10.1), ASCE 41-17) = \text{Min}(, 1.25 * \sqrt{(l_b/d)^2/3}) = 0.44757577$
 with f_c^* ((12.3), ACI 440) = 24.12975
 $f_c = 20.00$
 $f_l = 1.3173$
 $k = 1$
 Effective FRP thickness, $t_f = NL * t * \text{Cos}(b1) = 1.016$
 e_{fe} ((12.5) and (12.7)) = 0.004
 $f_u = 0.01$
 $E_f = 64828.00$

Calculation of ratio l_b/d

Lap Length: $l_d/l_{d,min} = 0.33490748$

$l_b = 300.00$

$l_d = 895.7698$

Calculation of I according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$db = 18.00$

Mean strength value of all re-bars: $f_y = 444.44$

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3 \text{ MPa}$ (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$cb = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \sqrt{2} * \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

End Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (b)

Calculation No. 14

column C1, Floor 1

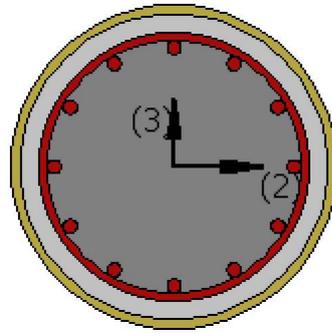
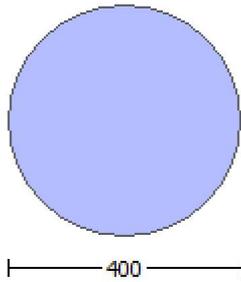
Limit State: Collapse Prevention (data interpolation between analysis steps 2 and 3)

Analysis: Uniform +X

Check: Chord rotation capacity (θ_u)

Edge: End

Local Axis: (2)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 3
 (Bending local axis: 2)
 Section Type: rccs

Constant Properties

Knowledge Factor, $\gamma = 1.00$
 Mean strength values are used for both shear and moment calculations.
 Consequently:
 Existing material of Primary Member: Concrete Strength, $f_c = f_{cm} = 20.00$
 Existing material of Primary Member: Steel Strength, $f_s = f_{sm} = 444.44$
 Concrete Elasticity, $E_c = 21019.039$
 Steel Elasticity, $E_s = 200000.00$

 Note: Especially for the calculation of moment strengths,
 the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14
 Existing material: Steel Strength, $f_s = 1.25 \cdot f_{sm} = 555.55$

 Diameter, $D = 400.00$
 Cover Thickness, $c = 25.00$
 Mean Confinement Factor overall section = 1.84055
 Element Length, $L = 3000.00$
 Primary Member
 Ribbed Bars
 Ductile Steel
 Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)
 Longitudinal Bars With Ends Lapped Starting at the End Sections
 Lap Length $l_o = 300.00$
 FRP Wrapping Data
 Type: Carbon
 Cured laminate properties (design values)
 Thickness, $t = 1.016$
 Tensile Strength, $f_{fu} = 1055.00$
 Tensile Modulus, $E_f = 64828.00$
 Elongation, $e_{fu} = 0.01$
 Number of directions, $NoDir = 1$
 Fiber orientations, $b_i: 0.00^\circ$
 Number of layers, $NL = 1$
 Radius of rounding corners, $R = 40.00$

Stepwise Properties

At local axis: 3
 EDGE -A-
 Shear Force, $V_a = 1.2107607E-031$
 EDGE -B-
 Shear Force, $V_b = -1.2107607E-031$
 BOTH EDGES
 Axial Force, $F = -4771.233$
 Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension: Aslt = 0.00

-Compression: Aslc = 3053.628

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension: Asl,ten = 1017.876

-Compression: Asl,com = 1017.876

-Middle: Asl,mid = 1017.876

Calculation of Shear Capacity ratio , $V_e/V_r = 0.30252729$

Member Controlled by Flexure ($V_e/V_r < 1$)

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14 $V_e = (M_{pr1} + M_{pr2})/l_n = 90202.132$

with

$M_{pr1} = \text{Max}(M_{u1+} , M_{u1-}) = 1.3530E+008$

$M_{u1+} = 1.3530E+008$, is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$M_{u1-} = 1.3530E+008$, is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(M_{u2+} , M_{u2-}) = 1.3530E+008$

$M_{u2+} = 1.3530E+008$, is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

$M_{u2-} = 1.3530E+008$, is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

Calculation of M_{u1+}

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), M_u

$M_u = 1.3530E+008$

$\phi = 0.90757121$

$\phi' = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$

conf. factor $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$\phi = \phi' \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of l_b ,min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$\beta = 1$

$d_b = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3 \text{ MPa}$ (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of Mu1-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu
Mu = 1.3530E+008

= 0.90757121
' = 0.80580716
error of function (3.68), Biskinis Phd = 28928.286
From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$
conf. factor $c = 1.84055$
 $f_c = 20.00$
From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$
 $l_b/l_d = 0.26792599$
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00155946$
 $N = 4771.233$
 $Ac = 125663.706$
= $\cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$
 $l_b = 300.00$
 $l_d = 1119.712$
Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
= 1
 $db = 18.00$
Mean strength value of all re-bars: $f_y = 555.55$
 $f_c' = 20.00$, but $f_c^{0.5} \leq 8.3 \text{ MPa}$ (22.5.3.1, ACI 318-14)
 $t = 1.00$
 $s = 0.80$
 $e = 1.00$
 $cb = 25.00$
 $K_{tr} = 1.14232$
 $A_{tr} = \sqrt{2} \cdot \text{Area of stirrup} = 123.3701$
 $s = 360.00$
 $n = 12.00$

Calculation of Mu2+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu
Mu = 1.3530E+008

= 0.90757121
' = 0.80580716
error of function (3.68), Biskinis Phd = 28928.286
From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$
conf. factor $c = 1.84055$
 $f_c = 20.00$
From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$
 $l_b/l_d = 0.26792599$
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00155946$
 $N = 4771.233$
 $Ac = 125663.706$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.28805051$$

Calculation of ratio l_b/d

Lap Length: $l_b/d = 0.26792599$

$l_b = 300.00$

$d = 1119.712$

Calculation of l_b, min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$d_b = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f'_c = 20.00$, but $f_c^{0.5} \leq 8.3 \text{ MPa}$ (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of μ_2

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ

$\mu = 1.3530 \times 10^8$

$= 0.90757121$

$' = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY: $f_{cc} = f'_c \cdot c = 36.81095$

conf. factor $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 288.6089$

$l_b/d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/d

Lap Length: $l_b/d = 0.26792599$

$l_b = 300.00$

$d = 1119.712$

Calculation of l_b, min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$d_b = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f'_c = 20.00$, but $f_c^{0.5} \leq 8.3 \text{ MPa}$ (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of Shear Strength $V_r = \text{Min}(V_{r1}, V_{r2}) = 298161.965$

Calculation of Shear Strength at edge 1, $V_{r1} = 298161.965$

$V_{r1} = V_{Col}$ ((10.3), ASCE 41-17) = $k_{nl} * V_{Col0}$

$V_{Col0} = 298161.965$

$k_{nl} = 1$ (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ '
where V_f is the contribution of FRPs (11.3), ACI 440).

 $\rho = 1$ (normal-weight concrete)

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$M / Vd = 2.00$

$\mu_u = 1.2899604E-011$

$\nu_u = 1.2107607E-031$

$d = 0.8 * D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14: $V_s = 0.00$

$A_v = \rho / 2 * A_{stirrup} = 123370.055$

$f_y = 444.44$

$s = 360.00$

V_s is multiplied by $\rho_{col} = 0.00$

$s/d = 1.125$

V_f ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$, for fully-wrapped sections

$w_f / s_f = 1$ (FRP strips adjacent to one another).

In (11.3) $\sin \theta + \cos \theta$ is replaced with $(\cot \theta + \csc \theta) \sin \alpha$ which is more a generalised expression,
where θ is the angle of the crack direction (see KANEPE).

This later relation, considered as a function $V_f(\theta, \alpha)$, is implemented for every different fiber orientation α_i ,
as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.

orientation 1: $\theta = b_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \alpha)|)$, with:

total thickness per orientation, $t_{f1} = N_L * t / N_{oDir} = 1.016$

$d_{fv} = d$ (figure 11.2, ACI 440) = 370.00

f_{fe} ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$, from (11.6a), ACI 440

with $f_u = 0.01$

From (11-11), ACI 440: $V_s + V_f \leq 238930.50$

$b_w * d = \rho * d^2 / 4 = 80424.772$

Calculation of Shear Strength at edge 2, $V_{r2} = 298161.965$

$V_{r2} = V_{Col}$ ((10.3), ASCE 41-17) = $k_{nl} * V_{Col0}$

$V_{Col0} = 298161.965$

$k_{nl} = 1$ (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ '
where V_f is the contribution of FRPs (11.3), ACI 440).

 $\rho = 1$ (normal-weight concrete)

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$M / Vd = 2.00$

$\mu_u = 1.2899604E-011$

$\nu_u = 1.2107607E-031$

$d = 0.8 * D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14: $V_s = 0.00$

$A_v = \rho / 2 * A_{stirrup} = 123370.055$

$f_y = 444.44$

$s = 360.00$

V_s is multiplied by $\rho_{col} = 0.00$

$s/d = 1.125$

Vf ((11-3)-(11.4), ACI 440) = 194961.134

f = 0.95, for fully-wrapped sections

wf/sf = 1 (FRP strips adjacent to one another).

In (11.3) $\sin \theta + \cos \theta$ is replaced with $(\cot \theta + \cot \alpha) \sin \alpha$ which is more a generalised expression, where θ is the angle of the crack direction (see KANEPE).

This later relation, considered as a function Vf(θ), is implemented for every different fiber orientation ai, as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.

orientation 1: $\theta_1 = b_1 + 90^\circ = 90.00$

Vf = Min(|Vf(45, 1)|, |Vf(-45, a1)|), with:

total thickness per orientation, tf1 = NL*t/NoDir = 1.016

dfv = d (figure 11.2, ACI 440) = 370.00

ffe ((11-5), ACI 440) = 259.312

Ef = 64828.00

fe = 0.004, from (11.6a), ACI 440

with fu = 0.01

From (11-11), ACI 440: Vs + Vf <= 238930.50

bw*d = $\frac{V_s + V_f}{4} = 80424.772$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At local axis: 3

Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 2

(Bending local axis: 3)

Section Type: rccs

Constant Properties

Knowledge Factor, $\phi = 1.00$

Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength, $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength, $f_s = f_{sm} = 444.44$

Concrete Elasticity, $E_c = 21019.039$

Steel Elasticity, $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength, $f_s = 1.25 * f_{sm} = 555.55$

#####

Diameter, D = 400.00

Cover Thickness, c = 25.00

Mean Confinement Factor overall section = 1.84055

Element Length, L = 3000.00

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length lo = 300.00

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness, t = 1.016

Tensile Strength, $f_{fu} = 1055.00$

Tensile Modulus, Ef = 64828.00

Elongation, $e_{fu} = 0.01$

Number of directions, NoDir = 1

Fiber orientations, bi: 0.00°

Number of layers, NL = 1

Radius of rounding corners, R = 40.00

Stepwise Properties

At local axis: 2

EDGE -A-

Shear Force, $V_a = -7.4135260E-048$

EDGE -B-

Shear Force, $V_b = 7.4135260E-048$

BOTH EDGES

Axial Force, $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension: $A_{sl,t} = 0.00$

-Compression: $A_{sl,c} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension: $A_{sl,ten} = 1017.876$

-Compression: $A_{sl,com} = 1017.876$

-Middle: $A_{sl,mid} = 1017.876$

Calculation of Shear Capacity ratio, $V_e/V_r = 0.30252729$

Member Controlled by Flexure ($V_e/V_r < 1$)

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14 $V_e = (M_{pr1} + M_{pr2})/l_n = 90202.132$

with

$M_{pr1} = \text{Max}(M_{u1+}, M_{u1-}) = 1.3530E+008$

$M_{u1+} = 1.3530E+008$, is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$M_{u1-} = 1.3530E+008$, is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(M_{u2+}, M_{u2-}) = 1.3530E+008$

$M_{u2+} = 1.3530E+008$, is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

$M_{u2-} = 1.3530E+008$, is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

Calculation of M_{u1+}

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), M_u

$M_u = 1.3530E+008$

$\lambda = 0.90757121$

$\lambda' = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$

conf. factor $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \lambda \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of l_b, min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$db = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$
 $f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)
 $t = 1.00$
 $s = 0.80$
 $e = 1.00$
 $cb = 25.00$
 $K_{tr} = 1.14232$
 $A_{tr} = \frac{1}{2} * \text{Area of stirrup} = 123.3701$
 $s = 360.00$
 $n = 12.00$

Calculation of μ_{1-}

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ
 $\mu = 1.3530E+008$

 $\phi = 0.90757121$
 $\lambda = 0.80580716$
error of function (3.68), Biskinis Phd = 28928.286
From 5A.2, TBDY: $f_{cc} = f_c' * c = 36.81095$
conf. factor $c = 1.84055$
 $f_c = 20.00$
From 10.3.5, ASCE 41-17, Final value of f_y : $f_y * \text{Min}(1, 1.25 * (l_b/d)^{2/3}) = 288.6089$
 $l_b/d = 0.26792599$
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00155946$
 $N = 4771.233$
 $A_c = 125663.706$
 $\phi * \text{Min}(1, 1.25 * (l_b/d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/d

Lap Length: $l_b/d = 0.26792599$
 $l_b = 300.00$
 $l_d = 1119.712$
Calculation of l_b, min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
 $\phi = 1$

$db = 18.00$
Mean strength value of all re-bars: $f_y = 555.55$
 $f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)
 $t = 1.00$
 $s = 0.80$
 $e = 1.00$
 $cb = 25.00$
 $K_{tr} = 1.14232$
 $A_{tr} = \frac{1}{2} * \text{Area of stirrup} = 123.3701$
 $s = 360.00$
 $n = 12.00$

Calculation of μ_{2+}

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ
 $\mu = 1.3530E+008$

 $\phi = 0.90757121$
 $\lambda = 0.80580716$
error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$

conf. factor $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$db = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$cb = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of μ_2

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ

$\mu = 1.3530E+008$

$= 0.90757121$

$' = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$

conf. factor $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$db = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

t = 1.00
s = 0.80
e = 1.00
cb = 25.00
Ktr = 1.14232
Atr = /2 * Area of stirrup = 123.3701
s = 360.00
n = 12.00

Calculation of Shear Strength $V_r = \text{Min}(V_{r1}, V_{r2}) = 298161.965$

Calculation of Shear Strength at edge 1, $V_{r1} = 298161.965$

$V_{r1} = V_{Col}$ ((10.3), ASCE 41-17) = knl*VCol0

VCol0 = 298161.965

knl = 1 (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ '
where V_f is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)
 $f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

M/Vd = 2.00

$\mu_u = 5.9321525E-012$

$\nu_u = 7.4135260E-048$

d = 0.8*D = 320.00

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14: $V_s = 0.00$

$A_v = /2 * A_{stirrup} = 123370.055$

$f_y = 444.44$

s = 360.00

V_s is multiplied by Col = 0.00

s/d = 1.125

V_f ((11-3)-(11.4), ACI 440) = 194961.134

f = 0.95, for fully-wrapped sections

wf/sf = 1 (FRP strips adjacent to one another).

In (11.3) $\sin + \cos$ is replaced with $(\cot + \cot_a) \sin a$ which is more a generalised expression,
where a is the angle of the crack direction (see KANEPE).

This later relation, considered as a function $V_f(\theta, a)$, is implemented for every different fiber orientation a_i ,
as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.

orientation 1: $\theta = b1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, a1)|)$, with:

total thickness per orientation, $t_{f1} = NL * t / N_{oDir} = 1.016$

$d_{fv} = d$ (figure 11.2, ACI 440) = 370.00

f_{fe} ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$, from (11.6a), ACI 440

with $f_u = 0.01$

From (11-11), ACI 440: $V_s + V_f \leq 238930.50$

$b_w * d = *d * d / 4 = 80424.772$

Calculation of Shear Strength at edge 2, $V_{r2} = 298161.965$

$V_{r2} = V_{Col}$ ((10.3), ASCE 41-17) = knl*VCol0

VCol0 = 298161.965

knl = 1 (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ '
where V_f is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

M/Vd = 2.00

$\mu_u = 5.9321525E-012$

$V_u = 7.4135260E-048$
 $d = 0.8 \cdot D = 320.00$
 $Nu = 4771.233$
 $Ag = 125663.706$
 From (11.5.4.8), ACI 318-14: $V_s = 0.00$
 $A_v = \frac{1}{2} \cdot A_{stirrup} = 123370.055$
 $f_y = 444.44$
 $s = 360.00$
 V_s is multiplied by $Col = 0.00$
 $s/d = 1.125$
 V_f ((11-3)-(11.4), ACI 440) = 194961.134
 $f = 0.95$, for fully-wrapped sections
 $w_f/s_f = 1$ (FRP strips adjacent to one another).
 In (11.3) $\sin \theta + \cos \theta$ is replaced with $(\cot \theta + \cot \alpha) \sin \alpha$ which is more a generalised expression,
 where θ is the angle of the crack direction (see KANEPE).
 This later relation, considered as a function $V_f(\theta, \alpha)$, is implemented for every different cyclic fiber orientation α_i ,
 as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.
 orientation 1: $\theta = \theta_1 + 90^\circ = 90.00$
 $V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, \theta_1)|)$, with:
 total thickness per orientation, $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$
 $d_{fv} = d$ (figure 11.2, ACI 440) = 370.00
 f_{fe} ((11-5), ACI 440) = 259.312
 $E_f = 64828.00$
 $f_e = 0.004$, from (11.6a), ACI 440
 with $f_u = 0.01$
 From (11-11), ACI 440: $V_s + V_f \leq 238930.50$
 $b_w \cdot d = \frac{1}{4} \cdot d \cdot d = 80424.772$

 End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1
 At local axis: 2

 Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1
 At local axis: 2
 Integration Section: (b)
 Section Type: rccs

Constant Properties

 Knowledge Factor, $\gamma = 1.00$
 Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE 41-17.
 Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17
 Consequently:
 Existing material of Primary Member: Concrete Strength, $f_c = f_{cm} = 20.00$
 Existing material of Primary Member: Steel Strength, $f_s = f_{sm} = 444.44$
 Concrete Elasticity, $E_c = 21019.039$
 Steel Elasticity, $E_s = 200000.00$
 Diameter, $D = 400.00$
 Cover Thickness, $c = 25.00$
 Element Length, $L = 3000.00$
 Primary Member
 Ribbed Bars
 Ductile Steel
 Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)
 Longitudinal Bars With Ends Lapped Starting at the End Sections
 Lap Length $l_b = 300.00$
 FRP Wrapping Data
 Type: Carbon
 Cured laminate properties (design values)
 Thickness, $t = 1.016$
 Tensile Strength, $f_{fu} = 1055.00$
 Tensile Modulus, $E_f = 64828.00$
 Elongation, $e_{fu} = 0.01$
 Number of directions, $\text{NoDir} = 1$

Fiber orientations, b_i : 0.00°
Number of layers, NL = 1
Radius of rounding corners, R = 40.00

Stepwise Properties

Bending Moment, M = -1.1086227E-010
Shear Force, V2 = 4486.792
Shear Force, V3 = 1.7064698E-013
Axial Force, F = -4783.229
Longitudinal Reinforcement Area Distribution (in 2 divisions)
-Tension: As_t = 0.00
-Compression: As_c = 3053.628
Longitudinal Reinforcement Area Distribution (in 3 divisions)
-Tension: As_{t,ten} = 1017.876
-Compression: As_{c,com} = 1017.876
-Middle: As_{l,mid} = 1017.876
Mean Diameter of Tension Reinforcement, DbL = 18.00

Existing component: From table 7-7, ASCE 41_17: Final chord rotation Capacity $u_R = \phi u = 0.05131733$
 $u = \phi u + p = 0.05131733$

- Calculation of y -

 $y = (M_y * L_s / 3) / E_{eff} = 0.00931733$ ((4.29), Biskinis Phd)
 $M_y = 1.4766E+008$
 $L_s = M/V$ (with $L_s > 0.1 * L$ and $L_s < 2 * L$) = 1500.00
From table 10.5, ASCE 41_17: $E_{eff} = factor * E_c * I_g = 7.9240E+012$
factor = 0.30
Ag = 125663.706
fc' = 20.00
N = 4783.229
 $E_c * I_g = 2.6413E+013$

Calculation of Yielding Moment M_y

Calculation of y and M_y according to (7) - (8) in Biskinis and Fardis

 $M_y = \text{Min}(M_{y_ten}, M_{y_com}) = 1.4766E+008$
 $y = 8.1688588E-006$
 M_{y_ten} (8c) = 1.4766E+008
 y_{ten} (7c) = 72.4069
error of function (7c) = 0.00117279
 M_{y_com} (8d) = 3.7493E+008
 y_{com} (7d) = 69.91139
error of function (7d) = 0.00300901
with ((10.1), ASCE 41-17) $e_y = \text{Min}(e_y, 1.25 * e_y * (I_b / I_d)^{2/3}) = 0.0022222$
eco = 0.002
apl = 0.45 ((9c) in Biskinis and Fardis for FRP Wrap)
d1 = 44.00
R = 200.00
v = 0.00157746
N = 4783.229
Ac = 125663.706
((10.1), ASCE 41-17) = $\text{Min}(, 1.25 * (I_b / I_d)^{2/3}) = 0.44757577$
with fc' ((12.3), ACI 440) = 24.12975
fc = 20.00
fl = 1.3173
k = 1
Effective FRP thickness, tf = NL * t * Cos(b1) = 1.016

$\epsilon_{fe} ((12.5) \text{ and } (12.7)) = 0.004$
 $f_u = 0.01$
 $E_f = 64828.00$

Calculation of ratio l_b/l_d

Lap Length: $l_d/l_{d,min} = 0.33490748$

$l_b = 300.00$

$l_d = 895.7698$

Calculation of λ according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
 $\lambda = 1$

$d_b = 18.00$

Mean strength value of all re-bars: $f_y = 444.44$

$f'_c = 20.00$, but $f'_c^{0.5} \leq 8.3 \text{ MPa}$ (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} * \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

- Calculation of ρ -

From table 10-9: $\rho = 0.042$

with:

- Columns controlled by inadequate development or splicing along the clear height because $l_b/l_d < 1$

shear control ratio $V_y E / V_{col} O E = 0.30252729$

$d = 0.00$

$s = 0.00$

$t = 2 * A_v / (d_c * s) + 4 * t_f / D * (f_{fe} / f_s) = 0.00721126$

$A_v = 78.53982$, is the area of the circular stirrup

$d_c = D - 2 * \text{cover} - \text{Hoop Diameter} = 340.00$

The term $2 * t_f / b_w * (f_{fe} / f_s)$ is implemented to account for FRP contribution

where $f = 2 * t_f / b_w$ is FRP ratio (EC8 - 3, A.4.4.3(6)) and f_{fe} / f_s normalises f to steel strength

All these variables have already been given in Shear control ratio calculation.

$N_{UD} = 4783.229$

$A_g = 125663.706$

$f'_c E = 20.00$

$f_{yt} E = f_{yl} E = 444.44$

$\rho_l = \text{Area}_{Tot_Long_Rein} / (A_g) = 0.0243$

$f'_c E = 20.00$

End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (b)

Calculation No. 15

column C1, Floor 1

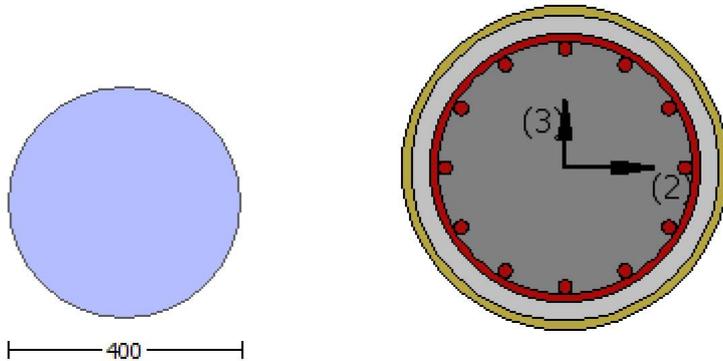
Limit State: Collapse Prevention (data interpolation between analysis steps 2 and 3)

Analysis: Uniform +X

Check: Shear capacity VRd

Edge: End

Local Axis: (3)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (b)

Section Type: rccs

Constant Properties

Knowledge Factor, $\gamma = 1.00$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE 41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength, $f_c = f_{c_lower_bound} = 16.00$

Existing material of Primary Member: Steel Strength, $f_s = f_{s_lower_bound} = 400.00$

Concrete Elasticity, $E_c = 21019.039$

Steel Elasticity, $E_s = 200000.00$

#####

Note: Especially for the calculation of γ for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE 41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE 41-17).

Existing material: Concrete Strength, $f_c = f_{cm} = 20.00$

Existing material: Steel Strength, $f_s = f_{sm} = 444.44$

#####

Diameter, $D = 400.00$

Cover Thickness, $c = 25.00$

Element Length, $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length $l_o = l_b = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness, $t = 1.016$

Tensile Strength, $f_{fu} = 1055.00$

Tensile Modulus, $E_f = 64828.00$

Elongation, $e_{fu} = 0.01$

Number of directions, $NoDir = 1$

Fiber orientations, θ_i : 0.00°
Number of layers, NL = 1
Radius of rounding corners, R = 40.00

Stepwise Properties

EDGE -A-
Bending Moment, M_a = 6.2300792E-010
Shear Force, V_a = -1.7064698E-013
EDGE -B-
Bending Moment, M_b = -1.1086227E-010
Shear Force, V_b = 1.7064698E-013
BOTH EDGES
Axial Force, F = -4783.229
Longitudinal Reinforcement Area Distribution (in 2 divisions)
-Tension: A_{st} = 0.00
-Compression: A_{sc} = 3053.628
Longitudinal Reinforcement Area Distribution (in 3 divisions)
-Tension: $A_{st,ten}$ = 1017.876
-Compression: $A_{sc,com}$ = 1017.876
-Middle: $A_{sc,mid}$ = 1017.876
Mean Diameter of Tension Reinforcement, $D_{bL,ten}$ = 18.00

Existing component: From table 7-7, ASCE 41_17: Final Shear Capacity $V_R = \phi V_n = 286339.856$
 V_n ((10.3), ASCE 41-17) = $k_n V_{CoI} = 286339.856$
 $V_{CoI} = 286339.856$
 $k_n = 1.00$
displacement_ductility_demand = 0.00

NOTE: In expression (10-3) ' $V_s = A_v f_y d/s$ ' is replaced by ' $V_s + \phi V_f$ '
where V_f is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)
 $f_c' = 16.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)
 $M/Vd = 2.00$
 $M_u = 1.1086227E-010$
 $V_u = 1.7064698E-013$
 $d = 0.8D = 320.00$
 $N_u = 4783.229$
 $A_g = 125663.706$
From (11.5.4.8), ACI 318-14: $V_s = 0.00$
 $A_v = \phi / 2 A_{stirrup} = 123370.055$
 $f_y = 400.00$
 $s = 360.00$
 V_s is multiplied by $\phi_{CoI} = 0.00$
 $s/d = 1.125$
 V_f ((11-3)-(11.4), ACI 440) = 194961.134
 $\phi = 0.95$, for fully-wrapped sections
 $w_f/s_f = 1$ (FRP strips adjacent to one another).
In (11.3) $\sin \theta + \cos \theta$ is replaced with $(\cot \theta + \csc \theta) \sin \alpha$ which is more a generalised expression,
where θ is the angle of the crack direction (see KANEPE).
This later relation, considered as a function $V_f(\theta, \alpha)$, is implemented for every different fiber orientation α_i ,
as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.
orientation 1: $\theta_1 = \theta + 90^\circ = 90.00$
 $V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, \theta_1)|)$, with:
total thickness per orientation, $t_{f1} = NL * t / \text{NoDir} = 1.016$
 $d_{fv} = d$ (figure 11.2, ACI 440) = 370.00
 f_{fe} ((11-5), ACI 440) = 259.312
 $E_f = 64828.00$
 $f_e = 0.004$, from (11.6a), ACI 440
with $f_u = 0.01$
From (11-11), ACI 440: $V_s + V_f \leq 213705.936$
 $b_w d = \phi d^2 / 4 = 80424.772$

displacement_ductility_demand is calculated as δ / y

- Calculation of δ / y for END B -
for rotation axis 2 and integ. section (b)

From analysis, chord rotation $\theta = 1.5066232E-020$
 $y = (M_y * L_s / 3) / E_{eff} = 0.00931733$ ((4.29), Biskinis Phd)
 $M_y = 1.4766E+008$
 $L_s = M/V$ (with $L_s > 0.1 * L$ and $L_s < 2 * L$) = 1500.00
From table 10.5, ASCE 41_17: $E_{eff} = factor * E_c * I_g = 7.9240E+012$
factor = 0.30
Ag = 125663.706
fc' = 20.00
N = 4783.229
 $E_c * I_g = 2.6413E+013$

Calculation of Yielding Moment M_y

Calculation of δ / y and M_y according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y_ten}, M_{y_com}) = 1.4766E+008$
 $y = 8.1688588E-006$
 M_{y_ten} (8c) = 1.4766E+008
 δ / y (7c) = 72.4069
error of function (7c) = 0.00117279
 M_{y_com} (8d) = 3.7493E+008
 δ / y (7d) = 69.91139
error of function (7d) = 0.00300901
with ((10.1), ASCE 41-17) $e_y = \text{Min}(e_y, 1.25 * e_y * (l_b / l_d)^{2/3}) = 0.0022222$
 $e_{co} = 0.002$
apl = 0.45 ((9c) in Biskinis and Fardis for FRP Wrap)
d1 = 44.00
R = 200.00
v = 0.00157746
N = 4783.229
Ac = 125663.706
((10.1), ASCE 41-17) $e_s = \text{Min}(e_s, 1.25 * e_s * (l_b / l_d)^{2/3}) = 0.44757577$
with fc' ((12.3), ACI 440) = 24.12975
fc = 20.00
fl = 1.3173
k = 1
Effective FRP thickness, tf = NL * t * Cos(b1) = 1.016
efe ((12.5) and (12.7)) = 0.004
fu = 0.01
Ef = 64828.00

Calculation of ratio l_b / l_d

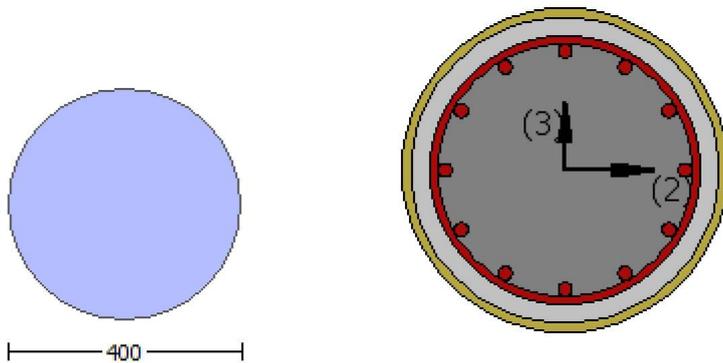
Lap Length: $l_d / l_{d,min} = 0.33490748$
 $l_b = 300.00$
 $l_d = 895.7698$
Calculation of I according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
= 1
db = 18.00
Mean strength value of all re-bars: $f_y = 444.44$
fc' = 20.00, but $fc'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)
t = 1.00
s = 0.80
e = 1.00
cb = 25.00

$K_{tr} = 1.14232$
 $A_{tr} = \sqrt{2} * \text{Area of stirrup} = 123.3701$
 $s = 360.00$
 $n = 12.00$

 End Of Calculation of Shear Capacity for element: column CC1 of floor 1
 At local axis: 3
 Integration Section: (b)

Calculation No. 16

column C1, Floor 1
 Limit State: Collapse Prevention (data interpolation between analysis steps 2 and 3)
 Analysis: Uniform +X
 Check: Chord rotation capacity (θ)
 Edge: End
 Local Axis: (3)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1
 At Shear local axis: 3
 (Bending local axis: 2)
 Section Type: rccs

Constant Properties

 Knowledge Factor, $\gamma = 1.00$
 Mean strength values are used for both shear and moment calculations.
 Consequently:
 Existing material of Primary Member: Concrete Strength, $f_c = f_{cm} = 20.00$
 Existing material of Primary Member: Steel Strength, $f_s = f_{sm} = 444.44$
 Concrete Elasticity, $E_c = 21019.039$
 Steel Elasticity, $E_s = 200000.00$

 Note: Especially for the calculation of moment strengths,
 the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14
 Existing material: Steel Strength, $f_s = 1.25 * f_{sm} = 555.55$
 #####

Diameter, D = 400.00
 Cover Thickness, c = 25.00
 Mean Confinement Factor overall section = 1.84055
 Element Length, L = 3000.00
 Primary Member
 Ribbed Bars
 Ductile Steel
 Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)
 Longitudinal Bars With Ends Lapped Starting at the End Sections
 Lap Length l_o = 300.00
 FRP Wrapping Data
 Type: Carbon
 Cured laminate properties (design values)
 Thickness, t = 1.016
 Tensile Strength, f_{fu} = 1055.00
 Tensile Modulus, E_f = 64828.00
 Elongation, e_{fu} = 0.01
 Number of directions, NoDir = 1
 Fiber orientations, b_i : 0.00°
 Number of layers, NL = 1
 Radius of rounding corners, R = 40.00

 Stepwise Properties

At local axis: 3
 EDGE -A-
 Shear Force, V_a = 1.2107607E-031
 EDGE -B-
 Shear Force, V_b = -1.2107607E-031
 BOTH EDGES
 Axial Force, F = -4771.233
 Longitudinal Reinforcement Area Distribution (in 2 divisions)
 -Tension: A_{sl} = 0.00
 -Compression: A_{slc} = 3053.628
 Longitudinal Reinforcement Area Distribution (in 3 divisions)
 -Tension: $A_{sl,ten}$ = 1017.876
 -Compression: $A_{sl,com}$ = 1017.876
 -Middle: $A_{sl,mid}$ = 1017.876

 Calculation of Shear Capacity ratio , V_e/V_r = 0.30252729
 Member Controlled by Flexure ($V_e/V_r < 1$)
 Calculation of Shear Demand from fig. R18.6.5, ACI 318-14 $V_e = (M_{pr1} + M_{pr2})/l_n = 90202.132$
 with
 $M_{pr1} = \text{Max}(Mu_{1+} , Mu_{1-}) = 1.3530E+008$
 $Mu_{1+} = 1.3530E+008$, is the ultimate moment strength at the edge 1 of the member in the actual moment direction
 which is defined for the static loading combination
 $Mu_{1-} = 1.3530E+008$, is the ultimate moment strength at the edge 1 of the member in the opposite moment
 direction which is defined for the static loading combination
 $M_{pr2} = \text{Max}(Mu_{2+} , Mu_{2-}) = 1.3530E+008$
 $Mu_{2+} = 1.3530E+008$, is the ultimate moment strength at the edge 2 of the member in the actual moment direction
 which is defined for the the static loading combination
 $Mu_{2-} = 1.3530E+008$, is the ultimate moment strength at the edge 2 of the member in the opposite moment
 direction which is defined for the the static loading combination

 Calculation of Mu_{1+}

 Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu
 $Mu = 1.3530E+008$

 = 0.90757121

$\lambda = 0.80580716$
 error of function (3.68), Biskinis Phd = 28928.286
 From 5A.2, TBDY: $f_{cc} = f_c \cdot \lambda = 36.81095$
 conf. factor $\lambda = 1.84055$
 $f_c = 20.00$
 From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$
 $l_b/l_d = 0.26792599$
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00155946$
 $N = 4771.233$
 $A_c = 125663.706$
 $= \lambda \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$
 $l_b = 300.00$
 $l_d = 1119.712$
 Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
 $= 1$
 $d_b = 18.00$
 Mean strength value of all re-bars: $f_y = 555.55$
 $f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)
 $t = 1.00$
 $s = 0.80$
 $e = 1.00$
 $c_b = 25.00$
 $K_{tr} = 1.14232$
 $A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$
 $s = 360.00$
 $n = 12.00$

Calculation of μ_1

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ
 $\mu = 1.3530E+008$

$\lambda = 0.90757121$
 $\lambda = 0.80580716$
 error of function (3.68), Biskinis Phd = 28928.286
 From 5A.2, TBDY: $f_{cc} = f_c \cdot \lambda = 36.81095$
 conf. factor $\lambda = 1.84055$
 $f_c = 20.00$
 From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$
 $l_b/l_d = 0.26792599$
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00155946$
 $N = 4771.233$
 $A_c = 125663.706$
 $= \lambda \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$
 $l_b = 300.00$
 $l_d = 1119.712$
 Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
 $= 1$
 $d_b = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$
 $f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)
 $t = 1.00$
 $s = 0.80$
 $e = 1.00$
 $cb = 25.00$
 $K_{tr} = 1.14232$
 $A_{tr} = \frac{1}{2} * \text{Area of stirrup} = 123.3701$
 $s = 360.00$
 $n = 12.00$

Calculation of μ_{2+}

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ
 $\mu = 1.3530E+008$

 $\phi = 0.90757121$
 $\lambda = 0.80580716$
error of function (3.68), Biskinis Phd = 28928.286
From 5A.2, TBDY: $f_{cc} = f_c' * c = 36.81095$
conf. factor $c = 1.84055$
 $f_c = 20.00$
From 10.3.5, ASCE 41-17, Final value of f_y : $f_y * \text{Min}(1, 1.25 * (l_b/d)^{2/3}) = 288.6089$
 $l_b/d = 0.26792599$
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00155946$
 $N = 4771.233$
 $A_c = 125663.706$
 $\phi * \text{Min}(1, 1.25 * (l_b/d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/d

Lap Length: $l_b/d = 0.26792599$
 $l_b = 300.00$
 $l_d = 1119.712$
Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
 $\phi = 1$

$db = 18.00$
Mean strength value of all re-bars: $f_y = 555.55$
 $f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)
 $t = 1.00$
 $s = 0.80$
 $e = 1.00$
 $cb = 25.00$
 $K_{tr} = 1.14232$
 $A_{tr} = \frac{1}{2} * \text{Area of stirrup} = 123.3701$
 $s = 360.00$
 $n = 12.00$

Calculation of μ_{2-}

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ
 $\mu = 1.3530E+008$

 $\phi = 0.90757121$
 $\lambda = 0.80580716$
error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$

conf. factor $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$d_b = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of Shear Strength $V_r = \text{Min}(V_{r1}, V_{r2}) = 298161.965$

Calculation of Shear Strength at edge 1, $V_{r1} = 298161.965$

$V_{r1} = V_{Col}$ ((10.3), ASCE 41-17) = $k_{nl} \cdot V_{Col0}$

$V_{Col0} = 298161.965$

$k_{nl} = 1$ (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where V_f is the contribution of FRPs (11.3), ACI 440).

$= 1$ (normal-weight concrete)

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.2899604E-011$

$\nu_u = 1.2107607E-031$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14: $V_s = 0.00$

$A_v = \frac{1}{2} \cdot A_{stirrup} = 123370.055$

$f_y = 444.44$

$s = 360.00$

V_s is multiplied by $\text{Col} = 0.00$

$s/d = 1.125$

V_f ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$, for fully-wrapped sections

$w_f/s_f = 1$ (FRP strips adjacent to one another).

In (11.3) $\sin \theta + \cos \theta$ is replaced with $(\cot \theta + \csc \theta) \sin \alpha$ which is more a generalised expression, where θ is the angle of the crack direction (see KANEPE).

This later relation, considered as a function $V_f(\theta, \alpha)$, is implemented for every different fiber orientation α_i , as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.

orientation 1: $\theta = 45^\circ + 90^\circ = 135^\circ$

$V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, a1)|)$, with:
 total thickness per orientation, $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$
 $d_{fv} = d$ (figure 11.2, ACI 440) = 370.00
 f_{fe} ((11-5), ACI 440) = 259.312
 $E_f = 64828.00$
 $f_e = 0.004$, from (11.6a), ACI 440
 with $f_u = 0.01$
 From (11-11), ACI 440: $V_s + V_f \leq 238930.50$
 $b_w \cdot d = \sqrt[4]{d \cdot d} = 80424.772$

Calculation of Shear Strength at edge 2, $V_{r2} = 298161.965$
 $V_{r2} = V_{Col}$ ((10.3), ASCE 41-17) = $k_{nl} \cdot V_{Col0}$
 $V_{Col0} = 298161.965$
 $k_{nl} = 1$ (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ '
 where V_f is the contribution of FRPs (11.3), ACI 440).

$\rho = 1$ (normal-weight concrete)
 $f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)
 $M / Vd = 2.00$
 $\mu_u = 1.2899604E-011$
 $\nu_u = 1.2107607E-031$
 $d = 0.8 \cdot D = 320.00$
 $N_u = 4771.233$
 $A_g = 125663.706$
 From (11.5.4.8), ACI 318-14: $V_s = 0.00$
 $A_v = \sqrt{2} \cdot A_{stirrup} = 123370.055$
 $f_y = 444.44$
 $s = 360.00$

V_s is multiplied by $\rho_{col} = 0.00$
 $s/d = 1.125$

V_f ((11-3)-(11.4), ACI 440) = 194961.134
 $f = 0.95$, for fully-wrapped sections

$w_f / s_f = 1$ (FRP strips adjacent to one another).

In (11.3) $\sin + \cos$ is replaced with $(\cot + \cot a) \sin a$ which is more a generalised expression,
 where a is the angle of the crack direction (see KANEPE).

This later relation, considered as a function $V_f(\theta, \rho)$, is implemented for every different fiber orientation a_i ,
 as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.

orientation 1: $\theta = b1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, a1)|)$, with:
 total thickness per orientation, $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$
 $d_{fv} = d$ (figure 11.2, ACI 440) = 370.00
 f_{fe} ((11-5), ACI 440) = 259.312
 $E_f = 64828.00$
 $f_e = 0.004$, from (11.6a), ACI 440
 with $f_u = 0.01$
 From (11-11), ACI 440: $V_s + V_f \leq 238930.50$
 $b_w \cdot d = \sqrt[4]{d \cdot d} = 80424.772$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1
 At local axis: 3

Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1
 At Shear local axis: 2
 (Bending local axis: 3)
 Section Type: rccs

Constant Properties

Knowledge Factor, $\gamma = 1.00$
 Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength, $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength, $f_s = f_{sm} = 444.44$

Concrete Elasticity, $E_c = 21019.039$

Steel Elasticity, $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength, $f_s = 1.25 \cdot f_{sm} = 555.55$

#####

Diameter, $D = 400.00$

Cover Thickness, $c = 25.00$

Mean Confinement Factor overall section = 1.84055

Element Length, $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length $l_o = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness, $t = 1.016$

Tensile Strength, $f_{fu} = 1055.00$

Tensile Modulus, $E_f = 64828.00$

Elongation, $e_{fu} = 0.01$

Number of directions, $N_{oDir} = 1$

Fiber orientations, $b_i = 0.00^\circ$

Number of layers, $N_L = 1$

Radius of rounding corners, $R = 40.00$

Stepwise Properties

At local axis: 2

EDGE -A-

Shear Force, $V_a = -7.4135260E-048$

EDGE -B-

Shear Force, $V_b = 7.4135260E-048$

BOTH EDGES

Axial Force, $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension: $A_{sl,t} = 0.00$

-Compression: $A_{sl,c} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension: $A_{sl,ten} = 1017.876$

-Compression: $A_{sl,com} = 1017.876$

-Middle: $A_{sl,mid} = 1017.876$

Calculation of Shear Capacity ratio, $V_e/V_r = 0.30252729$

Member Controlled by Flexure ($V_e/V_r < 1$)

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14 $V_e = (M_{pr1} + M_{pr2})/l_n = 90202.132$

with

$M_{pr1} = \text{Max}(M_{u1+}, M_{u1-}) = 1.3530E+008$

$M_{u1+} = 1.3530E+008$, is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$M_{u1-} = 1.3530E+008$, is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(M_{u2+}, M_{u2-}) = 1.3530E+008$

$M_{u2+} = 1.3530E+008$, is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination

$M_{u2-} = 1.3530E+008$, is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

Calculation of Mu1+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu
Mu = 1.3530E+008

= 0.90757121
' = 0.80580716
error of function (3.68), Biskinis Phd = 28928.286
From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$
conf. factor $c = 1.84055$
 $f_c = 20.00$
From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$
 $l_b/l_d = 0.26792599$
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00155946$
 $N = 4771.233$
 $A_c = 125663.706$
= $\cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$
 $l_b = 300.00$
 $l_d = 1119.712$
Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
= 1
 $d_b = 18.00$
Mean strength value of all re-bars: $f_y = 555.55$
 $f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3 \text{ MPa}$ (22.5.3.1, ACI 318-14)
 $t = 1.00$
 $s = 0.80$
 $e = 1.00$
 $cb = 25.00$
 $K_{tr} = 1.14232$
 $A_{tr} = \sqrt{2} \cdot \text{Area of stirrup} = 123.3701$
 $s = 360.00$
 $n = 12.00$

Calculation of Mu1-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu
Mu = 1.3530E+008

= 0.90757121
' = 0.80580716
error of function (3.68), Biskinis Phd = 28928.286
From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$
conf. factor $c = 1.84055$
 $f_c = 20.00$
From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$
 $l_b/l_d = 0.26792599$
 $d_1 = 44.00$
 $R = 200.00$
 $v = 0.00155946$
 $N = 4771.233$
 $A_c = 125663.706$
= $\cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of l_b, \min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$d_b = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f'_c = 20.00$, but $f_c^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of μ_{2+}

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ

$\mu = 1.3530E+008$

$= 0.90757121$

$' = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY: $f_{cc} = f'_c \cdot c = 36.81095$

conf. factor $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of l_b, \min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$d_b = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f'_c = 20.00$, but $f_c^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of μ_2 -

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), μ

$\mu = 1.3530E+008$

$\mu = 0.90757121$

$\mu = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY: $f_{cc} = f_c \cdot c = 36.81095$

conf. factor $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of f_y : $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio l_b/l_d

Lap Length: $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of l_b , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$d_b = 18.00$

Mean strength value of all re-bars: $f_y = 555.55$

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of Shear Strength $V_r = \text{Min}(V_{r1}, V_{r2}) = 298161.965$

Calculation of Shear Strength at edge 1, $V_{r1} = 298161.965$

$V_{r1} = V_{Co1}$ ((10.3), ASCE 41-17) = $k_{nl} \cdot V_{Co10}$

$V_{Co10} = 298161.965$

$k_{nl} = 1$ (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where V_f is the contribution of FRPs (11.3), ACI 440).

$= 1$ (normal-weight concrete)

$f_c' = 20.00$, but $f_c'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu = 5.9321525E-012$

$V_u = 7.4135260E-048$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14: $V_s = 0.00$

$A_v = \frac{1}{2} \cdot A_{\text{stirrup}} = 123370.055$

$f_y = 444.44$

$s = 360.00$

Vs is multiplied by Col = 0.00

$$s/d = 1.125$$

$$V_f \text{ ((11-3)-(11.4), ACI 440) } = 194961.134$$

f = 0.95, for fully-wrapped sections

wf/sf = 1 (FRP strips adjacent to one another).

In (11.3) $\sin \theta + \cos \theta$ is replaced with $(\cot \theta + \csc \theta) \sin \alpha$ which is more a generalised expression, where θ is the angle of the crack direction (see KANEPE).

This later relation, considered as a function $V_f(\theta, \alpha)$, is implemented for every different fiber orientation α_i , as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.

$$\text{orientation 1: } \theta = b1 + 90^\circ = 90.00$$

$$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \theta)|), \text{ with:}$$

$$\text{total thickness per orientation, } t_{f1} = NL * t / \text{NoDir} = 1.016$$

$$d_{fv} = d \text{ (figure 11.2, ACI 440) } = 370.00$$

$$f_{fe} \text{ ((11-5), ACI 440) } = 259.312$$

$$E_f = 64828.00$$

$$f_e = 0.004, \text{ from (11.6a), ACI 440}$$

$$\text{with } f_u = 0.01$$

$$\text{From (11-11), ACI 440: } V_s + V_f \leq 238930.50$$

$$b_w * d = \rho * d^2 / 4 = 80424.772$$

Calculation of Shear Strength at edge 2, $V_{r2} = 298161.965$

$$V_{r2} = V_{Col} \text{ ((10.3), ASCE 41-17) } = k_{nl} * V_{Col0}$$

$$V_{Col0} = 298161.965$$

$$k_{nl} = 1 \text{ (zero step-static loading)}$$

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ ' where V_f is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

$$f_c' = 20.00, \text{ but } f_c'^{0.5} \leq 8.3 \text{ MPa (22.5.3.1, ACI 318-14)}$$

$$M / Vd = 2.00$$

$$\mu_u = 5.9321525E-012$$

$$V_u = 7.4135260E-048$$

$$d = 0.8 * D = 320.00$$

$$N_u = 4771.233$$

$$A_g = 125663.706$$

$$\text{From (11.5.4.8), ACI 318-14: } V_s = 0.00$$

$$A_v = \rho / 2 * A_{stirup} = 123370.055$$

$$f_y = 444.44$$

$$s = 360.00$$

Vs is multiplied by Col = 0.00

$$s/d = 1.125$$

$$V_f \text{ ((11-3)-(11.4), ACI 440) } = 194961.134$$

f = 0.95, for fully-wrapped sections

wf/sf = 1 (FRP strips adjacent to one another).

In (11.3) $\sin \theta + \cos \theta$ is replaced with $(\cot \theta + \csc \theta) \sin \alpha$ which is more a generalised expression, where θ is the angle of the crack direction (see KANEPE).

This later relation, considered as a function $V_f(\theta, \alpha)$, is implemented for every different fiber orientation α_i , as well as for 2 crack directions, $\theta = 45^\circ$ and $\theta = -45^\circ$ to take into consideration the cyclic seismic loading.

$$\text{orientation 1: } \theta = b1 + 90^\circ = 90.00$$

$$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \theta)|), \text{ with:}$$

$$\text{total thickness per orientation, } t_{f1} = NL * t / \text{NoDir} = 1.016$$

$$d_{fv} = d \text{ (figure 11.2, ACI 440) } = 370.00$$

$$f_{fe} \text{ ((11-5), ACI 440) } = 259.312$$

$$E_f = 64828.00$$

$$f_e = 0.004, \text{ from (11.6a), ACI 440}$$

$$\text{with } f_u = 0.01$$

$$\text{From (11-11), ACI 440: } V_s + V_f \leq 238930.50$$

$$b_w * d = \rho * d^2 / 4 = 80424.772$$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At local axis: 2

Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (b)

Section Type: rccs

Constant Properties

Knowledge Factor, $k = 1.00$

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE 41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength, $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength, $f_s = f_{sm} = 444.44$

Concrete Elasticity, $E_c = 21019.039$

Steel Elasticity, $E_s = 200000.00$

Diameter, $D = 400.00$

Cover Thickness, $c = 25.00$

Element Length, $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length $l_b = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness, $t = 1.016$

Tensile Strength, $f_{fu} = 1055.00$

Tensile Modulus, $E_f = 64828.00$

Elongation, $e_{fu} = 0.01$

Number of directions, $N_{oDir} = 1$

Fiber orientations, $b_i = 0.00^\circ$

Number of layers, $N_L = 1$

Radius of rounding corners, $R = 40.00$

Stepwise Properties

Bending Moment, $M = 2684.487$

Shear Force, $V_2 = 4486.792$

Shear Force, $V_3 = 1.7064698E-013$

Axial Force, $F = -4783.229$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension: $A_{st} = 0.00$

-Compression: $A_{sc} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension: $A_{st,ten} = 1017.876$

-Compression: $A_{st,com} = 1017.876$

-Middle: $A_{st,mid} = 1017.876$

Mean Diameter of Tension Reinforcement, $D_bL = 18.00$

Existing component: From table 7-7, ASCE 41_17: Final chord rotation Capacity $u_{R} = \gamma + p = 0.04386347$

$u = \gamma + p = 0.04386347$

- Calculation of γ -

$\gamma = (M \gamma L_s / 3) / E_{eff} = 0.00186347$ ((4.29), Biskinis Phd))

$M \gamma = 1.4766E+008$

$L_s = M / V$ (with $L_s > 0.1 * L$ and $L_s < 2 * L$) = 300.00

From table 10.5, ASCE 41_17: $E_{eff} = factor * E_c * I_g = 7.9240E+012$

factor = 0.30

Ag = 125663.706
fc' = 20.00
N = 4783.229
Ec*Ig = 2.6413E+013

Calculation of Yielding Moment My

Calculation of ϕ_y and My according to (7) - (8) in Biskinis and Fardis

My = Min(My_ten, My_com) = 1.4766E+008
y = 8.1688588E-006
My_ten (8c) = 1.4766E+008
 ten (7c) = 72.4069
error of function (7c) = 0.00117279
My_com (8d) = 3.7493E+008
 com (7d) = 69.91139
error of function (7d) = 0.00300901
with ((10.1), ASCE 41-17) $\phi_y = \text{Min}(\phi_y, 1.25 * \phi_y * (I_b/I_d)^{2/3}) = 0.0022222$
 eco = 0.002
 apl = 0.45 ((9c) in Biskinis and Fardis for FRP Wrap)
 d1 = 44.00
 R = 200.00
 v = 0.00157746
 N = 4783.229
 Ac = 125663.706
 ((10.1), ASCE 41-17) $\phi_y = \text{Min}(\phi_y, 1.25 * \phi_y * (I_b/I_d)^{2/3}) = 0.44757577$
with fc' ((12.3), ACI 440) = 24.12975
 fc = 20.00
 fl = 1.3173
 k = 1
 Effective FRP thickness, tf = NL*t*cos(b1) = 1.016
 efe ((12.5) and (12.7)) = 0.004
 fu = 0.01
 Ef = 64828.00

Calculation of ratio lb/l_d

Lap Length: l_d/l_{d,min} = 0.33490748
lb = 300.00
ld = 895.7698
Calculation of l according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
 = 1
 db = 18.00
Mean strength value of all re-bars: fy = 444.44
fc' = 20.00, but $fc'^{0.5} \leq 8.3$ MPa (22.5.3.1, ACI 318-14)
 t = 1.00
 s = 0.80
 e = 1.00
 cb = 25.00
 Ktr = 1.14232
 Atr = $\sqrt{2} * \text{Area of stirrup} = 123.3701$
 s = 360.00
 n = 12.00

- Calculation of ϕ_p -

From table 10-9: $\phi_p = 0.042$

with:

- Columns controlled by inadequate development or splicing along the clear height because $l_b/l_d < 1$
shear control ratio $V_y E / V_{CoI} E = 0.30252729$
d = 0.00
s = 0.00

$$t = 2 \cdot A_v / (d_c \cdot s) + 4 \cdot t_f / D \cdot (f_{fe} / f_s) = 0.00721126$$

$A_v = 78.53982$, is the area of the circular stirrup

$d_c = D - 2 \cdot \text{cover}$ - Hoop Diameter = 340.00

The term $2 \cdot t_f / bw \cdot (f_{fe} / f_s)$ is implemented to account for FRP contribution

where $f = 2 \cdot t_f / bw$ is FRP ratio (EC8 - 3, A.4.4.3(6)) and f_{fe} / f_s normalises f to steel strength

All these variables have already been given in Shear control ratio calculation.

$$N_{UD} = 4783.229$$

$$A_g = 125663.706$$

$$f_{cE} = 20.00$$

$$f_{yE} = f_{yI} = 444.44$$

$$p_l = \text{Area_Tot_Long_Rein} / (A_g) = 0.0243$$

$$f_{cE} = 20.00$$

End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (b)
