

# Detailed Member Calculations

**Units: N&mm**

**Regulation: ASCE 41-17**

## Calculation No. 1

column C1, Floor 1

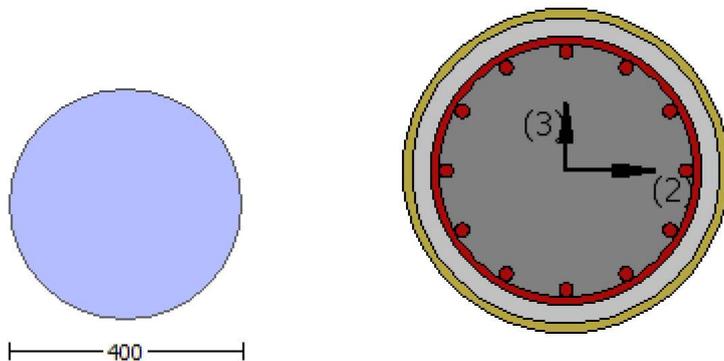
Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Shear capacity VRd

Edge: Start

Local Axis: (2)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (a)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 0.90$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

New material of Secondary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 25.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 500.00$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

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Note: Especially for the calculation of  $\gamma$  for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE41-17).

New material: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material: Steel Strength,  $f_s = f_{sm} = 555.56$

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Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Secondary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{o,min} = l_b/l_d \geq 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $NoDir = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

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Stepwise Properties

EDGE -A-

Bending Moment,  $M_a = -1.1282E+007$

Shear Force,  $V_a = -3759.209$

EDGE -B-

Bending Moment,  $M_b = 0.06021356$

Shear Force,  $V_b = 3759.209$

BOTH EDGES

Axial Force,  $F = -4769.729$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 1272.345$

-Compression:  $As_c = 1781.283$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{l,ten} = 1017.876$

-Compression:  $As_{l,com} = 1017.876$

-Middle:  $As_{l,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $Db_{L,ten} = 18.00$

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New component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = 1.0 \cdot V_n = 330216.75$

$V_n$  ((10.3), ASCE 41-17) =  $knI \cdot V_{CoI} = 330216.75$

$V_{CoI} = 330216.75$

$knI = 1.00$

$displacement\_ductility\_demand = 0.01018442$

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NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

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 $= 1$  (normal-weight concrete)

$f_c' = 25.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 4.00$

$M_u = 1.1282E+007$

$V_u = 3759.209$

$d = 0.8 \cdot D = 320.00$

$N_u = 4769.729$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 197392.088$   
 $A_v = \frac{1}{2} A_{stirrup} = 123370.055$   
 $f_y = 500.00$   
 $s = 100.00$   
 $V_s$  is multiplied by  $Col = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin^2 + \cos^2$  is replaced with  $(\cot^2 + \csc^2)\sin^2 \alpha$  which is more a generalised expression,  
 where  $\alpha$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ ,  
 as well as for 2 crack directions,  $\alpha = 45^\circ$  and  $\alpha = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\alpha_1 = b_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \alpha_1)|, |V_f(-45, \alpha_1)|)$ , with:  
 total thickness per orientation,  $tf_1 = NL * t / NoDir = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}$  ((11-5), ACI 440) = 259.312  
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 267132.42$   
 $bw * d = A_v * d / 4 = 80424.772$

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 displacement ductility demand is calculated as  $\delta / y$

- Calculation of  $\delta / y$  for END A -  
 for rotation axis 3 and integ. section (a)

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 From analysis, chord rotation  $\theta = 0.00030347$   
 $y = (M_y * L_s / 3) / E_{eff} = 0.02979765$  ((4.29), Biskinis Phd))  
 $M_y = 3.0318E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 3001.151  
 From table 10.5, ASCE 41\_17:  $E_{eff} = \text{factor} * E_c * I_g = 1.0179E+013$   
 factor = 0.30  
 $A_g = 125663.706$   
 $f_c' = 33.00$   
 $N = 4769.729$   
 $E_c * I_g = 3.3929E+013$

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 Calculation of Yielding Moment  $M_y$

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 Calculation of  $\delta / y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

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 $M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 3.0318E+008$   
 $y$  ((10a) or (10b)) = 1.6222680E-005  
 $M_{y\_ten}$  (8a) = 3.0318E+008  
 $y_{ten}$  (7a) = 69.35397  
 error of function (7a) = 0.00743258  
 $M_{y\_com}$  (8b) = 4.6746E+008  
 $y_{com}$  (7b) = 67.1564  
 error of function (7b) = -0.00265541  
 with  $e_y = 0.0027778$   
 $e_{co} = 0.002$   
 $\alpha_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00102226$   
 $N = 4769.729$   
 $A_c = 125663.706$   
 $\alpha = 0.36359274$

with  $f_c^*$  ((12.3), ACI 440) = 37.12975  
 $f_c = 33.00$   
 $f_l = 1.3173$   
 $k = 1$   
Effective FRP thickness,  $t_f = NL*t*\cos(b1) = 1.016$   
 $e_{fe}$  ((12.5) and (12.7)) = 0.004  
 $E_f = 64828.00$

Calculation of ratio  $l_b/l_d$

Adequate Lap Length:  $l_b/l_d \geq 1$

End Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (a)

## Calculation No. 2

column C1, Floor 1

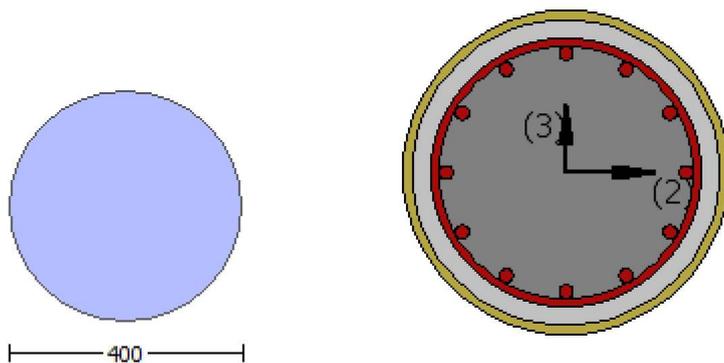
Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity (  $\theta$  )

Edge: Start

Local Axis: (2)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 0.90$

Mean strength values are used for both shear and moment calculations.

Consequently:

New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 694.45$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.56406

Element Length,  $L = 3000.00$

Secondary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou}, \min > 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $NoDir = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

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Stepwise Properties

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At local axis: 3

EDGE -A-

Shear Force,  $V_a = 1.1167452E-031$

EDGE -B-

Shear Force,  $V_b = -1.1167452E-031$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 0.00$

-Compression:  $As_c = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{l,ten} = 1017.876$

-Compression:  $As_{l,com} = 1017.876$

-Middle:  $As_{l,mid} = 1017.876$

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Calculation of Shear Capacity ratio,  $V_e/V_r = 0.44725617$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 202038.041$

with

$M_{pr1} = \max(\mu_{1+}, \mu_{1-}) = 3.0306E+008$

$\mu_{1+} = 3.0306E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$\mu_{1-} = 3.0306E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \max(\mu_{2+}, \mu_{2-}) = 3.0306E+008$

$\mu_{2+} = 3.0306E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

$\mu_{2-} = 3.0306E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

Calculation of Mu1+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 3.0306E+008

= 1.02974  
' = 0.91217079  
error of function (3.68), Biskinis Phd = 60360.02  
From 5A.2, TBDY: fcc = fc\* c = 51.61391  
conf. factor c = 1.56406  
fc = 33.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 694.45  
lb/d = 1.00  
d1 = 44.00  
R = 200.00  
v = 0.00101663  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.45184585

Calculation of ratio lb/d

Adequate Lap Length: lb/d >= 1

Calculation of Mu1-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 3.0306E+008

= 1.02974  
' = 0.91217079  
error of function (3.68), Biskinis Phd = 60360.02  
From 5A.2, TBDY: fcc = fc\* c = 51.61391  
conf. factor c = 1.56406  
fc = 33.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 694.45  
lb/d = 1.00  
d1 = 44.00  
R = 200.00  
v = 0.00101663  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.45184585

Calculation of ratio lb/d

Adequate Lap Length: lb/d >= 1

Calculation of Mu2+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 3.0306E+008

= 1.02974  
' = 0.91217079

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 51.61391$

conf. factor  $c = 1.56406$

$f_c = 33.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00101663$

$N = 4771.233$

$A_c = 125663.706$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of  $\mu_2$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$\mu = 3.0306E+008$

$= 1.02974$

$' = 0.91217079$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 51.61391$

conf. factor  $c = 1.56406$

$f_c = 33.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00101663$

$N = 4771.233$

$A_c = 125663.706$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 451727.786$

Calculation of Shear Strength at edge 1,  $V_{r1} = 451727.786$

$V_{r1} = V_{Co1} ((10.3), ASCE 41-17) = k_{nl} \cdot V_{Co10}$

$V_{Co10} = 451727.786$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_{s+} = f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

$f_c' = 33.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/d = 2.00$

$\mu = 1.5353150E-011$

$V_u = 1.1167452E-031$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

Ag = 125663.706

From (11.5.4.8), ACI 318-14: Vs = 219326.297

Av =  $\sqrt{2} \cdot A_{\text{stirrup}} = 123370.055$

fy = 555.56

s = 100.00

Vs is multiplied by Col = 0.00

s/d = 0.3125

Vf ((11-3)-(11.4), ACI 440) = 194961.134

f = 0.95, for fully-wrapped sections

wf/sf = 1 (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function Vf(  $\theta, \alpha$  ), is implemented for every different fiber orientation ai, as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta_1 = \theta_1 + 90^\circ = 90.00$

Vf = Min(|Vf(45,  $\theta_1$ )|, |Vf(-45,  $\theta_1$ )|), with:

total thickness per orientation, tf1 = NL\*t/NoDir = 1.016

dfv = d (figure 11.2, ACI 440) = 370.00

ffe ((11-5), ACI 440) = 259.312

Ef = 64828.00

fe = 0.004, from (11.6a), ACI 440

with fu = 0.01

From (11-11), ACI 440: Vs + Vf <= 306911.784

bw\*d =  $\sqrt{2} \cdot d^2 / 4 = 80424.772$

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Calculation of Shear Strength at edge 2, Vr2 = 451727.786

Vr2 = VCol ((10.3), ASCE 41-17) = knl\*VCol0

VCol0 = 451727.786

kn1 = 1 (zero step-static loading)

-----  
NOTE: In expression (10-3) 'Vs = Av\*fy\*d/s' is replaced by 'Vs+ f\*Vf' where Vf is the contribution of FRPs (11.3), ACI 440).

-----  
= 1 (normal-weight concrete)

fc' = 33.00, but  $fc'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

M/Vd = 2.00

Mu = 1.5353150E-011

Vu = 1.1167452E-031

d = 0.8\*D = 320.00

Nu = 4771.233

Ag = 125663.706

From (11.5.4.8), ACI 318-14: Vs = 219326.297

Av =  $\sqrt{2} \cdot A_{\text{stirrup}} = 123370.055$

fy = 555.56

s = 100.00

Vs is multiplied by Col = 0.00

s/d = 0.3125

Vf ((11-3)-(11.4), ACI 440) = 194961.134

f = 0.95, for fully-wrapped sections

wf/sf = 1 (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function Vf(  $\theta, \alpha$  ), is implemented for every different fiber orientation ai, as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta_1 = \theta_1 + 90^\circ = 90.00$

Vf = Min(|Vf(45,  $\theta_1$ )|, |Vf(-45,  $\theta_1$ )|), with:

total thickness per orientation, tf1 = NL\*t/NoDir = 1.016

dfv = d (figure 11.2, ACI 440) = 370.00

ffe ((11-5), ACI 440) = 259.312

Ef = 64828.00

fe = 0.004, from (11.6a), ACI 440

with fu = 0.01

From (11-11), ACI 440: Vs + Vf <= 306911.784

bw\*d =  $\sqrt{2} \cdot d^2 / 4 = 80424.772$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At local axis: 3

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Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At Shear local axis: 2  
(Bending local axis: 3)  
Section Type: rccs

Constant Properties

-----  
Knowledge Factor,  $\gamma = 0.90$   
Mean strength values are used for both shear and moment calculations.  
Consequently:  
New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$   
New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 555.56$   
Concrete Elasticity,  $E_c = 26999.444$   
Steel Elasticity,  $E_s = 200000.00$   
#####  
Note: Especially for the calculation of moment strengths,  
the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14  
New material: Steel Strength,  $f_s = 1.25 * f_{sm} = 694.45$   
#####  
Diameter,  $D = 400.00$   
Cover Thickness,  $c = 25.00$   
Mean Confinement Factor overall section = 1.56406  
Element Length,  $L = 3000.00$   
Secondary Member  
Ribbed Bars  
Ductile Steel  
Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Adequate Lap Length ( $l_o/l_{ou}, \min >= 1$ )  
FRP Wrapping Data  
Type: Carbon  
Cured laminate properties (design values)  
Thickness,  $t = 1.016$   
Tensile Strength,  $f_{fu} = 1055.00$   
Tensile Modulus,  $E_f = 64828.00$   
Elongation,  $e_{fu} = 0.01$   
Number of directions,  $N_{oDir} = 1$   
Fiber orientations,  $b_i = 0.00^\circ$   
Number of layers,  $N_L = 1$   
Radius of rounding corners,  $R = 40.00$

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Stepwise Properties

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At local axis: 2  
EDGE -A-  
Shear Force,  $V_a = -6.8378665E-048$   
EDGE -B-  
Shear Force,  $V_b = 6.8378665E-048$   
BOTH EDGES  
Axial Force,  $F = -4771.233$   
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $As_t = 0.00$   
-Compression:  $As_c = 3053.628$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $As_{t,ten} = 1017.876$   
-Compression:  $As_{l,com} = 1017.876$   
-Middle:  $As_{l,mid} = 1017.876$   
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Calculation of Shear Capacity ratio ,  $V_e/V_r = 0.44725617$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 202038.041$   
with

$M_{pr1} = \text{Max}(M_{u1+} , M_{u1-}) = 3.0306E+008$

$M_{u1+} = 3.0306E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$M_{u1-} = 3.0306E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(M_{u2+} , M_{u2-}) = 3.0306E+008$

$M_{u2+} = 3.0306E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

$M_{u2-} = 3.0306E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

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Calculation of  $M_{u1+}$   
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Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$

$M_u = 3.0306E+008$

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 $\phi = 1.02974$

$\lambda = 0.91217079$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c^* \quad c = 51.61391$

conf. factor  $c = 1.56406$

$f_c = 33.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00101663$

$N = 4771.233$

$A_c = 125663.706$

$= \phi \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$   
-----

Calculation of ratio  $l_b/d$   
-----

Adequate Lap Length:  $l_b/d \geq 1$   
-----  
-----

Calculation of  $M_{u1-}$   
-----

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$

$M_u = 3.0306E+008$

-----  
 $\phi = 1.02974$

$\lambda = 0.91217079$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c^* \quad c = 51.61391$

conf. factor  $c = 1.56406$

$f_c = 33.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00101663$

$N = 4771.233$

$A_c = 125663.706$

$= \phi \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$

-----  
Calculation of ratio lb/d  
-----

Adequate Lap Length:  $lb/d \geq 1$   
-----  
-----  
-----

Calculation of  $\mu_{2+}$   
-----  
-----

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 3.0306E+008$   
-----

$\lambda = 1.02974$   
 $\lambda' = 0.91217079$   
error of function (3.68), Biskinis Phd = 60360.02  
From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 51.61391$   
conf. factor  $c = 1.56406$   
 $f_c = 33.00$   
From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (lb/d)^{2/3}) = 694.45$   
 $lb/d = 1.00$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00101663$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (lb/d)^{2/3}) = 0.45184585$   
-----

Calculation of ratio lb/d  
-----

Adequate Lap Length:  $lb/d \geq 1$   
-----  
-----  
-----

Calculation of  $\mu_{2-}$   
-----  
-----

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 3.0306E+008$   
-----

$\lambda = 1.02974$   
 $\lambda' = 0.91217079$   
error of function (3.68), Biskinis Phd = 60360.02  
From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 51.61391$   
conf. factor  $c = 1.56406$   
 $f_c = 33.00$   
From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (lb/d)^{2/3}) = 694.45$   
 $lb/d = 1.00$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00101663$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (lb/d)^{2/3}) = 0.45184585$   
-----

Calculation of ratio lb/d  
-----

Adequate Lap Length:  $lb/d \geq 1$   
-----  
-----  
-----

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 451727.786$

Calculation of Shear Strength at edge 1,  $V_{r1} = 451727.786$

$V_{r1} = V_{CoI} ((10.3), \text{ASCE } 41-17) = k_{nl} * V_{CoI0}$

$V_{CoI0} = 451727.786$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\lambda = 1$  (normal-weight concrete)

$f_c' = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M / Vd = 2.00$

$\mu_u = 5.4715235E-012$

$\nu_u = 6.8378665E-048$

$d = 0.8 * D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 219326.297$

$A_v = \sqrt{2} * A_{stirrup} = 123370.055$

$f_y = 555.56$

$s = 100.00$

$V_s$  is multiplied by  $\text{Col} = 0.00$

$s/d = 0.3125$

$V_f ((11-3)-(11.4), \text{ACI } 440) = 194961.134$

$f = 0.95$ , for fully-wrapped sections

$w_f / s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression,

where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = 45^\circ$  and  $\alpha = 90^\circ$

$V_f = \text{Min}(|V_f(45^\circ, 90^\circ)|, |V_f(-45^\circ, 90^\circ)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L * t / N_{oDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe} ((11-5), \text{ACI } 440) = 259.312$

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 306911.784$

$b_w * d = \sqrt{2} * d^2 / 4 = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 451727.786$

$V_{r2} = V_{CoI} ((10.3), \text{ASCE } 41-17) = k_{nl} * V_{CoI0}$

$V_{CoI0} = 451727.786$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\lambda = 1$  (normal-weight concrete)

$f_c' = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M / Vd = 2.00$

$\mu_u = 5.4715235E-012$

$\nu_u = 6.8378665E-048$

$d = 0.8 * D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 219326.297$

$A_v = \sqrt{2} * A_{stirrup} = 123370.055$

$f_y = 555.56$

$s = 100.00$

$V_s$  is multiplied by  $\text{Col} = 0.00$

$s/d = 0.3125$

$V_f ((11-3)-(11.4), \text{ACI } 440) = 194961.134$

$f = 0.95$ , for fully-wrapped sections

$wf/sf = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a)\sin a$  which is more a generalised expression, where  $a$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = b1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, a1)|)$ , with:

total thickness per orientation,  $tf1 = NL * t / \text{NoDir} = 1.016$

$dfv = d$  (figure 11.2, ACI 440) = 370.00

$ffe$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 306911.784$

$bw * d = \rho * d * d / 4 = 80424.772$

-----  
End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At local axis: 2

-----  
Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (a)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 0.90$

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Secondary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $b/d \geq 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $ff_u = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $ef_u = 0.01$

Number of directions,  $\text{NoDir} = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

-----  
Stepwise Properties

Bending Moment,  $M = 6.3952987E-010$

Shear Force,  $V2 = -3759.209$

Shear Force,  $V3 = -1.9910600E-013$

Axial Force,  $F = -4769.729$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_{lt} = 1272.345$

-Compression:  $As_{lc} = 1781.283$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{l,ten} = 1017.876$

-Compression:  $As_{l,com} = 1017.876$

-Middle:  $As_{l,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $Db_L = 18.00$

New component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_{,R} = 1.0^*$   $u = 0.01989311$   
 $u = y + p = 0.01989311$

- Calculation of  $y$  -

$y = (M_y * L_s / 3) / E_{eff} = 0.01489311$  ((4.29), Biskinis Phd)

$M_y = 3.0318E+008$

$L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 1500.00

From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 1.0179E+013$

factor = 0.30

$A_g = 125663.706$

$f_c' = 33.00$

$N = 4769.729$

$E_c * I_g = 3.3929E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 3.0318E+008$

$y$  ((10a) or (10b)) = 1.6222680E-005

$M_{y\_ten}$  (8a) = 3.0318E+008

$y_{ten}$  (7a) = 69.35397

error of function (7a) = 0.00743258

$M_{y\_com}$  (8b) = 4.6746E+008

$y_{com}$  (7b) = 67.1564

error of function (7b) = -0.00265541

with  $e_y = 0.0027778$

$e_{co} = 0.002$

$a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)

$d_1 = 44.00$

$R = 200.00$

$v = 0.00102226$

$N = 4769.729$

$A_c = 125663.706$

= 0.36359274

with  $f_c^*$  ((12.3), ACI 440) = 37.12975

$f_c = 33.00$

$f_l = 1.3173$

$k = 1$

Effective FRP thickness,  $t_f = NL * t * \text{Cos}(b_1) = 1.016$

$e_{fe}$  ((12.5) and (12.7)) = 0.004

$E_f = 64828.00$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

- Calculation of  $p$  -

From table 10-9:  $p = 0.005$

with:

- Columns controlled by inadequate development or splicing along the clear height because  $l_b/l_d < 1$
- shear control ratio  $V_{yE}/V_{CoI0E} = 0.44725617$
- $d = 0.00$
- $s = 0.00$
- $t = 2 \cdot A_v / (d_c \cdot s) + 4 \cdot t_f / D \cdot (f_{fe} / f_s) = 0.00$
- $A_v = 78.53982$ , is the area of the circular stirrup
- $d_c = D - 2 \cdot \text{cover} - \text{Hoop Diameter} = 340.00$
- The term  $2 \cdot t_f / b_w \cdot (f_{fe} / f_s)$  is implemented to account for FRP contribution
- where  $f = 2 \cdot t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength
- All these variables have already been given in Shear control ratio calculation.
- $NUD = 4769.729$
- $A_g = 125663.706$
- $f_{cE} = 33.00$
- $f_{ytE} = f_{ylE} = 555.56$
- $p_l = \text{Area\_Tot\_Long\_Rein} / (A_g) = 0.0243$
- $f_{cE} = 33.00$

End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (a)

### Calculation No. 3

column C1, Floor 1

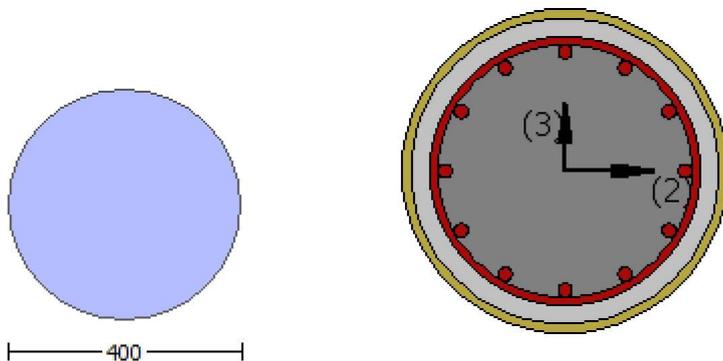
Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Shear capacity  $V_{Rd}$

Edge: Start

Local Axis: (3)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (a)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\phi = 0.90$   
 Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE41-17.  
 Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17  
 Consequently:  
 New material of Secondary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 25.00$   
 New material of Secondary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 500.00$   
 Concrete Elasticity,  $E_c = 26999.444$   
 Steel Elasticity,  $E_s = 200000.00$   
 #####  
 Note: Especially for the calculation of  $\mu_y$  for displacement ductility demand,  
 the expected (mean value) strengths are used (7.5.1.3, ASCE41-17) because bending is considered as  
 Deformation-Controlled Action (Table C7-1, ASCE41-17).  
 New material: Concrete Strength,  $f_c = f_{cm} = 33.00$   
 New material: Steel Strength,  $f_s = f_{sm} = 555.56$   
 #####  
 Diameter,  $D = 400.00$   
 Cover Thickness,  $c = 25.00$   
 Element Length,  $L = 3000.00$   
 Secondary Member  
 Ribbed Bars  
 Ductile Steel  
 Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Adequate Lap Length ( $l_o/l_{ou,min} = l_b/l_d \geq 1$ )  
 FRP Wrapping Data  
 Type: Carbon  
 Cured laminate properties (design values)  
 Thickness,  $t = 1.016$   
 Tensile Strength,  $f_{fu} = 1055.00$   
 Tensile Modulus,  $E_f = 64828.00$   
 Elongation,  $e_{fu} = 0.01$   
 Number of directions,  $N_{oDir} = 1$   
 Fiber orientations,  $b_i = 0.00^\circ$   
 Number of layers,  $N_L = 1$   
 Radius of rounding corners,  $R = 40.00$

Stepwise Properties

EDGE -A-  
 Bending Moment,  $M_a = 6.3952987E-010$   
 Shear Force,  $V_a = -1.9910600E-013$   
 EDGE -B-  
 Bending Moment,  $M_b = -4.1943110E-011$   
 Shear Force,  $V_b = 1.9910600E-013$   
 BOTH EDGES  
 Axial Force,  $F = -4769.729$   
 Longitudinal Reinforcement Area Distribution (in 2 divisions)  
   -Tension:  $A_{sl,t} = 1272.345$   
   -Compression:  $A_{sl,c} = 1781.283$   
 Longitudinal Reinforcement Area Distribution (in 3 divisions)  
   -Tension:  $A_{sl,ten} = 1017.876$   
   -Compression:  $A_{sl,com} = 1017.876$   
   -Middle:  $A_{sl,mid} = 1017.876$   
 Mean Diameter of Tension Reinforcement,  $Db_{L,ten} = 18.00$

New component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = 1.0 \cdot V_n = 393301.081$   
 $V_n ((10.3), ASCE 41-17) = k_n l \cdot V_{CoI} = 393301.081$   
 $V_{CoI} = 393301.081$   
 $k_n l = 1.00$   
 displacement\_ductility\_demand = 0.00

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ '

where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

$f_c' = 25.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 6.3952987E-010$

$\nu_u = 1.9910600E-013$

$d = 0.8 \cdot D = 320.00$

$N_u = 4769.729$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 197392.088$

$A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 123370.055$

$f_y = 500.00$

$s = 100.00$

$V_s$  is multiplied by  $\text{Col} = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot_a) \sin \alpha$  which is more a generalised expression, where  $\alpha$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\alpha = 45^\circ$  and  $\alpha = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\alpha = 45^\circ$

$V_f = \text{Min}(|V_f(45, \alpha)|, |V_f(-45, \alpha)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{\text{Dir}} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 267132.42$

$b_w \cdot d = N_u \cdot d / 4 = 80424.772$

displacement ductility demand is calculated as  $\delta / y$

- Calculation of  $\delta / y$  for END A -

for rotation axis 2 and integ. section (a)

From analysis, chord rotation  $\theta = 1.8835917E-020$

$y = (M_y \cdot L_s / 3) / E_{\text{eff}} = 0.01489311$  ((4.29), Biskinis Phd)

$M_y = 3.0318E+008$

$L_s = M/V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) = 1500.00

From table 10.5, ASCE 41\_17:  $E_{\text{eff}} = \text{factor} \cdot E_c \cdot I_g = 1.0179E+013$

factor = 0.30

$A_g = 125663.706$

$f_c' = 33.00$

$N = 4769.729$

$E_c \cdot I_g = 3.3929E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\delta / y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 3.0318E+008$

$y$  ((10a) or (10b)) = 1.6222680E-005

$M_{y\_ten}$  (8a) = 3.0318E+008

$\delta / y$  (7a) = 69.35397

error of function (7a) = 0.00743258

$M_{y\_com}$  (8b) = 4.6746E+008

$\delta / y$  (7b) = 67.1564

error of function (7b) = -0.00265541

with  $e_y = 0.0027778$

$\epsilon_{co} = 0.002$   
 $\alpha_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00102226$   
 $N = 4769.729$   
 $A_c = 125663.706$   
 $= 0.36359274$   
 with  $f_c^*$  ((12.3), ACI 440) = 37.12975  
 $f_c = 33.00$   
 $f_l = 1.3173$   
 $k = 1$   
 Effective FRP thickness,  $t_f = NL * t * \cos(b_1) = 1.016$   
 $\epsilon_{fe}$  ((12.5) and (12.7)) = 0.004  
 $E_f = 64828.00$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

End Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (a)

## Calculation No. 4

column C1, Floor 1

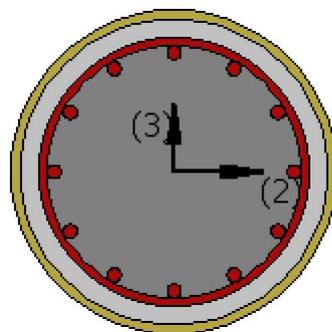
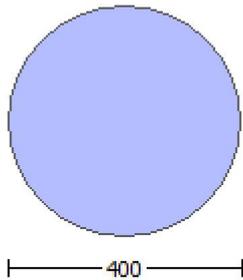
Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity ( $\theta_u$ )

Edge: Start

Local Axis: (3)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 0.90$

Mean strength values are used for both shear and moment calculations.

Consequently:

New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 694.45$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.56406

Element Length,  $L = 3000.00$

Secondary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou}, \min > = 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

Stepwise Properties

At local axis: 3

EDGE -A-

Shear Force,  $V_a = 1.1167452E-031$

EDGE -B-

Shear Force,  $V_b = -1.1167452E-031$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{sl,t} = 0.00$

-Compression:  $A_{sl,c} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{sl,ten} = 1017.876$

-Compression:  $A_{sl,com} = 1017.876$

-Middle:  $A_{sl,mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.44725617$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 202038.041$

with

$M_{pr1} = \text{Max}(M_{u1+}, M_{u1-}) = 3.0306E+008$

$M_{u1+} = 3.0306E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$M_{u1-} = 3.0306E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment

direction which is defined for the static loading combination

$$M_{pr2} = \text{Max}(M_{u2+}, M_{u2-}) = 3.0306E+008$$

$M_{u2+} = 3.0306E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination

$M_{u2-} = 3.0306E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

-----  
Calculation of  $M_{u1+}$   
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$

$$M_u = 3.0306E+008$$

$$\phi = 1.02974$$

$$\lambda = 0.91217079$$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 51.61391$

$$\text{conf. factor } c = 1.56406$$

$$f_c = 33.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$$l_b/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00101663$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \phi \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$$

-----  
Calculation of ratio  $l_b/d$   
-----

Adequate Lap Length:  $l_b/d \geq 1$   
-----  
-----

-----  
Calculation of  $M_{u1-}$   
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$

$$M_u = 3.0306E+008$$

$$\phi = 1.02974$$

$$\lambda = 0.91217079$$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 51.61391$

$$\text{conf. factor } c = 1.56406$$

$$f_c = 33.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$$l_b/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00101663$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \phi \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$$

-----  
Calculation of ratio  $l_b/d$   
-----

Adequate Lap Length:  $l_b/d \geq 1$   
-----  
-----

-----  
Calculation of  $M_{u2+}$   
-----

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 3.0306E+008

= 1.02974  
' = 0.91217079  
error of function (3.68), Biskinis Phd = 60360.02  
From 5A.2, TBDY: fcc = fc\* c = 51.61391  
conf. factor c = 1.56406  
fc = 33.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 694.45  
lb/d = 1.00  
d1 = 44.00  
R = 200.00  
v = 0.00101663  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.45184585

Calculation of ratio lb/d

Adequate Lap Length: lb/d >= 1

Calculation of Mu2-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 3.0306E+008

= 1.02974  
' = 0.91217079  
error of function (3.68), Biskinis Phd = 60360.02  
From 5A.2, TBDY: fcc = fc\* c = 51.61391  
conf. factor c = 1.56406  
fc = 33.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 694.45  
lb/d = 1.00  
d1 = 44.00  
R = 200.00  
v = 0.00101663  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.45184585

Calculation of ratio lb/d

Adequate Lap Length: lb/d >= 1

Calculation of Shear Strength Vr = Min(Vr1,Vr2) = 451727.786

Calculation of Shear Strength at edge 1, Vr1 = 451727.786  
Vr1 = VCol ((10.3), ASCE 41-17) = knl\*VCo10  
VCo10 = 451727.786  
knl = 1 (zero step-static loading)

NOTE: In expression (10-3) 'Vs = Av\*fy\*d/s' is replaced by 'Vs+ f\*Vf  
where Vf is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)  
 $f_c' = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $\mu_u = 1.5353150E-011$   
 $\nu_u = 1.1167452E-031$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 219326.297$   
 $A_v = \frac{1}{2} \cdot A_{\text{stirrup}} = 123370.055$   
 $f_y = 555.56$   
 $s = 100.00$   
 $V_s$  is multiplied by  $\text{Col} = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = 45^\circ$   
 $V_f = \text{Min}(|V_f(45, a)|, |V_f(-45, a)|)$ , with:  
 total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{\text{Dir}} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}$  ((11-5), ACI 440) = 259.312  
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 306911.784$   
 $b_w \cdot d = \frac{1}{4} \cdot d^2 = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 451727.786$   
 $V_{r2} = V_{\text{Col}}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{\text{Col}0}$   
 $V_{\text{Col}0} = 451727.786$   
 $k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ '  
 where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)  
 $f_c' = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $\mu_u = 1.5353150E-011$   
 $\nu_u = 1.1167452E-031$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 219326.297$   
 $A_v = \frac{1}{2} \cdot A_{\text{stirrup}} = 123370.055$   
 $f_y = 555.56$   
 $s = 100.00$   
 $V_s$  is multiplied by  $\text{Col} = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = 45^\circ$   
 $V_f = \text{Min}(|V_f(45, a)|, |V_f(-45, a)|)$ , with:  
 total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{\text{Dir}} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

ffe ((11-5), ACI 440) = 259.312  
Ef = 64828.00  
fe = 0.004, from (11.6a), ACI 440  
with fu = 0.01  
From (11-11), ACI 440: Vs + Vf <= 306911.784  
bw\*d = \*d\*d/4 = 80424.772

-----  
-----  
End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At local axis: 3

-----  
-----  
Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At Shear local axis: 2  
(Bending local axis: 3)  
Section Type: rccs

Constant Properties

-----  
Knowledge Factor, = 0.90  
Mean strength values are used for both shear and moment calculations.  
Consequently:  
New material of Secondary Member: Concrete Strength, fc = fcm = 33.00  
New material of Secondary Member: Steel Strength, fs = fsm = 555.56  
Concrete Elasticity, Ec = 26999.444  
Steel Elasticity, Es = 200000.00  
#####  
Note: Especially for the calculation of moment strengths,  
the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14  
New material: Steel Strength, fs = 1.25\*fsm = 694.45  
#####  
Diameter, D = 400.00  
Cover Thickness, c = 25.00  
Mean Confinement Factor overall section = 1.56406  
Element Length, L = 3000.00  
Secondary Member  
Ribbed Bars  
Ductile Steel  
Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Adequate Lap Length (lo/lou,min>=1)  
FRP Wrapping Data  
Type: Carbon  
Cured laminate properties (design values)  
Thickness, t = 1.016  
Tensile Strength, ffu = 1055.00  
Tensile Modulus, Ef = 64828.00  
Elongation, efu = 0.01  
Number of directions, NoDir = 1  
Fiber orientations, bi: 0.00°  
Number of layers, NL = 1  
Radius of rounding corners, R = 40.00

-----  
Stepwise Properties

-----  
At local axis: 2  
EDGE -A-  
Shear Force, Va = -6.8378665E-048  
EDGE -B-  
Shear Force, Vb = 6.8378665E-048  
BOTH EDGES  
Axial Force, F = -4771.233  
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension: Aslt = 0.00

-Compression:  $A_{sc} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{s,ten} = 1017.876$

-Compression:  $A_{s,com} = 1017.876$

-Middle:  $A_{s,mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.44725617$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 202038.041$

with

$M_{pr1} = \text{Max}(M_{u1+}, M_{u1-}) = 3.0306E+008$

$M_{u1+} = 3.0306E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$M_{u1-} = 3.0306E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(M_{u2+}, M_{u2-}) = 3.0306E+008$

$M_{u2+} = 3.0306E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

$M_{u2-} = 3.0306E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

Calculation of  $M_{u1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$

$M_u = 3.0306E+008$

$\phi = 1.02974$

$\lambda = 0.91217079$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c^* \quad c = 51.61391$

conf. factor  $c = 1.56406$

$f_c = 33.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00101663$

$N = 4771.233$

$A_c = 125663.706$

$\phi = \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of  $M_{u1-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$

$M_u = 3.0306E+008$

$\phi = 1.02974$

$\lambda = 0.91217079$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c^* \quad c = 51.61391$

conf. factor  $c = 1.56406$

$f_c = 33.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$$l_b/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00101663$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of  $\mu_{2+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$$\mu = 3.0306E+008$$

$$= 1.02974$$

$$' = 0.91217079$$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 51.61391$

$$\text{conf. factor } c = 1.56406$$

$$f_c = 33.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$$l_b/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00101663$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of  $\mu_{2-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$$\mu = 3.0306E+008$$

$$= 1.02974$$

$$' = 0.91217079$$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 51.61391$

$$\text{conf. factor } c = 1.56406$$

$$f_c = 33.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$$l_b/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00101663$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 451727.786$

Calculation of Shear Strength at edge 1,  $V_{r1} = 451727.786$

$V_{r1} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_n l V_{Col0}$

$V_{Col0} = 451727.786$

$k_n l = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v f_y d/s$ ' is replaced by ' $V_s + f^* V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

$f_c' = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 5.4715235E-012$

$V_u = 6.8378665E-048$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 219326.297$

$A_v = \sqrt{2} \cdot A_{stirrup} = 123370.055$

$f_y = 555.56$

$s = 100.00$

$V_s$  is multiplied by  $\text{Col} = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) =  $194961.134$

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different cyclic fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = b1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \theta)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{Dir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) =  $370.00$

$f_{fe}$  ((11-5), ACI 440) =  $259.312$

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 306911.784$

$b_w \cdot d = \sqrt{V_u \cdot d} / 4 = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 451727.786$

$V_{r2} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_n l V_{Col0}$

$V_{Col0} = 451727.786$

$k_n l = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v f_y d/s$ ' is replaced by ' $V_s + f^* V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

$f_c' = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 5.4715235E-012$

$V_u = 6.8378665E-048$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 219326.297$

$A_v = \sqrt{2} \cdot A_{stirrup} = 123370.055$

$f_y = 555.56$

$s = 100.00$

$V_s$  is multiplied by  $Col = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \cot \alpha) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta_1 = \theta_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, \theta_1)|)$ , with:

total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 306911.784$

$b_w \cdot d = \sqrt{4} \cdot d^2 = 80424.772$

-----  
End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At local axis: 2

-----  
Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (a)

Section Type: rccs

Constant Properties

-----  
Knowledge Factor,  $\phi = 0.90$

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Secondary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_b/d \geq 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $\text{NoDir} = 1$

Fiber orientations,  $\theta_i = 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

## Stepwise Properties

Bending Moment,  $M = -1.1282E+007$

Shear Force,  $V2 = -3759.209$

Shear Force,  $V3 = -1.9910600E-013$

Axial Force,  $F = -4769.729$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{st} = 1272.345$

-Compression:  $A_{sc} = 1781.283$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{st,ten} = 1017.876$

-Compression:  $A_{sc,com} = 1017.876$

-Middle:  $A_{st,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $DbL = 18.00$

New component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u,R = 1.0^*$   $u = 0.03479765$

$u = y + p = 0.03479765$

- Calculation of  $y$  -

$y = (My * L_s / 3) / E_{eff} = 0.02979765$  ((4.29), Biskinis Phd))

$My = 3.0318E+008$

$L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 3001.151

From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 1.0179E+013$

factor = 0.30

$A_g = 125663.706$

$f_c' = 33.00$

$N = 4769.729$

$E_c * I_g = 3.3929E+013$

Calculation of Yielding Moment  $My$

Calculation of  $y$  and  $My$  according to (7) - (8) in Biskinis and Fardis

$My = \text{Min}(My_{ten}, My_{com}) = 3.0318E+008$

$y$  ((10a) or (10b)) = 1.6222680E-005

$My_{ten}$  (8a) = 3.0318E+008

$_{ten}$  (7a) = 69.35397

error of function (7a) = 0.00743258

$My_{com}$  (8b) = 4.6746E+008

$_{com}$  (7b) = 67.1564

error of function (7b) = -0.00265541

with  $e_y = 0.0027778$

$e_{co} = 0.002$

$apl = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)

$d1 = 44.00$

$R = 200.00$

$v = 0.00102226$

$N = 4769.729$

$A_c = 125663.706$

= 0.36359274

with  $f_c^*$  ((12.3), ACI 440) = 37.12975

$f_c = 33.00$

$f_l = 1.3173$

$k = 1$

Effective FRP thickness,  $t_f = NL * t * \text{Cos}(b1) = 1.016$

$e_{fe}$  ((12.5) and (12.7)) = 0.004

$E_f = 64828.00$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

- Calculation of  $p$  -

From table 10-9:  $p = 0.005$

with:

- Columns controlled by inadequate development or splicing along the clear height because  $l_b/d < 1$

shear control ratio  $V_y E / V_{CoI} E = 0.44725617$

$d = 0.00$

$s = 0.00$

$t = 2 \cdot A_v / (d_c \cdot s) + 4 \cdot t_f / D \cdot (f_{fe} / f_s) = 0.00$

$A_v = 78.53982$ , is the area of the circular stirrup

$d_c = D - 2 \cdot \text{cover} - \text{Hoop Diameter} = 340.00$

The term  $2 \cdot t_f / b_w \cdot (f_{fe} / f_s)$  is implemented to account for FRP contribution

where  $f = 2 \cdot t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength

All these variables have already been given in Shear control ratio calculation.

$NUD = 4769.729$

$A_g = 125663.706$

$f_{cE} = 33.00$

$f_{ytE} = f_{ylE} = 555.56$

$p_l = \text{Area\_Tot\_Long\_Rein} / (A_g) = 0.0243$

$f_{cE} = 33.00$

End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (a)

## Calculation No. 5

column C1, Floor 1

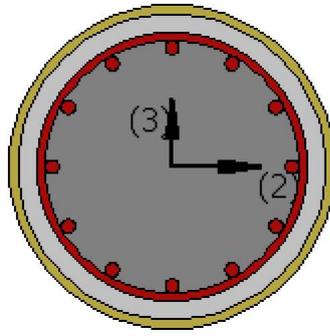
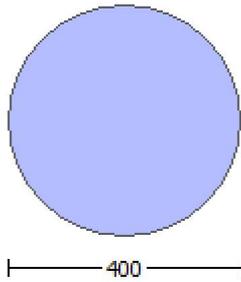
Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Shear capacity VRd

Edge: End

Local Axis: (2)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (b)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 0.90$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

New material of Secondary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 25.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 500.00$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of  $\gamma$  for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE41-17).

New material: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material: Steel Strength,  $f_s = f_{sm} = 555.56$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Secondary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou}, \min = l_b/l_d \geq 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $NoDir = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

Stepwise Properties

EDGE -A-

Bending Moment,  $M_a = -1.1282E+007$

Shear Force,  $V_a = -3759.209$

EDGE -B-

Bending Moment,  $M_b = 0.06021356$

Shear Force,  $V_b = 3759.209$

BOTH EDGES

Axial Force,  $F = -4769.729$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_{t1} = 0.00$

-Compression:  $As_{c1} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t,ten} = 1017.876$

-Compression:  $As_{l,com} = 1017.876$

-Middle:  $As_{l,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $Db_{L,ten} = 18.00$

New component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = 1.0 \cdot V_n = 393301.081$

$V_n$  ((10.3), ASCE 41-17) =  $k_n \cdot V_{Col0} = 393301.081$

$V_{Col} = 393301.081$

$k_n = 1.00$

$displacement\_ductility\_demand = 0.05581234$

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

$f'_c = 25.00$ , but  $f'_c \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 0.06021356$

$V_u = 3759.209$

$d = 0.8 \cdot D = 320.00$

$N_u = 4769.729$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 197392.088$

$A_v = \frac{1}{2} \cdot A_{stirrup} = 123370.055$

$f_y = 500.00$

$s = 100.00$

$V_s$  is multiplied by  $Col = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) =  $194961.134$

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = 45^\circ$  and  $\alpha = 90^\circ$  =  $90.00$

$V_f = \text{Min}(|V_f(45^\circ, 90^\circ)|, |V_f(-45^\circ, 90^\circ)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{oDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) =  $370.00$

$f_{fe}$  ((11-5), ACI 440) =  $259.312$

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 267132.42$

$b_w \cdot d = \frac{1}{4} \cdot A_{stirrup} = 80424.772$

$displacement\_ductility\_demand$  is calculated as  $\frac{1}{y}$

- Calculation of  $\frac{1}{y}$  for END B -

for rotation axis 3 and integ. section (b)

From analysis, chord rotation =  $0.00016624$

$y = (M_y \cdot L_s / 3) / E_{eff} = 0.00297862$  ((4.29), Biskinis Phd))

$M_y = 3.0318E+008$

$L_s = M/V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) =  $300.00$

From table 10.5, ASCE 41\_17:  $E_{eff} = factor \cdot E_c \cdot I_g = 1.0179E+013$

factor = 0.30  
Ag = 125663.706  
fc' = 33.00  
N = 4769.729  
Ec\*Ig = 3.3929E+013

-----  
-----  
Calculation of Yielding Moment My

-----  
Calculation of  $\rho_y$  and My according to (7) - (8) in Biskinis and Fardis

-----  
My = Min(My\_ten, My\_com) = 3.0318E+008  
 $\rho_y$  ((10a) or (10b)) = 1.6222680E-005  
My\_ten (8a) = 3.0318E+008  
 $\rho_{y\_ten}$  (7a) = 69.35397  
error of function (7a) = 0.00743258  
My\_com (8b) = 4.6746E+008  
 $\rho_{y\_com}$  (7b) = 67.1564  
error of function (7b) = -0.00265541  
with  $e_y$  = 0.0027778  
 $e_{co}$  = 0.002  
 $a_{pl}$  = 0.45 ((9c) in Biskinis and Fardis for FRP Wrap)  
d1 = 44.00  
R = 200.00  
 $v$  = 0.00102226  
N = 4769.729  
Ac = 125663.706  
= 0.36359274  
with  $f_c^*$  ((12.3), ACI 440) = 37.12975  
fc = 33.00  
fl = 1.3173  
k = 1  
Effective FRP thickness,  $t_f$  = NL\*t\*cos(b1) = 1.016  
efe ((12.5) and (12.7)) = 0.004  
Ef = 64828.00

-----  
Calculation of ratio lb/d

-----  
Adequate Lap Length: lb/d >= 1

-----  
End Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (b)

-----  
**Calculation No. 6**

column C1, Floor 1

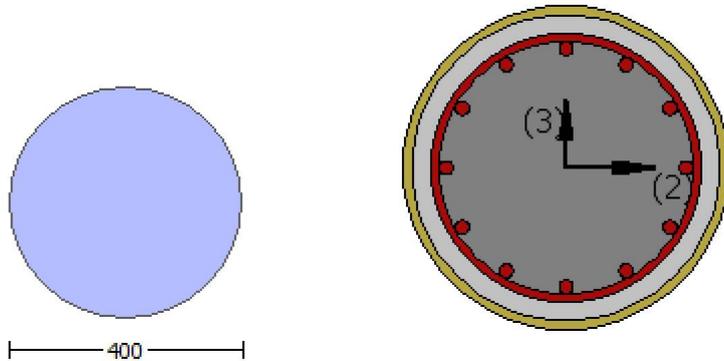
Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity ( $\theta_u$ )

Edge: End

Local Axis: (2)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 0.90$

Mean strength values are used for both shear and moment calculations.

Consequently:

New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 694.45$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.56406

Element Length,  $L = 3000.00$

Secondary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou, \min} > 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i: 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

Stepwise Properties

At local axis: 3

EDGE -A-

Shear Force,  $V_a = 1.1167452E-031$

EDGE -B-

Shear Force,  $V_b = -1.1167452E-031$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 0.00$

-Compression:  $As_c = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t,ten} = 1017.876$

-Compression:  $As_{c,com} = 1017.876$

-Middle:  $As_{mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.44725617$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 202038.041$

with

$M_{pr1} = \text{Max}(Mu_{1+}, Mu_{1-}) = 3.0306E+008$

$Mu_{1+} = 3.0306E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$Mu_{1-} = 3.0306E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(Mu_{2+}, Mu_{2-}) = 3.0306E+008$

$Mu_{2+} = 3.0306E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

$Mu_{2-} = 3.0306E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

Calculation of  $Mu_{1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$

$M_u = 3.0306E+008$

$\phi = 1.02974$

$\lambda = 0.91217079$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c \cdot \lambda = 51.61391$

conf. factor  $\lambda = 1.56406$

$f_c = 33.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00101663$

$N = 4771.233$

$A_c = 125663.706$

$\phi \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of Mu1-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 3.0306E+008

= 1.02974  
' = 0.91217079  
error of function (3.68), Biskinis Phd = 60360.02  
From 5A.2, TBDY: fcc = fc\* c = 51.61391  
conf. factor c = 1.56406  
fc = 33.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 694.45  
lb/d = 1.00  
d1 = 44.00  
R = 200.00  
v = 0.00101663  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.45184585

Calculation of ratio lb/d

Adequate Lap Length: lb/d >= 1

Calculation of Mu2+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 3.0306E+008

= 1.02974  
' = 0.91217079  
error of function (3.68), Biskinis Phd = 60360.02  
From 5A.2, TBDY: fcc = fc\* c = 51.61391  
conf. factor c = 1.56406  
fc = 33.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 694.45  
lb/d = 1.00  
d1 = 44.00  
R = 200.00  
v = 0.00101663  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.45184585

Calculation of ratio lb/d

Adequate Lap Length: lb/d >= 1

Calculation of Mu2-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 3.0306E+008

= 1.02974

$\rho = 0.91217079$   
 error of function (3.68), Biskinis Phd = 60360.02  
 From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 51.61391$   
 conf. factor  $c = 1.56406$   
 $f_c = 33.00$   
 From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$   
 $l_b/d = 1.00$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00101663$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 451727.786$

Calculation of Shear Strength at edge 1,  $V_{r1} = 451727.786$

$V_{r1} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{Col0}$

$V_{Col0} = 451727.786$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

$f_c' = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/d = 2.00$

$\mu_u = 1.5353150E-011$

$\nu_u = 1.1167452E-031$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 219326.297$

$A_v = \rho_s \cdot A_{stirup} = 123370.055$

$f_y = 555.56$

$s = 100.00$

$V_s$  is multiplied by  $\rho_{col} = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression,

where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = 45^\circ + 90^\circ = 135^\circ$

$V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, 1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{oDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_{e1} = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 306911.784$

$b_w \cdot d = \rho_s \cdot d^2 / 4 = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 451727.786$

$V_{r2} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{Col0}$

VCoIO = 451727.786  
knl = 1 (zero step-static loading)

NOTE: In expression (10-3) 'Vs = Av\*fy\*d/s' is replaced by 'Vs+ f\*VF'  
where Vf is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)  
fc' = 33.00, but  $fc^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
M/Vd = 2.00  
Mu = 1.5353150E-011  
Vu = 1.1167452E-031  
d = 0.8\*D = 320.00  
Nu = 4771.233  
Ag = 125663.706  
From (11.5.4.8), ACI 318-14: Vs = 219326.297  
Av = /2\*A\_stirrup = 123370.055  
fy = 555.56  
s = 100.00  
Vs is multiplied by Col = 0.00  
s/d = 0.3125  
Vf ((11-3)-(11.4), ACI 440) = 194961.134  
f = 0.95, for fully-wrapped sections  
wf/sf = 1 (FRP strips adjacent to one another).  
In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \cot \alpha) \sin \alpha$  which is more a generalised expression,  
where  $\theta$  is the angle of the crack direction (see KANEPE).  
This later relation, considered as a function Vf(  $\theta, \alpha$  ), is implemented for every different fiber orientation ai,  
as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
orientation 1:  $\theta = b1 + 90^\circ = 90.00$   
Vf = Min(|Vf(45,  $\theta$ )|, |Vf(-45,  $\theta$ )|), with:  
total thickness per orientation, tf1 = NL\*t/NoDir = 1.016  
dfv = d (figure 11.2, ACI 440) = 370.00  
ffe ((11-5), ACI 440) = 259.312  
Ef = 64828.00  
fe = 0.004, from (11.6a), ACI 440  
with fu = 0.01  
From (11-11), ACI 440: Vs + Vf <= 306911.784  
bw\*d = \*d\*d/4 = 80424.772

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At local axis: 3

Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At Shear local axis: 2  
(Bending local axis: 3)  
Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 0.90$   
Mean strength values are used for both shear and moment calculations.  
Consequently:  
New material of Secondary Member: Concrete Strength, fc = fcm = 33.00  
New material of Secondary Member: Steel Strength, fs = fsm = 555.56  
Concrete Elasticity, Ec = 26999.444  
Steel Elasticity, Es = 200000.00  
#####  
Note: Especially for the calculation of moment strengths,  
the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14  
New material: Steel Strength, fs = 1.25\*fsm = 694.45  
#####  
Diameter, D = 400.00  
Cover Thickness, c = 25.00  
Mean Confinement Factor overall section = 1.56406

Element Length, L = 3000.00  
Secondary Member  
Ribbed Bars  
Ductile Steel  
Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Adequate Lap Length ( $l_o/l_{ou}, \min > = 1$ )  
FRP Wrapping Data  
Type: Carbon  
Cured laminate properties (design values)  
Thickness, t = 1.016  
Tensile Strength,  $f_{fu} = 1055.00$   
Tensile Modulus,  $E_f = 64828.00$   
Elongation,  $e_{fu} = 0.01$   
Number of directions, NoDir = 1  
Fiber orientations,  $b_i = 0.00^\circ$   
Number of layers, NL = 1  
Radius of rounding corners, R = 40.00

#### Stepwise Properties

At local axis: 2  
EDGE -A-  
Shear Force,  $V_a = -6.8378665E-048$   
EDGE -B-  
Shear Force,  $V_b = 6.8378665E-048$   
BOTH EDGES  
Axial Force, F = -4771.233  
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $A_{st} = 0.00$   
-Compression:  $A_{sc} = 3053.628$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $A_{st,ten} = 1017.876$   
-Compression:  $A_{sl,com} = 1017.876$   
-Middle:  $A_{sl,mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.44725617$   
Member Controlled by Flexure ( $V_e/V_r < 1$ )  
Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 202038.041$   
with  
 $M_{pr1} = \text{Max}(\mu_{1+}, \mu_{1-}) = 3.0306E+008$   
 $\mu_{1+} = 3.0306E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination  
 $\mu_{1-} = 3.0306E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination  
 $M_{pr2} = \text{Max}(\mu_{2+}, \mu_{2-}) = 3.0306E+008$   
 $\mu_{2+} = 3.0306E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination  
 $\mu_{2-} = 3.0306E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

#### Calculation of $\mu_{1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$   
 $\mu_u = 3.0306E+008$

= 1.02974  
' = 0.91217079  
error of function (3.68), Biskinis Phd = 60360.02  
From 5A.2, TBDY:  $f_{cc} = f_c^* \quad c = 51.61391$

conf. factor  $c = 1.56406$

$f_c = 33.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00101663$

$N = 4771.233$

$Ac = 125663.706$

$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of  $\mu_{1-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$\mu = 3.0306E+008$

$= 1.02974$

$' = 0.91217079$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 51.61391$

conf. factor  $c = 1.56406$

$f_c = 33.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00101663$

$N = 4771.233$

$Ac = 125663.706$

$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of  $\mu_{2+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$\mu = 3.0306E+008$

$= 1.02974$

$' = 0.91217079$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 51.61391$

conf. factor  $c = 1.56406$

$f_c = 33.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00101663$

$N = 4771.233$

$Ac = 125663.706$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of  $\mu_2$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 3.0306 \times 10^8$

$$= 1.02974$$

$$' = 0.91217079$$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 51.61391$

conf. factor  $c = 1.56406$

$f_c = 33.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$$l_b/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00101663$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 451727.786$

Calculation of Shear Strength at edge 1,  $V_{r1} = 451727.786$

$V_{r1} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{Col0}$

$$V_{Col0} = 451727.786$$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

$f_c' = 33.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$$M/d = 2.00$$

$$\mu_u = 5.4715235 \times 10^{-12}$$

$$V_u = 6.8378665 \times 10^{-48}$$

$$d = 0.8 \cdot D = 320.00$$

$$N_u = 4771.233$$

$$A_g = 125663.706$$

From (11.5.4.8), ACI 318-14:  $V_s = 219326.297$

$$A_v = \sqrt{2} \cdot A_{stirrup} = 123370.055$$

$$f_y = 555.56$$

$$s = 100.00$$

$V_s$  is multiplied by  $\text{Col} = 0.00$

$$s/d = 0.3125$$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression,

where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta)$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = b1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, a1)|)$ , with:

total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 306911.784$

$b_w \cdot d = \rho \cdot d^2 / 4 = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 451727.786$

$V_{r2} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{Col0}$

$V_{Col0} = 451727.786$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\rho = 1$  (normal-weight concrete)

$f_c' = 33.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M / Vd = 2.00$

$\mu_u = 5.4715235E-012$

$\nu_u = 6.8378665E-048$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 219326.297$

$A_v = \rho \cdot A_{stirrup} = 123370.055$

$f_y = 555.56$

$s = 100.00$

$V_s$  is multiplied by  $\rho_{Col} = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections

$w_f / s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \cot a) \sin a$  which is more a generalised expression,

where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta)$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = b1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, a1)|)$ , with:

total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 306911.784$

$b_w \cdot d = \rho \cdot d^2 / 4 = 80424.772$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At local axis: 2

Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (b)

Section Type: rccs

## Constant Properties

Knowledge Factor,  $\phi = 0.90$

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Secondary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_b/l_d \geq 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $\epsilon_{fu} = 0.01$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

## Stepwise Properties

Bending Moment,  $M = -4.1943110E-011$

Shear Force,  $V_2 = 3759.209$

Shear Force,  $V_3 = 1.9910600E-013$

Axial Force,  $F = -4769.729$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{st} = 0.00$

-Compression:  $A_{sc} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{st,ten} = 1017.876$

-Compression:  $A_{sc,com} = 1017.876$

-Middle:  $A_{st,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $D_bL = 18.00$

New component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_{,R} = 1.0^* u = 0.01989311$

$u = y + p = 0.01989311$

- Calculation of  $y$  -

$y = (M_y * L_s / 3) / E_{eff} = 0.01489311$  ((4.29), Biskinis Phd)

$M_y = 3.0318E+008$

$L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 1500.00

From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 1.0179E+013$

factor = 0.30

$A_g = 125663.706$

$f_c' = 33.00$

$N = 4769.729$

$E_c * I_g = 3.3929E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\rho_y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 3.0318E+008$   
 $\rho_y \text{ ((10a) or (10b))} = 1.6222680E-005$   
 $M_{y\_ten} \text{ (8a)} = 3.0318E+008$   
 $\rho_{y\_ten} \text{ (7a)} = 69.35397$   
error of function (7a) = 0.00743258  
 $M_{y\_com} \text{ (8b)} = 4.6746E+008$   
 $\rho_{y\_com} \text{ (7b)} = 67.1564$   
error of function (7b) = -0.00265541  
with  $e_y = 0.0027778$   
 $e_{co} = 0.002$   
 $a_{pl} = 0.45 \text{ ((9c) in Biskinis and Fardis for FRP Wrap)}$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00102226$   
 $N = 4769.729$   
 $A_c = 125663.706$   
 $\rho = 0.36359274$   
with  $f_c^* \text{ ((12.3), ACI 440)} = 37.12975$   
 $f_c = 33.00$   
 $f_l = 1.3173$   
 $k = 1$   
Effective FRP thickness,  $t_f = N L^* t^* \text{Cos}(b_1) = 1.016$   
 $e_{fe} \text{ ((12.5) and (12.7))} = 0.004$   
 $E_f = 64828.00$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

- Calculation of  $\rho_p$  -

From table 10-9:  $\rho_p = 0.005$

with:

- Columns controlled by inadequate development or splicing along the clear height because  $l_b/d < 1$

shear control ratio  $V_y E / V_{CoI} E = 0.44725617$

$d = 0.00$

$s = 0.00$

$t = 2^*A_v / (d_c^*s) + 4^*t_f / D^*(f_{fe} / f_s) = 0.00$

$A_v = 78.53982$ , is the area of the circular stirrup

$d_c = D - 2^*cover$  - Hoop Diameter = 340.00

The term  $2^*t_f / b_w^*(f_{fe} / f_s)$  is implemented to account for FRP contribution

where  $f = 2^*t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength

All these variables have already been given in Shear control ratio calculation.

$N_{UD} = 4769.729$

$A_g = 125663.706$

$f_{cE} = 33.00$

$f_{yE} = f_{yI} E = 555.56$

$\rho_l = \text{Area\_Tot\_Long\_Rein} / (A_g) = 0.0243$

$f_{cE} = 33.00$

End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (b)

## Calculation No. 7

column C1, Floor 1

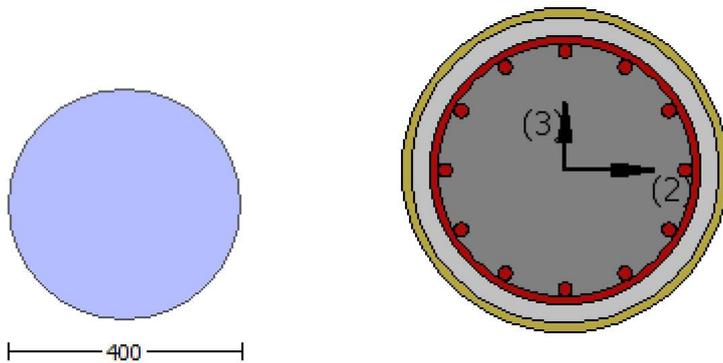
Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Shear capacity VRd

Edge: End

Local Axis: (3)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (b)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 0.90$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

New material of Secondary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 25.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 500.00$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of  $\gamma$  for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE41-17).

New material: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material: Steel Strength,  $f_s = f_{sm} = 555.56$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Secondary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou,min} = l_b/l_d \geq 1$ )

FRP Wrapping Data

Type: Carbon  
Cured laminate properties (design values)  
Thickness,  $t = 1.016$   
Tensile Strength,  $f_{fu} = 1055.00$   
Tensile Modulus,  $E_f = 64828.00$   
Elongation,  $e_{fu} = 0.01$   
Number of directions,  $NoDir = 1$   
Fiber orientations,  $b_i = 0.00^\circ$   
Number of layers,  $NL = 1$   
Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

EDGE -A-  
Bending Moment,  $M_a = 6.3952987E-010$   
Shear Force,  $V_a = -1.9910600E-013$   
EDGE -B-  
Bending Moment,  $M_b = -4.1943110E-011$   
Shear Force,  $V_b = 1.9910600E-013$   
BOTH EDGES  
Axial Force,  $F = -4769.729$   
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $As_{t1} = 0.00$   
-Compression:  $As_{c1} = 3053.628$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $As_{t,ten} = 1017.876$   
-Compression:  $As_{l,com} = 1017.876$   
-Middle:  $As_{l,mid} = 1017.876$   
Mean Diameter of Tension Reinforcement,  $Db_{L,ten} = 18.00$

New component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = 1.0 * V_n = 393301.081$   
 $V_n$  ((10.3), ASCE 41-17) =  $k_n * V_{Col} = 393301.081$   
 $V_{Col} = 393301.081$   
 $k_n = 1.00$   
 $displacement\_ductility\_demand = 0.00$

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)  
 $f'_c = 25.00$ , but  $f'_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M / Vd = 2.00$   
 $M_u = 4.1943110E-011$   
 $V_u = 1.9910600E-013$   
 $d = 0.8 * D = 320.00$   
 $N_u = 4769.729$   
 $A_g = 125663.706$   
From (11.5.4.8), ACI 318-14:  $V_s = 197392.088$   
 $A_v = \sqrt{2} * A_{stirrup} = 123370.055$   
 $f_y = 500.00$   
 $s = 100.00$   
 $V_s$  is multiplied by  $Col = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) =  $194961.134$   
 $f = 0.95$ , for fully-wrapped sections  
 $w_f / s_f = 1$  (FRP strips adjacent to one another).  
In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \csc) \sin \alpha$  which is more a generalised expression, where  $\alpha$  is the angle of the crack direction (see KANEPE).  
This later relation, considered as a function  $V_f(\alpha)$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $\alpha = 45^\circ$  and  $\alpha = -45^\circ$  to take into consideration the cyclic seismic loading.  
orientation 1:  $\alpha = b1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, a1)|)$ , with:  
total thickness per orientation,  $tf1 = NL * t / NoDir = 1.016$

$dfv = d$  (figure 11.2, ACI 440) = 370.00  
 $ffe$  ((11-5), ACI 440) = 259.312  
 $Ef = 64828.00$   
 $fe = 0.004$ , from (11.6a), ACI 440  
with  $fu = 0.01$   
From (11-11), ACI 440:  $Vs + Vf \leq 267132.42$   
 $bw*d = *d*d/4 = 80424.772$

displacement ductility demand is calculated as  $\phi / y$

- Calculation of  $\phi / y$  for END B -  
for rotation axis 2 and integ. section (b)

From analysis, chord rotation =  $9.9377781E-021$   
 $y = (My*Ls/3)/Eleff = 0.01489311$  ((4.29), Biskinis Phd)  
 $My = 3.0318E+008$   
 $Ls = M/V$  (with  $Ls > 0.1*L$  and  $Ls < 2*L$ ) = 1500.00  
From table 10.5, ASCE 41\_17:  $Eleff = factor*Ec*Ig = 1.0179E+013$   
 $factor = 0.30$   
 $Ag = 125663.706$   
 $fc' = 33.00$   
 $N = 4769.729$   
 $Ec*Ig = 3.3929E+013$

Calculation of Yielding Moment  $My$

Calculation of  $\phi$  and  $My$  according to (7) - (8) in Biskinis and Fardis

$My = \text{Min}(My_{ten}, My_{com}) = 3.0318E+008$   
 $y$  ((10a) or (10b)) =  $1.6222680E-005$   
 $My_{ten}$  (8a) =  $3.0318E+008$   
 $_{ten}$  (7a) = 69.35397  
error of function (7a) = 0.00743258  
 $My_{com}$  (8b) =  $4.6746E+008$   
 $_{com}$  (7b) = 67.1564  
error of function (7b) = -0.00265541  
with  $ey = 0.0027778$   
 $eco = 0.002$   
 $apl = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d1 = 44.00$   
 $R = 200.00$   
 $v = 0.00102226$   
 $N = 4769.729$   
 $Ac = 125663.706$   
= 0.36359274  
with  $fc^*$  ((12.3), ACI 440) = 37.12975  
 $fc = 33.00$   
 $fl = 1.3173$   
 $k = 1$   
Effective FRP thickness,  $tf = NL*t*\text{Cos}(b1) = 1.016$   
 $efe$  ((12.5) and (12.7)) = 0.004  
 $Ef = 64828.00$

Calculation of ratio  $lb/ld$

Adequate Lap Length:  $lb/ld \geq 1$

End Of Calculation of Shear Capacity for element: column CC1 of floor 1  
At local axis: 3  
Integration Section: (b)

## Calculation No. 8

column C1, Floor 1

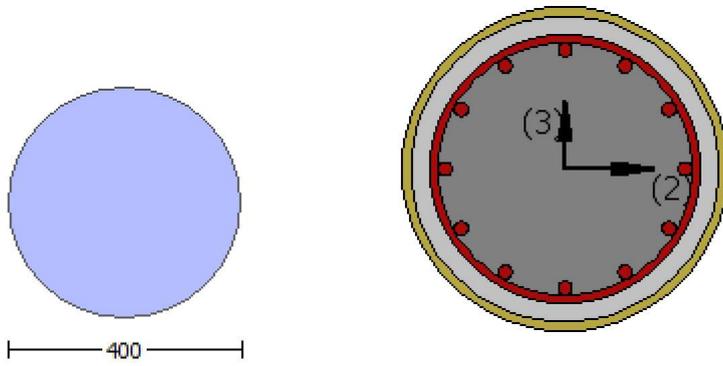
Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity ( $\theta_r$ )

Edge: End

Local Axis: (3)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\phi = 0.90$

Mean strength values are used for both shear and moment calculations.

Consequently:

New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

New material: Steel Strength,  $f_s = 1.25 * f_{sm} = 694.45$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.56406

Element Length,  $L = 3000.00$

Secondary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou, min} \geq 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $\epsilon_{fu} = 0.01$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

Stepwise Properties

At local axis: 3

EDGE -A-

Shear Force,  $V_a = 1.1167452E-031$

EDGE -B-

Shear Force,  $V_b = -1.1167452E-031$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{sl,t} = 0.00$

-Compression:  $A_{sl,c} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{sl,ten} = 1017.876$

-Compression:  $A_{sl,com} = 1017.876$

-Middle:  $A_{sl,mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.44725617$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 202038.041$

with

$M_{pr1} = \text{Max}(Mu_{1+}, Mu_{1-}) = 3.0306E+008$

$Mu_{1+} = 3.0306E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$Mu_{1-} = 3.0306E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(Mu_{2+}, Mu_{2-}) = 3.0306E+008$

$Mu_{2+} = 3.0306E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination

$Mu_{2-} = 3.0306E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

Calculation of  $Mu_{1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $Mu$

$Mu = 3.0306E+008$

$\lambda = 1.02974$

$\lambda' = 0.91217079$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 51.61391$

conf. factor  $c = 1.56406$

$f_c = 33.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00101663$

$$N = 4771.233$$
$$Ac = 125663.706$$
$$= *Min(1,1.25*(lb/d)^{2/3}) = 0.45184585$$

Calculation of ratio lb/d

Adequate Lap Length:  $lb/d \geq 1$

Calculation of Mu1-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 3.0306E+008

$$= 1.02974$$
$$' = 0.91217079$$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c * c = 51.61391$   
conf. factor  $c = 1.56406$   
 $f_c = 33.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y * Min(1,1.25*(lb/d)^{2/3}) = 694.45$   
 $lb/d = 1.00$   
 $d1 = 44.00$   
 $R = 200.00$   
 $v = 0.00101663$   
 $N = 4771.233$   
 $Ac = 125663.706$   
 $= *Min(1,1.25*(lb/d)^{2/3}) = 0.45184585$

Calculation of ratio lb/d

Adequate Lap Length:  $lb/d \geq 1$

Calculation of Mu2+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 3.0306E+008

$$= 1.02974$$
$$' = 0.91217079$$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c * c = 51.61391$   
conf. factor  $c = 1.56406$   
 $f_c = 33.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y * Min(1,1.25*(lb/d)^{2/3}) = 694.45$   
 $lb/d = 1.00$   
 $d1 = 44.00$   
 $R = 200.00$   
 $v = 0.00101663$   
 $N = 4771.233$   
 $Ac = 125663.706$   
 $= *Min(1,1.25*(lb/d)^{2/3}) = 0.45184585$

Calculation of ratio lb/d

Adequate Lap Length:  $lb/d \geq 1$

Calculation of Mu2-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 3.0306E+008

$$= 1.02974$$

$$' = 0.91217079$$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c' \cdot c = 51.61391$

$$\text{conf. factor } c = 1.56406$$

$$f_c = 33.00$$

From 10.3.5, ASCE41-17, Final value of fy:  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$$l_b/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00101663$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$$

Calculation of ratio lb/d

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 451727.786$

Calculation of Shear Strength at edge 1,  $V_{r1} = 451727.786$

$V_{r1} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{Col0}$

$$V_{Col0} = 451727.786$$

$$k_{nl} = 1 \text{ (zero step-static loading)}$$

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$$= 1 \text{ (normal-weight concrete)}$$

$$f_c' = 33.00, \text{ but } f_c'^{0.5} \leq 8.3 \text{ MPa (22.5.3.1, ACI 318-14)}$$

$$M/Vd = 2.00$$

$$Mu = 1.5353150E-011$$

$$Vu = 1.1167452E-031$$

$$d = 0.8 \cdot D = 320.00$$

$$Nu = 4771.233$$

$$Ag = 125663.706$$

From (11.5.4.8), ACI 318-14:  $V_s = 219326.297$

$$A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 123370.055$$

$$f_y = 555.56$$

$$s = 100.00$$

$V_s$  is multiplied by  $Col = 0.00$

$$s/d = 0.3125$$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$$f = 0.95, \text{ for fully-wrapped sections}$$

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = 45^\circ$  and  $\alpha = 90^\circ$

$$V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, 1)|), \text{ with:}$$

$$\text{total thickness per orientation, } t_{f1} = N_L \cdot t / N_{\text{Dir}} = 1.016$$

$$d_{fv} = d \text{ (figure 11.2, ACI 440)} = 370.00$$

$f_{fe}$  ((11-5), ACI 440) = 259.312  
 $E_f$  = 64828.00  
 $f_e$  = 0.004, from (11.6a), ACI 440  
with  $f_u$  = 0.01  
From (11-11), ACI 440:  $V_s + V_f \leq 306911.784$   
 $b_w \cdot d = \frac{V_s \cdot d}{4} = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 451727.786$   
 $V_{r2} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_n l \cdot V_{Col0}$   
 $V_{Col0} = 451727.786$   
 $k_n l = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ '  
where  $V_f$  is the contribution of FRPs ((11.3), ACI 440).

= 1 (normal-weight concrete)  
 $f_c' = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $M_u = 1.5353150E-011$   
 $V_u = 1.1167452E-031$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
From (11.5.4.8), ACI 318-14:  $V_s = 219326.297$   
 $A_v = \frac{V_s \cdot s}{f_y \cdot d} = 123370.055$   
 $f_y = 555.56$   
 $s = 100.00$

$V_s$  is multiplied by  $\phi_{Col} = 0.00$   
 $s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression,  
where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ ,  
as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = 45^\circ + 90^\circ = 135^\circ$

$V_f = \text{Min}(|V_f(45, \alpha_1)|, |V_f(-45, \alpha_1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{oDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f$  = 64828.00

$f_e$  = 0.004, from (11.6a), ACI 440

with  $f_u$  = 0.01

From (11-11), ACI 440:  $V_s + V_f \leq 306911.784$

$b_w \cdot d = \frac{V_s \cdot d}{4} = 80424.772$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At local axis: 3

Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At Shear local axis: 2  
(Bending local axis: 3)  
Section Type: rccs

Constant Properties

Knowledge Factor,  $\phi = 0.90$

Mean strength values are used for both shear and moment calculations.

Consequently:

New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,  
the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

New material: Steel Strength,  $f_s = 1.25 * f_{sm} = 694.45$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.56406

Element Length,  $L = 3000.00$

Secondary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou, \min} >= 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $NoDir = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

-----  
Stepwise Properties

-----  
At local axis: 2

EDGE -A-

Shear Force,  $V_a = -6.8378665E-048$

EDGE -B-

Shear Force,  $V_b = 6.8378665E-048$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{st} = 0.00$

-Compression:  $A_{sc} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{st, \text{ten}} = 1017.876$

-Compression:  $A_{sl, \text{com}} = 1017.876$

-Middle:  $A_{sl, \text{mid}} = 1017.876$

-----  
-----  
Calculation of Shear Capacity ratio,  $V_e/V_r = 0.44725617$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 202038.041$

with

$M_{pr1} = \text{Max}(M_{u1+}, M_{u1-}) = 3.0306E+008$

$M_{u1+} = 3.0306E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$M_{u1-} = 3.0306E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(M_{u2+}, M_{u2-}) = 3.0306E+008$

$M_{u2+} = 3.0306E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

$M_{u2-} = 3.0306E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

-----  
Calculation of  $M_{u1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 3.0306E+008

= 1.02974  
' = 0.91217079  
error of function (3.68), Biskinis Phd = 60360.02  
From 5A.2, TBDY: fcc = fc\* c = 51.61391  
conf. factor c = 1.56406  
fc = 33.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 694.45  
lb/d = 1.00  
d1 = 44.00  
R = 200.00  
v = 0.00101663  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.45184585

Calculation of ratio lb/d

Adequate Lap Length: lb/d >= 1

Calculation of Mu1-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 3.0306E+008

= 1.02974  
' = 0.91217079  
error of function (3.68), Biskinis Phd = 60360.02  
From 5A.2, TBDY: fcc = fc\* c = 51.61391  
conf. factor c = 1.56406  
fc = 33.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 694.45  
lb/d = 1.00  
d1 = 44.00  
R = 200.00  
v = 0.00101663  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.45184585

Calculation of ratio lb/d

Adequate Lap Length: lb/d >= 1

Calculation of Mu2+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 3.0306E+008

= 1.02974  
' = 0.91217079  
error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 51.61391$

conf. factor  $c = 1.56406$

$f_c = 33.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00101663$

$N = 4771.233$

$A_c = 125663.706$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of  $\mu_2$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$\mu = 3.0306E+008$

$\mu = 1.02974$

$\mu = 0.91217079$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 51.61391$

conf. factor  $c = 1.56406$

$f_c = 33.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00101663$

$N = 4771.233$

$A_c = 125663.706$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 451727.786$

Calculation of Shear Strength at edge 1,  $V_{r1} = 451727.786$

$V_{r1} = V_{Co1} ((10.3), ASCE 41-17) = k_{nl} \cdot V_{Co10}$

$V_{Co10} = 451727.786$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$f = 1$  (normal-weight concrete)

$f_c' = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/d = 2.00$

$\mu = 5.4715235E-012$

$V_u = 6.8378665E-048$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 219326.297$

$A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 123370.055$

$f_y = 555.56$

$s = 100.00$

$V_s$  is multiplied by  $\text{Col} = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin^2 + \cos^2$  is replaced with  $(\cot^2 + \csc^2)\sin^2\alpha$  which is more a generalised expression, where  $\alpha$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\alpha = 45^\circ$  and  $\alpha = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\alpha = b_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, a1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / \text{NoDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 306911.784$

$b_w \cdot d = \sqrt{2} \cdot d^2 / 4 = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 451727.786$

$V_{r2} = V_{\text{Col}}$  ((10.3), ASCE 41-17) =  $k_n \cdot V_{\text{Col0}}$

$V_{\text{Col0}} = 451727.786$

$k_n = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\gamma = 1$  (normal-weight concrete)

$f'_c = 33.00$ , but  $f'_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 5.4715235E-012$

$\nu_u = 6.8378665E-048$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 219326.297$

$A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 123370.055$

$f_y = 555.56$

$s = 100.00$

$V_s$  is multiplied by  $\text{Col} = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin^2 + \cos^2$  is replaced with  $(\cot^2 + \csc^2)\sin^2\alpha$  which is more a generalised expression, where  $\alpha$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\alpha = 45^\circ$  and  $\alpha = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\alpha = b_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, a1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / \text{NoDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 306911.784$

$b_w \cdot d = \sqrt{2} \cdot d^2 / 4 = 80424.772$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At local axis: 2

Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (b)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 0.90$

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Secondary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_b/l_d >= 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

Stepwise Properties

Bending Moment,  $M = 0.06021356$

Shear Force,  $V_2 = 3759.209$

Shear Force,  $V_3 = 1.9910600E-013$

Axial Force,  $F = -4769.729$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{st} = 0.00$

-Compression:  $A_{sc} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{s,ten} = 1017.876$

-Compression:  $A_{s,com} = 1017.876$

-Middle:  $A_{s,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $DbL = 18.00$

New component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u,R = 1.0^*$   $u = 0.00797862$

$u = \gamma + \rho = 0.00797862$

- Calculation of  $\gamma$  -

$\gamma = (M_y * L_s / 3) / E_{eff} = 0.00297862$  ((4.29), Biskinis Phd))

$M_y = 3.0318E+008$   
 $L_s = M/V$  (with  $L_s > 0.1*L$  and  $L_s < 2*L$ ) = 300.00  
From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 1.0179E+013$   
factor = 0.30  
Ag = 125663.706  
fc' = 33.00  
N = 4769.729  
 $E_c * I_g = 3.3929E+013$

-----  
-----  
Calculation of Yielding Moment  $M_y$

-----  
Calculation of  $\rho_y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

-----  
 $M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 3.0318E+008$   
 $\rho_y$  ((10a) or (10b)) = 1.6222680E-005  
 $M_{y\_ten}$  (8a) = 3.0318E+008  
 $\rho_{y\_ten}$  (7a) = 69.35397  
error of function (7a) = 0.00743258  
 $M_{y\_com}$  (8b) = 4.6746E+008  
 $\rho_{y\_com}$  (7b) = 67.1564  
error of function (7b) = -0.00265541  
with  $e_y = 0.0027778$   
eco = 0.002  
apl = 0.45 ((9c) in Biskinis and Fardis for FRP Wrap)  
d1 = 44.00  
R = 200.00  
v = 0.00102226  
N = 4769.729  
Ac = 125663.706  
= 0.36359274  
with  $f_c^*$  ((12.3), ACI 440) = 37.12975  
fc = 33.00  
fl = 1.3173  
k = 1  
Effective FRP thickness,  $t_f = NL * t * \text{Cos}(b1) = 1.016$   
efe ((12.5) and (12.7)) = 0.004  
Ef = 64828.00

-----  
Calculation of ratio  $l_b/d$

-----  
Adequate Lap Length:  $l_b/d \geq 1$

-----  
- Calculation of  $\rho_p$  -

-----  
From table 10-9:  $\rho_p = 0.005$

with:

- Columns controlled by inadequate development or splicing along the clear height because  $l_b/d < 1$

shear control ratio  $V_y E / V_{CoI} E = 0.44725617$

d = 0.00

s = 0.00

$t = 2 * A_v / (d_c * s) + 4 * t_f / D * (f_{fe} / f_s) = 0.00$

$A_v = 78.53982$ , is the area of the circular stirrup

$d_c = D - 2 * \text{cover} - \text{Hoop Diameter} = 340.00$

The term  $2 * t_f / b_w * (f_{fe} / f_s)$  is implemented to account for FRP contribution

where  $f = 2 * t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength

All these variables have already been given in Shear control ratio calculation.

NUD = 4769.729

Ag = 125663.706

f'cE = 33.00

fytE = fyIE = 555.56

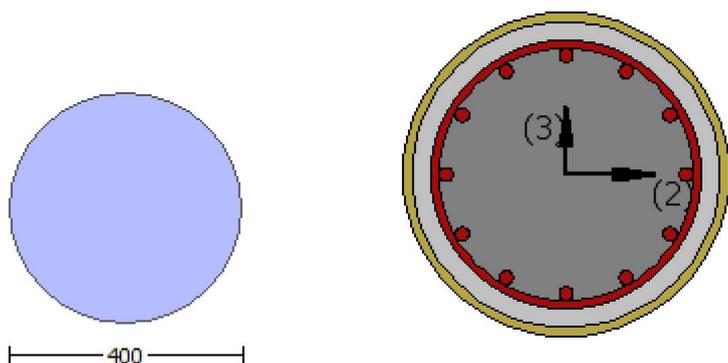
$\rho_l = \text{Area\_Tot\_Long\_Rein} / (A_g) = 0.0243$

f'cE = 33.00

End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1  
 At local axis: 3  
 Integration Section: (b)

## Calculation No. 9

column C1, Floor 1  
 Limit State: Collapse Prevention (data interpolation between analysis steps 1 and 2)  
 Analysis: Uniform +X  
 Check: Shear capacity VRd  
 Edge: Start  
 Local Axis: (2)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1  
 At local axis: 2  
 Integration Section: (a)  
 Section Type: rccs

### Constant Properties

Knowledge Factor,  $\gamma = 0.90$   
 Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE41-17.  
 Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17  
 Consequently:  
 New material of Secondary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 25.00$   
 New material of Secondary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 500.00$   
 Concrete Elasticity,  $E_c = 26999.444$   
 Steel Elasticity,  $E_s = 200000.00$   
 #####  
 Note: Especially for the calculation of  $\gamma$  for displacement ductility demand,  
 the expected (mean value) strengths are used (7.5.1.3, ASCE41-17) because bending is considered as  
 Deformation-Controlled Action (Table C7-1, ASCE41-17).  
 New material: Concrete Strength,  $f_c = f_{cm} = 33.00$   
 New material: Steel Strength,  $f_s = f_{sm} = 555.56$   
 #####  
 Diameter,  $D = 400.00$   
 Cover Thickness,  $c = 25.00$

Element Length, L = 3000.00  
Secondary Member  
Ribbed Bars  
Ductile Steel  
Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Adequate Lap Length ( $l_o/l_{ou,min} = l_b/l_d \geq 1$ )  
FRP Wrapping Data  
Type: Carbon  
Cured laminate properties (design values)  
Thickness, t = 1.016  
Tensile Strength,  $f_{fu} = 1055.00$   
Tensile Modulus,  $E_f = 64828.00$   
Elongation,  $e_{fu} = 0.01$   
Number of directions, NoDir = 1  
Fiber orientations,  $b_i = 0.00^\circ$   
Number of layers, NL = 1  
Radius of rounding corners, R = 40.00

#### Stepwise Properties

EDGE -A-  
Bending Moment,  $M_a = -9.0263E+006$   
Shear Force,  $V_a = -3007.62$   
EDGE -B-  
Bending Moment,  $M_b = 0.0481749$   
Shear Force,  $V_b = 3007.62$   
BOTH EDGES  
Axial Force, F = -4770.03  
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $As_t = 1272.345$   
-Compression:  $As_c = 1781.283$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $As_{t,ten} = 1017.876$   
-Compression:  $As_{c,com} = 1017.876$   
-Middle:  $As_{mid} = 1017.876$   
Mean Diameter of Tension Reinforcement,  $Db_{L,ten} = 18.00$

New component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = 1.0 \cdot V_n = 330216.78$   
 $V_n$  ((10.3), ASCE 41-17) =  $k_n \cdot V_{CoI} = 330216.78$   
 $V_{CoI} = 330216.78$   
 $k_n = 1.00$   
displacement\_ductility\_demand = 0.00814822

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ '  
where  $V_f$  is the contribution of FRPs ((11.3), ACI 440).

= 1 (normal-weight concrete)  
 $f_c' = 25.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 4.00$   
 $M_u = 9.0263E+006$   
 $V_u = 3007.62$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4770.03$   
 $A_g = 125663.706$   
From (11.5.4.8), ACI 318-14:  $V_s = 197392.088$   
 $A_v = /2 \cdot A_{stirrup} = 123370.055$   
 $f_y = 500.00$   
 $s = 100.00$   
 $V_s$  is multiplied by  $CoI = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections

$wf/sf = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \cot \alpha) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = b1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \alpha)|)$ , with:

total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 267132.42$

$b_w \cdot d = \frac{V_s \cdot d}{4} = 80424.772$

displacement ductility demand is calculated as  $\delta / y$

- Calculation of  $\delta / y$  for END A -  
for rotation axis 3 and integ. section (a)

From analysis, chord rotation  $\theta = 0.0002428$

$y = (M_y \cdot L_s / 3) / E_{eff} = 0.02979765$  ((4.29), Biskinis Phd))

$M_y = 3.0318E+008$

$L_s = M/V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) = 3001.151

From table 10.5, ASCE 41\_17:  $E_{eff} = \text{factor} \cdot E_c \cdot I_g = 1.0179E+013$

factor = 0.30

$A_g = 125663.706$

$f_c' = 33.00$

$N = 4770.03$

$E_c \cdot I_g = 3.3929E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\delta / y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 3.0318E+008$

$y$  ((10a) or (10b)) = 1.6222681E-005

$M_{y\_ten}$  (8a) = 3.0318E+008

$\delta_{y\_ten}$  (7a) = 69.35398

error of function (7a) = 0.00743251

$M_{y\_com}$  (8b) = 4.6746E+008

$\delta_{y\_com}$  (7b) = 67.1564

error of function (7b) = -0.0026554

with  $e_y = 0.0027778$

$e_{co} = 0.002$

$\alpha_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)

$d_1 = 44.00$

$R = 200.00$

$v = 0.00102233$

$N = 4770.03$

$A_c = 125663.706$

$= 0.36359274$

with  $f_c'$  ((12.3), ACI 440) = 37.12975

$f_c = 33.00$

$f_l = 1.3173$

$k = 1$

Effective FRP thickness,  $t_f = NL \cdot t \cdot \cos(b_1) = 1.016$

$e_{fe}$  ((12.5) and (12.7)) = 0.004

$E_f = 64828.00$

Calculation of ratio  $I_b / I_d$

Adequate Lap Length:  $l_b/d \geq 1$

End Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (a)

## Calculation No. 10

column C1, Floor 1

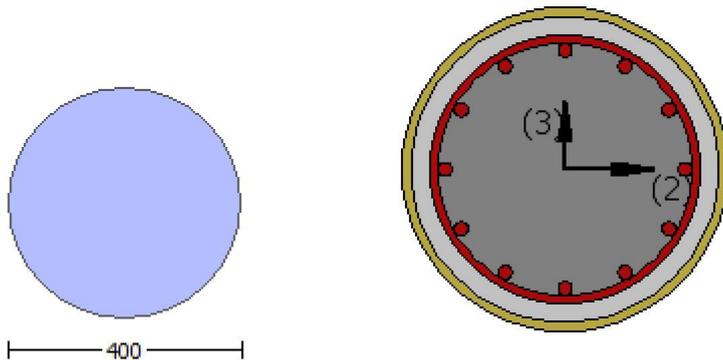
Limit State: Collapse Prevention (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity (  $\theta$  )

Edge: Start

Local Axis: (2)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\phi = 0.90$

Mean strength values are used for both shear and moment calculations.

Consequently:

New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 694.45$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.56406  
Element Length, L = 3000.00  
Secondary Member  
Ribbed Bars  
Ductile Steel  
Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Adequate Lap Length ( $l_o/l_{ou}, \min > = 1$ )  
FRP Wrapping Data  
Type: Carbon  
Cured laminate properties (design values)  
Thickness, t = 1.016  
Tensile Strength,  $f_{fu} = 1055.00$   
Tensile Modulus,  $E_f = 64828.00$   
Elongation,  $e_{fu} = 0.01$   
Number of directions, NoDir = 1  
Fiber orientations,  $b_i = 0.00^\circ$   
Number of layers, NL = 1  
Radius of rounding corners, R = 40.00

-----  
Stepwise Properties  
-----

At local axis: 3  
EDGE -A-  
Shear Force,  $V_a = 1.1167452E-031$   
EDGE -B-  
Shear Force,  $V_b = -1.1167452E-031$   
BOTH EDGES  
Axial Force, F = -4771.233  
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $A_{st} = 0.00$   
-Compression:  $A_{sc} = 3053.628$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $A_{st,ten} = 1017.876$   
-Compression:  $A_{st,com} = 1017.876$   
-Middle:  $A_{st,mid} = 1017.876$   
-----  
-----

Calculation of Shear Capacity ratio ,  $V_e/V_r = 0.44725617$   
Member Controlled by Flexure ( $V_e/V_r < 1$ )  
Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 202038.041$   
with  
 $M_{pr1} = \text{Max}(\mu_{u1+} , \mu_{u1-}) = 3.0306E+008$   
 $\mu_{u1+} = 3.0306E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction  
which is defined for the static loading combination  
 $\mu_{u1-} = 3.0306E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment  
direction which is defined for the static loading combination  
 $M_{pr2} = \text{Max}(\mu_{u2+} , \mu_{u2-}) = 3.0306E+008$   
 $\mu_{u2+} = 3.0306E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction  
which is defined for the the static loading combination  
 $\mu_{u2-} = 3.0306E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment  
direction which is defined for the the static loading combination

-----  
Calculation of  $\mu_{u1+}$   
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$   
 $\mu_u = 3.0306E+008$

-----  
= 1.02974  
' = 0.91217079  
error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 51.61391$

conf. factor  $c = 1.56406$

$f_c = 33.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00101663$

$N = 4771.233$

$A_c = 125663.706$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of  $\mu_{1-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$\mu = 3.0306E+008$

$= 1.02974$

$' = 0.91217079$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 51.61391$

conf. factor  $c = 1.56406$

$f_c = 33.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00101663$

$N = 4771.233$

$A_c = 125663.706$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of  $\mu_{2+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$\mu = 3.0306E+008$

$= 1.02974$

$' = 0.91217079$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 51.61391$

conf. factor  $c = 1.56406$

$f_c = 33.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00101663$

$N = 4771.233$

$$A_c = 125663.706$$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of  $\mu_2$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 3.0306E+008$

$$= 1.02974$$

$$\mu = 0.91217079$$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 51.61391$

conf. factor  $c = 1.56406$

$f_c = 33.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00101663$

$N = 4771.233$

$A_c = 125663.706$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 451727.786$

Calculation of Shear Strength at edge 1,  $V_{r1} = 451727.786$

$V_{r1} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{Col0}$

$V_{Col0} = 451727.786$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)  
 $f_c' = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/d = 2.00$

$\mu = 1.5353150E-011$

$V_u = 1.1167452E-031$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 219326.297$

$A_v = \sqrt{2} \cdot A_{stirrup} = 123370.055$

$f_y = 555.56$

$s = 100.00$

$V_s$  is multiplied by  $Col = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE). This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = 45^\circ + 90^\circ = 135^\circ$   
 $V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \alpha)|)$ , with:  
total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}$  ((11-5), ACI 440) = 259.312  
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
with  $f_u = 0.01$   
From (11-11), ACI 440:  $V_s + V_f \leq 306911.784$   
 $b_w \cdot d = \sqrt{V_s \cdot V_f} = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 451727.786$   
 $V_{r2} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{Col0}$   
 $V_{Col0} = 451727.786$   
 $k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\gamma = 1$  (normal-weight concrete)  
 $f'_c = 33.00$ , but  $f'_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M / Vd = 2.00$   
 $\mu_u = 1.5353150E-011$   
 $\nu_u = 1.1167452E-031$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
From (11.5.4.8), ACI 318-14:  $V_s = 219326.297$   
 $A_v = \sqrt{2} \cdot A_{stirrup} = 123370.055$   
 $f_y = 555.56$   
 $s = 100.00$   
 $V_s$  is multiplied by  $\gamma_{Col} = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections  
 $w_f / s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE). This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = 45^\circ + 90^\circ = 135^\circ$   
 $V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \alpha)|)$ , with:  
total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}$  ((11-5), ACI 440) = 259.312  
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
with  $f_u = 0.01$   
From (11-11), ACI 440:  $V_s + V_f \leq 306911.784$   
 $b_w \cdot d = \sqrt{V_s \cdot V_f} = 80424.772$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At local axis: 3

Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At Shear local axis: 2  
(Bending local axis: 3)  
Section Type: rccs

Constant Properties

Knowledge Factor,  $\phi = 0.90$

Mean strength values are used for both shear and moment calculations.

Consequently:

New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 694.45$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.56406

Element Length,  $L = 3000.00$

Secondary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou, \min} >= 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

Stepwise Properties

At local axis: 2

EDGE -A-

Shear Force,  $V_a = -6.8378665E-048$

EDGE -B-

Shear Force,  $V_b = 6.8378665E-048$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{sl,t} = 0.00$

-Compression:  $A_{sl,c} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{sl,ten} = 1017.876$

-Compression:  $A_{sl,com} = 1017.876$

-Middle:  $A_{sl,mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.44725617$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 202038.041$

with

$M_{pr1} = \text{Max}(M_{u1+}, M_{u1-}) = 3.0306E+008$

$M_{u1+} = 3.0306E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$M_{u1-} = 3.0306E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$$M_{pr2} = \text{Max}(M_{u2+}, M_{u2-}) = 3.0306E+008$$

$M_{u2+} = 3.0306E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

$M_{u2-} = 3.0306E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

-----  
Calculation of  $M_{u1+}$   
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$   
 $M_u = 3.0306E+008$   
-----

$$= 1.02974$$

$$\lambda = 0.91217079$$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c \cdot \lambda = 51.61391$

$$\text{conf. factor } \lambda = 1.56406$$

$$f_c = 33.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$$l_b/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00101663$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \lambda \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$$
  
-----

Calculation of ratio  $l_b/d$   
-----

Adequate Lap Length:  $l_b/d \geq 1$   
-----  
-----

Calculation of  $M_{u1-}$   
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$   
 $M_u = 3.0306E+008$   
-----

$$= 1.02974$$

$$\lambda = 0.91217079$$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c \cdot \lambda = 51.61391$

$$\text{conf. factor } \lambda = 1.56406$$

$$f_c = 33.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$$l_b/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00101663$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \lambda \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$$
  
-----

Calculation of ratio  $l_b/d$   
-----

Adequate Lap Length:  $l_b/d \geq 1$   
-----  
-----

Calculation of  $M_{u2+}$   
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu

$$\text{Mu} = 3.0306\text{E}+008$$

$$= 1.02974$$

$$' = 0.91217079$$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 51.61391$

$$\text{conf. factor } c = 1.56406$$

$$f_c = 33.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$$l_b/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00101663$$

$$N = 4771.233$$

$$Ac = 125663.706$$

$$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$$

-----  
Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

-----  
Calculation of Mu2-

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu

$$\text{Mu} = 3.0306\text{E}+008$$

$$= 1.02974$$

$$' = 0.91217079$$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 51.61391$

$$\text{conf. factor } c = 1.56406$$

$$f_c = 33.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$$l_b/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00101663$$

$$N = 4771.233$$

$$Ac = 125663.706$$

$$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$$

-----  
Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

-----  
Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 451727.786$

-----  
Calculation of Shear Strength at edge 1,  $V_{r1} = 451727.786$

$V_{r1} = V_{Co1}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{Co10}$

$$V_{Co10} = 451727.786$$

$$k_{nl} = 1 \text{ (zero step-static loading)}$$

-----  
NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ '  
where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\lambda = 1$  (normal-weight concrete)  
 $f_c' = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $M_u = 5.4715235E-012$   
 $V_u = 6.8378665E-048$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 219326.297$   
 $A_v = \lambda/2 \cdot A_{stirrup} = 123370.055$   
 $f_y = 555.56$   
 $s = 100.00$   
 $V_s$  is multiplied by  $\lambda = 1.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $\lambda = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin^2 \theta + \cos^2 \theta$  is replaced with  $(\cot^2 \theta + \csc^2 \theta) \sin^2 \theta$  which is more a generalised expression,  
 where  $\theta$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, \lambda)$ , is implemented for every different fiber orientation  $\theta_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\lambda = b_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \lambda)|, |V_f(-45, \lambda)|)$ , with:  
 total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{Dir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}$  ((11-5), ACI 440) = 259.312  
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 306911.784$   
 $b_w \cdot d = \lambda \cdot d^2 / 4 = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 451727.786$   
 $V_{r2} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_n \lambda V_{Col0}$   
 $V_{Col0} = 451727.786$   
 $k_n = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + \lambda \cdot V_f$ '  
 where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\lambda = 1$  (normal-weight concrete)  
 $f_c' = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $M_u = 5.4715235E-012$   
 $V_u = 6.8378665E-048$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 219326.297$   
 $A_v = \lambda/2 \cdot A_{stirrup} = 123370.055$   
 $f_y = 555.56$   
 $s = 100.00$   
 $V_s$  is multiplied by  $\lambda = 1.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $\lambda = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin^2 \theta + \cos^2 \theta$  is replaced with  $(\cot^2 \theta + \csc^2 \theta) \sin^2 \theta$  which is more a generalised expression,  
 where  $\theta$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, \lambda)$ , is implemented for every different fiber orientation  $\theta_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\lambda = b_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \lambda)|, |V_f(-45, \lambda)|)$ , with:  
 total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{Dir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}$  ((11-5), ACI 440) = 259.312

Ef = 64828.00

fe = 0.004, from (11.6a), ACI 440

with fu = 0.01

From (11-11), ACI 440: Vs + Vf <= 306911.784

bw\*d = \*d\*d/4 = 80424.772

-----  
End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At local axis: 2

-----  
Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (a)

Section Type: rccs

Constant Properties

-----  
Knowledge Factor, = 0.90

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

New material of Secondary Member: Concrete Strength, fc = fcm = 33.00

New material of Secondary Member: Steel Strength, fs = fsm = 555.56

Concrete Elasticity, Ec = 26999.444

Steel Elasticity, Es = 200000.00

Diameter, D = 400.00

Cover Thickness, c = 25.00

Element Length, L = 3000.00

Secondary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length (lb/lD >= 1)

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness, t = 1.016

Tensile Strength, ffu = 1055.00

Tensile Modulus, Ef = 64828.00

Elongation, efu = 0.01

Number of directions, NoDir = 1

Fiber orientations, bi: 0.00°

Number of layers, NL = 1

Radius of rounding corners, R = 40.00

-----  
Stepwise Properties

Bending Moment, M = 5.1473654E-010

Shear Force, V2 = -3007.62

Shear Force, V3 = -1.5929820E-013

Axial Force, F = -4770.03

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension: Aslt = 1272.345

-Compression: Aslc = 1781.283

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension: Asl,ten = 1017.876

-Compression: Asl,com = 1017.876

-Middle: Asl,mid = 1017.876

Mean Diameter of Tension Reinforcement, DbL = 18.00

New component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u, R = 1.0^*$   $u = 0.0864352$   
 $u = y + p = 0.0864352$

-----  
- Calculation of  $y$  -  
-----

$y = (M_y * L_s / 3) / E_{eff} = 0.01489311$  ((4.29), Biskinis Phd)  
 $M_y = 3.0318E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 1500.00  
From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 1.0179E+013$   
factor = 0.30  
Ag = 125663.706  
fc' = 33.00  
N = 4770.03  
 $E_c * I_g = 3.3929E+013$

-----  
-----  
Calculation of Yielding Moment  $M_y$   
-----

Calculation of  $y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis  
-----

$M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 3.0318E+008$   
 $y$  ((10a) or (10b)) = 1.6222681E-005  
 $M_{y\_ten}$  (8a) = 3.0318E+008  
\_ten (7a) = 69.35398  
error of function (7a) = 0.00743251  
 $M_{y\_com}$  (8b) = 4.6746E+008  
\_com (7b) = 67.1564  
error of function (7b) = -0.0026554  
with  $e_y = 0.0027778$   
eco = 0.002  
apl = 0.45 ((9c) in Biskinis and Fardis for FRP Wrap)  
d1 = 44.00  
R = 200.00  
v = 0.00102233  
N = 4770.03  
Ac = 125663.706  
= 0.36359274  
with  $f_c^*$  ((12.3), ACI 440) = 37.12975  
fc = 33.00  
fl = 1.3173  
k = 1  
Effective FRP thickness,  $t_f = NL * t * \text{Cos}(b1) = 1.016$   
efe ((12.5) and (12.7)) = 0.004  
Ef = 64828.00

-----  
Calculation of ratio  $l_b / l_d$   
-----

Adequate Lap Length:  $l_b / l_d >= 1$   
-----

- Calculation of  $p$  -  
-----

From table 10-9:  $p = 0.07154209$

with:

- Columns controlled by inadequate development or splicing along the clear height because  $l_b / l_d < 1$   
shear control ratio  $V_y E / V_{CoI} O E = 0.44725617$   
d = 0.00  
s = 0.00  
 $t = 2 * A_v / (d_c * s) + 4 * t_f / D * (f_{fe} / f_s) = 0.00$   
Av = 78.53982, is the area of the circular stirrup  
dc = D - 2 \* cover - Hoop Diameter = 340.00  
The term  $2 * t_f / b_w * (f_{fe} / f_s)$  is implemented to account for FRP contribution  
where  $f = 2 * t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength

All these variables have already been given in Shear control ratio calculation.

$NUD = 4770.03$

$Ag = 125663.706$

$f_{cE} = 33.00$

$f_{ytE} = f_{ylE} = 555.56$

$pl = \text{Area\_Tot\_Long\_Rein}/(Ag) = 0.0243$

$f_{cE} = 33.00$

End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (a)

## Calculation No. 11

column C1, Floor 1

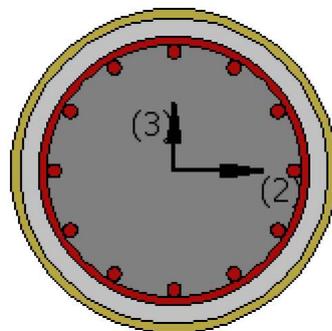
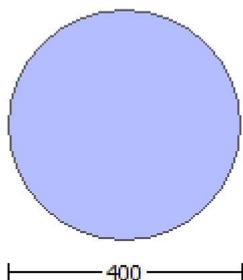
Limit State: Collapse Prevention (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Shear capacity  $VRd$

Edge: Start

Local Axis: (3)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (a)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 0.90$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

New material of Secondary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 25.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 500.00$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of  $\mu$  for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE41-17).

New material: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material: Steel Strength,  $f_s = f_{sm} = 555.56$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Secondary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou,min} = l_b/l_d \geq 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $NoDir = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

-----  
Stepwise Properties

-----  
EDGE -A-

Bending Moment,  $M_a = 5.1473654E-010$

Shear Force,  $V_a = -1.5929820E-013$

EDGE -B-

Bending Moment,  $M_b = -3.6626908E-011$

Shear Force,  $V_b = 1.5929820E-013$

BOTH EDGES

Axial Force,  $F = -4770.03$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 1272.345$

-Compression:  $As_c = 1781.283$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{l,ten} = 1017.876$

-Compression:  $As_{l,com} = 1017.876$

-Middle:  $As_{l,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $Db_{L,ten} = 18.00$

-----  
New component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = 1.0 \cdot V_n = 393301.141$

$V_n$  ((10.3), ASCE 41-17) =  $k_n \cdot V_{CoI} = 393301.141$

$V_{CoI} = 393301.141$

$k_n = 1.00$

$displacement\_ductility\_demand = 0.00$

-----  
NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_{s+} = f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

-----  
 $\gamma = 1$  (normal-weight concrete)

$f_c' = 25.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$M_u = 5.1473654E-010$

$V_u = 1.5929820E-013$

$d = 0.8 \cdot D = 320.00$

$N_u = 4770.03$

$$A_g = 125663.706$$

$$\text{From (11.5.4.8), ACI 318-14: } V_s = 197392.088$$

$$A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 123370.055$$

$$f_y = 500.00$$

$$s = 100.00$$

$$V_s \text{ is multiplied by } \text{Col} = 0.00$$

$$s/d = 0.3125$$

$$V_f \text{ ((11-3)-(11.4), ACI 440) = 194961.134}$$

$$f = 0.95, \text{ for fully-wrapped sections}$$

$$w_f/s_f = 1 \text{ (FRP strips adjacent to one another).}$$

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression,

where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

$$\text{orientation 1: } \theta_1 = \theta_1 + 90^\circ = 90.00$$

$$V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, \theta_1)|), \text{ with:}$$

$$\text{total thickness per orientation, } t_{f1} = N_L \cdot t / N_{\text{Dir}} = 1.016$$

$$d_{fv} = d \text{ (figure 11.2, ACI 440) = 370.00}$$

$$f_{fe} \text{ ((11-5), ACI 440) = 259.312}$$

$$E_f = 64828.00$$

$$f_e = 0.004, \text{ from (11.6a), ACI 440}$$

$$\text{with } f_u = 0.01$$

$$\text{From (11-11), ACI 440: } V_s + V_f \leq 267132.42$$

$$b_w \cdot d = \sqrt{2} \cdot d^2 / 4 = 80424.772$$

-----  
displacement\_ductility\_demand is calculated as  $\delta / y$

- Calculation of  $\delta / y$  for END A -

for rotation axis 2 and integ. section (a)

$$\text{From analysis, chord rotation } \theta = 1.5070001\text{E-}020$$

$$y = (M_y \cdot L_s / 3) / E_{\text{eff}} = 0.01489311 \text{ ((4.29), Biskinis Phd)}$$

$$M_y = 3.0318\text{E+}008$$

$$L_s = M/V \text{ (with } L_s > 0.1 \cdot L \text{ and } L_s < 2 \cdot L) = 1500.00$$

$$\text{From table 10.5, ASCE 41_17: } E_{\text{eff}} = \text{factor} \cdot E_c \cdot I_g = 1.0179\text{E+}013$$

$$\text{factor} = 0.30$$

$$A_g = 125663.706$$

$$f_c' = 33.00$$

$$N = 4770.03$$

$$E_c \cdot I_g = 3.3929\text{E+}013$$

-----  
Calculation of Yielding Moment  $M_y$

-----  
Calculation of  $\delta / y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$$M_y = \text{Min}(M_{y_{\text{ten}}}, M_{y_{\text{com}}}) = 3.0318\text{E+}008$$

$$y \text{ ((10a) or (10b))} = 1.6222681\text{E-}005$$

$$M_{y_{\text{ten}}} \text{ (8a)} = 3.0318\text{E+}008$$

$$y_{\text{ten}} \text{ (7a)} = 69.35398$$

$$\text{error of function (7a)} = 0.00743251$$

$$M_{y_{\text{com}}} \text{ (8b)} = 4.6746\text{E+}008$$

$$y_{\text{com}} \text{ (7b)} = 67.1564$$

$$\text{error of function (7b)} = -0.0026554$$

$$\text{with } e_y = 0.0027778$$

$$e_{co} = 0.002$$

$$a_{pl} = 0.45 \text{ ((9c) in Biskinis and Fardis for FRP Wrap)}$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00102233$$

$$N = 4770.03$$

$$A_c = 125663.706$$

$$= 0.36359274$$

$$\text{with } f_c^* \text{ ((12.3), ACI 440) = 37.12975}$$

$f_c = 33.00$   
 $f_l = 1.3173$   
 $k = 1$   
Effective FRP thickness,  $t_f = NL \cdot t \cdot \cos(b_1) = 1.016$   
efe ((12.5) and (12.7)) = 0.004  
 $E_f = 64828.00$

Calculation of ratio  $l_b/l_d$

Adequate Lap Length:  $l_b/l_d \geq 1$

End Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (a)

## Calculation No. 12

column C1, Floor 1

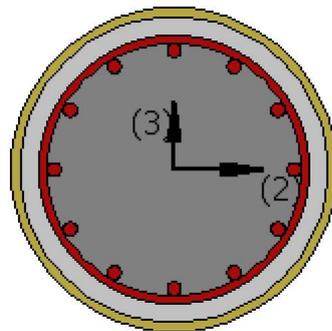
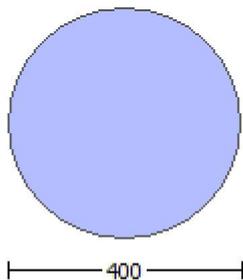
Limit State: Collapse Prevention (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity (  $\theta$  )

Edge: Start

Local Axis: (3)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 0.90$

Mean strength values are used for both shear and moment calculations.

Consequently:

New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,  
the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

New material: Steel Strength,  $f_s = 1.25 * f_{sm} = 694.45$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.56406

Element Length,  $L = 3000.00$

Secondary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou, \min} >= 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $NoDir = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

-----  
Stepwise Properties

-----  
At local axis: 3

EDGE -A-

Shear Force,  $V_a = 1.1167452E-031$

EDGE -B-

Shear Force,  $V_b = -1.1167452E-031$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{st} = 0.00$

-Compression:  $A_{sc} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{st, \text{ten}} = 1017.876$

-Compression:  $A_{st, \text{com}} = 1017.876$

-Middle:  $A_{st, \text{mid}} = 1017.876$

-----  
Calculation of Shear Capacity ratio,  $V_e/V_r = 0.44725617$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 202038.041$

with

$M_{pr1} = \text{Max}(M_{u1+}, M_{u1-}) = 3.0306E+008$

$M_{u1+} = 3.0306E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$M_{u1-} = 3.0306E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(M_{u2+}, M_{u2-}) = 3.0306E+008$

$M_{u2+} = 3.0306E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

$M_{u2-} = 3.0306E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

-----  
Calculation of  $M_{u1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 3.0306E+008

= 1.02974  
' = 0.91217079  
error of function (3.68), Biskinis Phd = 60360.02  
From 5A.2, TBDY: fcc = fc\* c = 51.61391  
conf. factor c = 1.56406  
fc = 33.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 694.45  
lb/d = 1.00  
d1 = 44.00  
R = 200.00  
v = 0.00101663  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.45184585

Calculation of ratio lb/d

Adequate Lap Length: lb/d >= 1

Calculation of Mu1-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 3.0306E+008

= 1.02974  
' = 0.91217079  
error of function (3.68), Biskinis Phd = 60360.02  
From 5A.2, TBDY: fcc = fc\* c = 51.61391  
conf. factor c = 1.56406  
fc = 33.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 694.45  
lb/d = 1.00  
d1 = 44.00  
R = 200.00  
v = 0.00101663  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.45184585

Calculation of ratio lb/d

Adequate Lap Length: lb/d >= 1

Calculation of Mu2+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 3.0306E+008

= 1.02974  
' = 0.91217079  
error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 51.61391$

conf. factor  $c = 1.56406$

$f_c = 33.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00101663$

$N = 4771.233$

$A_c = 125663.706$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of  $\mu_2$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$\mu = 3.0306E+008$

$\mu = 1.02974$

$\mu = 0.91217079$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 51.61391$

conf. factor  $c = 1.56406$

$f_c = 33.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00101663$

$N = 4771.233$

$A_c = 125663.706$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 451727.786$

Calculation of Shear Strength at edge 1,  $V_{r1} = 451727.786$

$V_{r1} = V_{Co1} ((10.3), ASCE 41-17) = k_{nl} \cdot V_{Co10}$

$V_{Co10} = 451727.786$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$f = 1$  (normal-weight concrete)

$f_c' = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/d = 2.00$

$\mu = 1.5353150E-011$

$\nu = 1.1167452E-031$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 219326.297$

$A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 123370.055$

$f_y = 555.56$

$s = 100.00$

$V_s$  is multiplied by  $\text{Col} = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin^2 + \cos^2$  is replaced with  $(\cot^2 + \csc^2)\sin^2\alpha$  which is more a generalised expression, where  $\alpha$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\alpha = 45^\circ$  and  $\alpha = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\alpha = b_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, a1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / \text{NoDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 306911.784$

$b_w \cdot d = \sqrt{2} \cdot d^2 / 4 = 80424.772$

-----  
Calculation of Shear Strength at edge 2,  $V_{r2} = 451727.786$

$V_{r2} = V_{\text{Col}}$  ((10.3), ASCE 41-17) =  $k_n \cdot V_{\text{Col0}}$

$V_{\text{Col0}} = 451727.786$

$k_n = 1$  (zero step-static loading)

-----  
NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

-----  
 $\alpha = 1$  (normal-weight concrete)

$f'_c = 33.00$ , but  $f'_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.5353150E-011$

$\nu_u = 1.1167452E-031$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 219326.297$

$A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 123370.055$

$f_y = 555.56$

$s = 100.00$

$V_s$  is multiplied by  $\text{Col} = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin^2 + \cos^2$  is replaced with  $(\cot^2 + \csc^2)\sin^2\alpha$  which is more a generalised expression, where  $\alpha$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\alpha = 45^\circ$  and  $\alpha = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\alpha = b_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, a1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / \text{NoDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 306911.784$

$b_w \cdot d = \sqrt{2} \cdot d^2 / 4 = 80424.772$

-----  
End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At local axis: 3

Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 2

(Bending local axis: 3)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 0.90$

Mean strength values are used for both shear and moment calculations.

Consequently:

New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

New material: Steel Strength,  $f_s = 1.25 * f_{sm} = 694.45$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.56406

Element Length,  $L = 3000.00$

Secondary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou}, \min >= 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

Stepwise Properties

At local axis: 2

EDGE -A-

Shear Force,  $V_a = -6.8378665E-048$

EDGE -B-

Shear Force,  $V_b = 6.8378665E-048$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 0.00$

-Compression:  $As_c = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{l,ten} = 1017.876$

-Compression:  $As_{l,com} = 1017.876$

-Middle:  $As_{l,mid} = 1017.876$

Calculation of Shear Capacity ratio ,  $V_e/V_r = 0.44725617$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 202038.041$

with

$M_{pr1} = \text{Max}(M_{u1+} , M_{u1-}) = 3.0306E+008$

$M_{u1+} = 3.0306E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$M_{u1-} = 3.0306E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(M_{u2+} , M_{u2-}) = 3.0306E+008$

$M_{u2+} = 3.0306E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

$M_{u2-} = 3.0306E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

-----  
Calculation of  $M_{u1+}$   
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$

$M_u = 3.0306E+008$

-----  
= 1.02974

' = 0.91217079

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c^* c = 51.61391$

conf. factor  $c = 1.56406$

$f_c = 33.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00101663$

$N = 4771.233$

$A_c = 125663.706$

=  $\cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$   
-----

Calculation of ratio  $l_b/d$   
-----

Adequate Lap Length:  $l_b/d \geq 1$   
-----  
-----

-----  
Calculation of  $M_{u1-}$   
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$

$M_u = 3.0306E+008$

-----  
= 1.02974

' = 0.91217079

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c^* c = 51.61391$

conf. factor  $c = 1.56406$

$f_c = 33.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00101663$

$N = 4771.233$

$A_c = 125663.706$

=  $\cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$   
-----

Calculation of ratio lb/d

Adequate Lap Length:  $lb/d \geq 1$

Calculation of  $\mu_{2+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 3.0306E+008$

$$= 1.02974$$

$$' = 0.91217079$$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 51.61391$

conf. factor  $c = 1.56406$

$f_c = 33.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (lb/d)^{2/3}) = 694.45$

$$lb/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00101663$$

$$N = 4771.233$$

$$Ac = 125663.706$$

$$= \cdot \text{Min}(1, 1.25 \cdot (lb/d)^{2/3}) = 0.45184585$$

Calculation of ratio lb/d

Adequate Lap Length:  $lb/d \geq 1$

Calculation of  $\mu_{2-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 3.0306E+008$

$$= 1.02974$$

$$' = 0.91217079$$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 51.61391$

conf. factor  $c = 1.56406$

$f_c = 33.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (lb/d)^{2/3}) = 694.45$

$$lb/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00101663$$

$$N = 4771.233$$

$$Ac = 125663.706$$

$$= \cdot \text{Min}(1, 1.25 \cdot (lb/d)^{2/3}) = 0.45184585$$

Calculation of ratio lb/d

Adequate Lap Length:  $lb/d \geq 1$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 451727.786$

-----  
Calculation of Shear Strength at edge 1,  $Vr1 = 451727.786$

$Vr1 = VCol$  ((10.3), ASCE 41-17) =  $knl * VColO$

$VColO = 451727.786$

$knl = 1$  (zero step-static loading)

-----  
NOTE: In expression (10-3) ' $Vs = Av * fy * d / s$ ' is replaced by ' $Vs + f * Vf$ '  
where  $Vf$  is the contribution of FRPs (11.3), ACI 440).

-----  
 $\lambda = 1$  (normal-weight concrete)

$fc' = 33.00$ , but  $fc^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$Mu = 5.4715235E-012$

$Vu = 6.8378665E-048$

$d = 0.8 * D = 320.00$

$Nu = 4771.233$

$Ag = 125663.706$

From (11.5.4.8), ACI 318-14:  $Vs = 219326.297$

$Av = \sqrt{2} * A_{stirup} = 123370.055$

$fy = 555.56$

$s = 100.00$

$Vs$  is multiplied by  $Col = 0.00$

$s/d = 0.3125$

$Vf$  ((11-3)-(11.4), ACI 440) =  $194961.134$

$f = 0.95$ , for fully-wrapped sections

$wf/sf = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression,  
where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $Vf(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ ,  
as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = b1 + 90^\circ = 90.00$

$Vf = \text{Min}(|Vf(45, \theta)|, |Vf(-45, \theta)|)$ , with:

total thickness per orientation,  $tf1 = NL * t / NoDir = 1.016$

$dfv = d$  (figure 11.2, ACI 440) =  $370.00$

$ffe$  ((11-5), ACI 440) =  $259.312$

$Ef = 64828.00$

$fe = 0.004$ , from (11.6a), ACI 440

with  $fu = 0.01$

From (11-11), ACI 440:  $Vs + Vf \leq 306911.784$

$bw * d = \sqrt{2} * d * d / 4 = 80424.772$

-----  
Calculation of Shear Strength at edge 2,  $Vr2 = 451727.786$

$Vr2 = VCol$  ((10.3), ASCE 41-17) =  $knl * VColO$

$VColO = 451727.786$

$knl = 1$  (zero step-static loading)

-----  
NOTE: In expression (10-3) ' $Vs = Av * fy * d / s$ ' is replaced by ' $Vs + f * Vf$ '  
where  $Vf$  is the contribution of FRPs (11.3), ACI 440).

-----  
 $\lambda = 1$  (normal-weight concrete)

$fc' = 33.00$ , but  $fc^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$Mu = 5.4715235E-012$

$Vu = 6.8378665E-048$

$d = 0.8 * D = 320.00$

$Nu = 4771.233$

$Ag = 125663.706$

From (11.5.4.8), ACI 318-14:  $Vs = 219326.297$

$Av = \sqrt{2} * A_{stirup} = 123370.055$

$fy = 555.56$

$s = 100.00$

$Vs$  is multiplied by  $Col = 0.00$

$s/d = 0.3125$

$Vf$  ((11-3)-(11.4), ACI 440) =  $194961.134$

$f = 0.95$ , for fully-wrapped sections

$wf/sf = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \theta$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE). This later relation, considered as a function  $V_f(\theta, \alpha_i)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta_1 = \theta_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, \theta_1)|)$ , with:  
total thickness per orientation,  $t_{f1} = NL * t / \text{NoDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}$  ((11-5), ACI 440) = 259.312  
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
with  $f_u = 0.01$   
From (11-11), ACI 440:  $V_s + V_f \leq 306911.784$   
 $b_w * d = \sqrt{V_s * d} / 4 = 80424.772$

-----  
-----  
End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At local axis: 2

-----  
-----  
Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1  
At local axis: 3  
Integration Section: (a)  
Section Type: rccs

#### Constant Properties

-----  
Knowledge Factor,  $\gamma = 0.90$   
Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE41-17.  
Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17  
Consequently:  
New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$   
New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 555.56$   
Concrete Elasticity,  $E_c = 26999.444$   
Steel Elasticity,  $E_s = 200000.00$   
Diameter,  $D = 400.00$   
Cover Thickness,  $c = 25.00$   
Element Length,  $L = 3000.00$   
Secondary Member  
Ribbed Bars  
Ductile Steel  
Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Adequate Lap Length ( $l_b / d \geq 1$ )  
FRP Wrapping Data  
Type: Carbon  
Cured laminate properties (design values)  
Thickness,  $t = 1.016$   
Tensile Strength,  $f_{fu} = 1055.00$   
Tensile Modulus,  $E_f = 64828.00$   
Elongation,  $e_{fu} = 0.01$   
Number of directions,  $\text{NoDir} = 1$   
Fiber orientations,  $\theta_i = 0.00^\circ$   
Number of layers,  $NL = 1$   
Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

-----  
-----  
Bending Moment,  $M = -9.0263E+006$   
Shear Force,  $V_2 = -3007.62$   
Shear Force,  $V_3 = -1.5929820E-013$   
Axial Force,  $F = -4770.03$   
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $A_{st} = 1272.345$

-Compression:  $As_c = 1781.283$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $As_{ten} = 1017.876$   
-Compression:  $As_{com} = 1017.876$   
-Middle:  $As_{mid} = 1017.876$   
Mean Diameter of Tension Reinforcement,  $Db_L = 18.00$

New component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_R = 1.0^*$   $u = 0.10133974$   
 $u = y + p = 0.10133974$

- Calculation of  $y$  -

$y = (M_y * L_s / 3) / E_{eff} = 0.02979765$  ((4.29), Biskinis Phd)  
 $M_y = 3.0318E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 3001.151  
From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 1.0179E+013$   
factor = 0.30  
 $A_g = 125663.706$   
 $f_c' = 33.00$   
 $N = 4770.03$   
 $E_c * I_g = 3.3929E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 3.0318E+008$   
 $y$  ((10a) or (10b)) =  $1.6222681E-005$   
 $M_{y\_ten}$  (8a) =  $3.0318E+008$   
 $y_{ten}$  (7a) = 69.35398  
error of function (7a) = 0.00743251  
 $M_{y\_com}$  (8b) =  $4.6746E+008$   
 $y_{com}$  (7b) = 67.1564  
error of function (7b) = -0.0026554  
with  $e_y = 0.0027778$   
 $e_{co} = 0.002$   
 $a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00102233$   
 $N = 4770.03$   
 $A_c = 125663.706$   
= 0.36359274  
with  $f_c^*$  ((12.3), ACI 440) = 37.12975  
 $f_c = 33.00$   
 $f_l = 1.3173$   
 $k = 1$   
Effective FRP thickness,  $t_f = NL * t * \text{Cos}(b_1) = 1.016$   
 $e_{fe}$  ((12.5) and (12.7)) = 0.004  
 $E_f = 64828.00$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

- Calculation of  $p$  -

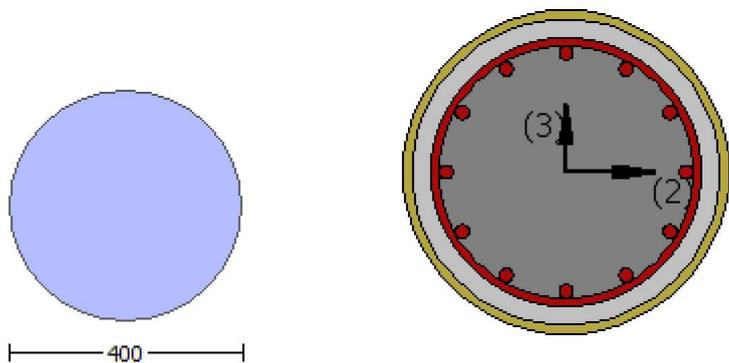
From table 10-9:  $p = 0.07154209$   
with:

- Columns controlled by inadequate development or splicing along the clear height because  $l_b/d < 1$   
 shear control ratio  $V_y E / V_{CoI} E = 0.44725617$   
 $d = 0.00$   
 $s = 0.00$   
 $t = 2 \cdot A_v / (d_c \cdot s) + 4 \cdot t_f / D \cdot (f_{fe} / f_s) = 0.00$   
 $A_v = 78.53982$ , is the area of the circular stirrup  
 $d_c = D - 2 \cdot \text{cover} - \text{Hoop Diameter} = 340.00$   
 The term  $2 \cdot t_f / b_w \cdot (f_{fe} / f_s)$  is implemented to account for FRP contribution  
 where  $f = 2 \cdot t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength  
 All these variables have already been given in Shear control ratio calculation.  
 $NUD = 4770.03$   
 $A_g = 125663.706$   
 $f_{cE} = 33.00$   
 $f_{ytE} = f_{ylE} = 555.56$   
 $\rho_l = \text{Area\_Tot\_Long\_Rein} / (A_g) = 0.0243$   
 $f_{cE} = 33.00$

-----  
 End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1  
 At local axis: 3  
 Integration Section: (a)

**Calculation No. 13**

column C1, Floor 1  
 Limit State: Collapse Prevention (data interpolation between analysis steps 1 and 2)  
 Analysis: Uniform +X  
 Check: Shear capacity  $V_{Rd}$   
 Edge: End  
 Local Axis: (2)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1  
 At local axis: 2  
 Integration Section: (b)  
 Section Type: rccs  
 Constant Properties

-----

Knowledge Factor,  $\phi = 0.90$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

New material of Secondary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 25.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 500.00$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of  $\gamma$  for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE41-17).

New material: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material: Steel Strength,  $f_s = f_{sm} = 555.56$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Secondary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou,min} = l_b/l_d \geq 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i: 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

-----

#### Stepwise Properties

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EDGE -A-

Bending Moment,  $M_a = -9.0263E+006$

Shear Force,  $V_a = -3007.62$

EDGE -B-

Bending Moment,  $M_b = 0.0481749$

Shear Force,  $V_b = 3007.62$

BOTH EDGES

Axial Force,  $F = -4770.03$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{sl,t} = 0.00$

-Compression:  $A_{sl,c} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{sl,ten} = 1017.876$

-Compression:  $A_{sl,com} = 1017.876$

-Middle:  $A_{sl,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $D_{bL,ten} = 18.00$

-----

New component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = 1.0 \cdot V_n = 393301.141$

$V_n ((10.3), ASCE 41-17) = k_n \cdot V_{CoI} = 393301.141$

$V_{CoI} = 393301.141$

$k_n = 1.00$

$displacement\_ductility\_demand = 0.04465363$

-----

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ '

where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

$f_c' = 25.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 0.0481749$

$V_u = 3007.62$

$d = 0.8 \cdot D = 320.00$

$N_u = 4770.03$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 197392.088$

$A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 123370.055$

$f_y = 500.00$

$s = 100.00$

$V_s$  is multiplied by  $\text{Col} = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression, where  $a$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta)$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = 45^\circ$

$V_f = \text{Min}(|V_f(45^\circ)|, |V_f(-45^\circ)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{\text{Dir}} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 267132.42$

$b_w \cdot d = \sqrt{2} \cdot d^2 / 4 = 80424.772$

displacement ductility demand is calculated as  $\delta / y$

- Calculation of  $\delta / y$  for END B -

for rotation axis 3 and integ. section (b)

From analysis, chord rotation  $\theta = 0.00013301$

$y = (M_y \cdot L_s / 3) / E_{\text{eff}} = 0.00297862$  ((4.29), Biskinis Phd))

$M_y = 3.0318 \text{E} + 008$

$L_s = M/V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) = 300.00

From table 10.5, ASCE 41\_17:  $E_{\text{eff}} = \text{factor} \cdot E_c \cdot I_g = 1.0179 \text{E} + 013$

factor = 0.30

$A_g = 125663.706$

$f_c' = 33.00$

$N = 4770.03$

$E_c \cdot I_g = 3.3929 \text{E} + 013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\delta / y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y_{\text{ten}}}, M_{y_{\text{com}}}) = 3.0318 \text{E} + 008$

$y$  ((10a) or (10b)) = 1.6222681E-005

$M_{y_{\text{ten}}} (8a) = 3.0318 \text{E} + 008$

$y_{\text{ten}} (7a) = 69.35398$

error of function (7a) = 0.00743251

$M_{y_{\text{com}}} (8b) = 4.6746 \text{E} + 008$

$y_{\text{com}} (7b) = 67.1564$

error of function (7b) = -0.0026554

with  $e_y = 0.0027778$

$e_{co} = 0.002$   
 $a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00102233$   
 $N = 4770.03$   
 $A_c = 125663.706$   
 $= 0.36359274$   
 with  $f_c^*$  ((12.3), ACI 440) = 37.12975  
 $f_c = 33.00$   
 $f_l = 1.3173$   
 $k = 1$   
 Effective FRP thickness,  $t_f = NL \cdot t \cdot \cos(b_1) = 1.016$   
 $e_{fe}$  ((12.5) and (12.7)) = 0.004  
 $E_f = 64828.00$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

End Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (b)

## Calculation No. 14

column C1, Floor 1

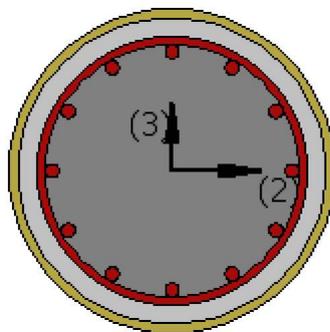
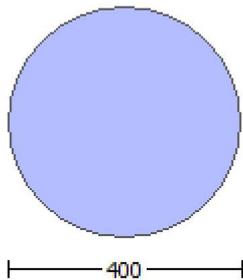
Limit State: Collapse Prevention (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity ( $\theta$ )

Edge: End

Local Axis: (2)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 0.90$

Mean strength values are used for both shear and moment calculations.

Consequently:

New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

New material: Steel Strength,  $f_s = 1.25 * f_{sm} = 694.45$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.56406

Element Length,  $L = 3000.00$

Secondary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou, \min} >= 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

Stepwise Properties

At local axis: 3

EDGE -A-

Shear Force,  $V_a = 1.1167452E-031$

EDGE -B-

Shear Force,  $V_b = -1.1167452E-031$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{sl,t} = 0.00$

-Compression:  $A_{sl,c} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{sl,ten} = 1017.876$

-Compression:  $A_{sl,com} = 1017.876$

-Middle:  $A_{sl,mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.44725617$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 202038.041$

with

$M_{pr1} = \text{Max}(M_{u1+}, M_{u1-}) = 3.0306E+008$

$M_{u1+} = 3.0306E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$M_{u1-} = 3.0306E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment

direction which is defined for the static loading combination

$$M_{pr2} = \text{Max}(M_{u2+}, M_{u2-}) = 3.0306E+008$$

$M_{u2+} = 3.0306E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

$M_{u2-} = 3.0306E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

-----  
Calculation of  $M_{u1+}$   
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$   
 $M_u = 3.0306E+008$   
-----

$$= 1.02974$$

$$\phi = 0.91217079$$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 51.61391$

conf. factor  $c = 1.56406$

$$f_c = 33.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$$l_b/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00101663$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \phi \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$$

-----  
Calculation of ratio  $l_b/d$   
-----

Adequate Lap Length:  $l_b/d \geq 1$   
-----  
-----  
-----

-----  
Calculation of  $M_{u1-}$   
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$   
 $M_u = 3.0306E+008$   
-----

$$= 1.02974$$

$$\phi = 0.91217079$$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 51.61391$

conf. factor  $c = 1.56406$

$$f_c = 33.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$$l_b/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00101663$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \phi \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$$

-----  
Calculation of ratio  $l_b/d$   
-----

Adequate Lap Length:  $l_b/d \geq 1$   
-----  
-----  
-----

-----  
Calculation of  $M_{u2+}$   
-----

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 3.0306E+008

= 1.02974  
' = 0.91217079  
error of function (3.68), Biskinis Phd = 60360.02  
From 5A.2, TBDY: fcc = fc\* c = 51.61391  
conf. factor c = 1.56406  
fc = 33.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 694.45  
lb/d = 1.00  
d1 = 44.00  
R = 200.00  
v = 0.00101663  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.45184585

Calculation of ratio lb/d

Adequate Lap Length: lb/d >= 1

Calculation of Mu2-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 3.0306E+008

= 1.02974  
' = 0.91217079  
error of function (3.68), Biskinis Phd = 60360.02  
From 5A.2, TBDY: fcc = fc\* c = 51.61391  
conf. factor c = 1.56406  
fc = 33.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 694.45  
lb/d = 1.00  
d1 = 44.00  
R = 200.00  
v = 0.00101663  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.45184585

Calculation of ratio lb/d

Adequate Lap Length: lb/d >= 1

Calculation of Shear Strength Vr = Min(Vr1,Vr2) = 451727.786

Calculation of Shear Strength at edge 1, Vr1 = 451727.786  
Vr1 = VCol ((10.3), ASCE 41-17) = knl\*VCoLO  
VCoLO = 451727.786  
knl = 1 (zero step-static loading)

NOTE: In expression (10-3) 'Vs = Av\*fy\*d/s' is replaced by 'Vs+ f\*Vf  
where Vf is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)  
 $f_c' = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $\mu_u = 1.5353150E-011$   
 $\mu_v = 1.1167452E-031$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 219326.297$   
 $A_v = \frac{1}{2} \cdot A_{\text{stirrup}} = 123370.055$   
 $f_y = 555.56$   
 $s = 100.00$   
 $V_s$  is multiplied by  $\text{Col} = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot_a) \sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = 45^\circ$   
 $V_f = \text{Min}(|V_f(45, a)|, |V_f(-45, a)|)$ , with:  
 total thickness per orientation,  $t_{f1} = NL \cdot t / N_{\text{Dir}} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}$  ((11-5), ACI 440) = 259.312  
 $E_f = 64828.00$   
 $f_{e} = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 306911.784$   
 $b_w \cdot d = \frac{1}{4} \cdot d^2 = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 451727.786$   
 $V_{r2} = V_{\text{Col}}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{\text{Col}0}$   
 $V_{\text{Col}0} = 451727.786$   
 $k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ '  
 where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)  
 $f_c' = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $\mu_u = 1.5353150E-011$   
 $\mu_v = 1.1167452E-031$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 219326.297$   
 $A_v = \frac{1}{2} \cdot A_{\text{stirrup}} = 123370.055$   
 $f_y = 555.56$   
 $s = 100.00$   
 $V_s$  is multiplied by  $\text{Col} = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot_a) \sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = 45^\circ$   
 $V_f = \text{Min}(|V_f(45, a)|, |V_f(-45, a)|)$ , with:  
 total thickness per orientation,  $t_{f1} = NL \cdot t / N_{\text{Dir}} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

ffe ((11-5), ACI 440) = 259.312  
Ef = 64828.00  
fe = 0.004, from (11.6a), ACI 440  
with fu = 0.01  
From (11-11), ACI 440: Vs + Vf <= 306911.784  
bw\*d = \*d\*d/4 = 80424.772

-----  
-----  
End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At local axis: 3

-----  
-----  
Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At Shear local axis: 2  
(Bending local axis: 3)  
Section Type: rccs

Constant Properties

-----  
Knowledge Factor, = 0.90  
Mean strength values are used for both shear and moment calculations.  
Consequently:  
New material of Secondary Member: Concrete Strength, fc = fcm = 33.00  
New material of Secondary Member: Steel Strength, fs = fsm = 555.56  
Concrete Elasticity, Ec = 26999.444  
Steel Elasticity, Es = 200000.00  
#####  
Note: Especially for the calculation of moment strengths,  
the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14  
New material: Steel Strength, fs = 1.25\*fsm = 694.45  
#####  
Diameter, D = 400.00  
Cover Thickness, c = 25.00  
Mean Confinement Factor overall section = 1.56406  
Element Length, L = 3000.00  
Secondary Member  
Ribbed Bars  
Ductile Steel  
Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Adequate Lap Length (lo/lou,min>=1)  
FRP Wrapping Data  
Type: Carbon  
Cured laminate properties (design values)  
Thickness, t = 1.016  
Tensile Strength, ffu = 1055.00  
Tensile Modulus, Ef = 64828.00  
Elongation, efu = 0.01  
Number of directions, NoDir = 1  
Fiber orientations, bi: 0.00°  
Number of layers, NL = 1  
Radius of rounding corners, R = 40.00

-----  
Stepwise Properties

-----  
At local axis: 2  
EDGE -A-  
Shear Force, Va = -6.8378665E-048  
EDGE -B-  
Shear Force, Vb = 6.8378665E-048  
BOTH EDGES  
Axial Force, F = -4771.233  
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension: Aslt = 0.00

-Compression:  $A_{sc} = 3053.628$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $A_{s,ten} = 1017.876$   
-Compression:  $A_{s,com} = 1017.876$   
-Middle:  $A_{s,mid} = 1017.876$

-----  
-----  
Calculation of Shear Capacity ratio,  $V_e/V_r = 0.44725617$   
Member Controlled by Flexure ( $V_e/V_r < 1$ )  
Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 202038.041$   
with

$M_{pr1} = \text{Max}(M_{u1+}, M_{u1-}) = 3.0306E+008$   
 $M_{u1+} = 3.0306E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination  
 $M_{u1-} = 3.0306E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination  
 $M_{pr2} = \text{Max}(M_{u2+}, M_{u2-}) = 3.0306E+008$   
 $M_{u2+} = 3.0306E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination  
 $M_{u2-} = 3.0306E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

-----  
Calculation of  $M_{u1+}$   
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$   
 $M_u = 3.0306E+008$

-----  
 $\lambda = 1.02974$   
 $\lambda' = 0.91217079$   
error of function (3.68), Biskinis Phd = 60360.02  
From 5A.2, TDY:  $f_{cc} = f_c^* \quad c = 51.61391$   
conf. factor  $c = 1.56406$   
 $f_c = 33.00$   
From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y * \text{Min}(1, 1.25 * (l_b/d)^{2/3}) = 694.45$   
 $l_b/d = 1.00$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00101663$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= * \text{Min}(1, 1.25 * (l_b/d)^{2/3}) = 0.45184585$

-----  
Calculation of ratio  $l_b/d$   
-----

Adequate Lap Length:  $l_b/d \geq 1$   
-----  
-----

-----  
Calculation of  $M_{u1-}$   
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$   
 $M_u = 3.0306E+008$

-----  
 $\lambda = 1.02974$   
 $\lambda' = 0.91217079$   
error of function (3.68), Biskinis Phd = 60360.02  
From 5A.2, TDY:  $f_{cc} = f_c^* \quad c = 51.61391$   
conf. factor  $c = 1.56406$   
 $f_c = 33.00$

From 10.3.5, ASCE41-17, Final value of fy:  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$$l_b/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00101663$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of  $\mu_{2+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$$\mu = 3.0306E+008$$

$$= 1.02974$$

$$' = 0.91217079$$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 51.61391$

$$\text{conf. factor } c = 1.56406$$

$$f_c = 33.00$$

From 10.3.5, ASCE41-17, Final value of fy:  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$$l_b/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00101663$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of  $\mu_{2-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$$\mu = 3.0306E+008$$

$$= 1.02974$$

$$' = 0.91217079$$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 51.61391$

$$\text{conf. factor } c = 1.56406$$

$$f_c = 33.00$$

From 10.3.5, ASCE41-17, Final value of fy:  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$$l_b/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00101663$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 451727.786$

Calculation of Shear Strength at edge 1,  $V_{r1} = 451727.786$

$V_{r1} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_n l V_{Col0}$

$V_{Col0} = 451727.786$

$k_n l = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v f_y d/s$ ' is replaced by ' $V_s + f^* V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\rho = 1$  (normal-weight concrete)

$f_c' = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 5.4715235E-012$

$\nu_u = 6.8378665E-048$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 219326.297$

$A_v = \rho A_{stirrup} = 123370.055$

$f_y = 555.56$

$s = 100.00$

$V_s$  is multiplied by  $\phi_{Col} = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) =  $194961.134$

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = b1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \theta)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{Dir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) =  $370.00$

$f_{fe}$  ((11-5), ACI 440) =  $259.312$

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 306911.784$

$b_w \cdot d = \rho \cdot d^2 / 4 = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 451727.786$

$V_{r2} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_n l V_{Col0}$

$V_{Col0} = 451727.786$

$k_n l = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v f_y d/s$ ' is replaced by ' $V_s + f^* V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\rho = 1$  (normal-weight concrete)

$f_c' = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 5.4715235E-012$

$\nu_u = 6.8378665E-048$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 219326.297$

$A_v = \sqrt{2} \cdot A_{stirrup} = 123370.055$

$f_y = 555.56$

$s = 100.00$

$V_s$  is multiplied by  $Col = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression, where  $a$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta)$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta_1 = b_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, a_1)|)$ , with:

total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 306911.784$

$b_w \cdot d = \sqrt{4} \cdot V_s = 80424.772$

-----  
End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At local axis: 2

-----  
Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (b)

Section Type: rccs

Constant Properties

-----  
Knowledge Factor,  $\lambda = 0.90$

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Secondary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_b/d \geq 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $\text{NoDir} = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

## Stepwise Properties

Bending Moment,  $M = -3.6626908E-011$

Shear Force,  $V2 = 3007.62$

Shear Force,  $V3 = 1.5929820E-013$

Axial Force,  $F = -4770.03$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{st} = 0.00$

-Compression:  $A_{sc} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{st,ten} = 1017.876$

-Compression:  $A_{sc,com} = 1017.876$

-Middle:  $A_{st,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $DbL = 18.00$

New component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u,R = 1.0^*$   $u = 0.0864352$

$u = y + p = 0.0864352$

- Calculation of  $y$  -

$y = (My * L_s / 3) / E_{eff} = 0.01489311$  ((4.29), Biskinis Phd))

$My = 3.0318E+008$

$L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 1500.00

From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 1.0179E+013$

factor = 0.30

$A_g = 125663.706$

$f_c' = 33.00$

$N = 4770.03$

$E_c * I_g = 3.3929E+013$

Calculation of Yielding Moment  $My$

Calculation of  $y$  and  $My$  according to (7) - (8) in Biskinis and Fardis

$My = \text{Min}(My_{ten}, My_{com}) = 3.0318E+008$

$y$  ((10a) or (10b)) = 1.6222681E-005

$My_{ten}$  (8a) = 3.0318E+008

$y_{ten}$  (7a) = 69.35398

error of function (7a) = 0.00743251

$My_{com}$  (8b) = 4.6746E+008

$y_{com}$  (7b) = 67.1564

error of function (7b) = -0.0026554

with  $e_y = 0.0027778$

$e_{co} = 0.002$

$a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)

$d1 = 44.00$

$R = 200.00$

$v = 0.00102233$

$N = 4770.03$

$A_c = 125663.706$

= 0.36359274

with  $f_c^*$  ((12.3), ACI 440) = 37.12975

$f_c = 33.00$

$f_l = 1.3173$

$k = 1$

Effective FRP thickness,  $t_f = NL * t * \text{Cos}(b1) = 1.016$

$e_{fe}$  ((12.5) and (12.7)) = 0.004

$E_f = 64828.00$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

- Calculation of  $p$  -

From table 10-9:  $p = 0.07154209$

with:

- Columns controlled by inadequate development or splicing along the clear height because  $l_b/d < 1$

shear control ratio  $V_y E / V_{CoI} E = 0.44725617$

$d = 0.00$

$s = 0.00$

$t = 2 \cdot A_v / (d_c \cdot s) + 4 \cdot t_f / D \cdot (f_{fe} / f_s) = 0.00$

$A_v = 78.53982$ , is the area of the circular stirrup

$d_c = D - 2 \cdot \text{cover} - \text{Hoop Diameter} = 340.00$

The term  $2 \cdot t_f / b_w \cdot (f_{fe} / f_s)$  is implemented to account for FRP contribution

where  $f = 2 \cdot t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength

All these variables have already been given in Shear control ratio calculation.

$N_{UD} = 4770.03$

$A_g = 125663.706$

$f_{cE} = 33.00$

$f_{ytE} = f_{ylE} = 555.56$

$p_l = \text{Area\_Tot\_Long\_Rein} / (A_g) = 0.0243$

$f_{cE} = 33.00$

End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (b)

## Calculation No. 15

column C1, Floor 1

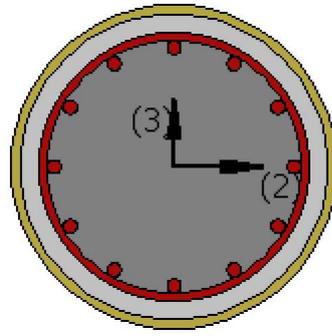
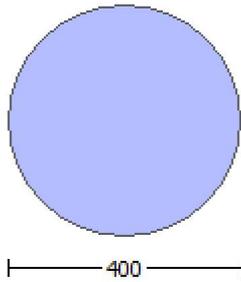
Limit State: Collapse Prevention (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Shear capacity  $V_{Rd}$

Edge: End

Local Axis: (3)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (b)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 0.90$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

New material of Secondary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 25.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 500.00$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of  $\gamma$  for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE41-17).

New material: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material: Steel Strength,  $f_s = f_{sm} = 555.56$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Secondary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou, \min} = l_b/l_d \geq 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $NoDir = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

Stepwise Properties

EDGE -A-

Bending Moment,  $M_a = 5.1473654E-010$

Shear Force,  $V_a = -1.5929820E-013$

EDGE -B-

Bending Moment,  $M_b = -3.6626908E-011$

Shear Force,  $V_b = 1.5929820E-013$

BOTH EDGES

Axial Force,  $F = -4770.03$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_{t1} = 0.00$

-Compression:  $As_{c1} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t,ten} = 1017.876$

-Compression:  $As_{l,com} = 1017.876$

-Middle:  $As_{l,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $Db_{L,ten} = 18.00$

New component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = 1.0 \cdot V_n = 393301.141$

$V_n$  ((10.3), ASCE 41-17) =  $k_n \cdot V_{Co10} = 393301.141$

$V_{Co10} = 393301.141$

$k_n = 1.00$

$displacement\_ductility\_demand = 0.00$

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs ((11.3), ACI 440).

= 1 (normal-weight concrete)

$f_c' = 25.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M / Vd = 2.00$

$\mu_u = 3.6626908E-011$

$\nu_u = 1.5929820E-013$

$d = 0.8 \cdot D = 320.00$

$N_u = 4770.03$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 197392.088$

$A_v = \sqrt{2} \cdot A_{stirrup} = 123370.055$

$f_y = 500.00$

$s = 100.00$

$V_s$  is multiplied by  $\phi_{col} = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) =  $194961.134$

$f = 0.95$ , for fully-wrapped sections

$w_f / s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = 45^\circ$  and  $\alpha = 90^\circ$  =  $90.00$

$V_f = \text{Min}(|V_f(45^\circ, 90^\circ)|, |V_f(-45^\circ, 90^\circ)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{oDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) =  $370.00$

$f_{fe}$  ((11-5), ACI 440) =  $259.312$

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 267132.42$

$b_w \cdot d = \sqrt{2} \cdot d^2 / 4 = 80424.772$

$displacement\_ductility\_demand$  is calculated as  $\phi / y$

- Calculation of  $\phi / y$  for END B -

for rotation axis 2 and integ. section (b)

From analysis, chord rotation =  $7.9508914E-021$

$y = (M_y \cdot L_s / 3) / E_{eff} = 0.01489311$  ((4.29), Biskinis Phd)

$M_y = 3.0318E+008$

$L_s = M / V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) =  $1500.00$

From table 10.5, ASCE 41\_17:  $E_{eff} = \text{factor} \cdot E_c \cdot I_g = 1.0179E+013$

factor = 0.30  
Ag = 125663.706  
fc' = 33.00  
N = 4770.03  
Ec\*Ig = 3.3929E+013

-----  
-----  
Calculation of Yielding Moment My

-----  
Calculation of  $\rho_y$  and My according to (7) - (8) in Biskinis and Fardis

-----  
My = Min(My\_ten, My\_com) = 3.0318E+008  
 $\rho_y$  ((10a) or (10b)) = 1.6222681E-005  
My\_ten (8a) = 3.0318E+008  
 $\rho_{y\_ten}$  (7a) = 69.35398  
error of function (7a) = 0.00743251  
My\_com (8b) = 4.6746E+008  
 $\rho_{y\_com}$  (7b) = 67.1564  
error of function (7b) = -0.0026554  
with  $e_y$  = 0.0027778  
eco = 0.002  
apl = 0.45 ((9c) in Biskinis and Fardis for FRP Wrap)  
d1 = 44.00  
R = 200.00  
v = 0.00102233  
N = 4770.03  
Ac = 125663.706  
= 0.36359274  
with  $f_c^*$  ((12.3), ACI 440) = 37.12975  
fc = 33.00  
fl = 1.3173  
k = 1  
Effective FRP thickness,  $t_f$  = NL\*t\*cos(b1) = 1.016  
efe ((12.5) and (12.7)) = 0.004  
Ef = 64828.00

-----  
Calculation of ratio lb/d

-----  
Adequate Lap Length: lb/d >= 1

-----  
End Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (b)

-----  
**Calculation No. 16**

column C1, Floor 1

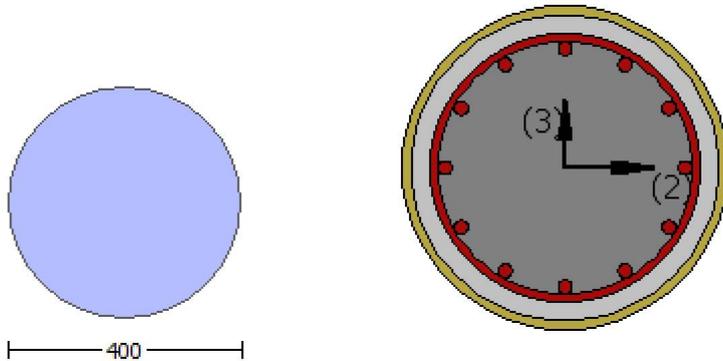
Limit State: Collapse Prevention (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity ( $\theta_u$ )

Edge: End

Local Axis: (3)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 0.90$

Mean strength values are used for both shear and moment calculations.

Consequently:

New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

New material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 694.45$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.56406

Element Length,  $L = 3000.00$

Secondary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou, \min} > 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i: 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

Stepwise Properties

At local axis: 3

EDGE -A-

Shear Force,  $V_a = 1.1167452E-031$

EDGE -B-

Shear Force,  $V_b = -1.1167452E-031$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 0.00$

-Compression:  $As_c = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t,ten} = 1017.876$

-Compression:  $As_{c,com} = 1017.876$

-Middle:  $As_{mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.44725617$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 202038.041$

with

$M_{pr1} = \text{Max}(Mu_{1+}, Mu_{1-}) = 3.0306E+008$

$Mu_{1+} = 3.0306E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$Mu_{1-} = 3.0306E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(Mu_{2+}, Mu_{2-}) = 3.0306E+008$

$Mu_{2+} = 3.0306E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

$Mu_{2-} = 3.0306E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

Calculation of  $Mu_{1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$

$M_u = 3.0306E+008$

$\phi = 1.02974$

$\lambda = 0.91217079$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c \cdot \lambda = 51.61391$

conf. factor  $\lambda = 1.56406$

$f_c = 33.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00101663$

$N = 4771.233$

$A_c = 125663.706$

$\phi \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of Mu1-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 3.0306E+008

= 1.02974  
' = 0.91217079  
error of function (3.68), Biskinis Phd = 60360.02  
From 5A.2, TBDY: fcc = fc\* c = 51.61391  
conf. factor c = 1.56406  
fc = 33.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 694.45  
lb/d = 1.00  
d1 = 44.00  
R = 200.00  
v = 0.00101663  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.45184585

Calculation of ratio lb/d

Adequate Lap Length: lb/d >= 1

Calculation of Mu2+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 3.0306E+008

= 1.02974  
' = 0.91217079  
error of function (3.68), Biskinis Phd = 60360.02  
From 5A.2, TBDY: fcc = fc\* c = 51.61391  
conf. factor c = 1.56406  
fc = 33.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 694.45  
lb/d = 1.00  
d1 = 44.00  
R = 200.00  
v = 0.00101663  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.45184585

Calculation of ratio lb/d

Adequate Lap Length: lb/d >= 1

Calculation of Mu2-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 3.0306E+008

= 1.02974

$\rho = 0.91217079$   
 error of function (3.68), Biskinis Phd = 60360.02  
 From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 51.61391$   
 conf. factor  $c = 1.56406$   
 $f_c = 33.00$   
 From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 694.45$   
 $l_b/l_d = 1.00$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00101663$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.45184585$

Calculation of ratio  $l_b/l_d$

Adequate Lap Length:  $l_b/l_d \geq 1$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 451727.786$

Calculation of Shear Strength at edge 1,  $V_{r1} = 451727.786$

$V_{r1} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{Col0}$

$V_{Col0} = 451727.786$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

$f_c' = 33.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.5353150E-011$

$\nu_u = 1.1167452E-031$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 219326.297$

$A_v = \rho_s \cdot A_{stirup} = 123370.055$

$f_y = 555.56$

$s = 100.00$

$V_s$  is multiplied by  $\rho_{col} = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression,

where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = 45^\circ + 90^\circ = 135^\circ$

$V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, 1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{oDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_{e} = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 306911.784$

$b_w \cdot d = \rho_s \cdot d^2 / 4 = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 451727.786$

$V_{r2} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{Col0}$

VCoIO = 451727.786  
knl = 1 (zero step-static loading)

NOTE: In expression (10-3) 'Vs = Av\*fy\*d/s' is replaced by 'Vs+ f\*VF'  
where Vf is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)  
fc' = 33.00, but  $fc^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
M/Vd = 2.00  
Mu = 1.5353150E-011  
Vu = 1.1167452E-031  
d = 0.8\*D = 320.00  
Nu = 4771.233  
Ag = 125663.706  
From (11.5.4.8), ACI 318-14: Vs = 219326.297  
Av = /2\*A\_stirrup = 123370.055  
fy = 555.56  
s = 100.00  
Vs is multiplied by Col = 0.00  
s/d = 0.3125  
Vf ((11-3)-(11.4), ACI 440) = 194961.134  
f = 0.95, for fully-wrapped sections  
wf/sf = 1 (FRP strips adjacent to one another).  
In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \cot \alpha) \sin \alpha$  which is more a generalised expression,  
where  $\theta$  is the angle of the crack direction (see KANEPE).  
This later relation, considered as a function Vf(  $\theta, \alpha$  ), is implemented for every different fiber orientation ai,  
as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
orientation 1:  $\theta = b1 + 90^\circ = 90.00$   
Vf = Min(|Vf(45,  $\theta$ )|, |Vf(-45,  $\theta$ )|), with:  
total thickness per orientation, tf1 = NL\*t/NoDir = 1.016  
dfv = d (figure 11.2, ACI 440) = 370.00  
ffe ((11-5), ACI 440) = 259.312  
Ef = 64828.00  
fe = 0.004, from (11.6a), ACI 440  
with fu = 0.01  
From (11-11), ACI 440: Vs + Vf <= 306911.784  
bw\*d = \*d\*d/4 = 80424.772

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At local axis: 3

Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At Shear local axis: 2  
(Bending local axis: 3)  
Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 0.90$   
Mean strength values are used for both shear and moment calculations.  
Consequently:  
New material of Secondary Member: Concrete Strength, fc = fcm = 33.00  
New material of Secondary Member: Steel Strength, fs = fsm = 555.56  
Concrete Elasticity, Ec = 26999.444  
Steel Elasticity, Es = 200000.00  
#####  
Note: Especially for the calculation of moment strengths,  
the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14  
New material: Steel Strength, fs = 1.25\*fsm = 694.45  
#####  
Diameter, D = 400.00  
Cover Thickness, c = 25.00  
Mean Confinement Factor overall section = 1.56406

Element Length, L = 3000.00  
Secondary Member  
Ribbed Bars  
Ductile Steel  
Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Adequate Lap Length ( $l_o/l_{ou}, \min \geq 1$ )  
FRP Wrapping Data  
Type: Carbon  
Cured laminate properties (design values)  
Thickness, t = 1.016  
Tensile Strength,  $f_{fu}$  = 1055.00  
Tensile Modulus,  $E_f$  = 64828.00  
Elongation,  $e_{fu}$  = 0.01  
Number of directions, NoDir = 1  
Fiber orientations,  $b_i$ : 0.00°  
Number of layers, NL = 1  
Radius of rounding corners, R = 40.00

#### Stepwise Properties

At local axis: 2  
EDGE -A-  
Shear Force,  $V_a$  = -6.8378665E-048  
EDGE -B-  
Shear Force,  $V_b$  = 6.8378665E-048  
BOTH EDGES  
Axial Force, F = -4771.233  
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $A_{st}$  = 0.00  
-Compression:  $A_{sc}$  = 3053.628  
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $A_{st,ten}$  = 1017.876  
-Compression:  $A_{sl,com}$  = 1017.876  
-Middle:  $A_{sl,mid}$  = 1017.876

Calculation of Shear Capacity ratio,  $V_e/V_r$  = 0.44725617  
Member Controlled by Flexure ( $V_e/V_r < 1$ )  
Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 202038.041$   
with  
 $M_{pr1} = \text{Max}(\mu_{1+}, \mu_{1-}) = 3.0306E+008$   
 $\mu_{1+} = 3.0306E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction  
which is defined for the static loading combination  
 $\mu_{1-} = 3.0306E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment  
direction which is defined for the static loading combination  
 $M_{pr2} = \text{Max}(\mu_{2+}, \mu_{2-}) = 3.0306E+008$   
 $\mu_{2+} = 3.0306E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction  
which is defined for the the static loading combination  
 $\mu_{2-} = 3.0306E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment  
direction which is defined for the the static loading combination

Calculation of  $\mu_{1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$   
 $\mu_u = 3.0306E+008$

= 1.02974  
' = 0.91217079  
error of function (3.68), Biskinis Phd = 60360.02  
From 5A.2, TBDY:  $f_{cc} = f_c^* c = 51.61391$

conf. factor  $c = 1.56406$

$f_c = 33.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00101663$

$N = 4771.233$

$Ac = 125663.706$

$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of  $\mu_{1-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$\mu = 3.0306E+008$

$= 1.02974$

$' = 0.91217079$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 51.61391$

conf. factor  $c = 1.56406$

$f_c = 33.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00101663$

$N = 4771.233$

$Ac = 125663.706$

$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of  $\mu_{2+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$\mu = 3.0306E+008$

$= 1.02974$

$' = 0.91217079$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 51.61391$

conf. factor  $c = 1.56406$

$f_c = 33.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00101663$

$N = 4771.233$

$Ac = 125663.706$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of  $\mu_2$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 3.0306 \times 10^8$

$$= 1.02974$$

$$' = 0.91217079$$

error of function (3.68), Biskinis Phd = 60360.02

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 51.61391$

conf. factor  $c = 1.56406$

$f_c = 33.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 694.45$

$$l_b/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00101663$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.45184585$$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 451727.786$

Calculation of Shear Strength at edge 1,  $V_{r1} = 451727.786$

$V_{r1} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{Col0}$

$$V_{Col0} = 451727.786$$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

$f_c' = 33.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$$M/d = 2.00$$

$$\mu_u = 5.4715235 \times 10^{-12}$$

$$V_u = 6.8378665 \times 10^{-48}$$

$$d = 0.8 \cdot D = 320.00$$

$$N_u = 4771.233$$

$$A_g = 125663.706$$

From (11.5.4.8), ACI 318-14:  $V_s = 219326.297$

$$A_v = \sqrt{2} \cdot A_{stirrup} = 123370.055$$

$$f_y = 555.56$$

$$s = 100.00$$

$V_s$  is multiplied by  $Col = 0.00$

$$s/d = 0.3125$$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression,

where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta)$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = b1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, a1)|)$ , with:

total thickness per orientation,  $t_{f1} = NL * t / \text{NoDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 306911.784$

$b_w * d = \rho * d^2 / 4 = 80424.772$

-----  
Calculation of Shear Strength at edge 2,  $V_{r2} = 451727.786$

$V_{r2} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_{nl} * V_{Col0}$

$V_{Col0} = 451727.786$

$k_{nl} = 1$  (zero step-static loading)

-----  
NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

-----  
 $\rho = 1$  (normal-weight concrete)

$f_c' = 33.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M / Vd = 2.00$

$\mu_u = 5.4715235E-012$

$\nu_u = 6.8378665E-048$

$d = 0.8 * D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 219326.297$

$A_v = \rho_s * A_{stirrup} = 123370.055$

$f_y = 555.56$

$s = 100.00$

$V_s$  is multiplied by  $\rho_{Col} = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections

$w_f / s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin a_i$  which is more a generalised expression,

where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta)$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = b1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, a1)|)$ , with:

total thickness per orientation,  $t_{f1} = NL * t / \text{NoDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 306911.784$

$b_w * d = \rho * d^2 / 4 = 80424.772$

-----  
End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At local axis: 2

-----  
Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (b)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\phi = 0.90$

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

New material of Secondary Member: Concrete Strength,  $f_c = f_{cm} = 33.00$

New material of Secondary Member: Steel Strength,  $f_s = f_{sm} = 555.56$

Concrete Elasticity,  $E_c = 26999.444$

Steel Elasticity,  $E_s = 200000.00$

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Secondary Member

Ribbed Bars

Ductile Steel

Without Detailing for Earthquake Resistance (including stirrups not closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_b/l_d \geq 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $\epsilon_{fu} = 0.01$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

Stepwise Properties

Bending Moment,  $M = 0.0481749$

Shear Force,  $V_2 = 3007.62$

Shear Force,  $V_3 = 1.5929820E-013$

Axial Force,  $F = -4770.03$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{st} = 0.00$

-Compression:  $A_{sc} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{st,ten} = 1017.876$

-Compression:  $A_{sc,com} = 1017.876$

-Middle:  $A_{st,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $D_bL = 18.00$

New component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_{,R} = 1.0^* u = 0.07452071$

$u = y + p = 0.07452071$

- Calculation of  $y$  -

$y = (M_y * L_s / 3) / E_{eff} = 0.00297862$  ((4.29), Biskinis Phd)

$M_y = 3.0318E+008$

$L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 300.00

From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 1.0179E+013$

factor = 0.30

$A_g = 125663.706$

$f_c' = 33.00$

$N = 4770.03$

$E_c * I_g = 3.3929E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\rho_y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 3.0318E+008$   
 $\rho_y \text{ ((10a) or (10b))} = 1.6222681E-005$   
 $M_{y\_ten} \text{ (8a)} = 3.0318E+008$   
 $\rho_{y\_ten} \text{ (7a)} = 69.35398$   
error of function (7a) = 0.00743251  
 $M_{y\_com} \text{ (8b)} = 4.6746E+008$   
 $\rho_{y\_com} \text{ (7b)} = 67.1564$   
error of function (7b) = -0.0026554  
with  $e_y = 0.0027778$   
 $e_{co} = 0.002$   
 $a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00102233$   
 $N = 4770.03$   
 $A_c = 125663.706$   
 $= 0.36359274$   
with  $f_c^*$  ((12.3), ACI 440) = 37.12975  
 $f_c = 33.00$   
 $f_l = 1.3173$   
 $k = 1$   
Effective FRP thickness,  $t_f = N L^* t^* \text{Cos}(b_1) = 1.016$   
 $e_{fe} \text{ ((12.5) and (12.7))} = 0.004$   
 $E_f = 64828.00$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

- Calculation of  $\rho_p$  -

From table 10-9:  $\rho_p = 0.07154209$

with:

- Columns controlled by inadequate development or splicing along the clear height because  $l_b/d < 1$

shear control ratio  $V_y E / V_{CoI} E = 0.44725617$

$d = 0.00$

$s = 0.00$

$t = 2^*A_v / (d_c^*s) + 4^*t_f / D^* (f_{fe} / f_s) = 0.00$

$A_v = 78.53982$ , is the area of the circular stirrup

$d_c = D - 2^*cover$  - Hoop Diameter = 340.00

The term  $2^*t_f / b_w^* (f_{fe} / f_s)$  is implemented to account for FRP contribution

where  $f = 2^*t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength

All these variables have already been given in Shear control ratio calculation.

$N_{UD} = 4770.03$

$A_g = 125663.706$

$f_c E = 33.00$

$f_{yt} E = f_{yl} E = 555.56$

$\rho_l = \text{Area\_Tot\_Long\_Rein} / (A_g) = 0.0243$

$f_c E = 33.00$

End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (b)