

# Detailed Member Calculations

**Units: N&mm**

**Regulation: ASCE 41-17**

## Calculation No. 1

column C1, Floor 1

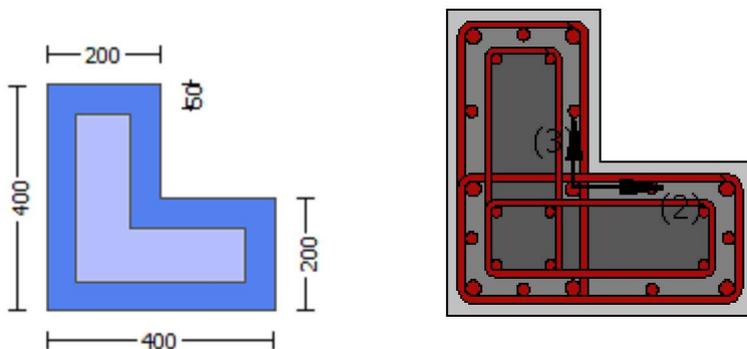
Limit State: Operational Level (data interpolation between analysis steps 50 and 51)

Analysis: Uniform +X

Check: Shear capacity  $V_{Rd}$

Edge: Start

Local Axis: (2)



Start Of Calculation of Shear Capacity for element: column JLC1 of floor 1

At local axis: 2

Integration Section: (a)

Section Type: rcjlc

Constant Properties

Knowledge Factor,  $\gamma = 0.90$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE 41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Jacket

New material of Secondary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 20.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 500.00$

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Concrete Elasticity, Ec = 25742.96
Steel Elasticity, Es = 200000.00
Existing Column
Existing material of Secondary Member: Concrete Strength, fc = fc_lower_bound = 12.00
Existing material of Secondary Member: Steel Strength, fs = fs_lower_bound = 400.00
Concrete Elasticity, Ec = 19940.411
Steel Elasticity, Es = 200000.00
#####
Note: Especially for the calculation of  $\gamma$  for displacement ductility demand,
the expected (mean value) strengths are used (7.5.1.3, ASCE 41-17) because bending is considered as
Deformation-Controlled Action (Table C7-1, ASCE 41-17).
Jacket
New material: Concrete Strength, fc = fcm = 30.00
New material: Steel Strength, fs = fsm = 625.00
Existing Column
Existing material: Concrete Strength, fc = fcm = 18.00
Existing material: Steel Strength, fs = fsm = 500.00
#####
Max Height, Hmax = 400.00
Min Height, Hmin = 200.00
Max Width, Wmax = 400.00
Min Width, Wmin = 200.00
Jacket Thickness, tj = 50.00
Cover Thickness, c = 15.00
Element Length, L = 3000.00
Secondary Member
Smooth Bars
Ductile Steel
Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)
Longitudinal Bars With Ends Lapped Starting at the End Sections
Adequate Lap Length ( $l_o/l_{o,u,min} = l_b/l_d >= 1$ )
No FRP Wrapping
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Stepwise Properties
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EDGE -A-
Bending Moment, Ma = -3.4056E+008
Shear Force, Va = -97644.268
EDGE -B-
Bending Moment, Mb = -7.2803E+007
Shear Force, Vb = 97644.268
BOTH EDGES
Axial Force, F = -1.8003E+006
Longitudinal Reinforcement Area Distribution (in 2 divisions)
  -Tension: Asl,t = 2161.416
  -Compression: Asl,c = 3191.858
Longitudinal Reinforcement Area Distribution (in 3 divisions)
  -Tension: Asl,ten = 2208.54
  -Compression: Asl,com = 1137.257
  -Middle: Asl,mid = 2007.478
Mean Diameter of Tension Reinforcement, DbL,ten = 16.60
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Existing component: From table 7-7, ASCE 41_17: Final Shear Capacity VR =  $\phi V_n = 258140.283$ 
Vn ((10.3), ASCE 41-17) = knl*VCol = 286822.537
VCol = 286822.537
knl = 1.00
displacement_ductility_demand = 1.53373
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NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + \phi \cdot V_f$ '
where Vf is the contribution of FRPs (11.3), ACI 440).
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= 1 (normal-weight concrete)
Mean concrete strength: fc' = (fc'_jacket*Area_jacket + fc'_core*Area_core)/Area_section = 16.66667, but fc'^0.5 <=

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8.3 MPa (22.5.3.1, ACI 318-14)

$$M/Vd = 4.00$$

$$\mu_u = 3.4056E+008$$

$$V_u = 97644.268$$

$$d = 0.8 \cdot h = 320.00$$

$$N_u = 1.8003E+006$$

$$A_g = 80000.00$$

$$\text{From (11.5.4.8), ACI 318-14: } V_s = V_{s,jacket} + V_{s,core} = 376991.118$$

where:

$$V_{s,jacket} = V_{s,j1} + V_{s,j2} = 376991.118$$

$V_{s,j1} = 125663.706$  is calculated for section web jacket, with:

$$d = 160.00$$

$$A_v = 157079.633$$

$$f_y = 500.00$$

$$s = 100.00$$

$V_{s,j1}$  is multiplied by  $Col,j1 = 1.00$

$$s/d = 0.625$$

$V_{s,j2} = 251327.412$  is calculated for section flange jacket, with:

$$d = 320.00$$

$$A_v = 157079.633$$

$$f_y = 500.00$$

$$s = 100.00$$

$V_{s,j2}$  is multiplied by  $Col,j2 = 1.00$

$$s/d = 0.3125$$

$$V_{s,core} = V_{s,c1} + V_{s,c2} = 0.00$$

$V_{s,c1} = 0.00$  is calculated for section web core, with:

$$d = 80.00$$

$$A_v = 100530.965$$

$$f_y = 400.00$$

$$s = 250.00$$

$V_{s,c1}$  is multiplied by  $Col,c1 = 0.00$

$$s/d = 3.125$$

$V_{s,c2} = 0.00$  is calculated for section flange core, with:

$$d = 240.00$$

$$A_v = 100530.965$$

$$f_y = 400.00$$

$$s = 250.00$$

$V_{s,c2}$  is multiplied by  $Col,c2 = 0.00$

$$s/d = 1.04167$$

$$V_f \text{ ((11-3)-(11.4), ACI 440)} = 0.00$$

$$\text{From (11-11), ACI 440: } V_s + V_f \leq 173568.578$$

$$b_w = 200.00$$

displacement ductility demand is calculated as  $\delta / y$

- Calculation of  $\delta / y$  for END A -  
for rotation axis 3 and integ. section (a)

From analysis, chord rotation  $\theta = 0.02234349$

$$y = (M_y \cdot L_s / 3) / E_{eff} = 0.0145681 \text{ ((4.29), Biskinis Phd)}$$

$$M_y = 3.1266E+008$$

$$L_s = M/V \text{ (with } L_s > 0.1 \cdot L \text{ and } L_s < 2 \cdot L) = 3487.751$$

$$\text{From table 10.5, ASCE 41_17: } E_{eff} = \text{factor} \cdot E_c \cdot I_g = 2.4951E+013$$

$$\text{factor} = 0.70$$

$$A_g = 120000.00$$

$$\text{Mean concrete strength: } f'_c = (f'_{c,jacket} \cdot \text{Area}_{jacket} + f'_{c,core} \cdot \text{Area}_{core}) / \text{Area}_{section} = 25.00$$

$$N = 1.8003E+006$$

$$E_c \cdot I_g = E_{c,jacket} \cdot I_{g,jacket} + E_{c,core} \cdot I_{g,core} = 3.5645E+013$$

Calculation of Yielding Moment  $M_y$

Calculation of  $\delta / y$  and  $M_y$  according to Annex 7 -

$y = \text{Min}(y_{\text{ten}}, y_{\text{com}})$   
 $y_{\text{ten}} = 1.9271316\text{E-}005$   
 with  $f_y = 596.2441$   
 $d = 367.00$   
 $y = 0.57848146$   
 $A = 0.11407012$   
 $B = 0.08752406$   
 with  $pt = 0.0300891$   
 $pc = 0.01549396$   
 $pv = 0.02734983$   
 $N = 1.8003\text{E+}006$   
 $b = 200.00$   
 $" = 0.08991826$   
 $y_{\text{comp}} = 7.6825402\text{E-}006$   
 with  $f_c = 30.00$   
 $E_c = 25742.96$   
 $y = 0.74398532$   
 $A = 0.01446815$   
 $B = 0.04638683$   
 with  $E_s = 200000.00$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

End Of Calculation of Shear Capacity for element: column JLC1 of floor 1

At local axis: 2

Integration Section: (a)

## Calculation No. 2

column C1, Floor 1

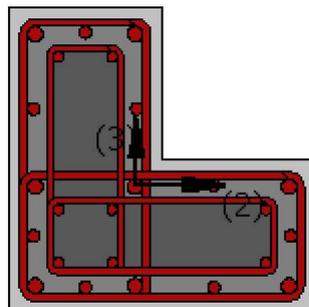
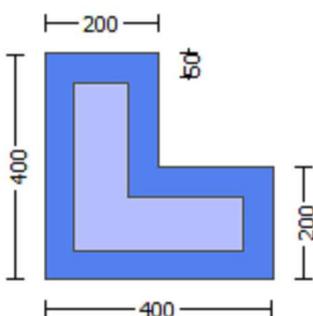
Limit State: Operational Level (data interpolation between analysis steps 50 and 51)

Analysis: Uniform +X

Check: Chord rotation capacity ( $\theta_r$ )

Edge: Start

Local Axis: (2)



Start Of Calculation of Shear Capacity ratio for element: column JLC1 of floor 1

At Shear local axis: 3  
(Bending local axis: 2)  
Section Type: rcjlc

Constant Properties

Knowledge Factor,  $\gamma = 0.90$   
Mean strength values are used for both shear and moment calculations.  
Consequently:

Jacket  
New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 30.00$   
New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 625.00$   
Concrete Elasticity,  $E_c = 25742.96$   
Steel Elasticity,  $E_s = 200000.00$

Existing Column  
Existing material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 18.00$   
Existing material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 500.00$   
Concrete Elasticity,  $E_c = 19940.411$   
Steel Elasticity,  $E_s = 200000.00$

#####  
Note: Especially for the calculation of moment strengths,  
the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Jacket  
New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 781.25$   
Existing Column  
Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 625.00$

#####  
Max Height,  $H_{max} = 400.00$   
Min Height,  $H_{min} = 200.00$   
Max Width,  $W_{max} = 400.00$   
Min Width,  $W_{min} = 200.00$   
Jacket Thickness,  $t_j = 50.00$   
Cover Thickness,  $c = 15.00$   
Mean Confinement Factor overall section = 1.00  
Element Length,  $L = 3000.00$

Secondary Member  
Smooth Bars  
Ductile Steel  
Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Adequate Lap Length ( $l_o/l_{ou, min} >= 1$ )  
No FRP Wrapping

Stepwise Properties

At local axis: 3  
EDGE -A-  
Shear Force,  $V_a = 2.3476251E-006$   
EDGE -B-  
Shear Force,  $V_b = -2.3476251E-006$   
BOTH EDGES  
Axial Force,  $F = -1.8026E+006$   
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $A_{st} = 0.00$   
-Compression:  $A_{sc} = 5353.274$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $A_{st, ten} = 2208.54$   
-Compression:  $A_{sc, com} = 1137.257$   
-Middle:  $A_{sl, mid} = 2007.478$

Calculation of Shear Capacity ratio ,  $V_e/V_r = 1.44151$

Member Controlled by Shear ( $V_e/V_r > 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 488890.189$

with

$M_{pr1} = \text{Max}(M_{u1+} , M_{u1-}) = 7.3334E+008$

$M_{u1+} = 7.3334E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$M_{u1-} = 6.3768E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(M_{u2+} , M_{u2-}) = 7.3334E+008$

$M_{u2+} = 7.3334E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

$M_{u2-} = 6.3768E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

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Calculation of  $M_{u1+}$   
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Calculation of ultimate curvature  $\phi_u$  according to 4.1, Biskinis/Fardis 2013:

$\phi_u = 4.2207094E-005$

$M_u = 7.3334E+008$   
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with full section properties:

$b = 400.00$

$d = 367.00$

$d' = 33.00$

$v = 0.40931126$

$N = 1.8026E+006$

$f_c = 30.00$

$\alpha_1$  (5A.5, TBDY) = 0.002

Final value of  $\phi_u$ :  $\phi_u^* = \text{shear\_factor} * \text{Max}(\phi_u, \phi_c) = 0.00925707$

The Shear\_factor is considered equal to 1 (pure moment strength)

From (5.4b), TBDY:  $\phi_u = 0.00925707$

$\phi_{we}$  (5.4c) = 0.0207149

$\alpha_{se}$  ((5.4d), TBDY) =  $(\alpha_{se1} * A_{ext} + \alpha_{se2} * A_{int}) / A_{sec} = 0.33836562$

$\alpha_{se1} = \text{Max}(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).

$\alpha_{se2} (> \alpha_{se1}) = \text{Max}(((A_{conf,max2} - A_{noConf2}) / A_{conf,max2}) * (A_{conf,min2} / A_{conf,max2}), 0) = 0.33836562$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe} , p_{sh,y} * F_{ywe}) = 6.12205$

Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

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 $p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$

$p_{sh1}$  ((5.4d), TBDY) =  $L_{stir1} * A_{stir1} / (A_{sec} * s_1) = 0.00680678$

Lstir1 (Length of stirrups along Y) = 1040.00  
Astir1 (stirrups area) = 78.53982  
psh2 (5.4d) = Lstir2\*Astir2/(Asec\*s2) = 0.0012868  
Lstir2 (Length of stirrups along Y) = 768.00  
Astir2 (stirrups area) = 50.26548

psh\_y\*Fywe = psh1\*Fywe1+ps2\*Fywe2 = 6.12205  
psh1 ((5.4d), TBDY) = Lstir1\*Astir1/(Asec\*s1) = 0.00680678  
Lstir1 (Length of stirrups along X) = 1040.00  
Astir1 (stirrups area) = 78.53982  
psh2 ((5.4d), TBDY) = Lstir2\*Astir2/(Asec\*s2) = 0.0012868  
Lstir2 (Length of stirrups along X) = 768.00  
Astir2 (stirrups area) = 50.26548

Asec = 120000.00

s1 = 100.00

s2 = 250.00

fywe1 = 781.25

fywe2 = 625.00

fce = 30.00

From ((5.A5), TBDY), TBDY: cc = 0.002

c = confinement factor = 1.00

y1 = 0.0025

sh1 = 0.008

ft1 = 898.293

fy1 = 748.5775

su1 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb = 1.00

su1 = 0.4\*esu1\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1,ft1,fy1, it is considered  
characteristic value fsy1 = fs1/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (\text{lb}/\text{ld})^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs1 = (fs,jacket\*Asl,ten,jacket + fs,core\*Asl,ten,core)/Asl,ten = 748.5775

with Es1 = (Es,jacket\*Asl,ten,jacket + Es,core\*Asl,ten,core)/Asl,ten = 200000.00

y2 = 0.0025

sh2 = 0.008

ft2 = 886.7403

fy2 = 738.9503

su2 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb,min = 1.00

su2 = 0.4\*esu2\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu2\_nominal = 0.08,

For calculation of esu2\_nominal and y2, sh2,ft2,fy2, it is considered  
characteristic value fsy2 = fs2/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (\text{lb}/\text{ld})^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs2 = (fs,jacket\*Asl,com,jacket + fs,core\*Asl,com,core)/Asl,com = 738.9503

with Es2 = (Es,jacket\*Asl,com,jacket + Es,core\*Asl,com,core)/Asl,com = 200000.00

yv = 0.0025

shv = 0.008

ftv = 894.3662

fyv = 745.3052

suv = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb = 1.00

suv = 0.4\*esuv\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esuv\_nominal = 0.08,

considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY

For calculation of esuv\_nominal and yv, shv,ftv,fyv, it is considered

characteristic value  $f_{sv} = f_{sv}/1.2$ , from table 5.1, TBDY.

$y_1, sh_1, ft_1, fy_1$ , are also multiplied by  $\text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $f_{sv} = (f_{s,jacket} \cdot A_{s,mid,jacket} + f_{s,mid} \cdot A_{s,mid,core}) / A_{s,mid} = 745.3052$

with  $E_{sv} = (E_{s,jacket} \cdot A_{s,mid,jacket} + E_{s,mid} \cdot A_{s,mid,core}) / A_{s,mid} = 200000.00$

1 =  $A_{s,ten} / (b \cdot d) \cdot (f_{s1}/f_c) = 0.37540035$

2 =  $A_{s,com} / (b \cdot d) \cdot (f_{s2}/f_c) = 0.19082108$

$v = A_{s,mid} / (b \cdot d) \cdot (f_{sv}/f_c) = 0.33973286$

and confined core properties:

$b = 360.00$

$d = 347.00$

$d' = 13.00$

$f_{cc}$  (5A.2, TBDY) = 30.00

$cc$  (5A.5, TBDY) = 0.002

$c$  = confinement factor = 1.00

1 =  $A_{s,ten} / (b \cdot d) \cdot (f_{s1}/f_c) = 0.44115251$

2 =  $A_{s,com} / (b \cdot d) \cdot (f_{s2}/f_c) = 0.22424379$

$v = A_{s,mid} / (b \cdot d) \cdot (f_{sv}/f_c) = 0.39923778$

Case/Assumption: Unconfined full section - Steel rupture

' satisfies Eq. (4.3)

---->

$v < v_{s,y2}$  - LHS eq.(4.5) is not satisfied

---->

$v < v_{s,c}$  - RHS eq.(4.5) is not satisfied

---->

Case/Assumption Rejected.

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New Case/Assumption: Unconfined full section - Spalling of concrete cover

' satisfies Eq. (4.4)

---->

$v < v_{s,y1}$  - LHS eq.(4.7) is not satisfied

---->

$v < v_{c,y1}$  - RHS eq.(4.6) is satisfied

---->

$c_u$  (4.10) = 0.59761571

$M_{Rc}$  (4.17) = 7.3334E+008

---->

New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover

In expressions below, the following modifications have been made

-  $b, d, d'$  replaced by geometric parameters of the core:  $b_o, d_o, d'_o$

-  $N, 1, 2, v$  normalised to  $b_o \cdot d_o$ , instead of  $b \cdot d$

-  $f_{cc}, cc$  parameters of confined concrete,  $f_{cc}, cc$ , used in lieu of  $f_c, e_{cu}$

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Subcase: Rupture of tension steel

---->

$v^* < v^*_{s,y2}$  - LHS eq.(4.5) is not satisfied

---->

$v^* < v^*_{s,c}$  - LHS eq.(4.5) is not satisfied

---->

Subcase rejected

---->

New Subcase: Failure of compression zone

---->

$v^* < v^*_{c,y2}$  - LHS eq.(4.6) is not satisfied

---->

$v^* < v^*_{c,y1}$  - RHS eq.(4.6) is not satisfied

---->

$c_u$  (4.11) = 0.62879536

$M_{Ro}$  (4.18) = 4.5866E+008

$M_{Ro} < 0.8 \cdot M_{Rc}$

---->

$u = c_u$  (unconfined full section) = 4.2207094E-005

$\mu = M_{Rc}$

-----

Calculation of ratio  $l_b/l_d$

-----

Adequate Lap Length:  $l_b/l_d \geq 1$

Calculation of Mu1-

Calculation of ultimate curvature  $\mu$  according to 4.1, Biskinis/Fardis 2013:

$$\mu = 4.8346386E-005$$

$$Mu = 6.3768E+008$$

with full section properties:

$$b = 200.00$$

$$d = 367.00$$

$$d' = 33.00$$

$$v = 0.81862252$$

$$N = 1.8026E+006$$

$$f_c = 30.00$$

$$c_o \text{ (5A.5, TBDY)} = 0.002$$

$$\text{Final value of } c_u: c_u^* = \text{shear\_factor} * \text{Max}(c_u, c_o) = 0.00925707$$

The Shear\_factor is considered equal to 1 (pure moment strength)

$$\text{From (5.4b), TBDY: } c_u = 0.00925707$$

$$w_e \text{ (5.4c)} = 0.0207149$$

$$a_{se} \text{ ((5.4d), TBDY)} = (a_{se1} * A_{ext} + a_{se2} * A_{int}) / A_{sec} = 0.33836562$$

$$a_{se1} = \text{Max}(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$a_{se2} (>= a_{se1}) = \text{Max}(((A_{conf,max2} - A_{noConf2}) / A_{conf,max2}) * (A_{conf,min2} / A_{conf,max2}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe}, p_{sh,y} * F_{ywe}) = 6.12205$$

Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

$$p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{s2} * F_{ywe2} = 6.12205$$

$$p_{sh1} \text{ ((5.4d), TBDY)} = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$$

$$L_{stir1} \text{ (Length of stirrups along Y)} = 1040.00$$

$$A_{stir1} \text{ (stirrups area)} = 78.53982$$

$$p_{sh2} \text{ (5.4d)} = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$$

$$L_{stir2} \text{ (Length of stirrups along Y)} = 768.00$$

$$A_{stir2} \text{ (stirrups area)} = 50.26548$$

$$p_{sh,y} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{s2} * F_{ywe2} = 6.12205$$

$$p_{sh1} \text{ ((5.4d), TBDY)} = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$$

$$L_{stir1} \text{ (Length of stirrups along X)} = 1040.00$$

$$A_{stir1} \text{ (stirrups area)} = 78.53982$$

$$p_{sh2} \text{ ((5.4d), TBDY)} = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$$

$$L_{stir2} \text{ (Length of stirrups along X)} = 768.00$$

$$A_{stir2} \text{ (stirrups area)} = 50.26548$$

Asec = 120000.00

s1 = 100.00

s2 = 250.00

fywe1 = 781.25

fywe2 = 625.00

fce = 30.00

From ((5.A.5), TBDY), TBDY: cc = 0.002

c = confinement factor = 1.00

y1 = 0.0025

sh1 = 0.008

ft1 = 886.7403

fy1 = 738.9503

su1 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb = 1.00

su1 = 0.4\*esu1\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1,ft1,fy1, it is considered  
characteristic value fsy1 = fs1/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (\text{lb}/\text{ld})^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs1 = (fsjacket\*Asl,ten,jacket + fs,core\*Asl,ten,core)/Asl,ten = 738.9503

with Es1 = (Es,jacket\*Asl,ten,jacket + Es,core\*Asl,ten,core)/Asl,ten = 200000.00

y2 = 0.0025

sh2 = 0.008

ft2 = 898.293

fy2 = 748.5775

su2 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb,min = 1.00

su2 = 0.4\*esu2\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu2\_nominal = 0.08,

For calculation of esu2\_nominal and y2, sh2,ft2,fy2, it is considered  
characteristic value fsy2 = fs2/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (\text{lb}/\text{ld})^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs2 = (fs,jacket\*Asl,com,jacket + fs,core\*Asl,com,core)/Asl,com = 748.5775

with Es2 = (Es,jacket\*Asl,com,jacket + Es,core\*Asl,com,core)/Asl,com = 200000.00

yv = 0.0025

shv = 0.008

ftv = 894.3662

fyv = 745.3052

suv = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb = 1.00

suv = 0.4\*esuv\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esuv\_nominal = 0.08,

considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY  
For calculation of esuv\_nominal and yv, shv,ftv,fyv, it is considered

characteristic value fsyv = fsv/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (\text{lb}/\text{ld})^{2/3})$ , from 10.3.5, ASCE 41-17.

with fsv = (fs,jacket\*Asl,mid,jacket + fs,mid\*Asl,mid,core)/Asl,mid = 745.3052

with Esv = (Es,jacket\*Asl,mid,jacket + Es,mid\*Asl,mid,core)/Asl,mid = 200000.00

1 = Asl,ten/(b\*d)\*(fs1/fc) = 0.38164216

2 = Asl,com/(b\*d)\*(fs2/fc) = 0.7508007

v = Asl,mid/(b\*d)\*(fsv/fc) = 0.67946571

and confined core properties:

b = 160.00

d = 347.00

d' = 13.00

fcc (5A.2, TBDY) = 30.00

cc (5A.5, TBDY) = 0.002

c = confinement factor = 1.00  
1 =  $Asl,ten/(b*d)*(fs1/fc) = 0.50454853$   
2 =  $Asl,com/(b*d)*(fs2/fc) = 0.99259314$   
v =  $Asl,mid/(b*d)*(fsv/fc) = 0.898285$

Case/Assumption: Unconfined full section - Steel rupture  
' satisfies Eq. (4.3)

---->  
v < vs,y2 - LHS eq.(4.5) is not satisfied

---->  
v < vs,c - RHS eq.(4.5) is not satisfied

---->  
Case/Assumption Rejected.

---->  
New Case/Assumption: Unconfined full section - Spalling of concrete cover  
' satisfies Eq. (4.4)

---->  
v < s,y1 - LHS eq.(4.7) is not satisfied

---->  
v < vc,y1 - RHS eq.(4.6) is satisfied

---->  
cu (4.10) = 0.52172715  
MRc (4.17) = 6.3768E+008

---->  
New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover  
In expressions below, the following modifications have been made

- b, d, d' replaced by geometric parameters of the core: bo, do, d'o
- N, 1, 2, v normalised to bo\*do, instead of b\*d
- - parameters of confined concrete, fcc, cc, used in lieu of fc, ecu

---->  
Subcase: Rupture of tension steel

---->  
v\* < v\*s,y2 - LHS eq.(4.5) is not satisfied

---->  
v\* < v\*s,c - LHS eq.(4.5) is not satisfied

---->  
Subcase rejected

---->  
New Subcase: Failure of compression zone

---->  
v\* < v\*c,y2 - LHS eq.(4.6) is not satisfied

---->  
v\* < v\*c,y1 - RHS eq.(4.6) is not satisfied

---->  
\*cu (4.11) = 0.5604845  
MRo (4.18) = 4.7738E+008

---->  
MRo < 0.8\*MRc

---->  
u = cu (unconfined full section) = 4.8346386E-005  
Mu = MRc

-----  
Calculation of ratio lb/d

-----  
Adequate Lap Length: lb/d >= 1

-----  
-----  
-----

-----  
Calculation of Mu2+

-----  
-----

-----  
Calculation of ultimate curvature u according to 4.1, Biskinis/Fardis 2013:

u = 4.2207094E-005  
Mu = 7.3334E+008

-----  
with full section properties:  
b = 400.00

$$d = 367.00$$

$$d' = 33.00$$

$$v = 0.40931126$$

$$N = 1.8026E+006$$

$$f_c = 30.00$$

$$c_o \text{ (5A.5, TBDY)} = 0.002$$

$$\text{Final value of } c_u: c_u^* = \text{shear\_factor} * \text{Max}(c_u, c_c) = 0.00925707$$

The Shear\_factor is considered equal to 1 (pure moment strength)

$$\text{From (5.4b), TBDY: } c_u = 0.00925707$$

$$w_e \text{ (5.4c)} = 0.0207149$$

$$a_{se} \text{ ((5.4d), TBDY)} = (a_{se1} * A_{ext} + a_{se2} * A_{int}) / A_{sec} = 0.33836562$$

$$a_{se1} = \text{Max}(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$a_{se2} (> a_{se1}) = \text{Max}(((A_{conf,max2} - A_{noConf2}) / A_{conf,max2}) * (A_{conf,min2} / A_{conf,max2}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe}, p_{sh,y} * F_{ywe}) = 6.12205$$

Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

$$p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$$

$$p_{sh1} \text{ ((5.4d), TBDY)} = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$$

$$L_{stir1} \text{ (Length of stirrups along Y)} = 1040.00$$

$$A_{stir1} \text{ (stirrups area)} = 78.53982$$

$$p_{sh2} \text{ (5.4d)} = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$$

$$L_{stir2} \text{ (Length of stirrups along Y)} = 768.00$$

$$A_{stir2} \text{ (stirrups area)} = 50.26548$$

$$p_{sh,y} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$$

$$p_{sh1} \text{ ((5.4d), TBDY)} = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$$

$$L_{stir1} \text{ (Length of stirrups along X)} = 1040.00$$

$$A_{stir1} \text{ (stirrups area)} = 78.53982$$

$$p_{sh2} \text{ ((5.4d), TBDY)} = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$$

$$L_{stir2} \text{ (Length of stirrups along X)} = 768.00$$

$$A_{stir2} \text{ (stirrups area)} = 50.26548$$

$$A_{sec} = 120000.00$$

$$s1 = 100.00$$

$$s2 = 250.00$$

$$f_{ywe1} = 781.25$$

$$f_{ywe2} = 625.00$$

$$f_{ce} = 30.00$$

$$\text{From ((5.A5), TBDY), TBDY: } c_c = 0.002$$

$$c = \text{confinement factor} = 1.00$$

$$y1 = 0.0025$$

$$sh1 = 0.008$$

$$ft1 = 898.293$$

$$fy1 = 748.5775$$

su1 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/ld = 1.00

su1 = 0.4\*esu1\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1,ft1,fy1, it is considered  
characteristic value fsy1 = fs1/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs1 = (fs,jacket\*Asl,ten,jacket + fs,core\*Asl,ten,core)/Asl,ten = 748.5775

with Es1 = (Es,jacket\*Asl,ten,jacket + Es,core\*Asl,ten,core)/Asl,ten = 200000.00

y2 = 0.0025

sh2 = 0.008

ft2 = 886.7403

fy2 = 738.9503

su2 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb,min = 1.00

su2 = 0.4\*esu2\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu2\_nominal = 0.08,

For calculation of esu2\_nominal and y2, sh2,ft2,fy2, it is considered  
characteristic value fsy2 = fs2/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs2 = (fs,jacket\*Asl,com,jacket + fs,core\*Asl,com,core)/Asl,com = 738.9503

with Es2 = (Es,jacket\*Asl,com,jacket + Es,core\*Asl,com,core)/Asl,com = 200000.00

yv = 0.0025

shv = 0.008

ftv = 894.3662

fyv = 745.3052

suv = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/ld = 1.00

suv = 0.4\*esuv\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esuv\_nominal = 0.08,

considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY

For calculation of esuv\_nominal and yv, shv,ftv,fyv, it is considered  
characteristic value fsyv = fsv/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fsv = (fs,jacket\*Asl,mid,jacket + fs,mid\*Asl,mid,core)/Asl,mid = 745.3052

with Esv = (Es,jacket\*Asl,mid,jacket + Es,mid\*Asl,mid,core)/Asl,mid = 200000.00

1 = Asl,ten/(b\*d)\*(fs1/fc) = 0.37540035

2 = Asl,com/(b\*d)\*(fs2/fc) = 0.19082108

v = Asl,mid/(b\*d)\*(fsv/fc) = 0.33973286

and confined core properties:

b = 360.00

d = 347.00

d' = 13.00

fcc (5A.2, TBDY) = 30.00

cc (5A.5, TBDY) = 0.002

c = confinement factor = 1.00

1 = Asl,ten/(b\*d)\*(fs1/fc) = 0.44115251

2 = Asl,com/(b\*d)\*(fs2/fc) = 0.22424379

v = Asl,mid/(b\*d)\*(fsv/fc) = 0.39923778

Case/Assumption: Unconfined full section - Steel rupture

' satisfies Eq. (4.3)

---->

v < vs,y2 - LHS eq.(4.5) is not satisfied

---->

v < vs,c - RHS eq.(4.5) is not satisfied

---->

Case/Assumption Rejected.

---->

New Case/Assumption: Unconfined full section - Spalling of concrete cover  
satisfies Eq. (4.4)

--->  
 $v < s, y1$  - LHS eq.(4.7) is not satisfied

--->  
 $v < v_c, y1$  - RHS eq.(4.6) is satisfied

--->  
 $c_u$  (4.10) = 0.59761571  
 $M_{Rc}$  (4.17) = 7.3334E+008

--->  
New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover

In expressions below, the following modifications have been made

- $b, d, d'$  replaced by geometric parameters of the core:  $b_o, d_o, d'_o$
- $N, 1, 2, v$  normalised to  $b_o * d_o$ , instead of  $b * d$
- $f_c, c_c$  parameters of confined concrete,  $f_{cc}, c_{cc}$  used in lieu of  $f_c, c_c$

--->  
Subcase: Rupture of tension steel

--->  
 $v^* < v^* s, y2$  - LHS eq.(4.5) is not satisfied

--->  
 $v^* < v^* s, c$  - LHS eq.(4.5) is not satisfied

--->  
Subcase rejected

--->  
New Subcase: Failure of compression zone

--->  
 $v^* < v^* c, y2$  - LHS eq.(4.6) is not satisfied

--->  
 $v^* < v^* c, y1$  - RHS eq.(4.6) is not satisfied

--->  
 $*c_u$  (4.11) = 0.62879536  
 $M_{Ro}$  (4.18) = 4.5866E+008

$M_{Ro} < 0.8 * M_{Rc}$

--->  
 $u = c_u$  (unconfined full section) = 4.2207094E-005  
 $M_u = M_{Rc}$

-----  
Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

-----  
Calculation of  $M_u2$ -

-----  
Calculation of ultimate curvature  $u$  according to 4.1, Biskinis/Fardis 2013:

$u = 4.8346386E-005$

$M_u = 6.3768E+008$

-----  
with full section properties:

$b = 200.00$

$d = 367.00$

$d' = 33.00$

$v = 0.81862252$

$N = 1.8026E+006$

$f_c = 30.00$

$c_o$  (5A.5, TBDY) = 0.002

Final value of  $c_u$ :  $c_u^* = \text{shear\_factor} * \text{Max}(c_u, c_c) = 0.00925707$

The Shear\_factor is considered equal to 1 (pure moment strength)

From (5.4b), TBDY:  $c_u = 0.00925707$

$w_e$  (5.4c) = 0.0207149

$a_{se}$  ((5.4d), TBDY) =  $(a_{se1} * A_{ext} + a_{se2} * A_{int}) / A_{sec} = 0.33836562$

$a_{se1} = \text{Max}(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$

The definitions of  $A_{noConf}, A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization

of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length

equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).

$ase2 (>= ase1) = \text{Max}(((A_{conf,max2} - A_{noConf2})/A_{conf,max2}) * (A_{conf,min2}/A_{conf,max2}), 0) = 0.33836562$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length

equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$psh_{min} * F_{ywe} = \text{Min}(psh_x * F_{ywe}, psh_y * F_{ywe}) = 6.12205$

Expression (5.4d) for  $psh_{min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

-----  
 $psh_x * F_{ywe} = psh1 * F_{ywe1} + ps2 * F_{ywe2} = 6.12205$

$psh1$  ((5.4d), TBDY) =  $L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$

$L_{stir1}$  (Length of stirrups along Y) = 1040.00

$A_{stir1}$  (stirrups area) = 78.53982

$psh2$  (5.4d) =  $L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$

$L_{stir2}$  (Length of stirrups along Y) = 768.00

$A_{stir2}$  (stirrups area) = 50.26548

-----  
 $psh_y * F_{ywe} = psh1 * F_{ywe1} + ps2 * F_{ywe2} = 6.12205$

$psh1$  ((5.4d), TBDY) =  $L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$

$L_{stir1}$  (Length of stirrups along X) = 1040.00

$A_{stir1}$  (stirrups area) = 78.53982

$psh2$  ((5.4d), TBDY) =  $L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$

$L_{stir2}$  (Length of stirrups along X) = 768.00

$A_{stir2}$  (stirrups area) = 50.26548

-----  
 $A_{sec} = 120000.00$

$s1 = 100.00$

$s2 = 250.00$

$f_{ywe1} = 781.25$

$f_{ywe2} = 625.00$

$f_{ce} = 30.00$

From ((5.A5), TBDY), TBDY:  $cc = 0.002$

$c = \text{confinement factor} = 1.00$

$y1 = 0.0025$

$sh1 = 0.008$

$ft1 = 886.7403$

$fy1 = 738.9503$

$su1 = 0.032$

using (30) in Biskinis/Fardis (2013) multiplied with  $shear\_factor$  and also multiplied by the  $shear\_factor$  according to 15.7.1.4, with  $Shear\_factor = 1.00$

$lo/lou_{min} = lb/ld = 1.00$

$su1 = 0.4 * esu1_{nominal}$  ((5.5), TBDY) = 0.032

From table 5A.1, TBDY:  $esu1_{nominal} = 0.08$ ,

For calculation of  $esu1_{nominal}$  and  $y1, sh1, ft1, fy1$ , it is considered characteristic value  $fsy1 = fs1/1.2$ , from table 5.1, TBDY.

$y1, sh1, ft1, fy1$ , are also multiplied by  $\text{Min}(1, 1.25 * (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $fs1 = (fs_{jacket} * A_{sl,ten,jacket} + fs_{core} * A_{sl,ten,core}) / A_{sl,ten} = 738.9503$

with  $Es1 = (Es_{jacket} * A_{sl,ten,jacket} + Es_{core} * A_{sl,ten,core}) / A_{sl,ten} = 200000.00$

$y2 = 0.0025$

```

sh2 = 0.008
ft2 = 898.293
fy2 = 748.5775
su2 = 0.032
  using (30) in Biskinis/Fardis (2013) multiplied with shear_factor
  and also multiplied by the shear_factor according to 15.7.1.4, with
  Shear_factor = 1.00
  lo/lou,min = lb/lb,min = 1.00
  su2 = 0.4*esu2_nominal ((5.5), TBDY) = 0.032
  From table 5A.1, TBDY: esu2_nominal = 0.08,
  For calculation of esu2_nominal and y2, sh2,ft2,fy2, it is considered
  characteristic value fsy2 = fs2/1.2, from table 5.1, TBDY.
  y1, sh1,ft1,fy1, are also multiplied by Min(1,1.25*(lb/d)^2/3), from 10.3.5, ASCE 41-17.
  with fs2 = (fs,jacket*Asl,com,jacket + fs,core*Asl,com,core)/Asl,com = 748.5775
  with Es2 = (Es,jacket*Asl,com,jacket + Es,core*Asl,com,core)/Asl,com = 200000.00
yv = 0.0025
shv = 0.008
ftv = 894.3662
fyv = 745.3052
suv = 0.032
  using (30) in Biskinis/Fardis (2013) multiplied with shear_factor
  and also multiplied by the shear_factor according to 15.7.1.4, with
  Shear_factor = 1.00
  lo/lou,min = lb/d = 1.00
  suv = 0.4*esuv_nominal ((5.5), TBDY) = 0.032
  From table 5A.1, TBDY: esuv_nominal = 0.08,
  considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY
  For calculation of esuv_nominal and yv, shv,ftv,fyv, it is considered
  characteristic value fsyv = fsv/1.2, from table 5.1, TBDY.
  y1, sh1,ft1,fy1, are also multiplied by Min(1,1.25*(lb/d)^2/3), from 10.3.5, ASCE 41-17.
  with fsv = (fs,jacket*Asl,mid,jacket + fs,mid*Asl,mid,core)/Asl,mid = 745.3052
  with Esv = (Es,jacket*Asl,mid,jacket + Es,mid*Asl,mid,core)/Asl,mid = 200000.00
1 = Asl,ten/(b*d)*(fs1/fc) = 0.38164216
2 = Asl,com/(b*d)*(fs2/fc) = 0.7508007
v = Asl,mid/(b*d)*(fsv/fc) = 0.67946571
and confined core properties:
b = 160.00
d = 347.00
d' = 13.00
fcc (5A.2, TBDY) = 30.00
cc (5A.5, TBDY) = 0.002
c = confinement factor = 1.00
1 = Asl,ten/(b*d)*(fs1/fc) = 0.50454853
2 = Asl,com/(b*d)*(fs2/fc) = 0.99259314
v = Asl,mid/(b*d)*(fsv/fc) = 0.898285
Case/Assumption: Unconfined full section - Steel rupture
' satisfies Eq. (4.3)
--->
v < vs,y2 - LHS eq.(4.5) is not satisfied
--->
v < vs,c - RHS eq.(4.5) is not satisfied
--->
Case/Assumption Rejected.
--->
New Case/Assumption: Unconfined full section - Spalling of concrete cover
' satisfies Eq. (4.4)
--->
v < s,y1 - LHS eq.(4.7) is not satisfied
--->
v < vc,y1 - RHS eq.(4.6) is satisfied
--->
cu (4.10) = 0.52172715
MRc (4.17) = 6.3768E+008
--->
New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover
In expressions below, the following modifications have been made
- b, d, d' replaced by geometric parameters of the core: bo, do, d'o

```

- N<sub>1</sub>, N<sub>2</sub>, v normalised to b<sub>o</sub>\*d<sub>o</sub>, instead of b\*d
- parameters of confined concrete, f<sub>cc</sub>, c<sub>c</sub>, used in lieu of f<sub>c</sub>, e<sub>cu</sub>

---->

Subcase: Rupture of tension steel

---->

v\* < v\*s<sub>y2</sub> - LHS eq.(4.5) is not satisfied

---->

v\* < v\*s<sub>c</sub> - LHS eq.(4.5) is not satisfied

---->

Subcase rejected

---->

New Subcase: Failure of compression zone

---->

v\* < v\*c<sub>y2</sub> - LHS eq.(4.6) is not satisfied

---->

v\* < v\*c<sub>y1</sub> - RHS eq.(4.6) is not satisfied

---->

\*c<sub>u</sub> (4.11) = 0.5604845

M<sub>Ro</sub> (4.18) = 4.7738E+008

M<sub>Ro</sub> < 0.8\*M<sub>Rc</sub>

---->

u = c<sub>u</sub> (unconfined full section) = 4.8346386E-005

Mu = M<sub>Rc</sub>

-----  
 Calculation of ratio l<sub>b</sub>/l<sub>d</sub>

-----  
 Adequate Lap Length: l<sub>b</sub>/l<sub>d</sub> >= 1

-----  
 Calculation of Shear Strength V<sub>r</sub> = Min(V<sub>r1</sub>, V<sub>r2</sub>) = 339150.739

-----  
 Calculation of Shear Strength at edge 1, V<sub>r1</sub> = 339150.739

V<sub>r1</sub> = V<sub>Col</sub> ((10.3), ASCE 41-17) = k<sub>nl</sub>\*V<sub>Col0</sub>

V<sub>Col0</sub> = 339150.739

k<sub>nl</sub> = 1 (zero step-static loading)

-----  
 NOTE: In expression (10-3) 'V<sub>s</sub> = A<sub>v</sub>\*f<sub>y</sub>\*d/s' is replaced by 'V<sub>s</sub>+ f\*V<sub>f</sub>'  
 where V<sub>f</sub> is the contribution of FRPs (11.3), ACI 440).

-----  
 = 1 (normal-weight concrete)

Mean concrete strength: f<sub>c</sub>' = (f<sub>c</sub>'<sub>jacket</sub>\*Area<sub>jacket</sub> + f<sub>c</sub>'<sub>core</sub>\*Area<sub>core</sub>)/Area<sub>section</sub> = 25.00, but f<sub>c</sub>'<sup>0.5</sup> <= 8.3  
 MPa (22.5.3.1, ACI 318-14)

M/V<sub>d</sub> = 4.00

Mu = 2.0345E+006

Vu = 2.3476251E-006

d = 0.8\*h = 320.00

Nu = 1.8026E+006

Ag = 80000.00

From (11.5.4.8), ACI 318-14: V<sub>s</sub> = V<sub>s,jacket</sub> + V<sub>s,core</sub> = 471238.898

where:

V<sub>s,jacket</sub> = V<sub>s,j1</sub> + V<sub>s,j2</sub> = 471238.898

V<sub>s,j1</sub> = 314159.265 is calculated for section web jacket, with:

d = 320.00

A<sub>v</sub> = 157079.633

f<sub>y</sub> = 625.00

s = 100.00

V<sub>s,j1</sub> is multiplied by Col<sub>j1</sub> = 1.00

s/d = 0.3125

V<sub>s,j2</sub> = 157079.633 is calculated for section flange jacket, with:

d = 160.00

A<sub>v</sub> = 157079.633

f<sub>y</sub> = 625.00

s = 100.00

V<sub>s,j2</sub> is multiplied by Col<sub>j2</sub> = 1.00

$s/d = 0.625$   
 $V_{s,core} = V_{s,c1} + V_{s,c2} = 0.00$   
 $V_{s,c1} = 0.00$  is calculated for section web core, with:  
 $d = 240.00$   
 $A_v = 100530.965$   
 $f_y = 500.00$   
 $s = 250.00$   
 $V_{s,c1}$  is multiplied by  $Col,c1 = 0.00$   
 $s/d = 1.04167$   
 $V_{s,c2} = 0.00$  is calculated for section flange core, with:  
 $d = 80.00$   
 $A_v = 100530.965$   
 $f_y = 500.00$   
 $s = 250.00$   
 $V_{s,c2}$  is multiplied by  $Col,c2 = 0.00$   
 $s/d = 3.125$   
 $V_f ((11-3)-(11.4), ACI 440) = 0.00$   
 From (11-11), ACI 440:  $V_s + V_f \leq 212577.225$   
 $bw = 200.00$

Calculation of Shear Strength at edge 2,  $V_{r2} = 339150.739$   
 $V_{r2} = V_{Col} ((10.3), ASCE 41-17) = knl * V_{Col0}$   
 $V_{Col0} = 339150.739$   
 $knl = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ '  
 where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)  
 Mean concrete strength:  $f'_c = (f'_{c,jacket} * Area_{jacket} + f'_{c,core} * Area_{core}) / Area_{section} = 25.00$ , but  $f'_c^{0.5} \leq 8.3$   
 MPa (22.5.3.1, ACI 318-14)

$M/Vd = 4.00$   
 $\mu_u = 6739.685$   
 $V_u = 2.3476251E-006$   
 $d = 0.8 * h = 320.00$   
 $N_u = 1.8026E+006$   
 $A_g = 80000.00$   
 From (11.5.4.8), ACI 318-14:  $V_s = V_{s,jacket} + V_{s,core} = 471238.898$

where:  
 $V_{s,jacket} = V_{s,j1} + V_{s,j2} = 471238.898$   
 $V_{s,j1} = 314159.265$  is calculated for section web jacket, with:  
 $d = 320.00$   
 $A_v = 157079.633$   
 $f_y = 625.00$   
 $s = 100.00$   
 $V_{s,j1}$  is multiplied by  $Col,j1 = 1.00$   
 $s/d = 0.3125$   
 $V_{s,j2} = 157079.633$  is calculated for section flange jacket, with:  
 $d = 160.00$   
 $A_v = 157079.633$   
 $f_y = 625.00$   
 $s = 100.00$   
 $V_{s,j2}$  is multiplied by  $Col,j2 = 1.00$   
 $s/d = 0.625$   
 $V_{s,core} = V_{s,c1} + V_{s,c2} = 0.00$   
 $V_{s,c1} = 0.00$  is calculated for section web core, with:  
 $d = 240.00$   
 $A_v = 100530.965$   
 $f_y = 500.00$   
 $s = 250.00$   
 $V_{s,c1}$  is multiplied by  $Col,c1 = 0.00$   
 $s/d = 1.04167$   
 $V_{s,c2} = 0.00$  is calculated for section flange core, with:  
 $d = 80.00$   
 $A_v = 100530.965$   
 $f_y = 500.00$

s = 250.00  
Vs,c2 is multiplied by Col,c2 = 0.00  
s/d = 3.125  
Vf ((11-3)-(11.4), ACI 440) = 0.00  
From (11-11), ACI 440: Vs + Vf <= 212577.225  
bw = 200.00

-----  
-----  
End Of Calculation of Shear Capacity ratio for element: column JLC1 of floor 1  
At local axis: 3

-----  
-----  
Start Of Calculation of Shear Capacity ratio for element: column JLC1 of floor 1  
At Shear local axis: 2  
(Bending local axis: 3)  
Section Type: rcjcs

Constant Properties

-----  
Knowledge Factor,  $\gamma = 0.90$   
Mean strength values are used for both shear and moment calculations.  
Consequently:  
Jacket  
New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 30.00$   
New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 625.00$   
Concrete Elasticity,  $E_c = 25742.96$   
Steel Elasticity,  $E_s = 200000.00$   
Existing Column  
Existing material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 18.00$   
Existing material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 500.00$   
Concrete Elasticity,  $E_c = 19940.411$   
Steel Elasticity,  $E_s = 200000.00$

#####  
Note: Especially for the calculation of moment strengths,  
the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Jacket  
New material: Steel Strength,  $f_s = 1.25 * f_{sm} = 781.25$   
Existing Column  
Existing material: Steel Strength,  $f_s = 1.25 * f_{sm} = 625.00$   
#####

Max Height,  $H_{max} = 400.00$   
Min Height,  $H_{min} = 200.00$   
Max Width,  $W_{max} = 400.00$   
Min Width,  $W_{min} = 200.00$   
Jacket Thickness,  $t_j = 50.00$   
Cover Thickness,  $c = 15.00$   
Mean Confinement Factor overall section = 1.00  
Element Length,  $L = 3000.00$   
Secondary Member  
Smooth Bars  
Ductile Steel  
Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Adequate Lap Length ( $l_o/l_{ou, min} >= 1$ )  
No FRP Wrapping

-----  
-----  
Stepwise Properties

-----  
-----  
At local axis: 2  
EDGE -A-  
Shear Force,  $V_a = -3.6347847E-006$   
EDGE -B-  
Shear Force,  $V_b = 3.6347847E-006$   
BOTH EDGES

Axial Force,  $F = -1.8026E+006$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 0.00$

-Compression:  $As_c = 5353.274$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t,ten} = 1137.257$

-Compression:  $As_{c,com} = 2208.54$

-Middle:  $As_{mid} = 2007.478$

Calculation of Shear Capacity ratio,  $V_e/V_r = 1.18753$

Member Controlled by Shear ( $V_e/V_r > 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 402750.026$

with

$M_{pr1} = \text{Max}(Mu_{1+}, Mu_{1-}) = 6.0413E+008$

$Mu_{1+} = 6.0413E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$Mu_{1-} = 5.4612E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(Mu_{2+}, Mu_{2-}) = 6.0413E+008$

$Mu_{2+} = 6.0413E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination

$Mu_{2-} = 5.4612E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

Calculation of  $Mu_{1+}$

Calculation of ultimate curvature  $\phi_u$  according to 4.1, Biskinis/Fardis 2013:

$\phi_u = 1.2613948E-005$

$M_u = 6.0413E+008$

with full section properties:

$b = 400.00$

$d = 367.00$

$d' = 33.00$

$v = 0.40931126$

$N = 1.8026E+006$

$f_c = 30.00$

$\omega (5A.5, TBDY) = 0.002$

Final value of  $\phi_u$ :  $\phi_u^* = \text{shear\_factor} * \text{Max}(\phi_u, \phi_c) = 0.00925707$

The Shear\_factor is considered equal to 1 (pure moment strength)

From (5.4b), TBDY:  $\phi_u = 0.00925707$

$\omega_e (5.4c) = 0.0207149$

$\omega_{ase} ((5.4d), TBDY) = (\omega_{ase1} * A_{ext} + \omega_{ase2} * A_{int}) / A_{sec} = 0.33836562$

$\omega_{ase1} = \text{Max}(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length

equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).

$\omega_{ase2} (> = \omega_{ase1}) = \text{Max}(((A_{conf,max2} - A_{noConf2}) / A_{conf,max2}) * (A_{conf,min2} / A_{conf,max2}), 0) = 0.33836562$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

Aconf,max2 = 45264.00 is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

Aconf,min2 = 0.00 is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area Aconf,max2 by a length equal to half the clear spacing between internal hoops.

AnoConf2 = 32309.333 is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$psh_{,min} * F_{ywe} = \text{Min}(psh_{,x} * F_{ywe}, psh_{,y} * F_{ywe}) = 6.12205$

Expression (5.4d) for  $psh_{,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

-----  
 $psh_{,x} * F_{ywe} = psh_1 * F_{ywe1} + ps_2 * F_{ywe2} = 6.12205$   
 $psh_1$  ((5.4d), TBDY) =  $L_{stir1} * A_{stir1} / (A_{sec} * s_1) = 0.00680678$   
Lstir1 (Length of stirrups along Y) = 1040.00  
Astir1 (stirrups area) = 78.53982  
 $psh_2$  (5.4d) =  $L_{stir2} * A_{stir2} / (A_{sec} * s_2) = 0.0012868$   
Lstir2 (Length of stirrups along Y) = 768.00  
Astir2 (stirrups area) = 50.26548

-----  
 $psh_{,y} * F_{ywe} = psh_1 * F_{ywe1} + ps_2 * F_{ywe2} = 6.12205$   
 $psh_1$  ((5.4d), TBDY) =  $L_{stir1} * A_{stir1} / (A_{sec} * s_1) = 0.00680678$   
Lstir1 (Length of stirrups along X) = 1040.00  
Astir1 (stirrups area) = 78.53982  
 $psh_2$  ((5.4d), TBDY) =  $L_{stir2} * A_{stir2} / (A_{sec} * s_2) = 0.0012868$   
Lstir2 (Length of stirrups along X) = 768.00  
Astir2 (stirrups area) = 50.26548

-----  
Asec = 120000.00  
s1 = 100.00  
s2 = 250.00  
fywe1 = 781.25  
fywe2 = 625.00  
fce = 30.00

From ((5.A5), TBDY), TBDY: cc = 0.002  
c = confinement factor = 1.00

y1 = 0.0025  
sh1 = 0.008  
ft1 = 886.7403  
fy1 = 738.9503  
su1 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

lo/lou,min = lb/ld = 1.00

su1 =  $0.4 * esu1_{nominal}$  ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1,ft1,fy1, it is considered characteristic value  $fsy_1 = fs_1/1.2$ , from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 * (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $fs_1 = (fs_{jacket} * A_{sl,ten,jacket} + fs_{core} * A_{sl,ten,core}) / A_{sl,ten} = 738.9503$

with  $Es_1 = (Es_{jacket} * A_{sl,ten,jacket} + Es_{core} * A_{sl,ten,core}) / A_{sl,ten} = 200000.00$

y2 = 0.0025  
sh2 = 0.008  
ft2 = 898.293  
fy2 = 748.5775  
su2 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

lo/lou,min = lb/lb,min = 1.00

su2 =  $0.4 * esu2_{nominal}$  ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu2\_nominal = 0.08,

For calculation of esu2\_nominal and y2, sh2,ft2,fy2, it is considered characteristic value  $fsy_2 = fs_2/1.2$ , from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 * (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $fs_2 = (fs_{jacket} * A_{sl,com,jacket} + fs_{core} * A_{sl,com,core}) / A_{sl,com} = 748.5775$

with  $Es_2 = (Es_{jacket} * A_{sl,com,jacket} + Es_{core} * A_{sl,com,core}) / A_{sl,com} = 200000.00$

yv = 0.0025

$shv = 0.008$   
 $ftv = 894.3662$   
 $fyv = 745.3052$   
 $suv = 0.032$   
 using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
 and also multiplied by the shear\_factor according to 15.7.1.4, with  
 $Shear\_factor = 1.00$   
 $lo/lou,min = lb/ld = 1.00$   
 $suv = 0.4 * esuv\_nominal ((5.5), TBDY) = 0.032$   
 From table 5A.1, TBDY:  $esuv\_nominal = 0.08$ ,  
 considering characteristic value  $fsyv = fsv/1.2$ , from table 5.1, TBDY  
 For calculation of  $esuv\_nominal$  and  $yv$ ,  $shv,ftv,fyv$ , it is considered  
 characteristic value  $fsyv = fsv/1.2$ , from table 5.1, TBDY.  
 $y1, sh1,ft1,fy1$ , are also multiplied by  $Min(1, 1.25 * (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.  
 with  $fsv = (fs,jacket * Asl,mid,jacket + fs,mid * Asl,mid,core) / Asl,mid = 745.3052$   
 with  $Esv = (Es,jacket * Asl,mid,jacket + Es,mid * Asl,mid,core) / Asl,mid = 200000.00$   
 $1 = Asl,ten / (b * d) * (fs1 / fc) = 0.19082108$   
 $2 = Asl,com / (b * d) * (fs2 / fc) = 0.37540035$   
 $v = Asl,mid / (b * d) * (fsv / fc) = 0.33973286$   
 and confined core properties:  
 $b = 360.00$   
 $d = 347.00$   
 $d' = 13.00$   
 $fcc (5A.2, TBDY) = 30.00$   
 $cc (5A.5, TBDY) = 0.002$   
 $c = \text{confinement factor} = 1.00$   
 $1 = Asl,ten / (b * d) * (fs1 / fc) = 0.22424379$   
 $2 = Asl,com / (b * d) * (fs2 / fc) = 0.44115251$   
 $v = Asl,mid / (b * d) * (fsv / fc) = 0.39923778$   
 Case/Assumption: Unconfined full section - Steel rupture  
 ' satisfies Eq. (4.3)  
 --->  
 $v < vs,y2$  - LHS eq.(4.5) is not satisfied  
 --->  
 $v < vs,c$  - RHS eq.(4.5) is not satisfied  
 --->  
 Case/Assumption Rejected.  
 --->  
 New Case/Assumption: Unconfined full section - Spalling of concrete cover  
 ' satisfies Eq. (4.4)  
 --->  
 $v < s,y1$  - LHS eq.(4.7) is not satisfied  
 --->  
 $v < vc,y1$  - RHS eq.(4.6) is satisfied  
 --->  
 $cu (4.10) = 0.37716748$   
 $MRC (4.17) = 7.0585E+008$   
 --->  
 New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover  
 In expressions below, the following modifications have been made  
 -  $b, d, d'$  replaced by geometric parameters of the core:  $bo, do, d'o$   
 -  $N, 1, 2, v$  normalised to  $bo * do$ , instead of  $b * d$   
 - - parameters of confined concrete,  $fcc, cc$ , used in lieu of  $fc, ecu$   
 --->  
 Subcase: Rupture of tension steel  
 --->  
 $v^* < v^*,y2$  - LHS eq.(4.5) is not satisfied  
 --->  
 $v^* < v^*,c$  - LHS eq.(4.5) is not satisfied  
 --->  
 Subcase rejected  
 --->  
 New Subcase: Failure of compression zone  
 --->  
 $v^* < v^*,y2$  - LHS eq.(4.6) is not satisfied  
 --->  
 $v^* < v^*,y1$  - RHS eq.(4.6) is not satisfied

--->

$$*cu(4.11) = 0.4569298$$

$$MRo(4.18) = 6.0413E+008$$

--->

$$u = cu(4.2) = 1.2613948E-005$$

$$Mu = MRo$$

-----  
Calculation of ratio lb/d

-----  
Adequate Lap Length: lb/d >= 1  
-----

-----  
Calculation of Mu1-  
-----

-----  
Calculation of ultimate curvature u according to 4.1, Biskinis/Fardis 2013:

$$u = 3.1475172E-005$$

$$Mu = 5.4612E+008$$

-----  
with full section properties:

$$b = 200.00$$

$$d = 367.00$$

$$d' = 33.00$$

$$v = 0.81862252$$

$$N = 1.8026E+006$$

$$fc = 30.00$$

$$co(5A.5, TBDY) = 0.002$$

$$\text{Final value of } cu: cu^* = \text{shear\_factor} * \text{Max}(cu, cc) = 0.00925707$$

The Shear\_factor is considered equal to 1 (pure moment strength)

$$\text{From (5.4b), TBDY: } cu = 0.00925707$$

$$we(5.4c) = 0.0207149$$

$$ase((5.4d), TBDY) = (ase1*Aext+ase2*Aint)/Asec = 0.33836562$$

$$ase1 = \text{Max}(((Aconf,max1-AnoConf1)/Aconf,max1)*(Aconf,min1/Aconf,max1),0) = 0.33836562$$

The definitions of AnoConf, Aconf,min and Aconf,max are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

Aconf,max1 = 89600.00 is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

Aconf,min1 = 59225.00 is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area Aconf,max1 by a length

equal to half the clear spacing between external hoops.

AnoConf1 = 43733.333 is the unconfined external core area which is equal to  $bi^2/6$  as defined at (A.2).

$$ase2 (>= ase1) = \text{Max}(((Aconf,max2-AnoConf2)/Aconf,max2)*(Aconf,min2/Aconf,max2),0) = 0.33836562$$

The definitions of AnoConf, Aconf,min and Aconf,max are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

Aconf,max2 = 45264.00 is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

Aconf,min2 = 0.00 is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area Aconf,max2 by a length

equal to half the clear spacing between internal hoops.

AnoConf2 = 32309.333 is the unconfined internal core area which is equal to  $bi^2/6$  as defined at (A.2).

$$psh,min*Fywe = \text{Min}(psh,x*Fywe, psh,y*Fywe) = 6.12205$$

Expression (5.4d) for psh,min\*Fywe has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

$$psh_x*Fywe = psh1*Fywe1+ps2*Fywe2 = 6.12205$$

$$psh1((5.4d), TBDY) = Lstir1*Astir1/(Asec*s1) = 0.00680678$$

$$Lstir1(\text{Length of stirrups along Y}) = 1040.00$$

$$Astir1(\text{stirrups area}) = 78.53982$$

psh2 (5.4d) =  $L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$   
Lstir2 (Length of stirrups along Y) = 768.00  
Astir2 (stirrups area) = 50.26548

psh\_y \* Fywe = psh1 \* Fywe1 + ps2 \* Fywe2 = 6.12205  
psh1 ((5.4d), TBDY) =  $L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$   
Lstir1 (Length of stirrups along X) = 1040.00  
Astir1 (stirrups area) = 78.53982  
psh2 ((5.4d), TBDY) =  $L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$   
Lstir2 (Length of stirrups along X) = 768.00  
Astir2 (stirrups area) = 50.26548

Asec = 120000.00

s1 = 100.00

s2 = 250.00

fywe1 = 781.25

fywe2 = 625.00

fce = 30.00

From ((5.A.5), TBDY), TBDY: cc = 0.002

c = confinement factor = 1.00

y1 = 0.0025

sh1 = 0.008

ft1 = 898.293

fy1 = 748.5775

su1 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb = 1.00

su1 =  $0.4 * esu1\_nominal ((5.5), TBDY) = 0.032$

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1, ft1, fy1, it is considered  
characteristic value fsy1 = fs1/1.2, from table 5.1, TBDY.

y1, sh1, ft1, fy1, are also multiplied by  $Min(1, 1.25 * (lb/lb)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs1 =  $(fs\_jacket * A_{sl,ten,jacket} + fs\_core * A_{sl,ten,core}) / A_{sl,ten} = 748.5775$

with Es1 =  $(Es\_jacket * A_{sl,ten,jacket} + Es\_core * A_{sl,ten,core}) / A_{sl,ten} = 200000.00$

y2 = 0.0025

sh2 = 0.008

ft2 = 886.7403

fy2 = 738.9503

su2 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb,min = 1.00

su2 =  $0.4 * esu2\_nominal ((5.5), TBDY) = 0.032$

From table 5A.1, TBDY: esu2\_nominal = 0.08,

For calculation of esu2\_nominal and y2, sh2, ft2, fy2, it is considered  
characteristic value fsy2 = fs2/1.2, from table 5.1, TBDY.

y1, sh1, ft1, fy1, are also multiplied by  $Min(1, 1.25 * (lb/lb)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs2 =  $(fs\_jacket * A_{sl,com,jacket} + fs\_core * A_{sl,com,core}) / A_{sl,com} = 738.9503$

with Es2 =  $(Es\_jacket * A_{sl,com,jacket} + Es\_core * A_{sl,com,core}) / A_{sl,com} = 200000.00$

yv = 0.0025

shv = 0.008

ftv = 894.3662

fyv = 745.3052

suv = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb = 1.00

suv =  $0.4 * esuv\_nominal ((5.5), TBDY) = 0.032$

From table 5A.1, TBDY: esuv\_nominal = 0.08,

considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY  
For calculation of esuv\_nominal and yv, shv, ftv, fyv, it is considered

characteristic value fsyv = fsv/1.2, from table 5.1, TBDY.

y1, sh1, ft1, fy1, are also multiplied by  $Min(1, 1.25 * (lb/lb)^{2/3})$ , from 10.3.5, ASCE 41-17.

```

with fsv = (fs,jacket*Asl,mid,jacket + fs,mid*Asl,mid,core)/Asl,mid = 745.3052
with Esv = (Es,jacket*Asl,mid,jacket + Es,mid*Asl,mid,core)/Asl,mid = 200000.00
1 = Asl,ten/(b*d)*(fs1/fc) = 0.7508007
2 = Asl,com/(b*d)*(fs2/fc) = 0.38164216
v = Asl,mid/(b*d)*(fsv/fc) = 0.67946571
and confined core properties:
b = 160.00
d = 347.00
d' = 13.00
fcc (5A.2, TBDY) = 30.00
cc (5A.5, TBDY) = 0.002
c = confinement factor = 1.00
1 = Asl,ten/(b*d)*(fs1/fc) = 0.99259314
2 = Asl,com/(b*d)*(fs2/fc) = 0.50454853
v = Asl,mid/(b*d)*(fsv/fc) = 0.898285
Case/Assumption: Unconfined full section - Steel rupture
' satisfies Eq. (4.3)
--->
v < vs,y2 - LHS eq.(4.5) is not satisfied
--->
v < vs,c - RHS eq.(4.5) is not satisfied
--->
Case/Assumption Rejected.
--->
New Case/Assumption: Unconfined full section - Spalling of concrete cover
' satisfies Eq. (4.4)
--->
v < s,y1 - LHS eq.(4.7) is not satisfied
--->
v < vc,y1 - RHS eq.(4.6) is not satisfied
--->
cu (4.11) = 0.80138156
MRc (4.18) = 5.4612E+008
--->
New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover
In expressions below, the following modifications have been made
- b, d, d' replaced by geometric parameters of the core: bo, do, d'o
- N, 1, 2, v normalised to bo*do, instead of b*d
- - parameters of confined concrete, fcc, cc, used in lieu of fc, ecu
--->
Subcase: Rupture of tension steel
--->
v* < v*s,y2 - LHS eq.(4.5) is not satisfied
--->
v* < v*s,c - LHS eq.(4.5) is not satisfied
--->
Subcase rejected
--->
New Subcase: Failure of compression zone
--->
v* < v*c,y2 - LHS eq.(4.6) is not satisfied
--->
v* < v*c,y1 - RHS eq.(4.6) is not satisfied
--->
*cu (4.11) = 0.80674817
MRo (4.18) = 2.8580E+008
MRo < 0.8*MRc
--->
u = cu (unconfined full section) = 3.1475172E-005
Mu = MRc
-----

Calculation of ratio lb/d
-----
Adequate Lap Length: lb/d >= 1
-----
-----

```

Calculation of  $\mu_{2+}$

Calculation of ultimate curvature  $\mu$  according to 4.1, Biskinis/Fardis 2013:

$$\mu = 1.2613948E-005$$

$$\mu = 6.0413E+008$$

with full section properties:

$$b = 400.00$$

$$d = 367.00$$

$$d' = 33.00$$

$$v = 0.40931126$$

$$N = 1.8026E+006$$

$$f_c = 30.00$$

$$c_o (5A.5, TBDY) = 0.002$$

$$\text{Final value of } \mu: \mu^* = \text{shear\_factor} * \text{Max}(\mu_c, \mu_{cc}) = 0.00925707$$

The Shear\_factor is considered equal to 1 (pure moment strength)

$$\text{From (5.4b), TBDY: } \mu_c = 0.00925707$$

$$\mu_{cc} (5.4c) = 0.0207149$$

$$\text{ase (5.4d), TBDY) = (ase1*Aext+ase2*Aint)/Asec = 0.33836562$$

$$\text{ase1} = \text{Max}(((A_{\text{conf,max1}} - A_{\text{noConf1}})/A_{\text{conf,max1}}) * (A_{\text{conf,min1}}/A_{\text{conf,max1}}), 0) = 0.33836562$$

The definitions of  $A_{\text{noConf}}$ ,  $A_{\text{conf,min}}$  and  $A_{\text{conf,max}}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{\text{conf,max1}} = 89600.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{\text{conf,min1}} = 59225.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{\text{conf,max1}}$  by a length equal to half the clear spacing between external hoops.

$A_{\text{noConf1}} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$\text{ase2} (> = \text{ase1}) = \text{Max}(((A_{\text{conf,max2}} - A_{\text{noConf2}})/A_{\text{conf,max2}}) * (A_{\text{conf,min2}}/A_{\text{conf,max2}}), 0) = 0.33836562$$

The definitions of  $A_{\text{noConf}}$ ,  $A_{\text{conf,min}}$  and  $A_{\text{conf,max}}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{\text{conf,max2}} = 45264.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{\text{conf,min2}} = 0.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{\text{conf,max2}}$  by a length equal to half the clear spacing between internal hoops.

$A_{\text{noConf2}} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe}, p_{sh,y} * F_{ywe}) = 6.12205$$

Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

$$p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$$

$$p_{sh1} ((5.4d), TBDY) = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$$

$$L_{stir1} (\text{Length of stirrups along Y}) = 1040.00$$

$$A_{stir1} (\text{stirrups area}) = 78.53982$$

$$p_{sh2} (5.4d) = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$$

$$L_{stir2} (\text{Length of stirrups along Y}) = 768.00$$

$$A_{stir2} (\text{stirrups area}) = 50.26548$$

$$p_{sh,y} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$$

$$p_{sh1} ((5.4d), TBDY) = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$$

$$L_{stir1} (\text{Length of stirrups along X}) = 1040.00$$

$$A_{stir1} (\text{stirrups area}) = 78.53982$$

$$p_{sh2} ((5.4d), TBDY) = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$$

$$L_{stir2} (\text{Length of stirrups along X}) = 768.00$$

$$A_{stir2} (\text{stirrups area}) = 50.26548$$

$$A_{sec} = 120000.00$$

s1 = 100.00  
s2 = 250.00  
fywe1 = 781.25  
fywe2 = 625.00  
fce = 30.00

From ((5.A5), TBDY), TBDY: cc = 0.002  
c = confinement factor = 1.00

y1 = 0.0025  
sh1 = 0.008  
ft1 = 886.7403  
fy1 = 738.9503  
su1 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/ld = 1.00

su1 = 0.4\*esu1\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1,ft1,fy1, it is considered  
characteristic value fsy1 = fs1/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (\text{lb}/\text{ld})^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs1 = (fs,jacket\*Asl,ten,jacket + fs,core\*Asl,ten,core)/Asl,ten = 738.9503

with Es1 = (Es,jacket\*Asl,ten,jacket + Es,core\*Asl,ten,core)/Asl,ten = 200000.00

y2 = 0.0025  
sh2 = 0.008  
ft2 = 898.293  
fy2 = 748.5775  
su2 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb,min = 1.00

su2 = 0.4\*esu2\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu2\_nominal = 0.08,

For calculation of esu2\_nominal and y2, sh2,ft2,fy2, it is considered  
characteristic value fsy2 = fs2/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (\text{lb}/\text{ld})^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs2 = (fs,jacket\*Asl,com,jacket + fs,core\*Asl,com,core)/Asl,com = 748.5775

with Es2 = (Es,jacket\*Asl,com,jacket + Es,core\*Asl,com,core)/Asl,com = 200000.00

yv = 0.0025  
shv = 0.008  
ftv = 894.3662  
fyv = 745.3052  
suv = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/ld = 1.00

suv = 0.4\*esuv\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esuv\_nominal = 0.08,

considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY

For calculation of esuv\_nominal and yv, shv,ftv,fyv, it is considered  
characteristic value fsyv = fsv/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (\text{lb}/\text{ld})^{2/3})$ , from 10.3.5, ASCE 41-17.

with fsv = (fs,jacket\*Asl,mid,jacket + fs,mid\*Asl,mid,core)/Asl,mid = 745.3052

with Esv = (Es,jacket\*Asl,mid,jacket + Es,mid\*Asl,mid,core)/Asl,mid = 200000.00

1 = Asl,ten/(b\*d)\*(fs1/fc) = 0.19082108

2 = Asl,com/(b\*d)\*(fs2/fc) = 0.37540035

v = Asl,mid/(b\*d)\*(fsv/fc) = 0.33973286

and confined core properties:

b = 360.00  
d = 347.00  
d' = 13.00  
fcc (5A.2, TBDY) = 30.00  
cc (5A.5, TBDY) = 0.002

c = confinement factor = 1.00

1 = Asl,ten/(b\*d)\*(fs1/fc) = 0.22424379

$$2 = A_{sl,com}/(b*d)*(f_s2/f_c) = 0.44115251$$

$$v = A_{sl,mid}/(b*d)*(f_{sv}/f_c) = 0.39923778$$

Case/Assumption: Unconfined full section - Steel rupture

' satisfies Eq. (4.3)

---->  
 $v < v_{s,y2}$  - LHS eq.(4.5) is not satisfied

---->  
 $v < v_{s,c}$  - RHS eq.(4.5) is not satisfied

---->  
Case/Assumption Rejected.

---->  
New Case/Assumption: Unconfined full section - Spalling of concrete cover

' satisfies Eq. (4.4)

---->  
 $v < v_{s,y1}$  - LHS eq.(4.7) is not satisfied

---->  
 $v < v_{c,y1}$  - RHS eq.(4.6) is satisfied

---->  
 $\epsilon_{cu}$  (4.10) = 0.37716748  
 $M_{Rc}$  (4.17) = 7.0585E+008

---->  
New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover

In expressions below, the following modifications have been made

- b, d, d' replaced by geometric parameters of the core:  $b_o$ ,  $d_o$ ,  $d'_o$
- $N_1$ ,  $N_2$ ,  $v$  normalised to  $b_o*d_o$ , instead of  $b*d$
- parameters of confined concrete,  $f_{cc}$ ,  $\epsilon_{cc}$ , used in lieu of  $f_c$ ,  $\epsilon_{cu}$

---->  
Subcase: Rupture of tension steel

---->  
 $v^* < v^*_{s,y2}$  - LHS eq.(4.5) is not satisfied

---->  
 $v^* < v^*_{s,c}$  - LHS eq.(4.5) is not satisfied

---->  
Subcase rejected

---->  
New Subcase: Failure of compression zone

---->  
 $v^* < v^*_{c,y2}$  - LHS eq.(4.6) is not satisfied

---->  
 $v^* < v^*_{c,y1}$  - RHS eq.(4.6) is not satisfied

---->  
 $\epsilon^*_{cu}$  (4.11) = 0.4569298  
 $M_{Ro}$  (4.18) = 6.0413E+008

---->  
 $u = \epsilon^*_{cu}$  (4.2) = 1.2613948E-005  
 $M_u = M_{Ro}$

-----  
Calculation of ratio  $l_b/d$

-----  
Adequate Lap Length:  $l_b/d \geq 1$

-----  
Calculation of  $M_{u2}$

-----  
Calculation of ultimate curvature  $u$  according to 4.1, Biskinis/Fardis 2013:

$u = 3.1475172E-005$   
 $M_u = 5.4612E+008$

-----  
with full section properties:

$b = 200.00$   
 $d = 367.00$   
 $d' = 33.00$   
 $v = 0.81862252$

$$N = 1.8026E+006$$

$$f_c = 30.00$$

$$c_o (5A.5, TBDY) = 0.002$$

$$\text{Final value of } c_u: c_u^* = \text{shear\_factor} * \text{Max}(c_u, c_c) = 0.00925707$$

The Shear\_factor is considered equal to 1 (pure moment strength)

$$\text{From (5.4b), TBDY: } c_u = 0.00925707$$

$$w_e (5.4c) = 0.0207149$$

$$a_{se} ((5.4d), TBDY) = (a_{se1} * A_{ext} + a_{se2} * A_{int}) / A_{sec} = 0.33836562$$

$$a_{se1} = \text{Max}(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$$

The definitions of  $A_{noConf1}$ ,  $A_{conf,min1}$  and  $A_{conf,max1}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$a_{se2} (>= a_{se1}) = \text{Max}(((A_{conf,max2} - A_{noConf2}) / A_{conf,max2}) * (A_{conf,min2} / A_{conf,max2}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe}, p_{sh,y} * F_{ywe}) = 6.12205$$

Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

$$p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$$

$$p_{sh1} ((5.4d), TBDY) = L_{stir1} * A_{stir1} / (A_{sec} * s_1) = 0.00680678$$

$$L_{stir1} (\text{Length of stirrups along Y}) = 1040.00$$

$$A_{stir1} (\text{stirrups area}) = 78.53982$$

$$p_{sh2} (5.4d) = L_{stir2} * A_{stir2} / (A_{sec} * s_2) = 0.0012868$$

$$L_{stir2} (\text{Length of stirrups along Y}) = 768.00$$

$$A_{stir2} (\text{stirrups area}) = 50.26548$$

$$p_{sh,y} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$$

$$p_{sh1} ((5.4d), TBDY) = L_{stir1} * A_{stir1} / (A_{sec} * s_1) = 0.00680678$$

$$L_{stir1} (\text{Length of stirrups along X}) = 1040.00$$

$$A_{stir1} (\text{stirrups area}) = 78.53982$$

$$p_{sh2} ((5.4d), TBDY) = L_{stir2} * A_{stir2} / (A_{sec} * s_2) = 0.0012868$$

$$L_{stir2} (\text{Length of stirrups along X}) = 768.00$$

$$A_{stir2} (\text{stirrups area}) = 50.26548$$

$$A_{sec} = 120000.00$$

$$s_1 = 100.00$$

$$s_2 = 250.00$$

$$f_{ywe1} = 781.25$$

$$f_{ywe2} = 625.00$$

$$f_{ce} = 30.00$$

$$\text{From ((5.A5), TBDY), TBDY: } c_c = 0.002$$

$$c = \text{confinement factor} = 1.00$$

$$y_1 = 0.0025$$

$$sh_1 = 0.008$$

$$ft_1 = 898.293$$

$$fy_1 = 748.5775$$

$$su_1 = 0.032$$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with

Shear\_factor = 1.00

lo/lou,min = lb/d = 1.00

su1 = 0.4\*esu1\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1,ft1,fy1, it is considered characteristic value fsy1 = fs1/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs1 = (fs,jacket\*Asl,ten,jacket + fs,core\*Asl,ten,core)/Asl,ten = 748.5775

with Es1 = (Es,jacket\*Asl,ten,jacket + Es,core\*Asl,ten,core)/Asl,ten = 200000.00

y2 = 0.0025

sh2 = 0.008

ft2 = 886.7403

fy2 = 738.9503

su2 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor

and also multiplied by the shear\_factor according to 15.7.1.4, with

Shear\_factor = 1.00

lo/lou,min = lb/lb,min = 1.00

su2 = 0.4\*esu2\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu2\_nominal = 0.08,

For calculation of esu2\_nominal and y2, sh2,ft2,fy2, it is considered characteristic value fsy2 = fs2/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs2 = (fs,jacket\*Asl,com,jacket + fs,core\*Asl,com,core)/Asl,com = 738.9503

with Es2 = (Es,jacket\*Asl,com,jacket + Es,core\*Asl,com,core)/Asl,com = 200000.00

yv = 0.0025

shv = 0.008

ftv = 894.3662

fyv = 745.3052

suv = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor

and also multiplied by the shear\_factor according to 15.7.1.4, with

Shear\_factor = 1.00

lo/lou,min = lb/d = 1.00

suv = 0.4\*esuv\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esuv\_nominal = 0.08,

considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY

For calculation of esuv\_nominal and yv, shv,ftv,fyv, it is considered characteristic value fsyv = fsv/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fsv = (fs,jacket\*Asl,mid,jacket + fs,mid\*Asl,mid,core)/Asl,mid = 745.3052

with Esv = (Es,jacket\*Asl,mid,jacket + Es,mid\*Asl,mid,core)/Asl,mid = 200000.00

1 = Asl,ten/(b\*d)\*(fs1/fc) = 0.7508007

2 = Asl,com/(b\*d)\*(fs2/fc) = 0.38164216

v = Asl,mid/(b\*d)\*(fsv/fc) = 0.67946571

and confined core properties:

b = 160.00

d = 347.00

d' = 13.00

fcc (5A.2, TBDY) = 30.00

cc (5A.5, TBDY) = 0.002

c = confinement factor = 1.00

1 = Asl,ten/(b\*d)\*(fs1/fc) = 0.99259314

2 = Asl,com/(b\*d)\*(fs2/fc) = 0.50454853

v = Asl,mid/(b\*d)\*(fsv/fc) = 0.898285

Case/Assumption: Unconfined full section - Steel rupture

' satisfies Eq. (4.3)

--->

v < vs,y2 - LHS eq.(4.5) is not satisfied

--->

v < vs,c - RHS eq.(4.5) is not satisfied

--->

Case/Assumption Rejected.

--->

New Case/Assumption: Unconfined full section - Spalling of concrete cover

' satisfies Eq. (4.4)

--->

v < s,y1 - LHS eq.(4.7) is not satisfied

---->

v < vc,y1 - RHS eq.(4.6) is not satisfied

---->

cu (4.11) = 0.80138156

MRC (4.18) = 5.4612E+008

---->

New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover

In expressions below, the following modifications have been made

- b, d, d' replaced by geometric parameters of the core: bo, do, d'o
- N, 1, 2, v normalised to bo\*do, instead of b\*d
- - parameters of confined concrete, fcc, cc, used in lieu of fc, ecu

---->

Subcase: Rupture of tension steel

---->

v\* < v\*s,y2 - LHS eq.(4.5) is not satisfied

---->

v\* < v\*s,c - LHS eq.(4.5) is not satisfied

---->

Subcase rejected

---->

New Subcase: Failure of compression zone

---->

v\* < v\*c,y2 - LHS eq.(4.6) is not satisfied

---->

v\* < v\*c,y1 - RHS eq.(4.6) is not satisfied

---->

\*cu (4.11) = 0.80674817

MRO (4.18) = 2.8580E+008

MRO < 0.8\*MRC

---->

u = cu (unconfined full section) = 3.1475172E-005

Mu = MRC

-----  
Calculation of ratio lb/d

Adequate Lap Length: lb/d >= 1

-----  
Calculation of Shear Strength Vr = Min(Vr1,Vr2) = 339150.739

-----  
Calculation of Shear Strength at edge 1, Vr1 = 339150.739

Vr1 = VCol ((10.3), ASCE 41-17) = knl\*VCol0

VCol0 = 339150.739

knl = 1 (zero step-static loading)

-----  
NOTE: In expression (10-3) 'Vs = Av\*fy\*d/s' is replaced by 'Vs+ f\*Vf'  
where Vf is the contribution of FRPs (11.3), ACI 440).

-----  
= 1 (normal-weight concrete)

Mean concrete strength: fc' = (fc'\_jacket\*Area\_jacket + fc'\_core\*Area\_core)/Area\_section = 25.00, but fc'^0.5 <= 8.3 MPa (22.5.3.1, ACI 318-14)

M/Vd = 4.00

Mu = 6.7415E+007

Vu = 3.6347847E-006

d = 0.8\*h = 320.00

Nu = 1.8026E+006

Ag = 80000.00

From (11.5.4.8), ACI 318-14: Vs = Vs\_jacket + Vs\_core = 471238.898

where:

Vs\_jacket = Vs\_j1 + Vs\_j2 = 471238.898

Vs\_j1 = 157079.633 is calculated for section web jacket, with:

d = 160.00

Av = 157079.633

$f_y = 625.00$   
 $s = 100.00$   
 $V_{s,j1}$  is multiplied by  $Col,j1 = 1.00$   
 $s/d = 0.625$   
 $V_{s,j2} = 314159.265$  is calculated for section flange jacket, with:  
 $d = 320.00$   
 $A_v = 157079.633$   
 $f_y = 625.00$   
 $s = 100.00$   
 $V_{s,j2}$  is multiplied by  $Col,j2 = 1.00$   
 $s/d = 0.3125$   
 $V_{s,core} = V_{s,c1} + V_{s,c2} = 0.00$   
 $V_{s,c1} = 0.00$  is calculated for section web core, with:  
 $d = 80.00$   
 $A_v = 100530.965$   
 $f_y = 500.00$   
 $s = 250.00$   
 $V_{s,c1}$  is multiplied by  $Col,c1 = 0.00$   
 $s/d = 3.125$   
 $V_{s,c2} = 0.00$  is calculated for section flange core, with:  
 $d = 240.00$   
 $A_v = 100530.965$   
 $f_y = 500.00$   
 $s = 250.00$   
 $V_{s,c2}$  is multiplied by  $Col,c2 = 0.00$   
 $s/d = 1.04167$   
 $V_f ((11-3)-(11.4), ACI 440) = 0.00$   
 From (11-11), ACI 440:  $V_s + V_f \leq 212577.225$   
 $bw = 200.00$

Calculation of Shear Strength at edge 2,  $V_{r2} = 339150.739$   
 $V_{r2} = V_{Col} ((10.3), ASCE 41-17) = knl * V_{Col0}$   
 $V_{Col0} = 339150.739$   
 $knl = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ '  
 where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)  
 Mean concrete strength:  $f_c' = (f_c'_{jacket} * Area_{jacket} + f_c'_{core} * Area_{core}) / Area_{section} = 25.00$ , but  $f_c'^{0.5} \leq 8.3$   
 MPa (22.5.3.1, ACI 318-14)  
 $M/d = 4.00$   
 $M_u = 5.3429E+007$   
 $V_u = 3.6347847E-006$   
 $d = 0.8 * h = 320.00$   
 $N_u = 1.8026E+006$   
 $A_g = 80000.00$   
 From (11.5.4.8), ACI 318-14:  $V_s = V_{s,jacket} + V_{s,core} = 471238.898$   
 where:  
 $V_{s,jacket} = V_{s,j1} + V_{s,j2} = 471238.898$   
 $V_{s,j1} = 157079.633$  is calculated for section web jacket, with:  
 $d = 160.00$   
 $A_v = 157079.633$   
 $f_y = 625.00$   
 $s = 100.00$   
 $V_{s,j1}$  is multiplied by  $Col,j1 = 1.00$   
 $s/d = 0.625$   
 $V_{s,j2} = 314159.265$  is calculated for section flange jacket, with:  
 $d = 320.00$   
 $A_v = 157079.633$   
 $f_y = 625.00$   
 $s = 100.00$   
 $V_{s,j2}$  is multiplied by  $Col,j2 = 1.00$   
 $s/d = 0.3125$   
 $V_{s,core} = V_{s,c1} + V_{s,c2} = 0.00$   
 $V_{s,c1} = 0.00$  is calculated for section web core, with:

d = 80.00

Av = 100530.965

fy = 500.00

s = 250.00

Vs,c1 is multiplied by Col,c1 = 0.00

s/d = 3.125

Vs,c2 = 0.00 is calculated for section flange core, with:

d = 240.00

Av = 100530.965

fy = 500.00

s = 250.00

Vs,c2 is multiplied by Col,c2 = 0.00

s/d = 1.04167

Vf ((11-3)-(11.4), ACI 440) = 0.00

From (11-11), ACI 440: Vs + Vf <= 212577.225

bw = 200.00

-----  
End Of Calculation of Shear Capacity ratio for element: column JLC1 of floor 1

At local axis: 2

-----  
Start Of Calculation of Chord Rotation Capacity for element: column JLC1 of floor 1

At local axis: 2

Integration Section: (a)

Section Type: rcjlc

Constant Properties

-----  
Knowledge Factor,  $\phi = 0.90$

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE 41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Jacket

New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 30.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 625.00$

Concrete Elasticity,  $E_c = 25742.96$

Steel Elasticity,  $E_s = 200000.00$

Existing Column

Existing material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 18.00$

Existing material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 500.00$

Concrete Elasticity,  $E_c = 19940.411$

Steel Elasticity,  $E_s = 200000.00$

Max Height,  $H_{max} = 400.00$

Min Height,  $H_{min} = 200.00$

Max Width,  $W_{max} = 400.00$

Min Width,  $W_{min} = 200.00$

Jacket Thickness,  $t_j = 50.00$

Cover Thickness,  $c = 15.00$

Element Length,  $L = 3000.00$

Secondary Member

Smooth Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_b/l_d >= 1$ )

No FRP Wrapping

-----  
Stepwise Properties

Bending Moment,  $M = 1.8819E+007$

Shear Force,  $V_2 = -97644.268$

Shear Force,  $V_3 = -30.34672$

Axial Force,  $F = -1.8003E+006$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_{t} = 2161.416$

-Compression:  $As_{c} = 3191.858$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t,ten} = 1137.257$

-Compression:  $As_{c,com} = 2208.54$

-Middle:  $As_{c,mid} = 2007.478$

Longitudinal External Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t,ten,jacket} = 829.3805$

-Compression:  $As_{c,com,jacket} = 1746.726$

-Middle:  $As_{c,mid,jacket} = 1545.664$

Longitudinal Internal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t,ten,core} = 307.8761$

-Compression:  $As_{c,com,core} = 461.8141$

-Middle:  $As_{c,mid,core} = 461.8141$

Mean Diameter of Tension Reinforcement,  $DbL = 16.80$

Existing component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_{R} = \gamma \cdot u = 0.02661509$   
 $u = \gamma + \rho = 0.02957233$

- Calculation of  $\gamma$  -

$\gamma = (M_y \cdot L_s / 3) / E_{eff} = 0.02957233$  ((4.29), Biskinis Phd)

$M_y = 3.6894E+008$

$L_s = M/V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) = 6000.00

From table 10.5, ASCE 41\_17:  $E_{eff} = factor \cdot E_c \cdot I_g = 2.4951E+013$

factor = 0.70

$A_g = 120000.00$

Mean concrete strength:  $f_c' = (f_c'_{jacket} \cdot Area_{jacket} + f_c'_{core} \cdot Area_{core}) / Area_{section} = 25.00$

$N = 1.8003E+006$

$E_c \cdot I_g = E_c \cdot I_{g,jacket} + E_c \cdot I_{g,core} = 3.5645E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\gamma$  and  $M_y$  according to Annex 7 -

$\gamma = \text{Min}(\gamma_{ten}, \gamma_{com})$

$\gamma_{ten} = 1.6451936E-005$

with  $f_y = 596.2441$

$d = 367.00$

$\gamma = 0.50624552$

$A = 0.11407012$

$B = 0.0742413$

with  $p_t = 0.01549396$

$p_c = 0.0300891$

$p_v = 0.02734983$

$N = 1.8003E+006$

$b = 200.00$

" = 0.08991826

$\gamma_{comp} = 9.3157495E-006$

with  $f_c = 30.00$

$E_c = 25742.96$

$\gamma = 0.61355204$

$A = 0.01446815$

$B = 0.03310406$

with  $E_s = 200000.00$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

- Calculation of  $\rho$  -

From table 10-8:  $\rho = 0.00$

with:

- Columns not controlled by inadequate development or splicing along the clear height because  $l_b/l_d \geq 1$   
shear control ratio  $V_{yE}/V_{CoIE} = 1.44151$

$d = d_{\text{external}} = 367.00$

$s = s_{\text{external}} = 100.00$

-  $t = s_1 + s_2 + 2 \cdot t_f/bw \cdot (f_{fe}/f_s) = 0.00809358$

jacket:  $s_1 = A_{v1} \cdot L_{\text{stir1}} / (s_1 \cdot A_g) = 0.00680678$

$A_{v1} = 78.53982$ , is the area of every stirrup parallel to loading (shear) direction

$L_{\text{stir1}} = 1040.00$ , is the total Length of all stirrups parallel to loading (shear) direction

$s_1 = 100.00$

core:  $s_2 = A_{v2} \cdot L_{\text{stir2}} / (s_2 \cdot A_g) = 0.0012868$

$A_{v2} = 50.26548$ , is the area of every stirrup parallel to loading (shear) direction

$L_{\text{stir2}} = 768.00$ , is the total Length of all stirrups parallel to loading (shear) direction

$s_2 = 250.00$

The term  $2 \cdot t_f/bw \cdot (f_{fe}/f_s)$  is implemented to account for FRP contribution

where  $f = 2 \cdot t_f/bw$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe}/f_s$  normalises  $f$  to steel strength

All these variables have already been given in Shear control ratio calculation.

For the normalisation  $f_s$  of jacket is used.

$N_{UD} = 1.8003E+006$

$A_g = 120000.00$

$f_{cE} = (f_{c\_jacket} \cdot \text{Area\_jacket} + f_{c\_core} \cdot \text{Area\_core}) / \text{section\_area} = 25.00$

$f_{yE} = (f_{y\_ext\_Long\_Reinf} \cdot \text{Area\_ext\_Long\_Reinf} + f_{y\_int\_Long\_Reinf} \cdot \text{Area\_int\_Long\_Reinf}) / \text{Area\_Tot\_Long\_Rein} =$

$596.2441$

$f_{yE} = (f_{y\_ext\_Trans\_Reinf} \cdot s_1 + f_{y\_int\_Trans\_Reinf} \cdot s_2) / (s_1 + s_2) = 605.1263$

$\rho = \text{Area\_Tot\_Long\_Rein} / (b \cdot d) = 0.07293289$

$b = 200.00$

$d = 367.00$

$f_{cE} = 25.00$

End Of Calculation of Chord Rotation Capacity for element: column JLC1 of floor 1

At local axis: 2

Integration Section: (a)

### Calculation No. 3

column C1, Floor 1

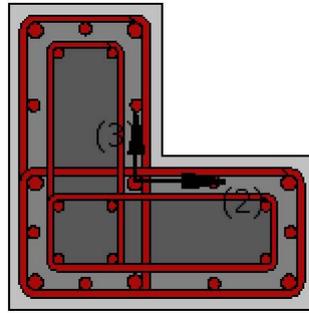
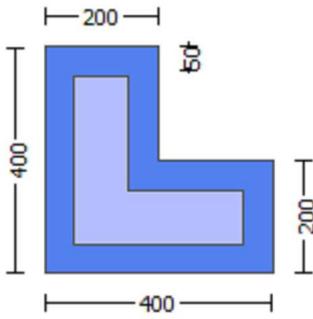
Limit State: Operational Level (data interpolation between analysis steps 50 and 51)

Analysis: Uniform +X

Check: Shear capacity  $V_{Rd}$

Edge: Start

Local Axis: (3)



Start Of Calculation of Shear Capacity for element: column JLC1 of floor 1

At local axis: 3

Integration Section: (a)

Section Type: rcjlcs

Constant Properties

Knowledge Factor,  $\gamma = 0.90$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE 41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Jacket

New material of Secondary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 20.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 500.00$

Concrete Elasticity,  $E_c = 25742.96$

Steel Elasticity,  $E_s = 200000.00$

Existing Column

Existing material of Secondary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 12.00$

Existing material of Secondary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 400.00$

Concrete Elasticity,  $E_c = 19940.411$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of  $\gamma$  for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE 41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE 41-17).

Jacket

New material: Concrete Strength,  $f_c = f_{cm} = 30.00$

New material: Steel Strength,  $f_s = f_{sm} = 625.00$

Existing Column

Existing material: Concrete Strength,  $f_c = f_{cm} = 18.00$

Existing material: Steel Strength,  $f_s = f_{sm} = 500.00$

#####

Max Height,  $H_{max} = 400.00$

Min Height,  $H_{min} = 200.00$

Max Width,  $W_{max} = 400.00$

Min Width,  $W_{min} = 200.00$

Jacket Thickness,  $t_j = 50.00$

Cover Thickness,  $c = 15.00$

Element Length,  $L = 3000.00$

Secondary Member

Smooth Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou,min} = l_b/l_d \geq 1$ )

No FRP Wrapping

Stepwise Properties

EDGE -A-

Bending Moment,  $M_a = 1.8819E+007$

Shear Force,  $V_a = -30.34672$   
 EDGE -B-  
 Bending Moment,  $M_b = 18804.004$   
 Shear Force,  $V_b = 30.34672$   
 BOTH EDGES  
 Axial Force,  $F = -1.8003E+006$   
 Longitudinal Reinforcement Area Distribution (in 2 divisions)  
   -Tension:  $As_t = 2161.416$   
   -Compression:  $As_c = 3191.858$   
 Longitudinal Reinforcement Area Distribution (in 3 divisions)  
   -Tension:  $As_{t,ten} = 1137.257$   
   -Compression:  $As_{c,com} = 2208.54$   
   -Middle:  $As_{c,mid} = 2007.478$   
 Mean Diameter of Tension Reinforcement,  $Db_{L,ten} = 16.80$

Existing component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = \phi V_n = 258140.283$   
 $V_n$  ((10.3), ASCE 41-17) =  $k_n \phi V_{CoI} = 286822.537$   
 $V_{CoI} = 286822.537$   
 $k_n = 1.00$   
 $displacement\_ductility\_demand = 0.11729439$

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + \phi \cdot V_f$ '  
 where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\phi = 1$  (normal-weight concrete)  
 Mean concrete strength:  $f'_c = (f'_{c,jacket} \cdot Area_{jacket} + f'_{c,core} \cdot Area_{core}) / Area_{section} = 16.66667$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/d = 4.00$   
 $\mu_u = 1.8819E+007$   
 $V_u = 30.34672$   
 $d = 0.8 \cdot h = 320.00$   
 $N_u = 1.8003E+006$   
 $A_g = 80000.00$   
 From (11.5.4.8), ACI 318-14:  $V_s = V_{s,jacket} + V_{s,core} = 376991.118$

where:

$V_{s,jacket} = V_{s,j1} + V_{s,j2} = 376991.118$   
 $V_{s,j1} = 251327.412$  is calculated for section web jacket, with:  
 $d = 320.00$   
 $A_v = 157079.633$   
 $f_y = 500.00$   
 $s = 100.00$   
 $V_{s,j1}$  is multiplied by  $Col_{j1} = 1.00$   
 $s/d = 0.3125$   
 $V_{s,j2} = 125663.706$  is calculated for section flange jacket, with:  
 $d = 160.00$   
 $A_v = 157079.633$   
 $f_y = 500.00$   
 $s = 100.00$   
 $V_{s,j2}$  is multiplied by  $Col_{j2} = 1.00$   
 $s/d = 0.625$

$V_{s,core} = V_{s,c1} + V_{s,c2} = 0.00$   
 $V_{s,c1} = 0.00$  is calculated for section web core, with:  
 $d = 240.00$   
 $A_v = 100530.965$   
 $f_y = 400.00$   
 $s = 250.00$

$V_{s,c1}$  is multiplied by  $Col_{c1} = 0.00$   
 $s/d = 1.04167$

$V_{s,c2} = 0.00$  is calculated for section flange core, with:  
 $d = 80.00$   
 $A_v = 100530.965$   
 $f_y = 400.00$   
 $s = 250.00$

$V_{s,c2}$  is multiplied by  $Col_{c2} = 0.00$

$s/d = 3.125$   
 $V_f((11-3)-(11.4), ACI 440) = 0.00$   
From (11-11), ACI 440:  $V_s + V_f \leq 173568.578$   
 $bw = 200.00$

displacement ductility demand is calculated as  $\phi / y$

- Calculation of  $\phi / y$  for END A -  
for rotation axis 2 and integ. section (a)

From analysis, chord rotation  $\theta = 0.00346867$   
 $y = (M_y * L_s / 3) / E_{eff} = 0.02957233$  ((4.29), Biskinis Phd))  
 $M_y = 3.6894E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 6000.00  
From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 2.4951E+013$   
factor = 0.70  
 $A_g = 120000.00$   
Mean concrete strength:  $f'_c = (f'_{c\_jacket} * Area_{jacket} + f'_{c\_core} * Area_{core}) / Area_{section} = 25.00$   
 $N = 1.8003E+006$   
 $E_c * I_g = E_{c\_jacket} * I_{g\_jacket} + E_{c\_core} * I_{g\_core} = 3.5645E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\phi$  and  $M_y$  according to Annex 7 -

$y = \text{Min}(y_{ten}, y_{com})$   
 $y_{ten} = 1.6451936E-005$   
with  $f_y = 596.2441$   
 $d = 367.00$   
 $y = 0.50624552$   
 $A = 0.11407012$   
 $B = 0.0742413$   
with  $p_t = 0.01549396$   
 $p_c = 0.0300891$   
 $p_v = 0.02734983$   
 $N = 1.8003E+006$   
 $b = 200.00$   
 $\rho = 0.08991826$   
 $y_{comp} = 9.3157495E-006$   
with  $f_c = 30.00$   
 $E_c = 25742.96$   
 $y = 0.61355204$   
 $A = 0.01446815$   
 $B = 0.03310406$   
with  $E_s = 200000.00$

Calculation of ratio  $I_b / I_d$

Adequate Lap Length:  $I_b / I_d \geq 1$

End Of Calculation of Shear Capacity for element: column JLC1 of floor 1  
At local axis: 3  
Integration Section: (a)

## Calculation No. 4

column C1, Floor 1

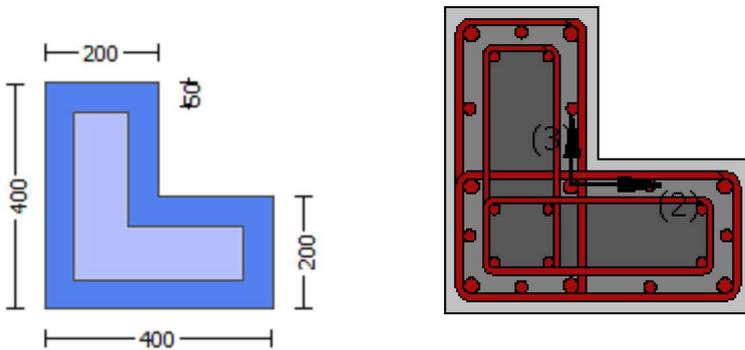
Limit State: Operational Level (data interpolation between analysis steps 50 and 51)

Analysis: Uniform +X

Check: Chord rotation capacity (  $\theta$  )

Edge: Start

Local Axis: (3)



Start Of Calculation of Shear Capacity ratio for element: column JLC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rcjlc

Constant Properties

Knowledge Factor,  $\gamma = 0.90$

Mean strength values are used for both shear and moment calculations.

Consequently:

Jacket

New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 30.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 625.00$

Concrete Elasticity,  $E_c = 25742.96$

Steel Elasticity,  $E_s = 200000.00$

Existing Column

Existing material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 18.00$

Existing material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 500.00$

Concrete Elasticity,  $E_c = 19940.411$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Jacket

New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 781.25$

Existing Column

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 625.00$

#####

Max Height,  $H_{max} = 400.00$

Min Height,  $H_{min} = 200.00$

Max Width,  $W_{max} = 400.00$

Min Width,  $W_{min} = 200.00$

Jacket Thickness,  $t_j = 50.00$   
Cover Thickness,  $c = 15.00$   
Mean Confinement Factor overall section = 1.00  
Element Length,  $L = 3000.00$   
Secondary Member  
Smooth Bars  
Ductile Steel  
Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Adequate Lap Length ( $l_o/l_{ou}, \min > = 1$ )  
No FRP Wrapping

#### Stepwise Properties

At local axis: 3  
EDGE -A-  
Shear Force,  $V_a = 2.3476251E-006$   
EDGE -B-  
Shear Force,  $V_b = -2.3476251E-006$   
BOTH EDGES  
Axial Force,  $F = -1.8026E+006$   
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $A_{st} = 0.00$   
-Compression:  $A_{sc} = 5353.274$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $A_{st,ten} = 2208.54$   
-Compression:  $A_{sc,com} = 1137.257$   
-Middle:  $A_{sc,mid} = 2007.478$

Calculation of Shear Capacity ratio,  $V_e/V_r = 1.44151$   
Member Controlled by Shear ( $V_e/V_r > 1$ )  
Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 488890.189$   
with  
 $M_{pr1} = \text{Max}(M_{u1+}, M_{u1-}) = 7.3334E+008$   
 $M_{u1+} = 7.3334E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination  
 $M_{u1-} = 6.3768E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination  
 $M_{pr2} = \text{Max}(M_{u2+}, M_{u2-}) = 7.3334E+008$   
 $M_{u2+} = 7.3334E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination  
 $M_{u2-} = 6.3768E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

#### Calculation of $M_{u1+}$

Calculation of ultimate curvature  $\phi_u$  according to 4.1, Biskinis/Fardis 2013:  
 $\phi_u = 4.2207094E-005$   
 $M_u = 7.3334E+008$

with full section properties:

$b = 400.00$   
 $d = 367.00$   
 $d' = 33.00$   
 $v = 0.40931126$   
 $N = 1.8026E+006$   
 $f_c = 30.00$   
 $\omega (5A.5, \text{TBDY}) = 0.002$   
Final value of  $\phi_u$ :  $\phi_u^* = \text{shear\_factor} * \text{Max}(\phi_u, \phi_c) = 0.00925707$   
The Shear\_factor is considered equal to 1 (pure moment strength)

From (5.4b), TBDY:  $c_u = 0.00925707$

$w_e$  (5.4c) = 0.0207149

$a_{se}$  ((5.4d), TBDY) =  $(a_{se1} * A_{ext} + a_{se2} * A_{int}) / A_{sec} = 0.33836562$

$a_{se1} = \text{Max}(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).

$a_{se2} (>= a_{se1}) = \text{Max}(((A_{conf,max2} - A_{noConf2}) / A_{conf,max2}) * (A_{conf,min2} / A_{conf,max2}), 0) = 0.33836562$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe}, p_{sh,y} * F_{ywe}) = 6.12205$

Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

-----  
 $p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{s2} * F_{ywe2} = 6.12205$

$p_{sh1}$  ((5.4d), TBDY) =  $L_{stir1} * A_{stir1} / (A_{sec} * s_1) = 0.00680678$

$L_{stir1}$  (Length of stirrups along Y) = 1040.00

$A_{stir1}$  (stirrups area) = 78.53982

$p_{sh2}$  (5.4d) =  $L_{stir2} * A_{stir2} / (A_{sec} * s_2) = 0.0012868$

$L_{stir2}$  (Length of stirrups along Y) = 768.00

$A_{stir2}$  (stirrups area) = 50.26548

-----  
 $p_{sh,y} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{s2} * F_{ywe2} = 6.12205$

$p_{sh1}$  ((5.4d), TBDY) =  $L_{stir1} * A_{stir1} / (A_{sec} * s_1) = 0.00680678$

$L_{stir1}$  (Length of stirrups along X) = 1040.00

$A_{stir1}$  (stirrups area) = 78.53982

$p_{sh2}$  ((5.4d), TBDY) =  $L_{stir2} * A_{stir2} / (A_{sec} * s_2) = 0.0012868$

$L_{stir2}$  (Length of stirrups along X) = 768.00

$A_{stir2}$  (stirrups area) = 50.26548

-----  
 $A_{sec} = 120000.00$

$s_1 = 100.00$

$s_2 = 250.00$

$f_{ywe1} = 781.25$

$f_{ywe2} = 625.00$

$f_{ce} = 30.00$

From ((5.A5), TBDY), TBDY:  $c_c = 0.002$

$c$  = confinement factor = 1.00

$y_1 = 0.0025$

$sh_1 = 0.008$

$ft_1 = 898.293$

$fy_1 = 748.5775$

$su_1 = 0.032$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor

and also multiplied by the shear\_factor according to 15.7.1.4, with

Shear\_factor = 1.00

$l_o/l_{ou,min} = l_b/d = 1.00$

$su_1 = 0.4 * esu_{1,nominal}$  ((5.5), TBDY) = 0.032

From table 5A.1, TBDY:  $esu_{1,nominal} = 0.08$ ,

For calculation of  $esu_{1,nominal}$  and  $y_1, sh_1, ft_1, fy_1$ , it is considered

characteristic value  $f_{sy1} = f_{s1}/1.2$ , from table 5.1, TBDY.

$y_1, sh_1, ft_1, fy_1$ , are also multiplied by  $\text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $f_{s1} = (f_{s,jacket} \cdot A_{s,ten,jacket} + f_{s,core} \cdot A_{s,ten,core})/A_{s,ten} = 748.5775$

with  $E_{s1} = (E_{s,jacket} \cdot A_{s,ten,jacket} + E_{s,core} \cdot A_{s,ten,core})/A_{s,ten} = 200000.00$

$y_2 = 0.0025$

$sh_2 = 0.008$

$ft_2 = 886.7403$

$fy_2 = 738.9503$

$su_2 = 0.032$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor

and also multiplied by the shear\_factor according to 15.7.1.4, with

Shear\_factor = 1.00

$l_o/l_{o,min} = l_b/l_{b,min} = 1.00$

$su_2 = 0.4 \cdot esu_{2,nominal} ((5.5), TBDY) = 0.032$

From table 5A.1, TBDY:  $esu_{2,nominal} = 0.08$ ,

For calculation of  $esu_{2,nominal}$  and  $y_2, sh_2, ft_2, fy_2$ , it is considered

characteristic value  $f_{sy2} = f_{s2}/1.2$ , from table 5.1, TBDY.

$y_1, sh_1, ft_1, fy_1$ , are also multiplied by  $\text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $f_{s2} = (f_{s,jacket} \cdot A_{s,com,jacket} + f_{s,core} \cdot A_{s,com,core})/A_{s,com} = 738.9503$

with  $E_{s2} = (E_{s,jacket} \cdot A_{s,com,jacket} + E_{s,core} \cdot A_{s,com,core})/A_{s,com} = 200000.00$

$y_v = 0.0025$

$sh_v = 0.008$

$ft_v = 894.3662$

$fy_v = 745.3052$

$su_v = 0.032$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor

and also multiplied by the shear\_factor according to 15.7.1.4, with

Shear\_factor = 1.00

$l_o/l_{o,min} = l_b/l_d = 1.00$

$su_v = 0.4 \cdot esuv_{nominal} ((5.5), TBDY) = 0.032$

From table 5A.1, TBDY:  $esuv_{nominal} = 0.08$ ,

considering characteristic value  $f_{syv} = f_{sv}/1.2$ , from table 5.1, TBDY

For calculation of  $esuv_{nominal}$  and  $y_v, sh_v, ft_v, fy_v$ , it is considered

characteristic value  $f_{syv} = f_{sv}/1.2$ , from table 5.1, TBDY.

$y_1, sh_1, ft_1, fy_1$ , are also multiplied by  $\text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $f_{sv} = (f_{s,jacket} \cdot A_{s,mid,jacket} + f_{s,mid} \cdot A_{s,mid,core})/A_{s,mid} = 745.3052$

with  $E_{sv} = (E_{s,jacket} \cdot A_{s,mid,jacket} + E_{s,mid} \cdot A_{s,mid,core})/A_{s,mid} = 200000.00$

$1 = A_{s,ten}/(b \cdot d) \cdot (f_{s1}/f_c) = 0.37540035$

$2 = A_{s,com}/(b \cdot d) \cdot (f_{s2}/f_c) = 0.19082108$

$v = A_{s,mid}/(b \cdot d) \cdot (f_{sv}/f_c) = 0.33973286$

and confined core properties:

$b = 360.00$

$d = 347.00$

$d' = 13.00$

$f_{cc} (5A.2, TBDY) = 30.00$

$cc (5A.5, TBDY) = 0.002$

$c = \text{confinement factor} = 1.00$

$1 = A_{s,ten}/(b \cdot d) \cdot (f_{s1}/f_c) = 0.44115251$

$2 = A_{s,com}/(b \cdot d) \cdot (f_{s2}/f_c) = 0.22424379$

$v = A_{s,mid}/(b \cdot d) \cdot (f_{sv}/f_c) = 0.39923778$

Case/Assumption: Unconfined full section - Steel rupture

' satisfies Eq. (4.3)

---->

$v < v_{s,y2}$  - LHS eq.(4.5) is not satisfied

---->

$v < v_{s,c}$  - RHS eq.(4.5) is not satisfied

---->

Case/Assumption Rejected.

---->

New Case/Assumption: Unconfined full section - Spalling of concrete cover

' satisfies Eq. (4.4)

---->

$v < s_{y1}$  - LHS eq.(4.7) is not satisfied

---->

$v < v_{c,y1}$  - RHS eq.(4.6) is satisfied

---->

$cu (4.10) = 0.59761571$

MRC (4.17) = 7.3334E+008

---->

New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover

In expressions below, the following modifications have been made

- b, d, d' replaced by geometric parameters of the core: bo, do, d'o
- N, 1, 2, v normalised to bo\*do, instead of b\*d
- - parameters of confined concrete, fcc, cc, used in lieu of fc, ec

---->

Subcase: Rupture of tension steel

---->

$v^* < v^*s_y2$  - LHS eq.(4.5) is not satisfied

---->

$v^* < v^*s_c$  - LHS eq.(4.5) is not satisfied

---->

Subcase rejected

---->

New Subcase: Failure of compression zone

---->

$v^* < v^*c_y2$  - LHS eq.(4.6) is not satisfied

---->

$v^* < v^*c_y1$  - RHS eq.(4.6) is not satisfied

---->

\*cu (4.11) = 0.62879536

MRO (4.18) = 4.5866E+008

MRO < 0.8\*MRC

---->

u = cu (unconfined full section) = 4.2207094E-005

Mu = MRC

-----  
Calculation of ratio lb/d

-----  
Adequate Lap Length: lb/d >= 1  
-----  
-----  
-----

-----  
Calculation of Mu1-  
-----  
-----

-----  
Calculation of ultimate curvature u according to 4.1, Biskinis/Fardis 2013:

u = 4.8346386E-005

Mu = 6.3768E+008  
-----

with full section properties:

b = 200.00

d = 367.00

d' = 33.00

v = 0.81862252

N = 1.8026E+006

fc = 30.00

co (5A.5, TBDY) = 0.002

Final value of cu:  $cu^* = \text{shear\_factor} * \text{Max}(cu, cc) = 0.00925707$

The Shear\_factor is considered equal to 1 (pure moment strength)

From (5.4b), TBDY:  $cu = 0.00925707$

we (5.4c) = 0.0207149

ase ((5.4d), TBDY) =  $(ase1*Aext+ase2*Aint)/Asec = 0.33836562$

ase1 =  $\text{Max}(((Aconf,max1-AnoConf1)/Aconf,max1)*(Aconf,min1/Aconf,max1),0) = 0.33836562$

The definitions of AnoConf, Aconf,min and Aconf,max are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

Aconf,max1 = 89600.00 is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

Aconf,min1 = 59225.00 is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area Aconf,max1 by a length

equal to half the clear spacing between external hoops.

AnoConf1 = 43733.333 is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).  
ase2 ( $\geq$  ase1) =  $\text{Max}(((A_{\text{conf,max2}} - \text{AnoConf2}) / A_{\text{conf,max2}}) * (A_{\text{conf,min2}} / A_{\text{conf,max2}}), 0) = 0.33836562$

The definitions of AnoConf, Aconf,min and Aconf,max are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

Aconf,max2 = 45264.00 is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

Aconf,min2 = 0.00 is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area Aconf,max2 by a length equal to half the clear spacing between internal hoops.

AnoConf2 = 32309.333 is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

psh,min\*Fywe =  $\text{Min}(psh_x * Fywe, psh_y * Fywe) = 6.12205$

Expression (5.4d) for psh,min\*Fywe has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

psh\_x\*Fywe = psh1\*Fywe1 + ps2\*Fywe2 = 6.12205

psh1 ((5.4d), TBDY) =  $L_{\text{stir1}} * A_{\text{stir1}} / (A_{\text{sec}} * s1) = 0.00680678$

Lstir1 (Length of stirrups along Y) = 1040.00

Astir1 (stirrups area) = 78.53982

psh2 (5.4d) =  $L_{\text{stir2}} * A_{\text{stir2}} / (A_{\text{sec}} * s2) = 0.0012868$

Lstir2 (Length of stirrups along Y) = 768.00

Astir2 (stirrups area) = 50.26548

psh\_y\*Fywe = psh1\*Fywe1 + ps2\*Fywe2 = 6.12205

psh1 ((5.4d), TBDY) =  $L_{\text{stir1}} * A_{\text{stir1}} / (A_{\text{sec}} * s1) = 0.00680678$

Lstir1 (Length of stirrups along X) = 1040.00

Astir1 (stirrups area) = 78.53982

psh2 ((5.4d), TBDY) =  $L_{\text{stir2}} * A_{\text{stir2}} / (A_{\text{sec}} * s2) = 0.0012868$

Lstir2 (Length of stirrups along X) = 768.00

Astir2 (stirrups area) = 50.26548

Asec = 120000.00

s1 = 100.00

s2 = 250.00

fywe1 = 781.25

fywe2 = 625.00

fce = 30.00

From ((5.A5), TBDY), TBDY: cc = 0.002

c = confinement factor = 1.00

y1 = 0.0025

sh1 = 0.008

ft1 = 886.7403

fy1 = 738.9503

su1 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

lo/lou,min = lb/lb = 1.00

su1 =  $0.4 * esu1_{\text{nominal}} ((5.5), \text{TBDY}) = 0.032$

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1, ft1, fy1, it is considered characteristic value fsy1 = fs1/1.2, from table 5.1, TBDY.

y1, sh1, ft1, fy1, are also multiplied by  $\text{Min}(1, 1.25 * (lb/lb)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs1 =  $(f_{s,\text{jacket}} * A_{s1,\text{ten,jacket}} + f_{s,\text{core}} * A_{s1,\text{ten,core}}) / A_{s1,\text{ten}} = 738.9503$

with Es1 =  $(E_{s,\text{jacket}} * A_{s1,\text{ten,jacket}} + E_{s,\text{core}} * A_{s1,\text{ten,core}}) / A_{s1,\text{ten}} = 200000.00$

y2 = 0.0025

sh2 = 0.008

ft2 = 898.293

fy2 = 748.5775

su2 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

lo/lou,min = lb/lb,min = 1.00

```

su2 = 0.4*esu2_nominal ((5.5), TBDY) = 0.032
From table 5A.1, TBDY: esu2_nominal = 0.08,
For calculation of esu2_nominal and y2, sh2,ft2,fy2, it is considered
characteristic value fsy2 = fs2/1.2, from table 5.1, TBDY.
y1, sh1,ft1,fy1, are also multiplied by Min(1,1.25*(lb/lc)^ 2/3), from 10.3.5, ASCE 41-17.
with fs2 = (fsjacket*Asl,com,jacket + fs,core*Asl,com,core)/Asl,com = 748.5775
with Es2 = (Esjacket*Asl,com,jacket + Es,core*Asl,com,core)/Asl,com = 200000.00
yv = 0.0025
shv = 0.008
ftv = 894.3662
fyv = 745.3052
suv = 0.032
using (30) in Biskinis/Fardis (2013) multiplied with shear_factor
and also multiplied by the shear_factor according to 15.7.1.4, with
Shear_factor = 1.00
lo/lou,min = lb/lc = 1.00
suv = 0.4*esuv_nominal ((5.5), TBDY) = 0.032
From table 5A.1, TBDY: esuv_nominal = 0.08,
considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY
For calculation of esuv_nominal and yv, shv,ftv,fyv, it is considered
characteristic value fsyv = fsv/1.2, from table 5.1, TBDY.
y1, sh1,ft1,fy1, are also multiplied by Min(1,1.25*(lb/lc)^ 2/3), from 10.3.5, ASCE 41-17.
with fsv = (fsjacket*Asl,mid,jacket + fs,mid*Asl,mid,core)/Asl,mid = 745.3052
with Esv = (Esjacket*Asl,mid,jacket + Es,mid*Asl,mid,core)/Asl,mid = 200000.00
1 = Asl,ten/(b*d)*(fs1/fc) = 0.38164216
2 = Asl,com/(b*d)*(fs2/fc) = 0.7508007
v = Asl,mid/(b*d)*(fsv/fc) = 0.67946571
and confined core properties:
b = 160.00
d = 347.00
d' = 13.00
fcc (5A.2, TBDY) = 30.00
cc (5A.5, TBDY) = 0.002
c = confinement factor = 1.00
1 = Asl,ten/(b*d)*(fs1/fc) = 0.50454853
2 = Asl,com/(b*d)*(fs2/fc) = 0.99259314
v = Asl,mid/(b*d)*(fsv/fc) = 0.898285
Case/Assumption: Unconfined full section - Steel rupture
'satisfies Eq. (4.3)
--->
v < vs,y2 - LHS eq.(4.5) is not satisfied
--->
v < vs,c - RHS eq.(4.5) is not satisfied
--->
Case/Assumption Rejected.
--->
New Case/Assumption: Unconfined full section - Spalling of concrete cover
'satisfies Eq. (4.4)
--->
v < sy1 - LHS eq.(4.7) is not satisfied
--->
v < vc,y1 - RHS eq.(4.6) is satisfied
--->
cu (4.10) = 0.52172715
MRc (4.17) = 6.3768E+008
--->
New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover
In expressions below, the following modifications have been made
- b, d, d' replaced by geometric parameters of the core: bo, do, d'o
- N, 1, 2, v normalised to bo*do, instead of b*d
- - parameters of confined concrete, fcc, cc, used in lieu of fc, ecu
--->
Subcase: Rupture of tension steel
--->
v* < v*s,y2 - LHS eq.(4.5) is not satisfied
--->
v* < v*s,c - LHS eq.(4.5) is not satisfied

```

--->

Subcase rejected

--->

New Subcase: Failure of compression zone

--->

$v^* < v^*c,y2$  - LHS eq.(4.6) is not satisfied

--->

$v^* < v^*c,y1$  - RHS eq.(4.6) is not satisfied

--->

\*cu (4.11) = 0.5604845

MRO (4.18) = 4.7738E+008

MRO < 0.8\*MRc

--->

u = cu (unconfined full section) = 4.8346386E-005

Mu = MRc

-----  
Calculation of ratio lb/d

-----  
Adequate Lap Length: lb/d >= 1  
-----  
-----

-----  
Calculation of Mu2+  
-----  
-----

-----  
Calculation of ultimate curvature u according to 4.1, Biskinis/Fardis 2013:

u = 4.2207094E-005

Mu = 7.3334E+008

-----  
with full section properties:

b = 400.00

d = 367.00

d' = 33.00

v = 0.40931126

N = 1.8026E+006

fc = 30.00

co (5A.5, TBDY) = 0.002

Final value of cu: cu\* = shear\_factor \* Max( cu, cc) = 0.00925707

The Shear\_factor is considered equal to 1 (pure moment strength)

From (5.4b), TBDY: cu = 0.00925707

we (5.4c) = 0.0207149

ase ((5.4d), TBDY) = (ase1\*Aext+ase2\*Aint)/Asec = 0.33836562

ase1 = Max(((Aconf,max1-AnoConf1)/Aconf,max1)\*(Aconf,min1/Aconf,max1),0) = 0.33836562

The definitions of AnoConf, Aconf,min and Aconf,max are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

Aconf,max1 = 89600.00 is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

Aconf,min1 = 59225.00 is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area Aconf,max1 by a length equal to half the clear spacing between external hoops.

AnoConf1 = 43733.333 is the unconfined external core area which is equal to bi<sup>2</sup>/6 as defined at (A.2).

ase2 (>=ase1) = Max(((Aconf,max2-AnoConf2)/Aconf,max2)\*(Aconf,min2/Aconf,max2),0) = 0.33836562

The definitions of AnoConf, Aconf,min and Aconf,max are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

Aconf,max2 = 45264.00 is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

Aconf,min2 = 0.00 is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area Aconf,max2 by a length equal to half the clear spacing between internal hoops.

AnoConf2 = 32309.333 is the unconfined internal core area which is equal to  $b^2/6$  as defined at (A.2).

$$psh_{min} * Fy_{we} = \text{Min}(psh_x * Fy_{we}, psh_y * Fy_{we}) = 6.12205$$

Expression (5.4d) for  $psh_{min} * Fy_{we}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

---

$$psh_x * Fy_{we} = psh1 * Fy_{we1} + ps2 * Fy_{we2} = 6.12205$$
$$psh1 \text{ ((5.4d), TBDY)} = Lstir1 * Astir1 / (Asec * s1) = 0.00680678$$

Lstir1 (Length of stirrups along Y) = 1040.00  
Astir1 (stirrups area) = 78.53982

$$psh2 \text{ (5.4d)} = Lstir2 * Astir2 / (Asec * s2) = 0.0012868$$

Lstir2 (Length of stirrups along Y) = 768.00  
Astir2 (stirrups area) = 50.26548

---

$$psh_y * Fy_{we} = psh1 * Fy_{we1} + ps2 * Fy_{we2} = 6.12205$$
$$psh1 \text{ ((5.4d), TBDY)} = Lstir1 * Astir1 / (Asec * s1) = 0.00680678$$

Lstir1 (Length of stirrups along X) = 1040.00  
Astir1 (stirrups area) = 78.53982

$$psh2 \text{ ((5.4d), TBDY)} = Lstir2 * Astir2 / (Asec * s2) = 0.0012868$$

Lstir2 (Length of stirrups along X) = 768.00  
Astir2 (stirrups area) = 50.26548

---

$$Asec = 120000.00$$

$$s1 = 100.00$$

$$s2 = 250.00$$

$$fy_{we1} = 781.25$$

$$fy_{we2} = 625.00$$

$$f_{ce} = 30.00$$

From ((5.A5), TBDY), TBDY:  $cc = 0.002$

$$c = \text{confinement factor} = 1.00$$

$$y1 = 0.0025$$

$$sh1 = 0.008$$

$$ft1 = 898.293$$

$$fy1 = 748.5775$$

$$su1 = 0.032$$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

$$lo/lou_{min} = lb/ld = 1.00$$

$$su1 = 0.4 * esu1_{nominal} \text{ ((5.5), TBDY)} = 0.032$$

From table 5A.1, TBDY:  $esu1_{nominal} = 0.08$ ,

For calculation of  $esu1_{nominal}$  and  $y1, sh1, ft1, fy1$ , it is considered characteristic value  $fsy1 = fs1/1.2$ , from table 5.1, TBDY.

$y1, sh1, ft1, fy1$ , are also multiplied by  $\text{Min}(1, 1.25 * (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

$$\text{with } fs1 = (fs_{jacket} * Asl_{ten,jacket} + fs_{core} * Asl_{ten,core}) / Asl_{ten} = 748.5775$$

$$\text{with } Es1 = (Es_{jacket} * Asl_{ten,jacket} + Es_{core} * Asl_{ten,core}) / Asl_{ten} = 200000.00$$

$$y2 = 0.0025$$

$$sh2 = 0.008$$

$$ft2 = 886.7403$$

$$fy2 = 738.9503$$

$$su2 = 0.032$$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

$$lo/lou_{min} = lb/lb_{min} = 1.00$$

$$su2 = 0.4 * esu2_{nominal} \text{ ((5.5), TBDY)} = 0.032$$

From table 5A.1, TBDY:  $esu2_{nominal} = 0.08$ ,

For calculation of  $esu2_{nominal}$  and  $y2, sh2, ft2, fy2$ , it is considered characteristic value  $fsy2 = fs2/1.2$ , from table 5.1, TBDY.

$y1, sh1, ft1, fy1$ , are also multiplied by  $\text{Min}(1, 1.25 * (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

$$\text{with } fs2 = (fs_{jacket} * Asl_{com,jacket} + fs_{core} * Asl_{com,core}) / Asl_{com} = 738.9503$$

$$\text{with } Es2 = (Es_{jacket} * Asl_{com,jacket} + Es_{core} * Asl_{com,core}) / Asl_{com} = 200000.00$$

$$y_v = 0.0025$$

$$sh_v = 0.008$$

$$ft_v = 894.3662$$

$$fy_v = 745.3052$$

$$su_v = 0.032$$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor

and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00  
 $l_o/l_{o,min} = l_b/l_d = 1.00$   
 $s_{uv} = 0.4 * e_{suv\_nominal} ((5.5), TBDY) = 0.032$   
From table 5A.1, TBDY:  $e_{suv\_nominal} = 0.08$ ,  
considering characteristic value  $f_{sv} = f_{sv}/1.2$ , from table 5.1, TBDY  
For calculation of  $e_{suv\_nominal}$  and  $\gamma_v$ ,  $sh_v, ft_v, fy_v$ , it is considered  
characteristic value  $f_{sv} = f_{sv}/1.2$ , from table 5.1, TBDY.  
 $\gamma_1$ ,  $sh_1, ft_1, fy_1$ , are also multiplied by  $\text{Min}(1, 1.25 * (l_b/l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.  
with  $f_{sv} = (f_{s,jacket} * A_{s,mid,jacket} + f_{s,mid} * A_{s,mid,core}) / A_{s,mid} = 745.3052$   
with  $E_{sv} = (E_{s,jacket} * A_{s,mid,jacket} + E_{s,mid} * A_{s,mid,core}) / A_{s,mid} = 200000.00$   
 $1 = A_{s,ten} / (b * d) * (f_{s1} / f_c) = 0.37540035$   
 $2 = A_{s,com} / (b * d) * (f_{s2} / f_c) = 0.19082108$   
 $v = A_{s,mid} / (b * d) * (f_{sv} / f_c) = 0.33973286$   
and confined core properties:  
 $b = 360.00$   
 $d = 347.00$   
 $d' = 13.00$   
 $f_{cc} (5A.2, TBDY) = 30.00$   
 $cc (5A.5, TBDY) = 0.002$   
 $c = \text{confinement factor} = 1.00$   
 $1 = A_{s,ten} / (b * d) * (f_{s1} / f_c) = 0.44115251$   
 $2 = A_{s,com} / (b * d) * (f_{s2} / f_c) = 0.22424379$   
 $v = A_{s,mid} / (b * d) * (f_{sv} / f_c) = 0.39923778$   
Case/Assumption: Unconfined full section - Steel rupture  
' satisfies Eq. (4.3)  
--->  
 $v < v_{s,y2}$  - LHS eq.(4.5) is not satisfied  
--->  
 $v < v_{s,c}$  - RHS eq.(4.5) is not satisfied  
--->  
Case/Assumption Rejected.  
--->  
New Case/Assumption: Unconfined full section - Spalling of concrete cover  
' satisfies Eq. (4.4)  
--->  
 $v < s_{y1}$  - LHS eq.(4.7) is not satisfied  
--->  
 $v < v_{c,y1}$  - RHS eq.(4.6) is satisfied  
--->  
 $cu (4.10) = 0.59761571$   
 $M_{Rc} (4.17) = 7.3334E+008$   
--->  
New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover  
In expressions below, the following modifications have been made  
-  $b, d, d'$  replaced by geometric parameters of the core:  $b_o, d_o, d'_o$   
-  $N, 1, 2, v$  normalised to  $b_o * d_o$ , instead of  $b * d$   
- - parameters of confined concrete,  $f_{cc}, cc$ , used in lieu of  $f_c, e_{cu}$   
--->  
Subcase: Rupture of tension steel  
--->  
 $v^* < v^*_{s,y2}$  - LHS eq.(4.5) is not satisfied  
--->  
 $v^* < v^*_{s,c}$  - LHS eq.(4.5) is not satisfied  
--->  
Subcase rejected  
--->  
New Subcase: Failure of compression zone  
--->  
 $v^* < v^*_{c,y2}$  - LHS eq.(4.6) is not satisfied  
--->  
 $v^* < v^*_{c,y1}$  - RHS eq.(4.6) is not satisfied  
--->  
 $*cu (4.11) = 0.62879536$   
 $M_{Ro} (4.18) = 4.5866E+008$   
 $M_{Ro} < 0.8 * M_{Rc}$   
--->

$u = c_u$  (unconfined full section) = 4.2207094E-005  
 $\mu = MRc$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of  $\mu_2$

Calculation of ultimate curvature  $\mu$  according to 4.1, Biskinis/Fardis 2013:

$u = 4.8346386E-005$   
 $\mu = 6.3768E+008$

with full section properties:

$b = 200.00$   
 $d = 367.00$   
 $d' = 33.00$   
 $v = 0.81862252$   
 $N = 1.8026E+006$

$f_c = 30.00$   
 $c_o$  (5A.5, TBDY) = 0.002  
Final value of  $c_u$ :  $c_u^* = \text{shear\_factor} * \text{Max}(c_u, c_c) = 0.00925707$

The Shear\_factor is considered equal to 1 (pure moment strength)

From (5.4b), TBDY:  $c_u = 0.00925707$   
 $w_e$  (5.4c) = 0.0207149

$a_{se}$  ((5.4d), TBDY) =  $(a_{se1} * A_{ext} + a_{se2} * A_{int}) / A_{sec} = 0.33836562$   
 $a_{se1} = \text{Max}(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988) "Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).  
 $a_{se2} (> a_{se1}) = \text{Max}(((A_{conf,max2} - A_{noConf2}) / A_{conf,max2}) * (A_{conf,min2} / A_{conf,max2}), 0) = 0.33836562$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988) "Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).  
 $p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe}, p_{sh,y} * F_{ywe}) = 6.12205$

Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

$p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$

$p_{sh1}$  ((5.4d), TBDY) =  $L_{stir1} * A_{stir1} / (A_{sec} * s_1) = 0.00680678$

$L_{stir1}$  (Length of stirrups along Y) = 1040.00

$A_{stir1}$  (stirrups area) = 78.53982

$p_{sh2}$  (5.4d) =  $L_{stir2} * A_{stir2} / (A_{sec} * s_2) = 0.0012868$

$L_{stir2}$  (Length of stirrups along Y) = 768.00

$A_{stir2}$  (stirrups area) = 50.26548

$psh\_y * Fywe = psh1 * Fywe1 + ps2 * Fywe2 = 6.12205$   
 $psh1 ((5.4d), TBDY) = Lstir1 * Astir1 / (Asec * s1) = 0.00680678$   
Lstir1 (Length of stirrups along X) = 1040.00  
Astir1 (stirrups area) = 78.53982  
 $psh2 ((5.4d), TBDY) = Lstir2 * Astir2 / (Asec * s2) = 0.0012868$   
Lstir2 (Length of stirrups along X) = 768.00  
Astir2 (stirrups area) = 50.26548

Asec = 120000.00

s1 = 100.00

s2 = 250.00

fywe1 = 781.25

fywe2 = 625.00

fce = 30.00

From ((5.A5), TBDY), TBDY: cc = 0.002

c = confinement factor = 1.00

y1 = 0.0025

sh1 = 0.008

ft1 = 886.7403

fy1 = 738.9503

su1 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/d = 1.00

su1 = 0.4 \* esu1\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1, ft1, fy1, it is considered  
characteristic value fsy1 = fs1/1.2, from table 5.1, TBDY.

y1, sh1, ft1, fy1, are also multiplied by  $\text{Min}(1, 1.25 * (lb/d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs1 = (fs,jacket \* Asl,ten,jacket + fs,core \* Asl,ten,core) / Asl,ten = 738.9503

with Es1 = (Es,jacket \* Asl,ten,jacket + Es,core \* Asl,ten,core) / Asl,ten = 200000.00

y2 = 0.0025

sh2 = 0.008

ft2 = 898.293

fy2 = 748.5775

su2 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb,min = 1.00

su2 = 0.4 \* esu2\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu2\_nominal = 0.08,

For calculation of esu2\_nominal and y2, sh2, ft2, fy2, it is considered  
characteristic value fsy2 = fs2/1.2, from table 5.1, TBDY.

y1, sh1, ft1, fy1, are also multiplied by  $\text{Min}(1, 1.25 * (lb/d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs2 = (fs,jacket \* Asl,com,jacket + fs,core \* Asl,com,core) / Asl,com = 748.5775

with Es2 = (Es,jacket \* Asl,com,jacket + Es,core \* Asl,com,core) / Asl,com = 200000.00

yv = 0.0025

shv = 0.008

ftv = 894.3662

fyv = 745.3052

suv = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/d = 1.00

suv = 0.4 \* esuv\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esuv\_nominal = 0.08,

considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY

For calculation of esuv\_nominal and yv, shv, ftv, fyv, it is considered  
characteristic value fsyv = fsv/1.2, from table 5.1, TBDY.

y1, sh1, ft1, fy1, are also multiplied by  $\text{Min}(1, 1.25 * (lb/d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fsv = (fs,jacket \* Asl,mid,jacket + fs,mid \* Asl,mid,core) / Asl,mid = 745.3052

with Esv = (Es,jacket \* Asl,mid,jacket + Es,mid \* Asl,mid,core) / Asl,mid = 200000.00

1 = Asl,ten / (b \* d) \* (fs1 / fc) = 0.38164216

2 = Asl,com / (b \* d) \* (fs2 / fc) = 0.7508007

$$v = A_{sl, mid} / (b * d) * (f_{sv} / f_c) = 0.67946571$$

and confined core properties:

$$b = 160.00$$

$$d = 347.00$$

$$d' = 13.00$$

$$f_{cc} (5A.2, TBDY) = 30.00$$

$$c_c (5A.5, TBDY) = 0.002$$

$$c = \text{confinement factor} = 1.00$$

$$1 = A_{sl, ten} / (b * d) * (f_{s1} / f_c) = 0.50454853$$

$$2 = A_{sl, com} / (b * d) * (f_{s2} / f_c) = 0.99259314$$

$$v = A_{sl, mid} / (b * d) * (f_{sv} / f_c) = 0.898285$$

Case/Assumption: Unconfined full section - Steel rupture

' satisfies Eq. (4.3)

--->

$v < v_{s, y2}$  - LHS eq.(4.5) is not satisfied

--->

$v < v_{s, c}$  - RHS eq.(4.5) is not satisfied

--->

Case/Assumption Rejected.

--->

New Case/Assumption: Unconfined full section - Spalling of concrete cover

' satisfies Eq. (4.4)

--->

$v < s_{y1}$  - LHS eq.(4.7) is not satisfied

--->

$v < v_{c, y1}$  - RHS eq.(4.6) is satisfied

--->

$$c_u (4.10) = 0.52172715$$

$$M_{Rc} (4.17) = 6.3768E+008$$

--->

New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover

In expressions below, the following modifications have been made

- b, d, d' replaced by geometric parameters of the core:  $b_o, d_o, d'_o$

-  $N, 1, 2, v$  normalised to  $b_o * d_o$ , instead of  $b * d$

- - parameters of confined concrete,  $f_{cc}, c_c$ , used in lieu of  $f_c, c_u$

--->

Subcase: Rupture of tension steel

--->

$v^* < v^*_{s, y2}$  - LHS eq.(4.5) is not satisfied

--->

$v^* < v^*_{s, c}$  - LHS eq.(4.5) is not satisfied

--->

Subcase rejected

--->

New Subcase: Failure of compression zone

--->

$v^* < v^*_{c, y2}$  - LHS eq.(4.6) is not satisfied

--->

$v^* < v^*_{c, y1}$  - RHS eq.(4.6) is not satisfied

--->

$$^*c_u (4.11) = 0.5604845$$

$$M_{Ro} (4.18) = 4.7738E+008$$

$$M_{Ro} < 0.8 * M_{Rc}$$

--->

$$u = c_u (\text{unconfined full section}) = 4.8346386E-005$$

$$M_u = M_{Rc}$$

-----  
Calculation of ratio  $l_b / l_d$

-----  
Adequate Lap Length:  $l_b / l_d \geq 1$   
-----  
-----  
-----

-----  
Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 339150.739$   
-----

Calculation of Shear Strength at edge 1,  $V_{r1} = 339150.739$

$V_{r1} = V_{Col} ((10.3), ASCE 41-17) = knl * V_{ColO}$

$V_{ColO} = 339150.739$

$kn1 = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ '  
where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

Mean concrete strength:  $f_c' = (f_c'_{jacket} * Area_{jacket} + f_c'_{core} * Area_{core}) / Area_{section} = 25.00$ , but  $f_c'^{0.5} \leq 8.3$   
MPa (22.5.3.1, ACI 318-14)

$M / Vd = 4.00$

$\mu_u = 2.0345E+006$

$V_u = 2.3476251E-006$

$d = 0.8 * h = 320.00$

$N_u = 1.8026E+006$

$A_g = 80000.00$

From (11.5.4.8), ACI 318-14:  $V_s = V_{sjacket} + V_{s,core} = 471238.898$

where:

$V_{sjacket} = V_{sj1} + V_{sj2} = 471238.898$

$V_{sj1} = 314159.265$  is calculated for section web jacket, with:

$d = 320.00$

$A_v = 157079.633$

$f_y = 625.00$

$s = 100.00$

$V_{sj1}$  is multiplied by  $Col_{j1} = 1.00$

$s/d = 0.3125$

$V_{sj2} = 157079.633$  is calculated for section flange jacket, with:

$d = 160.00$

$A_v = 157079.633$

$f_y = 625.00$

$s = 100.00$

$V_{sj2}$  is multiplied by  $Col_{j2} = 1.00$

$s/d = 0.625$

$V_{s,core} = V_{s,c1} + V_{s,c2} = 0.00$

$V_{s,c1} = 0.00$  is calculated for section web core, with:

$d = 240.00$

$A_v = 100530.965$

$f_y = 500.00$

$s = 250.00$

$V_{s,c1}$  is multiplied by  $Col_{c1} = 0.00$

$s/d = 1.04167$

$V_{s,c2} = 0.00$  is calculated for section flange core, with:

$d = 80.00$

$A_v = 100530.965$

$f_y = 500.00$

$s = 250.00$

$V_{s,c2}$  is multiplied by  $Col_{c2} = 0.00$

$s/d = 3.125$

$V_f ((11-3)-(11.4), ACI 440) = 0.00$

From (11-11), ACI 440:  $V_s + V_f \leq 212577.225$

$bw = 200.00$

Calculation of Shear Strength at edge 2,  $V_{r2} = 339150.739$

$V_{r2} = V_{Col} ((10.3), ASCE 41-17) = knl * V_{ColO}$

$V_{ColO} = 339150.739$

$kn1 = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ '  
where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

Mean concrete strength:  $f_c' = (f_c'_{jacket} * Area_{jacket} + f_c'_{core} * Area_{core}) / Area_{section} = 25.00$ , but  $f_c'^{0.5} \leq 8.3$   
MPa (22.5.3.1, ACI 318-14)

$M / Vd = 4.00$

$\mu_u = 6739.685$

$V_u = 2.3476251E-006$   
 $d = 0.8 \cdot h = 320.00$   
 $N_u = 1.8026E+006$   
 $A_g = 80000.00$   
 From (11.5.4.8), ACI 318-14:  $V_s = V_{s,jacket} + V_{s,core} = 471238.898$   
 where:  
 $V_{s,jacket} = V_{s,j1} + V_{s,j2} = 471238.898$   
 $V_{s,j1} = 314159.265$  is calculated for section web jacket, with:  
 $d = 320.00$   
 $A_v = 157079.633$   
 $f_y = 625.00$   
 $s = 100.00$   
 $V_{s,j1}$  is multiplied by  $Col,j1 = 1.00$   
 $s/d = 0.3125$   
 $V_{s,j2} = 157079.633$  is calculated for section flange jacket, with:  
 $d = 160.00$   
 $A_v = 157079.633$   
 $f_y = 625.00$   
 $s = 100.00$   
 $V_{s,j2}$  is multiplied by  $Col,j2 = 1.00$   
 $s/d = 0.625$   
 $V_{s,core} = V_{s,c1} + V_{s,c2} = 0.00$   
 $V_{s,c1} = 0.00$  is calculated for section web core, with:  
 $d = 240.00$   
 $A_v = 100530.965$   
 $f_y = 500.00$   
 $s = 250.00$   
 $V_{s,c1}$  is multiplied by  $Col,c1 = 0.00$   
 $s/d = 1.04167$   
 $V_{s,c2} = 0.00$  is calculated for section flange core, with:  
 $d = 80.00$   
 $A_v = 100530.965$   
 $f_y = 500.00$   
 $s = 250.00$   
 $V_{s,c2}$  is multiplied by  $Col,c2 = 0.00$   
 $s/d = 3.125$   
 $V_f ((11-3)-(11.4), ACI 440) = 0.00$   
 From (11-11), ACI 440:  $V_s + V_f \leq 212577.225$   
 $bw = 200.00$

-----  
 End Of Calculation of Shear Capacity ratio for element: column JLC1 of floor 1  
 At local axis: 3

-----  
 Start Of Calculation of Shear Capacity ratio for element: column JLC1 of floor 1  
 At Shear local axis: 2  
 (Bending local axis: 3)  
 Section Type: rcjlc

Constant Properties

-----  
 Knowledge Factor,  $\phi = 0.90$   
 Mean strength values are used for both shear and moment calculations.  
 Consequently:  
 Jacket  
 New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 30.00$   
 New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 625.00$   
 Concrete Elasticity,  $E_c = 25742.96$   
 Steel Elasticity,  $E_s = 200000.00$   
 Existing Column  
 Existing material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 18.00$   
 Existing material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 500.00$   
 Concrete Elasticity,  $E_c = 19940.411$   
 Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,  
the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Jacket

New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 781.25$

Existing Column

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 625.00$

#####

Max Height,  $H_{max} = 400.00$

Min Height,  $H_{min} = 200.00$

Max Width,  $W_{max} = 400.00$

Min Width,  $W_{min} = 200.00$

Jacket Thickness,  $t_j = 50.00$

Cover Thickness,  $c = 15.00$

Mean Confinement Factor overall section = 1.00

Element Length,  $L = 3000.00$

Secondary Member

Smooth Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{o,min} > 1$ )

No FRP Wrapping

-----  
Stepwise Properties

-----  
At local axis: 2

EDGE -A-

Shear Force,  $V_a = -3.6347847E-006$

EDGE -B-

Shear Force,  $V_b = 3.6347847E-006$

BOTH EDGES

Axial Force,  $F = -1.8026E+006$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 0.00$

-Compression:  $As_c = 5353.274$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t,ten} = 1137.257$

-Compression:  $As_{c,com} = 2208.54$

-Middle:  $As_{mid} = 2007.478$

-----  
Calculation of Shear Capacity ratio ,  $V_e/V_r = 1.18753$

Member Controlled by Shear ( $V_e/V_r > 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 402750.026$

with

$M_{pr1} = \text{Max}(M_{u1+} , M_{u1-}) = 6.0413E+008$

$M_{u1+} = 6.0413E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction  
which is defined for the static loading combination

$M_{u1-} = 5.4612E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment  
direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(M_{u2+} , M_{u2-}) = 6.0413E+008$

$M_{u2+} = 6.0413E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction  
which is defined for the the static loading combination

$M_{u2-} = 5.4612E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment  
direction which is defined for the the static loading combination

-----  
Calculation of  $M_{u1+}$

-----  
Calculation of ultimate curvature  $\phi_u$  according to 4.1, Biskinis/Fardis 2013:

$\phi_u = 1.2613948E-005$

$$\mu = 6.0413E+008$$

with full section properties:

$$b = 400.00$$

$$d = 367.00$$

$$d' = 33.00$$

$$v = 0.40931126$$

$$N = 1.8026E+006$$

$$f_c = 30.00$$

$$c_o \text{ (5A.5, TBDY)} = 0.002$$

$$\text{Final value of } c_u: c_u^* = \text{shear\_factor} * \text{Max}(c_u, c_c) = 0.00925707$$

The Shear\_factor is considered equal to 1 (pure moment strength)

$$\text{From (5.4b), TBDY: } c_u = 0.00925707$$

$$w_e \text{ (5.4c)} = 0.0207149$$

$$a_{se} \text{ ((5.4d), TBDY)} = (a_{se1} * A_{ext} + a_{se2} * A_{int}) / A_{sec} = 0.33836562$$

$$a_{se1} = \text{Max}(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$a_{se2} (>= a_{se1}) = \text{Max}(((A_{conf,max2} - A_{noConf2}) / A_{conf,max2}) * (A_{conf,min2} / A_{conf,max2}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe}, p_{sh,y} * F_{ywe}) = 6.12205$$

Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

$$p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{s2} * F_{ywe2} = 6.12205$$

$$p_{sh1} \text{ ((5.4d), TBDY)} = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$$

$$L_{stir1} \text{ (Length of stirrups along Y)} = 1040.00$$

$$A_{stir1} \text{ (stirrups area)} = 78.53982$$

$$p_{sh2} \text{ (5.4d)} = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$$

$$L_{stir2} \text{ (Length of stirrups along Y)} = 768.00$$

$$A_{stir2} \text{ (stirrups area)} = 50.26548$$

$$p_{sh,y} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{s2} * F_{ywe2} = 6.12205$$

$$p_{sh1} \text{ ((5.4d), TBDY)} = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$$

$$L_{stir1} \text{ (Length of stirrups along X)} = 1040.00$$

$$A_{stir1} \text{ (stirrups area)} = 78.53982$$

$$p_{sh2} \text{ ((5.4d), TBDY)} = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$$

$$L_{stir2} \text{ (Length of stirrups along X)} = 768.00$$

$$A_{stir2} \text{ (stirrups area)} = 50.26548$$

$$A_{sec} = 120000.00$$

$$s1 = 100.00$$

$$s2 = 250.00$$

$$f_{ywe1} = 781.25$$

$$f_{ywe2} = 625.00$$

$$f_{ce} = 30.00$$

$$\text{From ((5.A5), TBDY), TBDY: } c_c = 0.002$$

$$c = \text{confinement factor} = 1.00$$

y1 = 0.0025  
sh1 = 0.008  
ft1 = 886.7403  
fy1 = 738.9503  
su1 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/ld = 1.00

su1 = 0.4\*esu1\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1,ft1,fy1, it is considered  
characteristic value fsy1 = fs1/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs1 = (fs,jacket\*Asl,ten,jacket + fs,core\*Asl,ten,core)/Asl,ten = 738.9503

with Es1 = (Es,jacket\*Asl,ten,jacket + Es,core\*Asl,ten,core)/Asl,ten = 200000.00

y2 = 0.0025  
sh2 = 0.008  
ft2 = 898.293  
fy2 = 748.5775  
su2 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb,min = 1.00

su2 = 0.4\*esu2\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu2\_nominal = 0.08,

For calculation of esu2\_nominal and y2, sh2,ft2,fy2, it is considered  
characteristic value fsy2 = fs2/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs2 = (fs,jacket\*Asl,com,jacket + fs,core\*Asl,com,core)/Asl,com = 748.5775

with Es2 = (Es,jacket\*Asl,com,jacket + Es,core\*Asl,com,core)/Asl,com = 200000.00

yv = 0.0025  
shv = 0.008  
ftv = 894.3662  
fyv = 745.3052  
suv = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/ld = 1.00

suv = 0.4\*esuv\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esuv\_nominal = 0.08,

considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY

For calculation of esuv\_nominal and yv, shv,ftv,fyv, it is considered  
characteristic value fsyv = fsv/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fsv = (fs,jacket\*Asl,mid,jacket + fs,mid\*Asl,mid,core)/Asl,mid = 745.3052

with Esv = (Es,jacket\*Asl,mid,jacket + Es,mid\*Asl,mid,core)/Asl,mid = 200000.00

1 = Asl,ten/(b\*d)\*(fs1/fc) = 0.19082108

2 = Asl,com/(b\*d)\*(fs2/fc) = 0.37540035

v = Asl,mid/(b\*d)\*(fsv/fc) = 0.33973286

and confined core properties:

b = 360.00

d = 347.00

d' = 13.00

fcc (5A.2, TBDY) = 30.00

cc (5A.5, TBDY) = 0.002

c = confinement factor = 1.00

1 = Asl,ten/(b\*d)\*(fs1/fc) = 0.22424379

2 = Asl,com/(b\*d)\*(fs2/fc) = 0.44115251

v = Asl,mid/(b\*d)\*(fsv/fc) = 0.39923778

Case/Assumption: Unconfined full section - Steel rupture

' satisfies Eq. (4.3)

---->

v < vs,y2 - LHS eq.(4.5) is not satisfied

---->

v < v<sub>s,c</sub> - RHS eq.(4.5) is not satisfied

---->

Case/Assumption Rejected.

---->

New Case/Assumption: Unconfined full section - Spalling of concrete cover

' satisfies Eq. (4.4)

---->

v < v<sub>s,y1</sub> - LHS eq.(4.7) is not satisfied

---->

v < v<sub>c,y1</sub> - RHS eq.(4.6) is satisfied

---->

cu (4.10) = 0.37716748

M<sub>Rc</sub> (4.17) = 7.0585E+008

---->

New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover

In expressions below, the following modifications have been made

- b, d, d' replaced by geometric parameters of the core: b<sub>o</sub>, d<sub>o</sub>, d'<sub>o</sub>
- N<sub>1</sub>, N<sub>2</sub>, v normalised to b<sub>o</sub>\*d<sub>o</sub>, instead of b\*d
- parameters of confined concrete, f<sub>cc</sub>, c<sub>c</sub>, used in lieu of f<sub>c</sub>, e<sub>c</sub>

---->

Subcase: Rupture of tension steel

---->

v\* < v\*s<sub>y2</sub> - LHS eq.(4.5) is not satisfied

---->

v\* < v\*s<sub>c</sub> - LHS eq.(4.5) is not satisfied

---->

Subcase rejected

---->

New Subcase: Failure of compression zone

---->

v\* < v\*c<sub>y2</sub> - LHS eq.(4.6) is not satisfied

---->

v\* < v\*c<sub>y1</sub> - RHS eq.(4.6) is not satisfied

---->

\*cu (4.11) = 0.4569298

M<sub>Ro</sub> (4.18) = 6.0413E+008

---->

u = cu (4.2) = 1.2613948E-005

Mu = M<sub>Ro</sub>

-----  
Calculation of ratio lb/d

-----  
Adequate Lap Length: lb/d >= 1

-----  
Calculation of Mu1-

-----  
Calculation of ultimate curvature u according to 4.1, Biskinis/Fardis 2013:

u = 3.1475172E-005

Mu = 5.4612E+008

-----  
with full section properties:

b = 200.00

d = 367.00

d' = 33.00

v = 0.81862252

N = 1.8026E+006

f<sub>c</sub> = 30.00

co (5A.5, TBDY) = 0.002

Final value of cu: cu\* = shear\_factor \* Max( cu, cc) = 0.00925707

The Shear\_factor is considered equal to 1 (pure moment strength)

From (5.4b), TBDY: cu = 0.00925707

we (5.4c) = 0.0207149

$$ase((5.4d), TBDY) = (ase1 * A_{ext} + ase2 * A_{int}) / A_{sec} = 0.33836562$$

$$ase1 = \text{Max}(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$ase2 (>= ase1) = \text{Max}(((A_{conf,max2} - A_{noConf2}) / A_{conf,max2}) * (A_{conf,min2} / A_{conf,max2}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$psh_{,min} * F_{ywe} = \text{Min}(psh_x * F_{ywe}, psh_y * F_{ywe}) = 6.12205$$

Expression (5.4d) for  $psh_{,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

$$psh_x * F_{ywe} = psh1 * F_{ywe1} + ps2 * F_{ywe2} = 6.12205$$

$$psh1((5.4d), TBDY) = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$$

$L_{stir1}$  (Length of stirrups along Y) = 1040.00

$A_{stir1}$  (stirrups area) = 78.53982

$$psh2(5.4d) = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$$

$L_{stir2}$  (Length of stirrups along Y) = 768.00

$A_{stir2}$  (stirrups area) = 50.26548

$$psh_y * F_{ywe} = psh1 * F_{ywe1} + ps2 * F_{ywe2} = 6.12205$$

$$psh1((5.4d), TBDY) = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$$

$L_{stir1}$  (Length of stirrups along X) = 1040.00

$A_{stir1}$  (stirrups area) = 78.53982

$$psh2((5.4d), TBDY) = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$$

$L_{stir2}$  (Length of stirrups along X) = 768.00

$A_{stir2}$  (stirrups area) = 50.26548

$$A_{sec} = 120000.00$$

$$s1 = 100.00$$

$$s2 = 250.00$$

$$fywe1 = 781.25$$

$$fywe2 = 625.00$$

$$f_{ce} = 30.00$$

From ((5.A5), TBDY), TBDY:  $cc = 0.002$

$c$  = confinement factor = 1.00

$$y1 = 0.0025$$

$$sh1 = 0.008$$

$$ft1 = 898.293$$

$$fy1 = 748.5775$$

$$su1 = 0.032$$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

$$l_o / l_{ou,min} = l_b / l_d = 1.00$$

$$su1 = 0.4 * esu1_{nominal}((5.5), TBDY) = 0.032$$

From table 5A.1, TBDY:  $esu1_{nominal} = 0.08$ ,

For calculation of  $esu1_{nominal}$  and  $y1$ ,  $sh1$ ,  $ft1$ ,  $fy1$ , it is considered characteristic value  $fsy1 = fs1/1.2$ , from table 5.1, TBDY.

$y1$ ,  $sh1$ ,  $ft1$ ,  $fy1$ , are also multiplied by  $\text{Min}(1, 1.25 * (l_b / l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $fs1 = (fs,jacket*Asl,ten,jacket + fs,core*Asl,ten,core)/Asl,ten = 748.5775$   
 with  $Es1 = (Es,jacket*Asl,ten,jacket + Es,core*Asl,ten,core)/Asl,ten = 200000.00$   
 $y2 = 0.0025$   
 $sh2 = 0.008$   
 $ft2 = 886.7403$   
 $fy2 = 738.9503$   
 $su2 = 0.032$   
 using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
 and also multiplied by the shear\_factor according to 15.7.1.4, with  
 Shear\_factor = 1.00  
 $lo/lou,min = lb/lb,min = 1.00$   
 $su2 = 0.4*esu2,nominal ((5.5), TBDY) = 0.032$   
 From table 5A.1, TBDY:  $esu2,nominal = 0.08$ ,  
 For calculation of  $esu2,nominal$  and  $y2, sh2,ft2,fy2$ , it is considered  
 characteristic value  $fsy2 = fs2/1.2$ , from table 5.1, TBDY.  
 $y1, sh1,ft1,fy1$ , are also multiplied by  $Min(1,1.25*(lb/l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.  
 with  $fs2 = (fs,jacket*Asl,com,jacket + fs,core*Asl,com,core)/Asl,com = 738.9503$   
 with  $Es2 = (Es,jacket*Asl,com,jacket + Es,core*Asl,com,core)/Asl,com = 200000.00$   
 $yv = 0.0025$   
 $shv = 0.008$   
 $ftv = 894.3662$   
 $fyv = 745.3052$   
 $suv = 0.032$   
 using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
 and also multiplied by the shear\_factor according to 15.7.1.4, with  
 Shear\_factor = 1.00  
 $lo/lou,min = lb/l_d = 1.00$   
 $suv = 0.4*esuv,nominal ((5.5), TBDY) = 0.032$   
 From table 5A.1, TBDY:  $esuv,nominal = 0.08$ ,  
 considering characteristic value  $fsyv = fsv/1.2$ , from table 5.1, TBDY  
 For calculation of  $esuv,nominal$  and  $yv, shv,ftv,fyv$ , it is considered  
 characteristic value  $fsyv = fsv/1.2$ , from table 5.1, TBDY.  
 $y1, sh1,ft1,fy1$ , are also multiplied by  $Min(1,1.25*(lb/l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.  
 with  $fsv = (fs,jacket*Asl,mid,jacket + fs,mid*Asl,mid,core)/Asl,mid = 745.3052$   
 with  $Esv = (Es,jacket*Asl,mid,jacket + Es,mid*Asl,mid,core)/Asl,mid = 200000.00$   
 $1 = Asl,ten/(b*d)*(fs1/fc) = 0.7508007$   
 $2 = Asl,com/(b*d)*(fs2/fc) = 0.38164216$   
 $v = Asl,mid/(b*d)*(fsv/fc) = 0.67946571$   
 and confined core properties:  
 $b = 160.00$   
 $d = 347.00$   
 $d' = 13.00$   
 $fcc (5A.2, TBDY) = 30.00$   
 $cc (5A.5, TBDY) = 0.002$   
 $c = \text{confinement factor} = 1.00$   
 $1 = Asl,ten/(b*d)*(fs1/fc) = 0.99259314$   
 $2 = Asl,com/(b*d)*(fs2/fc) = 0.50454853$   
 $v = Asl,mid/(b*d)*(fsv/fc) = 0.898285$   
 Case/Assumption: Unconfined full section - Steel rupture  
 ' satisfies Eq. (4.3)  
 --->  
 $v < vs,y2$  - LHS eq.(4.5) is not satisfied  
 --->  
 $v < vs,c$  - RHS eq.(4.5) is not satisfied  
 --->  
 Case/Assumption Rejected.  
 --->  
 New Case/Assumption: Unconfined full section - Spalling of concrete cover  
 ' satisfies Eq. (4.4)  
 --->  
 $v < s,y1$  - LHS eq.(4.7) is not satisfied  
 --->  
 $v < vc,y1$  - RHS eq.(4.6) is not satisfied  
 --->  
 $cu (4.11) = 0.80138156$   
 $MRC (4.18) = 5.4612E+008$   
 --->

New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover

In expressions below, the following modifications have been made

- b, d, d' replaced by geometric parameters of the core: bo, do, d'o
- N, 1, 2, v normalised to bo\*do, instead of b\*d
- parameters of confined concrete, fcc, cc, used in lieu of fc, ec

--->

Subcase: Rupture of tension steel

--->

$v^* < v^*s_y2$  - LHS eq.(4.5) is not satisfied

--->

$v^* < v^*s_c$  - LHS eq.(4.5) is not satisfied

--->

Subcase rejected

--->

New Subcase: Failure of compression zone

--->

$v^* < v^*c_y2$  - LHS eq.(4.6) is not satisfied

--->

$v^* < v^*c_y1$  - RHS eq.(4.6) is not satisfied

--->

\*cu (4.11) = 0.80674817

MRo (4.18) = 2.8580E+008

MRo < 0.8\*MRc

--->

u = cu (unconfined full section) = 3.1475172E-005

Mu = MRc

-----  
Calculation of ratio lb/d

-----  
Adequate Lap Length: lb/d >= 1  
-----  
-----  
-----

-----  
Calculation of Mu2+  
-----  
-----  
-----

-----  
Calculation of ultimate curvature u according to 4.1, Biskinis/Fardis 2013:

u = 1.2613948E-005

Mu = 6.0413E+008  
-----

with full section properties:

b = 400.00

d = 367.00

d' = 33.00

v = 0.40931126

N = 1.8026E+006

fc = 30.00

co (5A.5, TBDY) = 0.002

Final value of cu: cu\* = shear\_factor \* Max( cu, cc) = 0.00925707

The Shear\_factor is considered equal to 1 (pure moment strength)

From (5.4b), TBDY: cu = 0.00925707

we (5.4c) = 0.0207149

ase ((5.4d), TBDY) = (ase1\*Aext+ase2\*Aint)/Asec = 0.33836562

ase1 = Max(((Aconf,max1-AnoConf1)/Aconf,max1)\*(Aconf,min1/Aconf,max1),0) = 0.33836562

The definitions of AnoConf, Aconf,min and Aconf,max are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

Aconf,max1 = 89600.00 is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

Aconf,min1 = 59225.00 is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area Aconf,max1 by a length equal to half the clear spacing between external hoops.

AnoConf1 = 43733.333 is the unconfined external core area which is equal to bi<sup>2</sup>/6 as defined at (A.2).

$$ase2 (>= ase1) = \text{Max}(((Aconf,max2 - AnoConf2)/Aconf,max2) * (Aconf,min2/Aconf,max2), 0) = 0.33836562$$

The definitions of AnoConf, Aconf,min and Aconf,max are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988) "Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

Aconf,max2 = 45264.00 is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

Aconf,min2 = 0.00 is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area Aconf,max2 by a length equal to half the clear spacing between internal hoops.

AnoConf2 = 32309.333 is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$psh,min * Fywe = \text{Min}(psh,x * Fywe, psh,y * Fywe) = 6.12205$$

Expression (5.4d) for psh,min \* Fywe has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

$$psh,x * Fywe = psh1 * Fywe1 + ps2 * Fywe2 = 6.12205$$

$$psh1 \text{ ((5.4d), TBDY)} = Lstir1 * Astir1 / (Asec * s1) = 0.00680678$$

$$Lstir1 \text{ (Length of stirrups along Y)} = 1040.00$$

$$Astir1 \text{ (stirrups area)} = 78.53982$$

$$psh2 \text{ (5.4d)} = Lstir2 * Astir2 / (Asec * s2) = 0.0012868$$

$$Lstir2 \text{ (Length of stirrups along Y)} = 768.00$$

$$Astir2 \text{ (stirrups area)} = 50.26548$$

$$psh,y * Fywe = psh1 * Fywe1 + ps2 * Fywe2 = 6.12205$$

$$psh1 \text{ ((5.4d), TBDY)} = Lstir1 * Astir1 / (Asec * s1) = 0.00680678$$

$$Lstir1 \text{ (Length of stirrups along X)} = 1040.00$$

$$Astir1 \text{ (stirrups area)} = 78.53982$$

$$psh2 \text{ ((5.4d), TBDY)} = Lstir2 * Astir2 / (Asec * s2) = 0.0012868$$

$$Lstir2 \text{ (Length of stirrups along X)} = 768.00$$

$$Astir2 \text{ (stirrups area)} = 50.26548$$

$$Asec = 120000.00$$

$$s1 = 100.00$$

$$s2 = 250.00$$

$$fywe1 = 781.25$$

$$fywe2 = 625.00$$

$$fce = 30.00$$

From ((5.A5), TBDY), TBDY: cc = 0.002

c = confinement factor = 1.00

$$y1 = 0.0025$$

$$sh1 = 0.008$$

$$ft1 = 886.7403$$

$$fy1 = 738.9503$$

$$su1 = 0.032$$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

$$lo/lou,min = lb/ld = 1.00$$

$$su1 = 0.4 * esu1\_nominal \text{ ((5.5), TBDY)} = 0.032$$

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1, ft1, fy1, it is considered characteristic value fsy1 = fs1/1.2, from table 5.1, TBDY.

y1, sh1, ft1, fy1, are also multiplied by  $\text{Min}(1, 1.25 * (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

$$\text{with } fs1 = (fs\_jacket * Asl,ten,jacket + fs\_core * Asl,ten,core) / Asl,ten = 738.9503$$

$$\text{with } Es1 = (Es\_jacket * Asl,ten,jacket + Es\_core * Asl,ten,core) / Asl,ten = 200000.00$$

$$y2 = 0.0025$$

$$sh2 = 0.008$$

$$ft2 = 898.293$$

$$fy2 = 748.5775$$

$$su2 = 0.032$$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

$$lo/lou,min = lb/lb,min = 1.00$$

$$su2 = 0.4 * esu2\_nominal \text{ ((5.5), TBDY)} = 0.032$$

From table 5A.1, TBDY: esu2\_nominal = 0.08,

For calculation of  $es_{u2\_nominal}$  and  $y_2$ ,  $sh_2, ft_2, fy_2$ , it is considered characteristic value  $fs_{y2} = fs_2/1.2$ , from table 5.1, TBDY.

$y_1$ ,  $sh_1, ft_1, fy_1$ , are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $fs_2 = (fs_{jacket} \cdot Asl_{,com,jacket} + fs_{core} \cdot Asl_{,com,core}) / Asl_{,com} = 748.5775$

with  $Es_2 = (Es_{jacket} \cdot Asl_{,com,jacket} + Es_{core} \cdot Asl_{,com,core}) / Asl_{,com} = 200000.00$

$y_v = 0.0025$

$sh_v = 0.008$

$ft_v = 894.3662$

$fy_v = 745.3052$

$suv = 0.032$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

$lo/lo_{u,min} = lb/ld = 1.00$

$suv = 0.4 \cdot es_{uv\_nominal} ((5.5), TBDY) = 0.032$

From table 5A.1, TBDY:  $es_{uv\_nominal} = 0.08$ ,

considering characteristic value  $fs_{yv} = fs_v/1.2$ , from table 5.1, TBDY

For calculation of  $es_{uv\_nominal}$  and  $y_v$ ,  $sh_v, ft_v, fy_v$ , it is considered characteristic value  $fs_{yv} = fs_v/1.2$ , from table 5.1, TBDY.

$y_1$ ,  $sh_1, ft_1, fy_1$ , are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $fs_v = (fs_{jacket} \cdot Asl_{,mid,jacket} + fs_{mid} \cdot Asl_{,mid,core}) / Asl_{,mid} = 745.3052$

with  $Es_v = (Es_{jacket} \cdot Asl_{,mid,jacket} + Es_{mid} \cdot Asl_{,mid,core}) / Asl_{,mid} = 200000.00$

$1 = Asl_{,ten} / (b \cdot d) \cdot (fs_1 / fc) = 0.19082108$

$2 = Asl_{,com} / (b \cdot d) \cdot (fs_2 / fc) = 0.37540035$

$v = Asl_{,mid} / (b \cdot d) \cdot (fs_v / fc) = 0.33973286$

and confined core properties:

$b = 360.00$

$d = 347.00$

$d' = 13.00$

$fcc$  (5A.2, TBDY) = 30.00

$cc$  (5A.5, TBDY) = 0.002

$c$  = confinement factor = 1.00

$1 = Asl_{,ten} / (b \cdot d) \cdot (fs_1 / fc) = 0.22424379$

$2 = Asl_{,com} / (b \cdot d) \cdot (fs_2 / fc) = 0.44115251$

$v = Asl_{,mid} / (b \cdot d) \cdot (fs_v / fc) = 0.39923778$

Case/Assumption: Unconfined full section - Steel rupture

' satisfies Eq. (4.3)

---

$v < v_{s,y2}$  - LHS eq.(4.5) is not satisfied

---

$v < v_{s,c}$  - RHS eq.(4.5) is not satisfied

---

Case/Assumption Rejected.

---

New Case/Assumption: Unconfined full section - Spalling of concrete cover

' satisfies Eq. (4.4)

---

$v < s_{y1}$  - LHS eq.(4.7) is not satisfied

---

$v < v_{c,y1}$  - RHS eq.(4.6) is satisfied

---

$cu$  (4.10) = 0.37716748

$MRC$  (4.17) = 7.0585E+008

---

New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover

In expressions below, the following modifications have been made

-  $b$ ,  $d$ ,  $d'$  replaced by geometric parameters of the core:  $bo$ ,  $do$ ,  $d'o$

-  $N$ ,  $1$ ,  $2$ ,  $v$  normalised to  $bo \cdot do$ , instead of  $b \cdot d$

-  $f_c$ ,  $cc$  parameters of confined concrete,  $fcc$ ,  $cc$ , used in lieu of  $fc$ ,  $ecu$

---

Subcase: Rupture of tension steel

---

$v^* < v^*_{s,y2}$  - LHS eq.(4.5) is not satisfied

---

$v^* < v^*_{s,c}$  - LHS eq.(4.5) is not satisfied

---

Subcase rejected

--->  
New Subcase: Failure of compression zone

--->  
 $v^* < v^*c,y2$  - LHS eq.(4.6) is not satisfied

--->  
 $v^* < v^*c,y1$  - RHS eq.(4.6) is not satisfied

--->  
 $*cu$  (4.11) = 0.4569298  
MRo (4.18) = 6.0413E+008

--->  
 $u = cu$  (4.2) = 1.2613948E-005  
Mu = MRo

-----  
Calculation of ratio lb/ld

-----  
Adequate Lap Length: lb/ld >= 1

-----  
Calculation of Mu2-

-----  
Calculation of ultimate curvature  $u$  according to 4.1, Biskinis/Fardis 2013:

$u = 3.1475172E-005$   
Mu = 5.4612E+008

-----  
with full section properties:

$b = 200.00$   
 $d = 367.00$   
 $d' = 33.00$   
 $v = 0.81862252$   
 $N = 1.8026E+006$

$fc = 30.00$   
 $co$  (5A.5, TBDY) = 0.002  
Final value of  $cu$ :  $cu^* = shear\_factor * Max(cu, cc) = 0.00925707$

The Shear\_factor is considered equal to 1 (pure moment strength)  
From (5.4b), TBDY:  $cu = 0.00925707$

$we$  (5.4c) = 0.0207149  
 $ase$  ((5.4d), TBDY) =  $(ase1 * Aext + ase2 * Aint) / Asec = 0.33836562$   
 $ase1 = Max(((Aconf,max1 - AnoConf1) / Aconf,max1) * (Aconf,min1 / Aconf,max1), 0) = 0.33836562$

The definitions of AnoConf, Aconf,min and Aconf,max are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$Aconf,max1 = 89600.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$Aconf,min1 = 59225.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $Aconf,max1$  by a length equal to half the clear spacing between external hoops.

$AnoConf1 = 43733.333$  is the unconfined external core area which is equal to  $bi^2/6$  as defined at (A.2).

$ase2 (>= ase1) = Max(((Aconf,max2 - AnoConf2) / Aconf,max2) * (Aconf,min2 / Aconf,max2), 0) = 0.33836562$

The definitions of AnoConf, Aconf,min and Aconf,max are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$Aconf,max2 = 45264.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$Aconf,min2 = 0.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $Aconf,max2$  by a length equal to half the clear spacing between internal hoops.

$AnoConf2 = 32309.333$  is the unconfined internal core area which is equal to  $bi^2/6$  as defined at (A.2).

$psh,min * Fywe = Min(psh,x * Fywe, psh,y * Fywe) = 6.12205$

Expression (5.4d) for  $psh,min * Fywe$  has been multiplied by 0.3 according to 15.7.1.3 for members without

earthquake detailing (90° closed stirrups)

psh\_x\*Fywe = psh1\*Fywe1+ps2\*Fywe2 = 6.12205  
psh1 ((5.4d), TBDY) = Lstir1\*Astir1/(Asec\*s1) = 0.00680678  
Lstir1 (Length of stirrups along Y) = 1040.00  
Astir1 (stirrups area) = 78.53982  
psh2 (5.4d) = Lstir2\*Astir2/(Asec\*s2) = 0.0012868  
Lstir2 (Length of stirrups along Y) = 768.00  
Astir2 (stirrups area) = 50.26548

psh\_y\*Fywe = psh1\*Fywe1+ps2\*Fywe2 = 6.12205  
psh1 ((5.4d), TBDY) = Lstir1\*Astir1/(Asec\*s1) = 0.00680678  
Lstir1 (Length of stirrups along X) = 1040.00  
Astir1 (stirrups area) = 78.53982  
psh2 ((5.4d), TBDY) = Lstir2\*Astir2/(Asec\*s2) = 0.0012868  
Lstir2 (Length of stirrups along X) = 768.00  
Astir2 (stirrups area) = 50.26548

Asec = 120000.00

s1 = 100.00

s2 = 250.00

fywe1 = 781.25

fywe2 = 625.00

fce = 30.00

From ((5.A5), TBDY), TBDY: cc = 0.002

c = confinement factor = 1.00

y1 = 0.0025

sh1 = 0.008

ft1 = 898.293

fy1 = 748.5775

su1 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/ld = 1.00

su1 = 0.4\*esu1\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1,ft1,fy1, it is considered  
characteristic value fsy1 = fs1/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (\text{lb}/\text{ld})^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs1 = (fs,jacket\*Asl,ten,jacket + fs,core\*Asl,ten,core)/Asl,ten = 748.5775

with Es1 = (Es,jacket\*Asl,ten,jacket + Es,core\*Asl,ten,core)/Asl,ten = 200000.00

y2 = 0.0025

sh2 = 0.008

ft2 = 886.7403

fy2 = 738.9503

su2 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb,min = 1.00

su2 = 0.4\*esu2\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu2\_nominal = 0.08,

For calculation of esu2\_nominal and y2, sh2,ft2,fy2, it is considered  
characteristic value fsy2 = fs2/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (\text{lb}/\text{ld})^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs2 = (fs,jacket\*Asl,com,jacket + fs,core\*Asl,com,core)/Asl,com = 738.9503

with Es2 = (Es,jacket\*Asl,com,jacket + Es,core\*Asl,com,core)/Asl,com = 200000.00

yv = 0.0025

shv = 0.008

ftv = 894.3662

fyv = 745.3052

suv = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/ld = 1.00

```

suv = 0.4*esuv_nominal ((5.5), TBDY) = 0.032
From table 5A.1, TBDY: esuv_nominal = 0.08,
considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY
For calculation of esuv_nominal and yv, shv,ftv,fyv, it is considered
characteristic value fsyv = fsv/1.2, from table 5.1, TBDY.
y1, sh1,ft1,fy1, are also multiplied by Min(1,1.25*(lb/lc)^2/3), from 10.3.5, ASCE 41-17.
with fsv = (fs,jacket*Asl,mid,jacket + fs,mid*Asl,mid,core)/Asl,mid = 745.3052
with Esv = (Es,jacket*Asl,mid,jacket + Es,mid*Asl,mid,core)/Asl,mid = 200000.00
1 = Asl,ten/(b*d)*(fs1/fc) = 0.7508007
2 = Asl,com/(b*d)*(fs2/fc) = 0.38164216
v = Asl,mid/(b*d)*(fsv/fc) = 0.67946571
and confined core properties:
b = 160.00
d = 347.00
d' = 13.00
fcc (5A.2, TBDY) = 30.00
cc (5A.5, TBDY) = 0.002
c = confinement factor = 1.00
1 = Asl,ten/(b*d)*(fs1/fc) = 0.99259314
2 = Asl,com/(b*d)*(fs2/fc) = 0.50454853
v = Asl,mid/(b*d)*(fsv/fc) = 0.898285
Case/Assumption: Unconfined full section - Steel rupture
'satisfies Eq. (4.3)
---->
v < vs,y2 - LHS eq.(4.5) is not satisfied
---->
v < vs,c - RHS eq.(4.5) is not satisfied
---->
Case/Assumption Rejected.
---->
New Case/Assumption: Unconfined full section - Spalling of concrete cover
'satisfies Eq. (4.4)
---->
v < sy1 - LHS eq.(4.7) is not satisfied
---->
v < vc,y1 - RHS eq.(4.6) is not satisfied
---->
cu (4.11) = 0.80138156
MRc (4.18) = 5.4612E+008
---->
New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover
In expressions below, the following modifications have been made
- b, d, d' replaced by geometric parameters of the core: bo, do, d'o
- N, 1, 2, v normalised to bo*do, instead of b*d
- parameters of confined concrete, fcc, cc, used in lieu of fc, ecu
---->
Subcase: Rupture of tension steel
---->
v* < v*s,y2 - LHS eq.(4.5) is not satisfied
---->
v* < v*s,c - LHS eq.(4.5) is not satisfied
---->
Subcase rejected
---->
New Subcase: Failure of compression zone
---->
v* < v*c,y2 - LHS eq.(4.6) is not satisfied
---->
v* < v*c,y1 - RHS eq.(4.6) is not satisfied
---->
*cu (4.11) = 0.80674817
MRo (4.18) = 2.8580E+008
MRo < 0.8*MRc
---->
u = cu (unconfined full section) = 3.1475172E-005
Mu = MRc
-----

```

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 339150.739$

Calculation of Shear Strength at edge 1,  $V_{r1} = 339150.739$

$V_{r1} = V_{Col} ((10.3), \text{ASCE } 41-17) = k_{nl} * V_{Col0}$

$V_{Col0} = 339150.739$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

Mean concrete strength:  $f_c' = (f_c'_{jacket} * \text{Area}_{jacket} + f_c'_{core} * \text{Area}_{core}) / \text{Area}_{section} = 25.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 4.00$

$\mu_u = 6.7415E+007$

$\nu_u = 3.6347847E-006$

$d = 0.8 * h = 320.00$

$N_u = 1.8026E+006$

$A_g = 80000.00$

From (11.5.4.8), ACI 318-14:  $V_s = V_{sjacket} + V_{s,core} = 471238.898$

where:

$V_{sjacket} = V_{sj1} + V_{sj2} = 471238.898$

$V_{sj1} = 157079.633$  is calculated for section web jacket, with:

$d = 160.00$

$A_v = 157079.633$

$f_y = 625.00$

$s = 100.00$

$V_{sj1}$  is multiplied by  $Col_{j1} = 1.00$

$s/d = 0.625$

$V_{sj2} = 314159.265$  is calculated for section flange jacket, with:

$d = 320.00$

$A_v = 157079.633$

$f_y = 625.00$

$s = 100.00$

$V_{sj2}$  is multiplied by  $Col_{j2} = 1.00$

$s/d = 0.3125$

$V_{s,core} = V_{s,c1} + V_{s,c2} = 0.00$

$V_{s,c1} = 0.00$  is calculated for section web core, with:

$d = 80.00$

$A_v = 100530.965$

$f_y = 500.00$

$s = 250.00$

$V_{s,c1}$  is multiplied by  $Col_{c1} = 0.00$

$s/d = 3.125$

$V_{s,c2} = 0.00$  is calculated for section flange core, with:

$d = 240.00$

$A_v = 100530.965$

$f_y = 500.00$

$s = 250.00$

$V_{s,c2}$  is multiplied by  $Col_{c2} = 0.00$

$s/d = 1.04167$

$V_f ((11-3)-(11.4), \text{ACI } 440) = 0.00$

From (11-11), ACI 440:  $V_s + V_f \leq 212577.225$

$bw = 200.00$

Calculation of Shear Strength at edge 2,  $V_{r2} = 339150.739$

$V_{r2} = V_{Col} ((10.3), \text{ASCE } 41-17) = k_{nl} * V_{Col0}$

$V_{Col0} = 339150.739$

knl = 1 (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ '  
where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

Mean concrete strength:  $f_c' = (f_c'_{\text{jacket}} \cdot \text{Area}_{\text{jacket}} + f_c'_{\text{core}} \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 25.00$ , but  $f_c'^{0.5} \leq 8.3$   
MPa (22.5.3.1, ACI 318-14)

$M/Vd = 4.00$

$\mu_u = 5.3429E+007$

$V_u = 3.6347847E-006$

$d = 0.8 \cdot h = 320.00$

$N_u = 1.8026E+006$

$A_g = 80000.00$

From (11.5.4.8), ACI 318-14:  $V_s = V_{s,\text{jacket}} + V_{s,\text{core}} = 471238.898$

where:

$V_{s,\text{jacket}} = V_{s,j1} + V_{s,j2} = 471238.898$

$V_{s,j1} = 157079.633$  is calculated for section web jacket, with:

$d = 160.00$

$A_v = 157079.633$

$f_y = 625.00$

$s = 100.00$

$V_{s,j1}$  is multiplied by  $\text{Col},j1 = 1.00$

$s/d = 0.625$

$V_{s,j2} = 314159.265$  is calculated for section flange jacket, with:

$d = 320.00$

$A_v = 157079.633$

$f_y = 625.00$

$s = 100.00$

$V_{s,j2}$  is multiplied by  $\text{Col},j2 = 1.00$

$s/d = 0.3125$

$V_{s,\text{core}} = V_{s,c1} + V_{s,c2} = 0.00$

$V_{s,c1} = 0.00$  is calculated for section web core, with:

$d = 80.00$

$A_v = 100530.965$

$f_y = 500.00$

$s = 250.00$

$V_{s,c1}$  is multiplied by  $\text{Col},c1 = 0.00$

$s/d = 3.125$

$V_{s,c2} = 0.00$  is calculated for section flange core, with:

$d = 240.00$

$A_v = 100530.965$

$f_y = 500.00$

$s = 250.00$

$V_{s,c2}$  is multiplied by  $\text{Col},c2 = 0.00$

$s/d = 1.04167$

$V_f$  ((11-3)-(11.4), ACI 440) = 0.00

From (11-11), ACI 440:  $V_s + V_f \leq 212577.225$

$b_w = 200.00$

End Of Calculation of Shear Capacity ratio for element: column JLC1 of floor 1

At local axis: 2

Start Of Calculation of Chord Rotation Capacity for element: column JLC1 of floor 1

At local axis: 3

Integration Section: (a)

Section Type: rcjlc

Constant Properties

Knowledge Factor,  $\gamma = 0.90$

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE 41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Jacket

New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 30.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 625.00$

Concrete Elasticity,  $E_c = 25742.96$

Steel Elasticity,  $E_s = 200000.00$

Existing Column

Existing material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 18.00$

Existing material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 500.00$

Concrete Elasticity,  $E_c = 19940.411$

Steel Elasticity,  $E_s = 200000.00$

Max Height,  $H_{max} = 400.00$

Min Height,  $H_{min} = 200.00$

Max Width,  $W_{max} = 400.00$

Min Width,  $W_{min} = 200.00$

Jacket Thickness,  $t_j = 50.00$

Cover Thickness,  $c = 15.00$

Element Length,  $L = 3000.00$

Secondary Member

Smooth Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_b/d >= 1$ )

No FRP Wrapping

-----  
Stepwise Properties

Bending Moment,  $M = -3.4056E+008$

Shear Force,  $V_2 = -97644.268$

Shear Force,  $V_3 = -30.34672$

Axial Force,  $F = -1.8003E+006$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 2161.416$

-Compression:  $As_c = 3191.858$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t,ten} = 2208.54$

-Compression:  $As_{c,com} = 1137.257$

-Middle:  $As_{mid} = 2007.478$

Longitudinal External Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t,ten,jacket} = 1746.726$

-Compression:  $As_{c,com,jacket} = 829.3805$

-Middle:  $As_{mid,jacket} = 1545.664$

Longitudinal Internal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t,ten,core} = 461.8141$

-Compression:  $As_{c,com,core} = 307.8761$

-Middle:  $As_{mid,core} = 461.8141$

Mean Diameter of Tension Reinforcement,  $Db_L = 16.60$

-----  
Existing component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_R = * u = 0.01311129$

$u = y + p = 0.0145681$

-----  
- Calculation of  $y$  -

$y = (M_y * L_s / 3) / E_{eff} = 0.0145681$  ((4.29), Biskinis Phd))

$M_y = 3.1266E+008$

$L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 3487.751

From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 2.4951E+013$

factor = 0.70

$A_g = 120000.00$

Mean concrete strength:  $f_c' = (f_c'_{jacket} * Area_{jacket} + f_c'_{core} * Area_{core}) / Area_{section} = 25.00$

$N = 1.8003E+006$

$$E_c I_g = E_c I_{g\_jacket} + E_c I_{g\_core} = 3.5645E+013$$

Calculation of Yielding Moment  $M_y$

Calculation of  $y$  and  $M_y$  according to Annex 7 -

$$y = \text{Min}(y_{ten}, y_{com})$$

$$y_{ten} = 1.9271316E-005$$

with  $f_y = 596.2441$

$$d = 367.00$$

$$y = 0.57848146$$

$$A = 0.11407012$$

$$B = 0.08752406$$

with  $p_t = 0.0300891$

$$p_c = 0.01549396$$

$$p_v = 0.02734983$$

$$N = 1.8003E+006$$

$$b = 200.00$$

$$" = 0.08991826$$

$$y_{comp} = 7.6825402E-006$$

with  $f_c = 30.00$

$$E_c = 25742.96$$

$$y = 0.74398532$$

$$A = 0.01446815$$

$$B = 0.04638683$$

with  $E_s = 200000.00$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

- Calculation of  $p$  -

From table 10-8:  $p = 0.00$

with:

- Columns not controlled by inadequate development or splicing along the clear height because  $l_b/d \geq 1$

shear control ratio  $V_y E / V_{CoI} O E = 1.18753$

$d = d_{external} = 367.00$

$s = s_{external} = 100.00$

-  $t = s_1 + s_2 + 2 * t_f / b_w * (f_{fe} / f_s) = 0.00809358$

jacket:  $s_1 = A_{v1} * L_{stir1} / (s_1 * A_g) = 0.00680678$

$A_{v1} = 78.53982$ , is the area of every stirrup parallel to loading (shear) direction

$L_{stir1} = 1040.00$ , is the total Length of all stirrups parallel to loading (shear) direction

$s_1 = 100.00$

core:  $s_2 = A_{v2} * L_{stir2} / (s_2 * A_g) = 0.0012868$

$A_{v2} = 50.26548$ , is the area of every stirrup parallel to loading (shear) direction

$L_{stir2} = 768.00$ , is the total Length of all stirrups parallel to loading (shear) direction

$s_2 = 250.00$

The term  $2 * t_f / b_w * (f_{fe} / f_s)$  is implemented to account for FRP contribution

where  $f = 2 * t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength

All these variables have already been given in Shear control ratio calculation.

For the normalisation  $f_s$  of jacket is used.

$N_{UD} = 1.8003E+006$

$A_g = 120000.00$

$f_c E = (f_c I_{jacket} * Area_{jacket} + f_c I_{core} * Area_{core}) / section\_area = 25.00$

$f_y I_E = (f_y I_{ext\_Long\_Reinf} * Area_{ext\_Long\_Reinf} + f_y I_{int\_Long\_Reinf} * Area_{int\_Long\_Reinf}) / Area_{Tot\_Long\_Rein} = 596.2441$

$f_y t E = (f_y I_{ext\_Trans\_Reinf} * s_1 + f_y I_{int\_Trans\_Reinf} * s_2) / (s_1 + s_2) = 605.1263$

$p_l = Area_{Tot\_Long\_Rein} / (b * d) = 0.07293289$

$b = 200.00$

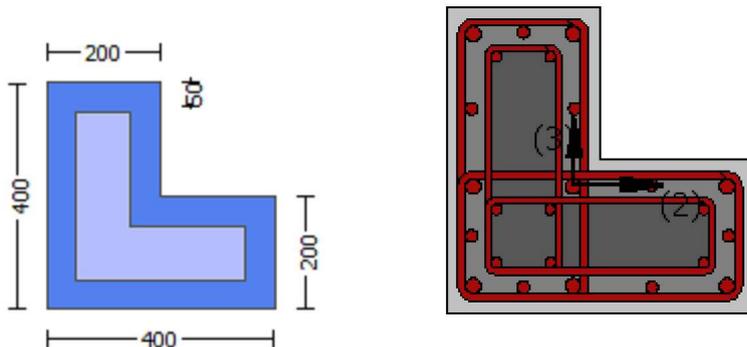
$d = 367.00$

$f_c E = 25.00$

End Of Calculation of Chord Rotation Capacity for element: column JLC1 of floor 1  
At local axis: 3  
Integration Section: (a)

## Calculation No. 5

column C1, Floor 1  
Limit State: Operational Level (data interpolation between analysis steps 50 and 51)  
Analysis: Uniform +X  
Check: Shear capacity VRd  
Edge: End  
Local Axis: (2)



Start Of Calculation of Shear Capacity for element: column JLC1 of floor 1  
At local axis: 2  
Integration Section: (b)  
Section Type: rcjcs

Constant Properties

Knowledge Factor,  $\gamma = 0.90$   
Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE 41-17.  
Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17  
Consequently:  
Jacket  
New material of Secondary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 20.00$   
New material of Secondary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 500.00$   
Concrete Elasticity,  $E_c = 25742.96$   
Steel Elasticity,  $E_s = 200000.00$   
Existing Column  
Existing material of Secondary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 12.00$   
Existing material of Secondary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 400.00$   
Concrete Elasticity,  $E_c = 19940.411$   
Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of  $\mu_y$  for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE 41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE 41-17).

Jacket

New material: Concrete Strength,  $f_c = f_{cm} = 30.00$

New material: Steel Strength,  $f_s = f_{sm} = 625.00$

Existing Column

Existing material: Concrete Strength,  $f_c = f_{cm} = 18.00$

Existing material: Steel Strength,  $f_s = f_{sm} = 500.00$

#####

Max Height,  $H_{max} = 400.00$

Min Height,  $H_{min} = 200.00$

Max Width,  $W_{max} = 400.00$

Min Width,  $W_{min} = 200.00$

Jacket Thickness,  $t_j = 50.00$

Cover Thickness,  $c = 15.00$

Element Length,  $L = 3000.00$

Secondary Member

Smooth Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou,min} = l_b/l_d \geq 1$ )

No FRP Wrapping

-----

Stepwise Properties

-----

EDGE -A-

Bending Moment,  $M_a = -3.4056E+008$

Shear Force,  $V_a = -97644.268$

EDGE -B-

Bending Moment,  $M_b = -7.2803E+007$

Shear Force,  $V_b = 97644.268$

BOTH EDGES

Axial Force,  $F = -1.8003E+006$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_{t} = 0.00$

-Compression:  $As_{c} = 5353.274$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t,ten} = 1137.257$

-Compression:  $As_{c,com} = 2208.54$

-Middle:  $As_{c,mid} = 2007.478$

Mean Diameter of Tension Reinforcement,  $Db_{L,ten} = 16.80$

-----

-----

Existing component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = \phi V_n = 331196.688$

$V_n$  ((10.3), ASCE 41-17) =  $k_n l V_{CoI} = 367996.32$

$V_{CoI} = 367996.32$

$k_n l = 1.00$

displacement\_ductility\_demand = 0.53336237

-----

NOTE: In expression (10-3) ' $V_s = A_v f_y d/s$ ' is replaced by ' $V_s + \phi V_f$ '

where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

-----

= 1 (normal-weight concrete)

Mean concrete strength:  $f'_c = (f'_{c,jacket} \text{Area}_{jacket} + f'_{c,core} \text{Area}_{core}) / \text{Area}_{section} = 16.66667$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.33$

$M_u = 7.2803E+007$

$V_u = 97644.268$

$d = 0.8 \cdot h = 320.00$

$N_u = 1.8003E+006$

$A_g = 80000.00$

From (11.5.4.8), ACI 318-14:  $V_s = V_{s,jacket} + V_{s,core} = 376991.118$

where:

$$V_{s,jacket} = V_{s,j1} + V_{s,j2} = 376991.118$$

$V_{s,j1} = 125663.706$  is calculated for section web jacket, with:

$$d = 160.00$$

$$A_v = 157079.633$$

$$f_y = 500.00$$

$$s = 100.00$$

$V_{s,j1}$  is multiplied by  $Col,j1 = 1.00$

$$s/d = 0.625$$

$V_{s,j2} = 251327.412$  is calculated for section flange jacket, with:

$$d = 320.00$$

$$A_v = 157079.633$$

$$f_y = 500.00$$

$$s = 100.00$$

$V_{s,j2}$  is multiplied by  $Col,j2 = 1.00$

$$s/d = 0.3125$$

$$V_{s,core} = V_{s,c1} + V_{s,c2} = 0.00$$

$V_{s,c1} = 0.00$  is calculated for section web core, with:

$$d = 80.00$$

$$A_v = 100530.965$$

$$f_y = 400.00$$

$$s = 250.00$$

$V_{s,c1}$  is multiplied by  $Col,c1 = 0.00$

$$s/d = 3.125$$

$V_{s,c2} = 0.00$  is calculated for section flange core, with:

$$d = 240.00$$

$$A_v = 100530.965$$

$$f_y = 400.00$$

$$s = 250.00$$

$V_{s,c2}$  is multiplied by  $Col,c2 = 0.00$

$$s/d = 1.04167$$

$V_f ((11-3)-(11.4), ACI 440) = 0.00$

From (11-11), ACI 440:  $V_s + V_f \leq 173568.578$

$$bw = 200.00$$

displacement ductility demand is calculated as  $\delta / y$

- Calculation of  $\delta / y$  for END B -

for rotation axis 3 and integ. section (b)

From analysis, chord rotation  $\theta = 0.00279899$

$$y = (M_y * L_s / 3) / E_{eff} = 0.00524781 \text{ ((4.29), Biskinis Phd)}$$

$$M_y = 5.2685E+008$$

$$L_s = M/V \text{ (with } L_s > 0.1 * L \text{ and } L_s < 2 * L) = 745.5987$$

From table 10.5, ASCE 41\_17:  $E_{eff} = \text{factor} * E_c * I_g = 2.4951E+013$

$$\text{factor} = 0.70$$

$$A_g = 120000.00$$

$$\text{Mean concrete strength: } f_c' = (f_c'_{jacket} * \text{Area}_{jacket} + f_c'_{core} * \text{Area}_{core}) / \text{Area}_{section} = 25.00$$

$$N = 1.8003E+006$$

$$E_c * I_g = E_c * I_{g,jacket} + E_c * I_{g,core} = 3.5645E+013$$

Calculation of Yielding Moment  $M_y$

Calculation of  $\delta / y$  and  $M_y$  according to Annex 7 -

Assuming neutral axis within flange ( $y < t/d$ , compression zone rectangular) with:

flange width,  $b = 400.00$

web width,  $bw = 200.00$

flange thickness,  $t = 200.00$

$$y = \text{Min}(y_{ten}, y_{com})$$

$$y_{ten} = 1.4407218E-005$$

with  $f_y = 596.2441$   
 $d = 367.00$   
 $y = 0.43617034$   
 $A = 0.05703506$   
 $B = 0.03712065$   
with  $p_t = 0.00774698$   
 $p_c = 0.01504455$   
 $p_v = 0.01367492$   
 $N = 1.8003E+006$   
 $b = 400.00$   
 $" = 0.08991826$   
 $y_{comp} = 1.2588515E-005$   
with  $f_c = 30.00$   
 $E_c = 25742.96$   
 $y = 0.45404063$   
 $A = 0.00723408$   
 $B = 0.01655203$   
with  $E_s = 200000.00$

CONFIRMATION:  $y = 0.45404063 < t/d$

Calculation of ratio  $I_b/I_d$

Adequate Lap Length:  $I_b/I_d \geq 1$

End Of Calculation of Shear Capacity for element: column JLC1 of floor 1

At local axis: 2

Integration Section: (b)

## Calculation No. 6

column C1, Floor 1

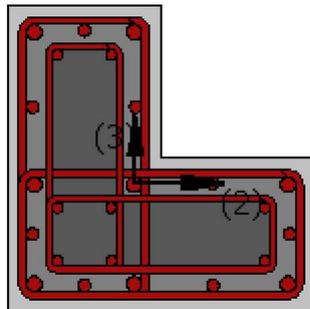
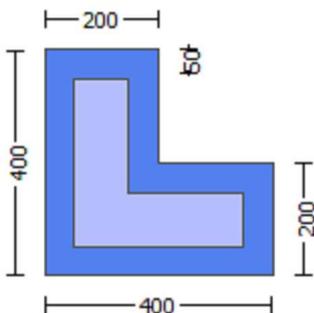
Limit State: Operational Level (data interpolation between analysis steps 50 and 51)

Analysis: Uniform +X

Check: Chord rotation capacity ( $\theta_u$ )

Edge: End

Local Axis: (2)



Start Of Calculation of Shear Capacity ratio for element: column JLC1 of floor 1

At Shear local axis: 3  
(Bending local axis: 2)  
Section Type: rcjlc

Constant Properties

Knowledge Factor,  $\gamma = 0.90$

Mean strength values are used for both shear and moment calculations.

Consequently:

Jacket

New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 30.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 625.00$

Concrete Elasticity,  $E_c = 25742.96$

Steel Elasticity,  $E_s = 200000.00$

Existing Column

Existing material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 18.00$

Existing material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 500.00$

Concrete Elasticity,  $E_c = 19940.411$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Jacket

New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 781.25$

Existing Column

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 625.00$

#####

Max Height,  $H_{max} = 400.00$

Min Height,  $H_{min} = 200.00$

Max Width,  $W_{max} = 400.00$

Min Width,  $W_{min} = 200.00$

Jacket Thickness,  $t_j = 50.00$

Cover Thickness,  $c = 15.00$

Mean Confinement Factor overall section = 1.00

Element Length,  $L = 3000.00$

Secondary Member

Smooth Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou, min} >= 1$ )

No FRP Wrapping

Stepwise Properties

At local axis: 3

EDGE -A-

Shear Force,  $V_a = 2.3476251E-006$

EDGE -B-

Shear Force,  $V_b = -2.3476251E-006$

BOTH EDGES

Axial Force,  $F = -1.8026E+006$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{st} = 0.00$

-Compression:  $A_{sc} = 5353.274$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{st, ten} = 2208.54$

-Compression:  $A_{sl, com} = 1137.257$

-Middle:  $A_{sl, mid} = 2007.478$

Calculation of Shear Capacity ratio ,  $V_e/V_r = 1.44151$

Member Controlled by Shear ( $V_e/V_r > 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 488890.189$

with  
 $M_{pr1} = \text{Max}(M_{u1+} , M_{u1-}) = 7.3334E+008$   
 $M_{u1+} = 7.3334E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination  
 $M_{u1-} = 6.3768E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination  
 $M_{pr2} = \text{Max}(M_{u2+} , M_{u2-}) = 7.3334E+008$   
 $M_{u2+} = 7.3334E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination  
 $M_{u2-} = 6.3768E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

-----  
Calculation of  $M_{u1+}$   
-----

-----  
Calculation of ultimate curvature  $\phi_u$  according to 4.1, Biskinis/Fardis 2013:

$\phi_u = 4.2207094E-005$

$M_u = 7.3334E+008$   
-----

with full section properties:

$b = 400.00$

$d = 367.00$

$d' = 33.00$

$v = 0.40931126$

$N = 1.8026E+006$

$f_c = 30.00$

$\alpha_1$  (5A.5, TBDY) = 0.002

Final value of  $\phi_u$ :  $\phi_u^* = \text{shear\_factor} * \text{Max}(\phi_u, \phi_c) = 0.00925707$

The Shear\_factor is considered equal to 1 (pure moment strength)

From (5.4b), TBDY:  $\phi_u = 0.00925707$

we (5.4c) = 0.0207149

$\alpha_{se}$  ((5.4d), TBDY) =  $(\alpha_{se1} * A_{ext} + \alpha_{se2} * A_{int}) / A_{sec} = 0.33836562$

$\alpha_{se1} = \text{Max}(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).

$\alpha_{se2} (> \alpha_{se1}) = \text{Max}(((A_{conf,max2} - A_{noConf2}) / A_{conf,max2}) * (A_{conf,min2} / A_{conf,max2}), 0) = 0.33836562$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe}, p_{sh,y} * F_{ywe}) = 6.12205$

Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

-----  
 $p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$

$p_{sh1}$  ((5.4d), TBDY) =  $L_{stir1} * A_{stir1} / (A_{sec} * s_1) = 0.00680678$

Lstir1 (Length of stirrups along Y) = 1040.00  
Astir1 (stirrups area) = 78.53982  
psh2 (5.4d) = Lstir2\*Astir2/(Asec\*s2) = 0.0012868  
Lstir2 (Length of stirrups along Y) = 768.00  
Astir2 (stirrups area) = 50.26548

psh\_y\*Fywe = psh1\*Fywe1+ps2\*Fywe2 = 6.12205  
psh1 ((5.4d), TBDY) = Lstir1\*Astir1/(Asec\*s1) = 0.00680678  
Lstir1 (Length of stirrups along X) = 1040.00  
Astir1 (stirrups area) = 78.53982  
psh2 ((5.4d), TBDY) = Lstir2\*Astir2/(Asec\*s2) = 0.0012868  
Lstir2 (Length of stirrups along X) = 768.00  
Astir2 (stirrups area) = 50.26548

Asec = 120000.00

s1 = 100.00

s2 = 250.00

fywe1 = 781.25

fywe2 = 625.00

fce = 30.00

From ((5.A5), TBDY), TBDY: cc = 0.002

c = confinement factor = 1.00

y1 = 0.0025

sh1 = 0.008

ft1 = 898.293

fy1 = 748.5775

su1 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb = 1.00

su1 = 0.4\*esu1\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1,ft1,fy1, it is considered  
characteristic value fsy1 = fs1/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (\text{lb}/\text{ld})^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs1 = (fs,jacket\*Asl,ten,jacket + fs,core\*Asl,ten,core)/Asl,ten = 748.5775

with Es1 = (Es,jacket\*Asl,ten,jacket + Es,core\*Asl,ten,core)/Asl,ten = 200000.00

y2 = 0.0025

sh2 = 0.008

ft2 = 886.7403

fy2 = 738.9503

su2 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb,min = 1.00

su2 = 0.4\*esu2\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu2\_nominal = 0.08,

For calculation of esu2\_nominal and y2, sh2,ft2,fy2, it is considered  
characteristic value fsy2 = fs2/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (\text{lb}/\text{ld})^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs2 = (fs,jacket\*Asl,com,jacket + fs,core\*Asl,com,core)/Asl,com = 738.9503

with Es2 = (Es,jacket\*Asl,com,jacket + Es,core\*Asl,com,core)/Asl,com = 200000.00

yv = 0.0025

shv = 0.008

ftv = 894.3662

fyv = 745.3052

suv = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb = 1.00

suv = 0.4\*esuv\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esuv\_nominal = 0.08,

considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY

For calculation of esuv\_nominal and yv, shv,ftv,fyv, it is considered

characteristic value  $f_{sv} = f_{sv}/1.2$ , from table 5.1, TBDY.

$y_1, sh_1, ft_1, fy_1$ , are also multiplied by  $\text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $f_{sv} = (f_{s,jacket} \cdot A_{s,mid,jacket} + f_{s,mid} \cdot A_{s,mid,core}) / A_{s,mid} = 745.3052$

with  $E_{sv} = (E_{s,jacket} \cdot A_{s,mid,jacket} + E_{s,mid} \cdot A_{s,mid,core}) / A_{s,mid} = 200000.00$

1 =  $A_{s,ten} / (b \cdot d) \cdot (f_{s1}/f_c) = 0.37540035$

2 =  $A_{s,com} / (b \cdot d) \cdot (f_{s2}/f_c) = 0.19082108$

$v = A_{s,mid} / (b \cdot d) \cdot (f_{sv}/f_c) = 0.33973286$

and confined core properties:

$b = 360.00$

$d = 347.00$

$d' = 13.00$

$f_{cc}$  (5A.2, TBDY) = 30.00

$cc$  (5A.5, TBDY) = 0.002

$c$  = confinement factor = 1.00

1 =  $A_{s,ten} / (b \cdot d) \cdot (f_{s1}/f_c) = 0.44115251$

2 =  $A_{s,com} / (b \cdot d) \cdot (f_{s2}/f_c) = 0.22424379$

$v = A_{s,mid} / (b \cdot d) \cdot (f_{sv}/f_c) = 0.39923778$

Case/Assumption: Unconfined full section - Steel rupture

' satisfies Eq. (4.3)

---->

$v < v_{s,y2}$  - LHS eq.(4.5) is not satisfied

---->

$v < v_{s,c}$  - RHS eq.(4.5) is not satisfied

---->

Case/Assumption Rejected.

---->

New Case/Assumption: Unconfined full section - Spalling of concrete cover

' satisfies Eq. (4.4)

---->

$v < v_{s,y1}$  - LHS eq.(4.7) is not satisfied

---->

$v < v_{c,y1}$  - RHS eq.(4.6) is satisfied

---->

$c_u$  (4.10) = 0.59761571

$M_{Rc}$  (4.17) = 7.3334E+008

---->

New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover

In expressions below, the following modifications have been made

-  $b, d, d'$  replaced by geometric parameters of the core:  $b_o, d_o, d'_o$

-  $N, 1, 2, v$  normalised to  $b_o \cdot d_o$ , instead of  $b \cdot d$

-  $f_{cc}, cc$  parameters of confined concrete,  $f_{cc}, cc$ , used in lieu of  $f_c, e_{cu}$

---->

Subcase: Rupture of tension steel

---->

$v^* < v^*_{s,y2}$  - LHS eq.(4.5) is not satisfied

---->

$v^* < v^*_{s,c}$  - LHS eq.(4.5) is not satisfied

---->

Subcase rejected

---->

New Subcase: Failure of compression zone

---->

$v^* < v^*_{c,y2}$  - LHS eq.(4.6) is not satisfied

---->

$v^* < v^*_{c,y1}$  - RHS eq.(4.6) is not satisfied

---->

$*c_u$  (4.11) = 0.62879536

$M_{Ro}$  (4.18) = 4.5866E+008

$M_{Ro} < 0.8 \cdot M_{Rc}$

---->

$u = c_u$  (unconfined full section) = 4.2207094E-005

$\mu = M_{Rc}$

-----  
Calculation of ratio  $l_b/l_d$

-----  
Adequate Lap Length:  $l_b/l_d \geq 1$

Calculation of Mu1-

Calculation of ultimate curvature  $\mu$  according to 4.1, Biskinis/Fardis 2013:

$$\mu = 4.8346386E-005$$

$$Mu = 6.3768E+008$$

with full section properties:

$$b = 200.00$$

$$d = 367.00$$

$$d' = 33.00$$

$$v = 0.81862252$$

$$N = 1.8026E+006$$

$$f_c = 30.00$$

$$c_o \text{ (5A.5, TBDY)} = 0.002$$

$$\text{Final value of } c_u: c_u^* = \text{shear\_factor} * \text{Max}(c_u, c_o) = 0.00925707$$

The Shear\_factor is considered equal to 1 (pure moment strength)

$$\text{From (5.4b), TBDY: } c_u = 0.00925707$$

$$w_e \text{ (5.4c)} = 0.0207149$$

$$a_{se} \text{ ((5.4d), TBDY)} = (a_{se1} * A_{ext} + a_{se2} * A_{int}) / A_{sec} = 0.33836562$$

$$a_{se1} = \text{Max}(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$a_{se2} (>= a_{se1}) = \text{Max}(((A_{conf,max2} - A_{noConf2}) / A_{conf,max2}) * (A_{conf,min2} / A_{conf,max2}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe}, p_{sh,y} * F_{ywe}) = 6.12205$$

Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

$$p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$$

$$p_{sh1} \text{ ((5.4d), TBDY)} = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$$

$$L_{stir1} \text{ (Length of stirrups along Y)} = 1040.00$$

$$A_{stir1} \text{ (stirrups area)} = 78.53982$$

$$p_{sh2} \text{ (5.4d)} = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$$

$$L_{stir2} \text{ (Length of stirrups along Y)} = 768.00$$

$$A_{stir2} \text{ (stirrups area)} = 50.26548$$

$$p_{sh,y} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$$

$$p_{sh1} \text{ ((5.4d), TBDY)} = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$$

$$L_{stir1} \text{ (Length of stirrups along X)} = 1040.00$$

$$A_{stir1} \text{ (stirrups area)} = 78.53982$$

$$p_{sh2} \text{ ((5.4d), TBDY)} = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$$

$$L_{stir2} \text{ (Length of stirrups along X)} = 768.00$$

$$A_{stir2} \text{ (stirrups area)} = 50.26548$$

Asec = 120000.00

s1 = 100.00

s2 = 250.00

fywe1 = 781.25

fywe2 = 625.00

fce = 30.00

From ((5.A.5), TBDY), TBDY: cc = 0.002

c = confinement factor = 1.00

y1 = 0.0025

sh1 = 0.008

ft1 = 886.7403

fy1 = 738.9503

su1 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb = 1.00

su1 = 0.4\*esu1\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1,ft1,fy1, it is considered  
characteristic value fsy1 = fs1/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (\text{lb}/\text{ld})^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs1 = (fsjacket\*Asl,ten,jacket + fs,core\*Asl,ten,core)/Asl,ten = 738.9503

with Es1 = (Es,jacket\*Asl,ten,jacket + Es,core\*Asl,ten,core)/Asl,ten = 200000.00

y2 = 0.0025

sh2 = 0.008

ft2 = 898.293

fy2 = 748.5775

su2 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb,min = 1.00

su2 = 0.4\*esu2\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu2\_nominal = 0.08,

For calculation of esu2\_nominal and y2, sh2,ft2,fy2, it is considered  
characteristic value fsy2 = fs2/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (\text{lb}/\text{ld})^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs2 = (fs,jacket\*Asl,com,jacket + fs,core\*Asl,com,core)/Asl,com = 748.5775

with Es2 = (Es,jacket\*Asl,com,jacket + Es,core\*Asl,com,core)/Asl,com = 200000.00

yv = 0.0025

shv = 0.008

ftv = 894.3662

fyv = 745.3052

suv = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb = 1.00

suv = 0.4\*esuv\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esuv\_nominal = 0.08,

considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY  
For calculation of esuv\_nominal and yv, shv,ftv,fyv, it is considered  
characteristic value fsyv = fsv/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (\text{lb}/\text{ld})^{2/3})$ , from 10.3.5, ASCE 41-17.

with fsv = (fs,jacket\*Asl,mid,jacket + fs,mid\*Asl,mid,core)/Asl,mid = 745.3052

with Esv = (Es,jacket\*Asl,mid,jacket + Es,mid\*Asl,mid,core)/Asl,mid = 200000.00

1 = Asl,ten/(b\*d)\*(fs1/fc) = 0.38164216

2 = Asl,com/(b\*d)\*(fs2/fc) = 0.7508007

v = Asl,mid/(b\*d)\*(fsv/fc) = 0.67946571

and confined core properties:

b = 160.00

d = 347.00

d' = 13.00

fcc (5A.2, TBDY) = 30.00

cc (5A.5, TBDY) = 0.002

c = confinement factor = 1.00  
1 =  $A_{sl,ten}/(b*d)*(f_{s1}/f_c) = 0.50454853$   
2 =  $A_{sl,com}/(b*d)*(f_{s2}/f_c) = 0.99259314$   
v =  $A_{sl,mid}/(b*d)*(f_{sv}/f_c) = 0.898285$

Case/Assumption: Unconfined full section - Steel rupture  
' satisfies Eq. (4.3)

---->  
v <  $v_{s,y2}$  - LHS eq.(4.5) is not satisfied

---->  
v <  $v_{s,c}$  - RHS eq.(4.5) is not satisfied

---->  
Case/Assumption Rejected.

---->  
New Case/Assumption: Unconfined full section - Spalling of concrete cover  
' satisfies Eq. (4.4)

---->  
v <  $s_{y1}$  - LHS eq.(4.7) is not satisfied

---->  
v <  $v_{c,y1}$  - RHS eq.(4.6) is satisfied

---->  
 $\epsilon_{cu}$  (4.10) = 0.52172715  
M<sub>Rc</sub> (4.17) = 6.3768E+008

---->  
New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover  
In expressions below, the following modifications have been made

- b, d, d' replaced by geometric parameters of the core: b<sub>o</sub>, d<sub>o</sub>, d'<sub>o</sub>
- N, 1, 2, v normalised to b<sub>o</sub>\*d<sub>o</sub>, instead of b\*d
- - parameters of confined concrete, f<sub>cc</sub>, c<sub>cc</sub>, used in lieu of f<sub>c</sub>, c<sub>u</sub>

---->  
Subcase: Rupture of tension steel

---->  
 $v^* < v^*_{s,y2}$  - LHS eq.(4.5) is not satisfied

---->  
 $v^* < v^*_{s,c}$  - LHS eq.(4.5) is not satisfied

---->  
Subcase rejected

---->  
New Subcase: Failure of compression zone

---->  
 $v^* < v^*_{c,y2}$  - LHS eq.(4.6) is not satisfied

---->  
 $v^* < v^*_{c,y1}$  - RHS eq.(4.6) is not satisfied

---->  
 $\epsilon^*_{cu}$  (4.11) = 0.5604845  
M<sub>Ro</sub> (4.18) = 4.7738E+008

M<sub>Ro</sub> < 0.8\*M<sub>Rc</sub>

---->  
u =  $\epsilon_{cu}$  (unconfined full section) = 4.8346386E-005  
Mu = M<sub>Rc</sub>

-----  
Calculation of ratio lb/d

-----  
Adequate Lap Length: lb/d >= 1

-----  
-----  
-----

Calculation of Mu2+

-----  
-----  
-----

Calculation of ultimate curvature u according to 4.1, Biskinis/Fardis 2013:

u = 4.2207094E-005  
Mu = 7.3334E+008

-----  
with full section properties:  
b = 400.00

$$d = 367.00$$

$$d' = 33.00$$

$$v = 0.40931126$$

$$N = 1.8026E+006$$

$$f_c = 30.00$$

$$c_o (5A.5, TBDY) = 0.002$$

$$\text{Final value of } c_u: c_u^* = \text{shear\_factor} * \text{Max}(c_u, c_c) = 0.00925707$$

The Shear\_factor is considered equal to 1 (pure moment strength)

$$\text{From (5.4b), TBDY: } c_u = 0.00925707$$

$$w_e (5.4c) = 0.0207149$$

$$a_{se} ((5.4d), TBDY) = (a_{se1} * A_{ext} + a_{se2} * A_{int}) / A_{sec} = 0.33836562$$

$$a_{se1} = \text{Max}(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$a_{se2} (> a_{se1}) = \text{Max}(((A_{conf,max2} - A_{noConf2}) / A_{conf,max2}) * (A_{conf,min2} / A_{conf,max2}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe}, p_{sh,y} * F_{ywe}) = 6.12205$$

Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

$$p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$$

$$p_{sh1} ((5.4d), TBDY) = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$$

$$L_{stir1} (\text{Length of stirrups along } Y) = 1040.00$$

$$A_{stir1} (\text{stirrups area}) = 78.53982$$

$$p_{sh2} (5.4d) = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$$

$$L_{stir2} (\text{Length of stirrups along } Y) = 768.00$$

$$A_{stir2} (\text{stirrups area}) = 50.26548$$

$$p_{sh,y} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$$

$$p_{sh1} ((5.4d), TBDY) = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$$

$$L_{stir1} (\text{Length of stirrups along } X) = 1040.00$$

$$A_{stir1} (\text{stirrups area}) = 78.53982$$

$$p_{sh2} ((5.4d), TBDY) = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$$

$$L_{stir2} (\text{Length of stirrups along } X) = 768.00$$

$$A_{stir2} (\text{stirrups area}) = 50.26548$$

$$A_{sec} = 120000.00$$

$$s1 = 100.00$$

$$s2 = 250.00$$

$$f_{ywe1} = 781.25$$

$$f_{ywe2} = 625.00$$

$$f_{ce} = 30.00$$

$$\text{From ((5.A5), TBDY), TBDY: } c_c = 0.002$$

$$c = \text{confinement factor} = 1.00$$

$$y1 = 0.0025$$

$$sh1 = 0.008$$

$$ft1 = 898.293$$

$$fy1 = 748.5775$$

su1 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/ld = 1.00

su1 = 0.4\*esu1\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1,ft1,fy1, it is considered  
characteristic value fsy1 = fs1/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs1 = (fs,jacket\*Asl,ten,jacket + fs,core\*Asl,ten,core)/Asl,ten = 748.5775

with Es1 = (Es,jacket\*Asl,ten,jacket + Es,core\*Asl,ten,core)/Asl,ten = 200000.00

y2 = 0.0025

sh2 = 0.008

ft2 = 886.7403

fy2 = 738.9503

su2 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb,min = 1.00

su2 = 0.4\*esu2\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu2\_nominal = 0.08,

For calculation of esu2\_nominal and y2, sh2,ft2,fy2, it is considered  
characteristic value fsy2 = fs2/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs2 = (fs,jacket\*Asl,com,jacket + fs,core\*Asl,com,core)/Asl,com = 738.9503

with Es2 = (Es,jacket\*Asl,com,jacket + Es,core\*Asl,com,core)/Asl,com = 200000.00

yv = 0.0025

shv = 0.008

ftv = 894.3662

fyv = 745.3052

suv = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/ld = 1.00

suv = 0.4\*esuv\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esuv\_nominal = 0.08,

considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY

For calculation of esuv\_nominal and yv, shv,ftv,fyv, it is considered  
characteristic value fsyv = fsv/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fsv = (fs,jacket\*Asl,mid,jacket + fs,mid\*Asl,mid,core)/Asl,mid = 745.3052

with Esv = (Es,jacket\*Asl,mid,jacket + Es,mid\*Asl,mid,core)/Asl,mid = 200000.00

1 = Asl,ten/(b\*d)\*(fs1/fc) = 0.37540035

2 = Asl,com/(b\*d)\*(fs2/fc) = 0.19082108

v = Asl,mid/(b\*d)\*(fsv/fc) = 0.33973286

and confined core properties:

b = 360.00

d = 347.00

d' = 13.00

fcc (5A.2, TBDY) = 30.00

cc (5A.5, TBDY) = 0.002

c = confinement factor = 1.00

1 = Asl,ten/(b\*d)\*(fs1/fc) = 0.44115251

2 = Asl,com/(b\*d)\*(fs2/fc) = 0.22424379

v = Asl,mid/(b\*d)\*(fsv/fc) = 0.39923778

Case/Assumption: Unconfined full section - Steel rupture

' satisfies Eq. (4.3)

---->

v < vs,y2 - LHS eq.(4.5) is not satisfied

---->

v < vs,c - RHS eq.(4.5) is not satisfied

---->

Case/Assumption Rejected.

---->

New Case/Assumption: Unconfined full section - Spalling of concrete cover  
satisfies Eq. (4.4)

--->  
 $v < s, y1$  - LHS eq.(4.7) is not satisfied

--->  
 $v < v_c, y1$  - RHS eq.(4.6) is satisfied

--->  
 $c_u$  (4.10) = 0.59761571  
 $M_{Rc}$  (4.17) = 7.3334E+008

--->  
New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover

In expressions below, the following modifications have been made

- $b, d, d'$  replaced by geometric parameters of the core:  $b_o, d_o, d'_o$
- $N, 1, 2, v$  normalised to  $b_o*d_o$ , instead of  $b*d$
- $f_c, c_c$  parameters of confined concrete,  $f_{cc}, c_{cc}$  used in lieu of  $f_c, c_c$

--->  
Subcase: Rupture of tension steel

--->  
 $v^* < v^*s, y2$  - LHS eq.(4.5) is not satisfied

--->  
 $v^* < v^*s, c$  - LHS eq.(4.5) is not satisfied

--->  
Subcase rejected

--->  
New Subcase: Failure of compression zone

--->  
 $v^* < v^*c, y2$  - LHS eq.(4.6) is not satisfied

--->  
 $v^* < v^*c, y1$  - RHS eq.(4.6) is not satisfied

--->  
 $*c_u$  (4.11) = 0.62879536  
 $M_{Ro}$  (4.18) = 4.5866E+008

$M_{Ro} < 0.8*M_{Rc}$

--->  
 $u = c_u$  (unconfined full section) = 4.2207094E-005  
 $M_u = M_{Rc}$

-----  
Calculation of ratio  $l_b/d$

-----  
Adequate Lap Length:  $l_b/d \geq 1$

-----  
Calculation of  $M_{u2}$

-----  
Calculation of ultimate curvature  $u$  according to 4.1, Biskinis/Fardis 2013:

$u = 4.8346386E-005$

$M_u = 6.3768E+008$

-----  
with full section properties:

$b = 200.00$

$d = 367.00$

$d' = 33.00$

$v = 0.81862252$

$N = 1.8026E+006$

$f_c = 30.00$

$c_o$  (5A.5, TBDY) = 0.002

Final value of  $c_u$ :  $c_u^* = \text{shear\_factor} * \text{Max}(c_u, c_c) = 0.00925707$

The Shear\_factor is considered equal to 1 (pure moment strength)

From (5.4b), TBDY:  $c_u = 0.00925707$

$w_e$  (5.4c) = 0.0207149

$a_{se}$  ((5.4d), TBDY) =  $(a_{se1}*A_{ext} + a_{se2}*A_{int})/A_{sec} = 0.33836562$

$a_{se1} = \text{Max}(((A_{conf,max1} - A_{noConf1})/A_{conf,max1}) * (A_{conf,min1}/A_{conf,max1}), 0) = 0.33836562$

The definitions of  $A_{noConf}, A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization

of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).

$ase2 (>= ase1) = \text{Max}(((A_{conf,max2} - A_{noConf2})/A_{conf,max2}) * (A_{conf,min2}/A_{conf,max2}), 0) = 0.33836562$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$psh_{min} * F_{ywe} = \text{Min}(psh_x * F_{ywe}, psh_y * F_{ywe}) = 6.12205$

Expression (5.4d) for  $psh_{min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

-----  
 $psh_x * F_{ywe} = psh1 * F_{ywe1} + ps2 * F_{ywe2} = 6.12205$

$psh1$  ((5.4d), TBDY) =  $L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$

$L_{stir1}$  (Length of stirrups along Y) = 1040.00

$A_{stir1}$  (stirrups area) = 78.53982

$psh2$  (5.4d) =  $L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$

$L_{stir2}$  (Length of stirrups along Y) = 768.00

$A_{stir2}$  (stirrups area) = 50.26548

-----  
 $psh_y * F_{ywe} = psh1 * F_{ywe1} + ps2 * F_{ywe2} = 6.12205$

$psh1$  ((5.4d), TBDY) =  $L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$

$L_{stir1}$  (Length of stirrups along X) = 1040.00

$A_{stir1}$  (stirrups area) = 78.53982

$psh2$  ((5.4d), TBDY) =  $L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$

$L_{stir2}$  (Length of stirrups along X) = 768.00

$A_{stir2}$  (stirrups area) = 50.26548

-----  
 $A_{sec} = 120000.00$

$s1 = 100.00$

$s2 = 250.00$

$f_{ywe1} = 781.25$

$f_{ywe2} = 625.00$

$f_{ce} = 30.00$

From ((5.A5), TBDY), TBDY:  $cc = 0.002$

$c = \text{confinement factor} = 1.00$

$y1 = 0.0025$

$sh1 = 0.008$

$ft1 = 886.7403$

$fy1 = 738.9503$

$su1 = 0.032$

using (30) in Biskinis/Fardis (2013) multiplied with  $shear\_factor$

and also multiplied by the  $shear\_factor$  according to 15.7.1.4, with

$Shear\_factor = 1.00$

$lo/lo_{min} = lb/ld = 1.00$

$su1 = 0.4 * esu1\_nominal$  ((5.5), TBDY) = 0.032

From table 5A.1, TBDY:  $esu1\_nominal = 0.08$ ,

For calculation of  $esu1\_nominal$  and  $y1, sh1, ft1, fy1$ , it is considered

characteristic value  $fsy1 = fs1/1.2$ , from table 5.1, TBDY.

$y1, sh1, ft1, fy1$ , are also multiplied by  $\text{Min}(1, 1.25 * (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $fs1 = (fs_{jacket} * A_{sl,ten,jacket} + fs_{core} * A_{sl,ten,core}) / A_{sl,ten} = 738.9503$

with  $Es1 = (Es_{jacket} * A_{sl,ten,jacket} + Es_{core} * A_{sl,ten,core}) / A_{sl,ten} = 200000.00$

$y2 = 0.0025$

$sh_2 = 0.008$   
 $ft_2 = 898.293$   
 $fy_2 = 748.5775$   
 $su_2 = 0.032$   
 using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
 and also multiplied by the shear\_factor according to 15.7.1.4, with  
 Shear\_factor = 1.00  
 $lo/lo_{min} = lb/lb_{min} = 1.00$   
 $su_2 = 0.4 * esu_{2,nominal} ((5.5), TBDY) = 0.032$   
 From table 5A.1, TBDY:  $esu_{2,nominal} = 0.08$ ,  
 For calculation of  $esu_{2,nominal}$  and  $y_2, sh_2, ft_2, fy_2$ , it is considered  
 characteristic value  $fsy_2 = fs_2/1.2$ , from table 5.1, TBDY.  
 $y_1, sh_1, ft_1, fy_1$ , are also multiplied by  $Min(1, 1.25 * (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.  
 with  $fs_2 = (fs_{jacket} * Asl_{com,jacket} + fs_{core} * Asl_{com,core}) / Asl_{com} = 748.5775$   
 with  $Es_2 = (Es_{jacket} * Asl_{com,jacket} + Es_{core} * Asl_{com,core}) / Asl_{com} = 200000.00$   
 $yv = 0.0025$   
 $shv = 0.008$   
 $ftv = 894.3662$   
 $fyv = 745.3052$   
 $suv = 0.032$   
 using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
 and also multiplied by the shear\_factor according to 15.7.1.4, with  
 Shear\_factor = 1.00  
 $lo/lo_{min} = lb/ld = 1.00$   
 $suv = 0.4 * esuv_{nominal} ((5.5), TBDY) = 0.032$   
 From table 5A.1, TBDY:  $esuv_{nominal} = 0.08$ ,  
 considering characteristic value  $fsyv = fsv/1.2$ , from table 5.1, TBDY  
 For calculation of  $esuv_{nominal}$  and  $yv, shv, ftv, fyv$ , it is considered  
 characteristic value  $fsyv = fsv/1.2$ , from table 5.1, TBDY.  
 $y_1, sh_1, ft_1, fy_1$ , are also multiplied by  $Min(1, 1.25 * (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.  
 with  $fsv = (fs_{jacket} * Asl_{mid,jacket} + fs_{mid} * Asl_{mid,core}) / Asl_{mid} = 745.3052$   
 with  $Es_v = (Es_{jacket} * Asl_{mid,jacket} + Es_{mid} * Asl_{mid,core}) / Asl_{mid} = 200000.00$   
 $1 = Asl_{ten} / (b * d) * (fs_1 / fc) = 0.38164216$   
 $2 = Asl_{com} / (b * d) * (fs_2 / fc) = 0.7508007$   
 $v = Asl_{mid} / (b * d) * (fsv / fc) = 0.67946571$   
 and confined core properties:  
 $b = 160.00$   
 $d = 347.00$   
 $d' = 13.00$   
 $fcc (5A.2, TBDY) = 30.00$   
 $cc (5A.5, TBDY) = 0.002$   
 $c = \text{confinement factor} = 1.00$   
 $1 = Asl_{ten} / (b * d) * (fs_1 / fc) = 0.50454853$   
 $2 = Asl_{com} / (b * d) * (fs_2 / fc) = 0.99259314$   
 $v = Asl_{mid} / (b * d) * (fsv / fc) = 0.898285$   
 Case/Assumption: Unconfined full section - Steel rupture  
 ' satisfies Eq. (4.3)  
 --->  
 $v < vs_{y2}$  - LHS eq.(4.5) is not satisfied  
 --->  
 $v < vs_c$  - RHS eq.(4.5) is not satisfied  
 --->  
 Case/Assumption Rejected.  
 --->  
 New Case/Assumption: Unconfined full section - Spalling of concrete cover  
 ' satisfies Eq. (4.4)  
 --->  
 $v < s_{y1}$  - LHS eq.(4.7) is not satisfied  
 --->  
 $v < vc_{y1}$  - RHS eq.(4.6) is satisfied  
 --->  
 $cu (4.10) = 0.52172715$   
 $MRC (4.17) = 6.3768E+008$   
 --->  
 New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover  
 In expressions below, the following modifications have been made  
 - b, d, d' replaced by geometric parameters of the core:  $bo, do, d'o$

- N<sub>1</sub>, N<sub>2</sub>, v normalised to b<sub>o</sub>\*d<sub>o</sub>, instead of b\*d
- parameters of confined concrete, f<sub>cc</sub>, c<sub>c</sub>, used in lieu of f<sub>c</sub>, e<sub>cu</sub>

--->  
Subcase: Rupture of tension steel

--->  
v\* < v\*s,y2 - LHS eq.(4.5) is not satisfied

--->  
v\* < v\*s,c - LHS eq.(4.5) is not satisfied

--->  
Subcase rejected

--->  
New Subcase: Failure of compression zone

--->  
v\* < v\*c,y2 - LHS eq.(4.6) is not satisfied

--->  
v\* < v\*c,y1 - RHS eq.(4.6) is not satisfied

--->  
\*c<sub>u</sub> (4.11) = 0.5604845

M<sub>Ro</sub> (4.18) = 4.7738E+008

M<sub>Ro</sub> < 0.8\*M<sub>Rc</sub>

--->  
u = c<sub>u</sub> (unconfined full section) = 4.8346386E-005  
Mu = M<sub>Rc</sub>

-----  
Calculation of ratio l<sub>b</sub>/l<sub>d</sub>

-----  
Adequate Lap Length: l<sub>b</sub>/l<sub>d</sub> >= 1

-----  
Calculation of Shear Strength V<sub>r</sub> = Min(V<sub>r1</sub>, V<sub>r2</sub>) = 339150.739

-----  
Calculation of Shear Strength at edge 1, V<sub>r1</sub> = 339150.739

V<sub>r1</sub> = V<sub>Col</sub> ((10.3), ASCE 41-17) = k<sub>nl</sub>\*V<sub>Col0</sub>

V<sub>Col0</sub> = 339150.739

k<sub>nl</sub> = 1 (zero step-static loading)

-----  
NOTE: In expression (10-3) 'V<sub>s</sub> = A<sub>v</sub>\*f<sub>y</sub>\*d/s' is replaced by 'V<sub>s</sub>+ f\*V<sub>f</sub>'  
where V<sub>f</sub> is the contribution of FRPs (11.3), ACI 440).

-----  
= 1 (normal-weight concrete)

Mean concrete strength: f<sub>c</sub>' = (f<sub>c</sub>'<sub>jacket</sub>\*Area<sub>jacket</sub> + f<sub>c</sub>'<sub>core</sub>\*Area<sub>core</sub>)/Area<sub>section</sub> = 25.00, but f<sub>c</sub>'<sup>0.5</sup> <= 8.3  
MPa (22.5.3.1, ACI 318-14)

M/V<sub>d</sub> = 4.00

Mu = 2.0345E+006

Vu = 2.3476251E-006

d = 0.8\*h = 320.00

Nu = 1.8026E+006

Ag = 80000.00

From (11.5.4.8), ACI 318-14: V<sub>s</sub> = V<sub>s,jacket</sub> + V<sub>s,core</sub> = 471238.898

where:

V<sub>s,jacket</sub> = V<sub>s,j1</sub> + V<sub>s,j2</sub> = 471238.898

V<sub>s,j1</sub> = 314159.265 is calculated for section web jacket, with:

d = 320.00

A<sub>v</sub> = 157079.633

f<sub>y</sub> = 625.00

s = 100.00

V<sub>s,j1</sub> is multiplied by Col,j1 = 1.00

s/d = 0.3125

V<sub>s,j2</sub> = 157079.633 is calculated for section flange jacket, with:

d = 160.00

A<sub>v</sub> = 157079.633

f<sub>y</sub> = 625.00

s = 100.00

V<sub>s,j2</sub> is multiplied by Col,j2 = 1.00

$s/d = 0.625$   
 $V_{s,core} = V_{s,c1} + V_{s,c2} = 0.00$   
 $V_{s,c1} = 0.00$  is calculated for section web core, with:  
 $d = 240.00$   
 $A_v = 100530.965$   
 $f_y = 500.00$   
 $s = 250.00$   
 $V_{s,c1}$  is multiplied by  $Col,c1 = 0.00$   
 $s/d = 1.04167$   
 $V_{s,c2} = 0.00$  is calculated for section flange core, with:  
 $d = 80.00$   
 $A_v = 100530.965$   
 $f_y = 500.00$   
 $s = 250.00$   
 $V_{s,c2}$  is multiplied by  $Col,c2 = 0.00$   
 $s/d = 3.125$   
 $V_f ((11-3)-(11.4), ACI 440) = 0.00$   
 From (11-11), ACI 440:  $V_s + V_f \leq 212577.225$   
 $bw = 200.00$

Calculation of Shear Strength at edge 2,  $V_{r2} = 339150.739$   
 $V_{r2} = V_{Col} ((10.3), ASCE 41-17) = knl * V_{Col0}$   
 $V_{Col0} = 339150.739$   
 $knl = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ '  
 where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)  
 Mean concrete strength:  $f'_c = (f'_{c,jacket} * Area_{jacket} + f'_{c,core} * Area_{core}) / Area_{section} = 25.00$ , but  $f'_c^{0.5} \leq 8.3$   
 MPa (22.5.3.1, ACI 318-14)

$M/d = 4.00$   
 $M_u = 6739.685$   
 $V_u = 2.3476251E-006$   
 $d = 0.8 * h = 320.00$   
 $N_u = 1.8026E+006$   
 $A_g = 80000.00$   
 From (11.5.4.8), ACI 318-14:  $V_s = V_{s,jacket} + V_{s,core} = 471238.898$

where:

$V_{s,jacket} = V_{s,j1} + V_{s,j2} = 471238.898$   
 $V_{s,j1} = 314159.265$  is calculated for section web jacket, with:  
 $d = 320.00$   
 $A_v = 157079.633$   
 $f_y = 625.00$   
 $s = 100.00$   
 $V_{s,j1}$  is multiplied by  $Col,j1 = 1.00$   
 $s/d = 0.3125$   
 $V_{s,j2} = 157079.633$  is calculated for section flange jacket, with:  
 $d = 160.00$   
 $A_v = 157079.633$   
 $f_y = 625.00$   
 $s = 100.00$   
 $V_{s,j2}$  is multiplied by  $Col,j2 = 1.00$   
 $s/d = 0.625$

$V_{s,core} = V_{s,c1} + V_{s,c2} = 0.00$   
 $V_{s,c1} = 0.00$  is calculated for section web core, with:  
 $d = 240.00$   
 $A_v = 100530.965$   
 $f_y = 500.00$   
 $s = 250.00$   
 $V_{s,c1}$  is multiplied by  $Col,c1 = 0.00$   
 $s/d = 1.04167$   
 $V_{s,c2} = 0.00$  is calculated for section flange core, with:  
 $d = 80.00$   
 $A_v = 100530.965$   
 $f_y = 500.00$

s = 250.00  
Vs,c2 is multiplied by Col,c2 = 0.00  
s/d = 3.125  
Vf ((11-3)-(11.4), ACI 440) = 0.00  
From (11-11), ACI 440: Vs + Vf <= 212577.225  
bw = 200.00

-----  
-----  
End Of Calculation of Shear Capacity ratio for element: column JLC1 of floor 1  
At local axis: 3

-----  
-----  
Start Of Calculation of Shear Capacity ratio for element: column JLC1 of floor 1  
At Shear local axis: 2  
(Bending local axis: 3)  
Section Type: rcjcs

Constant Properties

-----  
Knowledge Factor,  $\gamma = 0.90$   
Mean strength values are used for both shear and moment calculations.  
Consequently:  
Jacket  
New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 30.00$   
New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 625.00$   
Concrete Elasticity,  $E_c = 25742.96$   
Steel Elasticity,  $E_s = 200000.00$   
Existing Column  
Existing material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 18.00$   
Existing material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 500.00$   
Concrete Elasticity,  $E_c = 19940.411$   
Steel Elasticity,  $E_s = 200000.00$

#####  
Note: Especially for the calculation of moment strengths,  
the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Jacket  
New material: Steel Strength,  $f_s = 1.25 * f_{sm} = 781.25$   
Existing Column  
Existing material: Steel Strength,  $f_s = 1.25 * f_{sm} = 625.00$   
#####

Max Height,  $H_{max} = 400.00$   
Min Height,  $H_{min} = 200.00$   
Max Width,  $W_{max} = 400.00$   
Min Width,  $W_{min} = 200.00$   
Jacket Thickness,  $t_j = 50.00$   
Cover Thickness,  $c = 15.00$   
Mean Confinement Factor overall section = 1.00  
Element Length,  $L = 3000.00$   
Secondary Member  
Smooth Bars  
Ductile Steel  
Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Adequate Lap Length ( $l_o/l_{ou, min} >= 1$ )  
No FRP Wrapping

-----  
-----  
Stepwise Properties

-----  
-----  
At local axis: 2  
EDGE -A-  
Shear Force,  $V_a = -3.6347847E-006$   
EDGE -B-  
Shear Force,  $V_b = 3.6347847E-006$   
BOTH EDGES

Axial Force,  $F = -1.8026E+006$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 0.00$

-Compression:  $As_c = 5353.274$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t,ten} = 1137.257$

-Compression:  $As_{c,com} = 2208.54$

-Middle:  $As_{mid} = 2007.478$

Calculation of Shear Capacity ratio,  $V_e/V_r = 1.18753$

Member Controlled by Shear ( $V_e/V_r > 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 402750.026$

with

$M_{pr1} = \text{Max}(Mu_{1+}, Mu_{1-}) = 6.0413E+008$

$Mu_{1+} = 6.0413E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$Mu_{1-} = 5.4612E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(Mu_{2+}, Mu_{2-}) = 6.0413E+008$

$Mu_{2+} = 6.0413E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination

$Mu_{2-} = 5.4612E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

Calculation of  $Mu_{1+}$

Calculation of ultimate curvature  $\phi_u$  according to 4.1, Biskinis/Fardis 2013:

$\phi_u = 1.2613948E-005$

$M_u = 6.0413E+008$

with full section properties:

$b = 400.00$

$d = 367.00$

$d' = 33.00$

$v = 0.40931126$

$N = 1.8026E+006$

$f_c = 30.00$

$\omega (5A.5, TBDY) = 0.002$

Final value of  $\phi_u$ :  $\phi_u^* = \text{shear\_factor} * \text{Max}(\phi_u, \phi_c) = 0.00925707$

The Shear\_factor is considered equal to 1 (pure moment strength)

From (5.4b), TBDY:  $\phi_u = 0.00925707$

$\omega_e (5.4c) = 0.0207149$

$\omega_{ase} ((5.4d), TBDY) = (\omega_{ase1} * A_{ext} + \omega_{ase2} * A_{int}) / A_{sec} = 0.33836562$

$\omega_{ase1} = \text{Max}(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length

equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).

$\omega_{ase2} (> = \omega_{ase1}) = \text{Max}(((A_{conf,max2} - A_{noConf2}) / A_{conf,max2}) * (A_{conf,min2} / A_{conf,max2}), 0) = 0.33836562$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

Aconf,max2 = 45264.00 is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

Aconf,min2 = 0.00 is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area Aconf,max2 by a length equal to half the clear spacing between internal hoops.

AnoConf2 = 32309.333 is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$psh_{,min} * F_{ywe} = \text{Min}(psh_{,x} * F_{ywe}, psh_{,y} * F_{ywe}) = 6.12205$

Expression (5.4d) for  $psh_{,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

-----  
 $psh_{,x} * F_{ywe} = psh_1 * F_{ywe1} + ps_2 * F_{ywe2} = 6.12205$   
 $psh_1$  ((5.4d), TBDY) =  $L_{stir1} * A_{stir1} / (A_{sec} * s_1) = 0.00680678$   
Lstir1 (Length of stirrups along Y) = 1040.00  
Astir1 (stirrups area) = 78.53982  
 $psh_2$  (5.4d) =  $L_{stir2} * A_{stir2} / (A_{sec} * s_2) = 0.0012868$   
Lstir2 (Length of stirrups along Y) = 768.00  
Astir2 (stirrups area) = 50.26548

-----  
 $psh_{,y} * F_{ywe} = psh_1 * F_{ywe1} + ps_2 * F_{ywe2} = 6.12205$   
 $psh_1$  ((5.4d), TBDY) =  $L_{stir1} * A_{stir1} / (A_{sec} * s_1) = 0.00680678$   
Lstir1 (Length of stirrups along X) = 1040.00  
Astir1 (stirrups area) = 78.53982  
 $psh_2$  ((5.4d), TBDY) =  $L_{stir2} * A_{stir2} / (A_{sec} * s_2) = 0.0012868$   
Lstir2 (Length of stirrups along X) = 768.00  
Astir2 (stirrups area) = 50.26548

-----  
Asec = 120000.00  
s1 = 100.00  
s2 = 250.00  
fywe1 = 781.25  
fywe2 = 625.00  
fce = 30.00

From ((5.A5), TBDY), TBDY: cc = 0.002  
c = confinement factor = 1.00

y1 = 0.0025  
sh1 = 0.008  
ft1 = 886.7403  
fy1 = 738.9503  
su1 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

lo/lou,min = lb/ld = 1.00

su1 =  $0.4 * esu1_{nominal}$  ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1,ft1,fy1, it is considered characteristic value  $fsy1 = fs1/1.2$ , from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 * (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $fs1 = (fs_{jacket} * A_{sl,ten,jacket} + fs_{core} * A_{sl,ten,core}) / A_{sl,ten} = 738.9503$

with  $Es1 = (Es_{jacket} * A_{sl,ten,jacket} + Es_{core} * A_{sl,ten,core}) / A_{sl,ten} = 200000.00$

y2 = 0.0025  
sh2 = 0.008  
ft2 = 898.293  
fy2 = 748.5775  
su2 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

lo/lou,min = lb/lb,min = 1.00

su2 =  $0.4 * esu2_{nominal}$  ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu2\_nominal = 0.08,

For calculation of esu2\_nominal and y2, sh2,ft2,fy2, it is considered characteristic value  $fsy2 = fs2/1.2$ , from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 * (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $fs2 = (fs_{jacket} * A_{sl,com,jacket} + fs_{core} * A_{sl,com,core}) / A_{sl,com} = 748.5775$

with  $Es2 = (Es_{jacket} * A_{sl,com,jacket} + Es_{core} * A_{sl,com,core}) / A_{sl,com} = 200000.00$

yv = 0.0025

```

shv = 0.008
ftv = 894.3662
fyv = 745.3052
suv = 0.032
  using (30) in Biskinis/Fardis (2013) multiplied with shear_factor
  and also multiplied by the shear_factor according to 15.7.1.4, with
  Shear_factor = 1.00
  lo/lou,min = lb/lb = 1.00
  suv = 0.4*esuv_nominal ((5.5), TBDY) = 0.032
  From table 5A.1, TBDY: esuv_nominal = 0.08,
  considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY
  For calculation of esuv_nominal and yv, shv,ftv,fyv, it is considered
  characteristic value fsyv = fsv/1.2, from table 5.1, TBDY.
  y1, sh1,ft1,fy1, are also multiplied by Min(1,1.25*(lb/lb)^2/3), from 10.3.5, ASCE 41-17.
  with fsv = (fs,jacket*Asl,mid,jacket + fs,mid*Asl,mid,core)/Asl,mid = 745.3052
  with Esv = (Es,jacket*Asl,mid,jacket + Es,mid*Asl,mid,core)/Asl,mid = 200000.00
  1 = Asl,ten/(b*d)*(fs1/fc) = 0.19082108
  2 = Asl,com/(b*d)*(fs2/fc) = 0.37540035
  v = Asl,mid/(b*d)*(fsv/fc) = 0.33973286
and confined core properties:
b = 360.00
d = 347.00
d' = 13.00
fcc (5A.2, TBDY) = 30.00
cc (5A.5, TBDY) = 0.002
  c = confinement factor = 1.00
  1 = Asl,ten/(b*d)*(fs1/fc) = 0.22424379
  2 = Asl,com/(b*d)*(fs2/fc) = 0.44115251
  v = Asl,mid/(b*d)*(fsv/fc) = 0.39923778
Case/Assumption: Unconfined full section - Steel rupture
' satisfies Eq. (4.3)
---->
v < vs,y2 - LHS eq.(4.5) is not satisfied
---->
v < vs,c - RHS eq.(4.5) is not satisfied
---->
Case/Assumption Rejected.
---->
New Case/Assumption: Unconfined full section - Spalling of concrete cover
' satisfies Eq. (4.4)
---->
v < s,y1 - LHS eq.(4.7) is not satisfied
---->
v < vc,y1 - RHS eq.(4.6) is satisfied
---->
cu (4.10) = 0.37716748
MRc (4.17) = 7.0585E+008
---->
New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover
In expressions below, the following modifications have been made
- b, d, d' replaced by geometric parameters of the core: bo, do, d'o
- N, 1, 2, v normalised to bo*do, instead of b*d
- - parameters of confined concrete, fcc, cc, used in lieu of fc, ecu
---->
Subcase: Rupture of tension steel
---->
v* < v*,y2 - LHS eq.(4.5) is not satisfied
---->
v* < v*,c - LHS eq.(4.5) is not satisfied
---->
Subcase rejected
---->
New Subcase: Failure of compression zone
---->
v* < v*,y2 - LHS eq.(4.6) is not satisfied
---->
v* < v*,y1 - RHS eq.(4.6) is not satisfied

```

--->

$$*cu(4.11) = 0.4569298$$

$$MRo(4.18) = 6.0413E+008$$

--->

$$u = cu(4.2) = 1.2613948E-005$$

$$Mu = MRo$$

-----  
Calculation of ratio lb/d

-----  
Adequate Lap Length: lb/d >= 1  
-----  
-----

-----  
Calculation of Mu1-  
-----  
-----

-----  
Calculation of ultimate curvature u according to 4.1, Biskinis/Fardis 2013:

$$u = 3.1475172E-005$$

$$Mu = 5.4612E+008$$

-----  
with full section properties:

$$b = 200.00$$

$$d = 367.00$$

$$d' = 33.00$$

$$v = 0.81862252$$

$$N = 1.8026E+006$$

$$fc = 30.00$$

$$co(5A.5, TBDY) = 0.002$$

$$\text{Final value of } cu: cu^* = \text{shear\_factor} * \text{Max}(cu, cc) = 0.00925707$$

The Shear\_factor is considered equal to 1 (pure moment strength)

$$\text{From (5.4b), TBDY: } cu = 0.00925707$$

$$we(5.4c) = 0.0207149$$

$$ase((5.4d), TBDY) = (ase1*Aext+ase2*Aint)/Asec = 0.33836562$$

$$ase1 = \text{Max}(((Aconf,max1-AnoConf1)/Aconf,max1)*(Aconf,min1/Aconf,max1),0) = 0.33836562$$

The definitions of AnoConf, Aconf,min and Aconf,max are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

Aconf,max1 = 89600.00 is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

Aconf,min1 = 59225.00 is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area Aconf,max1 by a length equal to half the clear spacing between external hoops.

AnoConf1 = 43733.333 is the unconfined external core area which is equal to  $bi^2/6$  as defined at (A.2).

$$ase2 (>=ase1) = \text{Max}(((Aconf,max2-AnoConf2)/Aconf,max2)*(Aconf,min2/Aconf,max2),0) = 0.33836562$$

The definitions of AnoConf, Aconf,min and Aconf,max are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

Aconf,max2 = 45264.00 is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

Aconf,min2 = 0.00 is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area Aconf,max2 by a length equal to half the clear spacing between internal hoops.

AnoConf2 = 32309.333 is the unconfined internal core area which is equal to  $bi^2/6$  as defined at (A.2).

$$psh,min*Fywe = \text{Min}(psh,x*Fywe, psh,y*Fywe) = 6.12205$$

Expression (5.4d) for psh,min\*Fywe has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

$$psh_x*Fywe = psh1*Fywe1+ps2*Fywe2 = 6.12205$$

$$psh1((5.4d), TBDY) = Lstir1*Astir1/(Asec*s1) = 0.00680678$$

$$Lstir1(\text{Length of stirrups along Y}) = 1040.00$$

$$Astir1(\text{stirrups area}) = 78.53982$$

psh2 (5.4d) =  $L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$   
Lstir2 (Length of stirrups along Y) = 768.00  
Astir2 (stirrups area) = 50.26548

psh\_y \* Fywe = psh1 \* Fywe1 + ps2 \* Fywe2 = 6.12205  
psh1 ((5.4d), TBDY) =  $L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$   
Lstir1 (Length of stirrups along X) = 1040.00  
Astir1 (stirrups area) = 78.53982  
psh2 ((5.4d), TBDY) =  $L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$   
Lstir2 (Length of stirrups along X) = 768.00  
Astir2 (stirrups area) = 50.26548

Asec = 120000.00

s1 = 100.00

s2 = 250.00

fywe1 = 781.25

fywe2 = 625.00

fce = 30.00

From ((5.A.5), TBDY), TBDY: cc = 0.002

c = confinement factor = 1.00

y1 = 0.0025

sh1 = 0.008

ft1 = 898.293

fy1 = 748.5775

su1 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb = 1.00

su1 =  $0.4 * esu1\_nominal ((5.5), TBDY) = 0.032$

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1, ft1, fy1, it is considered  
characteristic value fsy1 = fs1/1.2, from table 5.1, TBDY.

y1, sh1, ft1, fy1, are also multiplied by  $Min(1, 1.25 * (lb/lb)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs1 =  $(fs_{jacket} * A_{sl,ten,jacket} + fs_{core} * A_{sl,ten,core}) / A_{sl,ten} = 748.5775$

with Es1 =  $(Es_{jacket} * A_{sl,ten,jacket} + Es_{core} * A_{sl,ten,core}) / A_{sl,ten} = 200000.00$

y2 = 0.0025

sh2 = 0.008

ft2 = 886.7403

fy2 = 738.9503

su2 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb,min = 1.00

su2 =  $0.4 * esu2\_nominal ((5.5), TBDY) = 0.032$

From table 5A.1, TBDY: esu2\_nominal = 0.08,

For calculation of esu2\_nominal and y2, sh2, ft2, fy2, it is considered  
characteristic value fsy2 = fs2/1.2, from table 5.1, TBDY.

y1, sh1, ft1, fy1, are also multiplied by  $Min(1, 1.25 * (lb/lb)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs2 =  $(fs_{jacket} * A_{sl,com,jacket} + fs_{core} * A_{sl,com,core}) / A_{sl,com} = 738.9503$

with Es2 =  $(Es_{jacket} * A_{sl,com,jacket} + Es_{core} * A_{sl,com,core}) / A_{sl,com} = 200000.00$

yv = 0.0025

shv = 0.008

ftv = 894.3662

fyv = 745.3052

suv = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb = 1.00

suv =  $0.4 * esuv\_nominal ((5.5), TBDY) = 0.032$

From table 5A.1, TBDY: esuv\_nominal = 0.08,

considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY  
For calculation of esuv\_nominal and yv, shv, ftv, fyv, it is considered

characteristic value fsyv = fsv/1.2, from table 5.1, TBDY.

y1, sh1, ft1, fy1, are also multiplied by  $Min(1, 1.25 * (lb/lb)^{2/3})$ , from 10.3.5, ASCE 41-17.

```

with fsv = (fs,jacket*Asl,mid,jacket + fs,mid*Asl,mid,core)/Asl,mid = 745.3052
with Esv = (Es,jacket*Asl,mid,jacket + Es,mid*Asl,mid,core)/Asl,mid = 200000.00
1 = Asl,ten/(b*d)*(fs1/fc) = 0.7508007
2 = Asl,com/(b*d)*(fs2/fc) = 0.38164216
v = Asl,mid/(b*d)*(fsv/fc) = 0.67946571
and confined core properties:
b = 160.00
d = 347.00
d' = 13.00
fcc (5A.2, TBDY) = 30.00
cc (5A.5, TBDY) = 0.002
c = confinement factor = 1.00
1 = Asl,ten/(b*d)*(fs1/fc) = 0.99259314
2 = Asl,com/(b*d)*(fs2/fc) = 0.50454853
v = Asl,mid/(b*d)*(fsv/fc) = 0.898285
Case/Assumption: Unconfined full section - Steel rupture
' satisfies Eq. (4.3)
--->
v < vs,y2 - LHS eq.(4.5) is not satisfied
--->
v < vs,c - RHS eq.(4.5) is not satisfied
--->
Case/Assumption Rejected.
--->
New Case/Assumption: Unconfined full section - Spalling of concrete cover
' satisfies Eq. (4.4)
--->
v < s,y1 - LHS eq.(4.7) is not satisfied
--->
v < vc,y1 - RHS eq.(4.6) is not satisfied
--->
cu (4.11) = 0.80138156
MRc (4.18) = 5.4612E+008
--->
New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover
In expressions below, the following modifications have been made
- b, d, d' replaced by geometric parameters of the core: bo, do, d'o
- N, 1, 2, v normalised to bo*do, instead of b*d
- - parameters of confined concrete, fcc, cc, used in lieu of fc, ecu
--->
Subcase: Rupture of tension steel
--->
v* < v*s,y2 - LHS eq.(4.5) is not satisfied
--->
v* < v*s,c - LHS eq.(4.5) is not satisfied
--->
Subcase rejected
--->
New Subcase: Failure of compression zone
--->
v* < v*c,y2 - LHS eq.(4.6) is not satisfied
--->
v* < v*c,y1 - RHS eq.(4.6) is not satisfied
--->
*cu (4.11) = 0.80674817
MRo (4.18) = 2.8580E+008
MRo < 0.8*MRc
--->
u = cu (unconfined full section) = 3.1475172E-005
Mu = MRc
-----
Calculation of ratio lb/d
-----
Adequate Lap Length: lb/d >= 1
-----
-----

```

Calculation of  $\mu_{2+}$

Calculation of ultimate curvature  $\mu$  according to 4.1, Biskinis/Fardis 2013:

$$\mu = 1.2613948E-005$$

$$\mu = 6.0413E+008$$

with full section properties:

$$b = 400.00$$

$$d = 367.00$$

$$d' = 33.00$$

$$v = 0.40931126$$

$$N = 1.8026E+006$$

$$f_c = 30.00$$

$$c_o (5A.5, TBDY) = 0.002$$

$$\text{Final value of } \mu: \mu^* = \text{shear\_factor} * \text{Max}(\mu_c, \mu_{cc}) = 0.00925707$$

The Shear\_factor is considered equal to 1 (pure moment strength)

$$\text{From (5.4b), TBDY: } \mu_c = 0.00925707$$

$$\mu_{cc} (5.4c) = 0.0207149$$

$$a_{se} (5.4d), TBDY) = (a_{se1} * A_{ext} + a_{se2} * A_{int}) / A_{sec} = 0.33836562$$

$$a_{se1} = \text{Max}(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$a_{se2} (>= a_{se1}) = \text{Max}(((A_{conf,max2} - A_{noConf2}) / A_{conf,max2}) * (A_{conf,min2} / A_{conf,max2}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe}, p_{sh,y} * F_{ywe}) = 6.12205$$

Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

$$p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$$

$$p_{sh1} ((5.4d), TBDY) = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$$

$$L_{stir1} (\text{Length of stirrups along Y}) = 1040.00$$

$$A_{stir1} (\text{stirrups area}) = 78.53982$$

$$p_{sh2} (5.4d) = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$$

$$L_{stir2} (\text{Length of stirrups along Y}) = 768.00$$

$$A_{stir2} (\text{stirrups area}) = 50.26548$$

$$p_{sh,y} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$$

$$p_{sh1} ((5.4d), TBDY) = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$$

$$L_{stir1} (\text{Length of stirrups along X}) = 1040.00$$

$$A_{stir1} (\text{stirrups area}) = 78.53982$$

$$p_{sh2} ((5.4d), TBDY) = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$$

$$L_{stir2} (\text{Length of stirrups along X}) = 768.00$$

$$A_{stir2} (\text{stirrups area}) = 50.26548$$

$$A_{sec} = 120000.00$$

s1 = 100.00  
s2 = 250.00  
fywe1 = 781.25  
fywe2 = 625.00  
fce = 30.00

From ((5.A5), TBDY), TBDY: cc = 0.002  
c = confinement factor = 1.00

y1 = 0.0025  
sh1 = 0.008  
ft1 = 886.7403  
fy1 = 738.9503  
su1 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/ld = 1.00

su1 = 0.4\*esu1\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1,ft1,fy1, it is considered  
characteristic value fsy1 = fs1/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (\text{lb}/\text{ld})^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs1 = (fs,jacket\*Asl,ten,jacket + fs,core\*Asl,ten,core)/Asl,ten = 738.9503

with Es1 = (Es,jacket\*Asl,ten,jacket + Es,core\*Asl,ten,core)/Asl,ten = 200000.00

y2 = 0.0025  
sh2 = 0.008  
ft2 = 898.293  
fy2 = 748.5775  
su2 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb,min = 1.00

su2 = 0.4\*esu2\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu2\_nominal = 0.08,

For calculation of esu2\_nominal and y2, sh2,ft2,fy2, it is considered  
characteristic value fsy2 = fs2/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (\text{lb}/\text{ld})^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs2 = (fs,jacket\*Asl,com,jacket + fs,core\*Asl,com,core)/Asl,com = 748.5775

with Es2 = (Es,jacket\*Asl,com,jacket + Es,core\*Asl,com,core)/Asl,com = 200000.00

yv = 0.0025  
shv = 0.008  
ftv = 894.3662  
fyv = 745.3052  
suv = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/ld = 1.00

suv = 0.4\*esuv\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esuv\_nominal = 0.08,

considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY

For calculation of esuv\_nominal and yv, shv,ftv,fyv, it is considered  
characteristic value fsyv = fsv/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (\text{lb}/\text{ld})^{2/3})$ , from 10.3.5, ASCE 41-17.

with fsv = (fs,jacket\*Asl,mid,jacket + fs,mid\*Asl,mid,core)/Asl,mid = 745.3052

with Esv = (Es,jacket\*Asl,mid,jacket + Es,mid\*Asl,mid,core)/Asl,mid = 200000.00

1 = Asl,ten/(b\*d)\*(fs1/fc) = 0.19082108  
2 = Asl,com/(b\*d)\*(fs2/fc) = 0.37540035  
v = Asl,mid/(b\*d)\*(fsv/fc) = 0.33973286

and confined core properties:

b = 360.00  
d = 347.00  
d' = 13.00

fcc (5A.2, TBDY) = 30.00

cc (5A.5, TBDY) = 0.002

c = confinement factor = 1.00

1 = Asl,ten/(b\*d)\*(fs1/fc) = 0.22424379

$$2 = A_{sl,com}/(b*d)*(f_s2/f_c) = 0.44115251$$

$$v = A_{sl,mid}/(b*d)*(f_{sv}/f_c) = 0.39923778$$

Case/Assumption: Unconfined full section - Steel rupture

' satisfies Eq. (4.3)

---->  
 $v < v_{s,y2}$  - LHS eq.(4.5) is not satisfied

---->  
 $v < v_{s,c}$  - RHS eq.(4.5) is not satisfied

---->  
Case/Assumption Rejected.

---->  
New Case/Assumption: Unconfined full section - Spalling of concrete cover

' satisfies Eq. (4.4)

---->  
 $v < v_{s,y1}$  - LHS eq.(4.7) is not satisfied

---->  
 $v < v_{c,y1}$  - RHS eq.(4.6) is satisfied

---->  
 $\epsilon_{cu}$  (4.10) = 0.37716748  
M<sub>Rc</sub> (4.17) = 7.0585E+008

---->  
New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover

In expressions below, the following modifications have been made

- b, d, d' replaced by geometric parameters of the core: b<sub>o</sub>, d<sub>o</sub>, d'<sub>o</sub>
- N<sub>1</sub>, N<sub>2</sub>, v normalised to b<sub>o</sub>\*d<sub>o</sub>, instead of b\*d
- parameters of confined concrete, f<sub>cc</sub>, ε<sub>cc</sub>, used in lieu of f<sub>c</sub>, ε<sub>cu</sub>

---->  
Subcase: Rupture of tension steel

---->  
 $v^* < v^*_{s,y2}$  - LHS eq.(4.5) is not satisfied

---->  
 $v^* < v^*_{s,c}$  - LHS eq.(4.5) is not satisfied

---->  
Subcase rejected

---->  
New Subcase: Failure of compression zone

---->  
 $v^* < v^*_{c,y2}$  - LHS eq.(4.6) is not satisfied

---->  
 $v^* < v^*_{c,y1}$  - RHS eq.(4.6) is not satisfied

---->  
 $\epsilon^*_{cu}$  (4.11) = 0.4569298  
M<sub>Ro</sub> (4.18) = 6.0413E+008

---->  
 $u = \epsilon^*_{cu}$  (4.2) = 1.2613948E-005  
M<sub>u</sub> = M<sub>Ro</sub>

-----  
Calculation of ratio l<sub>b</sub>/l<sub>d</sub>

-----  
Adequate Lap Length: l<sub>b</sub>/l<sub>d</sub> >= 1

-----  
Calculation of M<sub>u2</sub>-

-----  
Calculation of ultimate curvature  $\epsilon^*_{cu}$  according to 4.1, Biskinis/Fardis 2013:

$\epsilon^*_{cu} = 3.1475172E-005$   
M<sub>u</sub> = 5.4612E+008

-----  
with full section properties:

b = 200.00  
d = 367.00  
d' = 33.00  
v = 0.81862252

$$N = 1.8026E+006$$

$$f_c = 30.00$$

$$c_o (5A.5, TBDY) = 0.002$$

$$\text{Final value of } c_u: c_u^* = \text{shear\_factor} * \text{Max}(c_u, c_c) = 0.00925707$$

The Shear\_factor is considered equal to 1 (pure moment strength)

$$\text{From (5.4b), TBDY: } c_u = 0.00925707$$

$$w_e (5.4c) = 0.0207149$$

$$a_{se} ((5.4d), TBDY) = (a_{se1} * A_{ext} + a_{se2} * A_{int}) / A_{sec} = 0.33836562$$

$$a_{se1} = \text{Max}(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$$

The definitions of  $A_{noConf1}$ ,  $A_{conf,min1}$  and  $A_{conf,max1}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$a_{se2} (>= a_{se1}) = \text{Max}(((A_{conf,max2} - A_{noConf2}) / A_{conf,max2}) * (A_{conf,min2} / A_{conf,max2}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe}, p_{sh,y} * F_{ywe}) = 6.12205$$

Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

$$p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$$

$$p_{sh1} ((5.4d), TBDY) = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$$

$$L_{stir1} (\text{Length of stirrups along Y}) = 1040.00$$

$$A_{stir1} (\text{stirrups area}) = 78.53982$$

$$p_{sh2} (5.4d) = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$$

$$L_{stir2} (\text{Length of stirrups along Y}) = 768.00$$

$$A_{stir2} (\text{stirrups area}) = 50.26548$$

$$p_{sh,y} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$$

$$p_{sh1} ((5.4d), TBDY) = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$$

$$L_{stir1} (\text{Length of stirrups along X}) = 1040.00$$

$$A_{stir1} (\text{stirrups area}) = 78.53982$$

$$p_{sh2} ((5.4d), TBDY) = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$$

$$L_{stir2} (\text{Length of stirrups along X}) = 768.00$$

$$A_{stir2} (\text{stirrups area}) = 50.26548$$

$$A_{sec} = 120000.00$$

$$s1 = 100.00$$

$$s2 = 250.00$$

$$f_{ywe1} = 781.25$$

$$f_{ywe2} = 625.00$$

$$f_{ce} = 30.00$$

$$\text{From ((5.A5), TBDY), TBDY: } c_c = 0.002$$

$$c = \text{confinement factor} = 1.00$$

$$y1 = 0.0025$$

$$sh1 = 0.008$$

$$ft1 = 898.293$$

$$fy1 = 748.5775$$

$$su1 = 0.032$$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with

Shear\_factor = 1.00

lo/lou,min = lb/d = 1.00

su1 = 0.4\*esu1\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1,ft1,fy1, it is considered characteristic value fsy1 = fs1/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs1 = (fs,jacket\*Asl,ten,jacket + fs,core\*Asl,ten,core)/Asl,ten = 748.5775

with Es1 = (Es,jacket\*Asl,ten,jacket + Es,core\*Asl,ten,core)/Asl,ten = 200000.00

y2 = 0.0025

sh2 = 0.008

ft2 = 886.7403

fy2 = 738.9503

su2 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor

and also multiplied by the shear\_factor according to 15.7.1.4, with

Shear\_factor = 1.00

lo/lou,min = lb/lb,min = 1.00

su2 = 0.4\*esu2\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu2\_nominal = 0.08,

For calculation of esu2\_nominal and y2, sh2,ft2,fy2, it is considered characteristic value fsy2 = fs2/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs2 = (fs,jacket\*Asl,com,jacket + fs,core\*Asl,com,core)/Asl,com = 738.9503

with Es2 = (Es,jacket\*Asl,com,jacket + Es,core\*Asl,com,core)/Asl,com = 200000.00

yv = 0.0025

shv = 0.008

ftv = 894.3662

fyv = 745.3052

suv = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor

and also multiplied by the shear\_factor according to 15.7.1.4, with

Shear\_factor = 1.00

lo/lou,min = lb/d = 1.00

suv = 0.4\*esuv\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esuv\_nominal = 0.08,

considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY

For calculation of esuv\_nominal and yv, shv,ftv,fyv, it is considered characteristic value fsyv = fsv/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fsv = (fs,jacket\*Asl,mid,jacket + fs,mid\*Asl,mid,core)/Asl,mid = 745.3052

with Esv = (Es,jacket\*Asl,mid,jacket + Es,mid\*Asl,mid,core)/Asl,mid = 200000.00

1 = Asl,ten/(b\*d)\*(fs1/fc) = 0.7508007

2 = Asl,com/(b\*d)\*(fs2/fc) = 0.38164216

v = Asl,mid/(b\*d)\*(fsv/fc) = 0.67946571

and confined core properties:

b = 160.00

d = 347.00

d' = 13.00

fcc (5A.2, TBDY) = 30.00

cc (5A.5, TBDY) = 0.002

c = confinement factor = 1.00

1 = Asl,ten/(b\*d)\*(fs1/fc) = 0.99259314

2 = Asl,com/(b\*d)\*(fs2/fc) = 0.50454853

v = Asl,mid/(b\*d)\*(fsv/fc) = 0.898285

Case/Assumption: Unconfined full section - Steel rupture

' satisfies Eq. (4.3)

--->

v < vs,y2 - LHS eq.(4.5) is not satisfied

--->

v < vs,c - RHS eq.(4.5) is not satisfied

--->

Case/Assumption Rejected.

--->

New Case/Assumption: Unconfined full section - Spalling of concrete cover

' satisfies Eq. (4.4)

--->

v < s,y1 - LHS eq.(4.7) is not satisfied

---->

v < vc,y1 - RHS eq.(4.6) is not satisfied

---->

cu (4.11) = 0.80138156

MRC (4.18) = 5.4612E+008

---->

New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover

In expressions below, the following modifications have been made

- b, d, d' replaced by geometric parameters of the core: bo, do, d'o
- N, 1, 2, v normalised to bo\*do, instead of b\*d
- - parameters of confined concrete, fcc, cc, used in lieu of fc, ecu

---->

Subcase: Rupture of tension steel

---->

v\* < v\*s,y2 - LHS eq.(4.5) is not satisfied

---->

v\* < v\*s,c - LHS eq.(4.5) is not satisfied

---->

Subcase rejected

---->

New Subcase: Failure of compression zone

---->

v\* < v\*c,y2 - LHS eq.(4.6) is not satisfied

---->

v\* < v\*c,y1 - RHS eq.(4.6) is not satisfied

---->

\*cu (4.11) = 0.80674817

MRO (4.18) = 2.8580E+008

MRO < 0.8\*MRC

---->

u = cu (unconfined full section) = 3.1475172E-005

Mu = MRC

-----  
Calculation of ratio lb/d

Adequate Lap Length: lb/d >= 1

-----  
Calculation of Shear Strength Vr = Min(Vr1,Vr2) = 339150.739

-----  
Calculation of Shear Strength at edge 1, Vr1 = 339150.739

Vr1 = VCol ((10.3), ASCE 41-17) = knl\*VCol0

VCol0 = 339150.739

knl = 1 (zero step-static loading)

-----  
NOTE: In expression (10-3) 'Vs = Av\*fy\*d/s' is replaced by 'Vs+ f\*Vf'  
where Vf is the contribution of FRPs (11.3), ACI 440).

-----  
= 1 (normal-weight concrete)

Mean concrete strength: fc' = (fc'\_jacket\*Area\_jacket + fc'\_core\*Area\_core)/Area\_section = 25.00, but fc'^0.5 <= 8.3 MPa (22.5.3.1, ACI 318-14)

M/Vd = 4.00

Mu = 6.7415E+007

Vu = 3.6347847E-006

d = 0.8\*h = 320.00

Nu = 1.8026E+006

Ag = 80000.00

From (11.5.4.8), ACI 318-14: Vs = Vs\_jacket + Vs\_core = 471238.898

where:

Vs\_jacket = Vs\_j1 + Vs\_j2 = 471238.898

Vs\_j1 = 157079.633 is calculated for section web jacket, with:

d = 160.00

Av = 157079.633

$f_y = 625.00$   
 $s = 100.00$   
 $V_{s,j1}$  is multiplied by  $Col,j1 = 1.00$   
 $s/d = 0.625$   
 $V_{s,j2} = 314159.265$  is calculated for section flange jacket, with:  
 $d = 320.00$   
 $A_v = 157079.633$   
 $f_y = 625.00$   
 $s = 100.00$   
 $V_{s,j2}$  is multiplied by  $Col,j2 = 1.00$   
 $s/d = 0.3125$   
 $V_{s,core} = V_{s,c1} + V_{s,c2} = 0.00$   
 $V_{s,c1} = 0.00$  is calculated for section web core, with:  
 $d = 80.00$   
 $A_v = 100530.965$   
 $f_y = 500.00$   
 $s = 250.00$   
 $V_{s,c1}$  is multiplied by  $Col,c1 = 0.00$   
 $s/d = 3.125$   
 $V_{s,c2} = 0.00$  is calculated for section flange core, with:  
 $d = 240.00$   
 $A_v = 100530.965$   
 $f_y = 500.00$   
 $s = 250.00$   
 $V_{s,c2}$  is multiplied by  $Col,c2 = 0.00$   
 $s/d = 1.04167$   
 $V_f ((11-3)-(11.4), ACI 440) = 0.00$   
 From (11-11), ACI 440:  $V_s + V_f \leq 212577.225$   
 $bw = 200.00$

Calculation of Shear Strength at edge 2,  $V_{r2} = 339150.739$   
 $V_{r2} = V_{Col} ((10.3), ASCE 41-17) = knl * V_{Col0}$   
 $V_{Col0} = 339150.739$   
 $knl = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ '  
 where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)  
 Mean concrete strength:  $f_c' = (f_c'_{jacket} * Area_{jacket} + f_c'_{core} * Area_{core}) / Area_{section} = 25.00$ , but  $f_c'^{0.5} \leq 8.3$   
 MPa (22.5.3.1, ACI 318-14)  
 $M/d = 4.00$   
 $M_u = 5.3429E+007$   
 $V_u = 3.6347847E-006$   
 $d = 0.8 * h = 320.00$   
 $N_u = 1.8026E+006$   
 $A_g = 80000.00$   
 From (11.5.4.8), ACI 318-14:  $V_s = V_{s,jacket} + V_{s,core} = 471238.898$   
 where:  
 $V_{s,jacket} = V_{s,j1} + V_{s,j2} = 471238.898$   
 $V_{s,j1} = 157079.633$  is calculated for section web jacket, with:  
 $d = 160.00$   
 $A_v = 157079.633$   
 $f_y = 625.00$   
 $s = 100.00$   
 $V_{s,j1}$  is multiplied by  $Col,j1 = 1.00$   
 $s/d = 0.625$   
 $V_{s,j2} = 314159.265$  is calculated for section flange jacket, with:  
 $d = 320.00$   
 $A_v = 157079.633$   
 $f_y = 625.00$   
 $s = 100.00$   
 $V_{s,j2}$  is multiplied by  $Col,j2 = 1.00$   
 $s/d = 0.3125$   
 $V_{s,core} = V_{s,c1} + V_{s,c2} = 0.00$   
 $V_{s,c1} = 0.00$  is calculated for section web core, with:

d = 80.00

Av = 100530.965

fy = 500.00

s = 250.00

Vs,c1 is multiplied by Col,c1 = 0.00

s/d = 3.125

Vs,c2 = 0.00 is calculated for section flange core, with:

d = 240.00

Av = 100530.965

fy = 500.00

s = 250.00

Vs,c2 is multiplied by Col,c2 = 0.00

s/d = 1.04167

Vf ((11-3)-(11.4), ACI 440) = 0.00

From (11-11), ACI 440: Vs + Vf <= 212577.225

bw = 200.00

-----  
End Of Calculation of Shear Capacity ratio for element: column JLC1 of floor 1

At local axis: 2

-----  
Start Of Calculation of Chord Rotation Capacity for element: column JLC1 of floor 1

At local axis: 2

Integration Section: (b)

Section Type: rcjlc

Constant Properties

-----  
Knowledge Factor,  $\phi = 0.90$

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE 41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Jacket

New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 30.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 625.00$

Concrete Elasticity,  $E_c = 25742.96$

Steel Elasticity,  $E_s = 200000.00$

Existing Column

Existing material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 18.00$

Existing material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 500.00$

Concrete Elasticity,  $E_c = 19940.411$

Steel Elasticity,  $E_s = 200000.00$

Max Height,  $H_{max} = 400.00$

Min Height,  $H_{min} = 200.00$

Max Width,  $W_{max} = 400.00$

Min Width,  $W_{min} = 200.00$

Jacket Thickness,  $t_j = 50.00$

Cover Thickness,  $c = 15.00$

Element Length,  $L = 3000.00$

Secondary Member

Smooth Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_b/l_d >= 1$ )

No FRP Wrapping

-----  
Stepwise Properties

Bending Moment,  $M = 18804.004$

Shear Force,  $V_2 = 97644.268$

Shear Force,  $V_3 = 30.34672$

Axial Force,  $F = -1.8003E+006$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{st} = 0.00$

-Compression:  $A_{sc} = 5353.274$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{st,ten} = 2208.54$

-Compression:  $A_{sc,com} = 1137.257$

-Middle:  $A_{st,mid} = 2007.478$

Longitudinal External Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{st,ten,jacket} = 1746.726$

-Compression:  $A_{sc,com,jacket} = 829.3805$

-Middle:  $A_{st,mid,jacket} = 1545.664$

Longitudinal Internal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{st,ten,core} = 461.8141$

-Compression:  $A_{sc,com,core} = 307.8761$

-Middle:  $A_{st,mid,core} = 461.8141$

Mean Diameter of Tension Reinforcement,  $DbL = 16.60$

Existing component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_{R} = \alpha u = 0.00389461$   
 $u = \gamma + \rho = 0.00432734$

- Calculation of  $\gamma$  -

$\gamma = (M_y * L_s / 3) / E_{eff} = 0.00432734$  ((4.29), Biskinis Phd))

$M_y = 5.2276E+008$

$L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 619.6389

From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 2.4951E+013$

factor = 0.70

$A_g = 120000.00$

Mean concrete strength:  $f_c' = (f_c'_{jacket} * Area_{jacket} + f_c'_{core} * Area_{core}) / Area_{section} = 25.00$

$N = 1.8003E+006$

$E_c * I_g = E_c * I_{g,jacket} + E_c * I_{g,core} = 3.5645E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\gamma$  and  $M_y$  according to Annex 7 -

Assuming neutral axis out of flange ( $\gamma > t/d$ , compression zone NOT rectangular)  
extended expressions (2.1)-(2.7) of Biskinis-Fardis (2009) are adopted with:

flange width,  $b = 400.00$

web width,  $b_w = 200.00$

flange thickness,  $t = 200.00$

$\gamma = \text{Min}(\gamma_{ten}, \gamma_{com})$

$\gamma_{ten} = 1.6044767E-005$

with  $f_y = 596.2441$

$d = 367.00$

$\gamma = 0.49371549$

$A = 0.18421443$

$B = 0.10663695$

with  $p_t = 0.01504455$

$p_c = 0.00774698$

$p_v = 0.01367492$

$N = 1.8003E+006$

$b = 400.00$

$\alpha = 0.08991826$

$\gamma_{comp} = 1.0454056E-005$

with  $f_c = 30.00$

$E_c = 25742.96$

$\gamma = 0.54674444$

$A = 0.08461246$

$B = 0.06549972$

with  $E_s = 200000.00$   
CONFIRMATION:  $y = 0.54674444 > t/d$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

- Calculation of  $p$  -

From table 10-8:  $p = 0.00$

with:

- Columns not controlled by inadequate development or splicing along the clear height because  $l_b/d \geq 1$

shear control ratio  $V_y E / V_{CoI} O E = 1.44151$

$d = d_{\text{external}} = 367.00$

$s = s_{\text{external}} = 100.00$

-  $t = s_1 + s_2 + 2 * t_f / b_w * (f_{fe} / f_s) = 0.00809358$

jacket:  $s_1 = A_{v1} * L_{\text{stir1}} / (s_1 * A_g) = 0.00680678$

$A_{v1} = 78.53982$ , is the area of every stirrup parallel to loading (shear) direction

$L_{\text{stir1}} = 1040.00$ , is the total Length of all stirrups parallel to loading (shear) direction

$s_1 = 100.00$

core:  $s_2 = A_{v2} * L_{\text{stir2}} / (s_2 * A_g) = 0.0012868$

$A_{v2} = 50.26548$ , is the area of every stirrup parallel to loading (shear) direction

$L_{\text{stir2}} = 768.00$ , is the total Length of all stirrups parallel to loading (shear) direction

$s_2 = 250.00$

The term  $2 * t_f / b_w * (f_{fe} / f_s)$  is implemented to account for FRP contribution

where  $f = 2 * t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength

All these variables have already been given in Shear control ratio calculation.

For the normalisation  $f_s$  of jacket is used.

$N_{UD} = 1.8003E+006$

$A_g = 120000.00$

$f_{cE} = (f_{c\_jacket} * \text{Area}_{\text{jacket}} + f_{c\_core} * \text{Area}_{\text{core}}) / \text{section\_area} = 25.00$

$f_{yIE} = (f_{y\_ext\_Long\_Reinf} * \text{Area}_{\text{ext\_Long\_Reinf}} + f_{y\_int\_Long\_Reinf} * \text{Area}_{\text{int\_Long\_Reinf}}) / \text{Area}_{\text{Tot\_Long\_Rein}} = 596.2441$

$f_{ytE} = (f_{y\_ext\_Trans\_Reinf} * s_1 + f_{y\_int\_Trans\_Reinf} * s_2) / (s_1 + s_2) = 605.1263$

$p_l = \text{Area}_{\text{Tot\_Long\_Rein}} / (b * d) = 0.03646644$

$b = 400.00$

$d = 367.00$

$f_{cE} = 25.00$

End Of Calculation of Chord Rotation Capacity for element: column JLC1 of floor 1

At local axis: 2

Integration Section: (b)

## Calculation No. 7

column C1, Floor 1

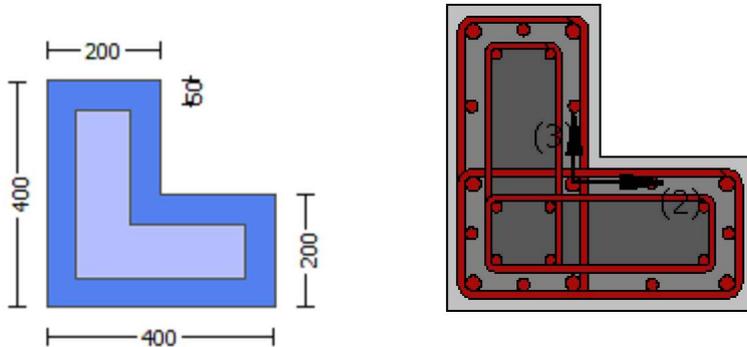
Limit State: Operational Level (data interpolation between analysis steps 50 and 51)

Analysis: Uniform +X

Check: Shear capacity VRd

Edge: End

Local Axis: (3)



Start Of Calculation of Shear Capacity for element: column JLC1 of floor 1

At local axis: 3

Integration Section: (b)

Section Type: rcjlcs

Constant Properties

Knowledge Factor,  $\gamma = 0.90$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE 41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Jacket

New material of Secondary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 20.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 500.00$

Concrete Elasticity,  $E_c = 25742.96$

Steel Elasticity,  $E_s = 200000.00$

Existing Column

Existing material of Secondary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 12.00$

Existing material of Secondary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 400.00$

Concrete Elasticity,  $E_c = 19940.411$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of  $\gamma$  for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE 41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE 41-17).

Jacket

New material: Concrete Strength,  $f_c = f_{cm} = 30.00$

New material: Steel Strength,  $f_s = f_{sm} = 625.00$

Existing Column

Existing material: Concrete Strength,  $f_c = f_{cm} = 18.00$

Existing material: Steel Strength,  $f_s = f_{sm} = 500.00$

#####

Max Height,  $H_{max} = 400.00$

Min Height,  $H_{min} = 200.00$

Max Width,  $W_{max} = 400.00$

Min Width,  $W_{min} = 200.00$

Jacket Thickness,  $t_j = 50.00$

Cover Thickness,  $c = 15.00$

Element Length,  $L = 3000.00$

Secondary Member  
Smooth Bars  
Ductile Steel  
Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Adequate Lap Length ( $l_o/l_{ou, \min} = l_b/l_d \geq 1$ )  
No FRP Wrapping

-----  
Stepwise Properties  
-----

EDGE -A-  
Bending Moment,  $M_a = 1.8819E+007$   
Shear Force,  $V_a = -30.34672$   
EDGE -B-  
Bending Moment,  $M_b = 18804.004$   
Shear Force,  $V_b = 30.34672$   
BOTH EDGES  
Axial Force,  $F = -1.8003E+006$   
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $A_{st} = 0.00$   
-Compression:  $A_{sc} = 5353.274$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $A_{st, \text{ten}} = 2208.54$   
-Compression:  $A_{st, \text{com}} = 1137.257$   
-Middle:  $A_{st, \text{mid}} = 2007.478$   
Mean Diameter of Tension Reinforcement,  $D_{bL, \text{ten}} = 16.60$

-----  
Existing component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = \phi V_n = 360068.846$   
 $V_n$  ((10.3), ASCE 41-17) =  $k_n l V_{c0} = 400076.496$   
 $V_{c0} = 400076.496$   
 $k_n = 1.00$   
 $\text{displacement\_ductility\_demand} = 0.01979823$

-----  
NOTE: In expression (10-3) ' $V_s = A_v f_y d/s$ ' is replaced by ' $V_s + \phi V_f$ '  
where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

-----  
= 1 (normal-weight concrete)  
Mean concrete strength:  $f'_c = (f'_{c, \text{jacket}} \cdot \text{Area}_{\text{jacket}} + f'_{c, \text{core}} \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 16.66667$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $M_u = 18804.004$   
 $V_u = 30.34672$   
 $d = 0.8 \cdot h = 320.00$   
 $N_u = 1.8003E+006$   
 $A_g = 80000.00$   
From (11.5.4.8), ACI 318-14:  $V_s = V_{s, \text{jacket}} + V_{s, \text{core}} = 376991.118$   
where:  
 $V_{s, \text{jacket}} = V_{s, j1} + V_{s, j2} = 376991.118$   
 $V_{s, j1} = 251327.412$  is calculated for section web jacket, with:  
 $d = 320.00$   
 $A_v = 157079.633$   
 $f_y = 500.00$   
 $s = 100.00$   
 $V_{s, j1}$  is multiplied by  $\text{Col}_{j1} = 1.00$   
 $s/d = 0.3125$   
 $V_{s, j2} = 125663.706$  is calculated for section flange jacket, with:  
 $d = 160.00$   
 $A_v = 157079.633$   
 $f_y = 500.00$   
 $s = 100.00$   
 $V_{s, j2}$  is multiplied by  $\text{Col}_{j2} = 1.00$   
 $s/d = 0.625$   
 $V_{s, \text{core}} = V_{s, c1} + V_{s, c2} = 0.00$

$V_{s,c1} = 0.00$  is calculated for section web core, with:

$$d = 240.00$$

$$A_v = 100530.965$$

$$f_y = 400.00$$

$$s = 250.00$$

$V_{s,c1}$  is multiplied by  $Col,c1 = 0.00$

$$s/d = 1.04167$$

$V_{s,c2} = 0.00$  is calculated for section flange core, with:

$$d = 80.00$$

$$A_v = 100530.965$$

$$f_y = 400.00$$

$$s = 250.00$$

$V_{s,c2}$  is multiplied by  $Col,c2 = 0.00$

$$s/d = 3.125$$

$V_f ((11-3)-(11.4), ACI 440) = 0.00$

From (11-11), ACI 440:  $V_s + V_f \leq 173568.578$

$$bw = 200.00$$

displacement ductility demand is calculated as  $\phi / y$

- Calculation of  $\phi / y$  for END B -

for rotation axis 2 and integ. section (b)

From analysis, chord rotation  $\theta = 8.5673666E-005$

$$y = (M_y * L_s / 3) / E_{eff} = 0.00432734 \text{ ((4.29), Biskinis Phd)}$$

$$M_y = 5.2276E+008$$

$$L_s = M/V \text{ (with } L_s > 0.1 * L \text{ and } L_s < 2 * L) = 619.6389$$

From table 10.5, ASCE 41\_17:  $E_{eff} = \text{factor} * E_c * I_g = 2.4951E+013$

$$\text{factor} = 0.70$$

$$A_g = 120000.00$$

$$\text{Mean concrete strength: } f_c' = (f_c'_{\text{jacket}} * \text{Area}_{\text{jacket}} + f_c'_{\text{core}} * \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 25.00$$

$$N = 1.8003E+006$$

$$E_c * I_g = E_c * I_{g,\text{jacket}} + E_c * I_{g,\text{core}} = 3.5645E+013$$

Calculation of Yielding Moment  $M_y$

Calculation of  $\phi / y$  and  $M_y$  according to Annex 7 -

Assuming neutral axis out of flange ( $y > t/d$ , compression zone NOT rectangular)  
extended expressions (2.1)-(2.7) of Biskinis-Fardis (2009) are adopted with:

$$\text{flange width, } b = 400.00$$

$$\text{web width, } bw = 200.00$$

$$\text{flange thickness, } t = 200.00$$

$$y = \text{Min}(y_{\text{ten}}, y_{\text{com}})$$

$$y_{\text{ten}} = 1.6044767E-005$$

$$\text{with } f_y = 596.2441$$

$$d = 367.00$$

$$y = 0.49371549$$

$$A = 0.18421443$$

$$B = 0.10663695$$

$$\text{with } p_t = 0.01504455$$

$$p_c = 0.00774698$$

$$p_v = 0.01367492$$

$$N = 1.8003E+006$$

$$b = 400.00$$

$$\lambda = 0.08991826$$

$$y_{\text{comp}} = 1.0454056E-005$$

$$\text{with } f_c = 30.00$$

$$E_c = 25742.96$$

$$y = 0.54674444$$

$$A = 0.08461246$$

$$B = 0.06549972$$

with  $E_s = 200000.00$   
CONFIRMATION:  $y = 0.54674444 > t/d$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

End Of Calculation of Shear Capacity for element: column JLC1 of floor 1

At local axis: 3

Integration Section: (b)

## Calculation No. 8

column C1, Floor 1

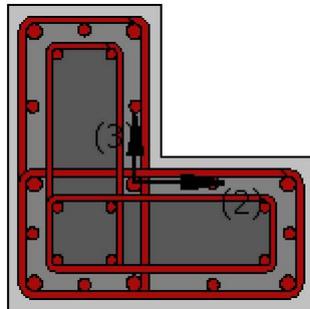
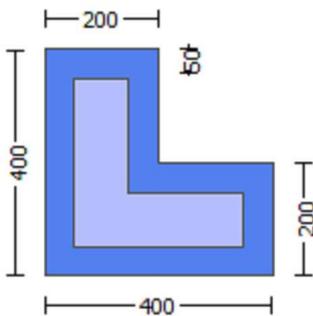
Limit State: Operational Level (data interpolation between analysis steps 50 and 51)

Analysis: Uniform +X

Check: Chord rotation capacity ( $\theta$ )

Edge: End

Local Axis: (3)



Start Of Calculation of Shear Capacity ratio for element: column JLC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rcjcs

Constant Properties

Knowledge Factor,  $\gamma = 0.90$

Mean strength values are used for both shear and moment calculations.

Consequently:

Jacket

New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 30.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 625.00$

Concrete Elasticity,  $E_c = 25742.96$

Steel Elasticity,  $E_s = 200000.00$

Existing Column

Existing material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 18.00$   
 Existing material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 500.00$   
 Concrete Elasticity,  $E_c = 19940.411$   
 Steel Elasticity,  $E_s = 200000.00$   
 #####  
 Note: Especially for the calculation of moment strengths,  
 the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14  
 Jacket  
 New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 781.25$   
 Existing Column  
 Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 625.00$   
 #####  
 Max Height,  $H_{max} = 400.00$   
 Min Height,  $H_{min} = 200.00$   
 Max Width,  $W_{max} = 400.00$   
 Min Width,  $W_{min} = 200.00$   
 Jacket Thickness,  $t_j = 50.00$   
 Cover Thickness,  $c = 15.00$   
 Mean Confinement Factor overall section = 1.00  
 Element Length,  $L = 3000.00$   
 Secondary Member  
 Smooth Bars  
 Ductile Steel  
 Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Adequate Lap Length ( $l_o/l_{ou}, \min >= 1$ )  
 No FRP Wrapping

-----  
 Stepwise Properties  
 -----

At local axis: 3  
 EDGE -A-  
 Shear Force,  $V_a = 2.3476251E-006$   
 EDGE -B-  
 Shear Force,  $V_b = -2.3476251E-006$   
 BOTH EDGES  
 Axial Force,  $F = -1.8026E+006$   
 Longitudinal Reinforcement Area Distribution (in 2 divisions)  
 -Tension:  $A_{st} = 0.00$   
 -Compression:  $A_{sc} = 5353.274$   
 Longitudinal Reinforcement Area Distribution (in 3 divisions)  
 -Tension:  $A_{st,ten} = 2208.54$   
 -Compression:  $A_{sc,com} = 1137.257$   
 -Middle:  $A_{sc,mid} = 2007.478$

-----  
 Calculation of Shear Capacity ratio ,  $V_e/V_r = 1.44151$   
 Member Controlled by Shear ( $V_e/V_r > 1$ )  
 Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 488890.189$   
 with  
 $M_{pr1} = \text{Max}(M_{u1+} , M_{u1-}) = 7.3334E+008$   
 $M_{u1+} = 7.3334E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction  
 which is defined for the static loading combination  
 $M_{u1-} = 6.3768E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment  
 direction which is defined for the static loading combination  
 $M_{pr2} = \text{Max}(M_{u2+} , M_{u2-}) = 7.3334E+008$   
 $M_{u2+} = 7.3334E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction  
 which is defined for the the static loading combination  
 $M_{u2-} = 6.3768E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment  
 direction which is defined for the the static loading combination

-----  
 Calculation of  $M_{u1+}$   
 -----

Calculation of ultimate curvature  $\phi_u$  according to 4.1, Biskinis/Fardis 2013:

$$u = 4.2207094E-005$$

$$\mu = 7.3334E+008$$

with full section properties:

$$b = 400.00$$

$$d = 367.00$$

$$d' = 33.00$$

$$v = 0.40931126$$

$$N = 1.8026E+006$$

$$f_c = 30.00$$

$$c_o(5A.5, TBDY) = 0.002$$

$$\text{Final value of } \phi_u: \phi_u^* = \text{shear\_factor} * \text{Max}(\phi_u, \phi_c) = 0.00925707$$

The Shear\_factor is considered equal to 1 (pure moment strength)

$$\text{From (5.4b), TBDY: } \phi_u = 0.00925707$$

$$\phi_{ue} \text{ (5.4c)} = 0.0207149$$

$$\phi_{ase} \text{ ((5.4d), TBDY)} = (\phi_{ase1} * A_{ext} + \phi_{ase2} * A_{int}) / A_{sec} = 0.33836562$$

$$\phi_{ase1} = \text{Max}(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$\phi_{ase2} (\geq \phi_{ase1}) = \text{Max}(((A_{conf,max2} - A_{noConf2}) / A_{conf,max2}) * (A_{conf,min2} / A_{conf,max2}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$\phi_{psh,min} * F_{ywe} = \text{Min}(\phi_{psh,x} * F_{ywe}, \phi_{psh,y} * F_{ywe}) = 6.12205$$

Expression (5.4d) for  $\phi_{psh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

$$\phi_{psh,x} * F_{ywe} = \phi_{psh1} * F_{ywe1} + \phi_{ps2} * F_{ywe2} = 6.12205$$

$$\phi_{psh1} \text{ ((5.4d), TBDY)} = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$$

$$L_{stir1} \text{ (Length of stirrups along Y)} = 1040.00$$

$$A_{stir1} \text{ (stirrups area)} = 78.53982$$

$$\phi_{psh2} \text{ (5.4d)} = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$$

$$L_{stir2} \text{ (Length of stirrups along Y)} = 768.00$$

$$A_{stir2} \text{ (stirrups area)} = 50.26548$$

$$\phi_{psh,y} * F_{ywe} = \phi_{psh1} * F_{ywe1} + \phi_{ps2} * F_{ywe2} = 6.12205$$

$$\phi_{psh1} \text{ ((5.4d), TBDY)} = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$$

$$L_{stir1} \text{ (Length of stirrups along X)} = 1040.00$$

$$A_{stir1} \text{ (stirrups area)} = 78.53982$$

$$\phi_{psh2} \text{ ((5.4d), TBDY)} = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$$

$$L_{stir2} \text{ (Length of stirrups along X)} = 768.00$$

$$A_{stir2} \text{ (stirrups area)} = 50.26548$$

$$A_{sec} = 120000.00$$

$$s1 = 100.00$$

$$s2 = 250.00$$

$$f_{ywe1} = 781.25$$

$$fy_{we2} = 625.00$$

$$f_{ce} = 30.00$$

From (5A.5), TBDY, TBDY:  $cc = 0.002$

$$c = \text{confinement factor} = 1.00$$

$$y_1 = 0.0025$$

$$sh_1 = 0.008$$

$$ft_1 = 898.293$$

$$fy_1 = 748.5775$$

$$su_1 = 0.032$$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

$$lo/lou, \min = lb/ld = 1.00$$

$$su_1 = 0.4 * esu_1 \text{ nominal} ((5.5), \text{TBDY}) = 0.032$$

From table 5A.1, TBDY:  $esu_1 \text{ nominal} = 0.08$ ,

For calculation of  $esu_1 \text{ nominal}$  and  $y_1, sh_1, ft_1, fy_1$ , it is considered  
characteristic value  $fsy_1 = fs_1/1.2$ , from table 5.1, TBDY.

$y_1, sh_1, ft_1, fy_1$ , are also multiplied by  $\text{Min}(1, 1.25 * (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

$$\text{with } fs_1 = (fs_{\text{jacket}} * A_{sl, \text{ten, jacket}} + fs_{\text{core}} * A_{sl, \text{ten, core}}) / A_{sl, \text{ten}} = 748.5775$$

$$\text{with } Es_1 = (Es_{\text{jacket}} * A_{sl, \text{ten, jacket}} + Es_{\text{core}} * A_{sl, \text{ten, core}}) / A_{sl, \text{ten}} = 200000.00$$

$$y_2 = 0.0025$$

$$sh_2 = 0.008$$

$$ft_2 = 886.7403$$

$$fy_2 = 738.9503$$

$$su_2 = 0.032$$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

$$lo/lou, \min = lb/lb, \min = 1.00$$

$$su_2 = 0.4 * esu_2 \text{ nominal} ((5.5), \text{TBDY}) = 0.032$$

From table 5A.1, TBDY:  $esu_2 \text{ nominal} = 0.08$ ,

For calculation of  $esu_2 \text{ nominal}$  and  $y_2, sh_2, ft_2, fy_2$ , it is considered  
characteristic value  $fsy_2 = fs_2/1.2$ , from table 5.1, TBDY.

$y_1, sh_1, ft_1, fy_1$ , are also multiplied by  $\text{Min}(1, 1.25 * (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

$$\text{with } fs_2 = (fs_{\text{jacket}} * A_{sl, \text{com, jacket}} + fs_{\text{core}} * A_{sl, \text{com, core}}) / A_{sl, \text{com}} = 738.9503$$

$$\text{with } Es_2 = (Es_{\text{jacket}} * A_{sl, \text{com, jacket}} + Es_{\text{core}} * A_{sl, \text{com, core}}) / A_{sl, \text{com}} = 200000.00$$

$$y_v = 0.0025$$

$$sh_v = 0.008$$

$$ft_v = 894.3662$$

$$fy_v = 745.3052$$

$$su_v = 0.032$$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

$$lo/lou, \min = lb/ld = 1.00$$

$$su_v = 0.4 * esu_v \text{ nominal} ((5.5), \text{TBDY}) = 0.032$$

From table 5A.1, TBDY:  $esu_v \text{ nominal} = 0.08$ ,

considering characteristic value  $fsy_v = fs_v/1.2$ , from table 5.1, TBDY  
For calculation of  $esu_v \text{ nominal}$  and  $y_v, sh_v, ft_v, fy_v$ , it is considered  
characteristic value  $fsy_v = fs_v/1.2$ , from table 5.1, TBDY.

$y_1, sh_1, ft_1, fy_1$ , are also multiplied by  $\text{Min}(1, 1.25 * (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

$$\text{with } fs_v = (fs_{\text{jacket}} * A_{sl, \text{mid, jacket}} + fs_{\text{mid}} * A_{sl, \text{mid, core}}) / A_{sl, \text{mid}} = 745.3052$$

$$\text{with } Es_v = (Es_{\text{jacket}} * A_{sl, \text{mid, jacket}} + Es_{\text{mid}} * A_{sl, \text{mid, core}}) / A_{sl, \text{mid}} = 200000.00$$

$$1 = A_{sl, \text{ten}} / (b * d) * (fs_1 / fc) = 0.37540035$$

$$2 = A_{sl, \text{com}} / (b * d) * (fs_2 / fc) = 0.19082108$$

$$v = A_{sl, \text{mid}} / (b * d) * (fs_v / fc) = 0.33973286$$

and confined core properties:

$$b = 360.00$$

$$d = 347.00$$

$$d' = 13.00$$

$$f_{cc} (5A.2, \text{TBDY}) = 30.00$$

$$cc (5A.5, \text{TBDY}) = 0.002$$

$$c = \text{confinement factor} = 1.00$$

$$1 = A_{sl, \text{ten}} / (b * d) * (fs_1 / fc) = 0.44115251$$

$$2 = A_{sl, \text{com}} / (b * d) * (fs_2 / fc) = 0.22424379$$

$$v = A_{sl, \text{mid}} / (b * d) * (fs_v / fc) = 0.39923778$$

Case/Assumption: Unconfined full section - Steel rupture

```

' satisfies Eq. (4.3)
---->
v < vs,y2 - LHS eq.(4.5) is not satisfied
---->
v < vs,c - RHS eq.(4.5) is not satisfied
---->
Case/Assumption Rejected.
---->
New Case/Assumption: Unconfined full section - Spalling of concrete cover
' satisfies Eq. (4.4)
---->
v < vs,y1 - LHS eq.(4.7) is not satisfied
---->
v < vc,y1 - RHS eq.(4.6) is satisfied
---->
cu (4.10) = 0.59761571
MRc (4.17) = 7.3334E+008
---->
New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover
In expressions below, the following modifications have been made
- b, d, d' replaced by geometric parameters of the core: bo, do, d'o
- N1, N2, v normalised to bo*do, instead of b*d
- parameters of confined concrete, fcc, cc, used in lieu of fc, ecu
---->
Subcase: Rupture of tension steel
---->
v* < v*s,y2 - LHS eq.(4.5) is not satisfied
---->
v* < v*s,c - LHS eq.(4.5) is not satisfied
---->
Subcase rejected
---->
New Subcase: Failure of compression zone
---->
v* < v*c,y2 - LHS eq.(4.6) is not satisfied
---->
v* < v*c,y1 - RHS eq.(4.6) is not satisfied
---->
*cu (4.11) = 0.62879536
MRo (4.18) = 4.5866E+008
MRo < 0.8*MRc
---->
u = cu (unconfined full section) = 4.2207094E-005
Mu = MRc
-----

Calculation of ratio lb/d
-----
Adequate Lap Length: lb/d >= 1
-----
-----

Calculation of Mu1-
-----
-----

Calculation of ultimate curvature u according to 4.1, Biskinis/Fardis 2013:
u = 4.8346386E-005
Mu = 6.3768E+008
-----

with full section properties:
b = 200.00
d = 367.00
d' = 33.00
v = 0.81862252
N = 1.8026E+006
fc = 30.00

```

$$c_o \text{ (5A.5, TBDY)} = 0.002$$

$$\text{Final value of } c_u^* = \text{shear\_factor} * \text{Max}(c_u, c_c) = 0.00925707$$

The Shear\_factor is considered equal to 1 (pure moment strength)

$$\text{From (5.4b), TBDY: } c_u = 0.00925707$$

$$w_e \text{ (5.4c)} = 0.0207149$$

$$a_{se} \text{ ((5.4d), TBDY)} = (a_{se1} * A_{ext} + a_{se2} * A_{int}) / A_{sec} = 0.33836562$$

$$a_{se1} = \text{Max}(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length

equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$a_{se2} (>= a_{se1}) = \text{Max}(((A_{conf,max2} - A_{noConf2}) / A_{conf,max2}) * (A_{conf,min2} / A_{conf,max2}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length

equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe}, p_{sh,y} * F_{ywe}) = 6.12205$$

Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

$$p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$$

$$p_{sh1} \text{ ((5.4d), TBDY)} = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$$

$$L_{stir1} \text{ (Length of stirrups along Y)} = 1040.00$$

$$A_{stir1} \text{ (stirrups area)} = 78.53982$$

$$p_{sh2} \text{ (5.4d)} = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$$

$$L_{stir2} \text{ (Length of stirrups along Y)} = 768.00$$

$$A_{stir2} \text{ (stirrups area)} = 50.26548$$

$$p_{sh,y} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$$

$$p_{sh1} \text{ ((5.4d), TBDY)} = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$$

$$L_{stir1} \text{ (Length of stirrups along X)} = 1040.00$$

$$A_{stir1} \text{ (stirrups area)} = 78.53982$$

$$p_{sh2} \text{ ((5.4d), TBDY)} = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$$

$$L_{stir2} \text{ (Length of stirrups along X)} = 768.00$$

$$A_{stir2} \text{ (stirrups area)} = 50.26548$$

$$A_{sec} = 120000.00$$

$$s1 = 100.00$$

$$s2 = 250.00$$

$$f_{ywe1} = 781.25$$

$$f_{ywe2} = 625.00$$

$$f_{ce} = 30.00$$

$$\text{From ((5.A5), TBDY), TBDY: } c_c = 0.002$$

$$c = \text{confinement factor} = 1.00$$

$$y1 = 0.0025$$

$$sh1 = 0.008$$

$$ft1 = 886.7403$$

$$fy1 = 738.9503$$

$$su1 = 0.032$$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor

and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

$$l_o / l_{ou,min} = l_b / l_d = 1.00$$

$su1 = 0.4 * esu1\_nominal ((5.5), TBDY) = 0.032$   
 From table 5A.1, TBDY:  $esu1\_nominal = 0.08$ ,  
 For calculation of  $esu1\_nominal$  and  $y1, sh1, ft1, fy1$ , it is considered  
 characteristic value  $fsy1 = fs1/1.2$ , from table 5.1, TBDY.  
 $y1, sh1, ft1, fy1$ , are also multiplied by  $Min(1, 1.25 * (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.  
 with  $fs1 = (fs\_jacket * Asl,ten,jacket + fs\_core * Asl,ten,core) / Asl,ten = 738.9503$   
 with  $Es1 = (Es\_jacket * Asl,ten,jacket + Es\_core * Asl,ten,core) / Asl,ten = 200000.00$   
 $y2 = 0.0025$   
 $sh2 = 0.008$   
 $ft2 = 898.293$   
 $fy2 = 748.5775$   
 $su2 = 0.032$   
 using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
 and also multiplied by the shear\_factor according to 15.7.1.4, with  
 Shear\_factor = 1.00  
 $lo/lou,min = lb/lb,min = 1.00$   
 $su2 = 0.4 * esu2\_nominal ((5.5), TBDY) = 0.032$   
 From table 5A.1, TBDY:  $esu2\_nominal = 0.08$ ,  
 For calculation of  $esu2\_nominal$  and  $y2, sh2, ft2, fy2$ , it is considered  
 characteristic value  $fsy2 = fs2/1.2$ , from table 5.1, TBDY.  
 $y1, sh1, ft1, fy1$ , are also multiplied by  $Min(1, 1.25 * (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.  
 with  $fs2 = (fs\_jacket * Asl,com,jacket + fs\_core * Asl,com,core) / Asl,com = 748.5775$   
 with  $Es2 = (Es\_jacket * Asl,com,jacket + Es\_core * Asl,com,core) / Asl,com = 200000.00$   
 $yv = 0.0025$   
 $shv = 0.008$   
 $ftv = 894.3662$   
 $fyv = 745.3052$   
 $suv = 0.032$   
 using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
 and also multiplied by the shear\_factor according to 15.7.1.4, with  
 Shear\_factor = 1.00  
 $lo/lou,min = lb/ld = 1.00$   
 $suv = 0.4 * esuv\_nominal ((5.5), TBDY) = 0.032$   
 From table 5A.1, TBDY:  $esuv\_nominal = 0.08$ ,  
 considering characteristic value  $fsyv = fsv/1.2$ , from table 5.1, TBDY  
 For calculation of  $esuv\_nominal$  and  $yv, shv, ftv, fyv$ , it is considered  
 characteristic value  $fsyv = fsv/1.2$ , from table 5.1, TBDY.  
 $y1, sh1, ft1, fy1$ , are also multiplied by  $Min(1, 1.25 * (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.  
 with  $fsv = (fs\_jacket * Asl,mid,jacket + fs\_mid * Asl,mid,core) / Asl,mid = 745.3052$   
 with  $Esv = (Es\_jacket * Asl,mid,jacket + Es\_mid * Asl,mid,core) / Asl,mid = 200000.00$   
 $1 = Asl,ten / (b * d) * (fs1 / fc) = 0.38164216$   
 $2 = Asl,com / (b * d) * (fs2 / fc) = 0.7508007$   
 $v = Asl,mid / (b * d) * (fsv / fc) = 0.67946571$   
 and confined core properties:  
 $b = 160.00$   
 $d = 347.00$   
 $d' = 13.00$   
 $fcc (5A.2, TBDY) = 30.00$   
 $cc (5A.5, TBDY) = 0.002$   
 $c = \text{confinement factor} = 1.00$   
 $1 = Asl,ten / (b * d) * (fs1 / fc) = 0.50454853$   
 $2 = Asl,com / (b * d) * (fs2 / fc) = 0.99259314$   
 $v = Asl,mid / (b * d) * (fsv / fc) = 0.898285$   
 Case/Assumption: Unconfined full section - Steel rupture  
 ' satisfies Eq. (4.3)  
 --->  
 $v < vs,y2$  - LHS eq.(4.5) is not satisfied  
 --->  
 $v < vs,c$  - RHS eq.(4.5) is not satisfied  
 --->  
 Case/Assumption Rejected.  
 --->  
 New Case/Assumption: Unconfined full section - Spalling of concrete cover  
 ' satisfies Eq. (4.4)  
 --->  
 $v < s,y1$  - LHS eq.(4.7) is not satisfied  
 --->

v < v<sub>c,y1</sub> - RHS eq.(4.6) is satisfied

---->

cu (4.10) = 0.52172715

M<sub>Rc</sub> (4.17) = 6.3768E+008

---->

New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover

In expressions below, the following modifications have been made

- b, d, d' replaced by geometric parameters of the core: b<sub>o</sub>, d<sub>o</sub>, d'<sub>o</sub>
- N<sub>1</sub>, N<sub>2</sub>, v normalised to b<sub>o</sub>\*d<sub>o</sub>, instead of b\*d
- parameters of confined concrete, f<sub>cc</sub>, c<sub>cc</sub>, used in lieu of f<sub>c</sub>, e<sub>cu</sub>

---->

Subcase: Rupture of tension steel

---->

v\* < v\*s<sub>y2</sub> - LHS eq.(4.5) is not satisfied

---->

v\* < v\*s<sub>c</sub> - LHS eq.(4.5) is not satisfied

---->

Subcase rejected

---->

New Subcase: Failure of compression zone

---->

v\* < v\*c<sub>y2</sub> - LHS eq.(4.6) is not satisfied

---->

v\* < v\*c<sub>y1</sub> - RHS eq.(4.6) is not satisfied

---->

\*cu (4.11) = 0.5604845

M<sub>Ro</sub> (4.18) = 4.7738E+008

M<sub>Ro</sub> < 0.8\*M<sub>Rc</sub>

---->

u = cu (unconfined full section) = 4.8346386E-005

Mu = M<sub>Rc</sub>

-----  
Calculation of ratio lb/d

-----  
Adequate Lap Length: lb/d >= 1  
-----  
-----

-----  
Calculation of Mu2+  
-----  
-----

-----  
Calculation of ultimate curvature u according to 4.1, Biskinis/Fardis 2013:

u = 4.2207094E-005

Mu = 7.3334E+008  
-----

with full section properties:

b = 400.00

d = 367.00

d' = 33.00

v = 0.40931126

N = 1.8026E+006

f<sub>c</sub> = 30.00

c<sub>o</sub> (5A.5, TBDY) = 0.002

Final value of cu: cu\* = shear\_factor \* Max( cu, cc) = 0.00925707

The Shear\_factor is considered equal to 1 (pure moment strength)

From (5.4b), TBDY: cu = 0.00925707

w<sub>e</sub> (5.4c) = 0.0207149

a<sub>se</sub> ((5.4d), TBDY) = (a<sub>se1</sub>\*A<sub>ext</sub>+a<sub>se2</sub>\*A<sub>int</sub>)/A<sub>sec</sub> = 0.33836562

a<sub>se1</sub> = Max(((A<sub>conf,max1</sub>-A<sub>noConf1</sub>)/A<sub>conf,max1</sub>)\*(A<sub>conf,min1</sub>/A<sub>conf,max1</sub>),0) = 0.33836562

The definitions of A<sub>noConf</sub>, A<sub>conf,min</sub> and A<sub>conf,max</sub> are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

A<sub>conf,max1</sub> = 89600.00 is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).  
 $ase_2 (>= ase_1) = \text{Max}(((A_{conf,max2} - A_{noConf2})/A_{conf,max2}) * (A_{conf,min2}/A_{conf,max2}), 0) = 0.33836562$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe}, p_{sh,y} * F_{ywe}) = 6.12205$

Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

-----  
 $p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$

$p_{sh1} ((5.4d), \text{TBDY}) = L_{stir1} * A_{stir1} / (A_{sec} * s_1) = 0.00680678$

$L_{stir1}$  (Length of stirrups along Y) = 1040.00

$A_{stir1}$  (stirrups area) = 78.53982

$p_{sh2} ((5.4d)) = L_{stir2} * A_{stir2} / (A_{sec} * s_2) = 0.0012868$

$L_{stir2}$  (Length of stirrups along Y) = 768.00

$A_{stir2}$  (stirrups area) = 50.26548

-----  
 $p_{sh,y} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$

$p_{sh1} ((5.4d), \text{TBDY}) = L_{stir1} * A_{stir1} / (A_{sec} * s_1) = 0.00680678$

$L_{stir1}$  (Length of stirrups along X) = 1040.00

$A_{stir1}$  (stirrups area) = 78.53982

$p_{sh2} ((5.4d), \text{TBDY}) = L_{stir2} * A_{stir2} / (A_{sec} * s_2) = 0.0012868$

$L_{stir2}$  (Length of stirrups along X) = 768.00

$A_{stir2}$  (stirrups area) = 50.26548

-----  
 $A_{sec} = 120000.00$

$s_1 = 100.00$

$s_2 = 250.00$

$f_{ywe1} = 781.25$

$f_{ywe2} = 625.00$

$f_{ce} = 30.00$

From ((5.A5), TBDY), TBDY:  $cc = 0.002$

$c = \text{confinement factor} = 1.00$

$y_1 = 0.0025$

$sh_1 = 0.008$

$ft_1 = 898.293$

$fy_1 = 748.5775$

$su_1 = 0.032$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor

and also multiplied by the shear\_factor according to 15.7.1.4, with

Shear\_factor = 1.00

$l_0/l_{ou,min} = l_b/l_d = 1.00$

$su_1 = 0.4 * esu_{1,nominal} ((5.5), \text{TBDY}) = 0.032$

From table 5A.1, TBDY:  $esu_{1,nominal} = 0.08$ ,

For calculation of  $esu_{1,nominal}$  and  $y_1, sh_1, ft_1, fy_1$ , it is considered characteristic value  $fsy_1 = fs_1/1.2$ , from table 5.1, TBDY.

$y_1, sh_1, ft_1, fy_1$ , are also multiplied by  $\text{Min}(1, 1.25 * (l_b/l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $fs_1 = (f_{s,jacket} * A_{sl,ten,jacket} + f_{s,core} * A_{sl,ten,core}) / A_{sl,ten} = 748.5775$

with  $Es_1 = (E_{s,jacket} * A_{sl,ten,jacket} + E_{s,core} * A_{sl,ten,core}) / A_{sl,ten} = 200000.00$

$y_2 = 0.0025$

$sh_2 = 0.008$

$ft_2 = 886.7403$

$fy_2 = 738.9503$

$su_2 = 0.032$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor

and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00  
 $l_o/l_{o,min} = l_b/l_{b,min} = 1.00$   
 $s_u2 = 0.4 * e_{su2\_nominal} ((5.5), TBDY) = 0.032$   
From table 5A.1, TBDY:  $e_{su2\_nominal} = 0.08$ ,  
For calculation of  $e_{su2\_nominal}$  and  $y_2, sh_2, ft_2, fy_2$ , it is considered  
characteristic value  $f_{sy2} = f_{s2}/1.2$ , from table 5.1, TBDY.  
 $y_1, sh_1, ft_1, fy_1$ , are also multiplied by  $\text{Min}(1, 1.25 * (l_b/l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.  
with  $f_{s2} = (f_{s,jacket} * A_{s1,com,jacket} + f_{s,core} * A_{s1,com,core}) / A_{s1,com} = 738.9503$   
with  $E_{s2} = (E_{s,jacket} * A_{s1,com,jacket} + E_{s,core} * A_{s1,com,core}) / A_{s1,com} = 200000.00$   
 $y_v = 0.0025$   
 $sh_v = 0.008$   
 $ft_v = 894.3662$   
 $fy_v = 745.3052$   
 $s_{uv} = 0.032$   
using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00  
 $l_o/l_{o,min} = l_b/l_d = 1.00$   
 $s_{uv} = 0.4 * e_{suv\_nominal} ((5.5), TBDY) = 0.032$   
From table 5A.1, TBDY:  $e_{suv\_nominal} = 0.08$ ,  
considering characteristic value  $f_{sv} = f_{sv}/1.2$ , from table 5.1, TBDY  
For calculation of  $e_{suv\_nominal}$  and  $y_v, sh_v, ft_v, fy_v$ , it is considered  
characteristic value  $f_{sv} = f_{sv}/1.2$ , from table 5.1, TBDY.  
 $y_1, sh_1, ft_1, fy_1$ , are also multiplied by  $\text{Min}(1, 1.25 * (l_b/l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.  
with  $f_{sv} = (f_{s,jacket} * A_{s1,mid,jacket} + f_{s,mid} * A_{s1,mid,core}) / A_{s1,mid} = 745.3052$   
with  $E_{sv} = (E_{s,jacket} * A_{s1,mid,jacket} + E_{s,mid} * A_{s1,mid,core}) / A_{s1,mid} = 200000.00$   
 $1 = A_{s1,ten} / (b * d) * (f_{s1} / f_c) = 0.37540035$   
 $2 = A_{s1,com} / (b * d) * (f_{s2} / f_c) = 0.19082108$   
 $v = A_{s1,mid} / (b * d) * (f_{sv} / f_c) = 0.33973286$   
and confined core properties:  
 $b = 360.00$   
 $d = 347.00$   
 $d' = 13.00$   
 $f_{cc} (5A.2, TBDY) = 30.00$   
 $cc (5A.5, TBDY) = 0.002$   
 $c = \text{confinement factor} = 1.00$   
 $1 = A_{s1,ten} / (b * d) * (f_{s1} / f_c) = 0.44115251$   
 $2 = A_{s1,com} / (b * d) * (f_{s2} / f_c) = 0.22424379$   
 $v = A_{s1,mid} / (b * d) * (f_{sv} / f_c) = 0.39923778$   
Case/Assumption: Unconfined full section - Steel rupture  
' satisfies Eq. (4.3)  
---->  
 $v < v_{s,y2}$  - LHS eq.(4.5) is not satisfied  
---->  
 $v < v_{s,c}$  - RHS eq.(4.5) is not satisfied  
---->  
Case/Assumption Rejected.  
---->  
New Case/Assumption: Unconfined full section - Spalling of concrete cover  
' satisfies Eq. (4.4)  
---->  
 $v < s_{y1}$  - LHS eq.(4.7) is not satisfied  
---->  
 $v < v_{c,y1}$  - RHS eq.(4.6) is satisfied  
---->  
 $c_u (4.10) = 0.59761571$   
 $M_{Rc} (4.17) = 7.3334E+008$   
---->  
New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover  
In expressions below, the following modifications have been made  
-  $b, d, d'$  replaced by geometric parameters of the core:  $b_o, d_o, d'_o$   
-  $N, 1, 2, v$  normalised to  $b_o * d_o$ , instead of  $b * d$   
- parameters of confined concrete,  $f_{cc}, cc$ , used in lieu of  $f_c, e_{cu}$   
---->  
Subcase: Rupture of tension steel  
---->

$v^* < v^*_{s,y2}$  - LHS eq.(4.5) is not satisfied

--->

$v^* < v^*_{s,c}$  - LHS eq.(4.5) is not satisfied

--->

Subcase rejected

--->

New Subcase: Failure of compression zone

--->

$v^* < v^*_{c,y2}$  - LHS eq.(4.6) is not satisfied

--->

$v^* < v^*_{c,y1}$  - RHS eq.(4.6) is not satisfied

--->

\*cu (4.11) = 0.62879536

MRO (4.18) = 4.5866E+008

MRO < 0.8\*MRc

--->

u = cu (unconfined full section) = 4.2207094E-005

Mu = MRc

-----  
Calculation of ratio lb/d

Adequate Lap Length: lb/d >= 1

-----  
Calculation of Mu2-

-----  
Calculation of ultimate curvature u according to 4.1, Biskinis/Fardis 2013:

u = 4.8346386E-005

Mu = 6.3768E+008

-----  
with full section properties:

b = 200.00

d = 367.00

d' = 33.00

v = 0.81862252

N = 1.8026E+006

fc = 30.00

co (5A.5, TBDY) = 0.002

Final value of cu: cu\* = shear\_factor \* Max( cu, cc) = 0.00925707

The Shear\_factor is considered equal to 1 (pure moment strength)

From (5.4b), TBDY: cu = 0.00925707

we (5.4c) = 0.0207149

ase ((5.4d), TBDY) = (ase1\*Aext+ase2\*Aint)/Asec = 0.33836562

ase1 = Max(((Aconf,max1-AnoConf1)/Aconf,max1)\*(Aconf,min1/Aconf,max1),0) = 0.33836562

The definitions of AnoConf, Aconf,min and Aconf,max are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

Aconf,max1 = 89600.00 is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

Aconf,min1 = 59225.00 is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area Aconf,max1 by a length

equal to half the clear spacing between external hoops.

AnoConf1 = 43733.333 is the unconfined external core area which is equal to bi<sup>2</sup>/6 as defined at (A.2).

ase2 (>=ase1) = Max(((Aconf,max2-AnoConf2)/Aconf,max2)\*(Aconf,min2/Aconf,max2),0) = 0.33836562

The definitions of AnoConf, Aconf,min and Aconf,max are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

Aconf,max2 = 45264.00 is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

Aconf,min2 = 0.00 is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area Aconf,max2 by a length equal to half the clear spacing between internal hoops.

AnoConf2 = 32309.333 is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).  
 $psh,min*Fywe = \text{Min}(psh,x*Fywe, psh,y*Fywe) = 6.12205$

Expression (5.4d) for  $psh,min*Fywe$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

-----  
 $psh_x*Fywe = psh1*Fywe1 + ps2*Fywe2 = 6.12205$   
 $psh1 \text{ ((5.4d), TBDY)} = Lstir1*Astir1/(Asec*s1) = 0.00680678$   
Lstir1 (Length of stirrups along Y) = 1040.00  
Astir1 (stirrups area) = 78.53982  
 $psh2 \text{ (5.4d)} = Lstir2*Astir2/(Asec*s2) = 0.0012868$   
Lstir2 (Length of stirrups along Y) = 768.00  
Astir2 (stirrups area) = 50.26548

-----  
 $psh_y*Fywe = psh1*Fywe1 + ps2*Fywe2 = 6.12205$   
 $psh1 \text{ ((5.4d), TBDY)} = Lstir1*Astir1/(Asec*s1) = 0.00680678$   
Lstir1 (Length of stirrups along X) = 1040.00  
Astir1 (stirrups area) = 78.53982  
 $psh2 \text{ ((5.4d), TBDY)} = Lstir2*Astir2/(Asec*s2) = 0.0012868$   
Lstir2 (Length of stirrups along X) = 768.00  
Astir2 (stirrups area) = 50.26548

-----  
Asec = 120000.00  
s1 = 100.00  
s2 = 250.00  
fywe1 = 781.25  
fywe2 = 625.00  
fce = 30.00

From ((5.A5), TBDY), TBDY: cc = 0.002  
c = confinement factor = 1.00

y1 = 0.0025  
sh1 = 0.008  
ft1 = 886.7403  
fy1 = 738.9503  
su1 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

lo/lou,min = lb/ld = 1.00

su1 = 0.4\*esu1\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1,ft1,fy1, it is considered characteristic value  $fsy1 = fs1/1.2$ , from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25*(lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $fs1 = (fs,jacket*Asl,ten,jacket + fs,core*Asl,ten,core)/Asl,ten = 738.9503$

with  $Es1 = (Es,jacket*Asl,ten,jacket + Es,core*Asl,ten,core)/Asl,ten = 200000.00$

y2 = 0.0025  
sh2 = 0.008  
ft2 = 898.293  
fy2 = 748.5775  
su2 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

lo/lou,min = lb/lb,min = 1.00

su2 = 0.4\*esu2\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu2\_nominal = 0.08,

For calculation of esu2\_nominal and y2, sh2,ft2,fy2, it is considered characteristic value  $fsy2 = fs2/1.2$ , from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25*(lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $fs2 = (fs,jacket*Asl,com,jacket + fs,core*Asl,com,core)/Asl,com = 748.5775$

with  $Es2 = (Es,jacket*Asl,com,jacket + Es,core*Asl,com,core)/Asl,com = 200000.00$

yv = 0.0025  
shv = 0.008  
ftv = 894.3662

$f_{yv} = 745.3052$   
 $s_{uv} = 0.032$   
 using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
 and also multiplied by the shear\_factor according to 15.7.1.4, with  
 Shear\_factor = 1.00  
 $l_o/l_{ou,min} = l_b/l_d = 1.00$   
 $s_{uv} = 0.4 * e_{suv\_nominal} ((5,5), TBDY) = 0.032$   
 From table 5A.1, TBDY:  $e_{suv\_nominal} = 0.08$ ,  
 considering characteristic value  $f_{sv} = f_{sv}/1.2$ , from table 5.1, TBDY  
 For calculation of  $e_{suv\_nominal}$  and  $\gamma_v$ ,  $sh_v, ft_v, f_{yv}$ , it is considered  
 characteristic value  $f_{sv} = f_{sv}/1.2$ , from table 5.1, TBDY.  
 $\gamma_1, sh_1, ft_1, f_{y1}$ , are also multiplied by  $\text{Min}(1, 1.25 * (l_b/l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.  
 with  $f_{sv} = (f_{s,jacket} * A_{s,mid,jacket} + f_{s,mid} * A_{s,mid,core}) / A_{s,mid} = 745.3052$   
 with  $E_{sv} = (E_{s,jacket} * A_{s,mid,jacket} + E_{s,mid} * A_{s,mid,core}) / A_{s,mid} = 200000.00$   
 $1 = A_{s,ten} / (b * d) * (f_{s1} / f_c) = 0.38164216$   
 $2 = A_{s,com} / (b * d) * (f_{s2} / f_c) = 0.7508007$   
 $v = A_{s,mid} / (b * d) * (f_{sv} / f_c) = 0.67946571$   
 and confined core properties:  
 $b = 160.00$   
 $d = 347.00$   
 $d' = 13.00$   
 $f_{cc} (5A.2, TBDY) = 30.00$   
 $cc (5A.5, TBDY) = 0.002$   
 $c = \text{confinement factor} = 1.00$   
 $1 = A_{s,ten} / (b * d) * (f_{s1} / f_c) = 0.50454853$   
 $2 = A_{s,com} / (b * d) * (f_{s2} / f_c) = 0.99259314$   
 $v = A_{s,mid} / (b * d) * (f_{sv} / f_c) = 0.898285$   
 Case/Assumption: Unconfined full section - Steel rupture  
 ' satisfies Eq. (4.3)  
 --->  
 $v < v_{s,y2}$  - LHS eq.(4.5) is not satisfied  
 --->  
 $v < v_{s,c}$  - RHS eq.(4.5) is not satisfied  
 --->  
 Case/Assumption Rejected.  
 --->  
 New Case/Assumption: Unconfined full section - Spalling of concrete cover  
 ' satisfies Eq. (4.4)  
 --->  
 $v < s_{y1}$  - LHS eq.(4.7) is not satisfied  
 --->  
 $v < v_{c,y1}$  - RHS eq.(4.6) is satisfied  
 --->  
 $cu (4.10) = 0.52172715$   
 $MRC (4.17) = 6.3768E+008$   
 --->  
 New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover  
 In expressions below, the following modifications have been made  
 -  $b, d, d'$  replaced by geometric parameters of the core:  $b_o, d_o, d'_o$   
 -  $N, 1, 2, v$  normalised to  $b_o * d_o$ , instead of  $b * d$   
 -  $f_{cc}, cc$  parameters of confined concrete,  $f_{cc}, cc$ , used in lieu of  $f_c, e_{cu}$   
 --->  
 Subcase: Rupture of tension steel  
 --->  
 $v^* < v^*_{s,y2}$  - LHS eq.(4.5) is not satisfied  
 --->  
 $v^* < v^*_{s,c}$  - LHS eq.(4.5) is not satisfied  
 --->  
 Subcase rejected  
 --->  
 New Subcase: Failure of compression zone  
 --->  
 $v^* < v^*_{c,y2}$  - LHS eq.(4.6) is not satisfied  
 --->  
 $v^* < v^*_{c,y1}$  - RHS eq.(4.6) is not satisfied  
 --->  
 $*cu (4.11) = 0.5604845$

$$M_{Ro} (4.18) = 4.7738E+008$$

$$M_{Ro} < 0.8 * M_{Rc}$$

--->

$$u = c_u \text{ (unconfined full section)} = 4.8346386E-005$$

$$M_u = M_{Rc}$$

-----  
Calculation of ratio  $l_b/d$

-----  
Adequate Lap Length:  $l_b/d \geq 1$

-----  
Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 339150.739$

-----  
Calculation of Shear Strength at edge 1,  $V_{r1} = 339150.739$

$$V_{r1} = V_{Col} \text{ ((10.3), ASCE 41-17)} = k_{nl} * V_{Col0}$$

$$V_{Col0} = 339150.739$$

$k_{nl} = 1$  (zero step-static loading)

-----  
NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ '  
where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

-----  
= 1 (normal-weight concrete)

Mean concrete strength:  $f'_c = (f'_c \text{_{jacket}} * \text{Area}_{\text{jacket}} + f'_c \text{_{core}} * \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 25.00$ , but  $f'_c^{0.5} \leq 8.3$   
MPa (22.5.3.1, ACI 318-14)

$$M/Vd = 4.00$$

$$M_u = 2.0345E+006$$

$$V_u = 2.3476251E-006$$

$$d = 0.8 * h = 320.00$$

$$N_u = 1.8026E+006$$

$$A_g = 80000.00$$

$$\text{From (11.5.4.8), ACI 318-14: } V_s = V_{s,jacket} + V_{s,core} = 471238.898$$

where:

$$V_{s,jacket} = V_{s,j1} + V_{s,j2} = 471238.898$$

$V_{s,j1} = 314159.265$  is calculated for section web jacket, with:

$$d = 320.00$$

$$A_v = 157079.633$$

$$f_y = 625.00$$

$$s = 100.00$$

$V_{s,j1}$  is multiplied by  $Col_{j1} = 1.00$

$$s/d = 0.3125$$

$V_{s,j2} = 157079.633$  is calculated for section flange jacket, with:

$$d = 160.00$$

$$A_v = 157079.633$$

$$f_y = 625.00$$

$$s = 100.00$$

$V_{s,j2}$  is multiplied by  $Col_{j2} = 1.00$

$$s/d = 0.625$$

$$V_{s,core} = V_{s,c1} + V_{s,c2} = 0.00$$

$V_{s,c1} = 0.00$  is calculated for section web core, with:

$$d = 240.00$$

$$A_v = 100530.965$$

$$f_y = 500.00$$

$$s = 250.00$$

$V_{s,c1}$  is multiplied by  $Col_{c1} = 0.00$

$$s/d = 1.04167$$

$V_{s,c2} = 0.00$  is calculated for section flange core, with:

$$d = 80.00$$

$$A_v = 100530.965$$

$$f_y = 500.00$$

$$s = 250.00$$

$V_{s,c2}$  is multiplied by  $Col_{c2} = 0.00$

$$s/d = 3.125$$

$$V_f \text{ ((11-3)-(11.4), ACI 440)} = 0.00$$

$$\text{From (11-11), ACI 440: } V_s + V_f \leq 212577.225$$

bw = 200.00

Calculation of Shear Strength at edge 2, Vr2 = 339150.739

Vr2 = VCol ((10.3), ASCE 41-17) = knl\*VCol0

VCol0 = 339150.739

knl = 1 (zero step-static loading)

NOTE: In expression (10-3) 'Vs = Av\*fy\*d/s' is replaced by 'Vs+ f\*Vf'  
where Vf is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

Mean concrete strength: fc' = (fc'\_jacket\*Area\_jacket + fc'\_core\*Area\_core)/Area\_section = 25.00, but fc'^0.5 <= 8.3 MPa (22.5.3.1, ACI 318-14)

M/Vd = 4.00

Mu = 6739.685

Vu = 2.3476251E+006

d = 0.8\*h = 320.00

Nu = 1.8026E+006

Ag = 80000.00

From (11.5.4.8), ACI 318-14: Vs = Vs\_jacket + Vs\_core = 471238.898

where:

Vs\_jacket = Vs\_j1 + Vs\_j2 = 471238.898

Vs\_j1 = 314159.265 is calculated for section web jacket, with:

d = 320.00

Av = 157079.633

fy = 625.00

s = 100.00

Vs\_j1 is multiplied by Col\_j1 = 1.00

s/d = 0.3125

Vs\_j2 = 157079.633 is calculated for section flange jacket, with:

d = 160.00

Av = 157079.633

fy = 625.00

s = 100.00

Vs\_j2 is multiplied by Col\_j2 = 1.00

s/d = 0.625

Vs\_core = Vs\_c1 + Vs\_c2 = 0.00

Vs\_c1 = 0.00 is calculated for section web core, with:

d = 240.00

Av = 100530.965

fy = 500.00

s = 250.00

Vs\_c1 is multiplied by Col\_c1 = 0.00

s/d = 1.04167

Vs\_c2 = 0.00 is calculated for section flange core, with:

d = 80.00

Av = 100530.965

fy = 500.00

s = 250.00

Vs\_c2 is multiplied by Col\_c2 = 0.00

s/d = 3.125

Vf ((11-3)-(11.4), ACI 440) = 0.00

From (11-11), ACI 440: Vs + Vf <= 212577.225

bw = 200.00

End Of Calculation of Shear Capacity ratio for element: column JLC1 of floor 1

At local axis: 3

Start Of Calculation of Shear Capacity ratio for element: column JLC1 of floor 1

At Shear local axis: 2

(Bending local axis: 3)

Section Type: rcjcs

Constant Properties

Knowledge Factor,  $\gamma = 0.90$

Mean strength values are used for both shear and moment calculations.

Consequently:

Jacket

New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 30.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 625.00$

Concrete Elasticity,  $E_c = 25742.96$

Steel Elasticity,  $E_s = 200000.00$

Existing Column

Existing material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 18.00$

Existing material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 500.00$

Concrete Elasticity,  $E_c = 19940.411$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Jacket

New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 781.25$

Existing Column

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 625.00$

#####

Max Height,  $H_{max} = 400.00$

Min Height,  $H_{min} = 200.00$

Max Width,  $W_{max} = 400.00$

Min Width,  $W_{min} = 200.00$

Jacket Thickness,  $t_j = 50.00$

Cover Thickness,  $c = 15.00$

Mean Confinement Factor overall section = 1.00

Element Length,  $L = 3000.00$

Secondary Member

Smooth Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou, min} > 1$ )

No FRP Wrapping

Stepwise Properties

At local axis: 2

EDGE -A-

Shear Force,  $V_a = -3.6347847E-006$

EDGE -B-

Shear Force,  $V_b = 3.6347847E-006$

BOTH EDGES

Axial Force,  $F = -1.8026E+006$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{st} = 0.00$

-Compression:  $A_{sc} = 5353.274$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{st, ten} = 1137.257$

-Compression:  $A_{sl, com} = 2208.54$

-Middle:  $A_{sl, mid} = 2007.478$

Calculation of Shear Capacity ratio,  $V_e/V_r = 1.18753$

Member Controlled by Shear ( $V_e/V_r > 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 402750.026$

with

$M_{pr1} = \text{Max}(M_{u1+}, M_{u1-}) = 6.0413E+008$

$M_{u1+} = 6.0413E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction

which is defined for the static loading combination

$Mu_{1-} = 5.4612E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$Mpr_2 = \text{Max}(Mu_{2+}, Mu_{2-}) = 6.0413E+008$

$Mu_{2+} = 6.0413E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

$Mu_{2-} = 5.4612E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

-----  
Calculation of  $Mu_{1+}$   
-----

-----  
Calculation of ultimate curvature  $\phi_u$  according to 4.1, Biskinis/Fardis 2013:

$\phi_u = 1.2613948E-005$

$Mu = 6.0413E+008$   
-----

with full section properties:

$b = 400.00$

$d = 367.00$

$d' = 33.00$

$v = 0.40931126$

$N = 1.8026E+006$

$f_c = 30.00$

$\alpha_{co} (5A.5, TBDY) = 0.002$

Final value of  $\phi_{cu}$ :  $\phi_{cu}^* = \text{shear\_factor} * \text{Max}(\phi_{cu}, \phi_{cc}) = 0.00925707$

The Shear\_factor is considered equal to 1 (pure moment strength)

From (5.4b), TBDY:  $\phi_{cu} = 0.00925707$

we (5.4c) = 0.0207149

$\alpha_{ase} ((5.4d), TBDY) = (\alpha_{ase1} * A_{ext} + \alpha_{ase2} * A_{int}) / A_{sec} = 0.33836562$

$\alpha_{ase1} = \text{Max}(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).

$\alpha_{ase2} (> \alpha_{ase1}) = \text{Max}(((A_{conf,max2} - A_{noConf2}) / A_{conf,max2}) * (A_{conf,min2} / A_{conf,max2}), 0) = 0.33836562$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe}, p_{sh,y} * F_{ywe}) = 6.12205$

Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

-----  
 $p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$

$p_{sh1} ((5.4d), TBDY) = L_{stir1} * A_{stir1} / (A_{sec} * s_1) = 0.00680678$

$L_{stir1}$  (Length of stirrups along Y) = 1040.00

$A_{stir1}$  (stirrups area) = 78.53982

$p_{sh2} (5.4d) = L_{stir2} * A_{stir2} / (A_{sec} * s_2) = 0.0012868$

$L_{stir2}$  (Length of stirrups along Y) = 768.00

$A_{stir2}$  (stirrups area) = 50.26548  
-----

psh\_y\*Fywe = psh1\*Fywe1+ps2\*Fywe2 = 6.12205  
psh1 ((5.4d), TBDY) = Lstir1\*Astir1/(Asec\*s1) = 0.00680678  
Lstir1 (Length of stirrups along X) = 1040.00  
Astir1 (stirrups area) = 78.53982  
psh2 ((5.4d), TBDY) = Lstir2\*Astir2/(Asec\*s2) = 0.0012868  
Lstir2 (Length of stirrups along X) = 768.00  
Astir2 (stirrups area) = 50.26548

Asec = 120000.00

s1 = 100.00

s2 = 250.00

fywe1 = 781.25

fywe2 = 625.00

fce = 30.00

From ((5.A5), TBDY), TBDY: cc = 0.002

c = confinement factor = 1.00

y1 = 0.0025

sh1 = 0.008

ft1 = 886.7403

fy1 = 738.9503

su1 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/d = 1.00

su1 = 0.4\*esu1\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1,ft1,fy1, it is considered  
characteristic value fsy1 = fs1/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by Min(1,1.25\*(lb/d)^2/3), from 10.3.5, ASCE 41-17.

with fs1 = (fs,jacket\*Asl,ten,jacket + fs,core\*Asl,ten,core)/Asl,ten = 738.9503

with Es1 = (Es,jacket\*Asl,ten,jacket + Es,core\*Asl,ten,core)/Asl,ten = 200000.00

y2 = 0.0025

sh2 = 0.008

ft2 = 898.293

fy2 = 748.5775

su2 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb,min = 1.00

su2 = 0.4\*esu2\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu2\_nominal = 0.08,

For calculation of esu2\_nominal and y2, sh2,ft2,fy2, it is considered  
characteristic value fsy2 = fs2/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by Min(1,1.25\*(lb/d)^2/3), from 10.3.5, ASCE 41-17.

with fs2 = (fs,jacket\*Asl,com,jacket + fs,core\*Asl,com,core)/Asl,com = 748.5775

with Es2 = (Es,jacket\*Asl,com,jacket + Es,core\*Asl,com,core)/Asl,com = 200000.00

yv = 0.0025

shv = 0.008

ftv = 894.3662

fyv = 745.3052

suv = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/d = 1.00

suv = 0.4\*esuv\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esuv\_nominal = 0.08,

considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY

For calculation of esuv\_nominal and yv, shv,ftv,fyv, it is considered  
characteristic value fsyv = fsv/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by Min(1,1.25\*(lb/d)^2/3), from 10.3.5, ASCE 41-17.

with fsv = (fs,jacket\*Asl,mid,jacket + fs,mid\*Asl,mid,core)/Asl,mid = 745.3052

with Esv = (Es,jacket\*Asl,mid,jacket + Es,mid\*Asl,mid,core)/Asl,mid = 200000.00

1 = Asl,ten/(b\*d)\*(fs1/fc) = 0.19082108

2 = Asl,com/(b\*d)\*(fs2/fc) = 0.37540035

$v = A_{sl,mid}/(b*d)*(f_{sv}/f_c) = 0.33973286$   
 and confined core properties:  
 $b = 360.00$   
 $d = 347.00$   
 $d' = 13.00$   
 $f_{cc} (5A.2, TBDY) = 30.00$   
 $cc (5A.5, TBDY) = 0.002$   
 $c = \text{confinement factor} = 1.00$   
 $1 = A_{sl,ten}/(b*d)*(f_{s1}/f_c) = 0.22424379$   
 $2 = A_{sl,com}/(b*d)*(f_{s2}/f_c) = 0.44115251$   
 $v = A_{sl,mid}/(b*d)*(f_{sv}/f_c) = 0.39923778$   
 Case/Assumption: Unconfined full section - Steel rupture  
 ' satisfies Eq. (4.3)  
 --->  
 $v < v_{s,y2}$  - LHS eq.(4.5) is not satisfied  
 --->  
 $v < v_{s,c}$  - RHS eq.(4.5) is not satisfied  
 --->  
 Case/Assumption Rejected.  
 --->  
 New Case/Assumption: Unconfined full section - Spalling of concrete cover  
 ' satisfies Eq. (4.4)  
 --->  
 $v < s_{y1}$  - LHS eq.(4.7) is not satisfied  
 --->  
 $v < v_{c,y1}$  - RHS eq.(4.6) is satisfied  
 --->  
 $c_u (4.10) = 0.37716748$   
 $M_{Rc} (4.17) = 7.0585E+008$   
 --->  
 New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover  
 In expressions below, the following modifications have been made  
 -  $b, d, d'$  replaced by geometric parameters of the core:  $b_o, d_o, d'_o$   
 -  $N, 1, 2, v$  normalised to  $b_o*d_o$ , instead of  $b*d$   
 -  $f_{cc}, cc$  parameters of confined concrete,  $f_{cc}, cc$ , used in lieu of  $f_c, e_{cu}$   
 --->  
 Subcase: Rupture of tension steel  
 --->  
 $v^* < v^*_{s,y2}$  - LHS eq.(4.5) is not satisfied  
 --->  
 $v^* < v^*_{s,c}$  - LHS eq.(4.5) is not satisfied  
 --->  
 Subcase rejected  
 --->  
 New Subcase: Failure of compression zone  
 --->  
 $v^* < v^*_{c,y2}$  - LHS eq.(4.6) is not satisfied  
 --->  
 $v^* < v^*_{c,y1}$  - RHS eq.(4.6) is not satisfied  
 --->  
 $*c_u (4.11) = 0.4569298$   
 $M_{Ro} (4.18) = 6.0413E+008$   
 --->  
 $u = c_u (4.2) = 1.2613948E-005$   
 $M_u = M_{Ro}$

-----  
 Calculation of ratio  $l_b/l_d$   
 -----

Adequate Lap Length:  $l_b/l_d \geq 1$   
 -----  
 -----  
 -----

Calculation of  $M_{u1}$ -  
 -----  
 -----  
 -----

Calculation of ultimate curvature  $\phi_u$  according to 4.1, Biskinis/Fardis 2013:

$$u = 3.1475172E-005$$

$$\mu = 5.4612E+008$$

with full section properties:

$$b = 200.00$$

$$d = 367.00$$

$$d' = 33.00$$

$$v = 0.81862252$$

$$N = 1.8026E+006$$

$$f_c = 30.00$$

$$\omega (5A.5, TBDY) = 0.002$$

$$\text{Final value of } \phi_u: \phi_u^* = \text{shear\_factor} * \text{Max}(\phi_u, \phi_c) = 0.00925707$$

The Shear\_factor is considered equal to 1 (pure moment strength)

$$\text{From (5.4b), TBDY: } \phi_u = 0.00925707$$

$$\omega_e (5.4c) = 0.0207149$$

$$\text{ase ((5.4d), TBDY) = (ase1*A_{ext} + ase2*A_{int})/A_{sec} = 0.33836562$$

$$\text{ase1} = \text{Max}(((A_{conf,max1} - A_{noConf1})/A_{conf,max1}) * (A_{conf,min1}/A_{conf,max1}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$\text{ase2} (>= \text{ase1}) = \text{Max}(((A_{conf,max2} - A_{noConf2})/A_{conf,max2}) * (A_{conf,min2}/A_{conf,max2}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe}, p_{sh,y} * F_{ywe}) = 6.12205$$

Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

$$p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{s2} * F_{ywe2} = 6.12205$$

$$p_{sh1} ((5.4d), TBDY) = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$$

$$L_{stir1} (\text{Length of stirrups along Y}) = 1040.00$$

$$A_{stir1} (\text{stirrups area}) = 78.53982$$

$$p_{sh2} (5.4d) = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$$

$$L_{stir2} (\text{Length of stirrups along Y}) = 768.00$$

$$A_{stir2} (\text{stirrups area}) = 50.26548$$

$$p_{sh,y} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{s2} * F_{ywe2} = 6.12205$$

$$p_{sh1} ((5.4d), TBDY) = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$$

$$L_{stir1} (\text{Length of stirrups along X}) = 1040.00$$

$$A_{stir1} (\text{stirrups area}) = 78.53982$$

$$p_{sh2} ((5.4d), TBDY) = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$$

$$L_{stir2} (\text{Length of stirrups along X}) = 768.00$$

$$A_{stir2} (\text{stirrups area}) = 50.26548$$

$$A_{sec} = 120000.00$$

$$s1 = 100.00$$

$$s2 = 250.00$$

$$f_{ywe1} = 781.25$$

$$f_{ywe2} = 625.00$$

$$f_{ce} = 30.00$$

From ((5.A.5), TBDY), TBDY:  $cc = 0.002$   
 $c = \text{confinement factor} = 1.00$   
 $y1 = 0.0025$   
 $sh1 = 0.008$   
 $ft1 = 898.293$   
 $fy1 = 748.5775$   
 $su1 = 0.032$   
 using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
 and also multiplied by the shear\_factor according to 15.7.1.4, with  
 $\text{Shear\_factor} = 1.00$   
 $lo/lou,min = lb/d = 1.00$   
 $su1 = 0.4*esu1\_nominal ((5.5), TBDY) = 0.032$   
 From table 5A.1, TBDY:  $esu1\_nominal = 0.08$ ,  
 For calculation of  $esu1\_nominal$  and  $y1, sh1, ft1, fy1$ , it is considered  
 characteristic value  $fsy1 = fs1/1.2$ , from table 5.1, TBDY.  
 $y1, sh1, ft1, fy1$ , are also multiplied by  $\text{Min}(1, 1.25*(lb/d)^{2/3})$ , from 10.3.5, ASCE 41-17.  
 with  $fs1 = (fs,jacket*Asl,ten,jacket + fs,core*Asl,ten,core)/Asl,ten = 748.5775$   
 with  $Es1 = (Es,jacket*Asl,ten,jacket + Es,core*Asl,ten,core)/Asl,ten = 200000.00$   
 $y2 = 0.0025$   
 $sh2 = 0.008$   
 $ft2 = 886.7403$   
 $fy2 = 738.9503$   
 $su2 = 0.032$   
 using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
 and also multiplied by the shear\_factor according to 15.7.1.4, with  
 $\text{Shear\_factor} = 1.00$   
 $lo/lou,min = lb/lb,min = 1.00$   
 $su2 = 0.4*esu2\_nominal ((5.5), TBDY) = 0.032$   
 From table 5A.1, TBDY:  $esu2\_nominal = 0.08$ ,  
 For calculation of  $esu2\_nominal$  and  $y2, sh2, ft2, fy2$ , it is considered  
 characteristic value  $fsy2 = fs2/1.2$ , from table 5.1, TBDY.  
 $y1, sh1, ft1, fy1$ , are also multiplied by  $\text{Min}(1, 1.25*(lb/d)^{2/3})$ , from 10.3.5, ASCE 41-17.  
 with  $fs2 = (fs,jacket*Asl,com,jacket + fs,core*Asl,com,core)/Asl,com = 738.9503$   
 with  $Es2 = (Es,jacket*Asl,com,jacket + Es,core*Asl,com,core)/Asl,com = 200000.00$   
 $yv = 0.0025$   
 $shv = 0.008$   
 $ftv = 894.3662$   
 $fyv = 745.3052$   
 $suv = 0.032$   
 using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
 and also multiplied by the shear\_factor according to 15.7.1.4, with  
 $\text{Shear\_factor} = 1.00$   
 $lo/lou,min = lb/d = 1.00$   
 $suv = 0.4*esuv\_nominal ((5.5), TBDY) = 0.032$   
 From table 5A.1, TBDY:  $esuv\_nominal = 0.08$ ,  
 considering characteristic value  $fsyv = fsv/1.2$ , from table 5.1, TBDY  
 For calculation of  $esuv\_nominal$  and  $yv, shv, ftv, fyv$ , it is considered  
 characteristic value  $fsyv = fsv/1.2$ , from table 5.1, TBDY.  
 $y1, sh1, ft1, fy1$ , are also multiplied by  $\text{Min}(1, 1.25*(lb/d)^{2/3})$ , from 10.3.5, ASCE 41-17.  
 with  $fsv = (fs,jacket*Asl,mid,jacket + fs,mid*Asl,mid,core)/Asl,mid = 745.3052$   
 with  $Es_v = (Es,jacket*Asl,mid,jacket + Es,mid*Asl,mid,core)/Asl,mid = 200000.00$   
 $1 = Asl,ten/(b*d)*(fs1/fc) = 0.7508007$   
 $2 = Asl,com/(b*d)*(fs2/fc) = 0.38164216$   
 $v = Asl,mid/(b*d)*(fsv/fc) = 0.67946571$   
 and confined core properties:  
 $b = 160.00$   
 $d = 347.00$   
 $d' = 13.00$   
 $fcc (5A.2, TBDY) = 30.00$   
 $cc (5A.5, TBDY) = 0.002$   
 $c = \text{confinement factor} = 1.00$   
 $1 = Asl,ten/(b*d)*(fs1/fc) = 0.99259314$   
 $2 = Asl,com/(b*d)*(fs2/fc) = 0.50454853$   
 $v = Asl,mid/(b*d)*(fsv/fc) = 0.898285$

Case/Assumption: Unconfined full section - Steel rupture  
 ' satisfies Eq. (4.3)

--->

v < vs,y2 - LHS eq.(4.5) is not satisfied

---->

v < vs,c - RHS eq.(4.5) is not satisfied

---->

Case/Assumption Rejected.

---->

New Case/Assumption: Unconfined full section - Spalling of concrete cover

' satisfies Eq. (4.4)

---->

v < vs,y1 - LHS eq.(4.7) is not satisfied

---->

v < vc,y1 - RHS eq.(4.6) is not satisfied

---->

cu (4.11) = 0.80138156

MRC (4.18) = 5.4612E+008

---->

New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover

In expressions below, the following modifications have been made

- b, d, d' replaced by geometric parameters of the core: bo, do, d'o
- N, 1, 2, v normalised to bo\*do, instead of b\*d
- - parameters of confined concrete, fcc, cc, used in lieu of fc, ecu

---->

Subcase: Rupture of tension steel

---->

v\* < v\*s,y2 - LHS eq.(4.5) is not satisfied

---->

v\* < v\*s,c - LHS eq.(4.5) is not satisfied

---->

Subcase rejected

---->

New Subcase: Failure of compression zone

---->

v\* < v\*c,y2 - LHS eq.(4.6) is not satisfied

---->

v\* < v\*c,y1 - RHS eq.(4.6) is not satisfied

---->

\*cu (4.11) = 0.80674817

MRO (4.18) = 2.8580E+008

MRO < 0.8\*MRC

---->

u = cu (unconfined full section) = 3.1475172E-005

Mu = MRC

-----  
Calculation of ratio lb/d

-----  
Adequate Lap Length: lb/d >= 1

-----  
Calculation of Mu2+

-----  
Calculation of ultimate curvature u according to 4.1, Biskinis/Fardis 2013:

u = 1.2613948E-005

Mu = 6.0413E+008

-----  
with full section properties:

b = 400.00

d = 367.00

d' = 33.00

v = 0.40931126

N = 1.8026E+006

fc = 30.00

co (5A.5, TBDY) = 0.002

Final value of cu: cu\* = shear\_factor \* Max( cu, cc) = 0.00925707

The Shear\_factor is considered equal to 1 (pure moment strength)

From (5.4b), TBDY:  $c_u = 0.00925707$

$w_e$  (5.4c) = 0.0207149

$a_{se}$  ((5.4d), TBDY) =  $(a_{se1} * A_{ext} + a_{se2} * A_{int}) / A_{sec} = 0.33836562$

$a_{se1} = \text{Max}(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).

$a_{se2} (>= a_{se1}) = \text{Max}(((A_{conf,max2} - A_{noConf2}) / A_{conf,max2}) * (A_{conf,min2} / A_{conf,max2}), 0) = 0.33836562$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe}, p_{sh,y} * F_{ywe}) = 6.12205$

Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

$p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$

$p_{sh1}$  ((5.4d), TBDY) =  $L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$

$L_{stir1}$  (Length of stirrups along Y) = 1040.00

$A_{stir1}$  (stirrups area) = 78.53982

$p_{sh2}$  (5.4d) =  $L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$

$L_{stir2}$  (Length of stirrups along Y) = 768.00

$A_{stir2}$  (stirrups area) = 50.26548

$p_{sh,y} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$

$p_{sh1}$  ((5.4d), TBDY) =  $L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$

$L_{stir1}$  (Length of stirrups along X) = 1040.00

$A_{stir1}$  (stirrups area) = 78.53982

$p_{sh2}$  ((5.4d), TBDY) =  $L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$

$L_{stir2}$  (Length of stirrups along X) = 768.00

$A_{stir2}$  (stirrups area) = 50.26548

$A_{sec} = 120000.00$

$s1 = 100.00$

$s2 = 250.00$

$f_{ywe1} = 781.25$

$f_{ywe2} = 625.00$

$f_{ce} = 30.00$

From ((5.A5), TBDY), TBDY:  $c_c = 0.002$

$c$  = confinement factor = 1.00

$y1 = 0.0025$

$sh1 = 0.008$

$ft1 = 886.7403$

$fy1 = 738.9503$

$su1 = 0.032$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

$l_o/l_{ou,min} = l_b/d = 1.00$

$su1 = 0.4 * e_{su1,nominal}$  ((5.5), TBDY) = 0.032

From table 5A.1, TBDY:  $e_{su1,nominal} = 0.08$ ,

For calculation of  $es_{u1\_nominal}$  and  $y_1, sh_1, ft_1, fy_1$ , it is considered characteristic value  $fs_{y1} = fs_1/1.2$ , from table 5.1, TBDY.

$y_1, sh_1, ft_1, fy_1$ , are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $fs_1 = (fs_{jacket} \cdot A_{s,ten,jacket} + fs_{core} \cdot A_{s,ten,core}) / A_{s,ten} = 738.9503$

with  $Es_1 = (Es_{jacket} \cdot A_{s,ten,jacket} + Es_{core} \cdot A_{s,ten,core}) / A_{s,ten} = 200000.00$

$y_2 = 0.0025$

$sh_2 = 0.008$

$ft_2 = 898.293$

$fy_2 = 748.5775$

$su_2 = 0.032$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

$lo/lou, min = lb/lb, min = 1.00$

$su_2 = 0.4 \cdot es_{u2\_nominal} ((5.5), TBDY) = 0.032$

From table 5A.1, TBDY:  $es_{u2\_nominal} = 0.08$ ,

For calculation of  $es_{u2\_nominal}$  and  $y_2, sh_2, ft_2, fy_2$ , it is considered characteristic value  $fs_{y2} = fs_2/1.2$ , from table 5.1, TBDY.

$y_2, sh_2, ft_2, fy_2$ , are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $fs_2 = (fs_{jacket} \cdot A_{s,com,jacket} + fs_{core} \cdot A_{s,com,core}) / A_{s,com} = 748.5775$

with  $Es_2 = (Es_{jacket} \cdot A_{s,com,jacket} + Es_{core} \cdot A_{s,com,core}) / A_{s,com} = 200000.00$

$y_v = 0.0025$

$sh_v = 0.008$

$ft_v = 894.3662$

$fy_v = 745.3052$

$suv = 0.032$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

$lo/lou, min = lb/l_d = 1.00$

$suv = 0.4 \cdot es_{uv\_nominal} ((5.5), TBDY) = 0.032$

From table 5A.1, TBDY:  $es_{uv\_nominal} = 0.08$ ,

considering characteristic value  $fs_{yv} = f_{sv}/1.2$ , from table 5.1, TBDY

For calculation of  $es_{uv\_nominal}$  and  $y_v, sh_v, ft_v, fy_v$ , it is considered characteristic value  $fs_{yv} = f_{sv}/1.2$ , from table 5.1, TBDY.

$y_1, sh_1, ft_1, fy_1$ , are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $f_{sv} = (f_{s,jacket} \cdot A_{s,mid,jacket} + f_{s,mid} \cdot A_{s,mid,core}) / A_{s,mid} = 745.3052$

with  $Es_v = (Es_{jacket} \cdot A_{s,mid,jacket} + Es_{mid} \cdot A_{s,mid,core}) / A_{s,mid} = 200000.00$

$1 = A_{s,ten} / (b \cdot d) \cdot (fs_1 / f_c) = 0.19082108$

$2 = A_{s,com} / (b \cdot d) \cdot (fs_2 / f_c) = 0.37540035$

$v = A_{s,mid} / (b \cdot d) \cdot (f_{sv} / f_c) = 0.33973286$

and confined core properties:

$b = 360.00$

$d = 347.00$

$d' = 13.00$

$f_{cc} (5A.2, TBDY) = 30.00$

$cc (5A.5, TBDY) = 0.002$

$c = \text{confinement factor} = 1.00$

$1 = A_{s,ten} / (b \cdot d) \cdot (fs_1 / f_c) = 0.22424379$

$2 = A_{s,com} / (b \cdot d) \cdot (fs_2 / f_c) = 0.44115251$

$v = A_{s,mid} / (b \cdot d) \cdot (f_{sv} / f_c) = 0.39923778$

Case/Assumption: Unconfined full section - Steel rupture

' satisfies Eq. (4.3)

--->

$v < v_{s,y2}$  - LHS eq.(4.5) is not satisfied

--->

$v < v_{s,c}$  - RHS eq.(4.5) is not satisfied

--->

Case/Assumption Rejected.

--->

New Case/Assumption: Unconfined full section - Spalling of concrete cover

' satisfies Eq. (4.4)

--->

$v < s_{y1}$  - LHS eq.(4.7) is not satisfied

--->

$v < v_{c,y1}$  - RHS eq.(4.6) is satisfied

--->

cu (4.10) = 0.37716748  
MRc (4.17) = 7.0585E+008

--->  
New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover  
In expressions below, the following modifications have been made

- b, d, d' replaced by geometric parameters of the core: bo, do, d'o
- N, 1, 2, v normalised to bo\*do, instead of b\*d
- - parameters of confined concrete, fcc, cc, used in lieu of fc, ecu

--->  
Subcase: Rupture of tension steel

--->  
 $v^* < v^*s_y2$  - LHS eq.(4.5) is not satisfied

--->  
 $v^* < v^*s_c$  - LHS eq.(4.5) is not satisfied

--->  
Subcase rejected

--->  
New Subcase: Failure of compression zone

--->  
 $v^* < v^*c_y2$  - LHS eq.(4.6) is not satisfied

--->  
 $v^* < v^*c_y1$  - RHS eq.(4.6) is not satisfied

--->  
\*cu (4.11) = 0.4569298  
MRo (4.18) = 6.0413E+008

--->  
u = cu (4.2) = 1.2613948E-005  
Mu = MRo

-----  
Calculation of ratio lb/d

-----  
Adequate Lap Length: lb/d >= 1

-----  
Calculation of Mu2-

-----  
Calculation of ultimate curvature u according to 4.1, Biskinis/Fardis 2013:

u = 3.1475172E-005  
Mu = 5.4612E+008

-----  
with full section properties:

b = 200.00  
d = 367.00  
d' = 33.00  
v = 0.81862252  
N = 1.8026E+006

fc = 30.00  
co (5A.5, TBDY) = 0.002  
Final value of cu:  $cu^* = \text{shear\_factor} * \text{Max}(cu, cc) = 0.00925707$

The Shear\_factor is considered equal to 1 (pure moment strength)

From (5.4b), TBDY:  $cu = 0.00925707$

we (5.4c) = 0.0207149

ase ((5.4d), TBDY) =  $(ase1*A_{ext} + ase2*A_{int})/A_{sec} = 0.33836562$

ase1 =  $\text{Max}(((A_{conf,max1} - A_{noConf1})/A_{conf,max1}) * (A_{conf,min1}/A_{conf,max1}), 0) = 0.33836562$

The definitions of A<sub>noConf</sub>, A<sub>conf,min</sub> and A<sub>conf,max</sub> are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

A<sub>conf,max1</sub> = 89600.00 is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

A<sub>conf,min1</sub> = 59225.00 is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area A<sub>conf,max1</sub> by a length

equal to half the clear spacing between external hoops.

AnoConf1 = 43733.333 is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).  
ase2 ( $\geq$  ase1) =  $\text{Max}(((A_{\text{conf,max2}} - \text{AnoConf2}) / A_{\text{conf,max2}}) * (A_{\text{conf,min2}} / A_{\text{conf,max2}}), 0) = 0.33836562$

The definitions of AnoConf, Aconf,min and Aconf,max are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

Aconf,max2 = 45264.00 is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

Aconf,min2 = 0.00 is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area Aconf,max2 by a length equal to half the clear spacing between internal hoops.

AnoConf2 = 32309.333 is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

psh,min\*Fywe =  $\text{Min}(psh_x * Fywe, psh_y * Fywe) = 6.12205$

Expression (5.4d) for psh,min\*Fywe has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

-----  
psh\_x\*Fywe = psh1\*Fywe1 + ps2\*Fywe2 = 6.12205  
psh1 ((5.4d), TBDY) =  $L_{\text{stir1}} * A_{\text{stir1}} / (A_{\text{sec}} * s1) = 0.00680678$   
Lstir1 (Length of stirrups along Y) = 1040.00  
Astir1 (stirrups area) = 78.53982  
psh2 (5.4d) =  $L_{\text{stir2}} * A_{\text{stir2}} / (A_{\text{sec}} * s2) = 0.0012868$   
Lstir2 (Length of stirrups along Y) = 768.00  
Astir2 (stirrups area) = 50.26548  
-----

-----  
psh\_y\*Fywe = psh1\*Fywe1 + ps2\*Fywe2 = 6.12205  
psh1 ((5.4d), TBDY) =  $L_{\text{stir1}} * A_{\text{stir1}} / (A_{\text{sec}} * s1) = 0.00680678$   
Lstir1 (Length of stirrups along X) = 1040.00  
Astir1 (stirrups area) = 78.53982  
psh2 ((5.4d), TBDY) =  $L_{\text{stir2}} * A_{\text{stir2}} / (A_{\text{sec}} * s2) = 0.0012868$   
Lstir2 (Length of stirrups along X) = 768.00  
Astir2 (stirrups area) = 50.26548  
-----

Asec = 120000.00

s1 = 100.00

s2 = 250.00

fywe1 = 781.25

fywe2 = 625.00

fce = 30.00

From ((5.A5), TBDY), TBDY: cc = 0.002

c = confinement factor = 1.00

y1 = 0.0025

sh1 = 0.008

ft1 = 898.293

fy1 = 748.5775

su1 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

lo/lou,min = lb/ld = 1.00

su1 =  $0.4 * esu1_{\text{nominal}} ((5.5), \text{TBDY}) = 0.032$

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1, ft1, fy1, it is considered characteristic value fsy1 = fs1/1.2, from table 5.1, TBDY.

y1, sh1, ft1, fy1, are also multiplied by  $\text{Min}(1, 1.25 * (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs1 =  $(f_{s,\text{jacket}} * A_{s1,\text{ten,jacket}} + f_{s,\text{core}} * A_{s1,\text{ten,core}}) / A_{s1,\text{ten}} = 748.5775$

with Es1 =  $(E_{s,\text{jacket}} * A_{s1,\text{ten,jacket}} + E_{s,\text{core}} * A_{s1,\text{ten,core}}) / A_{s1,\text{ten}} = 200000.00$

y2 = 0.0025

sh2 = 0.008

ft2 = 886.7403

fy2 = 738.9503

su2 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

lo/lou,min = lb/lb,min = 1.00

```

su2 = 0.4*esu2_nominal ((5.5), TBDY) = 0.032
From table 5A.1, TBDY: esu2_nominal = 0.08,
For calculation of esu2_nominal and y2, sh2,ft2,fy2, it is considered
characteristic value fsy2 = fs2/1.2, from table 5.1, TBDY.
y1, sh1,ft1,fy1, are also multiplied by Min(1,1.25*(lb/lc)^ 2/3), from 10.3.5, ASCE 41-17.
with fs2 = (fsjacket*Asl,com,jacket + fs,core*Asl,com,core)/Asl,com = 738.9503
with Es2 = (Es,jacket*Asl,com,jacket + Es,core*Asl,com,core)/Asl,com = 200000.00
yv = 0.0025
shv = 0.008
ftv = 894.3662
fyv = 745.3052
suv = 0.032
using (30) in Biskinis/Fardis (2013) multiplied with shear_factor
and also multiplied by the shear_factor according to 15.7.1.4, with
Shear_factor = 1.00
lo/lou,min = lb/lc = 1.00
suv = 0.4*esuv_nominal ((5.5), TBDY) = 0.032
From table 5A.1, TBDY: esuv_nominal = 0.08,
considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY
For calculation of esuv_nominal and yv, shv,ftv,fyv, it is considered
characteristic value fsyv = fsv/1.2, from table 5.1, TBDY.
y1, sh1,ft1,fy1, are also multiplied by Min(1,1.25*(lb/lc)^ 2/3), from 10.3.5, ASCE 41-17.
with fsv = (fsjacket*Asl,mid,jacket + fs,mid*Asl,mid,core)/Asl,mid = 745.3052
with Esv = (Es,jacket*Asl,mid,jacket + Es,mid*Asl,mid,core)/Asl,mid = 200000.00
1 = Asl,ten/(b*d)*(fs1/fc) = 0.7508007
2 = Asl,com/(b*d)*(fs2/fc) = 0.38164216
v = Asl,mid/(b*d)*(fsv/fc) = 0.67946571
and confined core properties:
b = 160.00
d = 347.00
d' = 13.00
fcc (5A.2, TBDY) = 30.00
cc (5A.5, TBDY) = 0.002
c = confinement factor = 1.00
1 = Asl,ten/(b*d)*(fs1/fc) = 0.99259314
2 = Asl,com/(b*d)*(fs2/fc) = 0.50454853
v = Asl,mid/(b*d)*(fsv/fc) = 0.898285
Case/Assumption: Unconfined full section - Steel rupture
'satisfies Eq. (4.3)
--->
v < vs,y2 - LHS eq.(4.5) is not satisfied
--->
v < vs,c - RHS eq.(4.5) is not satisfied
--->
Case/Assumption Rejected.
--->
New Case/Assumption: Unconfined full section - Spalling of concrete cover
'satisfies Eq. (4.4)
--->
v < sy1 - LHS eq.(4.7) is not satisfied
--->
v < vc,y1 - RHS eq.(4.6) is not satisfied
--->
cu (4.11) = 0.80138156
MRc (4.18) = 5.4612E+008
--->
New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover
In expressions below, the following modifications have been made
- b, d, d' replaced by geometric parameters of the core: bo, do, d'o
- N1, N2, v normalised to bo*do, instead of b*d
- parameters of confined concrete, fcc, cc, used in lieu of fc, ecu
--->
Subcase: Rupture of tension steel
--->
v* < v*s,y2 - LHS eq.(4.5) is not satisfied
--->
v* < v*s,c - LHS eq.(4.5) is not satisfied

```

```

---->
Subcase rejected
---->
New Subcase: Failure of compression zone
---->
v* < v*c,y2 - LHS eq.(4.6) is not satisfied
---->
v* < v*c,y1 - RHS eq.(4.6) is not satisfied
---->
*cu (4.11) = 0.80674817
MRo (4.18) = 2.8580E+008
MRo < 0.8*MRc
---->
u = cu (unconfined full section) = 3.1475172E-005
Mu = MRc
-----

Calculation of ratio lb/d
-----
Adequate Lap Length: lb/d >= 1
-----
-----
-----
Calculation of Shear Strength Vr = Min(Vr1,Vr2) = 339150.739
-----
Calculation of Shear Strength at edge 1, Vr1 = 339150.739
Vr1 = VCol ((10.3), ASCE 41-17) = knl*VCol0
VCol0 = 339150.739
knl = 1 (zero step-static loading)
-----
NOTE: In expression (10-3) 'Vs = Av*fy*d/s' is replaced by 'Vs+ f*Vf'
where Vf is the contribution of FRPs (11.3), ACI 440).
-----
= 1 (normal-weight concrete)
Mean concrete strength: fc' = (fc'_jacket*Area_jacket + fc'_core*Area_core)/Area_section = 25.00, but fc'^0.5 <= 8.3
MPa (22.5.3.1, ACI 318-14)
M/Vd = 4.00
Mu = 6.7415E+007
Vu = 3.6347847E-006
d = 0.8*h = 320.00
Nu = 1.8026E+006
Ag = 80000.00
From (11.5.4.8), ACI 318-14: Vs = Vs,jacket + Vs,core = 471238.898
where:
Vs,jacket = Vs,j1 + Vs,j2 = 471238.898
Vs,j1 = 157079.633 is calculated for section web jacket, with:
d = 160.00
Av = 157079.633
fy = 625.00
s = 100.00
Vs,j1 is multiplied by Col,j1 = 1.00
s/d = 0.625
Vs,j2 = 314159.265 is calculated for section flange jacket, with:
d = 320.00
Av = 157079.633
fy = 625.00
s = 100.00
Vs,j2 is multiplied by Col,j2 = 1.00
s/d = 0.3125
Vs,core = Vs,c1 + Vs,c2 = 0.00
Vs,c1 = 0.00 is calculated for section web core, with:
d = 80.00
Av = 100530.965
fy = 500.00
s = 250.00
Vs,c1 is multiplied by Col,c1 = 0.00

```

s/d = 3.125  
Vs,c2 = 0.00 is calculated for section flange core, with:  
d = 240.00  
Av = 100530.965  
fy = 500.00  
s = 250.00  
Vs,c2 is multiplied by Col,c2 = 0.00  
s/d = 1.04167  
Vf ((11-3)-(11.4), ACI 440) = 0.00  
From (11-11), ACI 440: Vs + Vf <= 212577.225  
bw = 200.00

Calculation of Shear Strength at edge 2, Vr2 = 339150.739  
Vr2 = VCol ((10.3), ASCE 41-17) = knl\*VCol0  
VCol0 = 339150.739  
knl = 1 (zero step-static loading)

NOTE: In expression (10-3) 'Vs = Av\*fy\*d/s' is replaced by 'Vs+ f\*Vf'  
where Vf is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)  
Mean concrete strength: fc' = (fc'\_jacket\*Area\_jacket + fc'\_core\*Area\_core)/Area\_section = 25.00, but fc'^0.5 <= 8.3  
MPa (22.5.3.1, ACI 318-14)

M/Vd = 4.00  
Mu = 5.3429E+007  
Vu = 3.6347847E-006  
d = 0.8\*h = 320.00  
Nu = 1.8026E+006  
Ag = 80000.00  
From (11.5.4.8), ACI 318-14: Vs = Vs\_jacket + Vs\_core = 471238.898

where:

Vs\_jacket = Vs\_j1 + Vs\_j2 = 471238.898  
Vs\_j1 = 157079.633 is calculated for section web jacket, with:  
d = 160.00  
Av = 157079.633  
fy = 625.00  
s = 100.00

Vs\_j1 is multiplied by Col,j1 = 1.00  
s/d = 0.625

Vs\_j2 = 314159.265 is calculated for section flange jacket, with:  
d = 320.00  
Av = 157079.633  
fy = 625.00  
s = 100.00

Vs\_j2 is multiplied by Col,j2 = 1.00  
s/d = 0.3125

Vs\_core = Vs,c1 + Vs,c2 = 0.00  
Vs,c1 = 0.00 is calculated for section web core, with:  
d = 80.00  
Av = 100530.965  
fy = 500.00  
s = 250.00

Vs,c1 is multiplied by Col,c1 = 0.00  
s/d = 3.125

Vs,c2 = 0.00 is calculated for section flange core, with:  
d = 240.00  
Av = 100530.965  
fy = 500.00  
s = 250.00

Vs,c2 is multiplied by Col,c2 = 0.00  
s/d = 1.04167

Vf ((11-3)-(11.4), ACI 440) = 0.00  
From (11-11), ACI 440: Vs + Vf <= 212577.225  
bw = 200.00

End Of Calculation of Shear Capacity ratio for element: column JLC1 of floor 1  
At local axis: 2

-----  
Start Of Calculation of Chord Rotation Capacity for element: column JLC1 of floor 1  
At local axis: 3  
Integration Section: (b)  
Section Type: rcjlcs

Constant Properties

-----  
Knowledge Factor,  $\gamma = 0.90$   
Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE 41-17.  
Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17  
Consequently:  
Jacket  
New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 30.00$   
New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 625.00$   
Concrete Elasticity,  $E_c = 25742.96$   
Steel Elasticity,  $E_s = 200000.00$   
Existing Column  
Existing material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 18.00$   
Existing material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 500.00$   
Concrete Elasticity,  $E_c = 19940.411$   
Steel Elasticity,  $E_s = 200000.00$   
Max Height,  $H_{max} = 400.00$   
Min Height,  $H_{min} = 200.00$   
Max Width,  $W_{max} = 400.00$   
Min Width,  $W_{min} = 200.00$   
Jacket Thickness,  $t_j = 50.00$   
Cover Thickness,  $c = 15.00$   
Element Length,  $L = 3000.00$   
Secondary Member  
Smooth Bars  
Ductile Steel  
Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Adequate Lap Length ( $l_b/l_d \geq 1$ )  
No FRP Wrapping

-----  
Stepwise Properties

-----  
Bending Moment,  $M = -7.2803E+007$   
Shear Force,  $V_2 = 97644.268$   
Shear Force,  $V_3 = 30.34672$   
Axial Force,  $F = -1.8003E+006$   
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $A_{sl,t} = 0.00$   
-Compression:  $A_{sl,c} = 5353.274$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $A_{sl,ten} = 1137.257$   
-Compression:  $A_{sl,com} = 2208.54$   
-Middle:  $A_{sl,mid} = 2007.478$   
Longitudinal External Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $A_{sl,ten,jacket} = 829.3805$   
-Compression:  $A_{sl,com,jacket} = 1746.726$   
-Middle:  $A_{sl,mid,jacket} = 1545.664$   
Longitudinal Internal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $A_{sl,ten,core} = 307.8761$   
-Compression:  $A_{sl,com,core} = 461.8141$   
-Middle:  $A_{sl,mid,core} = 461.8141$   
Mean Diameter of Tension Reinforcement,  $DbL = 16.80$   
-----  
-----

Existing component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_{R} = u = 0.00472303$   
 $u = y + p = 0.00524781$

-----  
- Calculation of  $y$  -  
-----

$y = (M_y * L_s / 3) / E_{eff} = 0.00524781$  ((4.29), Biskinis Phd)  
 $M_y = 5.2685E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 745.5987  
From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 2.4951E+013$   
 $factor = 0.70$   
 $A_g = 120000.00$   
Mean concrete strength:  $f_c' = (f_c'_{jacket} * Area_{jacket} + f_c'_{core} * Area_{core}) / Area_{section} = 25.00$   
 $N = 1.8003E+006$   
 $E_c * I_g = E_{c,jacket} * I_{g,jacket} + E_{c,core} * I_{g,core} = 3.5645E+013$   
-----

-----  
Calculation of Yielding Moment  $M_y$   
-----

Calculation of  $y$  and  $M_y$  according to Annex 7 -  
-----

Assuming neutral axis within flange ( $y < t/d$ , compression zone rectangular) with:  
flange width,  $b = 400.00$   
web width,  $b_w = 200.00$   
flange thickness,  $t = 200.00$   
-----

$y = \text{Min}(y_{ten}, y_{com})$   
 $y_{ten} = 1.4407218E-005$   
with  $f_y = 596.2441$   
 $d = 367.00$   
 $y = 0.43617034$   
 $A = 0.05703506$   
 $B = 0.03712065$   
with  $p_t = 0.00774698$   
 $p_c = 0.01504455$   
 $p_v = 0.01367492$   
 $N = 1.8003E+006$   
 $b = 400.00$   
 $" = 0.08991826$   
 $y_{comp} = 1.2588515E-005$   
with  $f_c = 30.00$   
 $E_c = 25742.96$   
 $y = 0.45404063$   
 $A = 0.00723408$   
 $B = 0.01655203$   
with  $E_s = 200000.00$   
CONFIRMATION:  $y = 0.45404063 < t/d$   
-----

Calculation of ratio  $l_b/d$   
-----

Adequate Lap Length:  $l_b/d \geq 1$   
-----

- Calculation of  $p$  -  
-----

From table 10-8:  $p = 0.00$   
with:  
- Columns not controlled by inadequate development or splicing along the clear height because  $l_b/d \geq 1$   
shear control ratio  $V_y E / C_o I_{OE} = 1.18753$   
 $d = d_{external} = 367.00$   
 $s = s_{external} = 100.00$   
-  $t = s_1 + s_2 + 2 * t_f / b_w * (f_{fe} / f_s) = 0.00809358$   
jacket:  $s_1 = A_{v1} * L_{stir1} / (s_1 * A_g) = 0.00680678$

$Av1 = 78.53982$ , is the area of every stirrup parallel to loading (shear) direction  
 $Lstir1 = 1040.00$ , is the total Length of all stirrups parallel to loading (shear) direction  
 $s1 = 100.00$

core:  $s2 = Av2 * Lstir2 / (s2 * Ag) = 0.0012868$

$Av2 = 50.26548$ , is the area of every stirrup parallel to loading (shear) direction  
 $Lstir2 = 768.00$ , is the total Length of all stirrups parallel to loading (shear) direction  
 $s2 = 250.00$

The term  $2 * tf / bw * (ffe / fs)$  is implemented to account for FRP contribution  
 where  $f = 2 * tf / bw$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $ffe / fs$  normalises  $f$  to steel strength  
 All these variables have already been given in Shear control ratio calculation.

For the normalisation  $fs$  of jacket is used.

$NUD = 1.8003E+006$

$Ag = 120000.00$

$f_{cE} = (f_{c\_jacket} * Area\_jacket + f_{c\_core} * Area\_core) / section\_area = 25.00$

$fyE = (fy\_ext\_Long\_Reinf * Area\_ext\_Long\_Reinf + fy\_int\_Long\_Reinf * Area\_int\_Long\_Reinf) / Area\_Tot\_Long\_Rein = 596.2441$

$fytE = (fy\_ext\_Trans\_Reinf * s1 + fy\_int\_Trans\_Reinf * s2) / (s1 + s2) = 605.1263$

$pl = Area\_Tot\_Long\_Rein / (b * d) = 0.03646644$

$b = 400.00$

$d = 367.00$

$f_{cE} = 25.00$

-----  
 End Of Calculation of Chord Rotation Capacity for element: column JLC1 of floor 1

At local axis: 3

Integration Section: (b)  
 -----

## Calculation No. 9

column C1, Floor 1

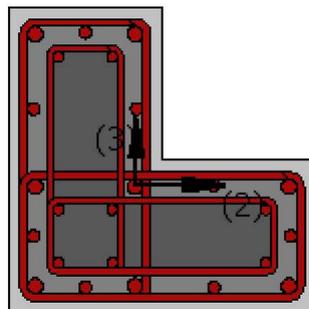
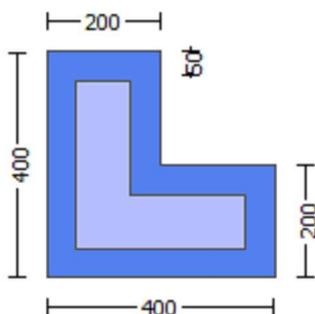
Limit State: Life Safety (data interpolation between analysis steps 50 and 51)

Analysis: Uniform +X

Check: Shear capacity  $VRd$

Edge: Start

Local Axis: (2)



Start Of Calculation of Shear Capacity for element: column JLC1 of floor 1

At local axis: 2

Integration Section: (a)

Section Type: rcjlc

Constant Properties

Knowledge Factor,  $\gamma = 0.90$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE 41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Jacket

New material of Secondary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 20.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 500.00$

Concrete Elasticity,  $E_c = 25742.96$

Steel Elasticity,  $E_s = 200000.00$

Existing Column

Existing material of Secondary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 12.00$

Existing material of Secondary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 400.00$

Concrete Elasticity,  $E_c = 19940.411$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of  $\gamma$  for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE 41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE 41-17).

Jacket

New material: Concrete Strength,  $f_c = f_{cm} = 30.00$

New material: Steel Strength,  $f_s = f_{sm} = 625.00$

Existing Column

Existing material: Concrete Strength,  $f_c = f_{cm} = 18.00$

Existing material: Steel Strength,  $f_s = f_{sm} = 500.00$

#####

Max Height,  $H_{max} = 400.00$

Min Height,  $H_{min} = 200.00$

Max Width,  $W_{max} = 400.00$

Min Width,  $W_{min} = 200.00$

Jacket Thickness,  $t_j = 50.00$

Cover Thickness,  $c = 15.00$

Element Length,  $L = 3000.00$

Secondary Member

Smooth Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou,min} = l_b/l_d >= 1$ )

No FRP Wrapping

Stepwise Properties

EDGE -A-

Bending Moment,  $M_a = -3.0601E+008$

Shear Force,  $V_a = -62256.122$

EDGE -B-

Bending Moment,  $M_b = -6.5602E+007$

Shear Force,  $V_b = 62256.122$

BOTH EDGES

Axial Force,  $F = -1.8013E+006$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{sl,t} = 2161.416$

-Compression:  $A_{sl,c} = 3191.858$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{sl,ten} = 2208.54$

-Compression:  $A_{sl,com} = 1137.257$

-Middle:  $A_{sl,mid} = 2007.478$

Mean Diameter of Tension Reinforcement,  $Db_{L,ten} = 16.60$

Existing component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = \phi V_n = 258166.022$   
 $V_n ((10.3), ASCE 41-17) = knl * V_{Col0} = 286851.135$   
 $V_{Col} = 286851.135$   
 $knl = 1.00$   
 $displacement\_ductility\_demand = 1.66929$

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + \phi * V_f$ '  
where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\phi = 1$  (normal-weight concrete)  
Mean concrete strength:  $f'_c = (f'_{c,jacket} * Area_{jacket} + f'_{c,core} * Area_{core}) / Area_{section} = 16.66667$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 4.00$   
 $\mu_u = 3.0601E+008$   
 $V_u = 62256.122$   
 $d = 0.8 * h = 320.00$   
 $N_u = 1.8013E+006$   
 $A_g = 80000.00$   
From (11.5.4.8), ACI 318-14:  $V_s = V_{s,jacket} + V_{s,core} = 376991.118$   
where:

$V_{s,jacket} = V_{s,j1} + V_{s,j2} = 376991.118$   
 $V_{s,j1} = 125663.706$  is calculated for section web jacket, with:  
 $d = 160.00$   
 $A_v = 157079.633$   
 $f_y = 500.00$   
 $s = 100.00$

$V_{s,j1}$  is multiplied by  $Col_{j1} = 1.00$   
 $s/d = 0.625$   
 $V_{s,j2} = 251327.412$  is calculated for section flange jacket, with:  
 $d = 320.00$   
 $A_v = 157079.633$   
 $f_y = 500.00$   
 $s = 100.00$

$V_{s,j2}$  is multiplied by  $Col_{j2} = 1.00$   
 $s/d = 0.3125$   
 $V_{s,core} = V_{s,c1} + V_{s,c2} = 0.00$   
 $V_{s,c1} = 0.00$  is calculated for section web core, with:  
 $d = 80.00$   
 $A_v = 100530.965$   
 $f_y = 400.00$   
 $s = 250.00$

$V_{s,c1}$  is multiplied by  $Col_{c1} = 0.00$   
 $s/d = 3.125$   
 $V_{s,c2} = 0.00$  is calculated for section flange core, with:  
 $d = 240.00$   
 $A_v = 100530.965$   
 $f_y = 400.00$   
 $s = 250.00$

$V_{s,c2}$  is multiplied by  $Col_{c2} = 0.00$   
 $s/d = 1.04167$   
 $V_f ((11-3)-(11.4), ACI 440) = 0.00$   
From (11-11), ACI 440:  $V_s + V_f \leq 173568.578$   
 $bw = 200.00$

$displacement\_ductility\_demand$  is calculated as  $\phi / y$

- Calculation of  $\phi / y$  for END A -  
for rotation axis 3 and integ. section (a)

From analysis, chord rotation  $\phi = 0.03426582$

$y = (M_y * L_s / 3) / E_{eff} = 0.02052718$  ((4.29), Biskinis Phd)  
 $M_y = 3.1260E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 4915.337  
From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 2.4951E+013$   
factor = 0.70  
 $A_g = 120000.00$   
Mean concrete strength:  $f_c' = (f_c'_{jacket} * Area_{jacket} + f_c'_{core} * Area_{core}) / Area_{section} = 25.00$   
 $N = 1.8013E+006$   
 $E_c * I_g = E_c_{jacket} * I_{g,jacket} + E_c_{core} * I_{g,core} = 3.5645E+013$

-----  
-----  
Calculation of Yielding Moment  $M_y$

-----  
Calculation of  $y$  and  $M_y$  according to Annex 7 -

-----  
 $y = \text{Min}(y_{ten}, y_{com})$   
 $y_{ten} = 1.9273633E-005$   
with  $f_y = 596.2441$   
 $d = 367.00$   
 $y = 0.57853212$   
 $A = 0.11409279$   
 $B = 0.08754673$   
with  $pt = 0.0300891$   
 $pc = 0.01549396$   
 $pv = 0.02734983$   
 $N = 1.8013E+006$   
 $b = 200.00$   
 $" = 0.08991826$   
 $y_{comp} = 7.6802956E-006$   
with  $f_c = 30.00$   
 $E_c = 25742.96$   
 $y = 0.74420275$   
 $A = 0.01443594$   
 $B = 0.04638683$   
with  $E_s = 200000.00$

-----  
Calculation of ratio  $I_b / I_d$

-----  
Adequate Lap Length:  $I_b / I_d \geq 1$

-----  
End Of Calculation of Shear Capacity for element: column JLC1 of floor 1

At local axis: 2

Integration Section: (a)

-----

**Calculation No. 10**

column C1, Floor 1

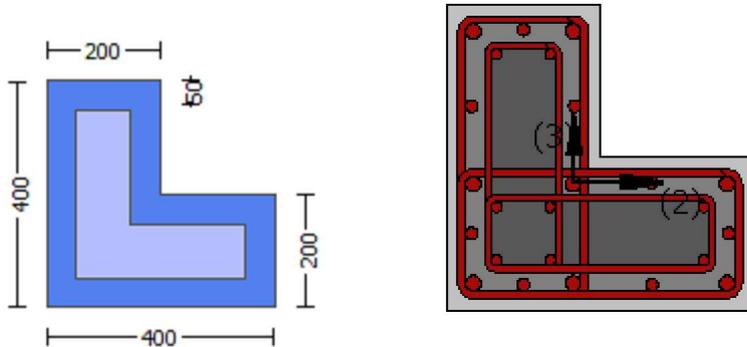
Limit State: Life Safety (data interpolation between analysis steps 50 and 51)

Analysis: Uniform +X

Check: Chord rotation capacity ( $\theta_u$ )

Edge: Start

Local Axis: (2)



Start Of Calculation of Shear Capacity ratio for element: column JLC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rcjlcs

Constant Properties

Knowledge Factor,  $\gamma = 0.90$

Mean strength values are used for both shear and moment calculations.

Consequently:

Jacket

New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 30.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 625.00$

Concrete Elasticity,  $E_c = 25742.96$

Steel Elasticity,  $E_s = 200000.00$

Existing Column

Existing material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 18.00$

Existing material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 500.00$

Concrete Elasticity,  $E_c = 19940.411$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Jacket

New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 781.25$

Existing Column

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 625.00$

#####

Max Height,  $H_{max} = 400.00$

Min Height,  $H_{min} = 200.00$

Max Width,  $W_{max} = 400.00$

Min Width,  $W_{min} = 200.00$

Jacket Thickness,  $t_j = 50.00$

Cover Thickness,  $c = 15.00$

Mean Confinement Factor overall section = 1.00

Element Length,  $L = 3000.00$

Secondary Member

Smooth Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Adequate Lap Length ( $l_o/l_{o,u}, \min \geq 1$ )  
No FRP Wrapping

-----  
Stepwise Properties  
-----

At local axis: 3  
EDGE -A-  
Shear Force,  $V_a = 2.3476251E-006$   
EDGE -B-  
Shear Force,  $V_b = -2.3476251E-006$   
BOTH EDGES  
Axial Force,  $F = -1.8026E+006$   
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $As_t = 0.00$   
-Compression:  $As_c = 5353.274$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $As_{t,ten} = 2208.54$   
-Compression:  $As_{c,com} = 1137.257$   
-Middle:  $As_{c,mid} = 2007.478$   
-----  
-----

Calculation of Shear Capacity ratio,  $V_e/V_r = 1.44151$   
Member Controlled by Shear ( $V_e/V_r > 1$ )  
Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 488890.189$   
with  
 $M_{pr1} = \max(\mu_{1+}, \mu_{1-}) = 7.3334E+008$   
 $\mu_{1+} = 7.3334E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination  
 $\mu_{1-} = 6.3768E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination  
 $M_{pr2} = \max(\mu_{2+}, \mu_{2-}) = 7.3334E+008$   
 $\mu_{2+} = 7.3334E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination  
 $\mu_{2-} = 6.3768E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

-----  
Calculation of  $\mu_{1+}$   
-----

-----  
Calculation of ultimate curvature  $\mu$  according to 4.1, Biskinis/Fardis 2013:

$$\mu = 4.2207094E-005$$

$$M_u = 7.3334E+008$$

-----  
with full section properties:

$$b = 400.00$$

$$d = 367.00$$

$$d' = 33.00$$

$$v = 0.40931126$$

$$N = 1.8026E+006$$

$$f_c = 30.00$$

$$c_o (5A.5, TBDY) = 0.002$$

$$\text{Final value of } \mu_c: \mu_c^* = \text{shear\_factor} * \max(\mu_c, \mu_{cc}) = 0.00925707$$

The Shear\_factor is considered equal to 1 (pure moment strength)

$$\text{From (5.4b), TBDY: } \mu_c = 0.00925707$$

$$\mu_{we} (5.4c) = 0.0207149$$

$$\mu_{ase} ((5.4d), TBDY) = (ase_1 * A_{ext} + ase_2 * A_{int}) / A_{sec} = 0.33836562$$

$$ase_1 = \max(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.  
 $A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).  
 $a_{se2} (>= a_{se1}) = \text{Max}(((A_{conf,max2} - A_{noConf2})/A_{conf,max2}) * (A_{conf,min2}/A_{conf,max2}), 0) = 0.33836562$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).  
 $p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe}, p_{sh,y} * F_{ywe}) = 6.12205$

Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

-----  
 $p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$   
 $p_{sh1}$  ((5.4d), TBDY) =  $L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$   
 $L_{stir1}$  (Length of stirrups along Y) = 1040.00  
 $A_{stir1}$  (stirrups area) = 78.53982  
 $p_{sh2}$  (5.4d) =  $L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$   
 $L_{stir2}$  (Length of stirrups along Y) = 768.00  
 $A_{stir2}$  (stirrups area) = 50.26548

-----  
 $p_{sh,y} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$   
 $p_{sh1}$  ((5.4d), TBDY) =  $L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$   
 $L_{stir1}$  (Length of stirrups along X) = 1040.00  
 $A_{stir1}$  (stirrups area) = 78.53982  
 $p_{sh2}$  ((5.4d), TBDY) =  $L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$   
 $L_{stir2}$  (Length of stirrups along X) = 768.00  
 $A_{stir2}$  (stirrups area) = 50.26548

-----  
 $A_{sec} = 120000.00$   
 $s1 = 100.00$   
 $s2 = 250.00$   
 $f_{ywe1} = 781.25$   
 $f_{ywe2} = 625.00$   
 $f_{ce} = 30.00$

From ((5.A5), TBDY), TBDY:  $cc = 0.002$   
 $c = \text{confinement factor} = 1.00$

$y1 = 0.0025$   
 $sh1 = 0.008$   
 $ft1 = 898.293$   
 $fy1 = 748.5775$   
 $su1 = 0.032$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

$l_o/l_{ou,min} = l_b/l_d = 1.00$

$su1 = 0.4 * esu1_{nominal}$  ((5.5), TBDY) = 0.032

From table 5A.1, TBDY:  $esu1_{nominal} = 0.08$ ,

For calculation of  $esu1_{nominal}$  and  $y1, sh1, ft1, fy1$ , it is considered characteristic value  $fsy1 = fs1/1.2$ , from table 5.1, TBDY.

$y1, sh1, ft1, fy1$ , are also multiplied by  $\text{Min}(1, 1.25 * (l_b/l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $fs1 = (f_{s,jacket} * A_{s1,ten,jacket} + f_{s,core} * A_{s1,ten,core}) / A_{s1,ten} = 748.5775$

with  $Es1 = (E_{s,jacket} * A_{s1,ten,jacket} + E_{s,core} * A_{s1,ten,core}) / A_{s1,ten} = 200000.00$

$y2 = 0.0025$   
 $sh2 = 0.008$   
 $ft2 = 886.7403$

```

fy2 = 738.9503
su2 = 0.032
using (30) in Biskinis/Fardis (2013) multiplied with shear_factor
and also multiplied by the shear_factor according to 15.7.1.4, with
Shear_factor = 1.00
lo/lou,min = lb/lb,min = 1.00
su2 = 0.4*esu2_nominal ((5.5), TBDY) = 0.032
From table 5A.1, TBDY: esu2_nominal = 0.08,
For calculation of esu2_nominal and y2, sh2,ft2,fy2, it is considered
characteristic value fsy2 = fs2/1.2, from table 5.1, TBDY.
y1, sh1,ft1,fy1, are also multiplied by Min(1,1.25*(lb/d)^ 2/3), from 10.3.5, ASCE 41-17.
with fs2 = (fs,jacket*Asl,com,jacket + fs,core*Asl,com,core)/Asl,com = 738.9503
with Es2 = (Es,jacket*Asl,com,jacket + Es,core*Asl,com,core)/Asl,com = 200000.00
yv = 0.0025
shv = 0.008
ftv = 894.3662
fyv = 745.3052
suv = 0.032
using (30) in Biskinis/Fardis (2013) multiplied with shear_factor
and also multiplied by the shear_factor according to 15.7.1.4, with
Shear_factor = 1.00
lo/lou,min = lb/d = 1.00
suv = 0.4*esuv_nominal ((5.5), TBDY) = 0.032
From table 5A.1, TBDY: esuv_nominal = 0.08,
considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY
For calculation of esuv_nominal and yv, shv,ftv,fyv, it is considered
characteristic value fsyv = fsv/1.2, from table 5.1, TBDY.
y1, sh1,ft1,fy1, are also multiplied by Min(1,1.25*(lb/d)^ 2/3), from 10.3.5, ASCE 41-17.
with fsv = (fs,jacket*Asl,mid,jacket + fs,mid*Asl,mid,core)/Asl,mid = 745.3052
with Esv = (Es,jacket*Asl,mid,jacket + Es,mid*Asl,mid,core)/Asl,mid = 200000.00
1 = Asl,ten/(b*d)*(fs1/fc) = 0.37540035
2 = Asl,com/(b*d)*(fs2/fc) = 0.19082108
v = Asl,mid/(b*d)*(fsv/fc) = 0.33973286
and confined core properties:
b = 360.00
d = 347.00
d' = 13.00
fcc (5A.2, TBDY) = 30.00
cc (5A.5, TBDY) = 0.002
c = confinement factor = 1.00
1 = Asl,ten/(b*d)*(fs1/fc) = 0.44115251
2 = Asl,com/(b*d)*(fs2/fc) = 0.22424379
v = Asl,mid/(b*d)*(fsv/fc) = 0.39923778
Case/Assumption: Unconfined full section - Steel rupture
' satisfies Eq. (4.3)
---->
v < vs,y2 - LHS eq.(4.5) is not satisfied
---->
v < vs,c - RHS eq.(4.5) is not satisfied
---->
Case/Assumption Rejected.
---->
New Case/Assumption: Unconfined full section - Spalling of concrete cover
' satisfies Eq. (4.4)
---->
v < s,y1 - LHS eq.(4.7) is not satisfied
---->
v < vc,y1 - RHS eq.(4.6) is satisfied
---->
cu (4.10) = 0.59761571
MRc (4.17) = 7.3334E+008
---->
New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover
In expressions below, the following modifications have been made
- b, d, d' replaced by geometric parameters of the core: bo, do, d'o
- N, 1, 2, v normalised to bo*do, instead of b*d
- - parameters of confined concrete, fcc, cc, used in lieu of fc, ecu

```

--->  
Subcase: Rupture of tension steel  
--->  
 $v^* < v^*_{s,y2}$  - LHS eq.(4.5) is not satisfied  
--->  
 $v^* < v^*_{s,c}$  - LHS eq.(4.5) is not satisfied  
--->  
Subcase rejected

--->  
New Subcase: Failure of compression zone  
--->  
 $v^* < v^*_{c,y2}$  - LHS eq.(4.6) is not satisfied  
--->  
 $v^* < v^*_{c,y1}$  - RHS eq.(4.6) is not satisfied  
--->  
 $*c_u$  (4.11) = 0.62879536  
MRo (4.18) = 4.5866E+008  
MRo < 0.8\*MRc

--->  
 $u = c_u$  (unconfined full section) = 4.2207094E-005  
Mu = MRc

-----  
Calculation of ratio  $l_b/d$   
-----

Adequate Lap Length:  $l_b/d \geq 1$   
-----  
-----  
-----

Calculation of  $\mu_1$ -  
-----  
-----

-----  
Calculation of ultimate curvature  $\mu$  according to 4.1, Biskinis/Fardis 2013:  
 $u = 4.8346386E-005$   
Mu = 6.3768E+008  
-----

with full section properties:

$b = 200.00$   
 $d = 367.00$   
 $d' = 33.00$   
 $v = 0.81862252$   
 $N = 1.8026E+006$   
 $f_c = 30.00$   
 $\omega$  (5A.5, TBDY) = 0.002

Final value of  $c_u$ :  $c_u^* = \text{shear\_factor} * \text{Max}(c_u, c_c) = 0.00925707$

The Shear\_factor is considered equal to 1 (pure moment strength)

From (5.4b), TBDY:  $c_u = 0.00925707$

$w_e$  (5.4c) = 0.0207149

$a_{se}$  ((5.4d), TBDY) =  $(a_{se1} * A_{ext} + a_{se2} * A_{int}) / A_{sec} = 0.33836562$

$a_{se1} = \text{Max}(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).

$a_{se2} (\geq a_{se1}) = \text{Max}(((A_{conf,max2} - A_{noConf2}) / A_{conf,max2}) * (A_{conf,min2} / A_{conf,max2}), 0) = 0.33836562$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe}, p_{sh,y} * F_{ywe}) = 6.12205$

Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

-----  
 $p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{s2} * F_{ywe2} = 6.12205$

$p_{sh1}$  ((5.4d), TBDY) =  $L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$

$L_{stir1}$  (Length of stirrups along Y) = 1040.00

$A_{stir1}$  (stirrups area) = 78.53982

$p_{sh2}$  (5.4d) =  $L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$

$L_{stir2}$  (Length of stirrups along Y) = 768.00

$A_{stir2}$  (stirrups area) = 50.26548

-----  
 $p_{sh,y} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{s2} * F_{ywe2} = 6.12205$

$p_{sh1}$  ((5.4d), TBDY) =  $L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$

$L_{stir1}$  (Length of stirrups along X) = 1040.00

$A_{stir1}$  (stirrups area) = 78.53982

$p_{sh2}$  ((5.4d), TBDY) =  $L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$

$L_{stir2}$  (Length of stirrups along X) = 768.00

$A_{stir2}$  (stirrups area) = 50.26548

-----  
 $A_{sec} = 120000.00$

$s1 = 100.00$

$s2 = 250.00$

$f_{ywe1} = 781.25$

$f_{ywe2} = 625.00$

$f_{ce} = 30.00$

From ((5.A5), TBDY), TBDY:  $c_c = 0.002$

$c = \text{confinement factor} = 1.00$

$y1 = 0.0025$

$sh1 = 0.008$

$ft1 = 886.7403$

$fy1 = 738.9503$

$su1 = 0.032$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

$l_o/l_{o,min} = l_b/l_d = 1.00$

$su1 = 0.4 * esu1_{nominal}$  ((5.5), TBDY) = 0.032

From table 5A.1, TBDY:  $esu1_{nominal} = 0.08$ ,

For calculation of  $esu1_{nominal}$  and  $y1, sh1, ft1, fy1$ , it is considered characteristic value  $fsy1 = fs1/1.2$ , from table 5.1, TBDY.

$y1, sh1, ft1, fy1$ , are also multiplied by  $\text{Min}(1, 1.25 * (l_b/l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $fs1 = (f_{s,jacket} * A_{sl,ten,jacket} + f_{s,core} * A_{sl,ten,core}) / A_{sl,ten} = 738.9503$

with  $Es1 = (E_{s,jacket} * A_{sl,ten,jacket} + E_{s,core} * A_{sl,ten,core}) / A_{sl,ten} = 200000.00$

$y2 = 0.0025$

$sh2 = 0.008$

$ft2 = 898.293$

$fy2 = 748.5775$

$su2 = 0.032$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

$l_o/l_{o,min} = l_b/l_{b,min} = 1.00$

$su2 = 0.4 * esu2_{nominal}$  ((5.5), TBDY) = 0.032

From table 5A.1, TBDY:  $esu2_{nominal} = 0.08$ ,

For calculation of  $esu2_{nominal}$  and  $y2, sh2, ft2, fy2$ , it is considered characteristic value  $fsy2 = fs2/1.2$ , from table 5.1, TBDY.

$y1, sh1, ft1, fy1$ , are also multiplied by  $\text{Min}(1, 1.25 * (l_b/l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $fs2 = (f_{s,jacket} * A_{sl,com,jacket} + f_{s,core} * A_{sl,com,core}) / A_{sl,com} = 748.5775$

with  $Es2 = (E_{s,jacket} * A_{sl,com,jacket} + E_{s,core} * A_{sl,com,core}) / A_{sl,com} = 200000.00$

$y_v = 0.0025$   
 $sh_v = 0.008$   
 $ft_v = 894.3662$   
 $fy_v = 745.3052$   
 $suv = 0.032$   
 using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
 and also multiplied by the shear\_factor according to 15.7.1.4, with  
 Shear\_factor = 1.00  
 $lo/lo_{min} = lb/ld = 1.00$   
 $suv = 0.4 * esuv_{nominal} ((5.5), TBDY) = 0.032$   
 From table 5A.1, TBDY:  $esuv_{nominal} = 0.08$ ,  
 considering characteristic value  $fs_v = fsv/1.2$ , from table 5.1, TBDY  
 For calculation of  $esuv_{nominal}$  and  $y_v, sh_v, ft_v, fy_v$ , it is considered  
 characteristic value  $fs_v = fsv/1.2$ , from table 5.1, TBDY.  
 $y_1, sh_1, ft_1, fy_1$ , are also multiplied by  $Min(1, 1.25 * (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.  
 with  $fsv = (fs_{jacket} * Asl_{mid,jacket} + fs_{mid} * Asl_{mid,core}) / Asl_{mid} = 745.3052$   
 with  $Esv = (Es_{jacket} * Asl_{mid,jacket} + Es_{mid} * Asl_{mid,core}) / Asl_{mid} = 200000.00$   
 $1 = Asl_{ten} / (b * d) * (fs_1 / fc) = 0.38164216$   
 $2 = Asl_{com} / (b * d) * (fs_2 / fc) = 0.7508007$   
 $v = Asl_{mid} / (b * d) * (fsv / fc) = 0.67946571$   
 and confined core properties:  
 $b = 160.00$   
 $d = 347.00$   
 $d' = 13.00$   
 $fcc (5A.2, TBDY) = 30.00$   
 $cc (5A.5, TBDY) = 0.002$   
 $c = \text{confinement factor} = 1.00$   
 $1 = Asl_{ten} / (b * d) * (fs_1 / fc) = 0.50454853$   
 $2 = Asl_{com} / (b * d) * (fs_2 / fc) = 0.99259314$   
 $v = Asl_{mid} / (b * d) * (fsv / fc) = 0.898285$   
 Case/Assumption: Unconfined full section - Steel rupture  
 ' satisfies Eq. (4.3)  
 --->  
 $v < v_{s,y2}$  - LHS eq.(4.5) is not satisfied  
 --->  
 $v < v_{s,c}$  - RHS eq.(4.5) is not satisfied  
 --->  
 Case/Assumption Rejected.  
 --->  
 New Case/Assumption: Unconfined full section - Spalling of concrete cover  
 ' satisfies Eq. (4.4)  
 --->  
 $v < s_{y1}$  - LHS eq.(4.7) is not satisfied  
 --->  
 $v < v_{c,y1}$  - RHS eq.(4.6) is satisfied  
 --->  
 $cu (4.10) = 0.52172715$   
 $MRC (4.17) = 6.3768E+008$   
 --->  
 New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover  
 In expressions below, the following modifications have been made  
 -  $b, d, d'$  replaced by geometric parameters of the core:  $bo, do, d'o$   
 -  $N, 1, 2, v$  normalised to  $bo * do$ , instead of  $b * d$   
 -  $fcc, cc$ , used in lieu of  $fc, ecu$   
 --->  
 Subcase: Rupture of tension steel  
 --->  
 $v^* < v^*_{s,y2}$  - LHS eq.(4.5) is not satisfied  
 --->  
 $v^* < v^*_{s,c}$  - LHS eq.(4.5) is not satisfied  
 --->  
 Subcase rejected  
 --->  
 New Subcase: Failure of compression zone  
 --->  
 $v^* < v^*_{c,y2}$  - LHS eq.(4.6) is not satisfied  
 --->

$v^* < v^*_{c,y1}$  - RHS eq.(4.6) is not satisfied

--->

$$*cu(4.11) = 0.5604845$$

$$MRo(4.18) = 4.7738E+008$$

$$MRo < 0.8*MRc$$

--->

$$u = cu(\text{unconfined full section}) = 4.8346386E-005$$

$$Mu = MRc$$

Calculation of ratio  $lb/d$

Adequate Lap Length:  $lb/d \geq 1$

Calculation of  $Mu_{2+}$

Calculation of ultimate curvature  $u$  according to 4.1, Biskinis/Fardis 2013:

$$u = 4.2207094E-005$$

$$Mu = 7.3334E+008$$

with full section properties:

$$b = 400.00$$

$$d = 367.00$$

$$d' = 33.00$$

$$v = 0.40931126$$

$$N = 1.8026E+006$$

$$fc = 30.00$$

$$co(5A.5, TBDY) = 0.002$$

$$\text{Final value of } cu: cu^* = \text{shear\_factor} * \text{Max}(cu, cc) = 0.00925707$$

The Shear\_factor is considered equal to 1 (pure moment strength)

$$\text{From (5.4b), TBDY: } cu = 0.00925707$$

$$we(5.4c) = 0.0207149$$

$$ase((5.4d), TBDY) = (ase1*Aext+ase2*Aint)/Asec = 0.33836562$$

$$ase1 = \text{Max}(((Aconf,max1-AnoConf1)/Aconf,max1)*(Aconf,min1/Aconf,max1),0) = 0.33836562$$

The definitions of  $AnoConf$ ,  $Aconf,min$  and  $Aconf,max$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$Aconf,max1 = 89600.00$  is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$Aconf,min1 = 59225.00$  is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area  $Aconf,max1$  by a length equal to half the clear spacing between external hoops.

$AnoConf1 = 43733.333$  is the unconfined external core area which is equal to  $bi^2/6$  as defined at (A.2).

$$ase2(\geq ase1) = \text{Max}(((Aconf,max2-AnoConf2)/Aconf,max2)*(Aconf,min2/Aconf,max2),0) = 0.33836562$$

The definitions of  $AnoConf$ ,  $Aconf,min$  and  $Aconf,max$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$Aconf,max2 = 45264.00$  is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$Aconf,min2 = 0.00$  is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area  $Aconf,max2$  by a length equal to half the clear spacing between internal hoops.

$AnoConf2 = 32309.333$  is the unconfined internal core area which is equal to  $bi^2/6$  as defined at (A.2).

$$psh,min*Fywe = \text{Min}(psh,x*Fywe, psh,y*Fywe) = 6.12205$$

Expression (5.4d) for  $psh,min*Fywe$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

$$psh_x*Fywe = psh1*Fywe1+ps2*Fywe2 = 6.12205$$

$$psh1((5.4d), TBDY) = Lstir1*Astir1/(Asec*s1) = 0.00680678$$

Lstir1 (Length of stirrups along Y) = 1040.00  
Astir1 (stirrups area) = 78.53982  
psh2 (5.4d) = Lstir2\*Astir2/(Asec\*s2) = 0.0012868  
Lstir2 (Length of stirrups along Y) = 768.00  
Astir2 (stirrups area) = 50.26548

psh\_y\*Fywe = psh1\*Fywe1+ps2\*Fywe2 = 6.12205  
psh1 ((5.4d), TBDY) = Lstir1\*Astir1/(Asec\*s1) = 0.00680678  
Lstir1 (Length of stirrups along X) = 1040.00  
Astir1 (stirrups area) = 78.53982  
psh2 ((5.4d), TBDY) = Lstir2\*Astir2/(Asec\*s2) = 0.0012868  
Lstir2 (Length of stirrups along X) = 768.00  
Astir2 (stirrups area) = 50.26548

Asec = 120000.00

s1 = 100.00

s2 = 250.00

fywe1 = 781.25

fywe2 = 625.00

fce = 30.00

From ((5.A5), TBDY), TBDY: cc = 0.002

c = confinement factor = 1.00

y1 = 0.0025

sh1 = 0.008

ft1 = 898.293

fy1 = 748.5775

su1 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb = 1.00

su1 = 0.4\*esu1\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1,ft1,fy1, it is considered  
characteristic value fsy1 = fs1/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 * (\text{lb}/\text{ld})^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs1 = (fs,jacket\*Asl,ten,jacket + fs,core\*Asl,ten,core)/Asl,ten = 748.5775

with Es1 = (Es,jacket\*Asl,ten,jacket + Es,core\*Asl,ten,core)/Asl,ten = 200000.00

y2 = 0.0025

sh2 = 0.008

ft2 = 886.7403

fy2 = 738.9503

su2 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb,min = 1.00

su2 = 0.4\*esu2\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu2\_nominal = 0.08,

For calculation of esu2\_nominal and y2, sh2,ft2,fy2, it is considered  
characteristic value fsy2 = fs2/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 * (\text{lb}/\text{ld})^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs2 = (fs,jacket\*Asl,com,jacket + fs,core\*Asl,com,core)/Asl,com = 738.9503

with Es2 = (Es,jacket\*Asl,com,jacket + Es,core\*Asl,com,core)/Asl,com = 200000.00

yv = 0.0025

shv = 0.008

ftv = 894.3662

fyv = 745.3052

suv = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb = 1.00

suv = 0.4\*esuv\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esuv\_nominal = 0.08,

considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY

For calculation of esuv\_nominal and yv, shv,ftv,fyv, it is considered

characteristic value  $f_{sv} = f_{sv}/1.2$ , from table 5.1, TBDY.

$y_1, sh_1, ft_1, fy_1$ , are also multiplied by  $\text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $f_{sv} = (f_{s,jacket} \cdot A_{s,mid,jacket} + f_{s,mid} \cdot A_{s,mid,core}) / A_{s,mid} = 745.3052$

with  $E_{sv} = (E_{s,jacket} \cdot A_{s,mid,jacket} + E_{s,mid} \cdot A_{s,mid,core}) / A_{s,mid} = 200000.00$

1 =  $A_{s,ten} / (b \cdot d) \cdot (f_{s1}/f_c) = 0.37540035$

2 =  $A_{s,com} / (b \cdot d) \cdot (f_{s2}/f_c) = 0.19082108$

$v = A_{s,mid} / (b \cdot d) \cdot (f_{sv}/f_c) = 0.33973286$

and confined core properties:

$b = 360.00$

$d = 347.00$

$d' = 13.00$

$f_{cc} (5A.2, TBDY) = 30.00$

$cc (5A.5, TBDY) = 0.002$

$c = \text{confinement factor} = 1.00$

1 =  $A_{s,ten} / (b \cdot d) \cdot (f_{s1}/f_c) = 0.44115251$

2 =  $A_{s,com} / (b \cdot d) \cdot (f_{s2}/f_c) = 0.22424379$

$v = A_{s,mid} / (b \cdot d) \cdot (f_{sv}/f_c) = 0.39923778$

Case/Assumption: Unconfined full section - Steel rupture

' satisfies Eq. (4.3)

---->

$v < v_{s,y2}$  - LHS eq.(4.5) is not satisfied

---->

$v < v_{s,c}$  - RHS eq.(4.5) is not satisfied

---->

Case/Assumption Rejected.

---->

New Case/Assumption: Unconfined full section - Spalling of concrete cover

' satisfies Eq. (4.4)

---->

$v < s_{y1}$  - LHS eq.(4.7) is not satisfied

---->

$v < v_{c,y1}$  - RHS eq.(4.6) is satisfied

---->

$c_u (4.10) = 0.59761571$

$M_{Rc} (4.17) = 7.3334E+008$

---->

New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover

In expressions below, the following modifications have been made

-  $b, d, d'$  replaced by geometric parameters of the core:  $b_o, d_o, d'_o$

-  $N, 1, 2, v$  normalised to  $b_o \cdot d_o$ , instead of  $b \cdot d$

-  $f_{cc}, cc$  parameters of confined concrete,  $f_{cc}, cc$ , used in lieu of  $f_c, e_{cu}$

---->

Subcase: Rupture of tension steel

---->

$v^* < v^*_{s,y2}$  - LHS eq.(4.5) is not satisfied

---->

$v^* < v^*_{s,c}$  - LHS eq.(4.5) is not satisfied

---->

Subcase rejected

---->

New Subcase: Failure of compression zone

---->

$v^* < v^*_{c,y2}$  - LHS eq.(4.6) is not satisfied

---->

$v^* < v^*_{c,y1}$  - RHS eq.(4.6) is not satisfied

---->

$*c_u (4.11) = 0.62879536$

$M_{Ro} (4.18) = 4.5866E+008$

$M_{Ro} < 0.8 \cdot M_{Rc}$

---->

$u = c_u$  (unconfined full section) =  $4.2207094E-005$

$\mu = M_{Rc}$

-----

Calculation of ratio  $l_b/l_d$

-----

Adequate Lap Length:  $l_b/l_d \geq 1$

Calculation of Mu2-

Calculation of ultimate curvature  $\mu$  according to 4.1, Biskinis/Fardis 2013:

$$\mu = 4.8346386E-005$$

$$Mu = 6.3768E+008$$

with full section properties:

$$b = 200.00$$

$$d = 367.00$$

$$d' = 33.00$$

$$v = 0.81862252$$

$$N = 1.8026E+006$$

$$f_c = 30.00$$

$$c_o \text{ (5A.5, TBDY)} = 0.002$$

$$\text{Final value of } c_u: c_u^* = \text{shear\_factor} * \text{Max}(c_u, c_o) = 0.00925707$$

The Shear\_factor is considered equal to 1 (pure moment strength)

$$\text{From (5.4b), TBDY: } c_u = 0.00925707$$

$$w_e \text{ (5.4c)} = 0.0207149$$

$$a_{se} \text{ ((5.4d), TBDY)} = (a_{se1} * A_{ext} + a_{se2} * A_{int}) / A_{sec} = 0.33836562$$

$$a_{se1} = \text{Max}(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$a_{se2} (>= a_{se1}) = \text{Max}(((A_{conf,max2} - A_{noConf2}) / A_{conf,max2}) * (A_{conf,min2} / A_{conf,max2}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe}, p_{sh,y} * F_{ywe}) = 6.12205$$

Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

$$p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$$

$$p_{sh1} \text{ ((5.4d), TBDY)} = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$$

$$L_{stir1} \text{ (Length of stirrups along Y)} = 1040.00$$

$$A_{stir1} \text{ (stirrups area)} = 78.53982$$

$$p_{sh2} \text{ (5.4d)} = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$$

$$L_{stir2} \text{ (Length of stirrups along Y)} = 768.00$$

$$A_{stir2} \text{ (stirrups area)} = 50.26548$$

$$p_{sh,y} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$$

$$p_{sh1} \text{ ((5.4d), TBDY)} = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$$

$$L_{stir1} \text{ (Length of stirrups along X)} = 1040.00$$

$$A_{stir1} \text{ (stirrups area)} = 78.53982$$

$$p_{sh2} \text{ ((5.4d), TBDY)} = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$$

$$L_{stir2} \text{ (Length of stirrups along X)} = 768.00$$

$$A_{stir2} \text{ (stirrups area)} = 50.26548$$

Asec = 120000.00

s1 = 100.00

s2 = 250.00

fywe1 = 781.25

fywe2 = 625.00

fce = 30.00

From ((5.A.5), TBDY), TBDY: cc = 0.002

c = confinement factor = 1.00

y1 = 0.0025

sh1 = 0.008

ft1 = 886.7403

fy1 = 738.9503

su1 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb = 1.00

su1 = 0.4\*esu1\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1,ft1,fy1, it is considered  
characteristic value fsy1 = fs1/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (\text{lb}/\text{ld})^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs1 = (fsjacket\*Asl,ten,jacket + fs,core\*Asl,ten,core)/Asl,ten = 738.9503

with Es1 = (Es,jacket\*Asl,ten,jacket + Es,core\*Asl,ten,core)/Asl,ten = 200000.00

y2 = 0.0025

sh2 = 0.008

ft2 = 898.293

fy2 = 748.5775

su2 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb,min = 1.00

su2 = 0.4\*esu2\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu2\_nominal = 0.08,

For calculation of esu2\_nominal and y2, sh2,ft2,fy2, it is considered  
characteristic value fsy2 = fs2/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (\text{lb}/\text{ld})^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs2 = (fsjacket\*Asl,com,jacket + fs,core\*Asl,com,core)/Asl,com = 748.5775

with Es2 = (Es,jacket\*Asl,com,jacket + Es,core\*Asl,com,core)/Asl,com = 200000.00

yv = 0.0025

shv = 0.008

ftv = 894.3662

fyv = 745.3052

suv = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb = 1.00

suv = 0.4\*esuv\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esuv\_nominal = 0.08,

considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY  
For calculation of esuv\_nominal and yv, shv,ftv,fyv, it is considered

characteristic value fsyv = fsv/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (\text{lb}/\text{ld})^{2/3})$ , from 10.3.5, ASCE 41-17.

with fsv = (fsjacket\*Asl,mid,jacket + fs,mid\*Asl,mid,core)/Asl,mid = 745.3052

with Esv = (Es,jacket\*Asl,mid,jacket + Es,mid\*Asl,mid,core)/Asl,mid = 200000.00

1 = Asl,ten/(b\*d)\*(fs1/fc) = 0.38164216

2 = Asl,com/(b\*d)\*(fs2/fc) = 0.7508007

v = Asl,mid/(b\*d)\*(fsv/fc) = 0.67946571

and confined core properties:

b = 160.00

d = 347.00

d' = 13.00

fcc (5A.2, TBDY) = 30.00

cc (5A.5, TBDY) = 0.002

$$c = \text{confinement factor} = 1.00$$

$$1 = A_{sl,ten}/(b*d)*(f_{s1}/f_c) = 0.50454853$$

$$2 = A_{sl,com}/(b*d)*(f_{s2}/f_c) = 0.99259314$$

$$v = A_{sl,mid}/(b*d)*(f_{sv}/f_c) = 0.898285$$

Case/Assumption: Unconfined full section - Steel rupture  
' satisfies Eq. (4.3)

---->  
 $v < v_{s,y2}$  - LHS eq.(4.5) is not satisfied

---->  
 $v < v_{s,c}$  - RHS eq.(4.5) is not satisfied

---->  
Case/Assumption Rejected.

---->  
New Case/Assumption: Unconfined full section - Spalling of concrete cover  
' satisfies Eq. (4.4)

---->  
 $v < s_{y1}$  - LHS eq.(4.7) is not satisfied

---->  
 $v < v_{c,y1}$  - RHS eq.(4.6) is satisfied

---->  
 $c_u$  (4.10) = 0.52172715  
MRC (4.17) = 6.3768E+008

---->  
New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover  
In expressions below, the following modifications have been made

- b, d, d' replaced by geometric parameters of the core:  $b_o, d_o, d'_o$
- $N, 1, 2, v$  normalised to  $b_o*d_o$ , instead of  $b*d$
- - parameters of confined concrete,  $f_{cc}, c_c$ , used in lieu of  $f_c, c_u$

---->  
Subcase: Rupture of tension steel

---->  
 $v^* < v^*_{s,y2}$  - LHS eq.(4.5) is not satisfied

---->  
 $v^* < v^*_{s,c}$  - LHS eq.(4.5) is not satisfied

---->  
Subcase rejected

---->  
New Subcase: Failure of compression zone

---->  
 $v^* < v^*_{c,y2}$  - LHS eq.(4.6) is not satisfied

---->  
 $v^* < v^*_{c,y1}$  - RHS eq.(4.6) is not satisfied

---->  
 $*c_u$  (4.11) = 0.5604845  
MRO (4.18) = 4.7738E+008

MRO < 0.8\*MRC

---->  
 $u = c_u$  (unconfined full section) = 4.8346386E-005  
 $\mu = MRC$

-----

Calculation of ratio  $l_b/l_d$

-----

Adequate Lap Length:  $l_b/l_d \geq 1$

-----

-----

-----

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 339150.739$

-----

Calculation of Shear Strength at edge 1,  $V_{r1} = 339150.739$

$$V_{r1} = V_{Co1} \text{ ((10.3), ASCE 41-17)} = k_{nl} * V_{Co10}$$

$$V_{Co10} = 339150.739$$

$$k_{nl} = 1 \text{ (zero step-static loading)}$$

-----

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_{s+} = f * V_f$   
where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

-----  
 = 1 (normal-weight concrete)  
 Mean concrete strength:  $f_c' = (f_c'_{\text{jacket}} \cdot \text{Area}_{\text{jacket}} + f_c'_{\text{core}} \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 25.00$ , but  $f_c'^{0.5} \leq 8.3$   
 MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 4.00$   
 $\mu_u = 2.0345E+006$   
 $V_u = 2.3476251E-006$   
 $d = 0.8 \cdot h = 320.00$   
 $N_u = 1.8026E+006$   
 $A_g = 80000.00$   
 From (11.5.4.8), ACI 318-14:  $V_s = V_{s,\text{jacket}} + V_{s,\text{core}} = 471238.898$   
 where:  
 $V_{s,\text{jacket}} = V_{s,j1} + V_{s,j2} = 471238.898$   
 $V_{s,j1} = 314159.265$  is calculated for section web jacket, with:  
 $d = 320.00$   
 $A_v = 157079.633$   
 $f_y = 625.00$   
 $s = 100.00$   
 $V_{s,j1}$  is multiplied by  $\text{Col},j1 = 1.00$   
 $s/d = 0.3125$   
 $V_{s,j2} = 157079.633$  is calculated for section flange jacket, with:  
 $d = 160.00$   
 $A_v = 157079.633$   
 $f_y = 625.00$   
 $s = 100.00$   
 $V_{s,j2}$  is multiplied by  $\text{Col},j2 = 1.00$   
 $s/d = 0.625$   
 $V_{s,\text{core}} = V_{s,c1} + V_{s,c2} = 0.00$   
 $V_{s,c1} = 0.00$  is calculated for section web core, with:  
 $d = 240.00$   
 $A_v = 100530.965$   
 $f_y = 500.00$   
 $s = 250.00$   
 $V_{s,c1}$  is multiplied by  $\text{Col},c1 = 0.00$   
 $s/d = 1.04167$   
 $V_{s,c2} = 0.00$  is calculated for section flange core, with:  
 $d = 80.00$   
 $A_v = 100530.965$   
 $f_y = 500.00$   
 $s = 250.00$   
 $V_{s,c2}$  is multiplied by  $\text{Col},c2 = 0.00$   
 $s/d = 3.125$   
 $V_f ((11-3)-(11.4), \text{ACI } 440) = 0.00$   
 From (11-11), ACI 440:  $V_s + V_f \leq 212577.225$   
 $bw = 200.00$   
 -----

-----  
 Calculation of Shear Strength at edge 2,  $V_{r2} = 339150.739$   
 $V_{r2} = V_{\text{Col}} ((10.3), \text{ASCE } 41-17) = k_{nl} \cdot V_{\text{Col}0}$   
 $V_{\text{Col}0} = 339150.739$   
 $k_{nl} = 1$  (zero step-static loading)  
 -----

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ '  
 where  $V_f$  is the contribution of FRPs (11.3), ACI 440).  
 -----

-----  
 = 1 (normal-weight concrete)  
 Mean concrete strength:  $f_c' = (f_c'_{\text{jacket}} \cdot \text{Area}_{\text{jacket}} + f_c'_{\text{core}} \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 25.00$ , but  $f_c'^{0.5} \leq 8.3$   
 MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 4.00$   
 $\mu_u = 6739.685$   
 $V_u = 2.3476251E-006$   
 $d = 0.8 \cdot h = 320.00$   
 $N_u = 1.8026E+006$   
 $A_g = 80000.00$   
 From (11.5.4.8), ACI 318-14:  $V_s = V_{s,\text{jacket}} + V_{s,\text{core}} = 471238.898$   
 where:  
 $V_{s,\text{jacket}} = V_{s,j1} + V_{s,j2} = 471238.898$   
 -----

Vs,j1 = 314159.265 is calculated for section web jacket, with:

d = 320.00  
Av = 157079.633  
fy = 625.00  
s = 100.00

Vs,j1 is multiplied by Col,j1 = 1.00

s/d = 0.3125

Vs,j2 = 157079.633 is calculated for section flange jacket, with:

d = 160.00  
Av = 157079.633  
fy = 625.00  
s = 100.00

Vs,j2 is multiplied by Col,j2 = 1.00

s/d = 0.625

Vs,core = Vs,c1 + Vs,c2 = 0.00

Vs,c1 = 0.00 is calculated for section web core, with:

d = 240.00  
Av = 100530.965  
fy = 500.00  
s = 250.00

Vs,c1 is multiplied by Col,c1 = 0.00

s/d = 1.04167

Vs,c2 = 0.00 is calculated for section flange core, with:

d = 80.00  
Av = 100530.965  
fy = 500.00  
s = 250.00

Vs,c2 is multiplied by Col,c2 = 0.00

s/d = 3.125

Vf ((11-3)-(11.4), ACI 440) = 0.00

From (11-11), ACI 440: Vs + Vf <= 212577.225

bw = 200.00

-----  
End Of Calculation of Shear Capacity ratio for element: column JLC1 of floor 1  
At local axis: 3

-----  
Start Of Calculation of Shear Capacity ratio for element: column JLC1 of floor 1  
At Shear local axis: 2  
(Bending local axis: 3)  
Section Type: rcjcs

Constant Properties

-----  
Knowledge Factor, = 0.90

Mean strength values are used for both shear and moment calculations.

Consequently:

Jacket

New material of Secondary Member: Concrete Strength, fc = fcm = 30.00

New material of Secondary Member: Steel Strength, fs = fsm = 625.00

Concrete Elasticity, Ec = 25742.96

Steel Elasticity, Es = 200000.00

Existing Column

Existing material of Secondary Member: Concrete Strength, fc = fcm = 18.00

Existing material of Secondary Member: Steel Strength, fs = fsm = 500.00

Concrete Elasticity, Ec = 19940.411

Steel Elasticity, Es = 200000.00

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Jacket

New material: Steel Strength, fs = 1.25\*fsm = 781.25

Existing Column

Existing material: Steel Strength, fs = 1.25\*fsm = 625.00

#####

Max Height, Hmax = 400.00  
Min Height, Hmin = 200.00  
Max Width, Wmax = 400.00  
Min Width, Wmin = 200.00  
Jacket Thickness, tj = 50.00  
Cover Thickness, c = 15.00  
Mean Confinement Factor overall section = 1.00  
Element Length, L = 3000.00  
Secondary Member  
Smooth Bars  
Ductile Steel  
Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Adequate Lap Length ( $l_o/l_{ou, \min} \geq 1$ )  
No FRP Wrapping

-----  
Stepwise Properties  
-----

At local axis: 2  
EDGE -A-  
Shear Force, Va = -3.6347847E-006  
EDGE -B-  
Shear Force, Vb = 3.6347847E-006  
BOTH EDGES  
Axial Force, F = -1.8026E+006  
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension: Aslt = 0.00  
-Compression: Asc = 5353.274  
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension: Asl,ten = 1137.257  
-Compression: Asl,com = 2208.54  
-Middle: Asl,mid = 2007.478

-----  
-----  
Calculation of Shear Capacity ratio ,  $V_e/V_r = 1.18753$   
Member Controlled by Shear ( $V_e/V_r > 1$ )  
Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 402750.026$   
with  
 $M_{pr1} = \text{Max}(M_{u1+} , M_{u1-}) = 6.0413E+008$   
 $M_{u1+} = 6.0413E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction  
which is defined for the static loading combination  
 $M_{u1-} = 5.4612E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment  
direction which is defined for the static loading combination  
 $M_{pr2} = \text{Max}(M_{u2+} , M_{u2-}) = 6.0413E+008$   
 $M_{u2+} = 6.0413E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction  
which is defined for the the static loading combination  
 $M_{u2-} = 5.4612E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment  
direction which is defined for the the static loading combination

-----  
Calculation of  $M_{u1+}$   
-----

-----  
Calculation of ultimate curvature  $\phi_u$  according to 4.1, Biskinis/Fardis 2013:  
 $\phi_u = 1.2613948E-005$   
 $M_u = 6.0413E+008$

-----  
with full section properties:  
b = 400.00  
d = 367.00  
d' = 33.00  
v = 0.40931126

$$N = 1.8026E+006$$

$$f_c = 30.00$$

$$c_o (5A.5, TBDY) = 0.002$$

$$\text{Final value of } c_u: c_u^* = \text{shear\_factor} * \text{Max}(c_u, c_c) = 0.00925707$$

The Shear\_factor is considered equal to 1 (pure moment strength)

$$\text{From (5.4b), TBDY: } c_u = 0.00925707$$

$$w_e (5.4c) = 0.0207149$$

$$a_{se} ((5.4d), TBDY) = (a_{se1} * A_{ext} + a_{se2} * A_{int}) / A_{sec} = 0.33836562$$

$$a_{se1} = \text{Max}(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$$

The definitions of  $A_{noConf1}$ ,  $A_{conf,min1}$  and  $A_{conf,max1}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$a_{se2} (>= a_{se1}) = \text{Max}(((A_{conf,max2} - A_{noConf2}) / A_{conf,max2}) * (A_{conf,min2} / A_{conf,max2}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe}, p_{sh,y} * F_{ywe}) = 6.12205$$

Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

$$p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$$

$$p_{sh1} ((5.4d), TBDY) = L_{stir1} * A_{stir1} / (A_{sec} * s_1) = 0.00680678$$

$$L_{stir1} (\text{Length of stirrups along Y}) = 1040.00$$

$$A_{stir1} (\text{stirrups area}) = 78.53982$$

$$p_{sh2} (5.4d) = L_{stir2} * A_{stir2} / (A_{sec} * s_2) = 0.0012868$$

$$L_{stir2} (\text{Length of stirrups along Y}) = 768.00$$

$$A_{stir2} (\text{stirrups area}) = 50.26548$$

$$p_{sh,y} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$$

$$p_{sh1} ((5.4d), TBDY) = L_{stir1} * A_{stir1} / (A_{sec} * s_1) = 0.00680678$$

$$L_{stir1} (\text{Length of stirrups along X}) = 1040.00$$

$$A_{stir1} (\text{stirrups area}) = 78.53982$$

$$p_{sh2} ((5.4d), TBDY) = L_{stir2} * A_{stir2} / (A_{sec} * s_2) = 0.0012868$$

$$L_{stir2} (\text{Length of stirrups along X}) = 768.00$$

$$A_{stir2} (\text{stirrups area}) = 50.26548$$

$$A_{sec} = 120000.00$$

$$s_1 = 100.00$$

$$s_2 = 250.00$$

$$f_{ywe1} = 781.25$$

$$f_{ywe2} = 625.00$$

$$f_{ce} = 30.00$$

$$\text{From ((5.A5), TBDY), TBDY: } c_c = 0.002$$

$$c = \text{confinement factor} = 1.00$$

$$y_1 = 0.0025$$

$$sh_1 = 0.008$$

$$ft_1 = 886.7403$$

$$fy_1 = 738.9503$$

$$su_1 = 0.032$$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with

Shear\_factor = 1.00

lo/lou,min = lb/d = 1.00

su1 = 0.4\*esu1\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1,ft1,fy1, it is considered characteristic value fsy1 = fs1/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs1 = (fs,jacket\*Asl,ten,jacket + fs,core\*Asl,ten,core)/Asl,ten = 738.9503

with Es1 = (Es,jacket\*Asl,ten,jacket + Es,core\*Asl,ten,core)/Asl,ten = 200000.00

y2 = 0.0025

sh2 = 0.008

ft2 = 898.293

fy2 = 748.5775

su2 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor

and also multiplied by the shear\_factor according to 15.7.1.4, with

Shear\_factor = 1.00

lo/lou,min = lb/lb,min = 1.00

su2 = 0.4\*esu2\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu2\_nominal = 0.08,

For calculation of esu2\_nominal and y2, sh2,ft2,fy2, it is considered characteristic value fsy2 = fs2/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs2 = (fs,jacket\*Asl,com,jacket + fs,core\*Asl,com,core)/Asl,com = 748.5775

with Es2 = (Es,jacket\*Asl,com,jacket + Es,core\*Asl,com,core)/Asl,com = 200000.00

yv = 0.0025

shv = 0.008

ftv = 894.3662

fyv = 745.3052

suv = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor

and also multiplied by the shear\_factor according to 15.7.1.4, with

Shear\_factor = 1.00

lo/lou,min = lb/d = 1.00

suv = 0.4\*esuv\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esuv\_nominal = 0.08,

considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY

For calculation of esuv\_nominal and yv, shv,ftv,fyv, it is considered characteristic value fsyv = fsv/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fsv = (fs,jacket\*Asl,mid,jacket + fs,mid\*Asl,mid,core)/Asl,mid = 745.3052

with Esv = (Es,jacket\*Asl,mid,jacket + Es,mid\*Asl,mid,core)/Asl,mid = 200000.00

1 = Asl,ten/(b\*d)\*(fs1/fc) = 0.19082108

2 = Asl,com/(b\*d)\*(fs2/fc) = 0.37540035

v = Asl,mid/(b\*d)\*(fsv/fc) = 0.33973286

and confined core properties:

b = 360.00

d = 347.00

d' = 13.00

fcc (5A.2, TBDY) = 30.00

cc (5A.5, TBDY) = 0.002

c = confinement factor = 1.00

1 = Asl,ten/(b\*d)\*(fs1/fc) = 0.22424379

2 = Asl,com/(b\*d)\*(fs2/fc) = 0.44115251

v = Asl,mid/(b\*d)\*(fsv/fc) = 0.39923778

Case/Assumption: Unconfined full section - Steel rupture

' satisfies Eq. (4.3)

--->

v < vs,y2 - LHS eq.(4.5) is not satisfied

--->

v < vs,c - RHS eq.(4.5) is not satisfied

--->

Case/Assumption Rejected.

--->

New Case/Assumption: Unconfined full section - Spalling of concrete cover

' satisfies Eq. (4.4)

--->

v < s,y1 - LHS eq.(4.7) is not satisfied

---->

v < vc,y1 - RHS eq.(4.6) is satisfied

---->

cu (4.10) = 0.37716748

MRC (4.17) = 7.0585E+008

---->

New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover

In expressions below, the following modifications have been made

- b, d, d' replaced by geometric parameters of the core: bo, do, d'o
- N, 1, 2, v normalised to bo\*do, instead of b\*d
- - parameters of confined concrete, fcc, cc, used in lieu of fc, ecu

---->

Subcase: Rupture of tension steel

---->

v\* < v\*s,y2 - LHS eq.(4.5) is not satisfied

---->

v\* < v\*s,c - LHS eq.(4.5) is not satisfied

---->

Subcase rejected

---->

New Subcase: Failure of compression zone

---->

v\* < v\*c,y2 - LHS eq.(4.6) is not satisfied

---->

v\* < v\*c,y1 - RHS eq.(4.6) is not satisfied

---->

\*cu (4.11) = 0.4569298

MRO (4.18) = 6.0413E+008

---->

u = cu (4.2) = 1.2613948E-005

Mu = MRO

-----  
Calculation of ratio lb/d

-----  
Adequate Lap Length: lb/d >= 1  
-----  
-----  
-----

-----  
Calculation of Mu1-  
-----  
-----  
-----

-----  
Calculation of ultimate curvature u according to 4.1, Biskinis/Fardis 2013:

u = 3.1475172E-005

Mu = 5.4612E+008  
-----

with full section properties:

b = 200.00

d = 367.00

d' = 33.00

v = 0.81862252

N = 1.8026E+006

fc = 30.00

co (5A.5, TBDY) = 0.002

Final value of cu: cu\* = shear\_factor \* Max( cu, cc) = 0.00925707

The Shear\_factor is considered equal to 1 (pure moment strength)

From (5.4b), TBDY: cu = 0.00925707

we (5.4c) = 0.0207149

ase ((5.4d), TBDY) = (ase1\*Aext+ase2\*Aint)/Asec = 0.33836562

ase1 = Max(((Aconf,max1-AnoConf1)/Aconf,max1)\*(Aconf,min1/Aconf,max1),0) = 0.33836562

The definitions of AnoConf, Aconf,min and Aconf,max are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

Aconf,max1 = 89600.00 is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

Aconf,min1 = 59225.00 is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area Aconf,max1 by a length equal to half the clear spacing between external hoops.

AnoConf1 = 43733.333 is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).  
ase2 (>=ase1) =  $\text{Max}(((Aconf,max2 - AnoConf2)/Aconf,max2) * (Aconf,min2/Aconf,max2), 0) = 0.33836562$

The definitions of AnoConf, Aconf,min and Aconf,max are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988) "Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

Aconf,max2 = 45264.00 is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

Aconf,min2 = 0.00 is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area Aconf,max2 by a length equal to half the clear spacing between internal hoops.

AnoConf2 = 32309.333 is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

psh,min\*Fywe =  $\text{Min}(psh,x*Fywe, psh,y*Fywe) = 6.12205$

Expression (5.4d) for psh,min\*Fywe has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

-----  
psh\_x\*Fywe = psh1\*Fywe1 + ps2\*Fywe2 = 6.12205

psh1 ((5.4d), TBDY) =  $Lstir1 * Astir1 / (Asec * s1) = 0.00680678$

Lstir1 (Length of stirrups along Y) = 1040.00

Astir1 (stirrups area) = 78.53982

psh2 (5.4d) =  $Lstir2 * Astir2 / (Asec * s2) = 0.0012868$

Lstir2 (Length of stirrups along Y) = 768.00

Astir2 (stirrups area) = 50.26548

-----  
psh\_y\*Fywe = psh1\*Fywe1 + ps2\*Fywe2 = 6.12205

psh1 ((5.4d), TBDY) =  $Lstir1 * Astir1 / (Asec * s1) = 0.00680678$

Lstir1 (Length of stirrups along X) = 1040.00

Astir1 (stirrups area) = 78.53982

psh2 ((5.4d), TBDY) =  $Lstir2 * Astir2 / (Asec * s2) = 0.0012868$

Lstir2 (Length of stirrups along X) = 768.00

Astir2 (stirrups area) = 50.26548

-----  
Asec = 120000.00

s1 = 100.00

s2 = 250.00

fywe1 = 781.25

fywe2 = 625.00

fce = 30.00

From ((5.A5), TBDY), TBDY: cc = 0.002

c = confinement factor = 1.00

y1 = 0.0025

sh1 = 0.008

ft1 = 898.293

fy1 = 748.5775

su1 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

lo/lou,min = lb/ld = 1.00

su1 =  $0.4 * esu1\_nominal$  ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1, ft1, fy1, it is considered characteristic value fsy1 = fs1/1.2, from table 5.1, TBDY.

y1, sh1, ft1, fy1, are also multiplied by  $\text{Min}(1, 1.25 * (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs1 =  $(fs\_jacket * Asl\_ten\_jacket + fs\_core * Asl\_ten\_core) / Asl\_ten = 748.5775$

with Es1 =  $(Es\_jacket * Asl\_ten\_jacket + Es\_core * Asl\_ten\_core) / Asl\_ten = 200000.00$

y2 = 0.0025

sh2 = 0.008

ft2 = 886.7403

fy2 = 738.9503

su2 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb,min = 1.00

su2 = 0.4\*esu2\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu2\_nominal = 0.08,

For calculation of esu2\_nominal and y2, sh2,ft2,fy2, it is considered  
characteristic value fsy2 = fs2/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs2 = (fs,jacket\*Asl,com,jacket + fs,core\*Asl,com,core)/Asl,com = 738.9503

with Es2 = (Es,jacket\*Asl,com,jacket + Es,core\*Asl,com,core)/Asl,com = 200000.00

yv = 0.0025

shv = 0.008

ftv = 894.3662

fyv = 745.3052

suv = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/ld = 1.00

suv = 0.4\*esuv\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esuv\_nominal = 0.08,

considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY

For calculation of esuv\_nominal and yv, shv,ftv,fyv, it is considered  
characteristic value fsyv = fsv/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fsv = (fs,jacket\*Asl,mid,jacket + fs,mid\*Asl,mid,core)/Asl,mid = 745.3052

with Esv = (Es,jacket\*Asl,mid,jacket + Es,mid\*Asl,mid,core)/Asl,mid = 200000.00

1 = Asl,ten/(b\*d)\*(fs1/fc) = 0.7508007

2 = Asl,com/(b\*d)\*(fs2/fc) = 0.38164216

v = Asl,mid/(b\*d)\*(fsv/fc) = 0.67946571

and confined core properties:

b = 160.00

d = 347.00

d' = 13.00

fcc (5A.2, TBDY) = 30.00

cc (5A.5, TBDY) = 0.002

c = confinement factor = 1.00

1 = Asl,ten/(b\*d)\*(fs1/fc) = 0.99259314

2 = Asl,com/(b\*d)\*(fs2/fc) = 0.50454853

v = Asl,mid/(b\*d)\*(fsv/fc) = 0.898285

Case/Assumption: Unconfined full section - Steel rupture

' satisfies Eq. (4.3)

---->

v < vs,y2 - LHS eq.(4.5) is not satisfied

---->

v < vs,c - RHS eq.(4.5) is not satisfied

---->

Case/Assumption Rejected.

---->

New Case/Assumption: Unconfined full section - Spalling of concrete cover

' satisfies Eq. (4.4)

---->

v < vs,y1 - LHS eq.(4.7) is not satisfied

---->

v < vc,y1 - RHS eq.(4.6) is not satisfied

---->

cu (4.11) = 0.80138156

MRc (4.18) = 5.4612E+008

---->

New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover

In expressions below, the following modifications have been made

- b, d, d' replaced by geometric parameters of the core: bo, do, d'o

- N, 1, 2, v normalised to bo\*do, instead of b\*d

- parameters of confined concrete, fcc, cc, used in lieu of fc, ecu

---->

Subcase: Rupture of tension steel

--->  
v\* < v\*s,y2 - LHS eq.(4.5) is not satisfied

--->  
v\* < v\*s,c - LHS eq.(4.5) is not satisfied

--->  
Subcase rejected

--->  
New Subcase: Failure of compression zone

--->  
v\* < v\*c,y2 - LHS eq.(4.6) is not satisfied

--->  
v\* < v\*c,y1 - RHS eq.(4.6) is not satisfied

--->  
\*cu (4.11) = 0.80674817

MRO (4.18) = 2.8580E+008

MRO < 0.8\*MRc

--->  
u = cu (unconfined full section) = 3.1475172E-005  
Mu = MRc

-----  
Calculation of ratio lb/d

-----  
Adequate Lap Length: lb/d >= 1

-----  
Calculation of Mu2+

-----  
Calculation of ultimate curvature u according to 4.1, Biskinis/Fardis 2013:

u = 1.2613948E-005

Mu = 6.0413E+008

-----  
with full section properties:

b = 400.00

d = 367.00

d' = 33.00

v = 0.40931126

N = 1.8026E+006

fc = 30.00

co (5A.5, TBDY) = 0.002

Final value of cu: cu\* = shear\_factor \* Max( cu, cc) = 0.00925707

The Shear\_factor is considered equal to 1 (pure moment strength)

From (5.4b), TBDY: cu = 0.00925707

we (5.4c) = 0.0207149

ase ((5.4d), TBDY) = (ase1\*Aext+ase2\*Aint)/Asec = 0.33836562

ase1 = Max(((Aconf,max1-AnoConf1)/Aconf,max1)\*(Aconf,min1/Aconf,max1),0) = 0.33836562

The definitions of AnoConf, Aconf,min and Aconf,max are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

Aconf,max1 = 89600.00 is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

Aconf,min1 = 59225.00 is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area Aconf,max1 by a length

equal to half the clear spacing between external hoops.

AnoConf1 = 43733.333 is the unconfined external core area which is equal to bi<sup>2</sup>/6 as defined at (A.2).

ase2 (>=ase1) = Max(((Aconf,max2-AnoConf2)/Aconf,max2)\*(Aconf,min2/Aconf,max2),0) = 0.33836562

The definitions of AnoConf, Aconf,min and Aconf,max are derived from generalization

of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

Aconf,max2 = 45264.00 is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe}, p_{sh,y} * F_{ywe}) = 6.12205$

Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

-----  
 $p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{s2} * F_{ywe2} = 6.12205$

$p_{sh1}$  ((5.4d), TBDY) =  $L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$

$L_{stir1}$  (Length of stirrups along Y) = 1040.00

$A_{stir1}$  (stirrups area) = 78.53982

$p_{sh2}$  (5.4d) =  $L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$

$L_{stir2}$  (Length of stirrups along Y) = 768.00

$A_{stir2}$  (stirrups area) = 50.26548  
-----

$p_{sh,y} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{s2} * F_{ywe2} = 6.12205$

$p_{sh1}$  ((5.4d), TBDY) =  $L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$

$L_{stir1}$  (Length of stirrups along X) = 1040.00

$A_{stir1}$  (stirrups area) = 78.53982

$p_{sh2}$  ((5.4d), TBDY) =  $L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$

$L_{stir2}$  (Length of stirrups along X) = 768.00

$A_{stir2}$  (stirrups area) = 50.26548  
-----

$A_{sec} = 120000.00$

$s1 = 100.00$

$s2 = 250.00$

$f_{ywe1} = 781.25$

$f_{ywe2} = 625.00$

$f_{ce} = 30.00$

From ((5.A5), TBDY), TBDY:  $c_c = 0.002$

$c$  = confinement factor = 1.00

$y1 = 0.0025$

$sh1 = 0.008$

$ft1 = 886.7403$

$fy1 = 738.9503$

$su1 = 0.032$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

$l_o/l_{ou,min} = l_b/l_d = 1.00$

$su1 = 0.4 * esu1_{nominal}$  ((5.5), TBDY) = 0.032

From table 5A.1, TBDY:  $esu1_{nominal} = 0.08$ ,

For calculation of  $esu1_{nominal}$  and  $y1, sh1, ft1, fy1$ , it is considered characteristic value  $fsy1 = fs1/1.2$ , from table 5.1, TBDY.

$y1, sh1, ft1, fy1$ , are also multiplied by  $\text{Min}(1, 1.25 * (l_b/l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $fs1 = (f_{s,jacket} * A_{sl,ten,jacket} + f_{s,core} * A_{sl,ten,core}) / A_{sl,ten} = 738.9503$

with  $Es1 = (E_{s,jacket} * A_{sl,ten,jacket} + E_{s,core} * A_{sl,ten,core}) / A_{sl,ten} = 200000.00$

$y2 = 0.0025$

$sh2 = 0.008$

$ft2 = 898.293$

$fy2 = 748.5775$

$su2 = 0.032$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

$l_o/l_{ou,min} = l_b/l_{b,min} = 1.00$

$su2 = 0.4 * esu2_{nominal}$  ((5.5), TBDY) = 0.032

From table 5A.1, TBDY:  $esu2_{nominal} = 0.08$ ,

For calculation of  $esu2_{nominal}$  and  $y2, sh2, ft2, fy2$ , it is considered characteristic value  $fsy2 = fs2/1.2$ , from table 5.1, TBDY.

$y1, sh1, ft1, fy1$ , are also multiplied by  $\text{Min}(1, 1.25 * (l_b/l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $fs2 = (f_{s,jacket} * A_{sl,com,jacket} + f_{s,core} * A_{sl,com,core}) / A_{sl,com} = 748.5775$

with  $Es2 = (E_{s,jacket} * A_{sl,com,jacket} + E_{s,core} * A_{sl,com,core}) / A_{sl,com} = 200000.00$

$yv = 0.0025$

$shv = 0.008$

```

ftv = 894.3662
fyv = 745.3052
suv = 0.032
using (30) in Biskinis/Fardis (2013) multiplied with shear_factor
and also multiplied by the shear_factor according to 15.7.1.4, with
Shear_factor = 1.00
lo/lo,min = lb/d = 1.00
suv = 0.4*esuv_nominal ((5.5), TBDY) = 0.032
From table 5A.1, TBDY: esuv_nominal = 0.08,
considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY
For calculation of esuv_nominal and yv, shv,ftv,fyv, it is considered
characteristic value fsyv = fsv/1.2, from table 5.1, TBDY.
y1, sh1,ft1,fy1, are also multiplied by Min(1,1.25*(lb/d)^2/3), from 10.3.5, ASCE 41-17.
with fsv = (fs,jacket*Asl,mid,jacket + fs,mid*Asl,mid,core)/Asl,mid = 745.3052
with Esv = (Es,jacket*Asl,mid,jacket + Es,mid*Asl,mid,core)/Asl,mid = 200000.00
1 = Asl,ten/(b*d)*(fs1/fc) = 0.19082108
2 = Asl,com/(b*d)*(fs2/fc) = 0.37540035
v = Asl,mid/(b*d)*(fsv/fc) = 0.33973286
and confined core properties:
b = 360.00
d = 347.00
d' = 13.00
fcc (5A.2, TBDY) = 30.00
cc (5A.5, TBDY) = 0.002
c = confinement factor = 1.00
1 = Asl,ten/(b*d)*(fs1/fc) = 0.22424379
2 = Asl,com/(b*d)*(fs2/fc) = 0.44115251
v = Asl,mid/(b*d)*(fsv/fc) = 0.39923778
Case/Assumption: Unconfined full section - Steel rupture
'satisfies Eq. (4.3)
---->
v < vs,y2 - LHS eq.(4.5) is not satisfied
---->
v < vs,c - RHS eq.(4.5) is not satisfied
---->
Case/Assumption Rejected.
---->
New Case/Assumption: Unconfined full section - Spalling of concrete cover
'satisfies Eq. (4.4)
---->
v < s,y1 - LHS eq.(4.7) is not satisfied
---->
v < vc,y1 - RHS eq.(4.6) is satisfied
---->
cu (4.10) = 0.37716748
MRc (4.17) = 7.0585E+008
---->
New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover
In expressions below, the following modifications have been made
- b, d, d' replaced by geometric parameters of the core: bo, do, d'o
- N, 1, 2, v normalised to bo*do, instead of b*d
- - parameters of confined concrete, fcc, cc, used in lieu of fc, ecu
---->
Subcase: Rupture of tension steel
---->
v* < v*s,y2 - LHS eq.(4.5) is not satisfied
---->
v* < v*s,c - LHS eq.(4.5) is not satisfied
---->
Subcase rejected
---->
New Subcase: Failure of compression zone
---->
v* < v*c,y2 - LHS eq.(4.6) is not satisfied
---->
v* < v*c,y1 - RHS eq.(4.6) is not satisfied
---->

```

$$*c_u(4.11) = 0.4569298$$
$$M_{Ro}(4.18) = 6.0413E+008$$

--->

$$u = c_u(4.2) = 1.2613948E-005$$
$$\mu = M_{Ro}$$

-----

Calculation of ratio  $l_b/l_d$

-----

Adequate Lap Length:  $l_b/l_d \geq 1$

-----

-----

Calculation of  $\mu_2$ -

-----

-----

Calculation of ultimate curvature  $u$  according to 4.1, Biskinis/Fardis 2013:

$$u = 3.1475172E-005$$
$$\mu = 5.4612E+008$$

-----

with full section properties:

$$b = 200.00$$
$$d = 367.00$$
$$d' = 33.00$$
$$v = 0.81862252$$
$$N = 1.8026E+006$$

$$f_c = 30.00$$

$$c_o(5A.5, TBDY) = 0.002$$

$$\text{Final value of } c_u: c_u^* = \text{shear\_factor} * \text{Max}(c_u, c_c) = 0.00925707$$

The Shear\_factor is considered equal to 1 (pure moment strength)

$$\text{From (5.4b), TBDY: } c_u = 0.00925707$$

$$w_e(5.4c) = 0.0207149$$

$$a_{se}((5.4d), TBDY) = (a_{se1} * A_{ext} + a_{se2} * A_{int}) / A_{sec} = 0.33836562$$

$$a_{se1} = \text{Max}(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$a_{se2} (\geq a_{se1}) = \text{Max}(((A_{conf,max2} - A_{noConf2}) / A_{conf,max2}) * (A_{conf,min2} / A_{conf,max2}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe}, p_{sh,y} * F_{ywe}) = 6.12205$$

Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

-----

$$p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$$

$$p_{sh1}((5.4d), TBDY) = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$$

$$L_{stir1}(\text{Length of stirrups along Y}) = 1040.00$$

$$A_{stir1}(\text{stirrups area}) = 78.53982$$

$$p_{sh2}(5.4d) = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$$

Lstir2 (Length of stirrups along Y) = 768.00  
Astir2 (stirrups area) = 50.26548

psh\_y\*Fywe = psh1\*Fywe1+ps2\*Fywe2 = 6.12205  
psh1 ((5.4d), TBDY) = Lstir1\*Astir1/(Asec\*s1) = 0.00680678  
Lstir1 (Length of stirrups along X) = 1040.00  
Astir1 (stirrups area) = 78.53982  
psh2 ((5.4d), TBDY) = Lstir2\*Astir2/(Asec\*s2) = 0.0012868  
Lstir2 (Length of stirrups along X) = 768.00  
Astir2 (stirrups area) = 50.26548

Asec = 120000.00  
s1 = 100.00  
s2 = 250.00

fywe1 = 781.25  
fywe2 = 625.00  
fce = 30.00

From ((5.A5), TBDY), TBDY: cc = 0.002  
c = confinement factor = 1.00

y1 = 0.0025  
sh1 = 0.008  
ft1 = 898.293  
fy1 = 748.5775  
su1 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/ld = 1.00

su1 = 0.4\*esu1\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1,ft1,fy1, it is considered  
characteristic value fsy1 = fs1/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by Min(1,1.25\*(lb/ld)^ 2/3), from 10.3.5, ASCE 41-17.

with fs1 = (fs,jacket\*Asl,ten,jacket + fs,core\*Asl,ten,core)/Asl,ten = 748.5775

with Es1 = (Es,jacket\*Asl,ten,jacket + Es,core\*Asl,ten,core)/Asl,ten = 200000.00

y2 = 0.0025  
sh2 = 0.008  
ft2 = 886.7403  
fy2 = 738.9503  
su2 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb,min = 1.00

su2 = 0.4\*esu2\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu2\_nominal = 0.08,

For calculation of esu2\_nominal and y2, sh2,ft2,fy2, it is considered  
characteristic value fsy2 = fs2/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by Min(1,1.25\*(lb/ld)^ 2/3), from 10.3.5, ASCE 41-17.

with fs2 = (fs,jacket\*Asl,com,jacket + fs,core\*Asl,com,core)/Asl,com = 738.9503

with Es2 = (Es,jacket\*Asl,com,jacket + Es,core\*Asl,com,core)/Asl,com = 200000.00

yv = 0.0025  
shv = 0.008  
ftv = 894.3662  
fyv = 745.3052  
suv = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/ld = 1.00

suv = 0.4\*esuv\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esuv\_nominal = 0.08,

considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY  
For calculation of esuv\_nominal and yv, shv,ftv,fyv, it is considered

characteristic value fsyv = fsv/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by Min(1,1.25\*(lb/ld)^ 2/3), from 10.3.5, ASCE 41-17.

with fsv = (fs,jacket\*Asl,mid,jacket + fs,mid\*Asl,mid,core)/Asl,mid = 745.3052

with  $E_{sv} = (E_{s,jacket} \cdot A_{sl,mid,jacket} + E_{s,mid} \cdot A_{sl,mid,core}) / A_{sl,mid} = 200000.00$

1 =  $A_{sl,ten} / (b \cdot d) \cdot (f_{s1} / f_c) = 0.7508007$

2 =  $A_{sl,com} / (b \cdot d) \cdot (f_{s2} / f_c) = 0.38164216$

v =  $A_{sl,mid} / (b \cdot d) \cdot (f_{sv} / f_c) = 0.67946571$

and confined core properties:

b = 160.00

d = 347.00

d' = 13.00

f<sub>cc</sub> (5A.2, TBDY) = 30.00

cc (5A.5, TBDY) = 0.002

c = confinement factor = 1.00

1 =  $A_{sl,ten} / (b \cdot d) \cdot (f_{s1} / f_c) = 0.99259314$

2 =  $A_{sl,com} / (b \cdot d) \cdot (f_{s2} / f_c) = 0.50454853$

v =  $A_{sl,mid} / (b \cdot d) \cdot (f_{sv} / f_c) = 0.898285$

Case/Assumption: Unconfined full section - Steel rupture

' satisfies Eq. (4.3)

--->

v < v<sub>s,y2</sub> - LHS eq.(4.5) is not satisfied

--->

v < v<sub>s,c</sub> - RHS eq.(4.5) is not satisfied

--->

Case/Assumption Rejected.

--->

New Case/Assumption: Unconfined full section - Spalling of concrete cover

' satisfies Eq. (4.4)

--->

v < s<sub>y1</sub> - LHS eq.(4.7) is not satisfied

--->

v < v<sub>c,y1</sub> - RHS eq.(4.6) is not satisfied

--->

cu (4.11) = 0.80138156

M<sub>Rc</sub> (4.18) = 5.4612E+008

--->

New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover

In expressions below, the following modifications have been made

- b, d, d' replaced by geometric parameters of the core: b<sub>o</sub>, d<sub>o</sub>, d'<sub>o</sub>
- N, 1, 2, v normalised to b<sub>o</sub>·d<sub>o</sub>, instead of b·d
- parameters of confined concrete, f<sub>cc</sub>, cc, used in lieu of f<sub>c</sub>, e<sub>c</sub>

--->

Subcase: Rupture of tension steel

--->

v\* < v\*<sub>s,y2</sub> - LHS eq.(4.5) is not satisfied

--->

v\* < v\*<sub>s,c</sub> - LHS eq.(4.5) is not satisfied

--->

Subcase rejected

--->

New Subcase: Failure of compression zone

--->

v\* < v\*<sub>c,y2</sub> - LHS eq.(4.6) is not satisfied

--->

v\* < v\*<sub>c,y1</sub> - RHS eq.(4.6) is not satisfied

--->

\*cu (4.11) = 0.80674817

M<sub>Ro</sub> (4.18) = 2.8580E+008

M<sub>Ro</sub> < 0.8·M<sub>Rc</sub>

--->

u = cu (unconfined full section) = 3.1475172E-005

Mu = M<sub>Rc</sub>

-----

Calculation of ratio lb/d

-----

Adequate Lap Length: lb/d >= 1

-----

-----

-----

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 339150.739$

Calculation of Shear Strength at edge 1,  $V_{r1} = 339150.739$

$V_{r1} = V_{\text{Col}} ((10.3), \text{ASCE } 41-17) = k_{nl} * V_{\text{Col}0}$

$V_{\text{Col}0} = 339150.739$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

Mean concrete strength:  $f_c' = (f_c'_{\text{jacket}} * \text{Area}_{\text{jacket}} + f_c'_{\text{core}} * \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 25.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 4.00$

$\mu_u = 6.7415E+007$

$\nu_u = 3.6347847E-006$

$d = 0.8 * h = 320.00$

$N_u = 1.8026E+006$

$A_g = 80000.00$

From (11.5.4.8), ACI 318-14:  $V_s = V_{s,\text{jacket}} + V_{s,\text{core}} = 471238.898$

where:

$V_{s,\text{jacket}} = V_{s,j1} + V_{s,j2} = 471238.898$

$V_{s,j1} = 157079.633$  is calculated for section web jacket, with:

$d = 160.00$

$A_v = 157079.633$

$f_y = 625.00$

$s = 100.00$

$V_{s,j1}$  is multiplied by  $\text{Col},j1 = 1.00$

$s/d = 0.625$

$V_{s,j2} = 314159.265$  is calculated for section flange jacket, with:

$d = 320.00$

$A_v = 157079.633$

$f_y = 625.00$

$s = 100.00$

$V_{s,j2}$  is multiplied by  $\text{Col},j2 = 1.00$

$s/d = 0.3125$

$V_{s,\text{core}} = V_{s,c1} + V_{s,c2} = 0.00$

$V_{s,c1} = 0.00$  is calculated for section web core, with:

$d = 80.00$

$A_v = 100530.965$

$f_y = 500.00$

$s = 250.00$

$V_{s,c1}$  is multiplied by  $\text{Col},c1 = 0.00$

$s/d = 3.125$

$V_{s,c2} = 0.00$  is calculated for section flange core, with:

$d = 240.00$

$A_v = 100530.965$

$f_y = 500.00$

$s = 250.00$

$V_{s,c2}$  is multiplied by  $\text{Col},c2 = 0.00$

$s/d = 1.04167$

$V_f ((11-3)-(11.4), \text{ACI } 440) = 0.00$

From (11-11), ACI 440:  $V_s + V_f \leq 212577.225$

$b_w = 200.00$

Calculation of Shear Strength at edge 2,  $V_{r2} = 339150.739$

$V_{r2} = V_{\text{Col}} ((10.3), \text{ASCE } 41-17) = k_{nl} * V_{\text{Col}0}$

$V_{\text{Col}0} = 339150.739$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

Mean concrete strength:  $f_c' = (f_c'_{\text{jacket}} * \text{Area}_{\text{jacket}} + f_c'_{\text{core}} * \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 25.00$ , but  $f_c'^{0.5} \leq 8.3$

MPa (22.5.3.1, ACI 318-14)

$$M/Vd = 4.00$$

$$\mu_u = 5.3429E+007$$

$$V_u = 3.6347847E-006$$

$$d = 0.8 \cdot h = 320.00$$

$$N_u = 1.8026E+006$$

$$A_g = 80000.00$$

$$\text{From (11.5.4.8), ACI 318-14: } V_s = V_{sjacket} + V_{s,core} = 471238.898$$

where:

$$V_{sjacket} = V_{sj1} + V_{sj2} = 471238.898$$

$V_{sj1} = 157079.633$  is calculated for section web jacket, with:

$$d = 160.00$$

$$A_v = 157079.633$$

$$f_y = 625.00$$

$$s = 100.00$$

$V_{sj1}$  is multiplied by  $Col,j1 = 1.00$

$$s/d = 0.625$$

$V_{sj2} = 314159.265$  is calculated for section flange jacket, with:

$$d = 320.00$$

$$A_v = 157079.633$$

$$f_y = 625.00$$

$$s = 100.00$$

$V_{sj2}$  is multiplied by  $Col,j2 = 1.00$

$$s/d = 0.3125$$

$$V_{s,core} = V_{s,c1} + V_{s,c2} = 0.00$$

$V_{s,c1} = 0.00$  is calculated for section web core, with:

$$d = 80.00$$

$$A_v = 100530.965$$

$$f_y = 500.00$$

$$s = 250.00$$

$V_{s,c1}$  is multiplied by  $Col,c1 = 0.00$

$$s/d = 3.125$$

$V_{s,c2} = 0.00$  is calculated for section flange core, with:

$$d = 240.00$$

$$A_v = 100530.965$$

$$f_y = 500.00$$

$$s = 250.00$$

$V_{s,c2}$  is multiplied by  $Col,c2 = 0.00$

$$s/d = 1.04167$$

$$V_f \text{ ((11-3)-(11.4), ACI 440)} = 0.00$$

$$\text{From (11-11), ACI 440: } V_s + V_f \leq 212577.225$$

$$b_w = 200.00$$

-----  
End Of Calculation of Shear Capacity ratio for element: column JLC1 of floor 1

At local axis: 2  
-----

Start Of Calculation of Chord Rotation Capacity for element: column JLC1 of floor 1

At local axis: 2

Integration Section: (a)

Section Type: rcjlc

Constant Properties  
-----

Knowledge Factor,  $\phi = 0.90$

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE 41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Jacket

New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 30.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 625.00$

Concrete Elasticity,  $E_c = 25742.96$

Steel Elasticity,  $E_s = 200000.00$

Existing Column

Existing material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 18.00$   
 Existing material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 500.00$   
 Concrete Elasticity,  $E_c = 19940.411$   
 Steel Elasticity,  $E_s = 200000.00$   
 Max Height,  $H_{max} = 400.00$   
 Min Height,  $H_{min} = 200.00$   
 Max Width,  $W_{max} = 400.00$   
 Min Width,  $W_{min} = 200.00$   
 Jacket Thickness,  $t_j = 50.00$   
 Cover Thickness,  $c = 15.00$   
 Element Length,  $L = 3000.00$   
 Secondary Member  
 Smooth Bars  
 Ductile Steel  
 Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Adequate Lap Length ( $l_b/l_d \geq 1$ )  
 No FRP Wrapping

-----  
 Stepwise Properties  
 -----

Bending Moment,  $M = 1.1948E+007$   
 Shear Force,  $V_2 = -62256.122$   
 Shear Force,  $V_3 = -174.4443$   
 Axial Force,  $F = -1.8013E+006$   
 Longitudinal Reinforcement Area Distribution (in 2 divisions)  
 -Tension:  $A_{sl,t} = 2161.416$   
 -Compression:  $A_{sl,c} = 3191.858$   
 Longitudinal Reinforcement Area Distribution (in 3 divisions)  
 -Tension:  $A_{sl,ten} = 1137.257$   
 -Compression:  $A_{sl,com} = 2208.54$   
 -Middle:  $A_{sl,mid} = 2007.478$   
 Longitudinal External Reinforcement Area Distribution (in 3 divisions)  
 -Tension:  $A_{sl,ten,jacket} = 829.3805$   
 -Compression:  $A_{sl,com,jacket} = 1746.726$   
 -Middle:  $A_{sl,mid,jacket} = 1545.664$   
 Longitudinal Internal Reinforcement Area Distribution (in 3 divisions)  
 -Tension:  $A_{sl,ten,core} = 307.8761$   
 -Compression:  $A_{sl,com,core} = 461.8141$   
 -Middle:  $A_{sl,mid,core} = 461.8141$   
 Mean Diameter of Tension Reinforcement,  $DbL = 16.80$

-----  
 Existing component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_{R,u} = u = 0.0380477$   
 $u = y + p = 0.04227522$

-----  
 - Calculation of  $y$  -  
 -----

$y = (M_y * L_s / 3) / E_{eff} = 0.02957008$  ((4.29), Biskinis Phd)  
 $M_y = 3.6891E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 6000.00  
 From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 2.4951E+013$   
 $factor = 0.70$   
 $A_g = 120000.00$   
 Mean concrete strength:  $f'_c = (f'_{c,jacket} * Area_{jacket} + f'_{c,core} * Area_{core}) / Area_{section} = 25.00$   
 $N = 1.8013E+006$   
 $E_c * I_g = E_{c,jacket} * I_{g,jacket} + E_{c,core} * I_{g,core} = 3.5645E+013$

-----  
 Calculation of Yielding Moment  $M_y$   
 -----

Calculation of  $y$  and  $M_y$  according to Annex 7 -

```

y = Min( y_ten, y_com)
y_ten = 1.6454016E-005
with fy = 596.2441
d = 367.00
y = 0.50630795
A = 0.11409279
B = 0.07426396
with pt = 0.01549396
pc = 0.0300891
pv = 0.02734983
N = 1.8013E+006
b = 200.00
" = 0.08991826
y_comp = 9.3125388E-006
with fc = 30.00
Ec = 25742.96
y = 0.61376358
A = 0.01443594
B = 0.03310406
with Es = 200000.00

```

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

- Calculation of  $p$  -

From table 10-8:  $p = 0.01270514$

with:

- Columns not controlled by inadequate development or splicing along the clear height because  $l_b/d \geq 1$

shear control ratio  $V_y E / V_{col} E = 1.44151$

$d = d_{external} = 367.00$

$s = s_{external} = 100.00$

-  $t = s_1 + s_2 + 2 * t_f / b_w * (f_{fe} / f_s) = 0.00809358$

jacket:  $s_1 = A_{v1} * L_{stir1} / (s_1 * A_g) = 0.00680678$

$A_{v1} = 78.53982$ , is the area of every stirrup parallel to loading (shear) direction

$L_{stir1} = 1040.00$ , is the total Length of all stirrups parallel to loading (shear) direction

$s_1 = 100.00$

core:  $s_2 = A_{v2} * L_{stir2} / (s_2 * A_g) = 0.0012868$

$A_{v2} = 50.26548$ , is the area of every stirrup parallel to loading (shear) direction

$L_{stir2} = 768.00$ , is the total Length of all stirrups parallel to loading (shear) direction

$s_2 = 250.00$

The term  $2 * t_f / b_w * (f_{fe} / f_s)$  is implemented to account for FRP contribution

where  $f = 2 * t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength

All these variables have already been given in Shear control ratio calculation.

For the normalisation  $f_s$  of jacket is used.

$N_{UD} = 1.8013E+006$

$A_g = 120000.00$

$f_{cE} = (f_{c\_jacket} * Area\_jacket + f_{c\_core} * Area\_core) / section\_area = 25.00$

$f_{yE} = (f_{y\_ext\_Long\_Reinf} * Area\_ext\_Long\_Reinf + f_{y\_int\_Long\_Reinf} * Area\_int\_Long\_Reinf) / Area\_Tot\_Long\_Rein =$

596.2441

$f_{yE} = (f_{y\_ext\_Trans\_Reinf} * s_1 + f_{y\_int\_Trans\_Reinf} * s_2) / (s_1 + s_2) = 605.1263$

$\rho_l = Area\_Tot\_Long\_Rein / (b * d) = 0.07293289$

$b = 200.00$

$d = 367.00$

$f_{cE} = 25.00$

End Of Calculation of Chord Rotation Capacity for element: column JLC1 of floor 1

At local axis: 2

Integration Section: (a)

## Calculation No. 11

column C1, Floor 1

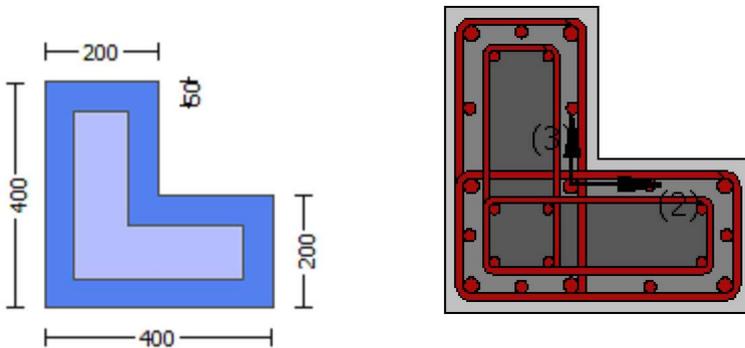
Limit State: Life Safety (data interpolation between analysis steps 50 and 51)

Analysis: Uniform +X

Check: Shear capacity VRd

Edge: Start

Local Axis: (3)



Start Of Calculation of Shear Capacity for element: column JLC1 of floor 1

At local axis: 3

Integration Section: (a)

Section Type: rcjlcs

Constant Properties

Knowledge Factor,  $\gamma = 0.90$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE 41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Jacket

New material of Secondary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 20.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 500.00$

Concrete Elasticity,  $E_c = 25742.96$

Steel Elasticity,  $E_s = 200000.00$

Existing Column

Existing material of Secondary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 12.00$

Existing material of Secondary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 400.00$

Concrete Elasticity,  $E_c = 19940.411$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of  $\gamma$  for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE 41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE 41-17).

Jacket

New material: Concrete Strength,  $f_c = f_{cm} = 30.00$

New material: Steel Strength,  $f_s = f_{sm} = 625.00$

Existing Column

Existing material: Concrete Strength,  $f_c = f_{cm} = 18.00$

Existing material: Steel Strength,  $f_s = f_{sm} = 500.00$

#####

Max Height,  $H_{max} = 400.00$

Min Height,  $H_{min} = 200.00$

Max Width,  $W_{max} = 400.00$

Min Width,  $W_{min} = 200.00$

Jacket Thickness,  $t_j = 50.00$

Cover Thickness,  $c = 15.00$

Element Length,  $L = 3000.00$

Secondary Member

Smooth Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou,min} = l_b/l_d \geq 1$ )

No FRP Wrapping

-----  
Stepwise Properties

EDGE -A-

Bending Moment,  $M_a = 1.1948E+007$

Shear Force,  $V_a = -174.4443$

EDGE -B-

Bending Moment,  $M_b = 246078.073$

Shear Force,  $V_b = 174.4443$

BOTH EDGES

Axial Force,  $F = -1.8013E+006$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{st} = 2161.416$

-Compression:  $A_{sc} = 3191.858$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{st,ten} = 1137.257$

-Compression:  $A_{st,com} = 2208.54$

-Middle:  $A_{st,mid} = 2007.478$

Mean Diameter of Tension Reinforcement,  $Db_{L,ten} = 16.80$

-----  
Existing component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = \phi V_n = 258166.022$

$V_n$  (10.3), ASCE 41-17) =  $k_n \phi V_{CoIO} = 286851.135$

$V_{CoI} = 286851.135$

$k_n = 1.00$

displacement\_ductility\_demand = 0.07294981

-----  
NOTE: In expression (10-3) ' $V_s = A_v \phi_f y d / s$ ' is replaced by ' $V_s + \phi_f V_f$ '  
where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

-----  
= 1 (normal-weight concrete)

Mean concrete strength:  $f'_c = (f'_{c,jacket} \cdot Area_{jacket} + f'_{c,core} \cdot Area_{core}) / Area_{section} = 16.66667$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 4.00$

$M_u = 1.1948E+007$

$V_u = 174.4443$

$d = 0.8 \cdot h = 320.00$

$N_u = 1.8013E+006$

$A_g = 80000.00$

From (11.5.4.8), ACI 318-14:  $V_s = V_{sjacket} + V_{s,core} = 376991.118$

where:

$V_{sjacket} = V_{sj1} + V_{sj2} = 376991.118$

$V_{sj1} = 251327.412$  is calculated for section web jacket, with:

$d = 320.00$

$A_v = 157079.633$

$f_y = 500.00$

$s = 100.00$   
 $V_{s,j1}$  is multiplied by  $Col,j1 = 1.00$   
 $s/d = 0.3125$   
 $V_{s,j2} = 125663.706$  is calculated for section flange jacket, with:  
 $d = 160.00$   
 $A_v = 157079.633$   
 $f_y = 500.00$   
 $s = 100.00$   
 $V_{s,j2}$  is multiplied by  $Col,j2 = 1.00$   
 $s/d = 0.625$   
 $V_{s,core} = V_{s,c1} + V_{s,c2} = 0.00$   
 $V_{s,c1} = 0.00$  is calculated for section web core, with:  
 $d = 240.00$   
 $A_v = 100530.965$   
 $f_y = 400.00$   
 $s = 250.00$   
 $V_{s,c1}$  is multiplied by  $Col,c1 = 0.00$   
 $s/d = 1.04167$   
 $V_{s,c2} = 0.00$  is calculated for section flange core, with:  
 $d = 80.00$   
 $A_v = 100530.965$   
 $f_y = 400.00$   
 $s = 250.00$   
 $V_{s,c2}$  is multiplied by  $Col,c2 = 0.00$   
 $s/d = 3.125$   
 $V_f((11-3)-(11.4), ACI 440) = 0.00$   
 From (11-11), ACI 440:  $V_s + V_f \leq 173568.578$   
 $bw = 200.00$

-----  
 displacement ductility demand is calculated as  $\phi / y$

- Calculation of  $\phi / y$  for END A -  
 for rotation axis 2 and integ. section (a)

-----  
 From analysis, chord rotation  $\theta = 0.00215713$   
 $y = (M_y * L_s / 3) / E_{eff} = 0.02957008$  ((4.29), Biskinis Phd))  
 $M_y = 3.6891E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 6000.00  
 From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 2.4951E+013$   
 $factor = 0.70$   
 $A_g = 120000.00$   
 Mean concrete strength:  $f_c' = (f_c'_{jacket} * Area_{jacket} + f_c'_{core} * Area_{core}) / Area_{section} = 25.00$   
 $N = 1.8013E+006$   
 $E_c * I_g = E_c_{jacket} * I_{g,jacket} + E_c_{core} * I_{g,core} = 3.5645E+013$

-----  
 Calculation of Yielding Moment  $M_y$

-----  
 Calculation of  $\phi / y$  and  $M_y$  according to Annex 7 -

-----  
 $y = \text{Min}(y_{ten}, y_{com})$   
 $y_{ten} = 1.6454016E-005$   
 with  $f_y = 596.2441$   
 $d = 367.00$   
 $y = 0.50630795$   
 $A = 0.11409279$   
 $B = 0.07426396$   
 with  $pt = 0.01549396$   
 $pc = 0.0300891$   
 $pv = 0.02734983$   
 $N = 1.8013E+006$   
 $b = 200.00$   
 $\mu = 0.08991826$   
 $y_{comp} = 9.3125388E-006$

with  $f_c = 30.00$   
 $E_c = 25742.96$   
 $y = 0.61376358$   
 $A = 0.01443594$   
 $B = 0.03310406$   
with  $E_s = 200000.00$

Calculation of ratio  $I_b/I_d$

Adequate Lap Length:  $I_b/I_d \geq 1$

End Of Calculation of Shear Capacity for element: column JLC1 of floor 1  
At local axis: 3  
Integration Section: (a)

## Calculation No. 12

column C1, Floor 1

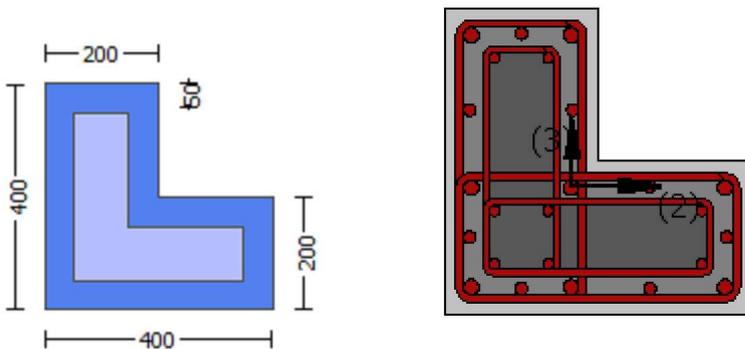
Limit State: Life Safety (data interpolation between analysis steps 50 and 51)

Analysis: Uniform +X

Check: Chord rotation capacity (  $\theta$  )

Edge: Start

Local Axis: (3)



Start Of Calculation of Shear Capacity ratio for element: column JLC1 of floor 1

At Shear local axis: 3  
(Bending local axis: 2)

Section Type: rcjcs

Constant Properties

Knowledge Factor,  $\gamma = 0.90$

Mean strength values are used for both shear and moment calculations.

Consequently:

Jacket

New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 30.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 625.00$   
 Concrete Elasticity,  $E_c = 25742.96$   
 Steel Elasticity,  $E_s = 200000.00$   
 Existing Column  
 Existing material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 18.00$   
 Existing material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 500.00$   
 Concrete Elasticity,  $E_c = 19940.411$   
 Steel Elasticity,  $E_s = 200000.00$   
 #####  
 Note: Especially for the calculation of moment strengths,  
 the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14  
 Jacket  
 New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 781.25$   
 Existing Column  
 Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 625.00$   
 #####  
 Max Height,  $H_{max} = 400.00$   
 Min Height,  $H_{min} = 200.00$   
 Max Width,  $W_{max} = 400.00$   
 Min Width,  $W_{min} = 200.00$   
 Jacket Thickness,  $t_j = 50.00$   
 Cover Thickness,  $c = 15.00$   
 Mean Confinement Factor overall section = 1.00  
 Element Length,  $L = 3000.00$   
 Secondary Member  
 Smooth Bars  
 Ductile Steel  
 Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Adequate Lap Length ( $l_o/l_{ou}, \min >= 1$ )  
 No FRP Wrapping  
 -----  
 Stepwise Properties  
 -----  
 At local axis: 3  
 EDGE -A-  
 Shear Force,  $V_a = 2.3476251E-006$   
 EDGE -B-  
 Shear Force,  $V_b = -2.3476251E-006$   
 BOTH EDGES  
 Axial Force,  $F = -1.8026E+006$   
 Longitudinal Reinforcement Area Distribution (in 2 divisions)  
   -Tension:  $A_{sl,t} = 0.00$   
   -Compression:  $A_{sl,c} = 5353.274$   
 Longitudinal Reinforcement Area Distribution (in 3 divisions)  
   -Tension:  $A_{sl,ten} = 2208.54$   
   -Compression:  $A_{sl,com} = 1137.257$   
   -Middle:  $A_{sl,mid} = 2007.478$   
 -----  
 -----  
 Calculation of Shear Capacity ratio,  $V_e/V_r = 1.44151$   
 Member Controlled by Shear ( $V_e/V_r > 1$ )  
 Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 488890.189$   
 with  
 $M_{pr1} = \text{Max}(Mu_{1+}, Mu_{1-}) = 7.3334E+008$   
 $Mu_{1+} = 7.3334E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction  
 which is defined for the static loading combination  
 $Mu_{1-} = 6.3768E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment  
 direction which is defined for the static loading combination  
 $M_{pr2} = \text{Max}(Mu_{2+}, Mu_{2-}) = 7.3334E+008$   
 $Mu_{2+} = 7.3334E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction  
 which is defined for the the static loading combination  
 $Mu_{2-} = 6.3768E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment  
 direction which is defined for the the static loading combination

-----  
Calculation of Mu1+  
-----

Calculation of ultimate curvature  $\mu$  according to 4.1, Biskinis/Fardis 2013:

$$\mu = 4.2207094E-005$$

$$\text{Mu} = 7.3334E+008$$

-----

with full section properties:

$$b = 400.00$$

$$d = 367.00$$

$$d' = 33.00$$

$$v = 0.40931126$$

$$N = 1.8026E+006$$

$$f_c = 30.00$$

$$c_o \text{ (5A.5, TBDY)} = 0.002$$

$$\text{Final value of } c_u: c_u^* = \text{shear\_factor} * \text{Max}(c_u, c_c) = 0.00925707$$

The Shear\_factor is considered equal to 1 (pure moment strength)

$$\text{From (5.4b), TBDY: } c_u = 0.00925707$$

$$w_e \text{ (5.4c)} = 0.0207149$$

$$a_{se} \text{ ((5.4d), TBDY)} = (a_{se1} * A_{ext} + a_{se2} * A_{int}) / A_{sec} = 0.33836562$$

$$a_{se1} = \text{Max}(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length

equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$a_{se2} (>= a_{se1}) = \text{Max}(((A_{conf,max2} - A_{noConf2}) / A_{conf,max2}) * (A_{conf,min2} / A_{conf,max2}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length

equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe}, p_{sh,y} * F_{ywe}) = 6.12205$$

Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

-----  
$$p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{s2} * F_{ywe2} = 6.12205$$

$$p_{sh1} \text{ ((5.4d), TBDY)} = L_{stir1} * A_{stir1} / (A_{sec} * s_1) = 0.00680678$$

$$L_{stir1} \text{ (Length of stirrups along Y)} = 1040.00$$

$$A_{stir1} \text{ (stirrups area)} = 78.53982$$

$$p_{sh2} \text{ (5.4d)} = L_{stir2} * A_{stir2} / (A_{sec} * s_2) = 0.0012868$$

$$L_{stir2} \text{ (Length of stirrups along Y)} = 768.00$$

$$A_{stir2} \text{ (stirrups area)} = 50.26548$$

-----

$$p_{sh,y} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{s2} * F_{ywe2} = 6.12205$$

$$p_{sh1} \text{ ((5.4d), TBDY)} = L_{stir1} * A_{stir1} / (A_{sec} * s_1) = 0.00680678$$

$$L_{stir1} \text{ (Length of stirrups along X)} = 1040.00$$

$$A_{stir1} \text{ (stirrups area)} = 78.53982$$

$$p_{sh2} \text{ ((5.4d), TBDY)} = L_{stir2} * A_{stir2} / (A_{sec} * s_2) = 0.0012868$$

$$L_{stir2} \text{ (Length of stirrups along X)} = 768.00$$

$$A_{stir2} \text{ (stirrups area)} = 50.26548$$

-----

$A_{sec} = 120000.00$   
 $s_1 = 100.00$   
 $s_2 = 250.00$   
 $fy_{we1} = 781.25$   
 $fy_{we2} = 625.00$   
 $f_{ce} = 30.00$   
 From ((5.A.5), TBDY), TBDY:  $cc = 0.002$   
 $c = \text{confinement factor} = 1.00$   
 $y_1 = 0.0025$   
 $sh_1 = 0.008$   
 $ft_1 = 898.293$   
 $fy_1 = 748.5775$   
 $su_1 = 0.032$   
 using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
 and also multiplied by the shear\_factor according to 15.7.1.4, with  
 $Shear\_factor = 1.00$   
 $lo/lou, min = lb/ld = 1.00$   
 $su_1 = 0.4 * esu_1\_nominal ((5.5), TBDY) = 0.032$   
 From table 5A.1, TBDY:  $esu_1\_nominal = 0.08$ ,  
 For calculation of  $esu_1\_nominal$  and  $y_1, sh_1, ft_1, fy_1$ , it is considered  
 characteristic value  $fsy_1 = fs_1/1.2$ , from table 5.1, TBDY.  
 $y_1, sh_1, ft_1, fy_1$ , are also multiplied by  $Min(1, 1.25 * (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.  
 with  $fs_1 = (fs_{jacket} * A_{sl, ten, jacket} + fs_{core} * A_{sl, ten, core}) / A_{sl, ten} = 748.5775$   
 with  $Es_1 = (Es_{jacket} * A_{sl, ten, jacket} + Es_{core} * A_{sl, ten, core}) / A_{sl, ten} = 200000.00$   
 $y_2 = 0.0025$   
 $sh_2 = 0.008$   
 $ft_2 = 886.7403$   
 $fy_2 = 738.9503$   
 $su_2 = 0.032$   
 using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
 and also multiplied by the shear\_factor according to 15.7.1.4, with  
 $Shear\_factor = 1.00$   
 $lo/lou, min = lb/lb, min = 1.00$   
 $su_2 = 0.4 * esu_2\_nominal ((5.5), TBDY) = 0.032$   
 From table 5A.1, TBDY:  $esu_2\_nominal = 0.08$ ,  
 For calculation of  $esu_2\_nominal$  and  $y_2, sh_2, ft_2, fy_2$ , it is considered  
 characteristic value  $fsy_2 = fs_2/1.2$ , from table 5.1, TBDY.  
 $y_1, sh_1, ft_1, fy_1$ , are also multiplied by  $Min(1, 1.25 * (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.  
 with  $fs_2 = (fs_{jacket} * A_{sl, com, jacket} + fs_{core} * A_{sl, com, core}) / A_{sl, com} = 738.9503$   
 with  $Es_2 = (Es_{jacket} * A_{sl, com, jacket} + Es_{core} * A_{sl, com, core}) / A_{sl, com} = 200000.00$   
 $y_v = 0.0025$   
 $sh_v = 0.008$   
 $ft_v = 894.3662$   
 $fy_v = 745.3052$   
 $suv = 0.032$   
 using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
 and also multiplied by the shear\_factor according to 15.7.1.4, with  
 $Shear\_factor = 1.00$   
 $lo/lou, min = lb/ld = 1.00$   
 $suv = 0.4 * esuv\_nominal ((5.5), TBDY) = 0.032$   
 From table 5A.1, TBDY:  $esuv\_nominal = 0.08$ ,  
 considering characteristic value  $fsyv = fsv/1.2$ , from table 5.1, TBDY  
 For calculation of  $esuv\_nominal$  and  $y_v, sh_v, ft_v, fy_v$ , it is considered  
 characteristic value  $fsyv = fsv/1.2$ , from table 5.1, TBDY.  
 $y_1, sh_1, ft_1, fy_1$ , are also multiplied by  $Min(1, 1.25 * (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.  
 with  $fsv = (fs_{jacket} * A_{sl, mid, jacket} + fs_{mid} * A_{sl, mid, core}) / A_{sl, mid} = 745.3052$   
 with  $Es_v = (Es_{jacket} * A_{sl, mid, jacket} + Es_{mid} * A_{sl, mid, core}) / A_{sl, mid} = 200000.00$   
 $1 = A_{sl, ten} / (b * d) * (fs_1 / f_c) = 0.37540035$   
 $2 = A_{sl, com} / (b * d) * (fs_2 / f_c) = 0.19082108$   
 $v = A_{sl, mid} / (b * d) * (fsv / f_c) = 0.33973286$   
 and confined core properties:  
 $b = 360.00$   
 $d = 347.00$   
 $d' = 13.00$   
 $f_{cc} (5A.2, TBDY) = 30.00$   
 $cc (5A.5, TBDY) = 0.002$   
 $c = \text{confinement factor} = 1.00$

$$1 = A_{sl,ten}/(b*d)*(f_{s1}/f_c) = 0.44115251$$

$$2 = A_{sl,com}/(b*d)*(f_{s2}/f_c) = 0.22424379$$

$$v = A_{sl,mid}/(b*d)*(f_{sv}/f_c) = 0.39923778$$

Case/Assumption: Unconfined full section - Steel rupture  
 satisfies Eq. (4.3)

---->  
 $v < v_{s,y2}$  - LHS eq.(4.5) is not satisfied  
 ---->  
 $v < v_{s,c}$  - RHS eq.(4.5) is not satisfied  
 ---->

Case/Assumption Rejected.

---->  
 New Case/Assumption: Unconfined full section - Spalling of concrete cover  
 satisfies Eq. (4.4)

---->  
 $v < v_{s,y1}$  - LHS eq.(4.7) is not satisfied  
 ---->  
 $v < v_{c,y1}$  - RHS eq.(4.6) is satisfied  
 ---->

$$c_u(4.10) = 0.59761571$$

$$M_{Rc}(4.17) = 7.3334E+008$$

---->  
 New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover

In expressions below, the following modifications have been made  
 - b, d, d' replaced by geometric parameters of the core:  $b_o, d_o, d'_o$   
 -  $N, 1, 2, v$  normalised to  $b_o*d_o$ , instead of  $b*d$   
 - - parameters of confined concrete,  $f_{cc}, c_c$ , used in lieu of  $f_c, c_u$

---->  
 Subcase: Rupture of tension steel

---->  
 $v^* < v^*_{s,y2}$  - LHS eq.(4.5) is not satisfied  
 ---->  
 $v^* < v^*_{s,c}$  - LHS eq.(4.5) is not satisfied  
 ---->

Subcase rejected

---->  
 New Subcase: Failure of compression zone

---->  
 $v^* < v^*_{c,y2}$  - LHS eq.(4.6) is not satisfied  
 ---->  
 $v^* < v^*_{c,y1}$  - RHS eq.(4.6) is not satisfied  
 ---->

$$^*c_u(4.11) = 0.62879536$$

$$M_{Ro}(4.18) = 4.5866E+008$$

$$M_{Ro} < 0.8*M_{Rc}$$

---->  
 $u = c_u$  (unconfined full section) =  $4.2207094E-005$   
 $\mu_u = M_{Rc}$

-----  
 Calculation of ratio  $l_b/d$   
 -----

Adequate Lap Length:  $l_b/d \geq 1$   
 -----  
 -----  
 -----

-----  
 Calculation of  $\mu_{u1}$ -  
 -----

-----  
 Calculation of ultimate curvature  $u$  according to 4.1, Biskinis/Fardis 2013:

$$u = 4.8346386E-005$$

$$\mu_u = 6.3768E+008$$

-----  
 with full section properties:

$$b = 200.00$$

$$d = 367.00$$

$d' = 33.00$   
 $v = 0.81862252$   
 $N = 1.8026E+006$   
 $f_c = 30.00$   
 $co (5A.5, TBDY) = 0.002$   
 Final value of  $cu$ :  $cu^* = shear\_factor * Max(cu, cc) = 0.00925707$   
 The Shear\_factor is considered equal to 1 (pure moment strength)  
 From (5.4b), TBDY:  $cu = 0.00925707$   
 $w_e (5.4c) = 0.0207149$   
 $ase ((5.4d), TBDY) = (ase1 * A_{ext} + ase2 * A_{int}) / A_{sec} = 0.33836562$   
 $ase1 = Max(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$   
 The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).  
 The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988) "Theoretical Stress-Strain Model for Confined Concrete."  
 J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.  
 $A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.  
 $A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length equal to half the clear spacing between external hoops.  
 $A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).  
 $ase2 (>= ase1) = Max(((A_{conf,max2} - A_{noConf2}) / A_{conf,max2}) * (A_{conf,min2} / A_{conf,max2}), 0) = 0.33836562$   
 The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).  
 The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988) "Theoretical Stress-Strain Model for Confined Concrete."  
 J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.  
 $A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.  
 $A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.  
 $A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).  
 $psh_{min} * F_{ywe} = Min(psh_x * F_{ywe}, psh_y * F_{ywe}) = 6.12205$   
 Expression (5.4d) for  $psh_{min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

$psh_x * F_{ywe} = psh1 * F_{ywe1} + ps2 * F_{ywe2} = 6.12205$   
 $psh1 ((5.4d), TBDY) = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$   
 $L_{stir1}$  (Length of stirrups along Y) = 1040.00  
 $A_{stir1}$  (stirrups area) = 78.53982  
 $psh2 (5.4d) = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$   
 $L_{stir2}$  (Length of stirrups along Y) = 768.00  
 $A_{stir2}$  (stirrups area) = 50.26548

$psh_y * F_{ywe} = psh1 * F_{ywe1} + ps2 * F_{ywe2} = 6.12205$   
 $psh1 ((5.4d), TBDY) = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$   
 $L_{stir1}$  (Length of stirrups along X) = 1040.00  
 $A_{stir1}$  (stirrups area) = 78.53982  
 $psh2 ((5.4d), TBDY) = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$   
 $L_{stir2}$  (Length of stirrups along X) = 768.00  
 $A_{stir2}$  (stirrups area) = 50.26548

$A_{sec} = 120000.00$   
 $s1 = 100.00$   
 $s2 = 250.00$   
 $f_{ywe1} = 781.25$   
 $f_{ywe2} = 625.00$   
 $f_{ce} = 30.00$

From ((5.A5), TBDY), TBDY:  $cc = 0.002$   
 $c =$  confinement factor = 1.00

$y1 = 0.0025$   
 $sh1 = 0.008$   
 $ft1 = 886.7403$   
 $fy1 = 738.9503$   
 $su1 = 0.032$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/d = 1.00

su1 = 0.4\*esu1\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1,ft1,fy1, it is considered  
characteristic value fsy1 = fs1/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs1 = (fs,jacket\*Asl,ten,jacket + fs,core\*Asl,ten,core)/Asl,ten = 738.9503

with Es1 = (Es,jacket\*Asl,ten,jacket + Es,core\*Asl,ten,core)/Asl,ten = 200000.00

y2 = 0.0025

sh2 = 0.008

ft2 = 898.293

fy2 = 748.5775

su2 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb,min = 1.00

su2 = 0.4\*esu2\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu2\_nominal = 0.08,

For calculation of esu2\_nominal and y2, sh2,ft2,fy2, it is considered  
characteristic value fsy2 = fs2/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs2 = (fs,jacket\*Asl,com,jacket + fs,core\*Asl,com,core)/Asl,com = 748.5775

with Es2 = (Es,jacket\*Asl,com,jacket + Es,core\*Asl,com,core)/Asl,com = 200000.00

yv = 0.0025

shv = 0.008

ftv = 894.3662

fyv = 745.3052

suv = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/d = 1.00

suv = 0.4\*esuv\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esuv\_nominal = 0.08,

considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY  
For calculation of esuv\_nominal and yv, shv,ftv,fyv, it is considered  
characteristic value fsyv = fsv/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fsv = (fs,jacket\*Asl,mid,jacket + fs,mid\*Asl,mid,core)/Asl,mid = 745.3052

with Esv = (Es,jacket\*Asl,mid,jacket + Es,mid\*Asl,mid,core)/Asl,mid = 200000.00

1 = Asl,ten/(b\*d)\*(fs1/fc) = 0.38164216

2 = Asl,com/(b\*d)\*(fs2/fc) = 0.7508007

v = Asl,mid/(b\*d)\*(fsv/fc) = 0.67946571

and confined core properties:

b = 160.00

d = 347.00

d' = 13.00

fcc (5A.2, TBDY) = 30.00

cc (5A.5, TBDY) = 0.002

c = confinement factor = 1.00

1 = Asl,ten/(b\*d)\*(fs1/fc) = 0.50454853

2 = Asl,com/(b\*d)\*(fs2/fc) = 0.99259314

v = Asl,mid/(b\*d)\*(fsv/fc) = 0.898285

Case/Assumption: Unconfined full section - Steel rupture

' satisfies Eq. (4.3)

--->

v < vs,y2 - LHS eq.(4.5) is not satisfied

--->

v < vs,c - RHS eq.(4.5) is not satisfied

--->

Case/Assumption Rejected.

--->

New Case/Assumption: Unconfined full section - Spalling of concrete cover

satisfies Eq. (4.4)

---->

$v < s_y1$  - LHS eq.(4.7) is not satisfied

---->

$v < v_{c,y1}$  - RHS eq.(4.6) is satisfied

---->

$c_u$  (4.10) = 0.52172715

$M_{Rc}$  (4.17) = 6.3768E+008

---->

New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover

In expressions below, the following modifications have been made

- $b$ ,  $d$ ,  $d'$  replaced by geometric parameters of the core:  $b_o$ ,  $d_o$ ,  $d'_o$
- $N$ ,  $1$ ,  $2$ ,  $v$  normalised to  $b_o*d_o$ , instead of  $b*d$
- $f_{cc}$ ,  $\epsilon_{cc}$  - parameters of confined concrete,  $f_{cc}$ ,  $\epsilon_{cc}$ , used in lieu of  $f_c$ ,  $\epsilon_{cu}$

---->

Subcase: Rupture of tension steel

---->

$v^* < v^*s_y2$  - LHS eq.(4.5) is not satisfied

---->

$v^* < v^*s_{c,c}$  - LHS eq.(4.5) is not satisfied

---->

Subcase rejected

---->

New Subcase: Failure of compression zone

---->

$v^* < v^*c_{y2}$  - LHS eq.(4.6) is not satisfied

---->

$v^* < v^*c_{y1}$  - RHS eq.(4.6) is not satisfied

---->

$c_u^*$  (4.11) = 0.5604845

$M_{Ro}$  (4.18) = 4.7738E+008

$M_{Ro} < 0.8*M_{Rc}$

---->

$u = c_u$  (unconfined full section) = 4.8346386E-005

$M_u = M_{Rc}$

-----  
Calculation of ratio  $l_b/l_d$

Adequate Lap Length:  $l_b/l_d \geq 1$

-----  
Calculation of  $M_{u2+}$

-----  
Calculation of ultimate curvature  $u$  according to 4.1, Biskinis/Fardis 2013:

$u = 4.2207094E-005$

$M_u = 7.3334E+008$

-----  
with full section properties:

$b = 400.00$

$d = 367.00$

$d' = 33.00$

$v = 0.40931126$

$N = 1.8026E+006$

$f_c = 30.00$

$\epsilon_{co}$  (5A.5, TBDY) = 0.002

Final value of  $c_u$ :  $c_u^* = \text{shear\_factor} * \text{Max}(c_u, \epsilon_{cc}) = 0.00925707$

The Shear\_factor is considered equal to 1 (pure moment strength)

From (5.4b), TBDY:  $c_u = 0.00925707$

$w_e$  (5.4c) = 0.0207149

$a_{se}$  ((5.4d), TBDY) =  $(a_{se1}*A_{ext} + a_{se2}*A_{int})/A_{sec} = 0.33836562$

$a_{se1} = \text{Max}(((A_{conf,max1} - A_{noConf1})/A_{conf,max1}) * (A_{conf,min1}/A_{conf,max1}), 0) = 0.33836562$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988) "Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_1^2/6$  as defined at (A.2).  
 $ase2 (>=ase1) = \text{Max}(((A_{conf,max2}-A_{noConf2})/A_{conf,max2}) * (A_{conf,min2}/A_{conf,max2}), 0) = 0.33836562$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988) "Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_2^2/6$  as defined at (A.2).  
 $psh_{min} * F_{ywe} = \text{Min}(psh_x * F_{ywe}, psh_y * F_{ywe}) = 6.12205$

Expression (5.4d) for  $psh_{min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

-----  
 $psh_x * F_{ywe} = psh1 * F_{ywe1} + ps2 * F_{ywe2} = 6.12205$

$psh1$  ((5.4d), TBDY) =  $L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$

$L_{stir1}$  (Length of stirrups along Y) = 1040.00

$A_{stir1}$  (stirrups area) = 78.53982

$psh2$  (5.4d) =  $L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$

$L_{stir2}$  (Length of stirrups along Y) = 768.00

$A_{stir2}$  (stirrups area) = 50.26548

-----  
 $psh_y * F_{ywe} = psh1 * F_{ywe1} + ps2 * F_{ywe2} = 6.12205$

$psh1$  ((5.4d), TBDY) =  $L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$

$L_{stir1}$  (Length of stirrups along X) = 1040.00

$A_{stir1}$  (stirrups area) = 78.53982

$psh2$  ((5.4d), TBDY) =  $L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$

$L_{stir2}$  (Length of stirrups along X) = 768.00

$A_{stir2}$  (stirrups area) = 50.26548

-----  
 $A_{sec} = 120000.00$

$s1 = 100.00$

$s2 = 250.00$

$f_{ywe1} = 781.25$

$f_{ywe2} = 625.00$

$f_{ce} = 30.00$

From ((5.A5), TBDY), TBDY:  $cc = 0.002$

$c = \text{confinement factor} = 1.00$

$y1 = 0.0025$

$sh1 = 0.008$

$ft1 = 898.293$

$fy1 = 748.5775$

$su1 = 0.032$

using (30) in Biskinis/Fardis (2013) multiplied with  $shear\_factor$  and also multiplied by the  $shear\_factor$  according to 15.7.1.4, with  $Shear\_factor = 1.00$

$l_o/l_{ou,min} = l_b/l_d = 1.00$

$su1 = 0.4 * esu1\_nominal$  ((5.5), TBDY) = 0.032

From table 5A.1, TBDY:  $esu1\_nominal = 0.08$ ,

For calculation of  $esu1\_nominal$  and  $y1, sh1, ft1, fy1$ , it is considered characteristic value  $fsy1 = fs1/1.2$ , from table 5.1, TBDY.

$y1, sh1, ft1, fy1$ , are also multiplied by  $\text{Min}(1, 1.25 * (l_b/l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $fs1 = (f_{s,jacket} * A_{s1,ten,jacket} + f_{s,core} * A_{s1,ten,core}) / A_{s1,ten} = 748.5775$

with  $Es1 = (E_{s,jacket} * A_{s1,ten,jacket} + E_{s,core} * A_{s1,ten,core}) / A_{s1,ten} = 200000.00$

$y2 = 0.0025$

$sh2 = 0.008$

$f_t2 = 886.7403$   
 $f_y2 = 738.9503$   
 $s_u2 = 0.032$   
 using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
 and also multiplied by the shear\_factor according to 15.7.1.4, with  
 Shear\_factor = 1.00  
 $l_o/l_{o,min} = l_b/l_{b,min} = 1.00$   
 $s_u2 = 0.4 * e_{s_u2,nominal} ((5.5), TBDY) = 0.032$   
 From table 5A.1, TBDY:  $e_{s_u2,nominal} = 0.08$ ,  
 For calculation of  $e_{s_u2,nominal}$  and  $y_2, sh_2, f_t2, f_y2$ , it is considered  
 characteristic value  $f_{s_y2} = f_{s_2}/1.2$ , from table 5.1, TBDY.  
 $y_1, sh_1, f_t1, f_y1$ , are also multiplied by  $\text{Min}(1, 1.25 * (l_b/l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.  
 with  $f_{s_2} = (f_{s,jacket} * A_{s_l,com,jacket} + f_{s,core} * A_{s_l,com,core}) / A_{s_l,com} = 738.9503$   
 with  $E_{s_2} = (E_{s,jacket} * A_{s_l,com,jacket} + E_{s,core} * A_{s_l,com,core}) / A_{s_l,com} = 200000.00$   
 $y_v = 0.0025$   
 $sh_v = 0.008$   
 $f_{t_v} = 894.3662$   
 $f_{y_v} = 745.3052$   
 $s_{u_v} = 0.032$   
 using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
 and also multiplied by the shear\_factor according to 15.7.1.4, with  
 Shear\_factor = 1.00  
 $l_o/l_{o,min} = l_b/l_d = 1.00$   
 $s_{u_v} = 0.4 * e_{s_{u_v},nominal} ((5.5), TBDY) = 0.032$   
 From table 5A.1, TBDY:  $e_{s_{u_v},nominal} = 0.08$ ,  
 considering characteristic value  $f_{s_{y_v}} = f_{s_v}/1.2$ , from table 5.1, TBDY  
 For calculation of  $e_{s_{u_v},nominal}$  and  $y_v, sh_v, f_{t_v}, f_{y_v}$ , it is considered  
 characteristic value  $f_{s_{y_v}} = f_{s_v}/1.2$ , from table 5.1, TBDY.  
 $y_1, sh_1, f_t1, f_y1$ , are also multiplied by  $\text{Min}(1, 1.25 * (l_b/l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.  
 with  $f_{s_v} = (f_{s,jacket} * A_{s_l,mid,jacket} + f_{s,mid} * A_{s_l,mid,core}) / A_{s_l,mid} = 745.3052$   
 with  $E_{s_v} = (E_{s,jacket} * A_{s_l,mid,jacket} + E_{s,mid} * A_{s_l,mid,core}) / A_{s_l,mid} = 200000.00$   
 $1 = A_{s_l,ten} / (b * d) * (f_{s_1} / f_c) = 0.37540035$   
 $2 = A_{s_l,com} / (b * d) * (f_{s_2} / f_c) = 0.19082108$   
 $v = A_{s_l,mid} / (b * d) * (f_{s_v} / f_c) = 0.33973286$   
 and confined core properties:  
 $b = 360.00$   
 $d = 347.00$   
 $d' = 13.00$   
 $f_{cc} (5A.2, TBDY) = 30.00$   
 $cc (5A.5, TBDY) = 0.002$   
 $c = \text{confinement factor} = 1.00$   
 $1 = A_{s_l,ten} / (b * d) * (f_{s_1} / f_c) = 0.44115251$   
 $2 = A_{s_l,com} / (b * d) * (f_{s_2} / f_c) = 0.22424379$   
 $v = A_{s_l,mid} / (b * d) * (f_{s_v} / f_c) = 0.39923778$   
 Case/Assumption: Unconfined full section - Steel rupture  
 ' satisfies Eq. (4.3)  
 --->  
 $v < v_{s,y2}$  - LHS eq.(4.5) is not satisfied  
 --->  
 $v < v_{s,c}$  - RHS eq.(4.5) is not satisfied  
 --->  
 Case/Assumption Rejected.  
 --->  
 New Case/Assumption: Unconfined full section - Spalling of concrete cover  
 ' satisfies Eq. (4.4)  
 --->  
 $v < s_{y1}$  - LHS eq.(4.7) is not satisfied  
 --->  
 $v < v_{c,y1}$  - RHS eq.(4.6) is satisfied  
 --->  
 $c_u (4.10) = 0.59761571$   
 $M_{Rc} (4.17) = 7.3334E+008$   
 --->  
 New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover  
 In expressions below, the following modifications have been made  
 -  $b, d, d'$  replaced by geometric parameters of the core:  $b_o, d_o, d'_o$   
 -  $N, 1, 2, v$  normalised to  $b_o * d_o$ , instead of  $b * d$

- - parameters of confined concrete,  $f_{cc}$ ,  $c_c$ , used in lieu of  $f_c$ ,  $e_{cu}$

---->

Subcase: Rupture of tension steel

---->

$v^* < v^*_{s,y2}$  - LHS eq.(4.5) is not satisfied

---->

$v^* < v^*_{s,c}$  - LHS eq.(4.5) is not satisfied

---->

Subcase rejected

---->

New Subcase: Failure of compression zone

---->

$v^* < v^*_{c,y2}$  - LHS eq.(4.6) is not satisfied

---->

$v^* < v^*_{c,y1}$  - RHS eq.(4.6) is not satisfied

---->

\* $c_u$  (4.11) = 0.62879536

$M_{Ro}$  (4.18) = 4.5866E+008

$M_{Ro} < 0.8 * M_{Rc}$

---->

$u = c_u$  (unconfined full section) = 4.2207094E-005

$\mu = M_{Rc}$

-----  
Calculation of ratio  $l_b/d$

-----  
Adequate Lap Length:  $l_b/d \geq 1$

-----  
Calculation of  $\mu_2$ -

-----  
Calculation of ultimate curvature  $\mu$  according to 4.1, Biskinis/Fardis 2013:

$u = 4.8346386E-005$

$\mu = 6.3768E+008$

-----  
with full section properties:

$b = 200.00$

$d = 367.00$

$d' = 33.00$

$v = 0.81862252$

$N = 1.8026E+006$

$f_c = 30.00$

$c_o$  (5A.5, TBDY) = 0.002

Final value of  $c_u$ :  $c_u^* = \text{shear\_factor} * \text{Max}(c_u, c_c) = 0.00925707$

The Shear\_factor is considered equal to 1 (pure moment strength)

From (5.4b), TBDY:  $c_u = 0.00925707$

$w_e$  (5.4c) = 0.0207149

$a_{se}$  ((5.4d), TBDY) =  $(a_{se1} * A_{ext} + a_{se2} * A_{int}) / A_{sec} = 0.33836562$

$a_{se1} = \text{Max}(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).

$a_{se2} (\geq a_{se1}) = \text{Max}(((A_{conf,max2} - A_{noConf2}) / A_{conf,max2}) * (A_{conf,min2} / A_{conf,max2}), 0) = 0.33836562$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.  
 $A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe}, p_{sh,y} * F_{ywe}) = 6.12205$

Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

-----  
 $p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$   
 $p_{sh1}$  ((5.4d), TBDY) =  $L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$   
 $L_{stir1}$  (Length of stirrups along Y) = 1040.00  
 $A_{stir1}$  (stirrups area) = 78.53982  
 $p_{sh2}$  (5.4d) =  $L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$   
 $L_{stir2}$  (Length of stirrups along Y) = 768.00  
 $A_{stir2}$  (stirrups area) = 50.26548

-----  
 $p_{sh,y} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$   
 $p_{sh1}$  ((5.4d), TBDY) =  $L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$   
 $L_{stir1}$  (Length of stirrups along X) = 1040.00  
 $A_{stir1}$  (stirrups area) = 78.53982  
 $p_{sh2}$  ((5.4d), TBDY) =  $L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$   
 $L_{stir2}$  (Length of stirrups along X) = 768.00  
 $A_{stir2}$  (stirrups area) = 50.26548

-----  
 $A_{sec} = 120000.00$

$s1 = 100.00$

$s2 = 250.00$

$f_{ywe1} = 781.25$

$f_{ywe2} = 625.00$

$f_{ce} = 30.00$

From ((5.A5), TBDY), TBDY:  $c_c = 0.002$

$c$  = confinement factor = 1.00

$y1 = 0.0025$

$sh1 = 0.008$

$ft1 = 886.7403$

$fy1 = 738.9503$

$su1 = 0.032$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

$l_o/l_{o,min} = l_b/l_d = 1.00$

$su1 = 0.4 * esu1_{nominal}$  ((5.5), TBDY) = 0.032

From table 5A.1, TBDY:  $esu1_{nominal} = 0.08$ ,

For calculation of  $esu1_{nominal}$  and  $y1, sh1, ft1, fy1$ , it is considered characteristic value  $fsy1 = fs1/1.2$ , from table 5.1, TBDY.

$y1, sh1, ft1, fy1$ , are also multiplied by  $\text{Min}(1, 1.25 * (l_b/l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $fs1 = (f_{s,jacket} * A_{s,ten,jacket} + f_{s,core} * A_{s,ten,core}) / A_{s,ten} = 738.9503$

with  $E_{s1} = (E_{s,jacket} * A_{s,ten,jacket} + E_{s,core} * A_{s,ten,core}) / A_{s,ten} = 200000.00$

$y2 = 0.0025$

$sh2 = 0.008$

$ft2 = 898.293$

$fy2 = 748.5775$

$su2 = 0.032$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

$l_o/l_{o,min} = l_b/l_{b,min} = 1.00$

$su2 = 0.4 * esu2_{nominal}$  ((5.5), TBDY) = 0.032

From table 5A.1, TBDY:  $esu2_{nominal} = 0.08$ ,

For calculation of  $esu2_{nominal}$  and  $y2, sh2, ft2, fy2$ , it is considered characteristic value  $fsy2 = fs2/1.2$ , from table 5.1, TBDY.

$y1, sh1, ft1, fy1$ , are also multiplied by  $\text{Min}(1, 1.25 * (l_b/l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $fs2 = (f_{s,jacket} * A_{s,com,jacket} + f_{s,core} * A_{s,com,core}) / A_{s,com} = 748.5775$

with  $E_s2 = (E_{s,jacket} \cdot A_{s,com,jacket} + E_{s,core} \cdot A_{s,com,core}) / A_{s,com} = 200000.00$

$y_v = 0.0025$   
 $sh_v = 0.008$   
 $ft_v = 894.3662$   
 $fy_v = 745.3052$   
 $suv = 0.032$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00  
 $lo/lo_{min} = lb/ld = 1.00$   
 $suv = 0.4 \cdot esuv_{nominal}((5.5), TBDY) = 0.032$   
From table 5A.1, TBDY:  $esuv_{nominal} = 0.08$ ,  
considering characteristic value  $fs_v = fsv/1.2$ , from table 5.1, TBDY  
For calculation of  $esuv_{nominal}$  and  $y_v, sh_v, ft_v, fy_v$ , it is considered  
characteristic value  $fs_v = fsv/1.2$ , from table 5.1, TBDY.  
 $y_1, sh_1, ft_1, fy_1$ , are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.  
with  $fsv = (f_{s,jacket} \cdot A_{s,mid,jacket} + f_{s,mid} \cdot A_{s,mid,core}) / A_{s,mid} = 745.3052$   
with  $E_{sv} = (E_{s,jacket} \cdot A_{s,mid,jacket} + E_{s,mid} \cdot A_{s,mid,core}) / A_{s,mid} = 200000.00$

$1 = A_{s,ten} / (b \cdot d) \cdot (fs_1 / fc) = 0.38164216$   
 $2 = A_{s,com} / (b \cdot d) \cdot (fs_2 / fc) = 0.7508007$   
 $v = A_{s,mid} / (b \cdot d) \cdot (fsv / fc) = 0.67946571$

and confined core properties:

$b = 160.00$   
 $d = 347.00$   
 $d' = 13.00$   
 $fcc(5A.2, TBDY) = 30.00$   
 $cc(5A.5, TBDY) = 0.002$   
 $c = \text{confinement factor} = 1.00$   
 $1 = A_{s,ten} / (b \cdot d) \cdot (fs_1 / fc) = 0.50454853$   
 $2 = A_{s,com} / (b \cdot d) \cdot (fs_2 / fc) = 0.99259314$   
 $v = A_{s,mid} / (b \cdot d) \cdot (fsv / fc) = 0.898285$

Case/Assumption: Unconfined full section - Steel rupture  
'satisfies Eq. (4.3)

--->  
 $v < v_{s,y2}$  - LHS eq.(4.5) is not satisfied  
--->  
 $v < v_{s,c}$  - RHS eq.(4.5) is not satisfied  
--->  
Case/Assumption Rejected.  
--->  
New Case/Assumption: Unconfined full section - Spalling of concrete cover  
'satisfies Eq. (4.4)

--->  
 $v < s_{y1}$  - LHS eq.(4.7) is not satisfied  
--->  
 $v < v_{c,y1}$  - RHS eq.(4.6) is satisfied  
--->  
 $cu(4.10) = 0.52172715$   
 $MRC(4.17) = 6.3768E+008$   
--->  
New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover  
In expressions below, the following modifications have been made

- $b, d, d'$  replaced by geometric parameters of the core:  $bo, do, d'o$
- $N, 1, 2, v$  normalised to  $bo \cdot do$ , instead of  $b \cdot d$
- $f_{cc}, cc$  parameters of confined concrete,  $f_{cc}, cc$ , used in lieu of  $fc, ec_u$

--->  
Subcase: Rupture of tension steel  
--->  
 $v^* < v^*_{s,y2}$  - LHS eq.(4.5) is not satisfied  
--->  
 $v^* < v^*_{s,c}$  - LHS eq.(4.5) is not satisfied  
--->  
Subcase rejected  
--->  
New Subcase: Failure of compression zone  
--->  
 $v^* < v^*_{c,y2}$  - LHS eq.(4.6) is not satisfied

```

--->
v* < v*c,y1 - RHS eq.(4.6) is not satisfied
--->
*cu (4.11) = 0.5604845
MRo (4.18) = 4.7738E+008
MRo < 0.8*MRc
--->
u = cu (unconfined full section) = 4.8346386E-005
Mu = MRc
-----

Calculation of ratio lb/d
-----

Adequate Lap Length: lb/d >= 1
-----
-----
-----

Calculation of Shear Strength Vr = Min(Vr1,Vr2) = 339150.739
-----

Calculation of Shear Strength at edge 1, Vr1 = 339150.739
Vr1 = VCol ((10.3), ASCE 41-17) = knl*VColO
VColO = 339150.739
knl = 1 (zero step-static loading)
-----

NOTE: In expression (10-3) 'Vs = Av*fy*d/s' is replaced by 'Vs+ f*Vf'
where Vf is the contribution of FRPs (11.3), ACI 440).
-----

= 1 (normal-weight concrete)
Mean concrete strength: fc' = (fc'_jacket*Area_jacket + fc'_core*Area_core)/Area_section = 25.00, but fc'^0.5 <= 8.3
MPa (22.5.3.1, ACI 318-14)
M/Vd = 4.00
Mu = 2.0345E+006
Vu = 2.3476251E-006
d = 0.8*h = 320.00
Nu = 1.8026E+006
Ag = 80000.00
From (11.5.4.8), ACI 318-14: Vs = Vs,jacket + Vs,core = 471238.898
where:
Vs,jacket = Vs,j1 + Vs,j2 = 471238.898
Vs,j1 = 314159.265 is calculated for section web jacket, with:
d = 320.00
Av = 157079.633
fy = 625.00
s = 100.00
Vs,j1 is multiplied by Col,j1 = 1.00
s/d = 0.3125
Vs,j2 = 157079.633 is calculated for section flange jacket, with:
d = 160.00
Av = 157079.633
fy = 625.00
s = 100.00
Vs,j2 is multiplied by Col,j2 = 1.00
s/d = 0.625
Vs,core = Vs,c1 + Vs,c2 = 0.00
Vs,c1 = 0.00 is calculated for section web core, with:
d = 240.00
Av = 100530.965
fy = 500.00
s = 250.00
Vs,c1 is multiplied by Col,c1 = 0.00
s/d = 1.04167
Vs,c2 = 0.00 is calculated for section flange core, with:
d = 80.00
Av = 100530.965
fy = 500.00
s = 250.00

```

$V_{s,c2}$  is multiplied by  $Col,c2 = 0.00$   
 $s/d = 3.125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 0.00  
From (11-11), ACI 440:  $V_s + V_f \leq 212577.225$   
 $bw = 200.00$

-----  
Calculation of Shear Strength at edge 2,  $V_{r2} = 339150.739$   
 $V_{r2} = V_{Col}$  ((10.3), ASCE 41-17) =  $knl * V_{Col0}$   
 $V_{Col0} = 339150.739$   
 $knl = 1$  (zero step-static loading)

-----  
NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ '  
where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

-----  
= 1 (normal-weight concrete)  
Mean concrete strength:  $f_c' = (f_c'_{jacket} * Area_{jacket} + f_c'_{core} * Area_{core}) / Area_{section} = 25.00$ , but  $f_c'^{0.5} \leq 8.3$   
MPa (22.5.3.1, ACI 318-14)

$M/Vd = 4.00$   
 $\mu_u = 6739.685$   
 $V_u = 2.3476251E+006$   
 $d = 0.8 * h = 320.00$   
 $N_u = 1.8026E+006$   
 $A_g = 80000.00$   
From (11.5.4.8), ACI 318-14:  $V_s = V_{s,jacket} + V_{s,core} = 471238.898$

where:

$V_{s,jacket} = V_{s,j1} + V_{s,j2} = 471238.898$   
 $V_{s,j1} = 314159.265$  is calculated for section web jacket, with:

$d = 320.00$   
 $A_v = 157079.633$   
 $f_y = 625.00$   
 $s = 100.00$

$V_{s,j1}$  is multiplied by  $Col,j1 = 1.00$   
 $s/d = 0.3125$

$V_{s,j2} = 157079.633$  is calculated for section flange jacket, with:

$d = 160.00$   
 $A_v = 157079.633$   
 $f_y = 625.00$   
 $s = 100.00$

$V_{s,j2}$  is multiplied by  $Col,j2 = 1.00$   
 $s/d = 0.625$

$V_{s,core} = V_{s,c1} + V_{s,c2} = 0.00$   
 $V_{s,c1} = 0.00$  is calculated for section web core, with:

$d = 240.00$   
 $A_v = 100530.965$   
 $f_y = 500.00$   
 $s = 250.00$

$V_{s,c1}$  is multiplied by  $Col,c1 = 0.00$   
 $s/d = 1.04167$

$V_{s,c2} = 0.00$  is calculated for section flange core, with:

$d = 80.00$   
 $A_v = 100530.965$   
 $f_y = 500.00$   
 $s = 250.00$

$V_{s,c2}$  is multiplied by  $Col,c2 = 0.00$   
 $s/d = 3.125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 0.00  
From (11-11), ACI 440:  $V_s + V_f \leq 212577.225$   
 $bw = 200.00$

-----  
End Of Calculation of Shear Capacity ratio for element: column JLC1 of floor 1  
At local axis: 3

Start Of Calculation of Shear Capacity ratio for element: column JLC1 of floor 1

At Shear local axis: 2

(Bending local axis: 3)

Section Type: rcjcs

Constant Properties

Knowledge Factor,  $\gamma = 0.90$

Mean strength values are used for both shear and moment calculations.

Consequently:

Jacket

New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 30.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 625.00$

Concrete Elasticity,  $E_c = 25742.96$

Steel Elasticity,  $E_s = 200000.00$

Existing Column

Existing material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 18.00$

Existing material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 500.00$

Concrete Elasticity,  $E_c = 19940.411$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Jacket

New material: Steel Strength,  $f_s = 1.25 * f_{sm} = 781.25$

Existing Column

Existing material: Steel Strength,  $f_s = 1.25 * f_{sm} = 625.00$

#####

Max Height,  $H_{max} = 400.00$

Min Height,  $H_{min} = 200.00$

Max Width,  $W_{max} = 400.00$

Min Width,  $W_{min} = 200.00$

Jacket Thickness,  $t_j = 50.00$

Cover Thickness,  $c = 15.00$

Mean Confinement Factor overall section = 1.00

Element Length,  $L = 3000.00$

Secondary Member

Smooth Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou,min} >= 1$ )

No FRP Wrapping

Stepwise Properties

At local axis: 2

EDGE -A-

Shear Force,  $V_a = -3.6347847E-006$

EDGE -B-

Shear Force,  $V_b = 3.6347847E-006$

BOTH EDGES

Axial Force,  $F = -1.8026E+006$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{sl,t} = 0.00$

-Compression:  $A_{sl,c} = 5353.274$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{sl,ten} = 1137.257$

-Compression:  $A_{sl,com} = 2208.54$

-Middle:  $A_{sl,mid} = 2007.478$

Calculation of Shear Capacity ratio,  $V_e/V_r = 1.18753$

Member Controlled by Shear ( $V_e/V_r > 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 402750.026$

with  
 $M_{pr1} = \text{Max}(\mu_{1+}, \mu_{1-}) = 6.0413E+008$   
 $\mu_{1+} = 6.0413E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination  
 $\mu_{1-} = 5.4612E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination  
 $M_{pr2} = \text{Max}(\mu_{2+}, \mu_{2-}) = 6.0413E+008$   
 $\mu_{2+} = 6.0413E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination  
 $\mu_{2-} = 5.4612E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

-----  
Calculation of  $\mu_{1+}$   
-----

-----  
Calculation of ultimate curvature  $\mu$  according to 4.1, Biskinis/Fardis 2013:

$\mu = 1.2613948E-005$   
 $\mu = 6.0413E+008$

-----  
with full section properties:

$b = 400.00$   
 $d = 367.00$   
 $d' = 33.00$   
 $v = 0.40931126$   
 $N = 1.8026E+006$   
 $f_c = 30.00$   
 $\alpha (5A.5, \text{TBDY}) = 0.002$   
Final value of  $\mu$ :  $\mu^* = \text{shear\_factor} * \text{Max}(\mu, \alpha) = 0.00925707$   
The Shear\_factor is considered equal to 1 (pure moment strength)  
From (5.4b), TBDY:  $\mu = 0.00925707$   
 $w_e (5.4c) = 0.0207149$

$a_{se} ((5.4d), \text{TBDY}) = (a_{se1} * A_{ext} + a_{se2} * A_{int}) / A_{sec} = 0.33836562$   
 $a_{se1} = \text{Max}(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$   
The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).  
The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988) "Theoretical Stress-Strain Model for Confined Concrete."  
J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.  
 $A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.  
 $A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length equal to half the clear spacing between external hoops.  
 $A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).  
 $a_{se2} (>= a_{se1}) = \text{Max}(((A_{conf,max2} - A_{noConf2}) / A_{conf,max2}) * (A_{conf,min2} / A_{conf,max2}), 0) = 0.33836562$   
The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).  
The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988) "Theoretical Stress-Strain Model for Confined Concrete."  
J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.  
 $A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.  
 $A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.  
 $A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).  
 $p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe}, p_{sh,y} * F_{ywe}) = 6.12205$   
Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

-----  
 $p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$   
 $p_{sh1} ((5.4d), \text{TBDY}) = L_{stir1} * A_{stir1} / (A_{sec} * s_1) = 0.00680678$   
 $L_{stir1}$  (Length of stirrups along Y) = 1040.00  
 $A_{stir1}$  (stirrups area) = 78.53982

psh2 (5.4d) =  $L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$   
Lstir2 (Length of stirrups along Y) = 768.00  
Astir2 (stirrups area) = 50.26548

psh\_y \* Fywe = psh1 \* Fywe1 + ps2 \* Fywe2 = 6.12205  
psh1 ((5.4d), TBDY) =  $L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$   
Lstir1 (Length of stirrups along X) = 1040.00  
Astir1 (stirrups area) = 78.53982  
psh2 ((5.4d), TBDY) =  $L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$   
Lstir2 (Length of stirrups along X) = 768.00  
Astir2 (stirrups area) = 50.26548

Asec = 120000.00

s1 = 100.00

s2 = 250.00

fywe1 = 781.25

fywe2 = 625.00

fce = 30.00

From ((5.A.5), TBDY), TBDY: cc = 0.002

c = confinement factor = 1.00

y1 = 0.0025

sh1 = 0.008

ft1 = 886.7403

fy1 = 738.9503

su1 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb = 1.00

su1 =  $0.4 * esu1\_nominal ((5.5), TBDY) = 0.032$

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1, ft1, fy1, it is considered  
characteristic value fsy1 = fs1/1.2, from table 5.1, TBDY.

y1, sh1, ft1, fy1, are also multiplied by  $Min(1, 1.25 * (lb/lb)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs1 =  $(fs_{jacket} * A_{sl,ten,jacket} + fs_{core} * A_{sl,ten,core}) / A_{sl,ten} = 738.9503$

with Es1 =  $(Es_{jacket} * A_{sl,ten,jacket} + Es_{core} * A_{sl,ten,core}) / A_{sl,ten} = 200000.00$

y2 = 0.0025

sh2 = 0.008

ft2 = 898.293

fy2 = 748.5775

su2 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb,min = 1.00

su2 =  $0.4 * esu2\_nominal ((5.5), TBDY) = 0.032$

From table 5A.1, TBDY: esu2\_nominal = 0.08,

For calculation of esu2\_nominal and y2, sh2, ft2, fy2, it is considered  
characteristic value fsy2 = fs2/1.2, from table 5.1, TBDY.

y1, sh1, ft1, fy1, are also multiplied by  $Min(1, 1.25 * (lb/lb)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs2 =  $(fs_{jacket} * A_{sl,com,jacket} + fs_{core} * A_{sl,com,core}) / A_{sl,com} = 748.5775$

with Es2 =  $(Es_{jacket} * A_{sl,com,jacket} + Es_{core} * A_{sl,com,core}) / A_{sl,com} = 200000.00$

yv = 0.0025

shv = 0.008

ftv = 894.3662

fyv = 745.3052

suv = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb = 1.00

suv =  $0.4 * esuv\_nominal ((5.5), TBDY) = 0.032$

From table 5A.1, TBDY: esuv\_nominal = 0.08,

considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY  
For calculation of esuv\_nominal and yv, shv, ftv, fyv, it is considered

characteristic value fsyv = fsv/1.2, from table 5.1, TBDY.

y1, sh1, ft1, fy1, are also multiplied by  $Min(1, 1.25 * (lb/lb)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $f_{sv} = (f_{s,jacket} \cdot A_{sl,mid,jacket} + f_{s,mid} \cdot A_{sl,mid,core}) / A_{sl,mid} = 745.3052$   
 with  $E_{sv} = (E_{s,jacket} \cdot A_{sl,mid,jacket} + E_{s,mid} \cdot A_{sl,mid,core}) / A_{sl,mid} = 200000.00$   
 $1 = A_{sl,ten} / (b \cdot d) \cdot (f_{s1} / f_c) = 0.19082108$   
 $2 = A_{sl,com} / (b \cdot d) \cdot (f_{s2} / f_c) = 0.37540035$   
 $v = A_{sl,mid} / (b \cdot d) \cdot (f_{sv} / f_c) = 0.33973286$

and confined core properties:

$b = 360.00$   
 $d = 347.00$   
 $d' = 13.00$   
 $f_{cc} (5A.2, TBDY) = 30.00$   
 $cc (5A.5, TBDY) = 0.002$   
 $c = \text{confinement factor} = 1.00$   
 $1 = A_{sl,ten} / (b \cdot d) \cdot (f_{s1} / f_c) = 0.22424379$   
 $2 = A_{sl,com} / (b \cdot d) \cdot (f_{s2} / f_c) = 0.44115251$   
 $v = A_{sl,mid} / (b \cdot d) \cdot (f_{sv} / f_c) = 0.39923778$

Case/Assumption: Unconfined full section - Steel rupture

' satisfies Eq. (4.3)

---

$v < v_{s,y2}$  - LHS eq.(4.5) is not satisfied

---

$v < v_{s,c}$  - RHS eq.(4.5) is not satisfied

---

Case/Assumption Rejected.

---

New Case/Assumption: Unconfined full section - Spalling of concrete cover

' satisfies Eq. (4.4)

---

$v < v_{s,y1}$  - LHS eq.(4.7) is not satisfied

---

$v < v_{c,y1}$  - RHS eq.(4.6) is satisfied

---

$c_u (4.10) = 0.37716748$

$M_{Rc} (4.17) = 7.0585E+008$

---

New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover

In expressions below, the following modifications have been made

- b, d, d' replaced by geometric parameters of the core:  $b_o$ ,  $d_o$ ,  $d'_o$
- $N$ ,  $1$ ,  $2$ ,  $v$  normalised to  $b_o \cdot d_o$ , instead of  $b \cdot d$
- $f_{cc}$ ,  $cc$ , used in lieu of  $f_c$ ,  $c_u$

---

Subcase: Rupture of tension steel

---

$v^* < v^*_{s,y2}$  - LHS eq.(4.5) is not satisfied

---

$v^* < v^*_{s,c}$  - LHS eq.(4.5) is not satisfied

---

Subcase rejected

---

New Subcase: Failure of compression zone

---

$v^* < v^*_{c,y2}$  - LHS eq.(4.6) is not satisfied

---

$v^* < v^*_{c,y1}$  - RHS eq.(4.6) is not satisfied

---

$*c_u (4.11) = 0.4569298$

$M_{Ro} (4.18) = 6.0413E+008$

---

$u = c_u (4.2) = 1.2613948E-005$

$\mu = M_{Ro}$

-----

Calculation of ratio  $l_b/d$

-----

Adequate Lap Length:  $l_b/d \geq 1$

-----

-----

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## Calculation of Mu1-

Calculation of ultimate curvature  $\mu$  according to 4.1, Biskinis/Fardis 2013:

$$\mu = 3.1475172E-005$$

$$Mu = 5.4612E+008$$

with full section properties:

$$b = 200.00$$

$$d = 367.00$$

$$d' = 33.00$$

$$v = 0.81862252$$

$$N = 1.8026E+006$$

$$f_c = 30.00$$

$$c_o (5A.5, TBDY) = 0.002$$

$$\text{Final value of } \mu: \mu^* = \text{shear\_factor} * \text{Max}(\mu, c_c) = 0.00925707$$

The Shear\_factor is considered equal to 1 (pure moment strength)

$$\text{From (5.4b), TBDY: } \mu = 0.00925707$$

$$\mu_e (5.4c) = 0.0207149$$

$$\mu_{se} ((5.4d), TBDY) = (\mu_{se1} * A_{ext} + \mu_{se2} * A_{int}) / A_{sec} = 0.33836562$$

$$\mu_{se1} = \text{Max}(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b^2/6$  as defined at (A.2).

$$\mu_{se2} (> \mu_{se1}) = \text{Max}(((A_{conf,max2} - A_{noConf2}) / A_{conf,max2}) * (A_{conf,min2} / A_{conf,max2}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b^2/6$  as defined at (A.2).

$$p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe}, p_{sh,y} * F_{ywe}) = 6.12205$$

Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

$$p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$$

$$p_{sh1} ((5.4d), TBDY) = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$$

$$L_{stir1} (\text{Length of stirrups along Y}) = 1040.00$$

$$A_{stir1} (\text{stirrups area}) = 78.53982$$

$$p_{sh2} (5.4d) = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$$

$$L_{stir2} (\text{Length of stirrups along Y}) = 768.00$$

$$A_{stir2} (\text{stirrups area}) = 50.26548$$

$$p_{sh,y} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$$

$$p_{sh1} ((5.4d), TBDY) = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$$

$$L_{stir1} (\text{Length of stirrups along X}) = 1040.00$$

$$A_{stir1} (\text{stirrups area}) = 78.53982$$

$$p_{sh2} ((5.4d), TBDY) = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$$

$$L_{stir2} (\text{Length of stirrups along X}) = 768.00$$

$$A_{stir2} (\text{stirrups area}) = 50.26548$$

$$A_{sec} = 120000.00$$

$$s1 = 100.00$$

$s_2 = 250.00$   
 $f_{ywe1} = 781.25$   
 $f_{ywe2} = 625.00$   
 $f_{ce} = 30.00$   
 From ((5.A.5), TBDY), TBDY:  $cc = 0.002$   
 $c = \text{confinement factor} = 1.00$   
 $y_1 = 0.0025$   
 $sh_1 = 0.008$   
 $ft_1 = 898.293$   
 $fy_1 = 748.5775$   
 $su_1 = 0.032$   
 using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
 and also multiplied by the shear\_factor according to 15.7.1.4, with  
 $Shear\_factor = 1.00$   
 $lo/lou, min = lb/d = 1.00$   
 $su_1 = 0.4 * esu_1\_nominal ((5.5), TBDY) = 0.032$   
 From table 5A.1, TBDY:  $esu_1\_nominal = 0.08$ ,  
 For calculation of  $esu_1\_nominal$  and  $y_1, sh_1, ft_1, fy_1$ , it is considered  
 characteristic value  $fsy_1 = fs_1/1.2$ , from table 5.1, TBDY.  
 $y_1, sh_1, ft_1, fy_1$ , are also multiplied by  $Min(1, 1.25 * (lb/d)^{2/3})$ , from 10.3.5, ASCE 41-17.  
 with  $fs_1 = (fs_{jacket} * A_{sl, ten, jacket} + fs_{core} * A_{sl, ten, core}) / A_{sl, ten} = 748.5775$   
 with  $Es_1 = (Es_{jacket} * A_{sl, ten, jacket} + Es_{core} * A_{sl, ten, core}) / A_{sl, ten} = 200000.00$   
 $y_2 = 0.0025$   
 $sh_2 = 0.008$   
 $ft_2 = 886.7403$   
 $fy_2 = 738.9503$   
 $su_2 = 0.032$   
 using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
 and also multiplied by the shear\_factor according to 15.7.1.4, with  
 $Shear\_factor = 1.00$   
 $lo/lou, min = lb/lb, min = 1.00$   
 $su_2 = 0.4 * esu_2\_nominal ((5.5), TBDY) = 0.032$   
 From table 5A.1, TBDY:  $esu_2\_nominal = 0.08$ ,  
 For calculation of  $esu_2\_nominal$  and  $y_2, sh_2, ft_2, fy_2$ , it is considered  
 characteristic value  $fsy_2 = fs_2/1.2$ , from table 5.1, TBDY.  
 $y_1, sh_1, ft_1, fy_1$ , are also multiplied by  $Min(1, 1.25 * (lb/d)^{2/3})$ , from 10.3.5, ASCE 41-17.  
 with  $fs_2 = (fs_{jacket} * A_{sl, com, jacket} + fs_{core} * A_{sl, com, core}) / A_{sl, com} = 738.9503$   
 with  $Es_2 = (Es_{jacket} * A_{sl, com, jacket} + Es_{core} * A_{sl, com, core}) / A_{sl, com} = 200000.00$   
 $y_v = 0.0025$   
 $sh_v = 0.008$   
 $ft_v = 894.3662$   
 $fy_v = 745.3052$   
 $suv = 0.032$   
 using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
 and also multiplied by the shear\_factor according to 15.7.1.4, with  
 $Shear\_factor = 1.00$   
 $lo/lou, min = lb/d = 1.00$   
 $suv = 0.4 * esuv\_nominal ((5.5), TBDY) = 0.032$   
 From table 5A.1, TBDY:  $esuv\_nominal = 0.08$ ,  
 considering characteristic value  $fsyv = fsv/1.2$ , from table 5.1, TBDY  
 For calculation of  $esuv\_nominal$  and  $y_v, sh_v, ft_v, fy_v$ , it is considered  
 characteristic value  $fsyv = fsv/1.2$ , from table 5.1, TBDY.  
 $y_1, sh_1, ft_1, fy_1$ , are also multiplied by  $Min(1, 1.25 * (lb/d)^{2/3})$ , from 10.3.5, ASCE 41-17.  
 with  $fsv = (fs_{jacket} * A_{sl, mid, jacket} + fs_{mid} * A_{sl, mid, core}) / A_{sl, mid} = 745.3052$   
 with  $Es_v = (Es_{jacket} * A_{sl, mid, jacket} + Es_{mid} * A_{sl, mid, core}) / A_{sl, mid} = 200000.00$   
 $1 = A_{sl, ten} / (b * d) * (fs_1 / fc) = 0.7508007$   
 $2 = A_{sl, com} / (b * d) * (fs_2 / fc) = 0.38164216$   
 $v = A_{sl, mid} / (b * d) * (fsv / fc) = 0.67946571$   
 and confined core properties:  
 $b = 160.00$   
 $d = 347.00$   
 $d' = 13.00$   
 $f_{cc} (5A.2, TBDY) = 30.00$   
 $cc (5A.5, TBDY) = 0.002$   
 $c = \text{confinement factor} = 1.00$   
 $1 = A_{sl, ten} / (b * d) * (fs_1 / fc) = 0.99259314$   
 $2 = A_{sl, com} / (b * d) * (fs_2 / fc) = 0.50454853$

$$v = A_{sl, mid} / (b \cdot d) \cdot (f_{sv} / f_c) = 0.898285$$

Case/Assumption: Unconfined full section - Steel rupture  
' satisfies Eq. (4.3)

--->  
 $v < v_{s, y2}$  - LHS eq.(4.5) is not satisfied

--->  
 $v < v_{s, c}$  - RHS eq.(4.5) is not satisfied

--->  
Case/Assumption Rejected.

--->  
New Case/Assumption: Unconfined full section - Spalling of concrete cover  
' satisfies Eq. (4.4)

--->  
 $v < v_{s, y1}$  - LHS eq.(4.7) is not satisfied

--->  
 $v < v_{c, y1}$  - RHS eq.(4.6) is not satisfied

--->  
 $\epsilon_{cu}$  (4.11) = 0.80138156  
 $M_{Rc}$  (4.18) = 5.4612E+008

--->  
New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover  
In expressions below, the following modifications have been made

- $b$ ,  $d$ ,  $d'$  replaced by geometric parameters of the core:  $b_o$ ,  $d_o$ ,  $d'_o$
- $N_1$ ,  $N_2$ ,  $v$  normalised to  $b_o \cdot d_o$ , instead of  $b \cdot d$
- $f_{cc}$ ,  $\epsilon_{cc}$  parameters of confined concrete,  $f_{cc}$ ,  $\epsilon_{cc}$ , used in lieu of  $f_c$ ,  $\epsilon_{cu}$

--->  
Subcase: Rupture of tension steel

--->  
 $v^* < v^*_{s, y2}$  - LHS eq.(4.5) is not satisfied

--->  
 $v^* < v^*_{s, c}$  - LHS eq.(4.5) is not satisfied

--->  
Subcase rejected

--->  
New Subcase: Failure of compression zone

--->  
 $v^* < v^*_{c, y2}$  - LHS eq.(4.6) is not satisfied

--->  
 $v^* < v^*_{c, y1}$  - RHS eq.(4.6) is not satisfied

--->  
 $\epsilon^*_{cu}$  (4.11) = 0.80674817  
 $M_{Ro}$  (4.18) = 2.8580E+008  
 $M_{Ro} < 0.8 \cdot M_{Rc}$

--->  
 $\epsilon_u = \epsilon_{cu}$  (unconfined full section) = 3.1475172E-005  
 $M_u = M_{Rc}$

-----  
Calculation of ratio  $l_b/d$

-----  
Adequate Lap Length:  $l_b/d \geq 1$

-----  
Calculation of  $M_{u2+}$

-----  
Calculation of ultimate curvature  $\epsilon_u$  according to 4.1, Biskinis/Fardis 2013:

$\epsilon_u = 1.2613948E-005$   
 $M_u = 6.0413E+008$

-----  
with full section properties:

$b = 400.00$   
 $d = 367.00$   
 $d' = 33.00$   
 $v = 0.40931126$

$$N = 1.8026E+006$$

$$f_c = 30.00$$

$$c_o (5A.5, TBDY) = 0.002$$

$$\text{Final value of } c_u: c_u^* = \text{shear\_factor} * \text{Max}(c_u, c_c) = 0.00925707$$

The Shear\_factor is considered equal to 1 (pure moment strength)

$$\text{From (5.4b), TBDY: } c_u = 0.00925707$$

$$w_e (5.4c) = 0.0207149$$

$$a_{se} ((5.4d), TBDY) = (a_{se1} * A_{ext} + a_{se2} * A_{int}) / A_{sec} = 0.33836562$$

$$a_{se1} = \text{Max}(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$$

The definitions of  $A_{noConf1}$ ,  $A_{conf,min1}$  and  $A_{conf,max1}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$a_{se2} (>= a_{se1}) = \text{Max}(((A_{conf,max2} - A_{noConf2}) / A_{conf,max2}) * (A_{conf,min2} / A_{conf,max2}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe}, p_{sh,y} * F_{ywe}) = 6.12205$$

Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

$$p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$$

$$p_{sh1} ((5.4d), TBDY) = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$$

$$L_{stir1} (\text{Length of stirrups along Y}) = 1040.00$$

$$A_{stir1} (\text{stirrups area}) = 78.53982$$

$$p_{sh2} (5.4d) = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$$

$$L_{stir2} (\text{Length of stirrups along Y}) = 768.00$$

$$A_{stir2} (\text{stirrups area}) = 50.26548$$

$$p_{sh,y} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$$

$$p_{sh1} ((5.4d), TBDY) = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$$

$$L_{stir1} (\text{Length of stirrups along X}) = 1040.00$$

$$A_{stir1} (\text{stirrups area}) = 78.53982$$

$$p_{sh2} ((5.4d), TBDY) = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$$

$$L_{stir2} (\text{Length of stirrups along X}) = 768.00$$

$$A_{stir2} (\text{stirrups area}) = 50.26548$$

$$A_{sec} = 120000.00$$

$$s1 = 100.00$$

$$s2 = 250.00$$

$$f_{ywe1} = 781.25$$

$$f_{ywe2} = 625.00$$

$$f_{ce} = 30.00$$

$$\text{From ((5.A5), TBDY), TBDY: } c_c = 0.002$$

$$c = \text{confinement factor} = 1.00$$

$$y1 = 0.0025$$

$$sh1 = 0.008$$

$$ft1 = 886.7403$$

$$fy1 = 738.9503$$

$$su1 = 0.032$$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with

Shear\_factor = 1.00

lo/lou,min = lb/d = 1.00

su1 = 0.4\*esu1\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1,ft1,fy1, it is considered characteristic value fsy1 = fs1/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs1 = (fs,jacket\*Asl,ten,jacket + fs,core\*Asl,ten,core)/Asl,ten = 738.9503

with Es1 = (Es,jacket\*Asl,ten,jacket + Es,core\*Asl,ten,core)/Asl,ten = 200000.00

y2 = 0.0025

sh2 = 0.008

ft2 = 898.293

fy2 = 748.5775

su2 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor

and also multiplied by the shear\_factor according to 15.7.1.4, with

Shear\_factor = 1.00

lo/lou,min = lb/lb,min = 1.00

su2 = 0.4\*esu2\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu2\_nominal = 0.08,

For calculation of esu2\_nominal and y2, sh2,ft2,fy2, it is considered characteristic value fsy2 = fs2/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs2 = (fs,jacket\*Asl,com,jacket + fs,core\*Asl,com,core)/Asl,com = 748.5775

with Es2 = (Es,jacket\*Asl,com,jacket + Es,core\*Asl,com,core)/Asl,com = 200000.00

yv = 0.0025

shv = 0.008

ftv = 894.3662

fyv = 745.3052

suv = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor

and also multiplied by the shear\_factor according to 15.7.1.4, with

Shear\_factor = 1.00

lo/lou,min = lb/d = 1.00

suv = 0.4\*esuv\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esuv\_nominal = 0.08,

considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY

For calculation of esuv\_nominal and yv, shv,ftv,fyv, it is considered characteristic value fsyv = fsv/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fsv = (fs,jacket\*Asl,mid,jacket + fs,mid\*Asl,mid,core)/Asl,mid = 745.3052

with Esv = (Es,jacket\*Asl,mid,jacket + Es,mid\*Asl,mid,core)/Asl,mid = 200000.00

1 = Asl,ten/(b\*d)\*(fs1/fc) = 0.19082108

2 = Asl,com/(b\*d)\*(fs2/fc) = 0.37540035

v = Asl,mid/(b\*d)\*(fsv/fc) = 0.33973286

and confined core properties:

b = 360.00

d = 347.00

d' = 13.00

fcc (5A.2, TBDY) = 30.00

cc (5A.5, TBDY) = 0.002

c = confinement factor = 1.00

1 = Asl,ten/(b\*d)\*(fs1/fc) = 0.22424379

2 = Asl,com/(b\*d)\*(fs2/fc) = 0.44115251

v = Asl,mid/(b\*d)\*(fsv/fc) = 0.39923778

Case/Assumption: Unconfined full section - Steel rupture

' satisfies Eq. (4.3)

--->

v < vs,y2 - LHS eq.(4.5) is not satisfied

--->

v < vs,c - RHS eq.(4.5) is not satisfied

--->

Case/Assumption Rejected.

--->

New Case/Assumption: Unconfined full section - Spalling of concrete cover

' satisfies Eq. (4.4)

--->

v < s,y1 - LHS eq.(4.7) is not satisfied

---->

v < vc,y1 - RHS eq.(4.6) is satisfied

---->

cu (4.10) = 0.37716748

MRC (4.17) = 7.0585E+008

---->

New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover

In expressions below, the following modifications have been made

- b, d, d' replaced by geometric parameters of the core: bo, do, d'o
- N, 1, 2, v normalised to bo\*do, instead of b\*d
- - parameters of confined concrete, fcc, cc, used in lieu of fc, ecu

---->

Subcase: Rupture of tension steel

---->

v\* < v\*s,y2 - LHS eq.(4.5) is not satisfied

---->

v\* < v\*s,c - LHS eq.(4.5) is not satisfied

---->

Subcase rejected

---->

New Subcase: Failure of compression zone

---->

v\* < v\*c,y2 - LHS eq.(4.6) is not satisfied

---->

v\* < v\*c,y1 - RHS eq.(4.6) is not satisfied

---->

\*cu (4.11) = 0.4569298

MRO (4.18) = 6.0413E+008

---->

u = cu (4.2) = 1.2613948E-005

Mu = MRO

-----  
Calculation of ratio lb/d

-----  
Adequate Lap Length: lb/d >= 1  
-----  
-----  
-----

-----  
Calculation of Mu2-  
-----  
-----  
-----

-----  
Calculation of ultimate curvature u according to 4.1, Biskinis/Fardis 2013:

u = 3.1475172E-005

Mu = 5.4612E+008  
-----

with full section properties:

b = 200.00

d = 367.00

d' = 33.00

v = 0.81862252

N = 1.8026E+006

fc = 30.00

co (5A.5, TBDY) = 0.002

Final value of cu: cu\* = shear\_factor \* Max( cu, cc) = 0.00925707

The Shear\_factor is considered equal to 1 (pure moment strength)

From (5.4b), TBDY: cu = 0.00925707

we (5.4c) = 0.0207149

ase ((5.4d), TBDY) = (ase1\*Aext+ase2\*Aint)/Asec = 0.33836562

ase1 = Max(((Aconf,max1-AnoConf1)/Aconf,max1)\*(Aconf,min1/Aconf,max1),0) = 0.33836562

The definitions of AnoConf, Aconf,min and Aconf,max are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

Aconf,max1 = 89600.00 is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

Aconf,min1 = 59225.00 is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area Aconf,max1 by a length equal to half the clear spacing between external hoops.

AnoConf1 = 43733.333 is the unconfined external core area which is equal to  $bi^2/6$  as defined at (A.2).  
ase2 ( $\geq$  ase1) =  $\text{Max}(((Aconf,max2 - AnoConf2)/Aconf,max2) * (Aconf,min2/Aconf,max2), 0) = 0.33836562$

The definitions of AnoConf, Aconf,min and Aconf,max are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988) "Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

Aconf,max2 = 45264.00 is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

Aconf,min2 = 0.00 is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area Aconf,max2 by a length equal to half the clear spacing between internal hoops.

AnoConf2 = 32309.333 is the unconfined internal core area which is equal to  $bi^2/6$  as defined at (A.2).

psh,min\*Fywe =  $\text{Min}(psh,x*Fywe, psh,y*Fywe) = 6.12205$

Expression (5.4d) for psh,min\*Fywe has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

-----  
psh\_x\*Fywe = psh1\*Fywe1 + ps2\*Fywe2 = 6.12205

psh1 ((5.4d), TBDY) =  $Lstir1 * Astir1 / (Asec * s1) = 0.00680678$

Lstir1 (Length of stirrups along Y) = 1040.00

Astir1 (stirrups area) = 78.53982

psh2 (5.4d) =  $Lstir2 * Astir2 / (Asec * s2) = 0.0012868$

Lstir2 (Length of stirrups along Y) = 768.00

Astir2 (stirrups area) = 50.26548

-----  
psh\_y\*Fywe = psh1\*Fywe1 + ps2\*Fywe2 = 6.12205

psh1 ((5.4d), TBDY) =  $Lstir1 * Astir1 / (Asec * s1) = 0.00680678$

Lstir1 (Length of stirrups along X) = 1040.00

Astir1 (stirrups area) = 78.53982

psh2 ((5.4d), TBDY) =  $Lstir2 * Astir2 / (Asec * s2) = 0.0012868$

Lstir2 (Length of stirrups along X) = 768.00

Astir2 (stirrups area) = 50.26548

-----  
Asec = 120000.00

s1 = 100.00

s2 = 250.00

fywe1 = 781.25

fywe2 = 625.00

fce = 30.00

From ((5.A5), TBDY), TBDY: cc = 0.002

c = confinement factor = 1.00

y1 = 0.0025

sh1 = 0.008

ft1 = 898.293

fy1 = 748.5775

su1 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

lo/lou,min = lb/ld = 1.00

su1 =  $0.4 * esu1\_nominal$  ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1, ft1, fy1, it is considered characteristic value fsy1 = fs1/1.2, from table 5.1, TBDY.

y1, sh1, ft1, fy1, are also multiplied by  $\text{Min}(1, 1.25 * (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs1 =  $(fs\_jacket * Asl\_ten\_jacket + fs\_core * Asl\_ten\_core) / Asl\_ten = 748.5775$

with Es1 =  $(Es\_jacket * Asl\_ten\_jacket + Es\_core * Asl\_ten\_core) / Asl\_ten = 200000.00$

y2 = 0.0025

sh2 = 0.008

ft2 = 886.7403

fy2 = 738.9503

su2 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb,min = 1.00

su2 = 0.4\*esu2\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu2\_nominal = 0.08,

For calculation of esu2\_nominal and y2, sh2,ft2,fy2, it is considered  
characteristic value fsy2 = fs2/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs2 = (fs,jacket\*Asl,com,jacket + fs,core\*Asl,com,core)/Asl,com = 738.9503

with Es2 = (Es,jacket\*Asl,com,jacket + Es,core\*Asl,com,core)/Asl,com = 200000.00

yv = 0.0025

shv = 0.008

ftv = 894.3662

fyv = 745.3052

suv = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/ld = 1.00

suv = 0.4\*esuv\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esuv\_nominal = 0.08,

considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY

For calculation of esuv\_nominal and yv, shv,ftv,fyv, it is considered  
characteristic value fsyv = fsv/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fsv = (fs,jacket\*Asl,mid,jacket + fs,mid\*Asl,mid,core)/Asl,mid = 745.3052

with Esv = (Es,jacket\*Asl,mid,jacket + Es,mid\*Asl,mid,core)/Asl,mid = 200000.00

1 = Asl,ten/(b\*d)\*(fs1/fc) = 0.7508007

2 = Asl,com/(b\*d)\*(fs2/fc) = 0.38164216

v = Asl,mid/(b\*d)\*(fsv/fc) = 0.67946571

and confined core properties:

b = 160.00

d = 347.00

d' = 13.00

fcc (5A.2, TBDY) = 30.00

cc (5A.5, TBDY) = 0.002

c = confinement factor = 1.00

1 = Asl,ten/(b\*d)\*(fs1/fc) = 0.99259314

2 = Asl,com/(b\*d)\*(fs2/fc) = 0.50454853

v = Asl,mid/(b\*d)\*(fsv/fc) = 0.898285

Case/Assumption: Unconfined full section - Steel rupture

' satisfies Eq. (4.3)

---->

v < vs,y2 - LHS eq.(4.5) is not satisfied

---->

v < vs,c - RHS eq.(4.5) is not satisfied

---->

Case/Assumption Rejected.

---->

New Case/Assumption: Unconfined full section - Spalling of concrete cover

' satisfies Eq. (4.4)

---->

v < s,y1 - LHS eq.(4.7) is not satisfied

---->

v < vc,y1 - RHS eq.(4.6) is not satisfied

---->

cu (4.11) = 0.80138156

MRc (4.18) = 5.4612E+008

---->

New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover

In expressions below, the following modifications have been made

- b, d, d' replaced by geometric parameters of the core: bo, do, d'o

- N, 1, 2, v normalised to bo\*do, instead of b\*d

- parameters of confined concrete, fcc, cc, used in lieu of fc, ecu

---->

Subcase: Rupture of tension steel

```

---->
v* < v*s,y2 - LHS eq.(4.5) is not satisfied
---->
v* < v*s,c - LHS eq.(4.5) is not satisfied
---->
Subcase rejected
---->
New Subcase: Failure of compression zone
---->
v* < v*c,y2 - LHS eq.(4.6) is not satisfied
---->
v* < v*c,y1 - RHS eq.(4.6) is not satisfied
---->
*cu (4.11) = 0.80674817
MRo (4.18) = 2.8580E+008
MRo < 0.8*MRc
---->
u = cu (unconfined full section) = 3.1475172E-005
Mu = MRc
-----

Calculation of ratio lb/d
-----
Adequate Lap Length: lb/d >= 1
-----
-----
-----
-----
Calculation of Shear Strength Vr = Min(Vr1,Vr2) = 339150.739
-----
Calculation of Shear Strength at edge 1, Vr1 = 339150.739
Vr1 = VCol ((10.3), ASCE 41-17) = knl*VColO
VColO = 339150.739
knl = 1 (zero step-static loading)
-----
NOTE: In expression (10-3) 'Vs = Av*fy*d/s' is replaced by 'Vs+ f*Vf'
where Vf is the contribution of FRPs (11.3), ACI 440).
-----
= 1 (normal-weight concrete)
Mean concrete strength: fc' = (fc'_jacket*Area_jacket + fc'_core*Area_core)/Area_section = 25.00, but fc'^0.5 <= 8.3
MPa (22.5.3.1, ACI 318-14)
M/Vd = 4.00
Mu = 6.7415E+007
Vu = 3.6347847E-006
d = 0.8*h = 320.00
Nu = 1.8026E+006
Ag = 80000.00
From (11.5.4.8), ACI 318-14: Vs = Vs,jacket + Vs,core = 471238.898
where:
Vs,jacket = Vs,j1 + Vs,j2 = 471238.898
Vs,j1 = 157079.633 is calculated for section web jacket, with:
d = 160.00
Av = 157079.633
fy = 625.00
s = 100.00
Vs,j1 is multiplied by Col,j1 = 1.00
s/d = 0.625
Vs,j2 = 314159.265 is calculated for section flange jacket, with:
d = 320.00
Av = 157079.633
fy = 625.00
s = 100.00
Vs,j2 is multiplied by Col,j2 = 1.00
s/d = 0.3125
Vs,core = Vs,c1 + Vs,c2 = 0.00
Vs,c1 = 0.00 is calculated for section web core, with:
d = 80.00

```

$$A_v = 100530.965$$

$$f_y = 500.00$$

$$s = 250.00$$

$V_{s,c1}$  is multiplied by  $Col,c1 = 0.00$

$$s/d = 3.125$$

$V_{s,c2} = 0.00$  is calculated for section flange core, with:

$$d = 240.00$$

$$A_v = 100530.965$$

$$f_y = 500.00$$

$$s = 250.00$$

$V_{s,c2}$  is multiplied by  $Col,c2 = 0.00$

$$s/d = 1.04167$$

$V_f$  ((11-3)-(11.4), ACI 440) = 0.00

From (11-11), ACI 440:  $V_s + V_f \leq 212577.225$

$$bw = 200.00$$

Calculation of Shear Strength at edge 2,  $V_r2 = 339150.739$

$V_r2 = V_{Col}$  ((10.3), ASCE 41-17) =  $k_n I V_{Col0}$

$$V_{Col0} = 339150.739$$

$k_n I = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v f_y d / s$ ' is replaced by ' $V_s + f V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

Mean concrete strength:  $f_c' = (f_c'_{jacket} \cdot Area_{jacket} + f_c'_{core} \cdot Area_{core}) / Area_{section} = 25.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$$M/Vd = 4.00$$

$$\mu_u = 5.3429E+007$$

$$V_u = 3.6347847E-006$$

$$d = 0.8 \cdot h = 320.00$$

$$N_u = 1.8026E+006$$

$$A_g = 80000.00$$

From (11.5.4.8), ACI 318-14:  $V_s = V_{sjacket} + V_{s,core} = 471238.898$

where:

$$V_{sjacket} = V_{sj1} + V_{sj2} = 471238.898$$

$V_{sj1} = 157079.633$  is calculated for section web jacket, with:

$$d = 160.00$$

$$A_v = 157079.633$$

$$f_y = 625.00$$

$$s = 100.00$$

$V_{sj1}$  is multiplied by  $Col,j1 = 1.00$

$$s/d = 0.625$$

$V_{sj2} = 314159.265$  is calculated for section flange jacket, with:

$$d = 320.00$$

$$A_v = 157079.633$$

$$f_y = 625.00$$

$$s = 100.00$$

$V_{sj2}$  is multiplied by  $Col,j2 = 1.00$

$$s/d = 0.3125$$

$V_{s,core} = V_{s,c1} + V_{s,c2} = 0.00$

$V_{s,c1} = 0.00$  is calculated for section web core, with:

$$d = 80.00$$

$$A_v = 100530.965$$

$$f_y = 500.00$$

$$s = 250.00$$

$V_{s,c1}$  is multiplied by  $Col,c1 = 0.00$

$$s/d = 3.125$$

$V_{s,c2} = 0.00$  is calculated for section flange core, with:

$$d = 240.00$$

$$A_v = 100530.965$$

$$f_y = 500.00$$

$$s = 250.00$$

$V_{s,c2}$  is multiplied by  $Col,c2 = 0.00$

$$s/d = 1.04167$$

$V_f$  ((11-3)-(11.4), ACI 440) = 0.00

From (11-11), ACI 440:  $V_s + V_f \leq 212577.225$   
 $bw = 200.00$

-----  
-----  
End Of Calculation of Shear Capacity ratio for element: column JLC1 of floor 1  
At local axis: 2

-----  
Start Of Calculation of Chord Rotation Capacity for element: column JLC1 of floor 1  
At local axis: 3  
Integration Section: (a)  
Section Type: rcjcs

#### Constant Properties

-----  
Knowledge Factor,  $\phi = 0.90$   
Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE 41-17.  
Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17  
Consequently:  
Jacket  
New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 30.00$   
New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 625.00$   
Concrete Elasticity,  $E_c = 25742.96$   
Steel Elasticity,  $E_s = 200000.00$   
Existing Column  
Existing material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 18.00$   
Existing material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 500.00$   
Concrete Elasticity,  $E_c = 19940.411$   
Steel Elasticity,  $E_s = 200000.00$   
Max Height,  $H_{max} = 400.00$   
Min Height,  $H_{min} = 200.00$   
Max Width,  $W_{max} = 400.00$   
Min Width,  $W_{min} = 200.00$   
Jacket Thickness,  $t_j = 50.00$   
Cover Thickness,  $c = 15.00$   
Element Length,  $L = 3000.00$   
Secondary Member  
Smooth Bars  
Ductile Steel  
Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Adequate Lap Length ( $l_b/d \geq 1$ )  
No FRP Wrapping

#### Stepwise Properties

-----  
Bending Moment,  $M = -3.0601E+008$   
Shear Force,  $V_2 = -62256.122$   
Shear Force,  $V_3 = -174.4443$   
Axial Force,  $F = -1.8013E+006$   
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $As_t = 2161.416$   
-Compression:  $As_c = 3191.858$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $As_{t,ten} = 2208.54$   
-Compression:  $As_{c,com} = 1137.257$   
-Middle:  $As_{l,mid} = 2007.478$   
Longitudinal External Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $As_{t,ten,jacket} = 1746.726$   
-Compression:  $As_{c,com,jacket} = 829.3805$   
-Middle:  $As_{l,mid,jacket} = 1545.664$   
Longitudinal Internal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $As_{t,ten,core} = 461.8141$   
-Compression:  $As_{c,com,core} = 307.8761$

-Middle:  $Asl_{mid,core} = 461.8141$

Mean Diameter of Tension Reinforcement,  $DbL = 16.60$

Existing component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_{R} = u = 0.02990909$   
 $u = y + p = 0.03323232$

- Calculation of  $y$  -

$y = (My * Ls / 3) / Eleff = 0.02052718$  ((4.29), Biskinis Phd)

$My = 3.1260E+008$

$Ls = M/V$  (with  $Ls > 0.1 * L$  and  $Ls < 2 * L$ ) = 4915.337

From table 10.5, ASCE 41\_17:  $Eleff = factor * Ec * Ig = 2.4951E+013$

factor = 0.70

$Ag = 120000.00$

Mean concrete strength:  $fc' = (fc'_{jacket} * Area_{jacket} + fc'_{core} * Area_{core}) / Area_{section} = 25.00$

$N = 1.8013E+006$

$Ec * Ig = Ec_{jacket} * Ig_{jacket} + Ec_{core} * Ig_{core} = 3.5645E+013$

Calculation of Yielding Moment  $My$

Calculation of  $y$  and  $My$  according to Annex 7 -

$y = \text{Min}(y_{ten}, y_{com})$

$y_{ten} = 1.9273633E-005$

with  $fy = 596.2441$

$d = 367.00$

$y = 0.57853212$

$A = 0.11409279$

$B = 0.08754673$

with  $pt = 0.0300891$

$pc = 0.01549396$

$p_v = 0.02734983$

$N = 1.8013E+006$

$b = 200.00$

" = 0.08991826

$y_{comp} = 7.6802956E-006$

with  $fc = 30.00$

$Ec = 25742.96$

$y = 0.74420275$

$A = 0.01443594$

$B = 0.04638683$

with  $Es = 200000.00$

Calculation of ratio  $lb/d$

Adequate Lap Length:  $lb/d \geq 1$

- Calculation of  $p$  -

From table 10-8:  $p = 0.01270514$

with:

- Columns not controlled by inadequate development or splicing along the clear height because  $lb/d \geq 1$

shear control ratio  $VyE/ColOE = 1.18753$

$d = d_{external} = 367.00$

$s = s_{external} = 100.00$

-  $t = s_1 + s_2 + 2 * tf / bw * (ffe / fs) = 0.00809358$

jacket:  $s_1 = Av_1 * Lstir_1 / (s_1 * Ag) = 0.00680678$

$Av_1 = 78.53982$ , is the area of every stirrup parallel to loading (shear) direction

$Lstir_1 = 1040.00$ , is the total Length of all stirrups parallel to loading (shear) direction

$$s1 = 100.00$$

$$\text{core: } s2 = Av2 * Lstir2 / (s2 * Ag) = 0.0012868$$

Av2 = 50.26548, is the area of every stirrup parallel to loading (shear) direction

Lstir2 = 768.00, is the total Length of all stirrups parallel to loading (shear) direction

$$s2 = 250.00$$

The term  $2 * tf / bw * (ffe / fs)$  is implemented to account for FRP contribution

where  $f = 2 * tf / bw$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $ffe / fs$  normalises  $f$  to steel strength

All these variables have already been given in Shear control ratio calculation.

For the normalisation  $fs$  of jacket is used.

$$NUD = 1.8013E+006$$

$$Ag = 120000.00$$

$$fcE = (fc\_jacket * Area\_jacket + fc\_core * Area\_core) / section\_area = 25.00$$

$$fyIE = (fy\_ext\_Long\_Reinf * Area\_ext\_Long\_Reinf + fy\_int\_Long\_Reinf * Area\_int\_Long\_Reinf) / Area\_Tot\_Long\_Rein = 596.2441$$

$$fytE = (fy\_ext\_Trans\_Reinf * s1 + fy\_int\_Trans\_Reinf * s2) / (s1 + s2) = 605.1263$$

$$pl = Area\_Tot\_Long\_Rein / (b * d) = 0.07293289$$

$$b = 200.00$$

$$d = 367.00$$

$$fcE = 25.00$$

End Of Calculation of Chord Rotation Capacity for element: column JLC1 of floor 1

At local axis: 3

Integration Section: (a)

## Calculation No. 13

column C1, Floor 1

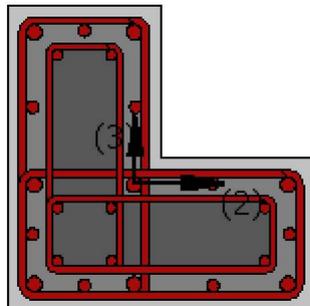
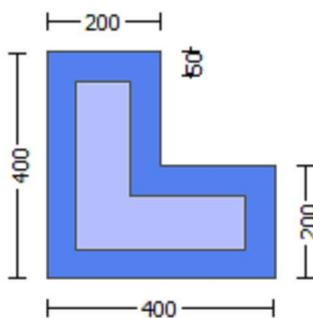
Limit State: Life Safety (data interpolation between analysis steps 50 and 51)

Analysis: Uniform +X

Check: Shear capacity VRd

Edge: End

Local Axis: (2)



Start Of Calculation of Shear Capacity for element: column JLC1 of floor 1

At local axis: 2  
Integration Section: (b)  
Section Type: rcjcs

Constant Properties

-----  
Knowledge Factor,  $\gamma = 0.90$   
Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE 41-17.  
Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17  
Consequently:  
Jacket  
New material of Secondary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 20.00$   
New material of Secondary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 500.00$   
Concrete Elasticity,  $E_c = 25742.96$   
Steel Elasticity,  $E_s = 200000.00$   
Existing Column  
Existing material of Secondary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 12.00$   
Existing material of Secondary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 400.00$   
Concrete Elasticity,  $E_c = 19940.411$   
Steel Elasticity,  $E_s = 200000.00$   
#####  
Note: Especially for the calculation of  $\gamma$  for displacement ductility demand,  
the expected (mean value) strengths are used (7.5.1.3, ASCE 41-17) because bending is considered as  
Deformation-Controlled Action (Table C7-1, ASCE 41-17).  
Jacket  
New material: Concrete Strength,  $f_c = f_{cm} = 30.00$   
New material: Steel Strength,  $f_s = f_{sm} = 625.00$   
Existing Column  
Existing material: Concrete Strength,  $f_c = f_{cm} = 18.00$   
Existing material: Steel Strength,  $f_s = f_{sm} = 500.00$   
#####  
Max Height,  $H_{max} = 400.00$   
Min Height,  $H_{min} = 200.00$   
Max Width,  $W_{max} = 400.00$   
Min Width,  $W_{min} = 200.00$   
Jacket Thickness,  $t_j = 50.00$   
Cover Thickness,  $c = 15.00$   
Element Length,  $L = 3000.00$   
Secondary Member  
Smooth Bars  
Ductile Steel  
Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Adequate Lap Length ( $l_o/l_{o,min} = l_b/l_d \geq 1$ )  
No FRP Wrapping

-----  
Stepwise Properties

-----  
EDGE -A-  
Bending Moment,  $M_a = -3.0601E+008$   
Shear Force,  $V_a = -62256.122$   
EDGE -B-  
Bending Moment,  $M_b = -6.5602E+007$   
Shear Force,  $V_b = 62256.122$   
BOTH EDGES  
Axial Force,  $F = -1.8013E+006$   
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $As_t = 0.00$   
-Compression:  $As_c = 5353.274$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $As_{t,ten} = 1137.257$   
-Compression:  $As_{l,com} = 2208.54$   
-Middle:  $As_{l,mid} = 2007.478$   
Mean Diameter of Tension Reinforcement,  $Db_{L,ten} = 16.80$   
-----  
-----

Existing component: From table 7-7, ASCE 41-17: Final Shear Capacity  $V_R = \phi V_n = 280056.413$   
 $V_n$  ((10.3), ASCE 41-17) =  $k_n V_{CoI} = 311173.792$   
 $V_{CoI} = 311173.792$   
 $k_n = 1.00$   
 $displacement\_ductility\_demand = 0.33443454$

NOTE: In expression (10-3) ' $V_s = A_v f_y d/s$ ' is replaced by ' $V_s + \phi V_f$ '  
where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)  
Mean concrete strength:  $f_c' = (f_c'_{jacket} Area_{jacket} + f_c'_{core} Area_{core}) / Area_{section} = 16.66667$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 3.29297$   
 $\mu_u = 6.5602E+007$   
 $V_u = 62256.122$   
 $d = 0.8 \cdot h = 320.00$   
 $N_u = 1.8013E+006$   
 $A_g = 80000.00$   
From (11.5.4.8), ACI 318-14:  $V_s = V_{sjacket} + V_{s,core} = 376991.118$   
where:  
 $V_{sjacket} = V_{sj1} + V_{sj2} = 376991.118$   
 $V_{sj1} = 125663.706$  is calculated for section web jacket, with:  
 $d = 160.00$   
 $A_v = 157079.633$   
 $f_y = 500.00$   
 $s = 100.00$   
 $V_{sj1}$  is multiplied by  $Col_{j1} = 1.00$   
 $s/d = 0.625$   
 $V_{sj2} = 251327.412$  is calculated for section flange jacket, with:  
 $d = 320.00$   
 $A_v = 157079.633$   
 $f_y = 500.00$   
 $s = 100.00$   
 $V_{sj2}$  is multiplied by  $Col_{j2} = 1.00$   
 $s/d = 0.3125$   
 $V_{s,core} = V_{s,c1} + V_{s,c2} = 0.00$   
 $V_{s,c1} = 0.00$  is calculated for section web core, with:  
 $d = 80.00$   
 $A_v = 100530.965$   
 $f_y = 400.00$   
 $s = 250.00$   
 $V_{s,c1}$  is multiplied by  $Col_{c1} = 0.00$   
 $s/d = 3.125$   
 $V_{s,c2} = 0.00$  is calculated for section flange core, with:  
 $d = 240.00$   
 $A_v = 100530.965$   
 $f_y = 400.00$   
 $s = 250.00$   
 $V_{s,c2}$  is multiplied by  $Col_{c2} = 0.00$   
 $s/d = 1.04167$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 0.00  
From (11-11), ACI 440:  $V_s + V_f \leq 173568.578$   
 $bw = 200.00$

$displacement\_ductility\_demand$  is calculated as  $\phi / y$

- Calculation of  $\phi / y$  for END B -  
for rotation axis 3 and integ. section (b)

From analysis, chord rotation  $\theta = 0.00248037$   
 $y = (M_y \cdot L_s / 3) / E_{eff} = 0.0074166$  ((4.29), Biskinis Phd))  
 $M_y = 5.2685E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) = 1053.751

From table 10.5, ASCE 41\_17:  $E_{eff} = \text{factor} * E_c * I_g = 2.4951E+013$

factor = 0.70

$A_g = 120000.00$

Mean concrete strength:  $f_c' = (f_c'_{jacket} * \text{Area}_{jacket} + f_c'_{core} * \text{Area}_{core}) / \text{Area}_{section} = 25.00$

$N = 1.8013E+006$

$E_c * I_g = E_c_{jacket} * I_{g,jacket} + E_c_{core} * I_{g,core} = 3.5645E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $y$  and  $M_y$  according to Annex 7 -

Assuming neutral axis within flange ( $y < t/d$ , compression zone rectangular) with:

flange width,  $b = 400.00$

web width,  $b_w = 200.00$

flange thickness,  $t = 200.00$

$y = \text{Min}(y_{ten}, y_{com})$

$y_{ten} = 1.4408661E-005$

with  $f_y = 596.2441$

$d = 367.00$

$y = 0.43622678$

$A = 0.05704639$

$B = 0.03713198$

with  $p_t = 0.00774698$

$p_c = 0.01504455$

$p_v = 0.01367492$

$N = 1.8013E+006$

$b = 400.00$

" = 0.08991826

$y_{comp} = 1.2585428E-005$

with  $f_c = 30.00$

$E_c = 25742.96$

$y = 0.45415198$

$A = 0.00721797$

$B = 0.01655203$

with  $E_s = 200000.00$

CONFIRMATION:  $y = 0.45415198 < t/d$

Calculation of ratio  $I_b/I_d$

Adequate Lap Length:  $I_b/I_d \geq 1$

End Of Calculation of Shear Capacity for element: column JLC1 of floor 1

At local axis: 2

Integration Section: (b)

**Calculation No. 14**

column C1, Floor 1

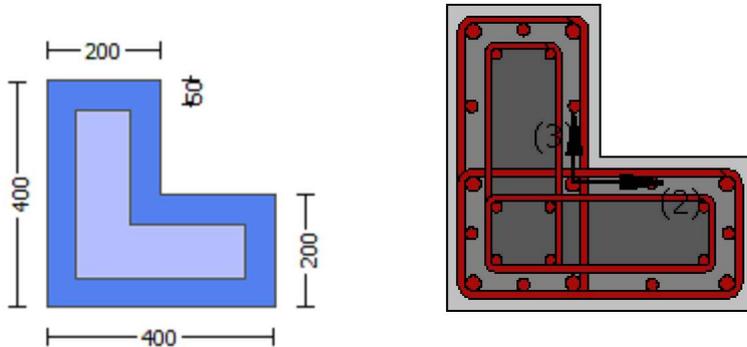
Limit State: Life Safety (data interpolation between analysis steps 50 and 51)

Analysis: Uniform +X

Check: Chord rotation capacity ( $\theta_u$ )

Edge: End

Local Axis: (2)



Start Of Calculation of Shear Capacity ratio for element: column JLC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rcjlcs

Constant Properties

Knowledge Factor,  $\gamma = 0.90$

Mean strength values are used for both shear and moment calculations.

Consequently:

Jacket

New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 30.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 625.00$

Concrete Elasticity,  $E_c = 25742.96$

Steel Elasticity,  $E_s = 200000.00$

Existing Column

Existing material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 18.00$

Existing material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 500.00$

Concrete Elasticity,  $E_c = 19940.411$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Jacket

New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 781.25$

Existing Column

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 625.00$

#####

Max Height,  $H_{max} = 400.00$

Min Height,  $H_{min} = 200.00$

Max Width,  $W_{max} = 400.00$

Min Width,  $W_{min} = 200.00$

Jacket Thickness,  $t_j = 50.00$

Cover Thickness,  $c = 15.00$

Mean Confinement Factor overall section = 1.00

Element Length,  $L = 3000.00$

Secondary Member

Smooth Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Adequate Lap Length ( $l_o/l_{ou}, \min \geq 1$ )  
No FRP Wrapping

-----  
Stepwise Properties  
-----

At local axis: 3  
EDGE -A-  
Shear Force,  $V_a = 2.3476251E-006$   
EDGE -B-  
Shear Force,  $V_b = -2.3476251E-006$   
BOTH EDGES  
Axial Force,  $F = -1.8026E+006$   
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $As_t = 0.00$   
-Compression:  $As_c = 5353.274$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $As_{t,ten} = 2208.54$   
-Compression:  $As_{c,com} = 1137.257$   
-Middle:  $As_{c,mid} = 2007.478$

-----  
Calculation of Shear Capacity ratio ,  $V_e/V_r = 1.44151$   
Member Controlled by Shear ( $V_e/V_r > 1$ )  
Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 488890.189$   
with  
 $M_{pr1} = \text{Max}(\mu_{1+}, \mu_{1-}) = 7.3334E+008$   
 $\mu_{1+} = 7.3334E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction  
which is defined for the static loading combination  
 $\mu_{1-} = 6.3768E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment  
direction which is defined for the static loading combination  
 $M_{pr2} = \text{Max}(\mu_{2+}, \mu_{2-}) = 7.3334E+008$   
 $\mu_{2+} = 7.3334E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction  
which is defined for the the static loading combination  
 $\mu_{2-} = 6.3768E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment  
direction which is defined for the the static loading combination

-----  
Calculation of  $\mu_{1+}$   
-----

-----  
Calculation of ultimate curvature  $\mu$  according to 4.1, Biskinis/Fardis 2013:

$$\mu = 4.2207094E-005$$

$$M_u = 7.3334E+008$$

-----  
with full section properties:

$$b = 400.00$$

$$d = 367.00$$

$$d' = 33.00$$

$$v = 0.40931126$$

$$N = 1.8026E+006$$

$$f_c = 30.00$$

$$c_o \text{ (5A.5, TBDY)} = 0.002$$

$$\text{Final value of } \mu_c: \mu_c^* = \text{shear\_factor} * \text{Max}(\mu_c, \mu_{cc}) = 0.00925707$$

The Shear\_factor is considered equal to 1 (pure moment strength)

$$\text{From (5.4b), TBDY: } \mu_c = 0.00925707$$

$$\mu_{we} \text{ (5.4c)} = 0.0207149$$

$$\mu_{ase} \text{ ((5.4d), TBDY)} = (ase1 * A_{ext} + ase2 * A_{int}) / A_{sec} = 0.33836562$$

$$ase1 = \text{Max}(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization  
of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.  
 $A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).  
 $a_{se2} (>= a_{se1}) = \text{Max}(((A_{conf,max2} - A_{noConf2})/A_{conf,max2}) * (A_{conf,min2}/A_{conf,max2}), 0) = 0.33836562$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).  
 $p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe}, p_{sh,y} * F_{ywe}) = 6.12205$

Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

-----  
 $p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$   
 $p_{sh1}$  ((5.4d), TBDY) =  $L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$   
 $L_{stir1}$  (Length of stirrups along Y) = 1040.00  
 $A_{stir1}$  (stirrups area) = 78.53982  
 $p_{sh2}$  (5.4d) =  $L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$   
 $L_{stir2}$  (Length of stirrups along Y) = 768.00  
 $A_{stir2}$  (stirrups area) = 50.26548

-----  
 $p_{sh,y} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$   
 $p_{sh1}$  ((5.4d), TBDY) =  $L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$   
 $L_{stir1}$  (Length of stirrups along X) = 1040.00  
 $A_{stir1}$  (stirrups area) = 78.53982  
 $p_{sh2}$  ((5.4d), TBDY) =  $L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$   
 $L_{stir2}$  (Length of stirrups along X) = 768.00  
 $A_{stir2}$  (stirrups area) = 50.26548

-----  
 $A_{sec} = 120000.00$   
 $s1 = 100.00$   
 $s2 = 250.00$   
 $f_{ywe1} = 781.25$   
 $f_{ywe2} = 625.00$   
 $f_{ce} = 30.00$

From ((5.A5), TBDY), TBDY:  $cc = 0.002$   
 $c = \text{confinement factor} = 1.00$

$y1 = 0.0025$   
 $sh1 = 0.008$   
 $ft1 = 898.293$   
 $fy1 = 748.5775$   
 $su1 = 0.032$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

$l_o/l_{ou,min} = l_b/l_d = 1.00$

$su1 = 0.4 * esu1_{nominal}$  ((5.5), TBDY) = 0.032

From table 5A.1, TBDY:  $esu1_{nominal} = 0.08$ ,

For calculation of  $esu1_{nominal}$  and  $y1, sh1, ft1, fy1$ , it is considered characteristic value  $fsy1 = fs1/1.2$ , from table 5.1, TBDY.

$y1, sh1, ft1, fy1$ , are also multiplied by  $\text{Min}(1, 1.25 * (l_b/l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $fs1 = (f_{s,jacket} * A_{s1,ten,jacket} + f_{s,core} * A_{s1,ten,core}) / A_{s1,ten} = 748.5775$

with  $Es1 = (E_{s,jacket} * A_{s1,ten,jacket} + E_{s,core} * A_{s1,ten,core}) / A_{s1,ten} = 200000.00$

$y2 = 0.0025$   
 $sh2 = 0.008$   
 $ft2 = 886.7403$

$f_y2 = 738.9503$   
 $s_u2 = 0.032$   
 using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
 and also multiplied by the shear\_factor according to 15.7.1.4, with  
 Shear\_factor = 1.00  
 $l_o/l_{o,min} = l_b/l_{b,min} = 1.00$   
 $s_u2 = 0.4 * e_{su2,nominal} ((5.5), TBDY) = 0.032$   
 From table 5A.1, TBDY:  $e_{su2,nominal} = 0.08$ ,  
 For calculation of  $e_{su2,nominal}$  and  $y_2, sh_2, ft_2, f_y2$ , it is considered  
 characteristic value  $f_{sy2} = f_s2/1.2$ , from table 5.1, TBDY.  
 $y_1, sh_1, ft_1, f_y1$ , are also multiplied by  $\text{Min}(1, 1.25 * (l_b/l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.  
 with  $f_{s2} = (f_{s,jacket} * A_{s1,com,jacket} + f_{s,core} * A_{s1,com,core}) / A_{s1,com} = 738.9503$   
 with  $E_{s2} = (E_{s,jacket} * A_{s1,com,jacket} + E_{s,core} * A_{s1,com,core}) / A_{s1,com} = 200000.00$   
 $y_v = 0.0025$   
 $sh_v = 0.008$   
 $ft_v = 894.3662$   
 $f_{yv} = 745.3052$   
 $s_{uv} = 0.032$   
 using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
 and also multiplied by the shear\_factor according to 15.7.1.4, with  
 Shear\_factor = 1.00  
 $l_o/l_{o,min} = l_b/l_d = 1.00$   
 $s_{uv} = 0.4 * e_{suv,nominal} ((5.5), TBDY) = 0.032$   
 From table 5A.1, TBDY:  $e_{suv,nominal} = 0.08$ ,  
 considering characteristic value  $f_{syv} = f_{sv}/1.2$ , from table 5.1, TBDY  
 For calculation of  $e_{suv,nominal}$  and  $y_v, sh_v, ft_v, f_{yv}$ , it is considered  
 characteristic value  $f_{syv} = f_{sv}/1.2$ , from table 5.1, TBDY.  
 $y_1, sh_1, ft_1, f_y1$ , are also multiplied by  $\text{Min}(1, 1.25 * (l_b/l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.  
 with  $f_{sv} = (f_{s,jacket} * A_{s1,mid,jacket} + f_{s,mid} * A_{s1,mid,core}) / A_{s1,mid} = 745.3052$   
 with  $E_{sv} = (E_{s,jacket} * A_{s1,mid,jacket} + E_{s,mid} * A_{s1,mid,core}) / A_{s1,mid} = 200000.00$   
 $1 = A_{s1,ten} / (b * d) * (f_{s1} / f_c) = 0.37540035$   
 $2 = A_{s1,com} / (b * d) * (f_{s2} / f_c) = 0.19082108$   
 $v = A_{s1,mid} / (b * d) * (f_{sv} / f_c) = 0.33973286$   
 and confined core properties:  
 $b = 360.00$   
 $d = 347.00$   
 $d' = 13.00$   
 $f_{cc} (5A.2, TBDY) = 30.00$   
 $cc (5A.5, TBDY) = 0.002$   
 $c = \text{confinement factor} = 1.00$   
 $1 = A_{s1,ten} / (b * d) * (f_{s1} / f_c) = 0.44115251$   
 $2 = A_{s1,com} / (b * d) * (f_{s2} / f_c) = 0.22424379$   
 $v = A_{s1,mid} / (b * d) * (f_{sv} / f_c) = 0.39923778$   
 Case/Assumption: Unconfined full section - Steel rupture  
 ' satisfies Eq. (4.3)  
 --->  
 $v < v_{s,y2}$  - LHS eq.(4.5) is not satisfied  
 --->  
 $v < v_{s,c}$  - RHS eq.(4.5) is not satisfied  
 --->  
 Case/Assumption Rejected.  
 --->  
 New Case/Assumption: Unconfined full section - Spalling of concrete cover  
 ' satisfies Eq. (4.4)  
 --->  
 $v < v_{s,y1}$  - LHS eq.(4.7) is not satisfied  
 --->  
 $v < v_{c,y1}$  - RHS eq.(4.6) is satisfied  
 --->  
 $c_u (4.10) = 0.59761571$   
 $MR_c (4.17) = 7.3334E+008$   
 --->  
 New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover  
 In expressions below, the following modifications have been made  
 -  $b, d, d'$  replaced by geometric parameters of the core:  $b_o, d_o, d'_o$   
 -  $N, 1, 2, v$  normalised to  $b_o * d_o$ , instead of  $b * d$   
 -  $f_{cc}, cc$  used in lieu of  $f_c, e_{cu}$

--->

Subcase: Rupture of tension steel

--->

$v^* < v^*_{s,y2}$  - LHS eq.(4.5) is not satisfied

--->

$v^* < v^*_{s,c}$  - LHS eq.(4.5) is not satisfied

--->

Subcase rejected

--->

New Subcase: Failure of compression zone

--->

$v^* < v^*_{c,y2}$  - LHS eq.(4.6) is not satisfied

--->

$v^* < v^*_{c,y1}$  - RHS eq.(4.6) is not satisfied

--->

\* $c_u$  (4.11) = 0.62879536

M<sub>Ro</sub> (4.18) = 4.5866E+008

M<sub>Ro</sub> < 0.8\*M<sub>Rc</sub>

--->

$u = c_u$  (unconfined full section) = 4.2207094E-005

$\mu = M_{Rc}$

-----  
Calculation of ratio  $l_b/d$

-----  
Adequate Lap Length:  $l_b/d \geq 1$

-----  
Calculation of  $\mu_{u1}$ -

-----  
Calculation of ultimate curvature  $\mu_u$  according to 4.1, Biskinis/Fardis 2013:

$u = 4.8346386E-005$

$\mu = 6.3768E+008$

-----  
with full section properties:

$b = 200.00$

$d = 367.00$

$d' = 33.00$

$v = 0.81862252$

$N = 1.8026E+006$

$f_c = 30.00$

$\omega$  (5A.5, TBDY) = 0.002

Final value of  $c_u$ :  $c_u^* = \text{shear\_factor} * \text{Max}(c_u, c_c) = 0.00925707$

The Shear\_factor is considered equal to 1 (pure moment strength)

From (5.4b), TBDY:  $c_u = 0.00925707$

$w_e$  (5.4c) = 0.0207149

$a_{se}$  ((5.4d), TBDY) =  $(a_{se1} * A_{ext} + a_{se2} * A_{int}) / A_{sec} = 0.33836562$

$a_{se1} = \text{Max}(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length

equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).

$a_{se2} (\geq a_{se1}) = \text{Max}(((A_{conf,max2} - A_{noConf2}) / A_{conf,max2}) * (A_{conf,min2} / A_{conf,max2}), 0) = 0.33836562$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe}, p_{sh,y} * F_{ywe}) = 6.12205$

Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

-----  
 $p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{s2} * F_{ywe2} = 6.12205$

$p_{sh1}$  ((5.4d), TBDY) =  $L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$

$L_{stir1}$  (Length of stirrups along Y) = 1040.00

$A_{stir1}$  (stirrups area) = 78.53982

$p_{sh2}$  (5.4d) =  $L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$

$L_{stir2}$  (Length of stirrups along Y) = 768.00

$A_{stir2}$  (stirrups area) = 50.26548

-----  
 $p_{sh,y} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{s2} * F_{ywe2} = 6.12205$

$p_{sh1}$  ((5.4d), TBDY) =  $L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$

$L_{stir1}$  (Length of stirrups along X) = 1040.00

$A_{stir1}$  (stirrups area) = 78.53982

$p_{sh2}$  ((5.4d), TBDY) =  $L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$

$L_{stir2}$  (Length of stirrups along X) = 768.00

$A_{stir2}$  (stirrups area) = 50.26548

-----  
 $A_{sec} = 120000.00$

$s1 = 100.00$

$s2 = 250.00$

$f_{ywe1} = 781.25$

$f_{ywe2} = 625.00$

$f_{ce} = 30.00$

From ((5.A5), TBDY), TBDY:  $c_c = 0.002$

$c$  = confinement factor = 1.00

$y1 = 0.0025$

$sh1 = 0.008$

$ft1 = 886.7403$

$fy1 = 738.9503$

$su1 = 0.032$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

$l_o/l_{o,min} = l_b/l_d = 1.00$

$su1 = 0.4 * esu1_{nominal}$  ((5.5), TBDY) = 0.032

From table 5A.1, TBDY:  $esu1_{nominal} = 0.08$ ,

For calculation of  $esu1_{nominal}$  and  $y1, sh1, ft1, fy1$ , it is considered characteristic value  $fsy1 = fs1/1.2$ , from table 5.1, TBDY.

$y1, sh1, ft1, fy1$ , are also multiplied by  $\text{Min}(1, 1.25 * (l_b/l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $fs1 = (f_{s,jacket} * A_{sl,ten,jacket} + f_{s,core} * A_{sl,ten,core}) / A_{sl,ten} = 738.9503$

with  $Es1 = (E_{s,jacket} * A_{sl,ten,jacket} + E_{s,core} * A_{sl,ten,core}) / A_{sl,ten} = 200000.00$

$y2 = 0.0025$

$sh2 = 0.008$

$ft2 = 898.293$

$fy2 = 748.5775$

$su2 = 0.032$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

$l_o/l_{o,min} = l_b/l_{b,min} = 1.00$

$su2 = 0.4 * esu2_{nominal}$  ((5.5), TBDY) = 0.032

From table 5A.1, TBDY:  $esu2_{nominal} = 0.08$ ,

For calculation of  $esu2_{nominal}$  and  $y2, sh2, ft2, fy2$ , it is considered characteristic value  $fsy2 = fs2/1.2$ , from table 5.1, TBDY.

$y1, sh1, ft1, fy1$ , are also multiplied by  $\text{Min}(1, 1.25 * (l_b/l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $fs2 = (f_{s,jacket} * A_{sl,com,jacket} + f_{s,core} * A_{sl,com,core}) / A_{sl,com} = 748.5775$

with  $Es2 = (E_{s,jacket} * A_{sl,com,jacket} + E_{s,core} * A_{sl,com,core}) / A_{sl,com} = 200000.00$

$y_v = 0.0025$   
 $sh_v = 0.008$   
 $ft_v = 894.3662$   
 $fy_v = 745.3052$   
 $suv = 0.032$   
 using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
 and also multiplied by the shear\_factor according to 15.7.1.4, with  
 Shear\_factor = 1.00  
 $lo/lo_{min} = lb/ld = 1.00$   
 $suv = 0.4 * esuv_{nominal} ((5.5), TBDY) = 0.032$   
 From table 5A.1, TBDY:  $esuv_{nominal} = 0.08$ ,  
 considering characteristic value  $fs_v = fsv/1.2$ , from table 5.1, TBDY  
 For calculation of  $esuv_{nominal}$  and  $y_v, sh_v, ft_v, fy_v$ , it is considered  
 characteristic value  $fs_v = fsv/1.2$ , from table 5.1, TBDY.  
 $y_1, sh_1, ft_1, fy_1$ , are also multiplied by  $Min(1, 1.25 * (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.  
 with  $fsv = (fs_{jacket} * Asl_{mid,jacket} + fs_{mid} * Asl_{mid,core}) / Asl_{mid} = 745.3052$   
 with  $Esv = (Es_{jacket} * Asl_{mid,jacket} + Es_{mid} * Asl_{mid,core}) / Asl_{mid} = 200000.00$   
 $1 = Asl_{ten} / (b * d) * (fs_1 / fc) = 0.38164216$   
 $2 = Asl_{com} / (b * d) * (fs_2 / fc) = 0.7508007$   
 $v = Asl_{mid} / (b * d) * (fsv / fc) = 0.67946571$   
 and confined core properties:  
 $b = 160.00$   
 $d = 347.00$   
 $d' = 13.00$   
 $fcc (5A.2, TBDY) = 30.00$   
 $cc (5A.5, TBDY) = 0.002$   
 $c = \text{confinement factor} = 1.00$   
 $1 = Asl_{ten} / (b * d) * (fs_1 / fc) = 0.50454853$   
 $2 = Asl_{com} / (b * d) * (fs_2 / fc) = 0.99259314$   
 $v = Asl_{mid} / (b * d) * (fsv / fc) = 0.898285$   
 Case/Assumption: Unconfined full section - Steel rupture  
 ' satisfies Eq. (4.3)  
 --->  
 $v < v_{s,y2}$  - LHS eq.(4.5) is not satisfied  
 --->  
 $v < v_{s,c}$  - RHS eq.(4.5) is not satisfied  
 --->  
 Case/Assumption Rejected.  
 --->  
 New Case/Assumption: Unconfined full section - Spalling of concrete cover  
 ' satisfies Eq. (4.4)  
 --->  
 $v < s_{y1}$  - LHS eq.(4.7) is not satisfied  
 --->  
 $v < v_{c,y1}$  - RHS eq.(4.6) is satisfied  
 --->  
 $cu (4.10) = 0.52172715$   
 $MRC (4.17) = 6.3768E+008$   
 --->  
 New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover  
 In expressions below, the following modifications have been made  
 -  $b, d, d'$  replaced by geometric parameters of the core:  $bo, do, d'o$   
 -  $N, 1, 2, v$  normalised to  $bo * do$ , instead of  $b * d$   
 - - parameters of confined concrete,  $fcc, cc$ , used in lieu of  $fc, ecu$   
 --->  
 Subcase: Rupture of tension steel  
 --->  
 $v^* < v^*_{s,y2}$  - LHS eq.(4.5) is not satisfied  
 --->  
 $v^* < v^*_{s,c}$  - LHS eq.(4.5) is not satisfied  
 --->  
 Subcase rejected  
 --->  
 New Subcase: Failure of compression zone  
 --->  
 $v^* < v^*_{c,y2}$  - LHS eq.(4.6) is not satisfied  
 --->

$v^* < v^*_{c,y1}$  - RHS eq.(4.6) is not satisfied

--->

$$*cu(4.11) = 0.5604845$$

$$MRo(4.18) = 4.7738E+008$$

$$MRo < 0.8*MRc$$

--->

$$u = cu(\text{unconfined full section}) = 4.8346386E-005$$

$$Mu = MRc$$

Calculation of ratio  $lb/d$

Adequate Lap Length:  $lb/d \geq 1$

Calculation of  $Mu_{2+}$

Calculation of ultimate curvature  $u$  according to 4.1, Biskinis/Fardis 2013:

$$u = 4.2207094E-005$$

$$Mu = 7.3334E+008$$

with full section properties:

$$b = 400.00$$

$$d = 367.00$$

$$d' = 33.00$$

$$v = 0.40931126$$

$$N = 1.8026E+006$$

$$fc = 30.00$$

$$co(5A.5, TBDY) = 0.002$$

$$\text{Final value of } cu: cu^* = \text{shear\_factor} * \text{Max}(cu, cc) = 0.00925707$$

The Shear\_factor is considered equal to 1 (pure moment strength)

$$\text{From (5.4b), TBDY: } cu = 0.00925707$$

$$we(5.4c) = 0.0207149$$

$$ase((5.4d), TBDY) = (ase1*Aext+ase2*Aint)/Asec = 0.33836562$$

$$ase1 = \text{Max}(((Aconf,max1-AnoConf1)/Aconf,max1)*(Aconf,min1/Aconf,max1),0) = 0.33836562$$

The definitions of  $AnoConf$ ,  $Aconf,min$  and  $Aconf,max$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$Aconf,max1 = 89600.00$  is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$Aconf,min1 = 59225.00$  is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area  $Aconf,max1$  by a length equal to half the clear spacing between external hoops.

$AnoConf1 = 43733.333$  is the unconfined external core area which is equal to  $bi^2/6$  as defined at (A.2).

$$ase2(\geq ase1) = \text{Max}(((Aconf,max2-AnoConf2)/Aconf,max2)*(Aconf,min2/Aconf,max2),0) = 0.33836562$$

The definitions of  $AnoConf$ ,  $Aconf,min$  and  $Aconf,max$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$Aconf,max2 = 45264.00$  is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$Aconf,min2 = 0.00$  is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area  $Aconf,max2$  by a length equal to half the clear spacing between internal hoops.

$AnoConf2 = 32309.333$  is the unconfined internal core area which is equal to  $bi^2/6$  as defined at (A.2).

$$psh,min*Fywe = \text{Min}(psh,x*Fywe, psh,y*Fywe) = 6.12205$$

Expression (5.4d) for  $psh,min*Fywe$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

$$psh_x*Fywe = psh1*Fywe1 + ps2*Fywe2 = 6.12205$$

$$psh1((5.4d), TBDY) = Lstir1*Astir1/(Asec*s1) = 0.00680678$$

Lstir1 (Length of stirrups along Y) = 1040.00  
Astir1 (stirrups area) = 78.53982  
psh2 (5.4d) = Lstir2\*Astir2/(Asec\*s2) = 0.0012868  
Lstir2 (Length of stirrups along Y) = 768.00  
Astir2 (stirrups area) = 50.26548

psh\_y\*Fywe = psh1\*Fywe1+ps2\*Fywe2 = 6.12205  
psh1 ((5.4d), TBDY) = Lstir1\*Astir1/(Asec\*s1) = 0.00680678  
Lstir1 (Length of stirrups along X) = 1040.00  
Astir1 (stirrups area) = 78.53982  
psh2 ((5.4d), TBDY) = Lstir2\*Astir2/(Asec\*s2) = 0.0012868  
Lstir2 (Length of stirrups along X) = 768.00  
Astir2 (stirrups area) = 50.26548

Asec = 120000.00

s1 = 100.00

s2 = 250.00

fywe1 = 781.25

fywe2 = 625.00

fce = 30.00

From ((5.A5), TBDY), TBDY: cc = 0.002

c = confinement factor = 1.00

y1 = 0.0025

sh1 = 0.008

ft1 = 898.293

fy1 = 748.5775

su1 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb = 1.00

su1 = 0.4\*esu1\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1,ft1,fy1, it is considered  
characteristic value fsy1 = fs1/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (\text{lb}/\text{ld})^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs1 = (fs,jacket\*Asl,ten,jacket + fs,core\*Asl,ten,core)/Asl,ten = 748.5775

with Es1 = (Es,jacket\*Asl,ten,jacket + Es,core\*Asl,ten,core)/Asl,ten = 200000.00

y2 = 0.0025

sh2 = 0.008

ft2 = 886.7403

fy2 = 738.9503

su2 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb,min = 1.00

su2 = 0.4\*esu2\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu2\_nominal = 0.08,

For calculation of esu2\_nominal and y2, sh2,ft2,fy2, it is considered  
characteristic value fsy2 = fs2/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (\text{lb}/\text{ld})^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs2 = (fs,jacket\*Asl,com,jacket + fs,core\*Asl,com,core)/Asl,com = 738.9503

with Es2 = (Es,jacket\*Asl,com,jacket + Es,core\*Asl,com,core)/Asl,com = 200000.00

yv = 0.0025

shv = 0.008

ftv = 894.3662

fyv = 745.3052

suv = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb = 1.00

suv = 0.4\*esuv\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esuv\_nominal = 0.08,

considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY

For calculation of esuv\_nominal and yv, shv,ftv,fyv, it is considered

characteristic value  $f_{sv} = f_{sv}/1.2$ , from table 5.1, TBDY.

$y_1, sh_1, ft_1, fy_1$ , are also multiplied by  $\text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $f_{sv} = (f_{s,jacket} \cdot A_{s,mid,jacket} + f_{s,mid} \cdot A_{s,mid,core}) / A_{s,mid} = 745.3052$

with  $E_{sv} = (E_{s,jacket} \cdot A_{s,mid,jacket} + E_{s,mid} \cdot A_{s,mid,core}) / A_{s,mid} = 200000.00$

1 =  $A_{s,ten} / (b \cdot d) \cdot (f_{s1}/f_c) = 0.37540035$

2 =  $A_{s,com} / (b \cdot d) \cdot (f_{s2}/f_c) = 0.19082108$

$v = A_{s,mid} / (b \cdot d) \cdot (f_{sv}/f_c) = 0.33973286$

and confined core properties:

$b = 360.00$

$d = 347.00$

$d' = 13.00$

$f_{cc}$  (5A.2, TBDY) = 30.00

$cc$  (5A.5, TBDY) = 0.002

$c$  = confinement factor = 1.00

1 =  $A_{s,ten} / (b \cdot d) \cdot (f_{s1}/f_c) = 0.44115251$

2 =  $A_{s,com} / (b \cdot d) \cdot (f_{s2}/f_c) = 0.22424379$

$v = A_{s,mid} / (b \cdot d) \cdot (f_{sv}/f_c) = 0.39923778$

Case/Assumption: Unconfined full section - Steel rupture

' satisfies Eq. (4.3)

---->

$v < v_{s,y2}$  - LHS eq.(4.5) is not satisfied

---->

$v < v_{s,c}$  - RHS eq.(4.5) is not satisfied

---->

Case/Assumption Rejected.

---->

New Case/Assumption: Unconfined full section - Spalling of concrete cover

' satisfies Eq. (4.4)

---->

$v < s_{y1}$  - LHS eq.(4.7) is not satisfied

---->

$v < v_{c,y1}$  - RHS eq.(4.6) is satisfied

---->

$c_u$  (4.10) = 0.59761571

$M_{Rc}$  (4.17) = 7.3334E+008

---->

New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover

In expressions below, the following modifications have been made

-  $b, d, d'$  replaced by geometric parameters of the core:  $b_o, d_o, d'_o$

-  $N, 1, 2, v$  normalised to  $b_o \cdot d_o$ , instead of  $b \cdot d$

-  $f_{cc}, cc$  parameters of confined concrete,  $f_{cc}, cc$ , used in lieu of  $f_c, e_{cu}$

---->

Subcase: Rupture of tension steel

---->

$v^* < v^*_{s,y2}$  - LHS eq.(4.5) is not satisfied

---->

$v^* < v^*_{s,c}$  - LHS eq.(4.5) is not satisfied

---->

Subcase rejected

---->

New Subcase: Failure of compression zone

---->

$v^* < v^*_{c,y2}$  - LHS eq.(4.6) is not satisfied

---->

$v^* < v^*_{c,y1}$  - RHS eq.(4.6) is not satisfied

---->

$*c_u$  (4.11) = 0.62879536

$M_{Ro}$  (4.18) = 4.5866E+008

$M_{Ro} < 0.8 \cdot M_{Rc}$

---->

$u = c_u$  (unconfined full section) = 4.2207094E-005

$\mu = M_{Rc}$

-----

Calculation of ratio  $l_b/l_d$

-----

Adequate Lap Length:  $l_b/l_d \geq 1$

Calculation of Mu2-

Calculation of ultimate curvature  $\mu$  according to 4.1, Biskinis/Fardis 2013:

$$\mu = 4.8346386E-005$$

$$Mu = 6.3768E+008$$

with full section properties:

$$b = 200.00$$

$$d = 367.00$$

$$d' = 33.00$$

$$v = 0.81862252$$

$$N = 1.8026E+006$$

$$f_c = 30.00$$

$$c_o \text{ (5A.5, TBDY)} = 0.002$$

$$\text{Final value of } c_u: c_u^* = \text{shear\_factor} * \text{Max}(c_u, c_o) = 0.00925707$$

The Shear\_factor is considered equal to 1 (pure moment strength)

$$\text{From (5.4b), TBDY: } c_u = 0.00925707$$

$$w_e \text{ (5.4c)} = 0.0207149$$

$$a_{se} \text{ ((5.4d), TBDY)} = (a_{se1} * A_{ext} + a_{se2} * A_{int}) / A_{sec} = 0.33836562$$

$$a_{se1} = \text{Max}(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$a_{se2} (>= a_{se1}) = \text{Max}(((A_{conf,max2} - A_{noConf2}) / A_{conf,max2}) * (A_{conf,min2} / A_{conf,max2}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe}, p_{sh,y} * F_{ywe}) = 6.12205$$

Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

$$p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{s2} * F_{ywe2} = 6.12205$$

$$p_{sh1} \text{ ((5.4d), TBDY)} = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$$

$$L_{stir1} \text{ (Length of stirrups along Y)} = 1040.00$$

$$A_{stir1} \text{ (stirrups area)} = 78.53982$$

$$p_{sh2} \text{ (5.4d)} = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$$

$$L_{stir2} \text{ (Length of stirrups along Y)} = 768.00$$

$$A_{stir2} \text{ (stirrups area)} = 50.26548$$

$$p_{sh,y} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{s2} * F_{ywe2} = 6.12205$$

$$p_{sh1} \text{ ((5.4d), TBDY)} = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$$

$$L_{stir1} \text{ (Length of stirrups along X)} = 1040.00$$

$$A_{stir1} \text{ (stirrups area)} = 78.53982$$

$$p_{sh2} \text{ ((5.4d), TBDY)} = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$$

$$L_{stir2} \text{ (Length of stirrups along X)} = 768.00$$

$$A_{stir2} \text{ (stirrups area)} = 50.26548$$

Asec = 120000.00

s1 = 100.00

s2 = 250.00

fywe1 = 781.25

fywe2 = 625.00

fce = 30.00

From ((5.A.5), TBDY), TBDY: cc = 0.002

c = confinement factor = 1.00

y1 = 0.0025

sh1 = 0.008

ft1 = 886.7403

fy1 = 738.9503

su1 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb = 1.00

su1 = 0.4\*esu1\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1,ft1,fy1, it is considered  
characteristic value fsy1 = fs1/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (\text{lb}/\text{ld})^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs1 = (fsjacket\*Asl,ten,jacket + fs,core\*Asl,ten,core)/Asl,ten = 738.9503

with Es1 = (Es,jacket\*Asl,ten,jacket + Es,core\*Asl,ten,core)/Asl,ten = 200000.00

y2 = 0.0025

sh2 = 0.008

ft2 = 898.293

fy2 = 748.5775

su2 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb,min = 1.00

su2 = 0.4\*esu2\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu2\_nominal = 0.08,

For calculation of esu2\_nominal and y2, sh2,ft2,fy2, it is considered  
characteristic value fsy2 = fs2/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (\text{lb}/\text{ld})^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs2 = (fs,jacket\*Asl,com,jacket + fs,core\*Asl,com,core)/Asl,com = 748.5775

with Es2 = (Es,jacket\*Asl,com,jacket + Es,core\*Asl,com,core)/Asl,com = 200000.00

yv = 0.0025

shv = 0.008

ftv = 894.3662

fyv = 745.3052

suv = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb = 1.00

suv = 0.4\*esuv\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esuv\_nominal = 0.08,

considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY

For calculation of esuv\_nominal and yv, shv,ftv,fyv, it is considered  
characteristic value fsyv = fsv/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (\text{lb}/\text{ld})^{2/3})$ , from 10.3.5, ASCE 41-17.

with fsv = (fs,jacket\*Asl,mid,jacket + fs,mid\*Asl,mid,core)/Asl,mid = 745.3052

with Esv = (Es,jacket\*Asl,mid,jacket + Es,mid\*Asl,mid,core)/Asl,mid = 200000.00

1 = Asl,ten/(b\*d)\*(fs1/fc) = 0.38164216

2 = Asl,com/(b\*d)\*(fs2/fc) = 0.7508007

v = Asl,mid/(b\*d)\*(fsv/fc) = 0.67946571

and confined core properties:

b = 160.00

d = 347.00

d' = 13.00

fcc (5A.2, TBDY) = 30.00

cc (5A.5, TBDY) = 0.002

c = confinement factor = 1.00  
1 =  $A_{sl,ten}/(b*d)*(f_{s1}/f_c) = 0.50454853$   
2 =  $A_{sl,com}/(b*d)*(f_{s2}/f_c) = 0.99259314$   
v =  $A_{sl,mid}/(b*d)*(f_{sv}/f_c) = 0.898285$

Case/Assumption: Unconfined full section - Steel rupture  
' satisfies Eq. (4.3)

---->  
v <  $v_{s,y2}$  - LHS eq.(4.5) is not satisfied

---->  
v <  $v_{s,c}$  - RHS eq.(4.5) is not satisfied

---->  
Case/Assumption Rejected.

---->  
New Case/Assumption: Unconfined full section - Spalling of concrete cover  
' satisfies Eq. (4.4)

---->  
v <  $s_{y1}$  - LHS eq.(4.7) is not satisfied

---->  
v <  $v_{c,y1}$  - RHS eq.(4.6) is satisfied

---->  
cu (4.10) = 0.52172715  
MRc (4.17) = 6.3768E+008

---->  
New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover  
In expressions below, the following modifications have been made

- b, d, d' replaced by geometric parameters of the core:  $b_o, d_o, d'_o$
- N, 1, 2, v normalised to  $b_o*d_o$ , instead of  $b*d$
- - parameters of confined concrete,  $f_{cc}, c_c$ , used in lieu of  $f_c, e_c$

---->  
Subcase: Rupture of tension steel

---->  
 $v^* < v^*_{s,y2}$  - LHS eq.(4.5) is not satisfied

---->  
 $v^* < v^*_{s,c}$  - LHS eq.(4.5) is not satisfied

---->  
Subcase rejected

---->  
New Subcase: Failure of compression zone

---->  
 $v^* < v^*_{c,y2}$  - LHS eq.(4.6) is not satisfied

---->  
 $v^* < v^*_{c,y1}$  - RHS eq.(4.6) is not satisfied

---->  
 $*c_u$  (4.11) = 0.5604845  
MRo (4.18) = 4.7738E+008  
MRo < 0.8\*MRc

---->  
u = cu (unconfined full section) = 4.8346386E-005  
Mu = MRc

-----  
Calculation of ratio  $l_b/l_d$

-----  
Adequate Lap Length:  $l_b/l_d \geq 1$

-----  
-----  
-----

-----  
Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 339150.739$

-----  
Calculation of Shear Strength at edge 1,  $V_{r1} = 339150.739$

$V_{r1} = V_{Co1}$  ((10.3), ASCE 41-17) =  $k_{nl}*V_{Co1}$   
 $V_{Co1} = 339150.739$   
 $k_{nl} = 1$  (zero step-static loading)

-----  
NOTE: In expression (10-3) ' $V_s = A_v*f_y*d/s$ ' is replaced by ' $V_{s+} = f^*V_f$ '  
where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

-----  
 = 1 (normal-weight concrete)  
 Mean concrete strength:  $f_c' = (f_c'_{\text{jacket}} \cdot \text{Area}_{\text{jacket}} + f_c'_{\text{core}} \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 25.00$ , but  $f_c'^{0.5} \leq 8.3$   
 MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 4.00$   
 $\mu_u = 2.0345E+006$   
 $V_u = 2.3476251E-006$   
 $d = 0.8 \cdot h = 320.00$   
 $N_u = 1.8026E+006$   
 $A_g = 80000.00$   
 From (11.5.4.8), ACI 318-14:  $V_s = V_{s,\text{jacket}} + V_{s,\text{core}} = 471238.898$   
 where:  
 $V_{s,\text{jacket}} = V_{s,j1} + V_{s,j2} = 471238.898$   
 $V_{s,j1} = 314159.265$  is calculated for section web jacket, with:  
 $d = 320.00$   
 $A_v = 157079.633$   
 $f_y = 625.00$   
 $s = 100.00$   
 $V_{s,j1}$  is multiplied by  $\text{Col},j1 = 1.00$   
 $s/d = 0.3125$   
 $V_{s,j2} = 157079.633$  is calculated for section flange jacket, with:  
 $d = 160.00$   
 $A_v = 157079.633$   
 $f_y = 625.00$   
 $s = 100.00$   
 $V_{s,j2}$  is multiplied by  $\text{Col},j2 = 1.00$   
 $s/d = 0.625$   
 $V_{s,\text{core}} = V_{s,c1} + V_{s,c2} = 0.00$   
 $V_{s,c1} = 0.00$  is calculated for section web core, with:  
 $d = 240.00$   
 $A_v = 100530.965$   
 $f_y = 500.00$   
 $s = 250.00$   
 $V_{s,c1}$  is multiplied by  $\text{Col},c1 = 0.00$   
 $s/d = 1.04167$   
 $V_{s,c2} = 0.00$  is calculated for section flange core, with:  
 $d = 80.00$   
 $A_v = 100530.965$   
 $f_y = 500.00$   
 $s = 250.00$   
 $V_{s,c2}$  is multiplied by  $\text{Col},c2 = 0.00$   
 $s/d = 3.125$   
 $V_f ((11-3)-(11.4), \text{ACI } 440) = 0.00$   
 From (11-11), ACI 440:  $V_s + V_f \leq 212577.225$   
 $b_w = 200.00$   
 -----

-----  
 Calculation of Shear Strength at edge 2,  $V_{r2} = 339150.739$   
 $V_{r2} = V_{\text{Col}} ((10.3), \text{ASCE } 41-17) = k_{nl} \cdot V_{\text{Col}0}$   
 $V_{\text{Col}0} = 339150.739$   
 $k_{nl} = 1$  (zero step-static loading)  
 -----

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ '  
 where  $V_f$  is the contribution of FRPs (11.3), ACI 440).  
 -----

-----  
 = 1 (normal-weight concrete)  
 Mean concrete strength:  $f_c' = (f_c'_{\text{jacket}} \cdot \text{Area}_{\text{jacket}} + f_c'_{\text{core}} \cdot \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 25.00$ , but  $f_c'^{0.5} \leq 8.3$   
 MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 4.00$   
 $\mu_u = 6739.685$   
 $V_u = 2.3476251E-006$   
 $d = 0.8 \cdot h = 320.00$   
 $N_u = 1.8026E+006$   
 $A_g = 80000.00$   
 From (11.5.4.8), ACI 318-14:  $V_s = V_{s,\text{jacket}} + V_{s,\text{core}} = 471238.898$   
 where:  
 $V_{s,\text{jacket}} = V_{s,j1} + V_{s,j2} = 471238.898$   
 -----

Vs,j1 = 314159.265 is calculated for section web jacket, with:

d = 320.00  
Av = 157079.633  
fy = 625.00  
s = 100.00

Vs,j1 is multiplied by Col,j1 = 1.00

s/d = 0.3125

Vs,j2 = 157079.633 is calculated for section flange jacket, with:

d = 160.00  
Av = 157079.633  
fy = 625.00  
s = 100.00

Vs,j2 is multiplied by Col,j2 = 1.00

s/d = 0.625

Vs,core = Vs,c1 + Vs,c2 = 0.00

Vs,c1 = 0.00 is calculated for section web core, with:

d = 240.00  
Av = 100530.965  
fy = 500.00  
s = 250.00

Vs,c1 is multiplied by Col,c1 = 0.00

s/d = 1.04167

Vs,c2 = 0.00 is calculated for section flange core, with:

d = 80.00  
Av = 100530.965  
fy = 500.00  
s = 250.00

Vs,c2 is multiplied by Col,c2 = 0.00

s/d = 3.125

Vf ((11-3)-(11.4), ACI 440) = 0.00

From (11-11), ACI 440: Vs + Vf <= 212577.225

bw = 200.00

-----  
End Of Calculation of Shear Capacity ratio for element: column JLC1 of floor 1  
At local axis: 3  
-----

-----  
Start Of Calculation of Shear Capacity ratio for element: column JLC1 of floor 1  
At Shear local axis: 2  
(Bending local axis: 3)  
Section Type: rcjlc

Constant Properties

-----  
Knowledge Factor, = 0.90

Mean strength values are used for both shear and moment calculations.

Consequently:

Jacket

New material of Secondary Member: Concrete Strength, fc = fcm = 30.00

New material of Secondary Member: Steel Strength, fs = fsm = 625.00

Concrete Elasticity, Ec = 25742.96

Steel Elasticity, Es = 200000.00

Existing Column

Existing material of Secondary Member: Concrete Strength, fc = fcm = 18.00

Existing material of Secondary Member: Steel Strength, fs = fsm = 500.00

Concrete Elasticity, Ec = 19940.411

Steel Elasticity, Es = 200000.00

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Jacket

New material: Steel Strength, fs = 1.25\*fsm = 781.25

Existing Column

Existing material: Steel Strength, fs = 1.25\*fsm = 625.00

#####

Max Height, Hmax = 400.00  
Min Height, Hmin = 200.00  
Max Width, Wmax = 400.00  
Min Width, Wmin = 200.00  
Jacket Thickness, tj = 50.00  
Cover Thickness, c = 15.00  
Mean Confinement Factor overall section = 1.00  
Element Length, L = 3000.00  
Secondary Member  
Smooth Bars  
Ductile Steel  
Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Adequate Lap Length ( $l_o/l_{ou, \min} \geq 1$ )  
No FRP Wrapping

-----  
Stepwise Properties

-----  
At local axis: 2  
EDGE -A-  
Shear Force, Va = -3.6347847E-006  
EDGE -B-  
Shear Force, Vb = 3.6347847E-006  
BOTH EDGES  
Axial Force, F = -1.8026E+006  
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension: Aslt = 0.00  
-Compression: Asc = 5353.274  
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension: Asl,ten = 1137.257  
-Compression: Asl,com = 2208.54  
-Middle: Asl,mid = 2007.478

-----  
-----  
Calculation of Shear Capacity ratio ,  $V_e/V_r = 1.18753$   
Member Controlled by Shear ( $V_e/V_r > 1$ )  
Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 402750.026$   
with  
 $M_{pr1} = \text{Max}(Mu_{1+} , Mu_{1-}) = 6.0413E+008$   
 $Mu_{1+} = 6.0413E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction  
which is defined for the static loading combination  
 $Mu_{1-} = 5.4612E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment  
direction which is defined for the static loading combination  
 $M_{pr2} = \text{Max}(Mu_{2+} , Mu_{2-}) = 6.0413E+008$   
 $Mu_{2+} = 6.0413E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction  
which is defined for the the static loading combination  
 $Mu_{2-} = 5.4612E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment  
direction which is defined for the the static loading combination

-----  
Calculation of  $Mu_{1+}$

-----  
Calculation of ultimate curvature  $\phi_u$  according to 4.1, Biskinis/Fardis 2013:  
 $\phi_u = 1.2613948E-005$   
 $M_u = 6.0413E+008$

-----  
with full section properties:  
b = 400.00  
d = 367.00  
d' = 33.00  
v = 0.40931126

$$N = 1.8026E+006$$

$$f_c = 30.00$$

$$c_o \text{ (5A.5, TBDY)} = 0.002$$

$$\text{Final value of } c_u: c_u^* = \text{shear\_factor} * \text{Max}(c_u, c_c) = 0.00925707$$

The Shear\_factor is considered equal to 1 (pure moment strength)

$$\text{From (5.4b), TBDY: } c_u = 0.00925707$$

$$w_e \text{ (5.4c)} = 0.0207149$$

$$a_{se} \text{ ((5.4d), TBDY)} = (a_{se1} * A_{ext} + a_{se2} * A_{int}) / A_{sec} = 0.33836562$$

$$a_{se1} = \text{Max}(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$$

The definitions of  $A_{noConf1}$ ,  $A_{conf,min1}$  and  $A_{conf,max1}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$a_{se2} (>= a_{se1}) = \text{Max}(((A_{conf,max2} - A_{noConf2}) / A_{conf,max2}) * (A_{conf,min2} / A_{conf,max2}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe}, p_{sh,y} * F_{ywe}) = 6.12205$$

Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

$$p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$$

$$p_{sh1} \text{ ((5.4d), TBDY)} = L_{stir1} * A_{stir1} / (A_{sec} * s_1) = 0.00680678$$

$$L_{stir1} \text{ (Length of stirrups along Y)} = 1040.00$$

$$A_{stir1} \text{ (stirrups area)} = 78.53982$$

$$p_{sh2} \text{ (5.4d)} = L_{stir2} * A_{stir2} / (A_{sec} * s_2) = 0.0012868$$

$$L_{stir2} \text{ (Length of stirrups along Y)} = 768.00$$

$$A_{stir2} \text{ (stirrups area)} = 50.26548$$

$$p_{sh,y} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$$

$$p_{sh1} \text{ ((5.4d), TBDY)} = L_{stir1} * A_{stir1} / (A_{sec} * s_1) = 0.00680678$$

$$L_{stir1} \text{ (Length of stirrups along X)} = 1040.00$$

$$A_{stir1} \text{ (stirrups area)} = 78.53982$$

$$p_{sh2} \text{ ((5.4d), TBDY)} = L_{stir2} * A_{stir2} / (A_{sec} * s_2) = 0.0012868$$

$$L_{stir2} \text{ (Length of stirrups along X)} = 768.00$$

$$A_{stir2} \text{ (stirrups area)} = 50.26548$$

$$A_{sec} = 120000.00$$

$$s_1 = 100.00$$

$$s_2 = 250.00$$

$$f_{ywe1} = 781.25$$

$$f_{ywe2} = 625.00$$

$$f_{ce} = 30.00$$

$$\text{From ((5.A5), TBDY), TBDY: } c_c = 0.002$$

$$c = \text{confinement factor} = 1.00$$

$$y_1 = 0.0025$$

$$sh_1 = 0.008$$

$$ft_1 = 886.7403$$

$$fy_1 = 738.9503$$

$$su_1 = 0.032$$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with

Shear\_factor = 1.00

lo/lou,min = lb/d = 1.00

su1 = 0.4\*esu1\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1,ft1,fy1, it is considered characteristic value fsy1 = fs1/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs1 = (fs,jacket\*Asl,ten,jacket + fs,core\*Asl,ten,core)/Asl,ten = 738.9503

with Es1 = (Es,jacket\*Asl,ten,jacket + Es,core\*Asl,ten,core)/Asl,ten = 200000.00

y2 = 0.0025

sh2 = 0.008

ft2 = 898.293

fy2 = 748.5775

su2 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor

and also multiplied by the shear\_factor according to 15.7.1.4, with

Shear\_factor = 1.00

lo/lou,min = lb/lb,min = 1.00

su2 = 0.4\*esu2\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu2\_nominal = 0.08,

For calculation of esu2\_nominal and y2, sh2,ft2,fy2, it is considered characteristic value fsy2 = fs2/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs2 = (fs,jacket\*Asl,com,jacket + fs,core\*Asl,com,core)/Asl,com = 748.5775

with Es2 = (Es,jacket\*Asl,com,jacket + Es,core\*Asl,com,core)/Asl,com = 200000.00

yv = 0.0025

shv = 0.008

ftv = 894.3662

fyv = 745.3052

suv = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor

and also multiplied by the shear\_factor according to 15.7.1.4, with

Shear\_factor = 1.00

lo/lou,min = lb/d = 1.00

suv = 0.4\*esuv\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esuv\_nominal = 0.08,

considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY

For calculation of esuv\_nominal and yv, shv,ftv,fyv, it is considered characteristic value fsyv = fsv/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fsv = (fs,jacket\*Asl,mid,jacket + fs,mid\*Asl,mid,core)/Asl,mid = 745.3052

with Esv = (Es,jacket\*Asl,mid,jacket + Es,mid\*Asl,mid,core)/Asl,mid = 200000.00

1 = Asl,ten/(b\*d)\*(fs1/fc) = 0.19082108

2 = Asl,com/(b\*d)\*(fs2/fc) = 0.37540035

v = Asl,mid/(b\*d)\*(fsv/fc) = 0.33973286

and confined core properties:

b = 360.00

d = 347.00

d' = 13.00

fcc (5A.2, TBDY) = 30.00

cc (5A.5, TBDY) = 0.002

c = confinement factor = 1.00

1 = Asl,ten/(b\*d)\*(fs1/fc) = 0.22424379

2 = Asl,com/(b\*d)\*(fs2/fc) = 0.44115251

v = Asl,mid/(b\*d)\*(fsv/fc) = 0.39923778

Case/Assumption: Unconfined full section - Steel rupture

' satisfies Eq. (4.3)

--->

v < vs,y2 - LHS eq.(4.5) is not satisfied

--->

v < vs,c - RHS eq.(4.5) is not satisfied

--->

Case/Assumption Rejected.

--->

New Case/Assumption: Unconfined full section - Spalling of concrete cover

' satisfies Eq. (4.4)

--->

v < s,y1 - LHS eq.(4.7) is not satisfied

---->

v < vc,y1 - RHS eq.(4.6) is satisfied

---->

cu (4.10) = 0.37716748

MRC (4.17) = 7.0585E+008

---->

New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover

In expressions below, the following modifications have been made

- b, d, d' replaced by geometric parameters of the core: bo, do, d'o
- N, 1, 2, v normalised to bo\*do, instead of b\*d
- - parameters of confined concrete, fcc, cc, used in lieu of fc, ecu

---->

Subcase: Rupture of tension steel

---->

v\* < v\*s,y2 - LHS eq.(4.5) is not satisfied

---->

v\* < v\*s,c - LHS eq.(4.5) is not satisfied

---->

Subcase rejected

---->

New Subcase: Failure of compression zone

---->

v\* < v\*c,y2 - LHS eq.(4.6) is not satisfied

---->

v\* < v\*c,y1 - RHS eq.(4.6) is not satisfied

---->

\*cu (4.11) = 0.4569298

MRO (4.18) = 6.0413E+008

---->

u = cu (4.2) = 1.2613948E-005

Mu = MRO

-----  
Calculation of ratio lb/d

-----  
Adequate Lap Length: lb/d >= 1  
-----  
-----

-----  
Calculation of Mu1-  
-----  
-----

-----  
Calculation of ultimate curvature u according to 4.1, Biskinis/Fardis 2013:

u = 3.1475172E-005

Mu = 5.4612E+008  
-----

with full section properties:

b = 200.00

d = 367.00

d' = 33.00

v = 0.81862252

N = 1.8026E+006

fc = 30.00

co (5A.5, TBDY) = 0.002

Final value of cu: cu\* = shear\_factor \* Max( cu, cc) = 0.00925707

The Shear\_factor is considered equal to 1 (pure moment strength)

From (5.4b), TBDY: cu = 0.00925707

we (5.4c) = 0.0207149

ase ((5.4d), TBDY) = (ase1\*Aext+ase2\*Aint)/Asec = 0.33836562

ase1 = Max(((Aconf,max1-AnoConf1)/Aconf,max1)\*(Aconf,min1/Aconf,max1),0) = 0.33836562

The definitions of AnoConf, Aconf,min and Aconf,max are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

Aconf,max1 = 89600.00 is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

Aconf,min1 = 59225.00 is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area Aconf,max1 by a length equal to half the clear spacing between external hoops.

AnoConf1 = 43733.333 is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).  
ase2 ( $\geq$  ase1) =  $\text{Max}(((A_{\text{conf,max2}} - A_{\text{noConf2}})/A_{\text{conf,max2}}) * (A_{\text{conf,min2}}/A_{\text{conf,max2}}), 0) = 0.33836562$

The definitions of AnoConf, Aconf,min and Aconf,max are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988) "Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

Aconf,max2 = 45264.00 is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

Aconf,min2 = 0.00 is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area Aconf,max2 by a length equal to half the clear spacing between internal hoops.

AnoConf2 = 32309.333 is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

psh,min\*Fywe =  $\text{Min}(psh_x * Fywe, psh_y * Fywe) = 6.12205$

Expression (5.4d) for psh,min\*Fywe has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

-----  
psh\_x\*Fywe = psh1\*Fywe1 + ps2\*Fywe2 = 6.12205

psh1 ((5.4d), TBDY) =  $L_{\text{stir1}} * A_{\text{stir1}} / (A_{\text{sec}} * s_1) = 0.00680678$

Lstir1 (Length of stirrups along Y) = 1040.00

Astir1 (stirrups area) = 78.53982

psh2 (5.4d) =  $L_{\text{stir2}} * A_{\text{stir2}} / (A_{\text{sec}} * s_2) = 0.0012868$

Lstir2 (Length of stirrups along Y) = 768.00

Astir2 (stirrups area) = 50.26548

-----  
psh\_y\*Fywe = psh1\*Fywe1 + ps2\*Fywe2 = 6.12205

psh1 ((5.4d), TBDY) =  $L_{\text{stir1}} * A_{\text{stir1}} / (A_{\text{sec}} * s_1) = 0.00680678$

Lstir1 (Length of stirrups along X) = 1040.00

Astir1 (stirrups area) = 78.53982

psh2 ((5.4d), TBDY) =  $L_{\text{stir2}} * A_{\text{stir2}} / (A_{\text{sec}} * s_2) = 0.0012868$

Lstir2 (Length of stirrups along X) = 768.00

Astir2 (stirrups area) = 50.26548

-----  
Asec = 120000.00

s1 = 100.00

s2 = 250.00

fywe1 = 781.25

fywe2 = 625.00

fce = 30.00

From ((5.A5), TBDY), TBDY: cc = 0.002

c = confinement factor = 1.00

y1 = 0.0025

sh1 = 0.008

ft1 = 898.293

fy1 = 748.5775

su1 = 0.032

using (30) in Bisikinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

lo/lou,min = lb/ld = 1.00

su1 =  $0.4 * e_{s1\_nominal} ((5.5), TBDY) = 0.032$

From table 5A.1, TBDY: esu1\_nominal = 0.08,

For calculation of esu1\_nominal and y1, sh1, ft1, fy1, it is considered characteristic value fsy1 = fs1/1.2, from table 5.1, TBDY.

y1, sh1, ft1, fy1, are also multiplied by  $\text{Min}(1, 1.25 * (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs1 =  $(f_{s,jacket} * A_{s,ten,jacket} + f_{s,core} * A_{s,ten,core}) / A_{s,ten} = 748.5775$

with Es1 =  $(E_{s,jacket} * A_{s,ten,jacket} + E_{s,core} * A_{s,ten,core}) / A_{s,ten} = 200000.00$

y2 = 0.0025

sh2 = 0.008

ft2 = 886.7403

fy2 = 738.9503

su2 = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/lb,min = 1.00

su2 = 0.4\*esu2\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu2\_nominal = 0.08,

For calculation of esu2\_nominal and y2, sh2,ft2,fy2, it is considered  
characteristic value fsy2 = fs2/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fs2 = (fs,jacket\*Asl,com,jacket + fs,core\*Asl,com,core)/Asl,com = 738.9503

with Es2 = (Es,jacket\*Asl,com,jacket + Es,core\*Asl,com,core)/Asl,com = 200000.00

yv = 0.0025

shv = 0.008

ftv = 894.3662

fyv = 745.3052

suv = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/ld = 1.00

suv = 0.4\*esuv\_nominal ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esuv\_nominal = 0.08,

considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY

For calculation of esuv\_nominal and yv, shv,ftv,fyv, it is considered  
characteristic value fsyv = fsv/1.2, from table 5.1, TBDY.

y1, sh1,ft1,fy1, are also multiplied by  $\text{Min}(1, 1.25 \cdot (lb/ld)^{2/3})$ , from 10.3.5, ASCE 41-17.

with fsv = (fs,jacket\*Asl,mid,jacket + fs,mid\*Asl,mid,core)/Asl,mid = 745.3052

with Esv = (Es,jacket\*Asl,mid,jacket + Es,mid\*Asl,mid,core)/Asl,mid = 200000.00

1 = Asl,ten/(b\*d)\*(fs1/fc) = 0.7508007

2 = Asl,com/(b\*d)\*(fs2/fc) = 0.38164216

v = Asl,mid/(b\*d)\*(fsv/fc) = 0.67946571

and confined core properties:

b = 160.00

d = 347.00

d' = 13.00

fcc (5A.2, TBDY) = 30.00

cc (5A.5, TBDY) = 0.002

c = confinement factor = 1.00

1 = Asl,ten/(b\*d)\*(fs1/fc) = 0.99259314

2 = Asl,com/(b\*d)\*(fs2/fc) = 0.50454853

v = Asl,mid/(b\*d)\*(fsv/fc) = 0.898285

Case/Assumption: Unconfined full section - Steel rupture

' satisfies Eq. (4.3)

---->

v < vs,y2 - LHS eq.(4.5) is not satisfied

---->

v < vs,c - RHS eq.(4.5) is not satisfied

---->

Case/Assumption Rejected.

---->

New Case/Assumption: Unconfined full section - Spalling of concrete cover

' satisfies Eq. (4.4)

---->

v < s,y1 - LHS eq.(4.7) is not satisfied

---->

v < vc,y1 - RHS eq.(4.6) is not satisfied

---->

cu (4.11) = 0.80138156

MRC (4.18) = 5.4612E+008

---->

New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover

In expressions below, the following modifications have been made

- b, d, d' replaced by geometric parameters of the core: bo, do, d'o

- N, 1, 2, v normalised to bo\*do, instead of b\*d

- parameters of confined concrete, fcc, cc, used in lieu of fc, ecu

---->

Subcase: Rupture of tension steel

--->  
 $v^* < v^*s,y2$  - LHS eq.(4.5) is not satisfied  
 --->  
 $v^* < v^*s,c$  - LHS eq.(4.5) is not satisfied  
 --->  
 Subcase rejected  
 --->  
 New Subcase: Failure of compression zone  
 --->  
 $v^* < v^*c,y2$  - LHS eq.(4.6) is not satisfied  
 --->  
 $v^* < v^*c,y1$  - RHS eq.(4.6) is not satisfied  
 --->  
 $*cu$  (4.11) = 0.80674817  
 $MRO$  (4.18) = 2.8580E+008  
 $MRO < 0.8*MRc$   
 --->  
 $u = cu$  (unconfined full section) = 3.1475172E-005  
 $Mu = MRc$

-----  
 Calculation of ratio  $lb/d$

-----  
 Adequate Lap Length:  $lb/d \geq 1$   
 -----  
 -----  
 -----

-----  
 Calculation of  $Mu_{2+}$   
 -----  
 -----

-----  
 Calculation of ultimate curvature  $u$  according to 4.1, Biskinis/Fardis 2013:  
 $u = 1.2613948E-005$   
 $Mu = 6.0413E+008$   
 -----

with full section properties:

$b = 400.00$   
 $d = 367.00$   
 $d' = 33.00$   
 $v = 0.40931126$   
 $N = 1.8026E+006$   
 $fc = 30.00$   
 $co$  (5A.5, TBDY) = 0.002

Final value of  $cu$ :  $cu^* = shear\_factor * Max(cu, cc) = 0.00925707$

The Shear\_factor is considered equal to 1 (pure moment strength)

From (5.4b), TBDY:  $cu = 0.00925707$

$w_e$  (5.4c) = 0.0207149

$ase$  ((5.4d), TBDY) =  $(ase1 * A_{ext} + ase2 * A_{int}) / A_{sec} = 0.33836562$

$ase1 = Max(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length

equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $bi^2/6$  as defined at (A.2).

$ase2 (\geq ase1) = Max(((A_{conf,max2} - A_{noConf2}) / A_{conf,max2}) * (A_{conf,min2} / A_{conf,max2}), 0) = 0.33836562$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and

is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and

is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe}, p_{sh,y} * F_{ywe}) = 6.12205$

Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

-----  
 $p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{s2} * F_{ywe2} = 6.12205$

$p_{sh1}$  ((5.4d), TBDY) =  $L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$

$L_{stir1}$  (Length of stirrups along Y) = 1040.00

$A_{stir1}$  (stirrups area) = 78.53982

$p_{sh2}$  (5.4d) =  $L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$

$L_{stir2}$  (Length of stirrups along Y) = 768.00

$A_{stir2}$  (stirrups area) = 50.26548  
-----

$p_{sh,y} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{s2} * F_{ywe2} = 6.12205$

$p_{sh1}$  ((5.4d), TBDY) =  $L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$

$L_{stir1}$  (Length of stirrups along X) = 1040.00

$A_{stir1}$  (stirrups area) = 78.53982

$p_{sh2}$  ((5.4d), TBDY) =  $L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$

$L_{stir2}$  (Length of stirrups along X) = 768.00

$A_{stir2}$  (stirrups area) = 50.26548  
-----

$A_{sec} = 120000.00$

$s1 = 100.00$

$s2 = 250.00$

$f_{ywe1} = 781.25$

$f_{ywe2} = 625.00$

$f_{ce} = 30.00$

From ((5.A5), TBDY), TBDY:  $c_c = 0.002$

$c$  = confinement factor = 1.00

$y1 = 0.0025$

$sh1 = 0.008$

$ft1 = 886.7403$

$fy1 = 738.9503$

$su1 = 0.032$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

$l_o/l_{ou,min} = l_b/l_d = 1.00$

$su1 = 0.4 * esu1_{nominal}$  ((5.5), TBDY) = 0.032

From table 5A.1, TBDY:  $esu1_{nominal} = 0.08$ ,

For calculation of  $esu1_{nominal}$  and  $y1, sh1, ft1, fy1$ , it is considered characteristic value  $fsy1 = fs1/1.2$ , from table 5.1, TBDY.

$y1, sh1, ft1, fy1$ , are also multiplied by  $\text{Min}(1, 1.25 * (l_b/l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $fs1 = (f_{s,jacket} * A_{sl,ten,jacket} + f_{s,core} * A_{sl,ten,core}) / A_{sl,ten} = 738.9503$

with  $Es1 = (E_{s,jacket} * A_{sl,ten,jacket} + E_{s,core} * A_{sl,ten,core}) / A_{sl,ten} = 200000.00$

$y2 = 0.0025$

$sh2 = 0.008$

$ft2 = 898.293$

$fy2 = 748.5775$

$su2 = 0.032$

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor and also multiplied by the shear\_factor according to 15.7.1.4, with Shear\_factor = 1.00

$l_o/l_{ou,min} = l_b/l_{b,min} = 1.00$

$su2 = 0.4 * esu2_{nominal}$  ((5.5), TBDY) = 0.032

From table 5A.1, TBDY:  $esu2_{nominal} = 0.08$ ,

For calculation of  $esu2_{nominal}$  and  $y2, sh2, ft2, fy2$ , it is considered characteristic value  $fsy2 = fs2/1.2$ , from table 5.1, TBDY.

$y1, sh1, ft1, fy1$ , are also multiplied by  $\text{Min}(1, 1.25 * (l_b/l_d)^{2/3})$ , from 10.3.5, ASCE 41-17.

with  $fs2 = (f_{s,jacket} * A_{sl,com,jacket} + f_{s,core} * A_{sl,com,core}) / A_{sl,com} = 748.5775$

with  $Es2 = (E_{s,jacket} * A_{sl,com,jacket} + E_{s,core} * A_{sl,com,core}) / A_{sl,com} = 200000.00$

$yv = 0.0025$

$shv = 0.008$

```

ftv = 894.3662
fyv = 745.3052
suv = 0.032
  using (30) in Biskinis/Fardis (2013) multiplied with shear_factor
  and also multiplied by the shear_factor according to 15.7.1.4, with
  Shear_factor = 1.00
  lo/lo,min = lb/ld = 1.00
  suv = 0.4*esuv_nominal ((5.5), TBDY) = 0.032
  From table 5A.1, TBDY: esuv_nominal = 0.08,
  considering characteristic value fsyv = fsv/1.2, from table 5.1, TBDY
  For calculation of esuv_nominal and yv, shv,ftv,fyv, it is considered
  characteristic value fsyv = fsv/1.2, from table 5.1, TBDY.
  y1, sh1,ft1,fy1, are also multiplied by Min(1,1.25*(lb/ld)^2/3), from 10.3.5, ASCE 41-17.
  with fsv = (fs,jacket*Asl,mid,jacket + fs,mid*Asl,mid,core)/Asl,mid = 745.3052
  with Esv = (Es,jacket*Asl,mid,jacket + Es,mid*Asl,mid,core)/Asl,mid = 200000.00
  1 = Asl,ten/(b*d)*(fs1/fc) = 0.19082108
  2 = Asl,com/(b*d)*(fs2/fc) = 0.37540035
  v = Asl,mid/(b*d)*(fsv/fc) = 0.33973286
and confined core properties:
b = 360.00
d = 347.00
d' = 13.00
fcc (5A.2, TBDY) = 30.00
cc (5A.5, TBDY) = 0.002
  c = confinement factor = 1.00
  1 = Asl,ten/(b*d)*(fs1/fc) = 0.22424379
  2 = Asl,com/(b*d)*(fs2/fc) = 0.44115251
  v = Asl,mid/(b*d)*(fsv/fc) = 0.39923778
Case/Assumption: Unconfined full section - Steel rupture
' satisfies Eq. (4.3)
---->
v < vs,y2 - LHS eq.(4.5) is not satisfied
---->
v < vs,c - RHS eq.(4.5) is not satisfied
---->
Case/Assumption Rejected.
---->
New Case/Assumption: Unconfined full section - Spalling of concrete cover
' satisfies Eq. (4.4)
---->
v < s,y1 - LHS eq.(4.7) is not satisfied
---->
v < vc,y1 - RHS eq.(4.6) is satisfied
---->
  cu (4.10) = 0.37716748
  MRc (4.17) = 7.0585E+008
---->
New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover
In expressions below, the following modifications have been made
- b, d, d' replaced by geometric parameters of the core: bo, do, d'o
- N, 1, 2, v normalised to bo*do, instead of b*d
- - parameters of confined concrete, fcc, cc, used in lieu of fc, ecu
---->
Subcase: Rupture of tension steel
---->
v* < v*s,y2 - LHS eq.(4.5) is not satisfied
---->
v* < v*s,c - LHS eq.(4.5) is not satisfied
---->
Subcase rejected
---->
New Subcase: Failure of compression zone
---->
v* < v*c,y2 - LHS eq.(4.6) is not satisfied
---->
v* < v*c,y1 - RHS eq.(4.6) is not satisfied
---->

```

$$*c_u(4.11) = 0.4569298$$
$$M_{Ro}(4.18) = 6.0413E+008$$

--->

$$u = c_u(4.2) = 1.2613948E-005$$
$$\mu = M_{Ro}$$

-----

Calculation of ratio  $l_b/l_d$

-----

Adequate Lap Length:  $l_b/l_d \geq 1$

-----

-----

Calculation of  $\mu_2$ -

-----

-----

Calculation of ultimate curvature  $u$  according to 4.1, Biskinis/Fardis 2013:

$$u = 3.1475172E-005$$
$$\mu = 5.4612E+008$$

-----

with full section properties:

$$b = 200.00$$
$$d = 367.00$$
$$d' = 33.00$$
$$v = 0.81862252$$
$$N = 1.8026E+006$$

$$f_c = 30.00$$

$$c_o(5A.5, TBDY) = 0.002$$

$$\text{Final value of } c_u: c_u^* = \text{shear\_factor} * \text{Max}(c_u, c_c) = 0.00925707$$

The Shear\_factor is considered equal to 1 (pure moment strength)

$$\text{From (5.4b), TBDY: } c_u = 0.00925707$$

$$w_e(5.4c) = 0.0207149$$

$$a_{se}((5.4d), TBDY) = (a_{se1} * A_{ext} + a_{se2} * A_{int}) / A_{sec} = 0.33836562$$

$$a_{se1} = \text{Max}(((A_{conf,max1} - A_{noConf1}) / A_{conf,max1}) * (A_{conf,min1} / A_{conf,max1}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max1} = 89600.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the external perimeter hoops.

$A_{conf,min1} = 59225.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max1}$  by a length equal to half the clear spacing between external hoops.

$A_{noConf1} = 43733.333$  is the unconfined external core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$a_{se2} (\geq a_{se1}) = \text{Max}(((A_{conf,max2} - A_{noConf2}) / A_{conf,max2}) * (A_{conf,min2} / A_{conf,max2}), 0) = 0.33836562$$

The definitions of  $A_{noConf}$ ,  $A_{conf,min}$  and  $A_{conf,max}$  are derived from generalization of the rectangular sections confinement, which is expressed by (5.4d).

The generalization is done according to Mander, J., Priestley, M., and Park, R. (1988)

"Theoretical Stress-Strain Model for Confined Concrete."

J. Struct. Eng., 10.1061/(ASCE)0733-9445(1988)114:8(1804), 1804-1826.

$A_{conf,max2} = 45264.00$  is the confined core area at levels of member with hoops and is calculated as the area of core enclosed by the center lines of the internal perimeter hoops.

$A_{conf,min2} = 0.00$  is the confined core area at midway between the levels of hoops and is calculated by reducing all the dimensions of the area  $A_{conf,max2}$  by a length equal to half the clear spacing between internal hoops.

$A_{noConf2} = 32309.333$  is the unconfined internal core area which is equal to  $b_i^2/6$  as defined at (A.2).

$$p_{sh,min} * F_{ywe} = \text{Min}(p_{sh,x} * F_{ywe}, p_{sh,y} * F_{ywe}) = 6.12205$$

Expression (5.4d) for  $p_{sh,min} * F_{ywe}$  has been multiplied by 0.3 according to 15.7.1.3 for members without earthquake detailing (90° closed stirrups)

-----

$$p_{sh,x} * F_{ywe} = p_{sh1} * F_{ywe1} + p_{sh2} * F_{ywe2} = 6.12205$$

$$p_{sh1}((5.4d), TBDY) = L_{stir1} * A_{stir1} / (A_{sec} * s1) = 0.00680678$$

$$L_{stir1}(\text{Length of stirrups along Y}) = 1040.00$$

$$A_{stir1}(\text{stirrups area}) = 78.53982$$

$$p_{sh2}(5.4d) = L_{stir2} * A_{stir2} / (A_{sec} * s2) = 0.0012868$$

Lstir2 (Length of stirrups along Y) = 768.00  
Astir2 (stirrups area) = 50.26548

psh<sub>y</sub>\*F<sub>ywe</sub> = psh<sub>1</sub>\*F<sub>ywe1</sub>+ps<sub>2</sub>\*F<sub>ywe2</sub> = 6.12205  
psh<sub>1</sub> ((5.4d), TBDY) = Lstir<sub>1</sub>\*Astir<sub>1</sub>/(Asec\*s<sub>1</sub>) = 0.00680678  
Lstir<sub>1</sub> (Length of stirrups along X) = 1040.00  
Astir<sub>1</sub> (stirrups area) = 78.53982  
psh<sub>2</sub> ((5.4d), TBDY) = Lstir<sub>2</sub>\*Astir<sub>2</sub>/(Asec\*s<sub>2</sub>) = 0.0012868  
Lstir<sub>2</sub> (Length of stirrups along X) = 768.00  
Astir<sub>2</sub> (stirrups area) = 50.26548

Asec = 120000.00  
s<sub>1</sub> = 100.00  
s<sub>2</sub> = 250.00

f<sub>ywe1</sub> = 781.25  
f<sub>ywe2</sub> = 625.00  
f<sub>ce</sub> = 30.00

From ((5.A5), TBDY), TBDY: cc = 0.002  
c = confinement factor = 1.00

y<sub>1</sub> = 0.0025  
sh<sub>1</sub> = 0.008  
ft<sub>1</sub> = 898.293  
fy<sub>1</sub> = 748.5775  
su<sub>1</sub> = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/l<sub>d</sub> = 1.00

su<sub>1</sub> = 0.4\*esu<sub>1\_nominal</sub> ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu<sub>1\_nominal</sub> = 0.08,

For calculation of esu<sub>1\_nominal</sub> and y<sub>1</sub>, sh<sub>1</sub>,ft<sub>1</sub>,fy<sub>1</sub>, it is considered  
characteristic value fs<sub>y1</sub> = fs<sub>1</sub>/1.2, from table 5.1, TBDY.

y<sub>1</sub>, sh<sub>1</sub>,ft<sub>1</sub>,fy<sub>1</sub>, are also multiplied by Min(1,1.25\*(lb/l<sub>d</sub>)<sup>2/3</sup>), from 10.3.5, ASCE 41-17.

with fs<sub>1</sub> = (fs<sub>jacket</sub>\*As<sub>l,ten,jacket</sub> + fs<sub>core</sub>\*As<sub>l,ten,core</sub>)/As<sub>l,ten</sub> = 748.5775

with Es<sub>1</sub> = (Es<sub>jacket</sub>\*As<sub>l,ten,jacket</sub> + Es<sub>core</sub>\*As<sub>l,ten,core</sub>)/As<sub>l,ten</sub> = 200000.00

y<sub>2</sub> = 0.0025

sh<sub>2</sub> = 0.008

ft<sub>2</sub> = 886.7403

fy<sub>2</sub> = 738.9503

su<sub>2</sub> = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/l<sub>b,min</sub> = 1.00

su<sub>2</sub> = 0.4\*esu<sub>2\_nominal</sub> ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esu<sub>2\_nominal</sub> = 0.08,

For calculation of esu<sub>2\_nominal</sub> and y<sub>2</sub>, sh<sub>2</sub>,ft<sub>2</sub>,fy<sub>2</sub>, it is considered  
characteristic value fs<sub>y2</sub> = fs<sub>2</sub>/1.2, from table 5.1, TBDY.

y<sub>2</sub>, sh<sub>2</sub>,ft<sub>2</sub>,fy<sub>2</sub>, are also multiplied by Min(1,1.25\*(lb/l<sub>d</sub>)<sup>2/3</sup>), from 10.3.5, ASCE 41-17.

with fs<sub>2</sub> = (fs<sub>jacket</sub>\*As<sub>l,com,jacket</sub> + fs<sub>core</sub>\*As<sub>l,com,core</sub>)/As<sub>l,com</sub> = 738.9503

with Es<sub>2</sub> = (Es<sub>jacket</sub>\*As<sub>l,com,jacket</sub> + Es<sub>core</sub>\*As<sub>l,com,core</sub>)/As<sub>l,com</sub> = 200000.00

y<sub>v</sub> = 0.0025

sh<sub>v</sub> = 0.008

ft<sub>v</sub> = 894.3662

fy<sub>v</sub> = 745.3052

suv = 0.032

using (30) in Biskinis/Fardis (2013) multiplied with shear\_factor  
and also multiplied by the shear\_factor according to 15.7.1.4, with  
Shear\_factor = 1.00

lo/lou,min = lb/l<sub>d</sub> = 1.00

suv = 0.4\*esuv<sub>nominal</sub> ((5.5), TBDY) = 0.032

From table 5A.1, TBDY: esuv<sub>nominal</sub> = 0.08,

considering characteristic value fs<sub>yv</sub> = fs<sub>v</sub>/1.2, from table 5.1, TBDY  
For calculation of esuv<sub>nominal</sub> and y<sub>v</sub>, sh<sub>v</sub>,ft<sub>v</sub>,fy<sub>v</sub>, it is considered

characteristic value fs<sub>yv</sub> = fs<sub>v</sub>/1.2, from table 5.1, TBDY.

y<sub>1</sub>, sh<sub>1</sub>,ft<sub>1</sub>,fy<sub>1</sub>, are also multiplied by Min(1,1.25\*(lb/l<sub>d</sub>)<sup>2/3</sup>), from 10.3.5, ASCE 41-17.

with fs<sub>v</sub> = (fs<sub>jacket</sub>\*As<sub>l,mid,jacket</sub> + fs<sub>mid</sub>\*As<sub>l,mid,core</sub>)/As<sub>l,mid</sub> = 745.3052

```

with Esv = (Es,jacket*Asl,mid,jacket + Es,mid*Asl,mid,core)/Asl,mid = 200000.00
1 = Asl,ten/(b*d)*(fs1/fc) = 0.7508007
2 = Asl,com/(b*d)*(fs2/fc) = 0.38164216
v = Asl,mid/(b*d)*(fsv/fc) = 0.67946571
and confined core properties:
b = 160.00
d = 347.00
d' = 13.00
fcc (5A.2, TBDY) = 30.00
cc (5A.5, TBDY) = 0.002
c = confinement factor = 1.00
1 = Asl,ten/(b*d)*(fs1/fc) = 0.99259314
2 = Asl,com/(b*d)*(fs2/fc) = 0.50454853
v = Asl,mid/(b*d)*(fsv/fc) = 0.898285
Case/Assumption: Unconfined full section - Steel rupture
' satisfies Eq. (4.3)
---->
v < vs,y2 - LHS eq.(4.5) is not satisfied
---->
v < vs,c - RHS eq.(4.5) is not satisfied
---->
Case/Assumption Rejected.
---->
New Case/Assumption: Unconfined full section - Spalling of concrete cover
' satisfies Eq. (4.4)
---->
v < s,y1 - LHS eq.(4.7) is not satisfied
---->
v < vc,y1 - RHS eq.(4.6) is not satisfied
---->
cu (4.11) = 0.80138156
MRc (4.18) = 5.4612E+008
---->
New Case/Assumption: Ultimate curvature of confined core after spalling of concrete cover
In expressions below, the following modifications have been made
- b, d, d' replaced by geometric parameters of the core: bo, do, d'o
- N, 1, 2, v normalised to bo*do, instead of b*d
- - parameters of confined concrete, fcc, cc, used in lieu of fc, ecu
---->
Subcase: Rupture of tension steel
---->
v* < v*s,y2 - LHS eq.(4.5) is not satisfied
---->
v* < v*s,c - LHS eq.(4.5) is not satisfied
---->
Subcase rejected
---->
New Subcase: Failure of compression zone
---->
v* < v*c,y2 - LHS eq.(4.6) is not satisfied
---->
v* < v*c,y1 - RHS eq.(4.6) is not satisfied
---->
*cu (4.11) = 0.80674817
MRo (4.18) = 2.8580E+008
MRo < 0.8*MRc
---->
u = cu (unconfined full section) = 3.1475172E-005
Mu = MRc

```

-----

Calculation of ratio lb/d

-----

Adequate Lap Length: lb/d >= 1

-----

-----

-----

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 339150.739$

Calculation of Shear Strength at edge 1,  $V_{r1} = 339150.739$

$V_{r1} = V_{\text{Col}} ((10.3), \text{ASCE } 41-17) = k_{nl} * V_{\text{Col}0}$

$V_{\text{Col}0} = 339150.739$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

Mean concrete strength:  $f_c' = (f_c'_{\text{jacket}} * \text{Area}_{\text{jacket}} + f_c'_{\text{core}} * \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 25.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M / Vd = 4.00$

$\mu_u = 6.7415E+007$

$\nu_u = 3.6347847E-006$

$d = 0.8 * h = 320.00$

$N_u = 1.8026E+006$

$A_g = 80000.00$

From (11.5.4.8), ACI 318-14:  $V_s = V_{s,\text{jacket}} + V_{s,\text{core}} = 471238.898$

where:

$V_{s,\text{jacket}} = V_{s,j1} + V_{s,j2} = 471238.898$

$V_{s,j1} = 157079.633$  is calculated for section web jacket, with:

$d = 160.00$

$A_v = 157079.633$

$f_y = 625.00$

$s = 100.00$

$V_{s,j1}$  is multiplied by  $\text{Col},j1 = 1.00$

$s/d = 0.625$

$V_{s,j2} = 314159.265$  is calculated for section flange jacket, with:

$d = 320.00$

$A_v = 157079.633$

$f_y = 625.00$

$s = 100.00$

$V_{s,j2}$  is multiplied by  $\text{Col},j2 = 1.00$

$s/d = 0.3125$

$V_{s,\text{core}} = V_{s,c1} + V_{s,c2} = 0.00$

$V_{s,c1} = 0.00$  is calculated for section web core, with:

$d = 80.00$

$A_v = 100530.965$

$f_y = 500.00$

$s = 250.00$

$V_{s,c1}$  is multiplied by  $\text{Col},c1 = 0.00$

$s/d = 3.125$

$V_{s,c2} = 0.00$  is calculated for section flange core, with:

$d = 240.00$

$A_v = 100530.965$

$f_y = 500.00$

$s = 250.00$

$V_{s,c2}$  is multiplied by  $\text{Col},c2 = 0.00$

$s/d = 1.04167$

$V_f ((11-3)-(11.4), \text{ACI } 440) = 0.00$

From (11-11), ACI 440:  $V_s + V_f \leq 212577.225$

$bw = 200.00$

Calculation of Shear Strength at edge 2,  $V_{r2} = 339150.739$

$V_{r2} = V_{\text{Col}} ((10.3), \text{ASCE } 41-17) = k_{nl} * V_{\text{Col}0}$

$V_{\text{Col}0} = 339150.739$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

Mean concrete strength:  $f_c' = (f_c'_{\text{jacket}} * \text{Area}_{\text{jacket}} + f_c'_{\text{core}} * \text{Area}_{\text{core}}) / \text{Area}_{\text{section}} = 25.00$ , but  $f_c'^{0.5} \leq 8.3$

MPa (22.5.3.1, ACI 318-14)

$$M/Vd = 4.00$$

$$\mu_u = 5.3429E+007$$

$$V_u = 3.6347847E-006$$

$$d = 0.8 \cdot h = 320.00$$

$$N_u = 1.8026E+006$$

$$A_g = 80000.00$$

$$\text{From (11.5.4.8), ACI 318-14: } V_s = V_{s,jacket} + V_{s,core} = 471238.898$$

where:

$$V_{s,jacket} = V_{s,j1} + V_{s,j2} = 471238.898$$

$V_{s,j1} = 157079.633$  is calculated for section web jacket, with:

$$d = 160.00$$

$$A_v = 157079.633$$

$$f_y = 625.00$$

$$s = 100.00$$

$V_{s,j1}$  is multiplied by  $Col,j1 = 1.00$

$$s/d = 0.625$$

$V_{s,j2} = 314159.265$  is calculated for section flange jacket, with:

$$d = 320.00$$

$$A_v = 157079.633$$

$$f_y = 625.00$$

$$s = 100.00$$

$V_{s,j2}$  is multiplied by  $Col,j2 = 1.00$

$$s/d = 0.3125$$

$$V_{s,core} = V_{s,c1} + V_{s,c2} = 0.00$$

$V_{s,c1} = 0.00$  is calculated for section web core, with:

$$d = 80.00$$

$$A_v = 100530.965$$

$$f_y = 500.00$$

$$s = 250.00$$

$V_{s,c1}$  is multiplied by  $Col,c1 = 0.00$

$$s/d = 3.125$$

$V_{s,c2} = 0.00$  is calculated for section flange core, with:

$$d = 240.00$$

$$A_v = 100530.965$$

$$f_y = 500.00$$

$$s = 250.00$$

$V_{s,c2}$  is multiplied by  $Col,c2 = 0.00$

$$s/d = 1.04167$$

$$V_f \text{ ((11-3)-(11.4), ACI 440)} = 0.00$$

$$\text{From (11-11), ACI 440: } V_s + V_f \leq 212577.225$$

$$b_w = 200.00$$

-----  
End Of Calculation of Shear Capacity ratio for element: column JLC1 of floor 1

At local axis: 2  
-----

Start Of Calculation of Chord Rotation Capacity for element: column JLC1 of floor 1

At local axis: 2

Integration Section: (b)

Section Type: rcjlcs

Constant Properties

-----  
Knowledge Factor,  $\phi = 0.90$

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE 41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Jacket

New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 30.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 625.00$

Concrete Elasticity,  $E_c = 25742.96$

Steel Elasticity,  $E_s = 200000.00$

Existing Column

Existing material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 18.00$   
 Existing material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 500.00$   
 Concrete Elasticity,  $E_c = 19940.411$   
 Steel Elasticity,  $E_s = 200000.00$   
 Max Height,  $H_{max} = 400.00$   
 Min Height,  $H_{min} = 200.00$   
 Max Width,  $W_{max} = 400.00$   
 Min Width,  $W_{min} = 200.00$   
 Jacket Thickness,  $t_j = 50.00$   
 Cover Thickness,  $c = 15.00$   
 Element Length,  $L = 3000.00$   
 Secondary Member  
 Smooth Bars  
 Ductile Steel  
 Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Adequate Lap Length ( $l_b/l_d \geq 1$ )  
 No FRP Wrapping

-----  
 Stepwise Properties

-----  
 Bending Moment,  $M = 246078.073$   
 Shear Force,  $V_2 = 62256.122$   
 Shear Force,  $V_3 = 174.4443$   
 Axial Force,  $F = -1.8013E+006$   
 Longitudinal Reinforcement Area Distribution (in 2 divisions)  
 -Tension:  $A_{sl,t} = 0.00$   
 -Compression:  $A_{sl,c} = 5353.274$   
 Longitudinal Reinforcement Area Distribution (in 3 divisions)  
 -Tension:  $A_{sl,ten} = 2208.54$   
 -Compression:  $A_{sl,com} = 1137.257$   
 -Middle:  $A_{sl,mid} = 2007.478$   
 Longitudinal External Reinforcement Area Distribution (in 3 divisions)  
 -Tension:  $A_{sl,ten,jacket} = 1746.726$   
 -Compression:  $A_{sl,com,jacket} = 829.3805$   
 -Middle:  $A_{sl,mid,jacket} = 1545.664$   
 Longitudinal Internal Reinforcement Area Distribution (in 3 divisions)  
 -Tension:  $A_{sl,ten,core} = 461.8141$   
 -Compression:  $A_{sl,com,core} = 307.8761$   
 -Middle:  $A_{sl,mid,core} = 461.8141$   
 Mean Diameter of Tension Reinforcement,  $DbL = 16.60$

-----  
 Existing component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_{R,u} = u = 0.02030057$   
 $u = y + p = 0.02255618$

-----  
 - Calculation of  $y$  -

-----  
 $y = (M_y * L_s / 3) / E_{eff} = 0.00985105$  ((4.29), Biskinis Phd)  
 $M_y = 5.2274E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 1410.64  
 From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 2.4951E+013$   
 $factor = 0.70$   
 $A_g = 120000.00$   
 Mean concrete strength:  $f'_c = (f'_{c,jacket} * Area_{jacket} + f'_{c,core} * Area_{core}) / Area_{section} = 25.00$   
 $N = 1.8013E+006$   
 $E_c * I_g = E_{c,jacket} * I_{g,jacket} + E_{c,core} * I_{g,core} = 3.5645E+013$

-----  
 Calculation of Yielding Moment  $M_y$

-----  
 Calculation of  $y$  and  $M_y$  according to Annex 7 -

Assuming neutral axis out of flange ( $y > t/d$ , compression zone NOT rectangular)  
extended expressions (2.1)-(2.7) of Biskinis-Fardis (2009) are adopted with:

flange width,  $b = 400.00$   
web width,  $b_w = 200.00$   
flange thickness,  $t = 200.00$

$y = \text{Min}(y_{\text{ten}}, y_{\text{com}})$   
 $y_{\text{ten}} = 1.6046235\text{E-}005$   
with  $f_y = 596.2441$   
 $d = 367.00$   
 $y = 0.49376179$   
 $A = 0.18423709$   
 $B = 0.10665962$   
with  $p_t = 0.01504455$   
 $p_c = 0.00774698$   
 $p_v = 0.01367492$   
 $N = 1.8013\text{E}+006$   
 $b = 400.00$   
 $" = 0.08991826$   
 $y_{\text{comp}} = 1.0451884\text{E-}005$   
with  $f_c = 30.00$   
 $E_c = 25742.96$   
 $y = 0.54685808$   
 $A = 0.08458025$   
 $B = 0.06549972$   
with  $E_s = 200000.00$   
CONFIRMATION:  $y = 0.54685808 > t/d$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

- Calculation of  $p$  -

From table 10-8:  $p = 0.01270514$

with:

- Columns not controlled by inadequate development or splicing along the clear height because  $l_b/d \geq 1$

shear control ratio  $V_y E / V_{CoI} O E = 1.44151$

$d = d_{\text{external}} = 367.00$

$s = s_{\text{external}} = 100.00$

-  $t = s_1 + s_2 + 2 \cdot t_f / b_w \cdot (f_{fe} / f_s) = 0.00809358$

jacket:  $s_1 = A_{v1} \cdot L_{\text{stir}1} / (s_1 \cdot A_g) = 0.00680678$

$A_{v1} = 78.53982$ , is the area of every stirrup parallel to loading (shear) direction

$L_{\text{stir}1} = 1040.00$ , is the total Length of all stirrups parallel to loading (shear) direction

$s_1 = 100.00$

core:  $s_2 = A_{v2} \cdot L_{\text{stir}2} / (s_2 \cdot A_g) = 0.0012868$

$A_{v2} = 50.26548$ , is the area of every stirrup parallel to loading (shear) direction

$L_{\text{stir}2} = 768.00$ , is the total Length of all stirrups parallel to loading (shear) direction

$s_2 = 250.00$

The term  $2 \cdot t_f / b_w \cdot (f_{fe} / f_s)$  is implemented to account for FRP contribution

where  $f = 2 \cdot t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength

All these variables have already been given in Shear control ratio calculation.

For the normalisation  $f_s$  of jacket is used.

$N_{UD} = 1.8013\text{E}+006$

$A_g = 120000.00$

$f_{cE} = (f_{c,jacket} \cdot \text{Area}_{jacket} + f_{c,core} \cdot \text{Area}_{core}) / \text{section\_area} = 25.00$

$f_{yE} = (f_{y,ext\_Long\_Reinf} \cdot \text{Area}_{ext\_Long\_Reinf} + f_{y,int\_Long\_Reinf} \cdot \text{Area}_{int\_Long\_Reinf}) / \text{Area}_{Tot\_Long\_Rein} = 596.2441$

$f_{yE} = (f_{y,ext\_Trans\_Reinf} \cdot s_1 + f_{y,int\_Trans\_Reinf} \cdot s_2) / (s_1 + s_2) = 605.1263$

$p_l = \text{Area}_{Tot\_Long\_Rein} / (b \cdot d) = 0.03646644$

$b = 400.00$

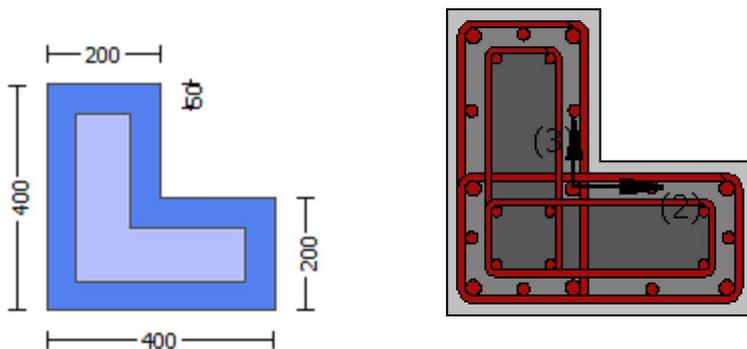
$d = 367.00$

$f_{cE} = 25.00$

End Of Calculation of Chord Rotation Capacity for element: column JLC1 of floor 1  
At local axis: 2  
Integration Section: (b)

## Calculation No. 15

column C1, Floor 1  
Limit State: Life Safety (data interpolation between analysis steps 50 and 51)  
Analysis: Uniform +X  
Check: Shear capacity VRd  
Edge: End  
Local Axis: (3)



Start Of Calculation of Shear Capacity for element: column JLC1 of floor 1  
At local axis: 3  
Integration Section: (b)  
Section Type: rcjcs

Constant Properties

Knowledge Factor,  $\gamma = 0.90$   
Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE 41-17.  
Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17  
Consequently:  
Jacket  
New material of Secondary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 20.00$   
New material of Secondary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 500.00$   
Concrete Elasticity,  $E_c = 25742.96$   
Steel Elasticity,  $E_s = 200000.00$   
Existing Column  
Existing material of Secondary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 12.00$   
Existing material of Secondary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 400.00$   
Concrete Elasticity,  $E_c = 19940.411$   
Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of  $\mu_y$  for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE 41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE 41-17).

Jacket

New material: Concrete Strength,  $f_c = f_{cm} = 30.00$

New material: Steel Strength,  $f_s = f_{sm} = 625.00$

Existing Column

Existing material: Concrete Strength,  $f_c = f_{cm} = 18.00$

Existing material: Steel Strength,  $f_s = f_{sm} = 500.00$

#####

Max Height,  $H_{max} = 400.00$

Min Height,  $H_{min} = 200.00$

Max Width,  $W_{max} = 400.00$

Min Width,  $W_{min} = 200.00$

Jacket Thickness,  $t_j = 50.00$

Cover Thickness,  $c = 15.00$

Element Length,  $L = 3000.00$

Secondary Member

Smooth Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou,min} = l_b/l_d \geq 1$ )

No FRP Wrapping

-----  
Stepwise Properties

EDGE -A-

Bending Moment,  $M_a = 1.1948E+007$

Shear Force,  $V_a = -174.4443$

EDGE -B-

Bending Moment,  $M_b = 246078.073$

Shear Force,  $V_b = 174.4443$

BOTH EDGES

Axial Force,  $F = -1.8013E+006$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 0.00$

-Compression:  $As_c = 5353.274$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t,ten} = 2208.54$

-Compression:  $As_{c,com} = 1137.257$

-Middle:  $As_{mid} = 2007.478$

Mean Diameter of Tension Reinforcement,  $Db_{L,ten} = 16.60$

-----  
Existing component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = \phi V_n = 258166.022$

$V_n$  ((10.3), ASCE 41-17) =  $k_n l V_{CoI} = 286851.135$

$V_{CoI} = 286851.135$

$k_n l = 1.00$

displacement\_ductility\_demand = 0.01679263

-----  
NOTE: In expression (10-3) ' $V_s = A_v f_y d/s$ ' is replaced by ' $V_s + \phi V_f$ '

where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

-----  
 $\phi = 1$  (normal-weight concrete)

Mean concrete strength:  $f'_c = (f'_{c,jacket} \text{Area}_{jacket} + f'_{c,core} \text{Area}_{core}) / \text{Area}_{section} = 16.66667$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 4.00$

$M_u = 246078.073$

$V_u = 174.4443$

$d = 0.8 \cdot h = 320.00$

$N_u = 1.8013E+006$

$A_g = 80000.00$

From (11.5.4.8), ACI 318-14:  $V_s = V_{s,jacket} + V_{s,core} = 376991.118$

where:

$V_{s,jacket} = V_{s,j1} + V_{s,j2} = 376991.118$

$V_{s,j1} = 251327.412$  is calculated for section web jacket, with:

$d = 320.00$

$A_v = 157079.633$

$f_y = 500.00$

$s = 100.00$

$V_{s,j1}$  is multiplied by  $Col,j1 = 1.00$

$s/d = 0.3125$

$V_{s,j2} = 125663.706$  is calculated for section flange jacket, with:

$d = 160.00$

$A_v = 157079.633$

$f_y = 500.00$

$s = 100.00$

$V_{s,j2}$  is multiplied by  $Col,j2 = 1.00$

$s/d = 0.625$

$V_{s,core} = V_{s,c1} + V_{s,c2} = 0.00$

$V_{s,c1} = 0.00$  is calculated for section web core, with:

$d = 240.00$

$A_v = 100530.965$

$f_y = 400.00$

$s = 250.00$

$V_{s,c1}$  is multiplied by  $Col,c1 = 0.00$

$s/d = 1.04167$

$V_{s,c2} = 0.00$  is calculated for section flange core, with:

$d = 80.00$

$A_v = 100530.965$

$f_y = 400.00$

$s = 250.00$

$V_{s,c2}$  is multiplied by  $Col,c2 = 0.00$

$s/d = 3.125$

$V_f$  ((11-3)-(11.4), ACI 440) = 0.00

From (11-11), ACI 440:  $V_s + V_f \leq 173568.578$

$bw = 200.00$

displacement ductility demand is calculated as  $\delta / y$

- Calculation of  $\delta / y$  for END B -

for rotation axis 2 and integ. section (b)

From analysis, chord rotation = 0.00016542

$y = (M_y * L_s / 3) / E_{eff} = 0.00985105$  ((4.29), Biskinis Phd))

$M_y = 5.2274E+008$

$L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 1410.64

From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 2.4951E+013$

factor = 0.70

$A_g = 120000.00$

Mean concrete strength:  $f'_c = (f'_{c,jacket} * Area_{jacket} + f'_{c,core} * Area_{core}) / Area_{section} = 25.00$

$N = 1.8013E+006$

$E_c * I_g = E_{c,jacket} * I_{g,jacket} + E_{c,core} * I_{g,core} = 3.5645E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\delta / y$  and  $M_y$  according to Annex 7 -

Assuming neutral axis out of flange ( $y > t/d$ , compression zone NOT rectangular)

extended expressions (2.1)-(2.7) of Biskinis-Fardis (2009) are adopted with:

flange width,  $b = 400.00$

web width,  $bw = 200.00$

flange thickness,  $t = 200.00$

$y = \text{Min}(y_{ten}, y_{com})$

y\_ten = 1.6046235E-005  
with fy = 596.2441  
d = 367.00  
y = 0.49376179  
A = 0.18423709  
B = 0.10665962  
with pt = 0.01504455  
pc = 0.00774698  
pv = 0.01367492  
N = 1.8013E+006  
b = 400.00  
" = 0.08991826  
y\_comp = 1.0451884E-005  
with fc = 30.00  
Ec = 25742.96  
y = 0.54685808  
A = 0.08458025  
B = 0.06549972  
with Es = 200000.00  
CONFIRMATION:  $y = 0.54685808 > t/d$

---

Calculation of ratio  $l_b/d$

---

Adequate Lap Length:  $l_b/d \geq 1$

---

End Of Calculation of Shear Capacity for element: column JLC1 of floor 1  
At local axis: 3  
Integration Section: (b)

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