

# Detailed Member Calculations

**Units: N&mm**

**Regulation: ASCE 41-17**

## Calculation No. 1

column C1, Floor 1

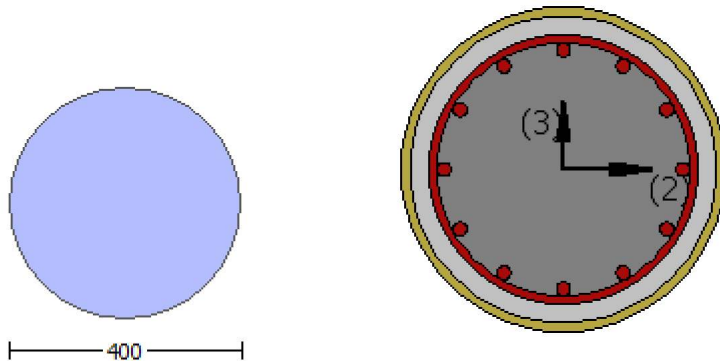
Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Shear capacity VRd

Edge: Start

Local Axis: (2)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (a)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE 41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 16.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 400.00$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$   
 #####  
 Note: Especially for the calculation of  $\gamma$  for displacement ductility demand,  
 the expected (mean value) strengths are used (7.5.1.3, ASCE 41-17) because bending is considered as  
 Deformation-Controlled Action (Table C7-1, ASCE 41-17).  
 Existing material: Concrete Strength,  $f_c = f_{cm} = 20.00$   
 Existing material: Steel Strength,  $f_s = f_{sm} = 444.44$   
 #####  
 Diameter,  $D = 400.00$   
 Cover Thickness,  $c = 25.00$   
 Element Length,  $L = 3000.00$   
 Primary Member  
 Ribbed Bars  
 Ductile Steel  
 Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Lap Length  $l_o = l_b = 300.00$   
 FRP Wrapping Data  
 Type: Carbon  
 Cured laminate properties (design values)  
 Thickness,  $t = 1.016$   
 Tensile Strength,  $f_{fu} = 1055.00$   
 Tensile Modulus,  $E_f = 64828.00$   
 Elongation,  $e_{fu} = 0.01$   
 Number of directions,  $NoDir = 1$   
 Fiber orientations,  $bi: 0.00^\circ$   
 Number of layers,  $NL = 1$   
 Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

EDGE -A-  
 Bending Moment,  $M_a = -1.0423E+007$   
 Shear Force,  $V_a = -3472.528$   
 EDGE -B-  
 Bending Moment,  $M_b = 0.09265189$   
 Shear Force,  $V_b = 3472.528$   
 BOTH EDGES  
 Axial Force,  $F = -4769.844$   
 Longitudinal Reinforcement Area Distribution (in 2 divisions)  
 -Tension:  $As_t = 1272.345$   
 -Compression:  $As_c = 1781.283$   
 Longitudinal Reinforcement Area Distribution (in 3 divisions)  
 -Tension:  $As_{l,ten} = 1017.876$   
 -Compression:  $As_{l,com} = 1017.876$   
 -Middle:  $As_{l,mid} = 1017.876$   
 Mean Diameter of Tension Reinforcement,  $Db_{L,ten} = 18.00$

Existing component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = V_n = 235775.141$   
 $V_n ((10.3), ASCE 41-17) = knl * V_{CoIO} = 235775.141$   
 $V_{CoI} = 235775.141$   
 $knl = 1.00$   
 $displacement\_ductility\_demand = 0.01890466$

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_{s+} = f^* V_f$ '  
 where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\gamma = 1$  (normal-weight concrete)  
 $f'_c = 16.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 4.00$   
 $M_u = 1.0423E+007$   
 $V_u = 3472.528$   
 $d = 0.8 * D = 320.00$

$Nu = 4769.844$   
 $Ag = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $Vs = 0.00$   
 $Av = \frac{1}{2} * A_{stirrup} = 123370.055$   
 $fy = 400.00$   
 $s = 360.00$   
 $Vs$  is multiplied by  $Col = 0.00$   
 $s/d = 1.125$   
 $Vf$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections  
 $wf/sf = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $Vf(\theta)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = b1 + 90^\circ = 90.00$   
 $Vf = \text{Min}(|Vf(45, \theta)|, |Vf(-45, a1)|)$ , with:  
 total thickness per orientation,  $tf1 = NL * t / NoDir = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $ffe$  ((11-5), ACI 440) = 259.312  
 $Ef = 64828.00$   
 $fe = 0.004$ , from (11.6a), ACI 440  
 with  $fu = 0.01$   
 From (11-11), ACI 440:  $Vs + Vf \leq 213705.936$   
 $bw * d = \frac{1}{4} * d * d = 80424.772$

displacement\_ductility\_demand is calculated as  $\delta / y$

- Calculation of  $\delta / y$  for END A -  
 for rotation axis 3 and integ. section (a)

From analysis, chord rotation  $\theta = 0.00035245$   
 $y = (My * Ls / 3) / Eleff = 0.01864342$  ((4.29), Biskinis Phd))  
 $My = 1.4766E+008$   
 $Ls = M/V$  (with  $Ls > 0.1 * L$  and  $Ls < 2 * L$ ) = 3001.447  
 From table 10.5, ASCE 41\_17:  $Eleff = factor * Ec * Ig = 7.9240E+012$   
 $factor = 0.30$   
 $Ag = 125663.706$   
 $fc' = 20.00$   
 $N = 4769.844$   
 $Ec * Ig = 2.6413E+013$

Calculation of Yielding Moment  $My$

Calculation of  $\delta$  and  $My$  according to (7) - (8) in Biskinis and Fardis

$My = \text{Min}(My_{ten}, My_{com}) = 1.4766E+008$   
 $y = 8.1688007E-006$   
 $My_{ten}$  (8c) = 1.4766E+008  
 $\delta_{ten}$  (7c) = 72.40642  
 error of function (7c) = 0.00117228  
 $My_{com}$  (8d) = 3.7493E+008  
 $\delta_{com}$  (7d) = 69.91126  
 error of function (7d) = 0.00301342  
 with ((10.1), ASCE 41-17)  $ey = \text{Min}(ey, 1.25 * ey * (lb/l_d)^{2/3}) = 0.0022222$   
 $eco = 0.002$   
 $apl = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d1 = 44.00$   
 $R = 200.00$   
 $v = 0.00157305$   
 $N = 4769.844$   
 $Ac = 125663.706$   
 ((10.1), ASCE 41-17)  $\delta = \text{Min}(\delta, 1.25 * \delta * (lb/l_d)^{2/3}) = 0.44757577$

with  $f_c^*$  ((12.3), ACI 440) = 24.12975  
 $f_c = 20.00$   
 $f_l = 1.3173$   
 $k = 1$   
Effective FRP thickness,  $t_f = NL * t * \cos(b_1) = 1.016$   
 $e_{fe}$  ((12.5) and (12.7)) = 0.004  
 $f_u = 0.01$   
 $E_f = 64828.00$

Calculation of ratio  $I_b/I_d$

Lap Length:  $I_d/I_{d,min} = 0.33490748$

$I_b = 300.00$

$I_d = 895.7698$

Calculation of  $I$  according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $= 1$

$db = 18.00$

Mean strength value of all re-bars:  $f_y = 444.44$

$f_c' = 20.00$ , but  $f_c^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$cb = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} * \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

End Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (a)

## Calculation No. 2

column C1, Floor 1

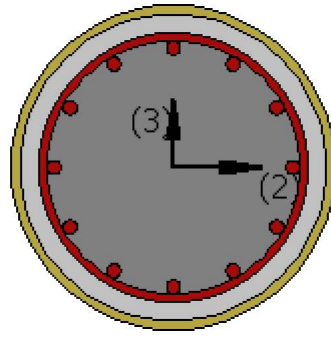
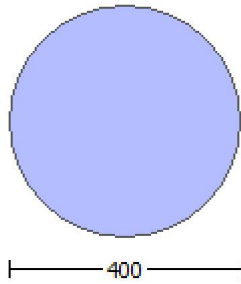
Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity (  $\phi$  )

Edge: Start

Local Axis: (2)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.44$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

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Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 555.55$

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Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.84055

Element Length,  $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length  $l_o = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $\epsilon_{fu} = 0.01$

Number of directions,  $NoDir = 1$

Fiber orientations,  $b_i: 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

Stepwise Properties

At local axis: 3

EDGE -A-

Shear Force,  $V_a = 1.2107607E-031$

EDGE -B-

Shear Force,  $V_b = -1.2107607E-031$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $Asl_t = 0.00$   
 -Compression:  $Asl_c = 3053.628$   
 Longitudinal Reinforcement Area Distribution (in 3 divisions)  
 -Tension:  $Asl_{ten} = 1017.876$   
 -Compression:  $Asl_{com} = 1017.876$   
 -Middle:  $Asl_{mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.30252729$   
 Member Controlled by Flexure ( $V_e/V_r < 1$ )  
 Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 90202.132$   
 with  
 $M_{pr1} = \max(\mu_{u1+}, \mu_{u1-}) = 1.3530E+008$   
 $\mu_{u1+} = 1.3530E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination  
 $\mu_{u1-} = 1.3530E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination  
 $M_{pr2} = \max(\mu_{u2+}, \mu_{u2-}) = 1.3530E+008$   
 $\mu_{u2+} = 1.3530E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination  
 $\mu_{u2-} = 1.3530E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

Calculation of  $\mu_{u1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$   
 $\mu_u = 1.3530E+008$

$\phi = 0.90757121$   
 $\phi' = 0.80580716$   
 error of function (3.68), Biskinis Phd = 28928.286  
 From 5A.2, TBDY:  $f_{cc} = f_c' \cdot c = 36.81095$   
 conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
 From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \min(1, 1.25 \cdot (l_b/d)^{2/3}) = 288.6089$   
 $l_b/d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \phi' \cdot \min(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.28805051$

Calculation of ratio  $l_b/d$

Lap Length:  $l_b/d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
 Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $\phi = 1$   
 $d_b = 18.00$   
 Mean strength value of all re-bars:  $f_y = 555.55$   
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \pi/2 \cdot \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

Calculation of Mu1-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.3530E+008

= 0.90757121  
' = 0.80580716  
error of function (3.68), Biskinis Phd = 28928.286  
From 5A.2, TDY:  $f_{cc} = f_c \cdot c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
=  $\cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio  $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
= 1  
 $db = 18.00$   
Mean strength value of all re-bars:  $f_y = 555.55$   
 $f_c' = 20.00$ , but  $f_c^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

Calculation of Mu2+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.3530E+008

= 0.90757121  
' = 0.80580716  
error of function (3.68), Biskinis Phd = 28928.286  
From 5A.2, TDY:  $f_{cc} = f_c \cdot c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.28805051$$

Calculation of ratio  $l_b/d$

Lap Length:  $l_b/d = 0.26792599$

$l_b = 300.00$

$d = 1119.712$

Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$d_b = 18.00$

Mean strength value of all re-bars:  $f_y = 555.55$

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of  $\mu_2$ -

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$\mu = 1.3530 \times 10^8$

$= 0.90757121$

$' = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY:  $f_{cc} = f'_c \cdot c = 36.81095$

conf. factor  $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 288.6089$

$l_b/d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.28805051$

Calculation of ratio  $l_b/d$

Lap Length:  $l_b/d = 0.26792599$

$l_b = 300.00$

$d = 1119.712$

Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$d_b = 18.00$

Mean strength value of all re-bars:  $f_y = 555.55$

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$



Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 298161.965$

Calculation of Shear Strength at edge 1,  $V_{r1} = 298161.965$

$V_{r1} = V_{Col} \text{ ((10.3), ASCE 41-17)} = k_{nl} * V_{Col0}$

$V_{Col0} = 298161.965$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\lambda = 1$  (normal-weight concrete)

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.2899604E-011$

$\nu_u = 1.2107607E-031$

$d = 0.8 * D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 0.00$

$A_v = \sqrt{2} * A_{\text{stirrup}} = 123370.055$

$f_y = 444.44$

$s = 360.00$

$V_s$  is multiplied by  $\text{Col} = 0.00$

$s/d = 1.125$

$V_f \text{ ((11-3)-(11.4), ACI 440)} = 194961.134$

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression, where  $a$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta)$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta_1 = b_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, a_1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L * t / N_{\text{Dir}} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe} \text{ ((11-5), ACI 440)} = 259.312$

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$

$b_w * d = \sqrt{2} * d^2 / 4 = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 298161.965$

$V_{r2} = V_{Col} \text{ ((10.3), ASCE 41-17)} = k_{nl} * V_{Col0}$

$V_{Col0} = 298161.965$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\lambda = 1$  (normal-weight concrete)

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.2899604E-011$

$\nu_u = 1.2107607E-031$

$d = 0.8 * D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 0.00$

$A_v = \sqrt{2} * A_{\text{stirrup}} = 123370.055$

$f_y = 444.44$

$s = 360.00$

$V_s$  is multiplied by  $\text{Col} = 0.00$

$s/d = 1.125$

$V_f((11-3)-(11.4), \text{ACI } 440) = 194961.134$   
 $f = 0.95$ , for fully-wrapped sections  
 $wf/sf = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a)\sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f( , )$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $a = 45^\circ$  and  $a = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $b_1 = b_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, b_1)|, |V_f(-45, a_1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = NL * t / \text{NoDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}((11-5), \text{ACI } 440) = 259.312$   
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $b_w * d = \frac{V_s * d}{4} = 80424.772$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At local axis: 3

Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At Shear local axis: 2  
 (Bending local axis: 3)  
 Section Type: rccs

#### Constant Properties

Knowledge Factor,  $\phi = 1.00$   
 Mean strength values are used for both shear and moment calculations.  
 Consequently:  
 Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$   
 Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.44$   
 Concrete Elasticity,  $E_c = 21019.039$   
 Steel Elasticity,  $E_s = 200000.00$   
 #####  
 Note: Especially for the calculation of moment strengths,  
 the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14  
 Existing material: Steel Strength,  $f_s = 1.25 * f_{sm} = 555.55$   
 #####  
 Diameter,  $D = 400.00$   
 Cover Thickness,  $c = 25.00$   
 Mean Confinement Factor overall section = 1.84055  
 Element Length,  $L = 3000.00$   
 Primary Member  
 Ribbed Bars  
 Ductile Steel  
 Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Lap Length  $l_o = 300.00$   
 FRP Wrapping Data  
 Type: Carbon  
 Cured laminate properties (design values)  
 Thickness,  $t = 1.016$   
 Tensile Strength,  $f_{fu} = 1055.00$   
 Tensile Modulus,  $E_f = 64828.00$   
 Elongation,  $e_{fu} = 0.01$   
 Number of directions,  $\text{NoDir} = 1$   
 Fiber orientations,  $b_i: 0.00^\circ$   
 Number of layers,  $NL = 1$   
 Radius of rounding corners,  $R = 40.00$

## Stepwise Properties

At local axis: 2

EDGE -A-

Shear Force,  $V_a = -7.4135260E-048$

EDGE -B-

Shear Force,  $V_b = 7.4135260E-048$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 0.00$

-Compression:  $As_c = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t,ten} = 1017.876$

-Compression:  $As_{c,com} = 1017.876$

-Middle:  $As_{l,mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.30252729$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 90202.132$

with

$M_{pr1} = \text{Max}(Mu_{1+}, Mu_{1-}) = 1.3530E+008$

$Mu_{1+} = 1.3530E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$Mu_{1-} = 1.3530E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(Mu_{2+}, Mu_{2-}) = 1.3530E+008$

$Mu_{2+} = 1.3530E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination

$Mu_{2-} = 1.3530E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

Calculation of  $Mu_{1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$

$M_u = 1.3530E+008$

$= 0.90757121$

$' = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY:  $f_{cc} = f_c^* \quad c = 36.81095$

conf. factor  $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y * \text{Min}(1, 1.25 * (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$Ac = 125663.706$

$= * \text{Min}(1, 1.25 * (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio  $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$db = 18.00$

Mean strength value of all re-bars:  $f_y = 555.55$   
 $f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \frac{1}{2} \times \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

#### Calculation of $\mu_1$ -

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 1.3530E+008$

$\phi = 0.90757121$   
 $\phi' = 0.80580716$   
error of function (3.68), Biskinis Phd = 28928.286  
From 5A.2, TBDY:  $f_{cc} = f'_c \times c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \times \text{Min}(1, 1.25 \times (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $\phi = \phi' \times \text{Min}(1, 1.25 \times (l_b/l_d)^{2/3}) = 0.28805051$

#### Calculation of ratio $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $\phi = 1$

$db = 18.00$   
Mean strength value of all re-bars:  $f_y = 555.55$   
 $f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \frac{1}{2} \times \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

#### Calculation of $\mu_2$ +

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 1.3530E+008$

$\phi = 0.90757121$   
 $\phi' = 0.80580716$   
error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY:  $f_{cc} = f_c' \cdot c = 36.81095$   
 conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
 From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio  $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
 Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $= 1$   
 $db = 18.00$   
 Mean strength value of all re-bars:  $f_y = 555.55$   
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = /2 \cdot \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

Calculation of  $\mu_2$ -

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 1.3530E+008$

$= 0.90757121$   
 $' = 0.80580716$   
 error of function (3.68), Biskinis Phd = 28928.286  
 From 5A.2, TBDY:  $f_{cc} = f_c' \cdot c = 36.81095$   
 conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
 From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio  $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
 Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $= 1$   
 $db = 18.00$   
 Mean strength value of all re-bars:  $f_y = 555.55$   
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)

$t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \frac{1}{2} \times \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 298161.965$

Calculation of Shear Strength at edge 1,  $V_{r1} = 298161.965$

$V_{r1} = V_{Col} \text{ ((10.3), ASCE 41-17)} = k_{nl} \times V_{Col0}$

$V_{Col0} = 298161.965$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \times f_y \times d / s$ ' is replaced by ' $V_s + f \times V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 5.9321525E-012$

$\mu_v = 7.4135260E-048$

$d = 0.8 \times D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 0.00$

$A_v = \frac{1}{2} \times A_{\text{stirrup}} = 123370.055$

$f_y = 444.44$

$s = 360.00$

$V_s$  is multiplied by  $\text{Col} = 0.00$

$s/d = 1.125$

$V_f \text{ ((11-3)-(11.4), ACI 440)} = 194961.134$

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression, where  $a$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = b1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, a1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \times t / N_{\text{Dir}} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe} \text{ ((11-5), ACI 440)} = 259.312$

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$

$b_w \times d = \frac{1}{4} \times d \times d = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 298161.965$

$V_{r2} = V_{Col} \text{ ((10.3), ASCE 41-17)} = k_{nl} \times V_{Col0}$

$V_{Col0} = 298161.965$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \times f_y \times d / s$ ' is replaced by ' $V_s + f \times V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 5.9321525E-012$

$V_u = 7.4135260E-048$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 0.00$   
 $A_v = \frac{1}{2} \cdot A_{stirrup} = 123370.055$   
 $f_y = 444.44$   
 $s = 360.00$   
 $V_s$  is multiplied by  $\text{Col} = 0.00$   
 $s/d = 1.125$   
 $V_f ((11-3)-(11.4), \text{ACI } 440) = 194961.134$   
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta_1 = \theta_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, a_1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = N_L \cdot t / \text{NoDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe} ((11-5), \text{ACI } 440) = 259.312$   
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $b_w \cdot d = \frac{1}{4} \cdot d \cdot d = 80424.772$

-----  
 End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At local axis: 2  
 -----

Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1  
 At local axis: 2  
 Integration Section: (a)  
 Section Type: rccs

Constant Properties

-----  
 Knowledge Factor,  $\gamma = 1.00$   
 Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE 41-17.  
 Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17  
 Consequently:  
 Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$   
 Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.44$   
 Concrete Elasticity,  $E_c = 21019.039$   
 Steel Elasticity,  $E_s = 200000.00$   
 Diameter,  $D = 400.00$   
 Cover Thickness,  $c = 25.00$   
 Element Length,  $L = 3000.00$   
 Primary Member  
 Ribbed Bars  
 Ductile Steel  
 Without Detailing for Earthquake Resistance (including stirrups not closed at  $135^\circ$ )  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Lap Length  $l_b = 300.00$   
 FRP Wrapping Data  
 Type: Carbon  
 Cured laminate properties (design values)  
 Thickness,  $t = 1.016$   
 Tensile Strength,  $f_{fu} = 1055.00$   
 Tensile Modulus,  $E_f = 64828.00$   
 Elongation,  $e_{fu} = 0.01$   
 Number of directions,  $\text{NoDir} = 1$

Fiber orientations, bi: 0.00°  
Number of layers, NL = 1  
Radius of rounding corners, R = 40.00

#### Stepwise Properties

Bending Moment, M = 4.7721810E-010  
Shear Force, V2 = -3472.528  
Shear Force, V3 = -1.2919419E-013  
Axial Force, F = -4769.844  
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension: Aslt = 1272.345  
-Compression: Aslc = 1781.283  
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension: Asl,ten = 1017.876  
-Compression: Asl,com = 1017.876  
-Middle: Asl,mid = 1017.876  
Mean Diameter of Tension Reinforcement, DbL = 18.00

Existing component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $\phi_{u,R} = \phi_u = 0.00931721$   
 $\phi_u = \phi_y + \phi_p = 0.00931721$

- Calculation of  $\phi_y$  -

$\phi_y = (M_y * L_s / 3) / E_{eff} = 0.00931721$  ((4.29), Biskinis Phd))  
 $M_y = 1.4766E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 1500.00  
From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 7.9240E+012$   
factor = 0.30  
 $A_g = 125663.706$   
 $f_c' = 20.00$   
 $N = 4769.844$   
 $E_c * I_g = 2.6413E+013$

#### Calculation of Yielding Moment $M_y$

Calculation of  $\phi_y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \min(M_{y\_ten}, M_{y\_com}) = 1.4766E+008$   
 $\phi_y = 8.1688007E-006$   
 $M_{y\_ten}$  (8c) = 1.4766E+008  
 $\phi_{y\_ten}$  (7c) = 72.40642  
error of function (7c) = 0.00117228  
 $M_{y\_com}$  (8d) = 3.7493E+008  
 $\phi_{y\_com}$  (7d) = 69.91126  
error of function (7d) = 0.00301342  
with ((10.1), ASCE 41-17)  $\phi_{ey} = \min(\phi_{ey}, 1.25 * \phi_{ey} * (I_b / I_d)^{2/3}) = 0.0022222$   
 $\phi_{eco} = 0.002$   
 $\phi_{apl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00157305$   
 $N = 4769.844$   
 $A_c = 125663.706$   
((10.1), ASCE 41-17)  $\phi_{ey} = \min(\phi_{ey}, 1.25 * \phi_{ey} * (I_b / I_d)^{2/3}) = 0.44757577$   
with  $f_c' * ((12.3), ACI 440) = 24.12975$   
 $f_c = 20.00$   
 $f_l = 1.3173$   
 $k = 1$   
Effective FRP thickness,  $t_f = NL * t * \cos(\theta_1) = 1.016$



$\phi_e((12.5) \text{ and } (12.7)) = 0.004$   
 $f_u = 0.01$   
 $E_f = 64828.00$

Calculation of ratio  $l_b/l_d$

Lap Length:  $l_d/l_{d,min} = 0.33490748$

$l_b = 300.00$

$l_d = 895.7698$

Calculation of  $l$  according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $= 1$

$d_b = 18.00$

Mean strength value of all re-bars:  $f_y = 444.44$

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \times \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

- Calculation of  $p$  -

From table 10-9:  $p = 0.00$

with:

- Columns controlled by inadequate development or splicing along the clear height because  $l_b/l_d < 1$

shear control ratio  $V_y E / V_{col} E = 0.30252729$

$d = 0.00$

$s = 0.00$

$t = 2 \cdot A_v / (d_c \cdot s) + 4 \cdot t_f / D \cdot (f_{fe} / f_s) = 0.00721126$

$A_v = 78.53982$ , is the area of the circular stirrup

$d_c = D - 2 \cdot \text{cover} - \text{Hoop Diameter} = 340.00$

The term  $2 \cdot t_f / b_w \cdot (f_{fe} / f_s)$  is implemented to account for FRP contribution

where  $f = 2 \cdot t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength

All these variables have already been given in Shear control ratio calculation.

$N_{UD} = 4769.844$

$A_g = 125663.706$

$f'_c E = 20.00$

$f_{yt} E = f_{yl} E = 444.44$

$p_l = \text{Area}_{\text{Tot Long Rein}} / (A_g) = 0.0243$

$f'_c E = 20.00$

End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (a)

**Calculation No. 3**

column C1, Floor 1

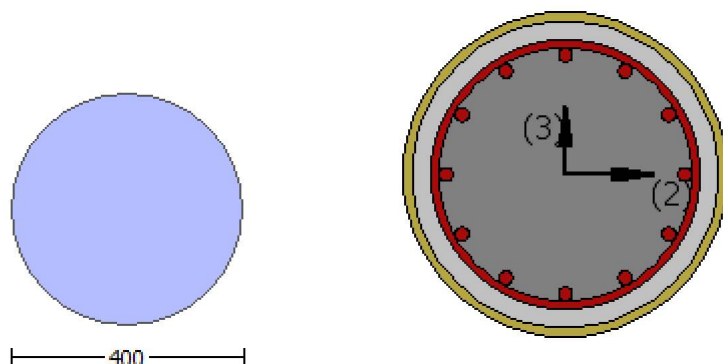
Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Shear capacity VRd

Edge: Start

Local Axis: (3)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (a)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE 41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 16.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 400.00$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of  $\gamma$  for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE 41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE 41-17).

Existing material: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material: Steel Strength,  $f_s = f_{sm} = 444.44$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length  $l_o = l_b = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $\epsilon_{fu} = 0.01$

Number of directions,  $NoDir = 1$

Fiber orientations, bi: 0.00°  
Number of layers, NL = 1  
Radius of rounding corners, R = 40.00

#### Stepwise Properties

EDGE -A-  
Bending Moment, Ma = 4.7721810E-010  
Shear Force, Va = -1.2919419E-013  
EDGE -B-  
Bending Moment, Mb = -8.9325839E-011  
Shear Force, Vb = 1.2919419E-013  
BOTH EDGES  
Axial Force, F = -4769.844  
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension: Aslt = 1272.345  
-Compression: Aslc = 1781.283  
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension: Asl,ten = 1017.876  
-Compression: Asl,com = 1017.876  
-Middle: Asl,mid = 1017.876  
Mean Diameter of Tension Reinforcement, DbL,ten = 18.00

Existing component: From table 7-7, ASCE 41\_17: Final Shear Capacity VR = \*Vn = 286337.204  
Vn ((10.3), ASCE 41-17) = knl\*VCol0 = 286337.204  
VCol = 286337.204  
knl = 1.00  
displacement\_ductility\_demand = 0.00

NOTE: In expression (10-3) 'Vs = Av\*fy\*d/s' is replaced by 'Vs+ f\*Vf'  
where Vf is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)  
fc' = 16.00, but  $fc'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
M/Vd = 2.00  
Mu = 4.7721810E-010  
Vu = 1.2919419E-013  
d = 0.8\*D = 320.00  
Nu = 4769.844  
Ag = 125663.706  
From (11.5.4.8), ACI 318-14: Vs = 0.00  
Av = /2\*A\_stirrup = 123370.055  
fy = 400.00  
s = 360.00  
Vs is multiplied by Col = 0.00  
s/d = 1.125  
Vf ((11-3)-(11.4), ACI 440) = 194961.134  
f = 0.95, for fully-wrapped sections  
wf/sf = 1 (FRP strips adjacent to one another).  
In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a)\sin a$  which is more a generalised expression,  
where is the angle of the crack direction (see KANEPE).  
This later relation, considered as a function Vf( , ), is implemented for every different fiber orientation ai,  
as well as for 2 crack directions, =45° and =-45° to take into consideration the cyclic seismic loading.  
orientation 1: 1 = b1 + 90° = 90.00  
Vf = Min(|Vf(45, 1)|, |Vf(-45,a1)|), with:  
total thickness per orientation, tf1 = NL\*t/NoDir = 1.016  
dfv = d (figure 11.2, ACI 440) = 370.00  
ffe ((11-5), ACI 440) = 259.312  
Ef = 64828.00  
fe = 0.004, from (11.6a), ACI 440  
with fu = 0.01  
From (11-11), ACI 440: Vs + Vf <= 213705.936  
bw\*d = \*d\*d/4 = 80424.772

displacement\_ductility\_demand is calculated as  $\delta / y$

- Calculation of  $\delta / y$  for END A -  
for rotation axis 2 and integ. section (a)

From analysis, chord rotation  $\theta = 2.1684378E-020$   
 $y = (M_y * L_s / 3) / E_{eff} = 0.00931721$  ((4.29), Biskinis Phd))  
 $M_y = 1.4766E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 1500.00  
From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 7.9240E+012$   
 $factor = 0.30$   
 $A_g = 125663.706$   
 $f_c' = 20.00$   
 $N = 4769.844$   
 $E_c * I_g = 2.6413E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\delta$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \min(M_{y\_ten}, M_{y\_com}) = 1.4766E+008$   
 $y = 8.1688007E-006$   
 $M_{y\_ten}$  (8c) = 1.4766E+008  
 $\delta_{ten}$  (7c) = 72.40642  
error of function (7c) = 0.00117228  
 $M_{y\_com}$  (8d) = 3.7493E+008  
 $\delta_{com}$  (7d) = 69.91126  
error of function (7d) = 0.00301342  
with ((10.1), ASCE 41-17)  $e_y = \min(e_y, 1.25 * e_y * (l_b / l_d)^{2/3}) = 0.0022222$   
 $e_{co} = 0.002$   
 $a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00157305$   
 $N = 4769.844$   
 $A_c = 125663.706$   
((10.1), ASCE 41-17)  $\phi = \min(\phi, 1.25 * \phi * (l_b / l_d)^{2/3}) = 0.44757577$   
with  $f_c' * ((12.3), ACI 440) = 24.12975$   
 $f_c = 20.00$   
 $f_l = 1.3173$   
 $k = 1$   
Effective FRP thickness,  $t_f = N L * t * \cos(\theta) = 1.016$   
 $e_{fe}$  ((12.5) and (12.7)) = 0.004  
 $f_u = 0.01$   
 $E_f = 64828.00$

Calculation of ratio  $l_b / l_d$

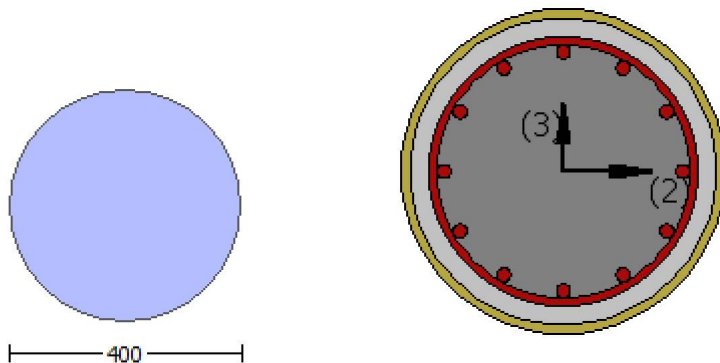
Lap Length:  $l_d / l_{d,min} = 0.33490748$   
 $l_b = 300.00$   
 $l_d = 895.7698$   
Calculation of  $l$  according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $\phi = 1$   
 $d_b = 18.00$   
Mean strength value of all re-bars:  $f_y = 444.44$   
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $c_b = 25.00$

$K_{tr} = 1.14232$   
 $A_{tr} = \frac{1}{2} * \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

-----  
 End Of Calculation of Shear Capacity for element: column CC1 of floor 1  
 At local axis: 3  
 Integration Section: (a)  
 -----

## Calculation No. 4

column C1, Floor 1  
 Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)  
 Analysis: Uniform +X  
 Check: Chord rotation capacity (  $\phi$  )  
 Edge: Start  
 Local Axis: (3)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At Shear local axis: 3  
 (Bending local axis: 2)  
 Section Type: rccs

Constant Properties

-----  
 Knowledge Factor,  $\gamma = 1.00$   
 Mean strength values are used for both shear and moment calculations.  
 Consequently:  
 Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$   
 Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.44$   
 Concrete Elasticity,  $E_c = 21019.039$   
 Steel Elasticity,  $E_s = 200000.00$   
 #####  
 Note: Especially for the calculation of moment strengths,  
 the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14  
 Existing material: Steel Strength,  $f_s = 1.25 * f_{sm} = 555.55$   
 #####

Diameter,  $D = 400.00$   
 Cover Thickness,  $c = 25.00$   
 Mean Confinement Factor overall section = 1.84055  
 Element Length,  $L = 3000.00$   
 Primary Member  
 Ribbed Bars  
 Ductile Steel  
 Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Lap Length  $l_o = 300.00$   
 FRP Wrapping Data  
 Type: Carbon  
 Cured laminate properties (design values)  
 Thickness,  $t = 1.016$   
 Tensile Strength,  $f_{fu} = 1055.00$   
 Tensile Modulus,  $E_f = 64828.00$   
 Elongation,  $ε_{fu} = 0.01$   
 Number of directions,  $N_{oDir} = 1$   
 Fiber orientations,  $b_i: 0.00^\circ$   
 Number of layers,  $N_L = 1$   
 Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

At local axis: 3  
 EDGE -A-  
 Shear Force,  $V_a = 1.2107607E-031$   
 EDGE -B-  
 Shear Force,  $V_b = -1.2107607E-031$   
 BOTH EDGES  
 Axial Force,  $F = -4771.233$   
 Longitudinal Reinforcement Area Distribution (in 2 divisions)  
     -Tension:  $A_{sl,t} = 0.00$   
     -Compression:  $A_{sl,c} = 3053.628$   
 Longitudinal Reinforcement Area Distribution (in 3 divisions)  
     -Tension:  $A_{sl,ten} = 1017.876$   
     -Compression:  $A_{sl,com} = 1017.876$   
     -Middle:  $A_{sl,mid} = 1017.876$

Calculation of Shear Capacity ratio ,  $V_e/V_r = 0.30252729$   
 Member Controlled by Flexure ( $V_e/V_r < 1$ )  
 Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 90202.132$   
 with  
 $M_{pr1} = \text{Max}(Mu_{1+}, Mu_{1-}) = 1.3530E+008$   
 $Mu_{1+} = 1.3530E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination  
 $Mu_{1-} = 1.3530E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination  
 $M_{pr2} = \text{Max}(Mu_{2+}, Mu_{2-}) = 1.3530E+008$   
 $Mu_{2+} = 1.3530E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination  
 $Mu_{2-} = 1.3530E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

#### Calculation of $Mu_{1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$   
 $M_u = 1.3530E+008$   
 $= 0.90757121$

```

' = 0.80580716
error of function (3.68), Biskinis Phd = 28928.286
From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$ 
  conf. factor  $c = 1.84055$ 
   $f_c = 20.00$ 
From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$ 
   $l_b/l_d = 0.26792599$ 
   $d_1 = 44.00$ 
   $R = 200.00$ 
   $v = 0.00155946$ 
   $N = 4771.233$ 
   $A_c = 125663.706$ 
   $= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$ 

```

#### Calculation of ratio $l_b/l_d$

```

Lap Length:  $l_b/l_d = 0.26792599$ 
 $l_b = 300.00$ 
 $l_d = 1119.712$ 
Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
  = 1
   $db = 18.00$ 
Mean strength value of all re-bars:  $f_y = 555.55$ 
 $f_c' = 20.00$ , but  $f_c^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)
   $t = 1.00$ 
   $s = 0.80$ 
   $e = 1.00$ 
   $cb = 25.00$ 
   $K_{tr} = 1.14232$ 
   $A_{tr} = \sqrt{2} \cdot \text{Area of stirrup} = 123.3701$ 
   $s = 360.00$ 
   $n = 12.00$ 

```

#### Calculation of $\mu_{u1}$

```

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$ 
 $\mu_u = 1.3530E+008$ 

```

```

  = 0.90757121
' = 0.80580716
error of function (3.68), Biskinis Phd = 28928.286
From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$ 
  conf. factor  $c = 1.84055$ 
   $f_c = 20.00$ 
From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$ 
   $l_b/l_d = 0.26792599$ 
   $d_1 = 44.00$ 
   $R = 200.00$ 
   $v = 0.00155946$ 
   $N = 4771.233$ 
   $A_c = 125663.706$ 
   $= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$ 

```

#### Calculation of ratio $l_b/l_d$

```

Lap Length:  $l_b/l_d = 0.26792599$ 
 $l_b = 300.00$ 
 $l_d = 1119.712$ 
Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.
  = 1
   $db = 18.00$ 

```

Mean strength value of all re-bars:  $f_y = 555.55$   
 $f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \frac{1}{2} \times \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

#### Calculation of $\mu_{2+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 1.3530E+008$

$\phi = 0.90757121$   
 $\phi' = 0.80580716$   
error of function (3.68), Biskinis Phd = 28928.286  
From 5A.2, TBDY:  $f_{cc} = f'_c \times c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \times \text{Min}(1, 1.25 \times (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $\phi = \phi' \times \text{Min}(1, 1.25 \times (l_b/l_d)^{2/3}) = 0.28805051$

#### Calculation of ratio $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $\phi = 1$   
 $db = 18.00$   
Mean strength value of all re-bars:  $f_y = 555.55$   
 $f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \frac{1}{2} \times \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

#### Calculation of $\mu_{2-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 1.3530E+008$

$\phi = 0.90757121$   
 $\phi' = 0.80580716$   
error of function (3.68), Biskinis Phd = 28928.286



From 5A.2, TBDY:  $f_{cc} = f_c' \cdot c = 36.81095$   
 conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
 From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio  $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
 Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $= 1$   
 $d_b = 18.00$   
 Mean strength value of all re-bars:  $f_y = 555.55$   
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 298161.965$

Calculation of Shear Strength at edge 1,  $V_{r1} = 298161.965$   
 $V_{r1} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{Col0}$   
 $V_{Col0} = 298161.965$   
 $k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)  
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $\mu_u = 1.2899604E-011$   
 $\nu_u = 1.2107607E-031$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 0.00$   
 $A_v = \frac{1}{2} \cdot A_{\text{stirrup}} = 123370.055$   
 $f_y = 444.44$   
 $s = 360.00$   
 $V_s$  is multiplied by  $\text{Col} = 0.00$   
 $s/d = 1.125$   
 $V_f$  ((11-3)-(11.4), ACI 440) =  $194961.134$   
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression, where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta)$ , is implemented for every different fiber orientation  $\theta_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta_1 = \theta_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, a1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}((11-5), \text{ACI 440}) = 259.312$   
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $b_w \cdot d = \frac{1}{4} \cdot d \cdot d = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 298161.965$   
 $V_{r2} = V_{Col}((10.3), \text{ASCE 41-17}) = k_{nl} \cdot V_{Col0}$   
 $V_{Col0} = 298161.965$   
 $k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ '  
 where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)  
 $f'_c = 20.00$ , but  $f'_c^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $\mu_u = 1.2899604E-011$   
 $\nu_u = 1.2107607E-031$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 0.00$   
 $A_v = \frac{1}{2} \cdot A_{stirrup} = 123370.055$   
 $f_y = 444.44$   
 $s = 360.00$   
 $V_s$  is multiplied by  $\text{Col} = 0.00$   
 $s/d = 1.125$   
 $V_f((11-3)-(11.4), \text{ACI 440}) = 194961.134$   
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = b1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, a1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}((11-5), \text{ACI 440}) = 259.312$   
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $b_w \cdot d = \frac{1}{4} \cdot d \cdot d = 80424.772$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At local axis: 3

Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At Shear local axis: 2  
 (Bending local axis: 3)  
 Section Type: rccs

Constant Properties

Knowledge Factor,  $\phi = 1.00$   
 Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.44$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 555.55$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.84055

Element Length,  $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length  $l_o = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $\epsilon_{fu} = 0.01$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i: 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

At local axis: 2

EDGE -A-

Shear Force,  $V_a = -7.4135260E-048$

EDGE -B-

Shear Force,  $V_b = 7.4135260E-048$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 0.00$

-Compression:  $As_c = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t,ten} = 1017.876$

-Compression:  $As_{l,com} = 1017.876$

-Middle:  $As_{l,mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.30252729$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 90202.132$

with

$M_{pr1} = \max(\mu_{u1+}, \mu_{u1-}) = 1.3530E+008$

$\mu_{u1+} = 1.3530E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$\mu_{u1-} = 1.3530E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \max(\mu_{u2+}, \mu_{u2-}) = 1.3530E+008$

$\mu_{u2+} = 1.3530E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination

$\mu_{u2-} = 1.3530E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

#### Calculation of Mu1+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.3530E+008

= 0.90757121  
' = 0.80580716  
error of function (3.68), Biskinis Phd = 28928.286  
From 5A.2, TBDY:  $f_{cc} = f_c^* c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y * \text{Min}(1, 1.25 * (l_b / l_d)^{2/3}) = 288.6089$   
 $l_b / l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
=  $*\text{Min}(1, 1.25 * (l_b / l_d)^{2/3}) = 0.28805051$

#### Calculation of ratio lb/l\_d

Lap Length:  $l_b / l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
= 1  
 $db = 18.00$   
Mean strength value of all re-bars:  $f_y = 555.55$   
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \sqrt{2} * \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

#### Calculation of Mu1-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.3530E+008

= 0.90757121  
' = 0.80580716  
error of function (3.68), Biskinis Phd = 28928.286  
From 5A.2, TBDY:  $f_{cc} = f_c^* c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y * \text{Min}(1, 1.25 * (l_b / l_d)^{2/3}) = 288.6089$   
 $l_b / l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
=  $*\text{Min}(1, 1.25 * (l_b / l_d)^{2/3}) = 0.28805051$

#### Calculation of ratio $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of  $l_b$ ,min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$db = 18.00$

Mean strength value of all re-bars:  $f_y = 555.55$

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$cb = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \times \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

#### Calculation of $\mu_{2+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$\mu = 1.3530E+008$

$= 0.90757121$

$' = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY:  $f_{cc} = f'_c \times c = 36.81095$

conf. factor  $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \times \text{Min}(1, 1.25 \times (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \text{Min}(1, 1.25 \times (l_b/l_d)^{2/3}) = 0.28805051$

#### Calculation of ratio $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of  $l_b$ ,min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$db = 18.00$

Mean strength value of all re-bars:  $f_y = 555.55$

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$cb = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \times \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

## Calculation of Mu2-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.3530E+008

= 0.90757121  
' = 0.80580716  
error of function (3.68), Biskinis Phd = 28928.286  
From 5A.2, TBDY:  $f_{cc} = f_c' \cdot c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
=  $\cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

## Calculation of ratio $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
= 1  
 $d_b = 18.00$   
Mean strength value of all re-bars:  $f_y = 555.55$   
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $c_b = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

## Calculation of Shear Strength $V_r = \text{Min}(V_{r1}, V_{r2}) = 298161.965$

Calculation of Shear Strength at edge 1,  $V_{r1} = 298161.965$   
 $V_{r1} = V_{CoI}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{CoI0}$   
 $V_{CoI0} = 298161.965$   
 $k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ '  
where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)  
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $\mu_u = 5.9321525E-012$   
 $V_u = 7.4135260E-048$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
From (11.5.4.8), ACI 318-14:  $V_s = 0.00$   
 $A_v = \frac{1}{2} \cdot A_{\text{stirrup}} = 123370.055$   
 $f_y = 444.44$   
 $s = 360.00$

$V_s$  is multiplied by  $\text{Col} = 0.00$   
 $s/d = 1.125$   
 $V_f ((11-3)-(11.4), \text{ACI } 440) = 194961.134$   
 $f = 0.95$ , for fully-wrapped sections  
 $wf/sf = 1$  (FRP strips adjacent to one another).  
In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a)\sin a$  which is more a generalised expression,  
where  $a$  is the angle of the crack direction (see KANEPE).  
This later relation, considered as a function  $V_f( , )$ , is implemented for every different fiber orientation  $a_i$ ,  
as well as for 2 crack directions,  $= 45^\circ$  and  $= -45^\circ$  to take into consideration the cyclic seismic loading.  
orientation 1:  $\theta_1 = \theta_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, a_1)|)$ , with:  
total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe} ((11-5), \text{ACI } 440) = 259.312$   
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
with  $f_u = 0.01$   
From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $b_w \cdot d = \text{Col} \cdot d / 4 = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 298161.965$   
 $V_{r2} = V_{\text{Col}} ((10.3), \text{ASCE } 41-17) = \text{knI} \cdot V_{\text{ColO}}$   
 $V_{\text{ColO}} = 298161.965$   
 $\text{knI} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ '  
where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\text{Col} = 1$  (normal-weight concrete)  
 $f'_c = 20.00$ , but  $f'_c^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $\mu_u = 5.9321525\text{E-}012$   
 $\nu_u = 7.4135260\text{E-}048$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
From (11.5.4.8), ACI 318-14:  $V_s = 0.00$   
 $A_v = \text{Col} / 2 \cdot A_{\text{stirrup}} = 123370.055$   
 $f_y = 444.44$   
 $s = 360.00$   
 $V_s$  is multiplied by  $\text{Col} = 0.00$   
 $s/d = 1.125$   
 $V_f ((11-3)-(11.4), \text{ACI } 440) = 194961.134$   
 $f = 0.95$ , for fully-wrapped sections  
 $wf/sf = 1$  (FRP strips adjacent to one another).  
In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a)\sin a$  which is more a generalised expression,  
where  $a$  is the angle of the crack direction (see KANEPE).  
This later relation, considered as a function  $V_f( , )$ , is implemented for every different fiber orientation  $a_i$ ,  
as well as for 2 crack directions,  $= 45^\circ$  and  $= -45^\circ$  to take into consideration the cyclic seismic loading.  
orientation 1:  $\theta_1 = \theta_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, a_1)|)$ , with:  
total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe} ((11-5), \text{ACI } 440) = 259.312$   
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
with  $f_u = 0.01$   
From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $b_w \cdot d = \text{Col} \cdot d / 4 = 80424.772$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At local axis: 2

Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (a)

Section Type: rccs

#### Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE 41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.44$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length  $l_b = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $\epsilon_{fu} = 0.01$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i: 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

Bending Moment,  $M = -1.0423E+007$

Shear Force,  $V_2 = -3472.528$

Shear Force,  $V_3 = -1.2919419E-013$

Axial Force,  $F = -4769.844$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{st} = 1272.345$

-Compression:  $A_{sc} = 1781.283$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{st,ten} = 1017.876$

-Compression:  $A_{sc,com} = 1017.876$

-Middle:  $A_{st,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $D_bL = 18.00$

Existing component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $\phi_{u,R} = \phi_u = 0.01864342$

$\phi_u = \phi_y + \phi_p = 0.01864342$

- Calculation of  $\phi_y$  -

$\phi_y = (M_y * L_s / 3) / E_{eff} = 0.01864342 ((4.29), \text{Biskinis Phd})$

$M_y = 1.4766E+008$

$L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 3001.447

From table 10.5, ASCE 41\_17:  $E_{eff} = \text{factor} * E_c * I_g = 7.9240E+012$

factor = 0.30



Ag = 125663.706  
fc' = 20.00  
N = 4769.844  
Ec\*Ig = 2.6413E+013

#### Calculation of Yielding Moment My

Calculation of  $\rho_y$  and My according to (7) - (8) in Biskinis and Fardis

My = Min(My\_ten, My\_com) = 1.4766E+008  
 $\rho_y = 8.1688007E-006$   
My\_ten (8c) = 1.4766E+008  
 $\rho_{y\_ten} (7c) = 72.40642$   
error of function (7c) = 0.00117228  
My\_com (8d) = 3.7493E+008  
 $\rho_{y\_com} (7d) = 69.91126$   
error of function (7d) = 0.00301342  
with ((10.1), ASCE 41-17)  $\rho_y = \text{Min}(\rho_y, 1.25 * \rho_y * (l_b/l_d)^{2/3}) = 0.0022222$   
 $\rho_{eco} = 0.002$   
 $\rho_{apl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
d1 = 44.00  
R = 200.00  
v = 0.00157305  
N = 4769.844  
Ac = 125663.706  
((10.1), ASCE 41-17)  $\rho_y = \text{Min}(\rho_y, 1.25 * \rho_y * (l_b/l_d)^{2/3}) = 0.44757577$   
with fc' ((12.3), ACI 440) = 24.12975  
fc = 20.00  
fl = 1.3173  
k = 1  
Effective FRP thickness, tf = NL\*t\*cos(b1) = 1.016  
efe ((12.5) and (12.7)) = 0.004  
fu = 0.01  
Ef = 64828.00

#### Calculation of ratio lb/l\_d

Lap Length:  $l_d/l_{d,min} = 0.33490748$   
lb = 300.00  
ld = 895.7698  
Calculation of l according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
= 1  
db = 18.00  
Mean strength value of all re-bars: fy = 444.44  
fc' = 20.00, but  $fc'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
t = 1.00  
s = 0.80  
e = 1.00  
cb = 25.00  
Ktr = 1.14232  
Atr =  $\sqrt{2} * \text{Area of stirrup} = 123.3701$   
s = 360.00  
n = 12.00

#### - Calculation of $\rho_p$ -

From table 10-9:  $\rho_p = 0.00$

with:

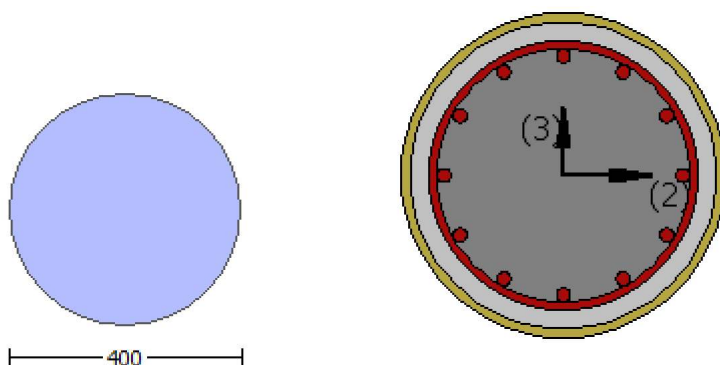
- Columns controlled by inadequate development or splicing along the clear height because  $l_b/l_d < 1$   
shear control ratio  $V_y E / V_{CoI OE} = 0.30252729$   
d = 0.00  
s = 0.00

$t = 2 \cdot A_v / (d_c \cdot s) + 4 \cdot t_f / D \cdot (f_{fe} / f_s) = 0.00721126$   
 $A_v = 78.53982$ , is the area of the circular stirrup  
 $d_c = D - 2 \cdot \text{cover} - \text{Hoop Diameter} = 340.00$   
 The term  $2 \cdot t_f / b_w \cdot (f_{fe} / f_s)$  is implemented to account for FRP contribution  
 where  $f = 2 \cdot t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength  
 All these variables have already been given in Shear control ratio calculation.  
 $NUD = 4769.844$   
 $Ag = 125663.706$   
 $f_{cE} = 20.00$   
 $f_{yE} = f_{yI} = 444.44$   
 $pl = \text{Area\_Tot\_Long\_Rein} / (Ag) = 0.0243$   
 $f_{cE} = 20.00$

-----  
 End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1  
 At local axis: 3  
 Integration Section: (a)  
 -----

## Calculation No. 5

column C1, Floor 1  
 Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)  
 Analysis: Uniform +X  
 Check: Shear capacity VRd  
 Edge: End  
 Local Axis: (2)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1  
 At local axis: 2  
 Integration Section: (b)  
 Section Type: rccs

Constant Properties

-----  
 Knowledge Factor,  $\gamma = 1.00$   
 Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE 41-17.  
 Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 16.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 400.00$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of  $\gamma$  for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE 41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE 41-17).

Existing material: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material: Steel Strength,  $f_s = f_{sm} = 444.44$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length  $l_o = l_b = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $ef_u = 0.01$

Number of directions,  $NoDir = 1$

Fiber orientations,  $bi: 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

EDGE -A-

Bending Moment,  $M_a = -1.0423E+007$

Shear Force,  $V_a = -3472.528$

EDGE -B-

Bending Moment,  $M_b = 0.09265189$

Shear Force,  $V_b = 3472.528$

BOTH EDGES

Axial Force,  $F = -4769.844$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 0.00$

-Compression:  $As_c = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{l,ten} = 1017.876$

-Compression:  $As_{l,com} = 1017.876$

-Middle:  $As_{l,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $Db_{L,ten} = 18.00$

Existing component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = V_n = 286337.204$

$V_n$  ((10.3), ASCE 41-17) =  $k_n V_{CoI} = 286337.204$

$V_{CoI} = 286337.204$

$k_n = 1.00$

$displacement\_ductility\_demand = 0.10586773$

NOTE: In expression (10-3) ' $V_s = A_v f_y d/s$ ' is replaced by ' $V_s + f' V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

$f'_c = 16.00$ , but  $f'_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$   
 $\mu_u = 0.09265189$   
 $V_u = 3472.528$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4769.844$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 0.00$   
 $A_v = \frac{1}{2} \cdot A_{stirrup} = 123370.055$   
 $f_y = 400.00$   
 $s = 360.00$   
 $V_s$  is multiplied by  $\text{Col} = 0.00$   
 $s/d = 1.125$   
 $V_f ((11-3)-(11.4), \text{ACI 440}) = 194961.134$   
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = 45^\circ$   
 $V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \theta)|)$ , with:  
 total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{Dir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe} ((11-5), \text{ACI 440}) = 259.312$   
 $E_f = 64828.00$   
 $f_{fe} = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 213705.936$   
 $b_w \cdot d = \frac{V_u}{\phi \cdot f_{fe}} = 80424.772$

displacement ductility demand is calculated as  $\delta_u / y$

- Calculation of  $\delta_u / y$  for END B -  
 for rotation axis 3 and integ. section (b)

From analysis, chord rotation  $\theta = 0.00019728$   
 $y = (M_y \cdot L_s / 3) / E_{eff} = 0.00186344$  ((4.29), Biskinis Phd))  
 $M_y = 1.4766E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) = 300.00  
 From table 10.5, ASCE 41-17:  $E_{eff} = \text{factor} \cdot E_c \cdot I_g = 7.9240E+012$   
 $\text{factor} = 0.30$   
 $A_g = 125663.706$   
 $f'_c = 20.00$   
 $N = 4769.844$   
 $E_c \cdot I_g = 2.6413E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\delta_u$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 1.4766E+008$   
 $y = 8.1688007E-006$   
 $M_{y\_ten} (8c) = 1.4766E+008$   
 $y_{ten} (7c) = 72.40642$   
 error of function (7c) = 0.00117228  
 $M_{y\_com} (8d) = 3.7493E+008$   
 $y_{com} (7d) = 69.91126$   
 error of function (7d) = 0.00301342  
 with ((10.1), ASCE 41-17)  $e_y = \text{Min}(e_y, 1.25 \cdot e_y \cdot (I_b/I_d)^{2/3}) = 0.0022222$   
 $e_{co} = 0.002$   
 $a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 200.00$

$v = 0.00157305$   
 $N = 4769.844$   
 $A_c = 125663.706$   
 $((10.1), ASCE 41-17) = \text{Min}( , 1.25 * ((lb/ld)^{2/3}) = 0.44757577$   
 with  $f_c' ((12.3), ACI 440) = 24.12975$   
 $f_c = 20.00$   
 $f_l = 1.3173$   
 $k = 1$   
 Effective FRP thickness,  $t_f = NL * t * \cos(b_1) = 1.016$   
 $e_{fe} ((12.5) \text{ and } (12.7)) = 0.004$   
 $f_u = 0.01$   
 $E_f = 64828.00$

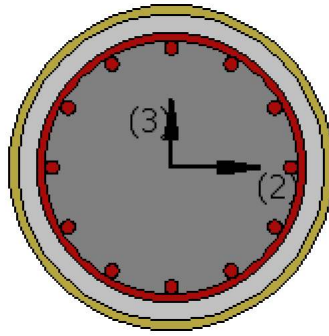
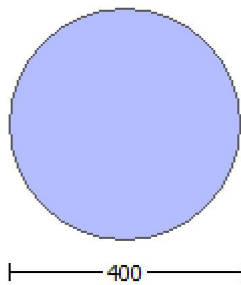
Calculation of ratio  $lb/ld$

Lap Length:  $ld/ld, \min = 0.33490748$   
 $lb = 300.00$   
 $ld = 895.7698$   
 Calculation of  $I$  according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $= 1$   
 $db = 18.00$   
 Mean strength value of all re-bars:  $f_y = 444.44$   
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \pi/4 * \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

End Of Calculation of Shear Capacity for element: column CC1 of floor 1  
 At local axis: 2  
 Integration Section: (b)

## Calculation No. 6

column C1, Floor 1  
 Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)  
 Analysis: Uniform +X  
 Check: Chord rotation capacity (  $\phi$  )  
 Edge: End  
 Local Axis: (2)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.44$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 555.55$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.84055

Element Length,  $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length  $l_o = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $\epsilon_{fu} = 0.01$

Number of directions,  $NoDir = 1$

Fiber orientations,  $b_i: 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

Stepwise Properties

At local axis: 3

EDGE -A-

Shear Force,  $V_a = 1.2107607E-031$

EDGE -B-

Shear Force,  $V_b = -1.2107607E-031$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $Asl_t = 0.00$   
 -Compression:  $Asl_c = 3053.628$   
 Longitudinal Reinforcement Area Distribution (in 3 divisions)  
 -Tension:  $Asl_{ten} = 1017.876$   
 -Compression:  $Asl_{com} = 1017.876$   
 -Middle:  $Asl_{mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.30252729$   
 Member Controlled by Flexure ( $V_e/V_r < 1$ )  
 Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 90202.132$   
 with  
 $M_{pr1} = \max(\mu_{u1+}, \mu_{u1-}) = 1.3530E+008$   
 $\mu_{u1+} = 1.3530E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination  
 $\mu_{u1-} = 1.3530E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination  
 $M_{pr2} = \max(\mu_{u2+}, \mu_{u2-}) = 1.3530E+008$   
 $\mu_{u2+} = 1.3530E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination  
 $\mu_{u2-} = 1.3530E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

Calculation of  $\mu_{u1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$   
 $\mu_u = 1.3530E+008$

$\phi = 0.90757121$   
 $\phi' = 0.80580716$   
 error of function (3.68), Biskinis Phd = 28928.286  
 From 5A.2, TBDY:  $f_{cc} = f_c' \cdot c = 36.81095$   
 conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
 From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \min(1, 1.25 \cdot (l_b/d)^{2/3}) = 288.6089$   
 $l_b/d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \phi' \cdot \min(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.28805051$

Calculation of ratio  $l_b/d$

Lap Length:  $l_b/d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
 Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $\phi = 1$   
 $d_b = 18.00$   
 Mean strength value of all re-bars:  $f_y = 555.55$   
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \pi \cdot s^2 \cdot n = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

Calculation of Mu1-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.3530E+008

= 0.90757121  
' = 0.80580716  
error of function (3.68), Biskinis Phd = 28928.286  
From 5A.2, TDY:  $f_{cc} = f_c \cdot c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
=  $\cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio  $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
= 1  
 $db = 18.00$   
Mean strength value of all re-bars:  $f_y = 555.55$   
 $f_c' = 20.00$ , but  $f_c^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

Calculation of Mu2+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.3530E+008

= 0.90757121  
' = 0.80580716  
error of function (3.68), Biskinis Phd = 28928.286  
From 5A.2, TDY:  $f_{cc} = f_c \cdot c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$



$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.28805051$$

Calculation of ratio  $l_b/d$

Lap Length:  $l_b/d = 0.26792599$

$l_b = 300.00$

$d = 1119.712$

Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$d_b = 18.00$

Mean strength value of all re-bars:  $f_y = 555.55$

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of  $\mu_2$ -

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$\mu = 1.3530E+008$

$= 0.90757121$

$' = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY:  $f_{cc} = f'_c \cdot c = 36.81095$

conf. factor  $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 288.6089$

$l_b/d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.28805051$

Calculation of ratio  $l_b/d$

Lap Length:  $l_b/d = 0.26792599$

$l_b = 300.00$

$d = 1119.712$

Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$d_b = 18.00$

Mean strength value of all re-bars:  $f_y = 555.55$

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of Shear Strength  $V_r = \min(V_{r1}, V_{r2}) = 298161.965$

Calculation of Shear Strength at edge 1,  $V_{r1} = 298161.965$

$V_{r1} = V_{Col} \text{ ((10.3), ASCE 41-17)} = k_{nl} * V_{Col0}$

$V_{Col0} = 298161.965$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\lambda = 1$  (normal-weight concrete)

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.2899604E-011$

$\nu_u = 1.2107607E-031$

$d = 0.8 * D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 0.00$

$A_v = \sqrt{2} * A_{stirrup} = 123370.055$

$f_y = 444.44$

$s = 360.00$

$V_s$  is multiplied by  $\lambda_{Col} = 0.00$

$s/d = 1.125$

$V_f \text{ ((11-3)-(11.4), ACI 440)} = 194961.134$

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression, where  $a$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta)$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta_1 = b_1 + 90^\circ = 90.00$

$V_f = \min(|V_f(45, \theta_1)|, |V_f(-45, a_1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L * t / N_{oDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe} \text{ ((11-5), ACI 440)} = 259.312$

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$

$b_w * d = \lambda * d * d / 4 = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 298161.965$

$V_{r2} = V_{Col} \text{ ((10.3), ASCE 41-17)} = k_{nl} * V_{Col0}$

$V_{Col0} = 298161.965$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\lambda = 1$  (normal-weight concrete)

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.2899604E-011$

$\nu_u = 1.2107607E-031$

$d = 0.8 * D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 0.00$

$A_v = \sqrt{2} * A_{stirrup} = 123370.055$

$f_y = 444.44$

$s = 360.00$

$V_s$  is multiplied by  $\lambda_{Col} = 0.00$

$s/d = 1.125$

$V_f((11-3)-(11.4), \text{ACI 440}) = 194961.134$   
 $f = 0.95$ , for fully-wrapped sections  
 $wf/sf = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a)\sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta_1 = \theta_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, \theta_1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}((11-5), \text{ACI 440}) = 259.312$   
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $b_w \cdot d = \frac{V_s \cdot d}{4} = 80424.772$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At local axis: 3

Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At Shear local axis: 2  
 (Bending local axis: 3)  
 Section Type: rccs

#### Constant Properties

Knowledge Factor,  $\phi = 1.00$   
 Mean strength values are used for both shear and moment calculations.  
 Consequently:  
 Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$   
 Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.44$   
 Concrete Elasticity,  $E_c = 21019.039$   
 Steel Elasticity,  $E_s = 200000.00$   
 #####  
 Note: Especially for the calculation of moment strengths,  
 the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14  
 Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 555.55$   
 #####  
 Diameter,  $D = 400.00$   
 Cover Thickness,  $c = 25.00$   
 Mean Confinement Factor overall section = 1.84055  
 Element Length,  $L = 3000.00$   
 Primary Member  
 Ribbed Bars  
 Ductile Steel  
 Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Lap Length  $l_o = 300.00$   
 FRP Wrapping Data  
 Type: Carbon  
 Cured laminate properties (design values)  
 Thickness,  $t = 1.016$   
 Tensile Strength,  $f_{fu} = 1055.00$   
 Tensile Modulus,  $E_f = 64828.00$   
 Elongation,  $e_{fu} = 0.01$   
 Number of directions,  $\text{NoDir} = 1$   
 Fiber orientations,  $b_i: 0.00^\circ$   
 Number of layers,  $NL = 1$   
 Radius of rounding corners,  $R = 40.00$

## Stepwise Properties

At local axis: 2

EDGE -A-

Shear Force,  $V_a = -7.4135260E-048$

EDGE -B-

Shear Force,  $V_b = 7.4135260E-048$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 0.00$

-Compression:  $As_c = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t,ten} = 1017.876$

-Compression:  $As_{c,com} = 1017.876$

-Middle:  $As_{l,mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.30252729$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 90202.132$

with

$M_{pr1} = \text{Max}(Mu_{1+}, Mu_{1-}) = 1.3530E+008$

$Mu_{1+} = 1.3530E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$Mu_{1-} = 1.3530E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(Mu_{2+}, Mu_{2-}) = 1.3530E+008$

$Mu_{2+} = 1.3530E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination

$Mu_{2-} = 1.3530E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

Calculation of  $Mu_{1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$

$M_u = 1.3530E+008$

$= 0.90757121$

$' = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY:  $f_{cc} = f_c^* \quad c = 36.81095$

conf. factor  $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$Ac = 125663.706$

$= \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio  $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$db = 18.00$

Mean strength value of all re-bars:  $f_y = 555.55$   
 $f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \frac{1}{2} \times \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

#### Calculation of $\mu_{1-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 1.3530E+008$

$\mu = 0.90757121$   
 $\mu' = 0.80580716$   
error of function (3.68), Biskinis Phd = 28928.286  
From 5A.2, TBDY:  $f_{cc} = f'_c \times c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \times \text{Min}(1, 1.25 \times (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $\mu = \mu' \times \text{Min}(1, 1.25 \times (l_b/l_d)^{2/3}) = 0.28805051$

#### Calculation of ratio $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $\mu = 1$   
 $db = 18.00$   
Mean strength value of all re-bars:  $f_y = 555.55$   
 $f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \frac{1}{2} \times \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

#### Calculation of $\mu_{2+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 1.3530E+008$

$\mu = 0.90757121$   
 $\mu' = 0.80580716$   
error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY:  $f_{cc} = f_c' \cdot c = 36.81095$   
 conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
 From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio  $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
 Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $= 1$   
 $db = 18.00$   
 Mean strength value of all re-bars:  $f_y = 555.55$   
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = /2 \cdot \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

Calculation of  $\mu_2$ -

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 1.3530E+008$

$= 0.90757121$   
 $' = 0.80580716$   
 error of function (3.68), Biskinis Phd = 28928.286  
 From 5A.2, TBDY:  $f_{cc} = f_c' \cdot c = 36.81095$   
 conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
 From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio  $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
 Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $= 1$   
 $db = 18.00$   
 Mean strength value of all re-bars:  $f_y = 555.55$   
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)

$t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \pi/2 \cdot \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

Calculation of Shear Strength  $V_r = \min(V_{r1}, V_{r2}) = 298161.965$

Calculation of Shear Strength at edge 1,  $V_{r1} = 298161.965$

$V_{r1} = V_{Col} \text{ ((10.3), ASCE 41-17)} = k_{nl} \cdot V_{Col0}$

$V_{Col0} = 298161.965$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 5.9321525E-012$

$\mu_v = 7.4135260E-048$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 0.00$

$A_v = \pi/2 \cdot A_{stirrup} = 123370.055$

$f_y = 444.44$

$s = 360.00$

$V_s$  is multiplied by  $\phi_{Col} = 0.00$

$s/d = 1.125$

$V_f \text{ ((11-3)-(11.4), ACI 440)} = 194961.134$

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \cot \alpha) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = b1 + 90^\circ = 90.00$

$V_f = \min(|V_f(45, \theta)|, |V_f(-45, \alpha)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{oDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe} \text{ ((11-5), ACI 440)} = 259.312$

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$

$b_w \cdot d = \pi \cdot d^2 / 4 = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 298161.965$

$V_{r2} = V_{Col} \text{ ((10.3), ASCE 41-17)} = k_{nl} \cdot V_{Col0}$

$V_{Col0} = 298161.965$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 5.9321525E-012$

$V_u = 7.4135260E-048$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 0.00$   
 $A_v = \frac{1}{2} \cdot A_{\text{stirrup}} = 123370.055$   
 $f_y = 444.44$   
 $s = 360.00$   
 $V_s$  is multiplied by  $\text{Col} = 0.00$   
 $s/d = 1.125$   
 $V_f ((11-3)-(11.4), \text{ACI } 440) = 194961.134$   
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta_1 = \theta_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, a_1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = N_L \cdot t / \text{NoDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe} ((11-5), \text{ACI } 440) = 259.312$   
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $b_w \cdot d = \frac{1}{4} \cdot d \cdot d = 80424.772$

-----  
 End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At local axis: 2  
 -----

Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1  
 At local axis: 2  
 Integration Section: (b)  
 Section Type: rccs

Constant Properties

-----  
 Knowledge Factor,  $\gamma = 1.00$   
 Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE 41-17.  
 Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17  
 Consequently:  
 Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$   
 Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.44$   
 Concrete Elasticity,  $E_c = 21019.039$   
 Steel Elasticity,  $E_s = 200000.00$   
 Diameter,  $D = 400.00$   
 Cover Thickness,  $c = 25.00$   
 Element Length,  $L = 3000.00$   
 Primary Member  
 Ribbed Bars  
 Ductile Steel  
 Without Detailing for Earthquake Resistance (including stirrups not closed at  $135^\circ$ )  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Lap Length  $l_b = 300.00$   
 FRP Wrapping Data  
 Type: Carbon  
 Cured laminate properties (design values)  
 Thickness,  $t = 1.016$   
 Tensile Strength,  $f_{fu} = 1055.00$   
 Tensile Modulus,  $E_f = 64828.00$   
 Elongation,  $e_{fu} = 0.01$   
 Number of directions,  $\text{NoDir} = 1$



Fiber orientations, bi: 0.00°  
Number of layers, NL = 1  
Radius of rounding corners, R = 40.00

#### Stepwise Properties

Bending Moment, M = -8.9325839E-011  
Shear Force, V2 = 3472.528  
Shear Force, V3 = 1.2919419E-013  
Axial Force, F = -4769.844  
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension: Aslt = 0.00  
-Compression: Aslc = 3053.628  
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension: Asl,ten = 1017.876  
-Compression: Asl,com = 1017.876  
-Middle: Asl,mid = 1017.876  
Mean Diameter of Tension Reinforcement, DbL = 18.00

Existing component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $\phi_{u,R} = \phi_u = 0.00931721$   
 $\phi_u = \phi_y + \phi_p = 0.00931721$

- Calculation of  $\phi_y$  -

$\phi_y = (M_y * L_s / 3) / E_{eff} = 0.00931721$  ((4.29), Biskinis Phd))  
 $M_y = 1.4766E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 1500.00  
From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 7.9240E+012$   
factor = 0.30  
 $A_g = 125663.706$   
 $f_c' = 20.00$   
 $N = 4769.844$   
 $E_c * I_g = 2.6413E+013$

#### Calculation of Yielding Moment $M_y$

Calculation of  $\phi_y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \min(M_{y\_ten}, M_{y\_com}) = 1.4766E+008$   
 $\phi_y = 8.1688007E-006$   
 $M_{y\_ten}$  (8c) = 1.4766E+008  
 $\phi_{y\_ten}$  (7c) = 72.40642  
error of function (7c) = 0.00117228  
 $M_{y\_com}$  (8d) = 3.7493E+008  
 $\phi_{y\_com}$  (7d) = 69.91126  
error of function (7d) = 0.00301342  
with ((10.1), ASCE 41-17)  $\phi_y = \min(\phi_y, 1.25 * \phi_{ey} * (I_b / I_d)^{2/3}) = 0.0022222$   
 $\phi_{ey} = 0.002$   
 $\phi_{apl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00157305$   
 $N = 4769.844$   
 $A_c = 125663.706$   
((10.1), ASCE 41-17)  $\phi_y = \min(\phi_y, 1.25 * \phi_{ey} * (I_b / I_d)^{2/3}) = 0.44757577$   
with  $f_c' * ((12.3), ACI 440) = 24.12975$   
 $f_c = 20.00$   
 $f_l = 1.3173$   
 $k = 1$   
Effective FRP thickness,  $t_f = NL * t * \cos(\beta_1) = 1.016$

$\phi_e((12.5) \text{ and } (12.7)) = 0.004$   
 $f_u = 0.01$   
 $E_f = 64828.00$

Calculation of ratio  $l_b/l_d$

Lap Length:  $l_d/l_{d,min} = 0.33490748$

$l_b = 300.00$

$l_d = 895.7698$

Calculation of  $\lambda$  according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $\lambda = 1$

$d_b = 18.00$

Mean strength value of all re-bars:  $f_y = 444.44$

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \pi/2 * \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

- Calculation of  $\rho$  -

From table 10-9:  $\rho = 0.00$

with:

- Columns controlled by inadequate development or splicing along the clear height because  $l_b/l_d < 1$

shear control ratio  $V_y E / V_{col} E = 0.30252729$

$d = 0.00$

$s = 0.00$

$t = 2 * A_v / (d_c * s) + 4 * t_f / D * (f_{fe} / f_s) = 0.00721126$

$A_v = 78.53982$ , is the area of the circular stirrup

$d_c = D - 2 * \text{cover} - \text{Hoop Diameter} = 340.00$

The term  $2 * t_f / b_w * (f_{fe} / f_s)$  is implemented to account for FRP contribution

where  $f = 2 * t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength

All these variables have already been given in Shear control ratio calculation.

$N_{UD} = 4769.844$

$A_g = 125663.706$

$f'_c E = 20.00$

$f_{yt} E = f_{yl} E = 444.44$

$\rho_l = \text{Area\_Tot\_Long\_Rein} / (A_g) = 0.0243$

$f'_c E = 20.00$

End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (b)

**Calculation No. 7**

column C1, Floor 1

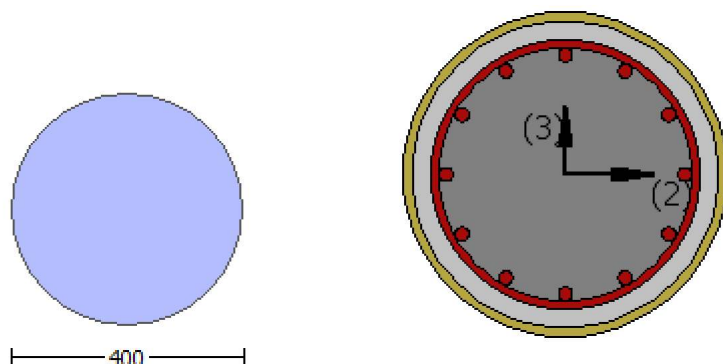
Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Shear capacity VRd

Edge: End

Local Axis: (3)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (b)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE 41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 16.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 400.00$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of  $\gamma$  for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE 41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE 41-17).

Existing material: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material: Steel Strength,  $f_s = f_{sm} = 444.44$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length  $l_o = l_b = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $\epsilon_{fu} = 0.01$

Number of directions,  $NoDir = 1$

Fiber orientations, bi: 0.00°  
Number of layers, NL = 1  
Radius of rounding corners, R = 40.00

#### Stepwise Properties

EDGE -A-  
Bending Moment, Ma = 4.7721810E-010  
Shear Force, Va = -1.2919419E-013  
EDGE -B-  
Bending Moment, Mb = -8.9325839E-011  
Shear Force, Vb = 1.2919419E-013  
BOTH EDGES  
Axial Force, F = -4769.844  
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension: Aslt = 0.00  
-Compression: Aslc = 3053.628  
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension: Asl,ten = 1017.876  
-Compression: Asl,com = 1017.876  
-Middle: Asl,mid = 1017.876  
Mean Diameter of Tension Reinforcement, DbL,ten = 18.00

Existing component: From table 7-7, ASCE 41\_17: Final Shear Capacity VR = \*Vn = 286337.204  
Vn ((10.3), ASCE 41-17) = knl\*VCol0 = 286337.204  
VCol = 286337.204  
knl = 1.00  
displacement\_ductility\_demand = 0.00

NOTE: In expression (10-3) 'Vs = Av\*fy\*d/s' is replaced by 'Vs+ f\*Vf'  
where Vf is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)  
fc' = 16.00, but  $fc'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
M/Vd = 2.00  
Mu = 8.9325839E-011  
Vu = 1.2919419E-013  
d = 0.8\*D = 320.00  
Nu = 4769.844  
Ag = 125663.706  
From (11.5.4.8), ACI 318-14: Vs = 0.00  
Av = /2\*A\_stirrup = 123370.055  
fy = 400.00  
s = 360.00  
Vs is multiplied by Col = 0.00  
s/d = 1.125  
Vf ((11-3)-(11.4), ACI 440) = 194961.134  
f = 0.95, for fully-wrapped sections  
wf/sf = 1 (FRP strips adjacent to one another).  
In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a)\sin a$  which is more a generalised expression,  
where is the angle of the crack direction (see KANEPE).  
This later relation, considered as a function Vf( , ), is implemented for every different fiber orientation ai,  
as well as for 2 crack directions, =45° and =-45° to take into consideration the cyclic seismic loading.  
orientation 1: 1 = b1 + 90° = 90.00  
Vf = Min(|Vf(45, 1)|, |Vf(-45,a1)|), with:  
total thickness per orientation, tf1 = NL\*t/NoDir = 1.016  
dfv = d (figure 11.2, ACI 440) = 370.00  
ffe ((11-5), ACI 440) = 259.312  
Ef = 64828.00  
fe = 0.004, from (11.6a), ACI 440  
with fu = 0.01  
From (11-11), ACI 440: Vs + Vf <= 213705.936  
bw\*d = \*d\*d/4 = 80424.772

displacement\_ductility\_demand is calculated as  $\delta / y$

- Calculation of  $\delta / y$  for END B -  
for rotation axis 2 and integ. section (b)

From analysis, chord rotation  $\theta = 1.1597918E-020$   
 $y = (M_y * L_s / 3) / E_{eff} = 0.00931721$  ((4.29), Biskinis Phd))  
 $M_y = 1.4766E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 1500.00  
From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 7.9240E+012$   
 $factor = 0.30$   
 $A_g = 125663.706$   
 $f_c' = 20.00$   
 $N = 4769.844$   
 $E_c * I_g = 2.6413E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\delta$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \min(M_{y\_ten}, M_{y\_com}) = 1.4766E+008$   
 $y = 8.1688007E-006$   
 $M_{y\_ten}$  (8c) = 1.4766E+008  
 $\delta_{ten}$  (7c) = 72.40642  
error of function (7c) = 0.00117228  
 $M_{y\_com}$  (8d) = 3.7493E+008  
 $\delta_{com}$  (7d) = 69.91126  
error of function (7d) = 0.00301342  
with ((10.1), ASCE 41-17)  $e_y = \min(e_y, 1.25 * e_y * (l_b / l_d)^{2/3}) = 0.0022222$   
 $e_{co} = 0.002$   
 $a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00157305$   
 $N = 4769.844$   
 $A_c = 125663.706$   
((10.1), ASCE 41-17)  $\phi = \min(\phi, 1.25 * \phi * (l_b / l_d)^{2/3}) = 0.44757577$   
with  $f_c' * ((12.3), ACI 440) = 24.12975$   
 $f_c = 20.00$   
 $f_l = 1.3173$   
 $k = 1$   
Effective FRP thickness,  $t_f = N L * t * \cos(\theta) = 1.016$   
 $e_{fe}$  ((12.5) and (12.7)) = 0.004  
 $f_u = 0.01$   
 $E_f = 64828.00$

Calculation of ratio  $l_b / l_d$

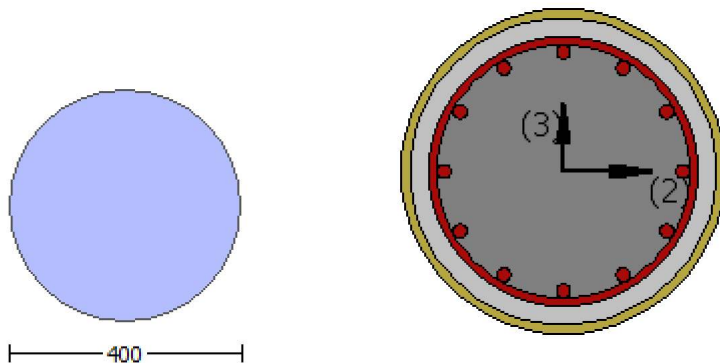
Lap Length:  $l_d / l_{d,min} = 0.33490748$   
 $l_b = 300.00$   
 $l_d = 895.7698$   
Calculation of  $l$  according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $\phi = 1$   
 $d_b = 18.00$   
Mean strength value of all re-bars:  $f_y = 444.44$   
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $c_b = 25.00$

$K_{tr} = 1.14232$   
 $A_{tr} = \frac{1}{2} * \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

-----  
 End Of Calculation of Shear Capacity for element: column CC1 of floor 1  
 At local axis: 3  
 Integration Section: (b)  
 -----

## Calculation No. 8

column C1, Floor 1  
 Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)  
 Analysis: Uniform +X  
 Check: Chord rotation capacity (  $\phi$  )  
 Edge: End  
 Local Axis: (3)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At Shear local axis: 3  
 (Bending local axis: 2)  
 Section Type: rccs

Constant Properties

-----  
 Knowledge Factor,  $\gamma = 1.00$   
 Mean strength values are used for both shear and moment calculations.  
 Consequently:  
 Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$   
 Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.44$   
 Concrete Elasticity,  $E_c = 21019.039$   
 Steel Elasticity,  $E_s = 200000.00$   
 #####  
 Note: Especially for the calculation of moment strengths,  
 the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14  
 Existing material: Steel Strength,  $f_s = 1.25 * f_{sm} = 555.55$   
 #####

Diameter,  $D = 400.00$   
 Cover Thickness,  $c = 25.00$   
 Mean Confinement Factor overall section = 1.84055  
 Element Length,  $L = 3000.00$   
 Primary Member  
 Ribbed Bars  
 Ductile Steel  
 Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Lap Length  $l_o = 300.00$   
 FRP Wrapping Data  
 Type: Carbon  
 Cured laminate properties (design values)  
 Thickness,  $t = 1.016$   
 Tensile Strength,  $f_{fu} = 1055.00$   
 Tensile Modulus,  $E_f = 64828.00$   
 Elongation,  $ε_{fu} = 0.01$   
 Number of directions,  $N_{oDir} = 1$   
 Fiber orientations,  $b_i: 0.00^\circ$   
 Number of layers,  $N_L = 1$   
 Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

At local axis: 3  
 EDGE -A-  
 Shear Force,  $V_a = 1.2107607E-031$   
 EDGE -B-  
 Shear Force,  $V_b = -1.2107607E-031$   
 BOTH EDGES  
 Axial Force,  $F = -4771.233$   
 Longitudinal Reinforcement Area Distribution (in 2 divisions)  
     -Tension:  $A_{sl,t} = 0.00$   
     -Compression:  $A_{sl,c} = 3053.628$   
 Longitudinal Reinforcement Area Distribution (in 3 divisions)  
     -Tension:  $A_{sl,ten} = 1017.876$   
     -Compression:  $A_{sl,com} = 1017.876$   
     -Middle:  $A_{sl,mid} = 1017.876$

Calculation of Shear Capacity ratio ,  $V_e/V_r = 0.30252729$   
 Member Controlled by Flexure ( $V_e/V_r < 1$ )  
 Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 90202.132$   
 with  
 $M_{pr1} = \text{Max}(Mu_{1+}, Mu_{1-}) = 1.3530E+008$   
 $Mu_{1+} = 1.3530E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination  
 $Mu_{1-} = 1.3530E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination  
 $M_{pr2} = \text{Max}(Mu_{2+}, Mu_{2-}) = 1.3530E+008$   
 $Mu_{2+} = 1.3530E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination  
 $Mu_{2-} = 1.3530E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

#### Calculation of $Mu_{1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$   
 $M_u = 1.3530E+008$   
 $= 0.90757121$

$\lambda = 0.80580716$   
 error of function (3.68), Biskinis Phd = 28928.286  
 From 5A.2, TBDY:  $f_{cc} = f_c \cdot \lambda = 36.81095$   
 conf. factor  $\lambda = 1.84055$   
 $f_c = 20.00$   
 From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

#### Calculation of ratio $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
 Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $= 1$   
 $d_b = 18.00$   
 Mean strength value of all re-bars:  $f_y = 555.55$   
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $c_b = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

#### Calculation of $\mu_{u1}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$   
 $\mu_u = 1.3530E+008$

$= 0.90757121$   
 $\lambda = 0.80580716$   
 error of function (3.68), Biskinis Phd = 28928.286  
 From 5A.2, TBDY:  $f_{cc} = f_c \cdot \lambda = 36.81095$   
 conf. factor  $\lambda = 1.84055$   
 $f_c = 20.00$   
 From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

#### Calculation of ratio $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
 Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $= 1$   
 $d_b = 18.00$



Mean strength value of all re-bars:  $f_y = 555.55$   
 $f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \frac{1}{2} \times \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

#### Calculation of $\mu_{2+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 1.3530E+008$

$\phi = 0.90757121$   
 $\phi' = 0.80580716$   
error of function (3.68), Biskinis Phd = 28928.286  
From 5A.2, TBDY:  $f_{cc} = f'_c \times c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \times \text{Min}(1, 1.25 \times (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $\phi = \phi' \times \text{Min}(1, 1.25 \times (l_b/l_d)^{2/3}) = 0.28805051$

#### Calculation of ratio $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $\phi = 1$   
 $db = 18.00$   
Mean strength value of all re-bars:  $f_y = 555.55$   
 $f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \frac{1}{2} \times \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

#### Calculation of $\mu_{2-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 1.3530E+008$

$\phi = 0.90757121$   
 $\phi' = 0.80580716$   
error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY:  $f_{cc} = f_c' \cdot c = 36.81095$   
 conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
 From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio  $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
 Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $= 1$   
 $d_b = 18.00$   
 Mean strength value of all re-bars:  $f_y = 555.55$   
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 298161.965$

Calculation of Shear Strength at edge 1,  $V_{r1} = 298161.965$   
 $V_{r1} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{Col0}$   
 $V_{Col0} = 298161.965$   
 $k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)  
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $\mu_u = 1.2899604E-011$   
 $\nu_u = 1.2107607E-031$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 0.00$   
 $A_v = \frac{1}{2} \cdot A_{stirrup} = 123370.055$   
 $f_y = 444.44$   
 $s = 360.00$   
 $V_s$  is multiplied by  $Col = 0.00$   
 $s/d = 1.125$   
 $V_f$  ((11-3)-(11.4), ACI 440) =  $194961.134$   
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot) \sin \alpha$  which is more a generalised expression, where  $\alpha$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\alpha = 45^\circ$  and  $\alpha = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\alpha_1 = b_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, a1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}((11-5), \text{ACI 440}) = 259.312$   
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $b_w \cdot d = \frac{1}{4} \cdot d \cdot d = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 298161.965$   
 $V_{r2} = V_{Col}((10.3), \text{ASCE 41-17}) = k_{nl} \cdot V_{Col0}$   
 $V_{Col0} = 298161.965$   
 $k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ '  
 where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)  
 $f'_c = 20.00$ , but  $f'_c^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $\mu_u = 1.2899604E-011$   
 $\nu_u = 1.2107607E-031$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 0.00$   
 $A_v = \frac{1}{2} \cdot A_{stirrup} = 123370.055$   
 $f_y = 444.44$   
 $s = 360.00$   
 $V_s$  is multiplied by  $\text{Col} = 0.00$   
 $s/d = 1.125$   
 $V_f((11-3)-(11.4), \text{ACI 440}) = 194961.134$   
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = \theta_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, a1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}((11-5), \text{ACI 440}) = 259.312$   
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $b_w \cdot d = \frac{1}{4} \cdot d \cdot d = 80424.772$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At local axis: 3

Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At Shear local axis: 2  
 (Bending local axis: 3)  
 Section Type: rccs

Constant Properties

Knowledge Factor,  $\phi = 1.00$   
 Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.44$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 555.55$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.84055

Element Length,  $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length  $l_o = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $\epsilon_{fu} = 0.01$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i: 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

At local axis: 2

EDGE -A-

Shear Force,  $V_a = -7.4135260E-048$

EDGE -B-

Shear Force,  $V_b = 7.4135260E-048$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{sl,t} = 0.00$

-Compression:  $A_{sl,c} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{sl,ten} = 1017.876$

-Compression:  $A_{sl,com} = 1017.876$

-Middle:  $A_{sl,mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.30252729$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 90202.132$

with

$M_{pr1} = \max(\mu_{u1+}, \mu_{u1-}) = 1.3530E+008$

$\mu_{u1+} = 1.3530E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$\mu_{u1-} = 1.3530E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \max(\mu_{u2+}, \mu_{u2-}) = 1.3530E+008$

$\mu_{u2+} = 1.3530E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination

$\mu_{u2-} = 1.3530E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

#### Calculation of Mu1+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.3530E+008

= 0.90757121  
' = 0.80580716  
error of function (3.68), Biskinis Phd = 28928.286  
From 5A.2, TBDY:  $f_{cc} = f_c^* c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y * \text{Min}(1, 1.25 * (l_b / l_d)^{2/3}) = 288.6089$   
 $l_b / l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
=  $*\text{Min}(1, 1.25 * (l_b / l_d)^{2/3}) = 0.28805051$

#### Calculation of ratio lb/l\_d

Lap Length:  $l_b / l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
= 1  
 $db = 18.00$   
Mean strength value of all re-bars:  $f_y = 555.55$   
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \sqrt{2} * \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

#### Calculation of Mu1-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.3530E+008

= 0.90757121  
' = 0.80580716  
error of function (3.68), Biskinis Phd = 28928.286  
From 5A.2, TBDY:  $f_{cc} = f_c^* c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y * \text{Min}(1, 1.25 * (l_b / l_d)^{2/3}) = 288.6089$   
 $l_b / l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
=  $*\text{Min}(1, 1.25 * (l_b / l_d)^{2/3}) = 0.28805051$

#### Calculation of ratio $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of  $l_b$ ,min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$db = 18.00$

Mean strength value of all re-bars:  $f_y = 555.55$

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$cb = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \times \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

#### Calculation of $\mu_{2+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$\mu = 1.3530E+008$

$= 0.90757121$

$' = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY:  $f_{cc} = f'_c \times c = 36.81095$

conf. factor  $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \times \text{Min}(1, 1.25 \times (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \text{Min}(1, 1.25 \times (l_b/l_d)^{2/3}) = 0.28805051$

#### Calculation of ratio $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of  $l_b$ ,min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$db = 18.00$

Mean strength value of all re-bars:  $f_y = 555.55$

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$cb = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \times \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

## Calculation of Mu2-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.3530E+008

= 0.90757121  
' = 0.80580716  
error of function (3.68), Biskinis Phd = 28928.286  
From 5A.2, TBDY:  $f_{cc} = f_c' \cdot c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
=  $\cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

## Calculation of ratio $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
= 1  
 $d_b = 18.00$   
Mean strength value of all re-bars:  $f_y = 555.55$   
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $c_b = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

## Calculation of Shear Strength $V_r = \text{Min}(V_{r1}, V_{r2}) = 298161.965$

Calculation of Shear Strength at edge 1,  $V_{r1} = 298161.965$   
 $V_{r1} = V_{CoI} \text{ ((10.3), ASCE 41-17)} = k_{nl} \cdot V_{CoI0}$   
 $V_{CoI0} = 298161.965$   
 $k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ '  
where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)  
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $M_u = 5.9321525E-012$   
 $V_u = 7.4135260E-048$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
From (11.5.4.8), ACI 318-14:  $V_s = 0.00$   
 $A_v = \frac{1}{2} \cdot A_{\text{stirrup}} = 123370.055$   
 $f_y = 444.44$   
 $s = 360.00$

$V_s$  is multiplied by  $\text{Col} = 0.00$   
 $s/d = 1.125$   
 $V_f ((11-3)-(11.4), \text{ACI } 440) = 194961.134$   
 $f = 0.95$ , for fully-wrapped sections  
 $wf/sf = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a)\sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f( , )$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $= 45^\circ$  and  $= -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta_1 = \theta_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, \theta_1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe} ((11-5), \text{ACI } 440) = 259.312$   
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $b_w \cdot d = \rho \cdot d^2 / 4 = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 298161.965$   
 $V_{r2} = V_{\text{Col}} ((10.3), \text{ASCE } 41-17) = k_{nl} \cdot V_{\text{Col}0}$   
 $V_{\text{Col}0} = 298161.965$   
 $k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ '  
 where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\rho = 1$  (normal-weight concrete)  
 $f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $\mu_u = 5.9321525\text{E-}012$   
 $\nu_u = 7.4135260\text{E-}048$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 0.00$   
 $A_v = \rho / 2 \cdot A_{\text{stirrup}} = 123370.055$   
 $f_y = 444.44$   
 $s = 360.00$

$V_s$  is multiplied by  $\text{Col} = 0.00$   
 $s/d = 1.125$   
 $V_f ((11-3)-(11.4), \text{ACI } 440) = 194961.134$   
 $f = 0.95$ , for fully-wrapped sections  
 $wf/sf = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a)\sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f( , )$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $= 45^\circ$  and  $= -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta_1 = \theta_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, \theta_1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe} ((11-5), \text{ACI } 440) = 259.312$   
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $b_w \cdot d = \rho \cdot d^2 / 4 = 80424.772$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At local axis: 2



Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (b)

Section Type: rccs

#### Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE 41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.44$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length  $l_b = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $\epsilon_{fu} = 0.01$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

Bending Moment,  $M = 0.09265189$

Shear Force,  $V_2 = 3472.528$

Shear Force,  $V_3 = 1.2919419E-013$

Axial Force,  $F = -4769.844$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{st} = 0.00$

-Compression:  $A_{sc} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{st,ten} = 1017.876$

-Compression:  $A_{st,com} = 1017.876$

-Middle:  $A_{st,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $D_bL = 18.00$

Existing component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_{R} = \gamma \cdot u = 0.00186344$

$u = \gamma \cdot u = 0.00186344$

- Calculation of  $\gamma$  -

$\gamma = (M_y \cdot L_s / 3) / E_{eff} = 0.00186344 ((4.29), \text{Biskinis Phd})$

$M_y = 1.4766E+008$

$L_s = M/V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) = 300.00

From table 10.5, ASCE 41\_17:  $E_{eff} = \text{factor} \cdot E_c \cdot I_g = 7.9240E+012$

factor = 0.30

$A_g = 125663.706$   
 $f_c' = 20.00$   
 $N = 4769.844$   
 $E_c I_g = 2.6413E+013$

#### Calculation of Yielding Moment $M_y$

Calculation of  $\rho_y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \min(M_{y\_ten}, M_{y\_com}) = 1.4766E+008$   
 $\rho_y = 8.1688007E-006$   
 $M_{y\_ten} (8c) = 1.4766E+008$   
 $\rho_{y\_ten} (7c) = 72.40642$   
error of function (7c) = 0.00117228  
 $M_{y\_com} (8d) = 3.7493E+008$   
 $\rho_{y\_com} (7d) = 69.91126$   
error of function (7d) = 0.00301342  
with ((10.1), ASCE 41-17)  $\rho_y = \min(\rho_y, 1.25 \cdot \rho_y \cdot (l_b/l_d)^{2/3}) = 0.0022222$   
 $\rho_{eco} = 0.002$   
 $\rho_{apl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00157305$   
 $N = 4769.844$   
 $A_c = 125663.706$   
((10.1), ASCE 41-17)  $\rho_y = \min(\rho_y, 1.25 \cdot \rho_y \cdot (l_b/l_d)^{2/3}) = 0.44757577$   
with  $f_c' ((12.3), ACI 440) = 24.12975$   
 $f_c = 20.00$   
 $f_l = 1.3173$   
 $k = 1$   
Effective FRP thickness,  $t_f = N L \cdot t \cdot \cos(b_1) = 1.016$   
 $\rho_{efe} ((12.5) \text{ and } (12.7)) = 0.004$   
 $f_u = 0.01$   
 $E_f = 64828.00$

#### Calculation of ratio $l_b/l_d$

Lap Length:  $l_d/l_{d,min} = 0.33490748$   
 $l_b = 300.00$   
 $l_d = 895.7698$   
Calculation of  $l$  according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $\rho = 1$   
 $d_b = 18.00$   
Mean strength value of all re-bars:  $f_y = 444.44$   
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $c_b = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \sqrt{2} \cdot \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

#### - Calculation of $\rho_p$ -

From table 10-9:  $\rho_p = 0.00$

with:

- Columns controlled by inadequate development or splicing along the clear height because  $l_b/l_d < 1$
- shear control ratio  $V_y E / V_{co} I_{OE} = 0.30252729$
- $d = 0.00$
- $s = 0.00$

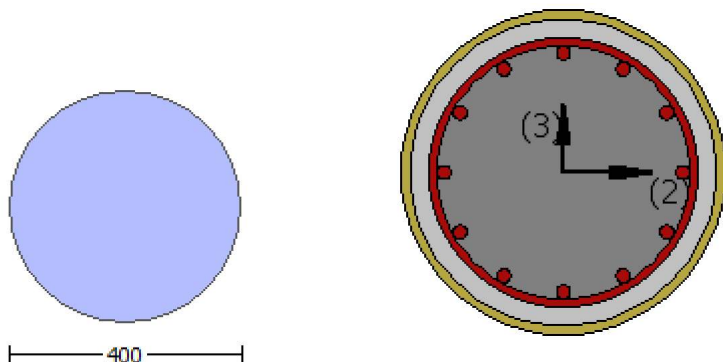
$t = 2 \cdot A_v / (d_c \cdot s) + 4 \cdot t_f / D \cdot (f_{fe} / f_s) = 0.00721126$   
 $A_v = 78.53982$ , is the area of the circular stirrup  
 $d_c = D - 2 \cdot \text{cover} - \text{Hoop Diameter} = 340.00$   
 The term  $2 \cdot t_f / b_w \cdot (f_{fe} / f_s)$  is implemented to account for FRP contribution  
 where  $f = 2 \cdot t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength  
 All these variables have already been given in Shear control ratio calculation.  
 $NUD = 4769.844$   
 $Ag = 125663.706$   
 $f_{cE} = 20.00$   
 $f_{ytE} = f_{ylE} = 444.44$   
 $pl = \text{Area\_Tot\_Long\_Rein} / (Ag) = 0.0243$   
 $f_{cE} = 20.00$

-----  
 End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1  
 At local axis: 3  
 Integration Section: (b)

-----

Calculation No. 9

column C1, Floor 1  
 Limit State: Collapse Prevention (data interpolation between analysis steps 2 and 3)  
 Analysis: Uniform +X  
 Check: Shear capacity VRd  
 Edge: Start  
 Local Axis: (2)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1  
 At local axis: 2  
 Integration Section: (a)  
 Section Type: rccs  
 Constant Properties  
 -----  
 Knowledge Factor,  $\gamma = 1.00$   
 Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE 41-17.  
 Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 16.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 400.00$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of  $\gamma$  for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE 41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE 41-17).

Existing material: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material: Steel Strength,  $f_s = f_{sm} = 444.44$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length  $l_o = l_b = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $ef_u = 0.01$

Number of directions,  $NoDir = 1$

Fiber orientations,  $bi: 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

EDGE -A-

Bending Moment,  $M_a = -1.3471E+007$

Shear Force,  $V_a = -4486.792$

EDGE -B-

Bending Moment,  $M_b = 2684.487$

Shear Force,  $V_b = 4486.792$

BOTH EDGES

Axial Force,  $F = -4783.229$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 1272.345$

-Compression:  $As_c = 1781.283$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t,ten} = 1017.876$

-Compression:  $As_{l,com} = 1017.876$

-Middle:  $As_{l,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $Db_{L,ten} = 18.00$

Existing component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = \phi V_n = 235776.467$

$V_n$  ((10.3), ASCE 41-17) =  $k_n \phi V_{Col0} = 235776.467$

$V_{Col} = 235776.467$

$k_n = 1.00$

$displacement\_ductility\_demand = 0.02901189$

NOTE: In expression (10-3) ' $V_s = A_v f_y d/s$ ' is replaced by ' $V_s + \phi V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\phi = 1$  (normal-weight concrete)

$f'_c = 16.00$ , but  $f'_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 4.00$   
 $\mu_u = 1.3471E+007$   
 $V_u = 4486.792$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4783.229$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 0.00$   
 $A_v = \frac{1}{2} \cdot A_{stirrup} = 123370.055$   
 $f_y = 400.00$   
 $s = 360.00$   
 $V_s$  is multiplied by  $\text{Col} = 0.00$   
 $s/d = 1.125$   
 $V_f ((11-3)-(11.4), \text{ACI 440}) = 194961.134$   
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = b1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, a1)|)$ , with:  
 total thickness per orientation,  $tf1 = N_L \cdot t / \text{NoDir} = 1.016$   
 $df_v = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe} ((11-5), \text{ACI 440}) = 259.312$   
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 213705.936$   
 $bw \cdot d = \frac{1}{4} \cdot d \cdot d = 80424.772$

displacement ductility demand is calculated as  $\delta / y$

- Calculation of  $\delta / y$  for END A -  
 for rotation axis 3 and integ. section (a)

From analysis, chord rotation  $\theta = 0.00054105$   
 $y = (M_y \cdot L_s / 3) / E_{eff} = 0.01864917$  ((4.29), Biskinis Phd))  
 $M_y = 1.4766E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) = 3002.334  
 From table 10.5, ASCE 41\_17:  $E_{eff} = \text{factor} \cdot E_c \cdot I_g = 7.9240E+012$   
 $\text{factor} = 0.30$   
 $A_g = 125663.706$   
 $f'_c = 20.00$   
 $N = 4783.229$   
 $E_c \cdot I_g = 2.6413E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\delta$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 1.4766E+008$   
 $y = 8.1688588E-006$   
 $M_{y\_ten} (8c) = 1.4766E+008$   
 $\delta_{ten} (7c) = 72.4069$   
 error of function (7c) = 0.00117279  
 $M_{y\_com} (8d) = 3.7493E+008$   
 $\delta_{com} (7d) = 69.91139$   
 error of function (7d) = 0.00300901  
 with ((10.1), ASCE 41-17)  $e_y = \text{Min}(e_y, 1.25 \cdot e_y \cdot (I_b/I_d)^{2/3}) = 0.0022222$   
 $e_{co} = 0.002$   
 $a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d1 = 44.00$   
 $R = 200.00$

$v = 0.00157746$   
 $N = 4783.229$   
 $A_c = 125663.706$   
 $((10.1), ASCE 41-17) = \text{Min}( , 1.25 * ((lb/ld)^{2/3}) = 0.44757577$   
 with  $f_c' ((12.3), ACI 440) = 24.12975$   
 $f_c = 20.00$   
 $f_l = 1.3173$   
 $k = 1$   
 Effective FRP thickness,  $t_f = NL * t * \cos(b_1) = 1.016$   
 $e_{fe} ((12.5) \text{ and } (12.7)) = 0.004$   
 $f_u = 0.01$   
 $E_f = 64828.00$

Calculation of ratio  $lb/ld$

Lap Length:  $ld/ld, \min = 0.33490748$

$lb = 300.00$

$ld = 895.7698$

Calculation of  $l$  according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$db = 18.00$

Mean strength value of all re-bars:  $f_y = 444.44$

$f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$cb = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \pi/2 * \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

End Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (a)

## Calculation No. 10

column C1, Floor 1

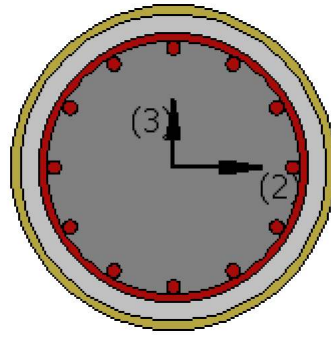
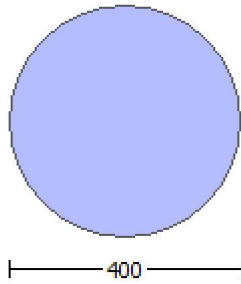
Limit State: Collapse Prevention (data interpolation between analysis steps 2 and 3)

Analysis: Uniform +X

Check: Chord rotation capacity (  $\phi$  )

Edge: Start

Local Axis: (2)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.44$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 555.55$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.84055

Element Length,  $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length  $l_o = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $\epsilon_{fu} = 0.01$

Number of directions,  $NoDir = 1$

Fiber orientations,  $b_i: 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

Stepwise Properties

At local axis: 3

EDGE -A-

Shear Force,  $V_a = 1.2107607E-031$

EDGE -B-

Shear Force,  $V_b = -1.2107607E-031$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $Asl_t = 0.00$   
 -Compression:  $Asl_c = 3053.628$   
 Longitudinal Reinforcement Area Distribution (in 3 divisions)  
 -Tension:  $Asl_{ten} = 1017.876$   
 -Compression:  $Asl_{com} = 1017.876$   
 -Middle:  $Asl_{mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.30252729$   
 Member Controlled by Flexure ( $V_e/V_r < 1$ )  
 Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 90202.132$   
 with  
 $M_{pr1} = \max(\mu_{u1+}, \mu_{u1-}) = 1.3530E+008$   
 $\mu_{u1+} = 1.3530E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination  
 $\mu_{u1-} = 1.3530E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination  
 $M_{pr2} = \max(\mu_{u2+}, \mu_{u2-}) = 1.3530E+008$   
 $\mu_{u2+} = 1.3530E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination  
 $\mu_{u2-} = 1.3530E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

Calculation of  $\mu_{u1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$   
 $\mu_u = 1.3530E+008$

$\phi = 0.90757121$   
 $\phi' = 0.80580716$   
 error of function (3.68), Biskinis Phd = 28928.286  
 From 5A.2, TBDY:  $f_{cc} = f_c' \cdot c = 36.81095$   
 conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
 From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \min(1, 1.25 \cdot (l_b/d)^{2/3}) = 288.6089$   
 $l_b/d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \phi' \cdot \min(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.28805051$

Calculation of ratio  $l_b/d$

Lap Length:  $l_b/d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
 Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $\phi = 1$   
 $\phi_b = 18.00$   
 Mean strength value of all re-bars:  $f_y = 555.55$   
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \pi \cdot \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$



Calculation of Mu1-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.3530E+008

= 0.90757121  
' = 0.80580716  
error of function (3.68), Biskinis Phd = 28928.286  
From 5A.2, TDY:  $f_{cc} = f_c \cdot c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
=  $\cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio  $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
= 1  
 $db = 18.00$   
Mean strength value of all re-bars:  $f_y = 555.55$   
 $f_c' = 20.00$ , but  $f_c^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \sqrt{2} \cdot \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

Calculation of Mu2+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.3530E+008

= 0.90757121  
' = 0.80580716  
error of function (3.68), Biskinis Phd = 28928.286  
From 5A.2, TDY:  $f_{cc} = f_c \cdot c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.28805051$$

Calculation of ratio  $l_b/d$

Lap Length:  $l_b/d = 0.26792599$

$l_b = 300.00$

$d = 1119.712$

Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$d_b = 18.00$

Mean strength value of all re-bars:  $f_y = 555.55$

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of  $\mu_2$ -

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$\mu = 1.3530E+008$

$= 0.90757121$

$' = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY:  $f_{cc} = f'_c \cdot c = 36.81095$

conf. factor  $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 288.6089$

$l_b/d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.28805051$

Calculation of ratio  $l_b/d$

Lap Length:  $l_b/d = 0.26792599$

$l_b = 300.00$

$d = 1119.712$

Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$d_b = 18.00$

Mean strength value of all re-bars:  $f_y = 555.55$

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 298161.965$

Calculation of Shear Strength at edge 1,  $V_{r1} = 298161.965$

$V_{r1} = V_{Col} \text{ ((10.3), ASCE 41-17)} = k_{nl} * V_{Col0}$

$V_{Col0} = 298161.965$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\lambda = 1$  (normal-weight concrete)

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.2899604E-011$

$\nu_u = 1.2107607E-031$

$d = 0.8 * D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 0.00$

$A_v = \sqrt{2} * A_{\text{stirrup}} = 123370.055$

$f_y = 444.44$

$s = 360.00$

$V_s$  is multiplied by  $\text{Col} = 0.00$

$s/d = 1.125$

$V_f \text{ ((11-3)-(11.4), ACI 440)} = 194961.134$

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression, where  $a$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta)$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta_1 = b_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, a_1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L * t / N_{\text{Dir}} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe} \text{ ((11-5), ACI 440)} = 259.312$

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$

$b_w * d = \sqrt{2} * d^2 / 4 = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 298161.965$

$V_{r2} = V_{Col} \text{ ((10.3), ASCE 41-17)} = k_{nl} * V_{Col0}$

$V_{Col0} = 298161.965$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\lambda = 1$  (normal-weight concrete)

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.2899604E-011$

$\nu_u = 1.2107607E-031$

$d = 0.8 * D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 0.00$

$A_v = \sqrt{2} * A_{\text{stirrup}} = 123370.055$

$f_y = 444.44$

$s = 360.00$

$V_s$  is multiplied by  $\text{Col} = 0.00$

$s/d = 1.125$

$V_f((11-3)-(11.4), \text{ACI 440}) = 194961.134$   
 $f = 0.95$ , for fully-wrapped sections  
 $wf/sf = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a)\sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta_1 = \theta_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, \theta_1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}((11-5), \text{ACI 440}) = 259.312$   
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $b_w \cdot d = \frac{V_s \cdot d}{4} = 80424.772$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At local axis: 3

Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At Shear local axis: 2  
 (Bending local axis: 3)  
 Section Type: rccs

#### Constant Properties

Knowledge Factor,  $\phi = 1.00$   
 Mean strength values are used for both shear and moment calculations.  
 Consequently:  
 Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$   
 Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.44$   
 Concrete Elasticity,  $E_c = 21019.039$   
 Steel Elasticity,  $E_s = 200000.00$   
 #####  
 Note: Especially for the calculation of moment strengths,  
 the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14  
 Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 555.55$   
 #####  
 Diameter,  $D = 400.00$   
 Cover Thickness,  $c = 25.00$   
 Mean Confinement Factor overall section = 1.84055  
 Element Length,  $L = 3000.00$   
 Primary Member  
 Ribbed Bars  
 Ductile Steel  
 Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Lap Length  $l_o = 300.00$   
 FRP Wrapping Data  
 Type: Carbon  
 Cured laminate properties (design values)  
 Thickness,  $t = 1.016$   
 Tensile Strength,  $f_{fu} = 1055.00$   
 Tensile Modulus,  $E_f = 64828.00$   
 Elongation,  $e_{fu} = 0.01$   
 Number of directions,  $\text{NoDir} = 1$   
 Fiber orientations,  $b_i: 0.00^\circ$   
 Number of layers,  $NL = 1$   
 Radius of rounding corners,  $R = 40.00$

## Stepwise Properties

At local axis: 2

EDGE -A-

Shear Force,  $V_a = -7.4135260E-048$

EDGE -B-

Shear Force,  $V_b = 7.4135260E-048$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 0.00$

-Compression:  $As_c = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t,ten} = 1017.876$

-Compression:  $As_{c,com} = 1017.876$

-Middle:  $As_{l,mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.30252729$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 90202.132$   
with

$M_{pr1} = \text{Max}(Mu_{1+}, Mu_{1-}) = 1.3530E+008$

$Mu_{1+} = 1.3530E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$Mu_{1-} = 1.3530E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(Mu_{2+}, Mu_{2-}) = 1.3530E+008$

$Mu_{2+} = 1.3530E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination

$Mu_{2-} = 1.3530E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

Calculation of  $Mu_{1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$

$M_u = 1.3530E+008$

$= 0.90757121$

$' = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY:  $f_{cc} = f_c^* \quad c = 36.81095$

conf. factor  $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y * \text{Min}(1, 1.25 * (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$Ac = 125663.706$

$= * \text{Min}(1, 1.25 * (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio  $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$db = 18.00$

Mean strength value of all re-bars:  $f_y = 555.55$   
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \frac{1}{2} \times \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

#### Calculation of $\mu_1$ -

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 1.3530E+008$

$\phi = 0.90757121$   
 $\phi' = 0.80580716$   
error of function (3.68), Biskinis Phd = 28928.286  
From 5A.2, TBDY:  $f_{cc} = f_c \times c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \times \text{Min}(1, 1.25 \times (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $\phi = \phi' \times \text{Min}(1, 1.25 \times (l_b/l_d)^{2/3}) = 0.28805051$

#### Calculation of ratio $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $\phi = 1$

$db = 18.00$   
Mean strength value of all re-bars:  $f_y = 555.55$   
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \frac{1}{2} \times \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

#### Calculation of $\mu_2$ +

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 1.3530E+008$

$\phi = 0.90757121$   
 $\phi' = 0.80580716$   
error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY:  $f_{cc} = f_c' \cdot c = 36.81095$   
 conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
 From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio  $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
 Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $= 1$   
 $db = 18.00$   
 Mean strength value of all re-bars:  $f_y = 555.55$   
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

Calculation of  $\mu_2$ -

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 1.3530E+008$

$= 0.90757121$   
 $' = 0.80580716$   
 error of function (3.68), Biskinis Phd = 28928.286  
 From 5A.2, TBDY:  $f_{cc} = f_c' \cdot c = 36.81095$   
 conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
 From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio  $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
 Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $= 1$   
 $db = 18.00$   
 Mean strength value of all re-bars:  $f_y = 555.55$   
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)

$t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \frac{1}{2} \times \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 298161.965$

Calculation of Shear Strength at edge 1,  $V_{r1} = 298161.965$

$V_{r1} = V_{Col} \text{ ((10.3), ASCE 41-17)} = k_{nl} \times V_{Col0}$

$V_{Col0} = 298161.965$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \times f_y \times d / s$ ' is replaced by ' $V_s + f \times V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 5.9321525E-012$

$\mu_v = 7.4135260E-048$

$d = 0.8 \times D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 0.00$

$A_v = \frac{1}{2} \times A_{\text{stirrup}} = 123370.055$

$f_y = 444.44$

$s = 360.00$

$V_s$  is multiplied by  $\text{Col} = 0.00$

$s/d = 1.125$

$V_f \text{ ((11-3)-(11.4), ACI 440)} = 194961.134$

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression, where  $a$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = b1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, a1)|)$ , with:

total thickness per orientation,  $tf1 = N_L \times t / N_{\text{Dir}} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe} \text{ ((11-5), ACI 440)} = 259.312$

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$

$b_w \times d = \frac{1}{4} \times d \times d = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 298161.965$

$V_{r2} = V_{Col} \text{ ((10.3), ASCE 41-17)} = k_{nl} \times V_{Col0}$

$V_{Col0} = 298161.965$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \times f_y \times d / s$ ' is replaced by ' $V_s + f \times V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 5.9321525E-012$



$V_u = 7.4135260E-048$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 0.00$   
 $A_v = \frac{1}{2} \cdot A_{\text{stirrup}} = 123370.055$   
 $f_y = 444.44$   
 $s = 360.00$   
 $V_s$  is multiplied by  $\text{Col} = 0.00$   
 $s/d = 1.125$   
 $V_f ((11-3)-(11.4), \text{ACI } 440) = 194961.134$   
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta_1 = \theta_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, a_1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = N_L \cdot t / \text{NoDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe} ((11-5), \text{ACI } 440) = 259.312$   
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $b_w \cdot d = \frac{1}{4} \cdot d \cdot d = 80424.772$

-----  
 End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At local axis: 2  
 -----

Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1  
 At local axis: 2  
 Integration Section: (a)  
 Section Type: rccs

Constant Properties

-----  
 Knowledge Factor,  $\gamma = 1.00$   
 Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE 41-17.  
 Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17  
 Consequently:  
 Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$   
 Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.44$   
 Concrete Elasticity,  $E_c = 21019.039$   
 Steel Elasticity,  $E_s = 200000.00$   
 Diameter,  $D = 400.00$   
 Cover Thickness,  $c = 25.00$   
 Element Length,  $L = 3000.00$   
 Primary Member  
 Ribbed Bars  
 Ductile Steel  
 Without Detailing for Earthquake Resistance (including stirrups not closed at  $135^\circ$ )  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Lap Length  $l_b = 300.00$   
 FRP Wrapping Data  
 Type: Carbon  
 Cured laminate properties (design values)  
 Thickness,  $t = 1.016$   
 Tensile Strength,  $f_{fu} = 1055.00$   
 Tensile Modulus,  $E_f = 64828.00$   
 Elongation,  $e_{fu} = 0.01$   
 Number of directions,  $\text{NoDir} = 1$

Fiber orientations, bi: 0.00°  
Number of layers, NL = 1  
Radius of rounding corners, R = 40.00

#### Stepwise Properties

Bending Moment, M = 6.2300792E-010  
Shear Force, V2 = -4486.792  
Shear Force, V3 = -1.7064698E-013  
Axial Force, F = -4783.229  
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension: Aslt = 1272.345  
-Compression: Aslc = 1781.283  
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension: Asl,ten = 1017.876  
-Compression: Asl,com = 1017.876  
-Middle: Asl,mid = 1017.876  
Mean Diameter of Tension Reinforcement, DbL = 18.00

Existing component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_R = \gamma + p = 0.05131733$

- Calculation of  $\gamma$  -

$\gamma = (M_y * L_s / 3) / E_{eff} = 0.00931733$  ((4.29), Biskinis Phd))  
 $M_y = 1.4766E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 1500.00  
From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 7.9240E+012$   
factor = 0.30  
 $A_g = 125663.706$   
 $f_c' = 20.00$   
 $N = 4783.229$   
 $E_c * I_g = 2.6413E+013$

#### Calculation of Yielding Moment $M_y$

Calculation of  $\gamma$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \min(M_{y\_ten}, M_{y\_com}) = 1.4766E+008$   
 $\gamma = 8.1688588E-006$   
 $M_{y\_ten}$  (8c) = 1.4766E+008  
 $\gamma_{ten}$  (7c) = 72.4069  
error of function (7c) = 0.00117279  
 $M_{y\_com}$  (8d) = 3.7493E+008  
 $\gamma_{com}$  (7d) = 69.91139  
error of function (7d) = 0.00300901  
with ((10.1), ASCE 41-17)  $e_y = \min(e_y, 1.25 * e_y * (I_b / I_d)^{2/3}) = 0.0022222$   
 $e_{co} = 0.002$   
 $a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00157746$   
 $N = 4783.229$   
 $A_c = 125663.706$   
((10.1), ASCE 41-17)  $e_y = \min(e_y, 1.25 * e_y * (I_b / I_d)^{2/3}) = 0.44757577$   
with  $f_c' = 20.00$  ((12.3), ACI 440) = 24.12975  
 $f_l = 1.3173$   
 $k = 1$   
Effective FRP thickness,  $t_f = NL * t * \cos(b_1) = 1.016$

$\phi_e((12.5) \text{ and } (12.7)) = 0.004$   
 $f_u = 0.01$   
 $E_f = 64828.00$

Calculation of ratio  $l_b/l_d$

Lap Length:  $l_d/l_{d,min} = 0.33490748$

$l_b = 300.00$

$l_d = 895.7698$

Calculation of  $\lambda$  according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $\lambda = 1$

$d_b = 18.00$

Mean strength value of all re-bars:  $f_y = 444.44$

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \times \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

- Calculation of  $\rho$  -

From table 10-9:  $\rho = 0.042$

with:

- Columns controlled by inadequate development or splicing along the clear height because  $l_b/l_d < 1$

shear control ratio  $V_y E / V_{col} E = 0.30252729$

$d = 0.00$

$s = 0.00$

$t = 2 \cdot A_v / (d_c \cdot s) + 4 \cdot t_f / D \cdot (f_{fe} / f_s) = 0.00721126$

$A_v = 78.53982$ , is the area of the circular stirrup

$d_c = D - 2 \cdot \text{cover} - \text{Hoop Diameter} = 340.00$

The term  $2 \cdot t_f / b_w \cdot (f_{fe} / f_s)$  is implemented to account for FRP contribution

where  $f = 2 \cdot t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength

All these variables have already been given in Shear control ratio calculation.

$N_{UD} = 4783.229$

$A_g = 125663.706$

$f'_c E = 20.00$

$f_{yt} E = f_{yl} E = 444.44$

$\rho_l = \text{Area\_Tot\_Long\_Rein} / (A_g) = 0.0243$

$f'_c E = 20.00$

End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (a)

**Calculation No. 11**

column C1, Floor 1

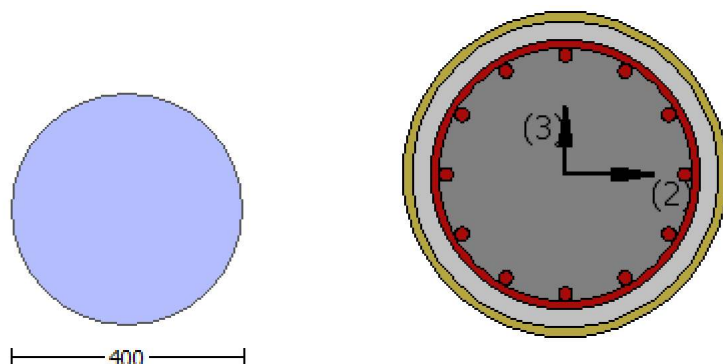
Limit State: Collapse Prevention (data interpolation between analysis steps 2 and 3)

Analysis: Uniform +X

Check: Shear capacity VRd

Edge: Start

Local Axis: (3)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (a)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE 41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 16.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 400.00$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of  $\gamma$  for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE 41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE 41-17).

Existing material: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material: Steel Strength,  $f_s = f_{sm} = 444.44$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length  $l_o = l_b = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $\epsilon_{fu} = 0.01$

Number of directions,  $NoDir = 1$

Fiber orientations, bi: 0.00°  
Number of layers, NL = 1  
Radius of rounding corners, R = 40.00

#### Stepwise Properties

EDGE -A-  
Bending Moment, Ma = 6.2300792E-010  
Shear Force, Va = -1.7064698E-013  
EDGE -B-  
Bending Moment, Mb = -1.1086227E-010  
Shear Force, Vb = 1.7064698E-013  
BOTH EDGES  
Axial Force, F = -4783.229  
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension: Aslt = 1272.345  
-Compression: Aslc = 1781.283  
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension: Asl,ten = 1017.876  
-Compression: Asl,com = 1017.876  
-Middle: Asl,mid = 1017.876  
Mean Diameter of Tension Reinforcement, DbL,ten = 18.00

Existing component: From table 7-7, ASCE 41\_17: Final Shear Capacity VR = \*Vn = 286339.856  
Vn ((10.3), ASCE 41-17) = knl\*VCol0 = 286339.856  
VCol = 286339.856  
knl = 1.00  
displacement\_ductility\_demand = 0.00

NOTE: In expression (10-3) 'Vs = Av\*fy\*d/s' is replaced by 'Vs+ f\*Vf'  
where Vf is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)  
fc' = 16.00, but  $fc'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
M/Vd = 2.00  
Mu = 6.2300792E-010  
Vu = 1.7064698E-013  
d = 0.8\*D = 320.00  
Nu = 4783.229  
Ag = 125663.706  
From (11.5.4.8), ACI 318-14: Vs = 0.00  
Av = /2\*A\_stirrup = 123370.055  
fy = 400.00  
s = 360.00  
Vs is multiplied by Col = 0.00  
s/d = 1.125  
Vf ((11-3)-(11.4), ACI 440) = 194961.134  
f = 0.95, for fully-wrapped sections  
wf/sf = 1 (FRP strips adjacent to one another).  
In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a)\sin a$  which is more a generalised expression,  
where is the angle of the crack direction (see KANEPE).  
This later relation, considered as a function Vf( , ), is implemented for every different fiber orientation ai,  
as well as for 2 crack directions, =45° and =-45° to take into consideration the cyclic seismic loading.  
orientation 1: 1 = b1 + 90° = 90.00  
Vf = Min(|Vf(45, 1)|, |Vf(-45,a1)|), with:  
total thickness per orientation, tf1 = NL\*t/NoDir = 1.016  
dfv = d (figure 11.2, ACI 440) = 370.00  
ffe ((11-5), ACI 440) = 259.312  
Ef = 64828.00  
fe = 0.004, from (11.6a), ACI 440  
with fu = 0.01  
From (11-11), ACI 440: Vs + Vf <= 213705.936  
bw\*d = \*d\*d/4 = 80424.772

displacement\_ductility\_demand is calculated as  $\delta / y$

- Calculation of  $\delta / y$  for END A -  
for rotation axis 2 and integ. section (a)

From analysis, chord rotation  $\theta = 1.4283643E-020$   
 $y = (M_y * L_s / 3) / E_{eff} = 0.00931733$  ((4.29), Biskinis Phd))  
 $M_y = 1.4766E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 1500.00  
From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 7.9240E+012$   
 $factor = 0.30$   
 $A_g = 125663.706$   
 $f_c' = 20.00$   
 $N = 4783.229$   
 $E_c * I_g = 2.6413E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\delta$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \min(M_{y\_ten}, M_{y\_com}) = 1.4766E+008$   
 $y = 8.1688588E-006$   
 $M_{y\_ten}$  (8c) = 1.4766E+008  
 $\delta_{ten}$  (7c) = 72.4069  
error of function (7c) = 0.00117279  
 $M_{y\_com}$  (8d) = 3.7493E+008  
 $\delta_{com}$  (7d) = 69.91139  
error of function (7d) = 0.00300901  
with ((10.1), ASCE 41-17)  $e_y = \min(e_y, 1.25 * e_y * (l_b / l_d)^{2/3}) = 0.0022222$   
 $e_{co} = 0.002$   
 $a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00157746$   
 $N = 4783.229$   
 $A_c = 125663.706$   
((10.1), ASCE 41-17)  $\phi = \min(\phi, 1.25 * \phi * (l_b / l_d)^{2/3}) = 0.44757577$   
with  $f_c' * ((12.3), ACI 440) = 24.12975$   
 $f_c = 20.00$   
 $f_l = 1.3173$   
 $k = 1$   
Effective FRP thickness,  $t_f = N L * t * \cos(\theta) = 1.016$   
 $e_{fe}$  ((12.5) and (12.7)) = 0.004  
 $f_u = 0.01$   
 $E_f = 64828.00$

Calculation of ratio  $l_b / l_d$

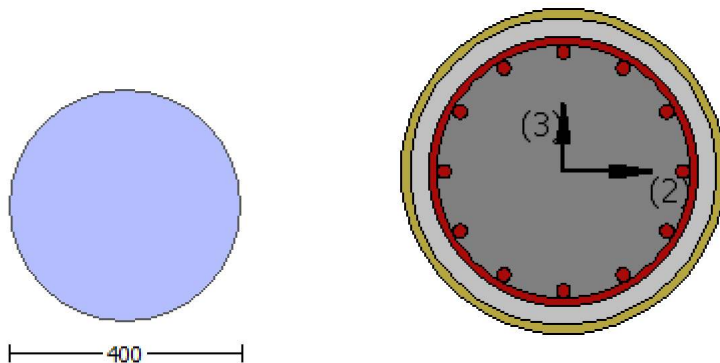
Lap Length:  $l_d / l_{d,min} = 0.33490748$   
 $l_b = 300.00$   
 $l_d = 895.7698$   
Calculation of  $l$  according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $\phi = 1$   
 $d_b = 18.00$   
Mean strength value of all re-bars:  $f_y = 444.44$   
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $c_b = 25.00$

$K_{tr} = 1.14232$   
 $A_{tr} = \frac{1}{2} * \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

-----  
 End Of Calculation of Shear Capacity for element: column CC1 of floor 1  
 At local axis: 3  
 Integration Section: (a)  
 -----

## Calculation No. 12

column C1, Floor 1  
 Limit State: Collapse Prevention (data interpolation between analysis steps 2 and 3)  
 Analysis: Uniform +X  
 Check: Chord rotation capacity (  $\phi$  )  
 Edge: Start  
 Local Axis: (3)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At Shear local axis: 3  
 (Bending local axis: 2)  
 Section Type: rccs

Constant Properties

-----  
 Knowledge Factor,  $\gamma = 1.00$   
 Mean strength values are used for both shear and moment calculations.  
 Consequently:  
 Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$   
 Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.44$   
 Concrete Elasticity,  $E_c = 21019.039$   
 Steel Elasticity,  $E_s = 200000.00$   
 #####  
 Note: Especially for the calculation of moment strengths,  
 the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14  
 Existing material: Steel Strength,  $f_s = 1.25 * f_{sm} = 555.55$   
 #####

Diameter,  $D = 400.00$   
 Cover Thickness,  $c = 25.00$   
 Mean Confinement Factor overall section = 1.84055  
 Element Length,  $L = 3000.00$   
 Primary Member  
 Ribbed Bars  
 Ductile Steel  
 Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Lap Length  $l_o = 300.00$   
 FRP Wrapping Data  
 Type: Carbon  
 Cured laminate properties (design values)  
 Thickness,  $t = 1.016$   
 Tensile Strength,  $f_{fu} = 1055.00$   
 Tensile Modulus,  $E_f = 64828.00$   
 Elongation,  $ε_{fu} = 0.01$   
 Number of directions,  $N_{oDir} = 1$   
 Fiber orientations,  $b_i: 0.00^\circ$   
 Number of layers,  $N_L = 1$   
 Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

At local axis: 3  
 EDGE -A-  
 Shear Force,  $V_a = 1.2107607E-031$   
 EDGE -B-  
 Shear Force,  $V_b = -1.2107607E-031$   
 BOTH EDGES  
 Axial Force,  $F = -4771.233$   
 Longitudinal Reinforcement Area Distribution (in 2 divisions)  
     -Tension:  $A_{sl,t} = 0.00$   
     -Compression:  $A_{sl,c} = 3053.628$   
 Longitudinal Reinforcement Area Distribution (in 3 divisions)  
     -Tension:  $A_{sl,ten} = 1017.876$   
     -Compression:  $A_{sl,com} = 1017.876$   
     -Middle:  $A_{sl,mid} = 1017.876$

Calculation of Shear Capacity ratio ,  $V_e/V_r = 0.30252729$   
 Member Controlled by Flexure ( $V_e/V_r < 1$ )  
 Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 90202.132$   
 with  
 $M_{pr1} = \text{Max}(Mu_{1+}, Mu_{1-}) = 1.3530E+008$   
 $Mu_{1+} = 1.3530E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination  
 $Mu_{1-} = 1.3530E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination  
 $M_{pr2} = \text{Max}(Mu_{2+}, Mu_{2-}) = 1.3530E+008$   
 $Mu_{2+} = 1.3530E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination  
 $Mu_{2-} = 1.3530E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

#### Calculation of $Mu_{1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$   
 $M_u = 1.3530E+008$   
 $= 0.90757121$



$\lambda = 0.80580716$   
 error of function (3.68), Biskinis Phd = 28928.286  
 From 5A.2, TBDY:  $f_{cc} = f_c \cdot \lambda = 36.81095$   
 conf. factor  $\lambda = 1.84055$   
 $f_c = 20.00$   
 From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

#### Calculation of ratio $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
 Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $= 1$   
 $d_b = 18.00$   
 Mean strength value of all re-bars:  $f_y = 555.55$   
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $c_b = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

#### Calculation of $\mu_{u1}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$   
 $\mu_u = 1.3530E+008$

$= 0.90757121$   
 $\lambda = 0.80580716$   
 error of function (3.68), Biskinis Phd = 28928.286  
 From 5A.2, TBDY:  $f_{cc} = f_c \cdot \lambda = 36.81095$   
 conf. factor  $\lambda = 1.84055$   
 $f_c = 20.00$   
 From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

#### Calculation of ratio $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
 Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $= 1$   
 $d_b = 18.00$

Mean strength value of all re-bars:  $f_y = 555.55$   
 $f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \frac{1}{2} \times \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

#### Calculation of $\mu_{2+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 1.3530E+008$

$\phi = 0.90757121$   
 $\phi' = 0.80580716$   
error of function (3.68), Biskinis Phd = 28928.286  
From 5A.2, TBDY:  $f_{cc} = f'_c \times c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \times \text{Min}(1, 1.25 \times (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $\phi = \phi' \times \text{Min}(1, 1.25 \times (l_b/l_d)^{2/3}) = 0.28805051$

#### Calculation of ratio $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $\phi = 1$   
 $db = 18.00$   
Mean strength value of all re-bars:  $f_y = 555.55$   
 $f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \frac{1}{2} \times \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

#### Calculation of $\mu_{2-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 1.3530E+008$

$\phi = 0.90757121$   
 $\phi' = 0.80580716$   
error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY:  $f_{cc} = f_c^* \quad c = 36.81095$

conf. factor  $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio  $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$d_b = 18.00$

Mean strength value of all re-bars:  $f_y = 555.55$

$f_c' = 20.00$ , but  $f_c^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 298161.965$

Calculation of Shear Strength at edge 1,  $V_{r1} = 298161.965$

$V_{r1} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{Col0}$

$V_{Col0} = 298161.965$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

$f_c' = 20.00$ , but  $f_c^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.2899604E-011$

$\nu_u = 1.2107607E-031$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 0.00$

$A_v = \frac{1}{2} \cdot A_{\text{stirrup}} = 123370.055$

$f_y = 444.44$

$s = 360.00$

$V_s$  is multiplied by  $\text{Col} = 0.00$

$s/d = 1.125$

$V_f$  ((11-3)-(11.4), ACI 440) =  $194961.134$

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \cot \alpha) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta_1 = \theta_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, a1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}((11-5), \text{ACI 440}) = 259.312$   
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $b_w \cdot d = \frac{1}{4} \cdot d \cdot d = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 298161.965$   
 $V_{r2} = V_{Col}((10.3), \text{ASCE 41-17}) = k_{nl} \cdot V_{Col0}$   
 $V_{Col0} = 298161.965$   
 $k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ '  
 where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)  
 $f'_c = 20.00$ , but  $f'_c^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $\mu_u = 1.2899604E-011$   
 $\nu_u = 1.2107607E-031$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 0.00$   
 $A_v = \frac{1}{2} \cdot A_{stirrup} = 123370.055$   
 $f_y = 444.44$   
 $s = 360.00$   
 $V_s$  is multiplied by  $\text{Col} = 0.00$   
 $s/d = 1.125$   
 $V_f((11-3)-(11.4), \text{ACI 440}) = 194961.134$   
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = b1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, a1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}((11-5), \text{ACI 440}) = 259.312$   
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $b_w \cdot d = \frac{1}{4} \cdot d \cdot d = 80424.772$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At local axis: 3

Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At Shear local axis: 2  
 (Bending local axis: 3)  
 Section Type: rccs

Constant Properties

Knowledge Factor,  $\phi = 1.00$   
 Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.44$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 555.55$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.84055

Element Length,  $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length  $l_o = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $\epsilon_{fu} = 0.01$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i: 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

At local axis: 2

EDGE -A-

Shear Force,  $V_a = -7.4135260E-048$

EDGE -B-

Shear Force,  $V_b = 7.4135260E-048$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{sl,t} = 0.00$

-Compression:  $A_{sl,c} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{sl,ten} = 1017.876$

-Compression:  $A_{sl,com} = 1017.876$

-Middle:  $A_{sl,mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.30252729$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 90202.132$

with

$M_{pr1} = \max(\mu_{u1+}, \mu_{u1-}) = 1.3530E+008$

$\mu_{u1+} = 1.3530E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$\mu_{u1-} = 1.3530E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \max(\mu_{u2+}, \mu_{u2-}) = 1.3530E+008$

$\mu_{u2+} = 1.3530E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination

$\mu_{u2-} = 1.3530E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

#### Calculation of Mu1+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.3530E+008

= 0.90757121  
' = 0.80580716  
error of function (3.68), Biskinis Phd = 28928.286  
From 5A.2, TBDY:  $f_{cc} = f_c^* c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y * \text{Min}(1, 1.25 * (l_b / l_d)^{2/3}) = 288.6089$   
 $l_b / l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
=  $*\text{Min}(1, 1.25 * (l_b / l_d)^{2/3}) = 0.28805051$

#### Calculation of ratio lb/l\_d

Lap Length:  $l_b / l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
= 1  
 $db = 18.00$   
Mean strength value of all re-bars:  $f_y = 555.55$   
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \sqrt{2} * \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

#### Calculation of Mu1-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.3530E+008

= 0.90757121  
' = 0.80580716  
error of function (3.68), Biskinis Phd = 28928.286  
From 5A.2, TBDY:  $f_{cc} = f_c^* c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y * \text{Min}(1, 1.25 * (l_b / l_d)^{2/3}) = 288.6089$   
 $l_b / l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
=  $*\text{Min}(1, 1.25 * (l_b / l_d)^{2/3}) = 0.28805051$

#### Calculation of ratio $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of  $l_b, \min$  according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$db = 18.00$

Mean strength value of all re-bars:  $f_y = 555.55$

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$cb = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \times \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

#### Calculation of $\mu_{2+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$\mu = 1.3530E+008$

$= 0.90757121$

$' = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY:  $f_{cc} = f'_c \times c = 36.81095$

conf. factor  $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \times \min(1, 1.25 \times (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \times \min(1, 1.25 \times (l_b/l_d)^{2/3}) = 0.28805051$

#### Calculation of ratio $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of  $l_b, \min$  according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$db = 18.00$

Mean strength value of all re-bars:  $f_y = 555.55$

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$cb = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \times \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

## Calculation of Mu2-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.3530E+008

= 0.90757121  
' = 0.80580716  
error of function (3.68), Biskinis Phd = 28928.286  
From 5A.2, TBDY:  $f_{cc} = f_c' \cdot c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
=  $\cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

## Calculation of ratio $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
= 1  
 $d_b = 18.00$   
Mean strength value of all re-bars:  $f_y = 555.55$   
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $c_b = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \pi/2 \cdot \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

## Calculation of Shear Strength $V_r = \text{Min}(V_{r1}, V_{r2}) = 298161.965$

Calculation of Shear Strength at edge 1,  $V_{r1} = 298161.965$   
 $V_{r1} = V_{CoI} \text{ ((10.3), ASCE 41-17)} = k_{nl} \cdot V_{CoI0}$   
 $V_{CoI0} = 298161.965$   
 $k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)  
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $M_u = 5.9321525E-012$   
 $V_u = 7.4135260E-048$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
From (11.5.4.8), ACI 318-14:  $V_s = 0.00$   
 $A_v = \pi/2 \cdot A_{\text{stirrup}} = 123370.055$   
 $f_y = 444.44$   
 $s = 360.00$



Vs is multiplied by Col = 0.00

$$s/d = 1.125$$

$$V_f((11-3)-(11.4), \text{ACI 440}) = 194961.134$$

f = 0.95, for fully-wrapped sections

wf/sf = 1 (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression, where  $a$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta)$ , is implemented for every different fiber orientation  $\theta_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

$$\theta_1 = \theta_1 + 90^\circ = 90.00$$

$$V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, \theta_1)|), \text{ with:}$$

$$\text{total thickness per orientation, } t_{f1} = NL \cdot t / \text{NoDir} = 1.016$$

$$d_{fv} = d \text{ (figure 11.2, ACI 440)} = 370.00$$

$$f_{fe}((11-5), \text{ACI 440}) = 259.312$$

$$E_f = 64828.00$$

$$f_e = 0.004, \text{ from (11.6a), ACI 440}$$

$$\text{with } f_u = 0.01$$

$$\text{From (11-11), ACI 440: } V_s + V_f \leq 238930.50$$

$$b_w \cdot d = \rho \cdot d^2 / 4 = 80424.772$$

Calculation of Shear Strength at edge 2,  $V_{r2} = 298161.965$

$$V_{r2} = V_{Col}((10.3), \text{ASCE 41-17}) = k_{nl} \cdot V_{Col0}$$

$$V_{Col0} = 298161.965$$

$$k_{nl} = 1 \text{ (zero step-static loading)}$$

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\rho = 1$  (normal-weight concrete)

$$f'_c = 20.00, \text{ but } f_c^{0.5} \leq 8.3 \text{ MPa (22.5.3.1, ACI 318-14)}$$

$$M/Vd = 2.00$$

$$\mu_u = 5.9321525E-012$$

$$V_u = 7.4135260E-048$$

$$d = 0.8 \cdot D = 320.00$$

$$N_u = 4771.233$$

$$A_g = 125663.706$$

$$\text{From (11.5.4.8), ACI 318-14: } V_s = 0.00$$

$$A_v = \rho / 2 \cdot A_{stirrup} = 123370.055$$

$$f_y = 444.44$$

$$s = 360.00$$

Vs is multiplied by Col = 0.00

$$s/d = 1.125$$

$$V_f((11-3)-(11.4), \text{ACI 440}) = 194961.134$$

f = 0.95, for fully-wrapped sections

wf/sf = 1 (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression, where  $a$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta)$ , is implemented for every different fiber orientation  $\theta_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

$$\theta_1 = \theta_1 + 90^\circ = 90.00$$

$$V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, \theta_1)|), \text{ with:}$$

$$\text{total thickness per orientation, } t_{f1} = NL \cdot t / \text{NoDir} = 1.016$$

$$d_{fv} = d \text{ (figure 11.2, ACI 440)} = 370.00$$

$$f_{fe}((11-5), \text{ACI 440}) = 259.312$$

$$E_f = 64828.00$$

$$f_e = 0.004, \text{ from (11.6a), ACI 440}$$

$$\text{with } f_u = 0.01$$

$$\text{From (11-11), ACI 440: } V_s + V_f \leq 238930.50$$

$$b_w \cdot d = \rho \cdot d^2 / 4 = 80424.772$$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At local axis: 2

Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (a)

Section Type: rccs

#### Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE 41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.44$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length  $l_b = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $\epsilon_{fu} = 0.01$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i: 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

Bending Moment,  $M = -1.3471E+007$

Shear Force,  $V_2 = -4486.792$

Shear Force,  $V_3 = -1.7064698E-013$

Axial Force,  $F = -4783.229$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{st} = 1272.345$

-Compression:  $A_{sc} = 1781.283$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{st,ten} = 1017.876$

-Compression:  $A_{st,com} = 1017.876$

-Middle:  $A_{st,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $D_{bL} = 18.00$

Existing component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_{R} = \gamma \cdot u = 0.06064917$

$u = \gamma \cdot u + p = 0.06064917$

- Calculation of  $\gamma$  -

$\gamma = (M_y \cdot L_s / 3) / E_{eff} = 0.01864917$  ((4.29), Biskinis Phd))

$M_y = 1.4766E+008$

$L_s = M/V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) = 3002.334

From table 10.5, ASCE 41\_17:  $E_{eff} = \text{factor} \cdot E_c \cdot I_g = 7.9240E+012$

factor = 0.30

Ag = 125663.706  
fc' = 20.00  
N = 4783.229  
Ec\*Ig = 2.6413E+013

#### Calculation of Yielding Moment My

Calculation of  $\rho_y$  and My according to (7) - (8) in Biskinis and Fardis

My = Min(My\_ten, My\_com) = 1.4766E+008  
 $\rho_y = 8.1688588E-006$   
My\_ten (8c) = 1.4766E+008  
 $\rho_{y\_ten} (7c) = 72.4069$   
error of function (7c) = 0.00117279  
My\_com (8d) = 3.7493E+008  
 $\rho_{y\_com} (7d) = 69.91139$   
error of function (7d) = 0.00300901  
with ((10.1), ASCE 41-17)  $\rho_y = \text{Min}(\rho_y, 1.25 * \rho_y * (l_b/l_d)^{2/3}) = 0.0022222$   
 $\rho_{eco} = 0.002$   
 $\rho_{apl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
d1 = 44.00  
R = 200.00  
v = 0.00157746  
N = 4783.229  
Ac = 125663.706  
((10.1), ASCE 41-17)  $\rho_y = \text{Min}(\rho_y, 1.25 * \rho_y * (l_b/l_d)^{2/3}) = 0.44757577$   
with fc' ((12.3), ACI 440) = 24.12975  
fc = 20.00  
fl = 1.3173  
k = 1  
Effective FRP thickness, tf = NL\*t\*cos(b1) = 1.016  
efe ((12.5) and (12.7)) = 0.004  
fu = 0.01  
Ef = 64828.00

#### Calculation of ratio lb/l\_d

Lap Length:  $l_d/l_{d,min} = 0.33490748$   
lb = 300.00  
ld = 895.7698  
Calculation of l according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
= 1  
db = 18.00  
Mean strength value of all re-bars: fy = 444.44  
fc' = 20.00, but  $fc'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
t = 1.00  
s = 0.80  
e = 1.00  
cb = 25.00  
Ktr = 1.14232  
Atr =  $\sqrt{2} * \text{Area of stirrup} = 123.3701$   
s = 360.00  
n = 12.00

#### - Calculation of $\rho_p$ -

From table 10-9:  $\rho_p = 0.042$

with:

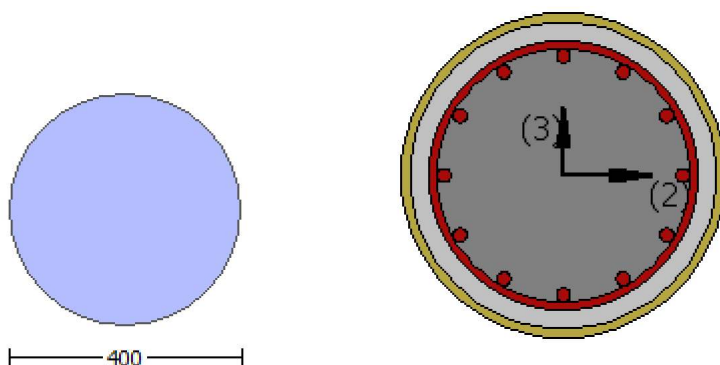
- Columns controlled by inadequate development or splicing along the clear height because  $l_b/l_d < 1$
- shear control ratio  $V_y E / V_{CoI} E = 0.30252729$
- d = 0.00
- s = 0.00

$t = 2 \cdot A_v / (d_c \cdot s) + 4 \cdot t_f / D \cdot (f_{fe} / f_s) = 0.00721126$   
 $A_v = 78.53982$ , is the area of the circular stirrup  
 $d_c = D - 2 \cdot \text{cover} - \text{Hoop Diameter} = 340.00$   
 The term  $2 \cdot t_f / b_w \cdot (f_{fe} / f_s)$  is implemented to account for FRP contribution  
 where  $f = 2 \cdot t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength  
 All these variables have already been given in Shear control ratio calculation.  
 $NUD = 4783.229$   
 $Ag = 125663.706$   
 $f_{cE} = 20.00$   
 $f_{yE} = f_{yI} = 444.44$   
 $pl = \text{Area\_Tot\_Long\_Rein} / (Ag) = 0.0243$   
 $f_{cE} = 20.00$

End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1  
 At local axis: 3  
 Integration Section: (a)

## Calculation No. 13

column C1, Floor 1  
 Limit State: Collapse Prevention (data interpolation between analysis steps 2 and 3)  
 Analysis: Uniform +X  
 Check: Shear capacity  $VR_d$   
 Edge: End  
 Local Axis: (2)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1  
 At local axis: 2  
 Integration Section: (b)  
 Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$   
 Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE 41-17.  
 Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 16.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 400.00$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of  $\gamma$  for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE 41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE 41-17).

Existing material: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material: Steel Strength,  $f_s = f_{sm} = 444.44$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length  $l_o = l_b = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $ef_u = 0.01$

Number of directions,  $NoDir = 1$

Fiber orientations,  $bi: 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

EDGE -A-

Bending Moment,  $M_a = -1.3471E+007$

Shear Force,  $V_a = -4486.792$

EDGE -B-

Bending Moment,  $M_b = 2684.487$

Shear Force,  $V_b = 4486.792$

BOTH EDGES

Axial Force,  $F = -4783.229$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 0.00$

-Compression:  $As_c = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t,ten} = 1017.876$

-Compression:  $As_{c,com} = 1017.876$

-Middle:  $As_{mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $Db_{L,ten} = 18.00$

Existing component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = \phi V_n = 286339.856$

$V_n$  ((10.3), ASCE 41-17) =  $k_n \phi V_{Col} = 286339.856$

$V_{Col} = 286339.856$

$k_n = 1.00$

$displacement\_ductility\_demand = 0.13687722$

NOTE: In expression (10-3) ' $V_s = A_v f_y d/s$ ' is replaced by ' $V_s + \phi V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\phi = 1$  (normal-weight concrete)

$f'_c = 16.00$ , but  $f'_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$   
 $\mu_u = 2684.487$   
 $V_u = 4486.792$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4783.229$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 0.00$   
 $A_v = \frac{1}{2} \cdot A_{stirrup} = 123370.055$   
 $f_y = 400.00$   
 $s = 360.00$   
 $V_s$  is multiplied by  $\text{Col} = 0.00$   
 $s/d = 1.125$   
 $V_f ((11-3)-(11.4), \text{ACI 440}) = 194961.134$   
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = 45^\circ + 90^\circ = 135^\circ$   
 $V_f = \text{Min}(|V_f(45, a_1)|, |V_f(-45, a_1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{Dir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe} ((11-5), \text{ACI 440}) = 259.312$   
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 213705.936$   
 $b_w \cdot d = \frac{V_u}{f_e \cdot E_f} = 80424.772$

displacement ductility demand is calculated as  $\delta_u / y$

- Calculation of  $\delta_u / y$  for END B -  
 for rotation axis 3 and integ. section (b)

From analysis, chord rotation  $\phi = 0.00025507$   
 $y = (M_y \cdot L_s / 3) / E_{eff} = 0.00186347$  ((4.29), Biskinis Phd))  
 $M_y = 1.4766E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) = 300.00  
 From table 10.5, ASCE 41-17:  $E_{eff} = \text{factor} \cdot E_c \cdot I_g = 7.9240E+012$   
 $\text{factor} = 0.30$   
 $A_g = 125663.706$   
 $f'_c = 20.00$   
 $N = 4783.229$   
 $E_c \cdot I_g = 2.6413E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\delta_u$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 1.4766E+008$   
 $y = 8.1688588E-006$   
 $M_{y\_ten} (8c) = 1.4766E+008$   
 $y_{ten} (7c) = 72.4069$   
 error of function (7c) = 0.00117279  
 $M_{y\_com} (8d) = 3.7493E+008$   
 $y_{com} (7d) = 69.91139$   
 error of function (7d) = 0.00300901  
 with ((10.1), ASCE 41-17)  $e_y = \text{Min}(e_y, 1.25 \cdot e_y \cdot (I_b/I_d)^{2/3}) = 0.0022222$   
 $e_{co} = 0.002$   
 $a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 200.00$

$v = 0.00157746$   
 $N = 4783.229$   
 $A_c = 125663.706$   
 $((10.1), ASCE 41-17) = \text{Min}( , 1.25 * ((lb/ld)^{2/3}) = 0.44757577$   
 with  $f_c' ((12.3), ACI 440) = 24.12975$   
 $f_c = 20.00$   
 $f_l = 1.3173$   
 $k = 1$   
 Effective FRP thickness,  $t_f = NL * t * \cos(b_1) = 1.016$   
 $e_{fe} ((12.5) \text{ and } (12.7)) = 0.004$   
 $f_u = 0.01$   
 $E_f = 64828.00$

Calculation of ratio  $lb/ld$

Lap Length:  $ld/ld, \min = 0.33490748$

$lb = 300.00$

$ld = 895.7698$

Calculation of  $l$  according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$db = 18.00$

Mean strength value of all re-bars:  $f_y = 444.44$

$f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$cb = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \pi/2 * \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

End Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (b)

## Calculation No. 14

column C1, Floor 1

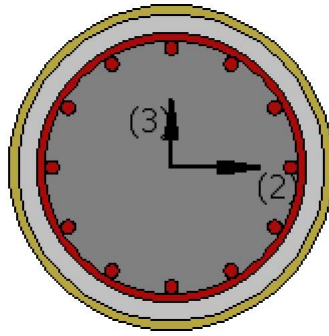
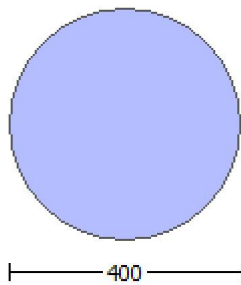
Limit State: Collapse Prevention (data interpolation between analysis steps 2 and 3)

Analysis: Uniform +X

Check: Chord rotation capacity (  $\phi$  )

Edge: End

Local Axis: (2)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.44$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 555.55$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.84055

Element Length,  $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length  $l_o = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $\epsilon_{fu} = 0.01$

Number of directions,  $NoDir = 1$

Fiber orientations,  $b_i: 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

Stepwise Properties

At local axis: 3

EDGE -A-

Shear Force,  $V_a = 1.2107607E-031$

EDGE -B-

Shear Force,  $V_b = -1.2107607E-031$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)



-Tension:  $Asl_t = 0.00$   
 -Compression:  $Asl_c = 3053.628$   
 Longitudinal Reinforcement Area Distribution (in 3 divisions)  
 -Tension:  $Asl_{ten} = 1017.876$   
 -Compression:  $Asl_{com} = 1017.876$   
 -Middle:  $Asl_{mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.30252729$   
 Member Controlled by Flexure ( $V_e/V_r < 1$ )  
 Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 90202.132$   
 with  
 $M_{pr1} = \max(\mu_{u1+}, \mu_{u1-}) = 1.3530E+008$   
 $\mu_{u1+} = 1.3530E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination  
 $\mu_{u1-} = 1.3530E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination  
 $M_{pr2} = \max(\mu_{u2+}, \mu_{u2-}) = 1.3530E+008$   
 $\mu_{u2+} = 1.3530E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination  
 $\mu_{u2-} = 1.3530E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

Calculation of  $\mu_{u1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$   
 $\mu_u = 1.3530E+008$

$\phi = 0.90757121$   
 $\phi' = 0.80580716$   
 error of function (3.68), Biskinis Phd = 28928.286  
 From 5A.2, TDY:  $f_{cc} = f_c' \cdot c = 36.81095$   
 conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
 From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \min(1, 1.25 \cdot (l_b/d)^{2/3}) = 288.6089$   
 $l_b/d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \phi' \cdot \min(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.28805051$

Calculation of ratio  $l_b/d$

Lap Length:  $l_b/d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
 Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $\phi = 1$   
 $d_b = 18.00$   
 Mean strength value of all re-bars:  $f_y = 555.55$   
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \pi \cdot \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

Calculation of Mu1-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.3530E+008

= 0.90757121  
' = 0.80580716  
error of function (3.68), Biskinis Phd = 28928.286  
From 5A.2, TDY:  $f_{cc} = f_c \cdot c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
=  $\cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio  $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
= 1  
 $db = 18.00$   
Mean strength value of all re-bars:  $f_y = 555.55$   
 $f_c' = 20.00$ , but  $f_c^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

Calculation of Mu2+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.3530E+008

= 0.90757121  
' = 0.80580716  
error of function (3.68), Biskinis Phd = 28928.286  
From 5A.2, TDY:  $f_{cc} = f_c \cdot c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.28805051$$

Calculation of ratio  $l_b/d$

Lap Length:  $l_b/d = 0.26792599$

$l_b = 300.00$

$d = 1119.712$

Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$d_b = 18.00$

Mean strength value of all re-bars:  $f_y = 555.55$

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of  $\mu_2$ -

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$\mu = 1.3530E+008$

$= 0.90757121$

$' = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY:  $f_{cc} = f'_c \cdot c = 36.81095$

conf. factor  $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 288.6089$

$l_b/d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.28805051$

Calculation of ratio  $l_b/d$

Lap Length:  $l_b/d = 0.26792599$

$l_b = 300.00$

$d = 1119.712$

Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$d_b = 18.00$

Mean strength value of all re-bars:  $f_y = 555.55$

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 298161.965$

Calculation of Shear Strength at edge 1,  $V_{r1} = 298161.965$

$V_{r1} = V_{Col} \text{ ((10.3), ASCE 41-17)} = k_{nl} * V_{Col0}$

$V_{Col0} = 298161.965$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\lambda = 1$  (normal-weight concrete)

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.2899604E-011$

$\nu_u = 1.2107607E-031$

$d = 0.8 * D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 0.00$

$A_v = \sqrt{2} * A_{\text{stirrup}} = 123370.055$

$f_y = 444.44$

$s = 360.00$

$V_s$  is multiplied by  $\text{Col} = 0.00$

$s/d = 1.125$

$V_f \text{ ((11-3)-(11.4), ACI 440)} = 194961.134$

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression, where  $a$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta)$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta_1 = b_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, a_1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L * t / N_{\text{Dir}} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe} \text{ ((11-5), ACI 440)} = 259.312$

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$

$b_w * d = \sqrt{2} * d^2 / 4 = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 298161.965$

$V_{r2} = V_{Col} \text{ ((10.3), ASCE 41-17)} = k_{nl} * V_{Col0}$

$V_{Col0} = 298161.965$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\lambda = 1$  (normal-weight concrete)

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.2899604E-011$

$\nu_u = 1.2107607E-031$

$d = 0.8 * D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 0.00$

$A_v = \sqrt{2} * A_{\text{stirrup}} = 123370.055$

$f_y = 444.44$

$s = 360.00$

$V_s$  is multiplied by  $\text{Col} = 0.00$

$s/d = 1.125$

$V_f((11-3)-(11.4), \text{ACI 440}) = 194961.134$   
 $f = 0.95$ , for fully-wrapped sections  
 $wf/sf = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a)\sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta_1 = \theta_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, \theta_1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}((11-5), \text{ACI 440}) = 259.312$   
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $b_w \cdot d = \frac{V_s \cdot d}{4} = 80424.772$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At local axis: 3

Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At Shear local axis: 2  
 (Bending local axis: 3)  
 Section Type: rccs

#### Constant Properties

Knowledge Factor,  $\phi = 1.00$   
 Mean strength values are used for both shear and moment calculations.  
 Consequently:  
 Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$   
 Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.44$   
 Concrete Elasticity,  $E_c = 21019.039$   
 Steel Elasticity,  $E_s = 200000.00$   
 #####  
 Note: Especially for the calculation of moment strengths,  
 the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14  
 Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 555.55$   
 #####  
 Diameter,  $D = 400.00$   
 Cover Thickness,  $c = 25.00$   
 Mean Confinement Factor overall section = 1.84055  
 Element Length,  $L = 3000.00$   
 Primary Member  
 Ribbed Bars  
 Ductile Steel  
 Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Lap Length  $l_o = 300.00$   
 FRP Wrapping Data  
 Type: Carbon  
 Cured laminate properties (design values)  
 Thickness,  $t = 1.016$   
 Tensile Strength,  $f_{fu} = 1055.00$   
 Tensile Modulus,  $E_f = 64828.00$   
 Elongation,  $e_{fu} = 0.01$   
 Number of directions,  $\text{NoDir} = 1$   
 Fiber orientations,  $b_i: 0.00^\circ$   
 Number of layers,  $NL = 1$   
 Radius of rounding corners,  $R = 40.00$

## Stepwise Properties

At local axis: 2

EDGE -A-

Shear Force,  $V_a = -7.4135260E-048$

EDGE -B-

Shear Force,  $V_b = 7.4135260E-048$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 0.00$

-Compression:  $As_c = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t,ten} = 1017.876$

-Compression:  $As_{c,com} = 1017.876$

-Middle:  $As_{l,mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.30252729$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 90202.132$   
with

$M_{pr1} = \text{Max}(\mu_{u1+}, \mu_{u1-}) = 1.3530E+008$

$\mu_{u1+} = 1.3530E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$\mu_{u1-} = 1.3530E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(\mu_{u2+}, \mu_{u2-}) = 1.3530E+008$

$\mu_{u2+} = 1.3530E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination

$\mu_{u2-} = 1.3530E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

Calculation of  $\mu_{u1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$

$\mu_u = 1.3530E+008$

$= 0.90757121$

$' = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY:  $f_{cc} = f_c^* \quad c = 36.81095$

conf. factor  $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio  $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$db = 18.00$

Mean strength value of all re-bars:  $f_y = 555.55$   
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \frac{1}{2} \times \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

#### Calculation of $\mu_{u1}$ -

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$   
 $\mu_u = 1.3530E+008$

$\phi = 0.90757121$   
 $\phi' = 0.80580716$   
error of function (3.68), Biskinis Phd = 28928.286  
From 5A.2, TBDY:  $f_{cc} = f_c' \times c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \times \text{Min}(1, 1.25 \times (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $\phi' \times \text{Min}(1, 1.25 \times (l_b/l_d)^{2/3}) = 0.28805051$

#### Calculation of ratio $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $\phi = 1$   
 $db = 18.00$   
Mean strength value of all re-bars:  $f_y = 555.55$   
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \frac{1}{2} \times \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

#### Calculation of $\mu_{u2+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$   
 $\mu_u = 1.3530E+008$

$\phi = 0.90757121$   
 $\phi' = 0.80580716$   
error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY:  $f_{cc} = f_c^* \quad c = 36.81095$   
 conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
 From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y * \text{Min}(1, 1.25 * (l_b / l_d)^{2/3}) = 288.6089$   
 $l_b / l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= * \text{Min}(1, 1.25 * (l_b / l_d)^{2/3}) = 0.28805051$

Calculation of ratio  $l_b / l_d$

Lap Length:  $l_b / l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
 Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $= 1$   
 $db = 18.00$   
 Mean strength value of all re-bars:  $f_y = 555.55$   
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = /2 * \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

Calculation of  $\mu_2$ -

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 1.3530E+008$

$= 0.90757121$   
 $' = 0.80580716$   
 error of function (3.68), Biskinis Phd = 28928.286  
 From 5A.2, TBDY:  $f_{cc} = f_c^* \quad c = 36.81095$   
 conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
 From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y * \text{Min}(1, 1.25 * (l_b / l_d)^{2/3}) = 288.6089$   
 $l_b / l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= * \text{Min}(1, 1.25 * (l_b / l_d)^{2/3}) = 0.28805051$

Calculation of ratio  $l_b / l_d$

Lap Length:  $l_b / l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
 Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $= 1$   
 $db = 18.00$   
 Mean strength value of all re-bars:  $f_y = 555.55$   
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)



$t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \frac{1}{2} \times \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 298161.965$

Calculation of Shear Strength at edge 1,  $V_{r1} = 298161.965$

$V_{r1} = V_{Col} \text{ ((10.3), ASCE 41-17)} = k_{nl} \times V_{Col0}$

$V_{Col0} = 298161.965$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \times f_y \times d / s$ ' is replaced by ' $V_s + f \times V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 5.9321525E-012$

$\mu_v = 7.4135260E-048$

$d = 0.8 \times D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 0.00$

$A_v = \frac{1}{2} \times A_{\text{stirrup}} = 123370.055$

$f_y = 444.44$

$s = 360.00$

$V_s$  is multiplied by  $\text{Col} = 0.00$

$s/d = 1.125$

$V_f \text{ ((11-3)-(11.4), ACI 440)} = 194961.134$

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression, where  $a$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = b1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, a1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \times t / N_{\text{Dir}} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe} \text{ ((11-5), ACI 440)} = 259.312$

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$

$b_w \times d = \frac{1}{4} \times d \times d = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 298161.965$

$V_{r2} = V_{Col} \text{ ((10.3), ASCE 41-17)} = k_{nl} \times V_{Col0}$

$V_{Col0} = 298161.965$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \times f_y \times d / s$ ' is replaced by ' $V_s + f \times V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 5.9321525E-012$

$V_u = 7.4135260E-048$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 0.00$   
 $A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 123370.055$   
 $f_y = 444.44$   
 $s = 360.00$   
 $V_s$  is multiplied by  $\text{Col} = 0.00$   
 $s/d = 1.125$   
 $V_f ((11-3)-(11.4), \text{ACI } 440) = 194961.134$   
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = \theta_1 = 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, a_1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = N_L \cdot t / \text{NoDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe} ((11-5), \text{ACI } 440) = 259.312$   
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $b_w \cdot d = \sqrt{2} \cdot d \cdot d / 4 = 80424.772$

-----  
 End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At local axis: 2  
 -----

Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1  
 At local axis: 2  
 Integration Section: (b)  
 Section Type: rccs

Constant Properties

-----  
 Knowledge Factor,  $\gamma = 1.00$   
 Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE 41-17.  
 Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17  
 Consequently:  
 Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$   
 Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.44$   
 Concrete Elasticity,  $E_c = 21019.039$   
 Steel Elasticity,  $E_s = 200000.00$   
 Diameter,  $D = 400.00$   
 Cover Thickness,  $c = 25.00$   
 Element Length,  $L = 3000.00$   
 Primary Member  
 Ribbed Bars  
 Ductile Steel  
 Without Detailing for Earthquake Resistance (including stirrups not closed at  $135^\circ$ )  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Lap Length  $l_b = 300.00$   
 FRP Wrapping Data  
 Type: Carbon  
 Cured laminate properties (design values)  
 Thickness,  $t = 1.016$   
 Tensile Strength,  $f_{fu} = 1055.00$   
 Tensile Modulus,  $E_f = 64828.00$   
 Elongation,  $e_{fu} = 0.01$   
 Number of directions,  $\text{NoDir} = 1$

Fiber orientations,  $\theta_i$ : 0.00°  
Number of layers, NL = 1  
Radius of rounding corners, R = 40.00

#### Stepwise Properties

Bending Moment, M = -1.1086227E-010  
Shear Force, V2 = 4486.792  
Shear Force, V3 = 1.7064698E-013  
Axial Force, F = -4783.229  
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $A_{st}$  = 0.00  
-Compression:  $A_{sc}$  = 3053.628  
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $A_{st,ten}$  = 1017.876  
-Compression:  $A_{sc,com}$  = 1017.876  
-Middle:  $A_{st,mid}$  = 1017.876  
Mean Diameter of Tension Reinforcement,  $D_bL$  = 18.00

Existing component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $\phi_{u,R} = \phi_u = 0.05131733$   
 $\phi_u = \phi_y + \phi_p = 0.05131733$

- Calculation of  $\phi_y$  -

$\phi_y = (M_y * L_s / 3) / E_{eff} = 0.00931733$  ((4.29), Biskinis Phd))  
 $M_y = 1.4766E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 1500.00  
From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 7.9240E+012$   
factor = 0.30  
 $A_g = 125663.706$   
 $f_c' = 20.00$   
N = 4783.229  
 $E_c * I_g = 2.6413E+013$

#### Calculation of Yielding Moment $M_y$

Calculation of  $\phi_y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \min(M_{y,ten}, M_{y,com}) = 1.4766E+008$   
 $\phi_y = 8.1688588E-006$   
 $M_{y,ten}$  (8c) = 1.4766E+008  
 $\phi_{y,ten}$  (7c) = 72.4069  
error of function (7c) = 0.00117279  
 $M_{y,com}$  (8d) = 3.7493E+008  
 $\phi_{y,com}$  (7d) = 69.91139  
error of function (7d) = 0.00300901  
with ((10.1), ASCE 41-17)  $\phi_y = \min(\phi_y, 1.25 * \phi_{y,ten} * (I_b / I_d)^{2/3}) = 0.0022222$   
 $\phi_{y,ten} = 0.002$   
 $\phi_{y,com} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
R = 200.00  
 $v = 0.00157746$   
N = 4783.229  
 $A_c = 125663.706$   
((10.1), ASCE 41-17)  $\phi_y = \min(\phi_y, 1.25 * \phi_{y,com} * (I_b / I_d)^{2/3}) = 0.44757577$   
with  $f_c' = 20.00$  ((12.3), ACI 440) = 24.12975  
 $f_l = 1.3173$   
k = 1  
Effective FRP thickness,  $t_f = NL * t * \cos(\theta_1) = 1.016$

$\phi_e((12.5) \text{ and } (12.7)) = 0.004$   
 $f_u = 0.01$   
 $E_f = 64828.00$

Calculation of ratio  $l_b/l_d$

Lap Length:  $l_d/l_{d,min} = 0.33490748$

$l_b = 300.00$

$l_d = 895.7698$

Calculation of  $\lambda$  according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $\lambda = 1$

$d_b = 18.00$

Mean strength value of all re-bars:  $f_y = 444.44$

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$c_b = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \pi/2 * \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

- Calculation of  $\rho$  -

From table 10-9:  $\rho = 0.042$

with:

- Columns controlled by inadequate development or splicing along the clear height because  $l_b/l_d < 1$

shear control ratio  $V_y E / V_{col} E = 0.30252729$

$d = 0.00$

$s = 0.00$

$t = 2 * A_v / (d_c * s) + 4 * t_f / D * (f_{fe} / f_s) = 0.00721126$

$A_v = 78.53982$ , is the area of the circular stirrup

$d_c = D - 2 * \text{cover} - \text{Hoop Diameter} = 340.00$

The term  $2 * t_f / b_w * (f_{fe} / f_s)$  is implemented to account for FRP contribution

where  $f = 2 * t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength

All these variables have already been given in Shear control ratio calculation.

$N_{UD} = 4783.229$

$A_g = 125663.706$

$f'_c E = 20.00$

$f_{yt} E = f_{yl} E = 444.44$

$\rho_l = \text{Area\_Tot\_Long\_Rein} / (A_g) = 0.0243$

$f'_c E = 20.00$

End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (b)

**Calculation No. 15**

column C1, Floor 1

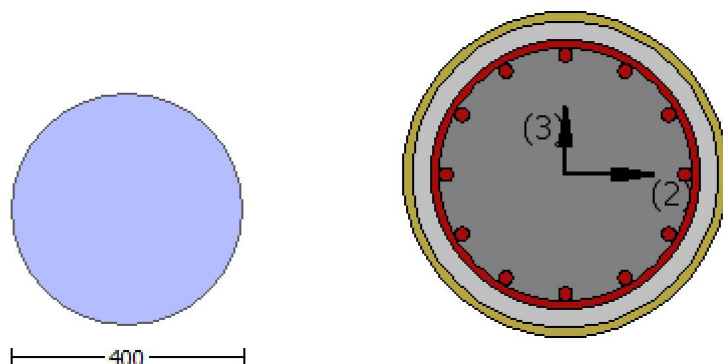
Limit State: Collapse Prevention (data interpolation between analysis steps 2 and 3)

Analysis: Uniform +X

Check: Shear capacity VRd

Edge: End

Local Axis: (3)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (b)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE 41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 16.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 400.00$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of  $\gamma$  for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE 41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE 41-17).

Existing material: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material: Steel Strength,  $f_s = f_{sm} = 444.44$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length  $l_o = l_b = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $\epsilon_{fu} = 0.01$

Number of directions,  $NoDir = 1$

Fiber orientations, bi: 0.00°  
Number of layers, NL = 1  
Radius of rounding corners, R = 40.00

#### Stepwise Properties

EDGE -A-  
Bending Moment, Ma = 6.2300792E-010  
Shear Force, Va = -1.7064698E-013  
EDGE -B-  
Bending Moment, Mb = -1.1086227E-010  
Shear Force, Vb = 1.7064698E-013  
BOTH EDGES  
Axial Force, F = -4783.229  
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension: Aslt = 0.00  
-Compression: Aslc = 3053.628  
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension: Asl,ten = 1017.876  
-Compression: Asl,com = 1017.876  
-Middle: Asl,mid = 1017.876  
Mean Diameter of Tension Reinforcement, DbL,ten = 18.00

Existing component: From table 7-7, ASCE 41\_17: Final Shear Capacity VR = \*Vn = 286339.856  
Vn ((10.3), ASCE 41-17) = knl\*VCol0 = 286339.856  
VCol = 286339.856  
knl = 1.00  
displacement\_ductility\_demand = 0.00

NOTE: In expression (10-3) 'Vs = Av\*fy\*d/s' is replaced by 'Vs+ f\*Vf'  
where Vf is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)  
fc' = 16.00, but  $fc'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
M/Vd = 2.00  
Mu = 1.1086227E-010  
Vu = 1.7064698E-013  
d = 0.8\*D = 320.00  
Nu = 4783.229  
Ag = 125663.706  
From (11.5.4.8), ACI 318-14: Vs = 0.00  
Av = /2\*A\_stirrup = 123370.055  
fy = 400.00  
s = 360.00  
Vs is multiplied by Col = 0.00  
s/d = 1.125  
Vf ((11-3)-(11.4), ACI 440) = 194961.134  
f = 0.95, for fully-wrapped sections  
wf/sf = 1 (FRP strips adjacent to one another).  
In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a)\sin a$  which is more a generalised expression,  
where is the angle of the crack direction (see KANEPE).  
This later relation, considered as a function Vf( , ), is implemented for every different fiber orientation ai,  
as well as for 2 crack directions, = 45° and = -45° to take into consideration the cyclic seismic loading.  
orientation 1: 1 = b1 + 90° = 90.00  
Vf = Min(|Vf(45, 1)|, |Vf(-45,a1)|), with:  
total thickness per orientation, tf1 = NL\*t/NoDir = 1.016  
dfv = d (figure 11.2, ACI 440) = 370.00  
ffe ((11-5), ACI 440) = 259.312  
Ef = 64828.00  
fe = 0.004, from (11.6a), ACI 440  
with fu = 0.01  
From (11-11), ACI 440: Vs + Vf <= 213705.936  
bw\*d = \*d\*d/4 = 80424.772

displacement\_ductility\_demand is calculated as  $\delta / y$

- Calculation of  $\delta / y$  for END B -  
for rotation axis 2 and integ. section (b)

From analysis, chord rotation  $\theta = 1.5066232E-020$   
 $y = (M_y * L_s / 3) / E_{eff} = 0.00931733$  ((4.29), Biskinis Phd))  
 $M_y = 1.4766E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 1500.00  
From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 7.9240E+012$   
 $factor = 0.30$   
 $A_g = 125663.706$   
 $f_c' = 20.00$   
 $N = 4783.229$   
 $E_c * I_g = 2.6413E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\delta$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \min(M_{y\_ten}, M_{y\_com}) = 1.4766E+008$   
 $y = 8.1688588E-006$   
 $M_{y\_ten}$  (8c) = 1.4766E+008  
 $\delta_{ten}$  (7c) = 72.4069  
error of function (7c) = 0.00117279  
 $M_{y\_com}$  (8d) = 3.7493E+008  
 $\delta_{com}$  (7d) = 69.91139  
error of function (7d) = 0.00300901  
with ((10.1), ASCE 41-17)  $e_y = \min(e_y, 1.25 * e_y * (l_b / l_d)^{2/3}) = 0.0022222$   
 $e_{co} = 0.002$   
 $a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00157746$   
 $N = 4783.229$   
 $A_c = 125663.706$   
((10.1), ASCE 41-17)  $\phi = \min(\phi, 1.25 * \phi * (l_b / l_d)^{2/3}) = 0.44757577$   
with  $f_c' * ((12.3), ACI 440) = 24.12975$   
 $f_c = 20.00$   
 $f_l = 1.3173$   
 $k = 1$   
Effective FRP thickness,  $t_f = N L * t * \cos(\theta_1) = 1.016$   
 $e_{fe}$  ((12.5) and (12.7)) = 0.004  
 $f_u = 0.01$   
 $E_f = 64828.00$

Calculation of ratio  $l_b / l_d$

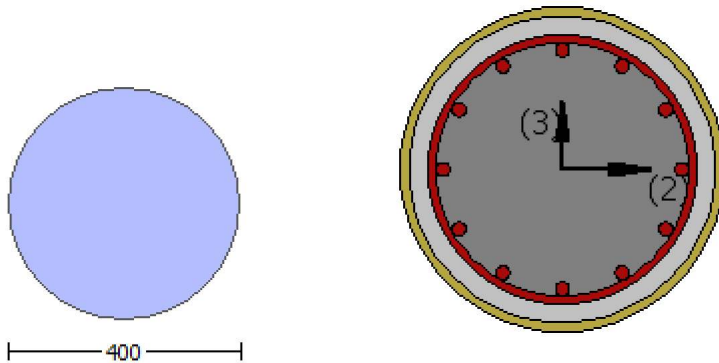
Lap Length:  $l_d / l_{d,min} = 0.33490748$   
 $l_b = 300.00$   
 $l_d = 895.7698$   
Calculation of  $l$  according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $\phi = 1$   
 $d_b = 18.00$   
Mean strength value of all re-bars:  $f_y = 444.44$   
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $c_b = 25.00$

$K_{tr} = 1.14232$   
 $A_{tr} = \frac{1}{2} * \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

-----  
 End Of Calculation of Shear Capacity for element: column CC1 of floor 1  
 At local axis: 3  
 Integration Section: (b)  
 -----

## Calculation No. 16

column C1, Floor 1  
 Limit State: Collapse Prevention (data interpolation between analysis steps 2 and 3)  
 Analysis: Uniform +X  
 Check: Chord rotation capacity (  $\phi$  )  
 Edge: End  
 Local Axis: (3)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At Shear local axis: 3  
 (Bending local axis: 2)  
 Section Type: rccs

Constant Properties

-----  
 Knowledge Factor,  $\gamma = 1.00$   
 Mean strength values are used for both shear and moment calculations.  
 Consequently:  
 Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$   
 Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.44$   
 Concrete Elasticity,  $E_c = 21019.039$   
 Steel Elasticity,  $E_s = 200000.00$   
 #####  
 Note: Especially for the calculation of moment strengths,  
 the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14  
 Existing material: Steel Strength,  $f_s = 1.25 * f_{sm} = 555.55$   
 #####



Diameter,  $D = 400.00$   
 Cover Thickness,  $c = 25.00$   
 Mean Confinement Factor overall section = 1.84055  
 Element Length,  $L = 3000.00$   
 Primary Member  
 Ribbed Bars  
 Ductile Steel  
 Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Lap Length  $l_o = 300.00$   
 FRP Wrapping Data  
 Type: Carbon  
 Cured laminate properties (design values)  
 Thickness,  $t = 1.016$   
 Tensile Strength,  $f_{fu} = 1055.00$   
 Tensile Modulus,  $E_f = 64828.00$   
 Elongation,  $ε_{fu} = 0.01$   
 Number of directions,  $N_{oDir} = 1$   
 Fiber orientations,  $b_i: 0.00^\circ$   
 Number of layers,  $N_L = 1$   
 Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

At local axis: 3  
 EDGE -A-  
 Shear Force,  $V_a = 1.2107607E-031$   
 EDGE -B-  
 Shear Force,  $V_b = -1.2107607E-031$   
 BOTH EDGES  
 Axial Force,  $F = -4771.233$   
 Longitudinal Reinforcement Area Distribution (in 2 divisions)  
     -Tension:  $A_{st} = 0.00$   
     -Compression:  $A_{sc} = 3053.628$   
 Longitudinal Reinforcement Area Distribution (in 3 divisions)  
     -Tension:  $A_{st,ten} = 1017.876$   
     -Compression:  $A_{st,com} = 1017.876$   
     -Middle:  $A_{st,mid} = 1017.876$

Calculation of Shear Capacity ratio ,  $V_e/V_r = 0.30252729$   
 Member Controlled by Flexure ( $V_e/V_r < 1$ )  
 Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 90202.132$   
 with  
 $M_{pr1} = \text{Max}(Mu_{1+}, Mu_{1-}) = 1.3530E+008$   
 $Mu_{1+} = 1.3530E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination  
 $Mu_{1-} = 1.3530E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination  
 $M_{pr2} = \text{Max}(Mu_{2+}, Mu_{2-}) = 1.3530E+008$   
 $Mu_{2+} = 1.3530E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination  
 $Mu_{2-} = 1.3530E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

#### Calculation of $Mu_{1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$   
 $M_u = 1.3530E+008$   
 $= 0.90757121$

$\rho = 0.80580716$   
 error of function (3.68), Biskinis Phd = 28928.286  
 From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$   
 conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
 From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

#### Calculation of ratio $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
 Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $= 1$   
 $db = 18.00$   
 Mean strength value of all re-bars:  $f_y = 555.55$   
 $f_c' = 20.00$ , but  $f_c^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \sqrt{2} \cdot \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

#### Calculation of $\mu_{u1}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$   
 $\mu_u = 1.3530E+008$

$= 0.90757121$   
 $\rho = 0.80580716$   
 error of function (3.68), Biskinis Phd = 28928.286  
 From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$   
 conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
 From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

#### Calculation of ratio $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
 Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $= 1$   
 $db = 18.00$

Mean strength value of all re-bars:  $f_y = 555.55$   
 $f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \frac{1}{2} \times \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

#### Calculation of $\mu_{2+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 1.3530E+008$

$\mu = 0.90757121$   
 $\mu' = 0.80580716$   
error of function (3.68), Biskinis Phd = 28928.286  
From 5A.2, TBDY:  $f_{cc} = f'_c \times c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \times \text{Min}(1, 1.25 \times (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $\mu = \mu' \times \text{Min}(1, 1.25 \times (l_b/l_d)^{2/3}) = 0.28805051$

#### Calculation of ratio $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $\mu = 1$   
 $db = 18.00$   
Mean strength value of all re-bars:  $f_y = 555.55$   
 $f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \frac{1}{2} \times \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

#### Calculation of $\mu_{2-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 1.3530E+008$

$\mu = 0.90757121$   
 $\mu' = 0.80580716$   
error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY:  $f_{cc} = f_c' \cdot c = 36.81095$   
 conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
 From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

Calculation of ratio  $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
 Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $= 1$   
 $d_b = 18.00$   
 Mean strength value of all re-bars:  $f_y = 555.55$   
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \frac{1}{2} \cdot \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 298161.965$

Calculation of Shear Strength at edge 1,  $V_{r1} = 298161.965$   
 $V_{r1} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{Col0}$   
 $V_{Col0} = 298161.965$   
 $k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)  
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $\mu_u = 1.2899604E-011$   
 $\nu_u = 1.2107607E-031$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 0.00$   
 $A_v = \frac{1}{2} \cdot A_{\text{stirrup}} = 123370.055$   
 $f_y = 444.44$   
 $s = 360.00$   
 $V_s$  is multiplied by  $\text{Col} = 0.00$   
 $s/d = 1.125$   
 $V_f$  ((11-3)-(11.4), ACI 440) =  $194961.134$   
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression, where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta)$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta_1 = \theta_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, a1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}((11-5), \text{ACI 440}) = 259.312$   
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $b_w \cdot d = \frac{1}{4} \cdot d \cdot d = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 298161.965$   
 $V_{r2} = V_{Col}((10.3), \text{ASCE 41-17}) = k_{nl} \cdot V_{Col0}$   
 $V_{Col0} = 298161.965$   
 $k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ '  
 where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)  
 $f'_c = 20.00$ , but  $f'_c^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $\mu_u = 1.2899604E-011$   
 $\nu_u = 1.2107607E-031$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 0.00$   
 $A_v = \frac{1}{2} \cdot A_{stirrup} = 123370.055$   
 $f_y = 444.44$   
 $s = 360.00$   
 $V_s$  is multiplied by  $\text{Col} = 0.00$   
 $s/d = 1.125$   
 $V_f((11-3)-(11.4), \text{ACI 440}) = 194961.134$   
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = b1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, a1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}((11-5), \text{ACI 440}) = 259.312$   
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $b_w \cdot d = \frac{1}{4} \cdot d \cdot d = 80424.772$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At local axis: 3

Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At Shear local axis: 2  
 (Bending local axis: 3)  
 Section Type: rccs

Constant Properties

Knowledge Factor,  $\phi = 1.00$   
 Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.44$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,  
the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 555.55$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.84055

Element Length,  $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length  $l_o = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $ε_{fu} = 0.01$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i: 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

-----

Stepwise Properties

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At local axis: 2

EDGE -A-

Shear Force,  $V_a = -7.4135260E-048$

EDGE -B-

Shear Force,  $V_b = 7.4135260E-048$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{sl,t} = 0.00$

-Compression:  $A_{sl,c} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{sl,ten} = 1017.876$

-Compression:  $A_{sl,com} = 1017.876$

-Middle:  $A_{sl,mid} = 1017.876$

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Calculation of Shear Capacity ratio ,  $V_e/V_r = 0.30252729$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 90202.132$

with

$M_{pr1} = \text{Max}(\mu_{u1+}, \mu_{u1-}) = 1.3530E+008$

$\mu_{u1+} = 1.3530E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$\mu_{u1-} = 1.3530E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(\mu_{u2+}, \mu_{u2-}) = 1.3530E+008$

$\mu_{u2+} = 1.3530E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination

$\mu_{u2-} = 1.3530E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

#### Calculation of Mu1+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.3530E+008

= 0.90757121  
' = 0.80580716  
error of function (3.68), Biskinis Phd = 28928.286  
From 5A.2, TDY:  $f_{cc} = f_c \cdot c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
=  $\cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

#### Calculation of ratio lb/l\_d

Lap Length:  $l_b/l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
= 1  
 $db = 18.00$   
Mean strength value of all re-bars:  $f_y = 555.55$   
 $f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $cb = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \sqrt{2} \cdot \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

#### Calculation of Mu1-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.3530E+008

= 0.90757121  
' = 0.80580716  
error of function (3.68), Biskinis Phd = 28928.286  
From 5A.2, TDY:  $f_{cc} = f_c \cdot c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
=  $\cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

#### Calculation of ratio $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of  $l_b, \min$  according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$db = 18.00$

Mean strength value of all re-bars:  $f_y = 555.55$

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$cb = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \times \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$

#### Calculation of $\mu_{2+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$\mu = 1.3530E+008$

$= 0.90757121$

$' = 0.80580716$

error of function (3.68), Biskinis Phd = 28928.286

From 5A.2, TBDY:  $f_{cc} = f'_c \cdot c = 36.81095$

conf. factor  $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \min(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$

$l_b/l_d = 0.26792599$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \cdot \min(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

#### Calculation of ratio $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$

$l_b = 300.00$

$l_d = 1119.712$

Calculation of  $l_b, \min$  according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.

$= 1$

$db = 18.00$

Mean strength value of all re-bars:  $f_y = 555.55$

$f'_c = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$t = 1.00$

$s = 0.80$

$e = 1.00$

$cb = 25.00$

$K_{tr} = 1.14232$

$A_{tr} = \frac{1}{2} \times \text{Area of stirrup} = 123.3701$

$s = 360.00$

$n = 12.00$



## Calculation of Mu2-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.3530E+008

= 0.90757121  
' = 0.80580716  
error of function (3.68), Biskinis Phd = 28928.286  
From 5A.2, TBDY:  $f_{cc} = f_c' \cdot c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE 41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 288.6089$   
 $l_b/l_d = 0.26792599$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
=  $\cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.28805051$

## Calculation of ratio $l_b/l_d$

Lap Length:  $l_b/l_d = 0.26792599$   
 $l_b = 300.00$   
 $l_d = 1119.712$   
Calculation of  $l_b$ , min according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
= 1  
 $d_b = 18.00$   
Mean strength value of all re-bars:  $f_y = 555.55$   
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $t = 1.00$   
 $s = 0.80$   
 $e = 1.00$   
 $c_b = 25.00$   
 $K_{tr} = 1.14232$   
 $A_{tr} = \pi/2 \cdot \text{Area of stirrup} = 123.3701$   
 $s = 360.00$   
 $n = 12.00$

## Calculation of Shear Strength $V_r = \text{Min}(V_{r1}, V_{r2}) = 298161.965$

Calculation of Shear Strength at edge 1,  $V_{r1} = 298161.965$   
 $V_{r1} = V_{CoI} \text{ ((10.3), ASCE 41-17)} = k_{nl} \cdot V_{CoI0}$   
 $V_{CoI0} = 298161.965$   
 $k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)  
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $M_u = 5.9321525E-012$   
 $V_u = 7.4135260E-048$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
From (11.5.4.8), ACI 318-14:  $V_s = 0.00$   
 $A_v = \pi/2 \cdot A_{\text{stirrup}} = 123370.055$   
 $f_y = 444.44$   
 $s = 360.00$

$V_s$  is multiplied by  $Col = 0.00$   
 $s/d = 1.125$   
 $V_f ((11-3)-(11.4), ACI 440) = 194961.134$   
 $f = 0.95$ , for fully-wrapped sections  
 $wf/sf = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a)\sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f( , )$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $= 45^\circ$  and  $= -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $1 = b1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, a1)|)$ , with:  
 total thickness per orientation,  $tf1 = NL * t / NoDir = 1.016$   
 $dfv = d$  (figure 11.2, ACI 440) = 370.00  
 $ffe ((11-5), ACI 440) = 259.312$   
 $Ef = 64828.00$   
 $fe = 0.004$ , from (11.6a), ACI 440  
 with  $fu = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $bw * d = *d * d / 4 = 80424.772$

Calculation of Shear Strength at edge 2,  $Vr2 = 298161.965$   
 $Vr2 = VCol ((10.3), ASCE 41-17) = knl * VCol0$   
 $VCol0 = 298161.965$   
 $knl = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ '  
 where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)  
 $fc' = 20.00$ , but  $fc^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $Mu = 5.9321525E-012$   
 $Vu = 7.4135260E-048$   
 $d = 0.8 * D = 320.00$   
 $Nu = 4771.233$   
 $Ag = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 0.00$   
 $A_v = /2 * A_{stirrup} = 123370.055$   
 $f_y = 444.44$   
 $s = 360.00$

$V_s$  is multiplied by  $Col = 0.00$   
 $s/d = 1.125$   
 $V_f ((11-3)-(11.4), ACI 440) = 194961.134$   
 $f = 0.95$ , for fully-wrapped sections  
 $wf/sf = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a)\sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f( , )$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $= 45^\circ$  and  $= -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $1 = b1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, a1)|)$ , with:  
 total thickness per orientation,  $tf1 = NL * t / NoDir = 1.016$   
 $dfv = d$  (figure 11.2, ACI 440) = 370.00  
 $ffe ((11-5), ACI 440) = 259.312$   
 $Ef = 64828.00$   
 $fe = 0.004$ , from (11.6a), ACI 440  
 with  $fu = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $bw * d = *d * d / 4 = 80424.772$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At local axis: 2

Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (b)

Section Type: rccs

#### Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE 41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.44$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Lap Length  $l_b = 300.00$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $\epsilon_{fu} = 0.01$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i: 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

Bending Moment,  $M = 2684.487$

Shear Force,  $V_2 = 4486.792$

Shear Force,  $V_3 = 1.7064698E-013$

Axial Force,  $F = -4783.229$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{st} = 0.00$

-Compression:  $A_{sc} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{st,ten} = 1017.876$

-Compression:  $A_{st,com} = 1017.876$

-Middle:  $A_{st,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $D_{bL} = 18.00$

Existing component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_R = \gamma \cdot u = 0.04386347$

$u = \gamma \cdot u + p = 0.04386347$

- Calculation of  $\gamma$  -

$\gamma = (M_y \cdot L_s / 3) / E_{eff} = 0.00186347 ((4.29), \text{Biskinis Phd})$

$M_y = 1.4766E+008$

$L_s = M/V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) = 300.00

From table 10.5, ASCE 41\_17:  $E_{eff} = \text{factor} \cdot E_c \cdot I_g = 7.9240E+012$

factor = 0.30

Ag = 125663.706  
fc' = 20.00  
N = 4783.229  
Ec\*Ig = 2.6413E+013

#### Calculation of Yielding Moment My

Calculation of  $\phi_y$  and My according to (7) - (8) in Biskinis and Fardis

My = Min(My\_ten, My\_com) = 1.4766E+008  
 $\phi_y = 8.1688588E-006$   
My\_ten (8c) = 1.4766E+008  
 $\phi_{ten} (7c) = 72.4069$   
error of function (7c) = 0.00117279  
My\_com (8d) = 3.7493E+008  
 $\phi_{com} (7d) = 69.91139$   
error of function (7d) = 0.00300901  
with ((10.1), ASCE 41-17)  $\phi_y = \text{Min}(\phi_y, 1.25 * \phi_y * (l_b/l_d)^{2/3}) = 0.0022222$   
 $\phi_{co} = 0.002$   
 $\phi_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
d1 = 44.00  
R = 200.00  
v = 0.00157746  
N = 4783.229  
Ac = 125663.706  
((10.1), ASCE 41-17)  $\phi_y = \text{Min}(\phi_y, 1.25 * \phi_y * (l_b/l_d)^{2/3}) = 0.44757577$   
with  $\phi_{co}^*$  ((12.3), ACI 440) = 24.12975  
fc = 20.00  
fl = 1.3173  
k = 1  
Effective FRP thickness,  $t_f = NL * t * \cos(b1) = 1.016$   
 $\phi_{fe}$  ((12.5) and (12.7)) = 0.004  
fu = 0.01  
Ef = 64828.00

#### Calculation of ratio $l_b/l_d$

Lap Length:  $l_d/l_d, \min = 0.33490748$   
 $l_b = 300.00$   
 $l_d = 895.7698$   
Calculation of l according to (25.4.3.2), ACI 318-14, using mean values for all the section re-bars.  
 $\phi = 1$   
db = 18.00  
Mean strength value of all re-bars:  $f_y = 444.44$   
fc' = 20.00, but  $fc'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
t = 1.00  
s = 0.80  
e = 1.00  
cb = 25.00  
Ktr = 1.14232  
Atr =  $\sqrt{2} * \text{Area of stirrup} = 123.3701$   
s = 360.00  
n = 12.00

#### - Calculation of $\phi_p$ -

From table 10-9:  $\phi_p = 0.042$

with:

- Columns controlled by inadequate development or splicing along the clear height because  $l_b/l_d < 1$
- shear control ratio  $V_y E / V_{co} I_{OE} = 0.30252729$   
d = 0.00  
s = 0.00

$$t = 2 \cdot A_v / (d_c \cdot s) + 4 \cdot t_f / D \cdot (f_{fe} / f_s) = 0.00721126$$

$A_v = 78.53982$ , is the area of the circular stirrup

$d_c = D - 2 \cdot \text{cover} - \text{Hoop Diameter} = 340.00$

The term  $2 \cdot t_f / b_w \cdot (f_{fe} / f_s)$  is implemented to account for FRP contribution

where  $f = 2 \cdot t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength

All these variables have already been given in Shear control ratio calculation.

$$N_{UD} = 4783.229$$

$$A_g = 125663.706$$

$$f_{cE} = 20.00$$

$$f_{yE} = f_{yI} = 444.44$$

$$p_l = \text{Area\_Tot\_Long\_Rein} / (A_g) = 0.0243$$

$$f_{cE} = 20.00$$

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End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (b)

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