

# Detailed Member Calculations

**Units: N&mm**

**Regulation: ASCE 41-17**

## Calculation No. 1

column C1, Floor 1

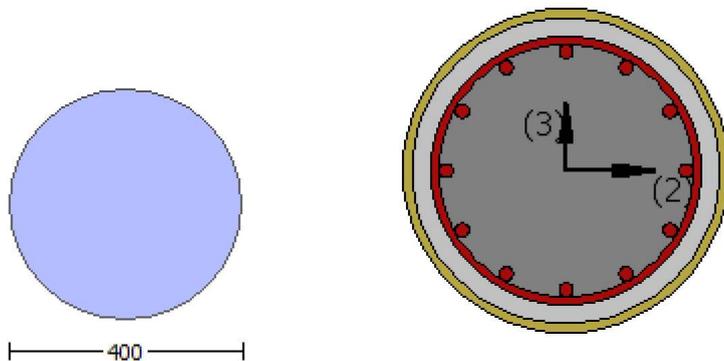
Limit State: Operational Level (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Shear capacity VRd

Edge: Start

Local Axis: (2)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (a)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 16.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 420.00$

Concrete Elasticity,  $E_c = 23025.204$

Steel Elasticity,  $E_s = 200000.00$

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Note: Especially for the calculation of  $\gamma$  for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE41-17).

Existing material: Concrete Strength,  $f_c = f_{cm} = 24.00$

Existing material: Steel Strength,  $f_s = f_{sm} = 525.00$

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Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at  $135^\circ$ )

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou,min} = l_b/l_d \geq 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $NoDir = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

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Stepwise Properties

EDGE -A-

Bending Moment,  $M_a = -1.0733E+007$

Shear Force,  $V_a = -3575.956$

EDGE -B-

Bending Moment,  $M_b = 0.09489048$

Shear Force,  $V_b = 3575.956$

BOTH EDGES

Axial Force,  $F = -4769.80$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{sl,t} = 1272.345$

-Compression:  $A_{sl,c} = 1781.283$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{sl,ten} = 1017.876$

-Compression:  $A_{sl,com} = 1017.876$

-Middle:  $A_{sl,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $Db_{L,ten} = 18.00$

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Existing component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = \phi V_n = 264267.995$

$V_n$  ((10.3), ASCE 41-17) =  $k_n l V_{CoI0} = 264267.995$

$V_{CoI} = 264267.995$

$k_n l = 1.00$

displacement\_ductility\_demand = 0.01013255

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NOTE: In expression (10-3) ' $V_s = A_v f_y d/s$ ' is replaced by ' $V_s + \phi V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

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= 1 (normal-weight concrete)

$f_c' = 16.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 4.00$

$M_u = 1.0733E+007$

$V_u = 3575.956$

$d = 0.8 \cdot D = 320.00$

$N_u = 4769.80$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 165809.354$   
 $A_v = \frac{1}{2} A_{stirrup} = 123370.055$   
 $f_y = 420.00$   
 $s = 100.00$   
 $V_s$  is multiplied by  $Col = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a)\sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = b1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, a1)|)$ , with:  
 total thickness per orientation,  $tf1 = NL * t / NoDir = 1.016$   
 $dfv = d$  (figure 11.2, ACI 440) = 370.00  
 $ffe$  ((11-5), ACI 440) = 259.312  
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 213705.936$   
 $bw * d = \frac{1}{4} * d * d = 80424.772$

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 displacement ductility demand is calculated as  $\delta / y$

- Calculation of  $\delta / y$  for END A -  
 for rotation axis 3 and integ. section (a)

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 From analysis, chord rotation  $\theta = 0.00033407$   
 $y = (M_y * L_s / 3) / E_{eff} = 0.03297001$  ((4.29), Biskinis Phd))  
 $M_y = 2.8606E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 3001.332  
 From table 10.5, ASCE 41\_17:  $E_{eff} = \text{factor} * E_c * I_g = 8.6803E+012$   
 $\text{factor} = 0.30$   
 $A_g = 125663.706$   
 $f_c' = 24.00$   
 $N = 4769.80$   
 $E_c * I_g = 2.8934E+013$

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 Calculation of Yielding Moment  $M_y$

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 Calculation of  $\delta / y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

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 $M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 2.8606E+008$   
 $y$  ((10a) or (10b)) = 1.6206346E-005  
 $M_{y\_ten}$  (8a) = 2.8606E+008  
 $y_{ten}$  (7a) = 73.23937  
 error of function (7a) = 0.00140751  
 $M_{y\_com}$  (8b) = 4.0293E+008  
 $y_{com}$  (7b) = 68.98129  
 error of function (7b) = -0.00043134  
 with  $e_y = 0.002625$   
 $e_{co} = 0.002$   
 $a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d1 = 44.00$   
 $R = 200.00$   
 $v = 0.00134935$   
 $N = 4769.80$   
 $A_c = 125663.706$   
 $= 0.45352339$

with  $f_c^*$  ((12.3), ACI 440) = 28.12975  
 $f_c = 24.00$   
 $f_l = 1.3173$   
 $k = 1$   
Effective FRP thickness,  $t_f = NL \cdot t \cdot \cos(b_1) = 1.016$   
 $e_{fe}$  ((12.5) and (12.7)) = 0.004  
 $E_f = 64828.00$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

End Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (a)

## Calculation No. 2

column C1, Floor 1

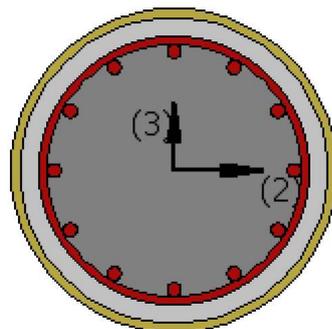
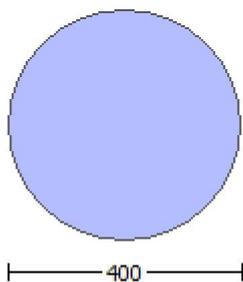
Limit State: Operational Level (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity (  $\theta$  )

Edge: Start

Local Axis: (2)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 24.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 525.00$

Concrete Elasticity,  $E_c = 23025.204$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,  
the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 656.25$

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Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.72976

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou, \min} > 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $NoDir = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

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Stepwise Properties

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At local axis: 3

EDGE -A-

Shear Force,  $V_a = -2.2164346E-030$

EDGE -B-

Shear Force,  $V_b = 2.2164346E-030$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 0.00$

-Compression:  $As_c = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t, \text{ten}} = 1017.876$

-Compression:  $As_{l, \text{com}} = 1017.876$

-Middle:  $As_{l, \text{mid}} = 1017.876$

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Calculation of Shear Capacity ratio,  $V_e/V_r = 0.48361756$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 186373.736$

with

$M_{pr1} = \text{Max}(M_{u1+}, M_{u1-}) = 2.7956E+008$

$M_{u1+} = 2.7956E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$M_{u1-} = 2.7956E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(M_{u2+}, M_{u2-}) = 2.7956E+008$

$M_{u2+} = 2.7956E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

$M_{u2-} = 2.7956E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

Calculation of Mu1+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.7956E+008

= 1.06465  
' = 0.94240061  
error of function (3.68), Biskinis Phd = 47223.857  
From 5A.2, TBDY: fcc = fc\* c = 41.51419  
conf. factor c = 1.72976  
fc = 24.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 656.25  
lb/d = 1.00  
d1 = 44.00  
R = 200.00  
v = 0.00133941  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.56255742

Calculation of ratio lb/d

Adequate Lap Length: lb/d >= 1

Calculation of Mu1-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.7956E+008

= 1.06465  
' = 0.94240061  
error of function (3.68), Biskinis Phd = 47223.857  
From 5A.2, TBDY: fcc = fc\* c = 41.51419  
conf. factor c = 1.72976  
fc = 24.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 656.25  
lb/d = 1.00  
d1 = 44.00  
R = 200.00  
v = 0.00133941  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.56255742

Calculation of ratio lb/d

Adequate Lap Length: lb/d >= 1

Calculation of Mu2+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.7956E+008

= 1.06465  
' = 0.94240061

error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_c' = f_c \cdot c = 41.51419$

conf. factor  $c = 1.72976$

$f_c = 24.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00133941$

$N = 4771.233$

$A_c = 125663.706$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of  $\mu_2$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$\mu = 2.7956E+008$

$= 1.06465$

$' = 0.94240061$

error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_c' = f_c \cdot c = 41.51419$

conf. factor  $c = 1.72976$

$f_c = 24.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00133941$

$N = 4771.233$

$A_c = 125663.706$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 385374.211$

Calculation of Shear Strength at edge 1,  $V_{r1} = 385374.211$

$V_{r1} = V_{Co1} ((10.3), ASCE 41-17) = k_{nl} \cdot V_{Co10}$

$V_{Co10} = 385374.211$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_{s+} = f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

$f_c' = 24.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu = 1.4753320E-011$

$V_u = 2.2164346E-030$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

Ag = 125663.706

From (11.5.4.8), ACI 318-14: Vs = 207261.692

Av =  $\sqrt{2} \cdot A_{\text{stirrup}} = 123370.055$

fy = 525.00

s = 100.00

Vs is multiplied by Col = 0.00

s/d = 0.3125

Vf ((11-3)-(11.4), ACI 440) = 194961.134

f = 0.95, for fully-wrapped sections

wf/sf = 1 (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression,

where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function Vf(  $\theta, \alpha$  ), is implemented for every different fiber orientation ai, as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta_1 = b_1 + 90^\circ = 90.00$

Vf = Min(|Vf(45,  $\theta_1$ )|, |Vf(-45,  $\theta_1$ )|), with:

total thickness per orientation, tf1 = NL\*t/NoDir = 1.016

dfv = d (figure 11.2, ACI 440) = 370.00

ffe ((11-5), ACI 440) = 259.312

Ef = 64828.00

fe = 0.004, from (11.6a), ACI 440

with fu = 0.01

From (11-11), ACI 440: Vs + Vf <= 261735.249

bw\*d =  $\sqrt{2} \cdot d^2 / 4 = 80424.772$

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Calculation of Shear Strength at edge 2, Vr2 = 385374.211

Vr2 = VCol ((10.3), ASCE 41-17) = knl\*VCol0

VCol0 = 385374.211

kn1 = 1 (zero step-static loading)

-----  
NOTE: In expression (10-3) 'Vs = Av\*fy\*d/s' is replaced by 'Vs+ f\*Vf' where Vf is the contribution of FRPs (11.3), ACI 440).

-----  
= 1 (normal-weight concrete)

fc' = 24.00, but  $fc'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

M/Vd = 2.00

Mu = 1.4753320E-011

Vu = 2.2164346E-030

d = 0.8\*D = 320.00

Nu = 4771.233

Ag = 125663.706

From (11.5.4.8), ACI 318-14: Vs = 207261.692

Av =  $\sqrt{2} \cdot A_{\text{stirrup}} = 123370.055$

fy = 525.00

s = 100.00

Vs is multiplied by Col = 0.00

s/d = 0.3125

Vf ((11-3)-(11.4), ACI 440) = 194961.134

f = 0.95, for fully-wrapped sections

wf/sf = 1 (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression,

where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function Vf(  $\theta, \alpha$  ), is implemented for every different fiber orientation ai, as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta_1 = b_1 + 90^\circ = 90.00$

Vf = Min(|Vf(45,  $\theta_1$ )|, |Vf(-45,  $\theta_1$ )|), with:

total thickness per orientation, tf1 = NL\*t/NoDir = 1.016

dfv = d (figure 11.2, ACI 440) = 370.00

ffe ((11-5), ACI 440) = 259.312

Ef = 64828.00

fe = 0.004, from (11.6a), ACI 440

with fu = 0.01

From (11-11), ACI 440: Vs + Vf <= 261735.249

bw\*d =  $\sqrt{2} \cdot d^2 / 4 = 80424.772$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At local axis: 3

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Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At Shear local axis: 2  
(Bending local axis: 3)  
Section Type: rccs

Constant Properties

-----  
Knowledge Factor,  $\gamma = 1.00$   
Mean strength values are used for both shear and moment calculations.  
Consequently:  
Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 24.00$   
Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 525.00$   
Concrete Elasticity,  $E_c = 23025.204$   
Steel Elasticity,  $E_s = 200000.00$   
#####  
Note: Especially for the calculation of moment strengths,  
the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14  
Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 656.25$   
#####  
Diameter,  $D = 400.00$   
Cover Thickness,  $c = 25.00$   
Mean Confinement Factor overall section = 1.72976  
Element Length,  $L = 3000.00$   
Primary Member  
Smooth Bars  
Ductile Steel  
With Detailing for Earthquake Resistance (including stirrups closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Adequate Lap Length ( $l_o/l_{ou}, \min >= 1$ )  
FRP Wrapping Data  
Type: Carbon  
Cured laminate properties (design values)  
Thickness,  $t = 1.016$   
Tensile Strength,  $f_{fu} = 1055.00$   
Tensile Modulus,  $E_f = 64828.00$   
Elongation,  $e_{fu} = 0.01$   
Number of directions,  $N_{oDir} = 1$   
Fiber orientations,  $b_i = 0.00^\circ$   
Number of layers,  $N_L = 1$   
Radius of rounding corners,  $R = 40.00$

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Stepwise Properties

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At local axis: 2  
EDGE -A-  
Shear Force,  $V_a = 7.8886091E-031$   
EDGE -B-  
Shear Force,  $V_b = -7.8886091E-031$   
BOTH EDGES  
Axial Force,  $F = -4771.233$   
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $A_{slt} = 0.00$   
-Compression:  $A_{slc} = 3053.628$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $A_{sl,ten} = 1017.876$   
-Compression:  $A_{sl,com} = 1017.876$   
-Middle:  $A_{sl,mid} = 1017.876$   
-----  
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Calculation of Shear Capacity ratio ,  $V_e/V_r = 0.48361756$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 186373.736$   
with

$M_{pr1} = \text{Max}(M_{u1+} , M_{u1-}) = 2.7956E+008$

$M_{u1+} = 2.7956E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$M_{u1-} = 2.7956E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(M_{u2+} , M_{u2-}) = 2.7956E+008$

$M_{u2+} = 2.7956E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

$M_{u2-} = 2.7956E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

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Calculation of  $M_{u1+}$   
-----

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$

$M_u = 2.7956E+008$

-----  
= 1.06465

' = 0.94240061

error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c^* \quad c = 41.51419$

conf. factor  $c = 1.72976$

$f_c = 24.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00133941$

$N = 4771.233$

$A_c = 125663.706$

=  $\cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$   
-----

Calculation of ratio  $l_b/d$   
-----

Adequate Lap Length:  $l_b/d \geq 1$   
-----  
-----

Calculation of  $M_{u1-}$   
-----

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$

$M_u = 2.7956E+008$

-----  
= 1.06465

' = 0.94240061

error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c^* \quad c = 41.51419$

conf. factor  $c = 1.72976$

$f_c = 24.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00133941$

$N = 4771.233$

$A_c = 125663.706$

=  $\cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$

-----  
Calculation of ratio lb/d  
-----

Adequate Lap Length:  $lb/d \geq 1$   
-----  
-----  
-----

Calculation of  $\mu_{2+}$   
-----  
-----

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 2.7956E+008$   
-----

$\beta = 1.06465$   
 $\beta' = 0.94240061$   
error of function (3.68), Biskinis Phd = 47223.857  
From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$   
conf. factor  $c = 1.72976$   
 $f_c = 24.00$   
From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (lb/d)^{2/3}) = 656.25$   
 $lb/d = 1.00$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00133941$   
 $N = 4771.233$   
 $Ac = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (lb/d)^{2/3}) = 0.56255742$   
-----

Calculation of ratio lb/d  
-----

Adequate Lap Length:  $lb/d \geq 1$   
-----  
-----  
-----

Calculation of  $\mu_{2-}$   
-----  
-----

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 2.7956E+008$   
-----

$\beta = 1.06465$   
 $\beta' = 0.94240061$   
error of function (3.68), Biskinis Phd = 47223.857  
From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$   
conf. factor  $c = 1.72976$   
 $f_c = 24.00$   
From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (lb/d)^{2/3}) = 656.25$   
 $lb/d = 1.00$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00133941$   
 $N = 4771.233$   
 $Ac = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (lb/d)^{2/3}) = 0.56255742$   
-----

Calculation of ratio lb/d  
-----

Adequate Lap Length:  $lb/d \geq 1$   
-----  
-----  
-----

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 385374.211$

Calculation of Shear Strength at edge 1,  $V_{r1} = 385374.211$

$V_{r1} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_n l * V_{Col0}$

$V_{Col0} = 385374.211$

$k_n l = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\lambda = 1$  (normal-weight concrete)

$f_c' = 24.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M / V d = 2.00$

$\mu_u = 7.3565446E-012$

$\nu_u = 7.8886091E-031$

$d = 0.8 * D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 207261.692$

$A_v = \sqrt{2} * A_{stirrup} = 123370.055$

$f_y = 525.00$

$s = 100.00$

$V_s$  is multiplied by  $\text{Col} = 0.00$

$s / d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections

$w_f / s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = 45^\circ$  and  $\alpha = 90^\circ$

$V_f = \text{Min}(|V_f(45, 90)|, |V_f(-45, 90)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L * t / N_{oDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_{e} = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 261735.249$

$b_w * d = \sqrt{2} * d^2 / 4 = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 385374.211$

$V_{r2} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_n l * V_{Col0}$

$V_{Col0} = 385374.211$

$k_n l = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\lambda = 1$  (normal-weight concrete)

$f_c' = 24.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M / V d = 2.00$

$\mu_u = 7.3565446E-012$

$\nu_u = 7.8886091E-031$

$d = 0.8 * D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 207261.692$

$A_v = \sqrt{2} * A_{stirrup} = 123370.055$

$f_y = 525.00$

$s = 100.00$

$V_s$  is multiplied by  $\text{Col} = 0.00$

$s / d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections

$wf/sf = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a)\sin a$  which is more a generalised expression, where  $a$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = 45^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, a)|)$ , with:

total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 261735.249$

$b_w \cdot d = \frac{A_s \cdot d}{4} = 80424.772$

-----  
End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At local axis: 2

-----  
Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (a)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 24.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 525.00$

Concrete Elasticity,  $E_c = 23025.204$

Steel Elasticity,  $E_s = 200000.00$

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at  $135^\circ$ )

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $b/d \geq 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $\text{NoDir} = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

-----  
Stepwise Properties

Bending Moment,  $M = 8.1497328E-010$

Shear Force,  $V_2 = -3575.956$

Shear Force,  $V_3 = -2.9854277E-013$

Axial Force,  $F = -4769.80$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_{lt} = 1272.345$

-Compression:  $As_{lc} = 1781.283$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{l,ten} = 1017.876$

-Compression:  $As_{l,com} = 1017.876$

-Middle:  $As_{l,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $Db_L = 18.00$

Existing component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u, R = * u = 0.02147769$   
 $u = y + p = 0.02147769$

- Calculation of  $y$  -

$y = (M_y * L_s / 3) / E_{eff} = 0.01647769$  ((4.29), Biskinis Phd)

$M_y = 2.8606E+008$

$L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 1500.00

From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 8.6803E+012$

factor = 0.30

$A_g = 125663.706$

$f_c' = 24.00$

$N = 4769.80$

$E_c * I_g = 2.8934E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y,ten}, M_{y,com}) = 2.8606E+008$

$y$  ((10a) or (10b)) =  $1.6206346E-005$

$M_{y,ten}$  (8a) =  $2.8606E+008$

$y_{ten}$  (7a) = 73.23937

error of function (7a) = 0.00140751

$M_{y,com}$  (8b) =  $4.0293E+008$

$y_{com}$  (7b) = 68.98129

error of function (7b) = -0.00043134

with  $e_y = 0.002625$

$e_{co} = 0.002$

$a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)

$d_1 = 44.00$

$R = 200.00$

$v = 0.00134935$

$N = 4769.80$

$A_c = 125663.706$

= 0.45352339

with  $f_c^*$  ((12.3), ACI 440) = 28.12975

$f_c = 24.00$

$f_l = 1.3173$

$k = 1$

Effective FRP thickness,  $t_f = NL * t * \text{Cos}(b_1) = 1.016$

$e_{fe}$  ((12.5) and (12.7)) = 0.004

$E_f = 64828.00$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

- Calculation of  $p$  -

From table 10-9:  $p = 0.005$

with:

- Columns controlled by inadequate development or splicing along the clear height because  $l_b/l_d < 1$

shear control ratio  $V_{yE}/V_{CoIE} = 0.48361756$

$d = 0.00$

$s = 0.00$

$t = 2 \cdot A_v / (d_c \cdot s) + 4 \cdot t_f / D \cdot (f_{fe} / f_s) = 0.00$

$A_v = 78.53982$ , is the area of the circular stirrup

$d_c = D - 2 \cdot \text{cover} - \text{Hoop Diameter} = 340.00$

The term  $2 \cdot t_f / b_w \cdot (f_{fe} / f_s)$  is implemented to account for FRP contribution

where  $f = 2 \cdot t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength

All these variables have already been given in Shear control ratio calculation.

$NUD = 4769.80$

$A_g = 125663.706$

$f_{cE} = 24.00$

$f_{ytE} = f_{ylE} = 525.00$

$\rho_l = \text{Area\_Tot\_Long\_Rein} / (A_g) = 0.0243$

$f_{cE} = 24.00$

-----  
End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (a)  
-----

### Calculation No. 3

column C1, Floor 1

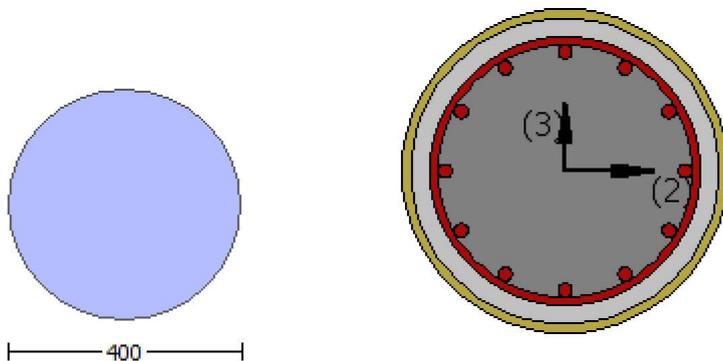
Limit State: Operational Level (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Shear capacity  $V_{Rd}$

Edge: Start

Local Axis: (3)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (a)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\phi = 1.00$   
 Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE41-17.  
 Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17  
 Consequently:  
 Existing material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 16.00$   
 Existing material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 420.00$   
 Concrete Elasticity,  $E_c = 23025.204$   
 Steel Elasticity,  $E_s = 200000.00$   
 #####  
 Note: Especially for the calculation of  $\gamma$  for displacement ductility demand,  
 the expected (mean value) strengths are used (7.5.1.3, ASCE41-17) because bending is considered as  
 Deformation-Controlled Action (Table C7-1, ASCE41-17).  
 Existing material: Concrete Strength,  $f_c = f_{cm} = 24.00$   
 Existing material: Steel Strength,  $f_s = f_{sm} = 525.00$   
 #####  
 Diameter,  $D = 400.00$   
 Cover Thickness,  $c = 25.00$   
 Element Length,  $L = 3000.00$   
 Primary Member  
 Smooth Bars  
 Ductile Steel  
 With Detailing for Earthquake Resistance (including stirrups closed at 135°)  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Adequate Lap Length ( $l_o/l_{ou,min} = l_b/l_d \geq 1$ )  
 FRP Wrapping Data  
 Type: Carbon  
 Cured laminate properties (design values)  
 Thickness,  $t = 1.016$   
 Tensile Strength,  $f_{fu} = 1055.00$   
 Tensile Modulus,  $E_f = 64828.00$   
 Elongation,  $e_{fu} = 0.01$   
 Number of directions,  $N_{oDir} = 1$   
 Fiber orientations,  $b_i: 0.00^\circ$   
 Number of layers,  $N_L = 1$   
 Radius of rounding corners,  $R = 40.00$

Stepwise Properties

EDGE -A-  
 Bending Moment,  $M_a = 8.1497328E-010$   
 Shear Force,  $V_a = -2.9854277E-013$   
 EDGE -B-  
 Bending Moment,  $M_b = 8.1069813E-011$   
 Shear Force,  $V_b = 2.9854277E-013$   
 BOTH EDGES  
 Axial Force,  $F = -4769.80$   
 Longitudinal Reinforcement Area Distribution (in 2 divisions)  
   -Tension:  $A_{sl,t} = 1272.345$   
   -Compression:  $A_{sl,c} = 1781.283$   
 Longitudinal Reinforcement Area Distribution (in 3 divisions)  
   -Tension:  $A_{sl,ten} = 1017.876$   
   -Compression:  $A_{sl,com} = 1017.876$   
   -Middle:  $A_{sl,mid} = 1017.876$   
 Mean Diameter of Tension Reinforcement,  $Db_{L,ten} = 18.00$

Existing component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = \phi V_n = 314830.054$   
 $V_n ((10.3), ASCE 41-17) = knl * V_{CoI} = 314830.054$   
 $V_{CoI} = 314830.054$   
 $knl = 1.00$   
 $displacement\_ductility\_demand = 0.00$

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + \phi * V_f$ '

where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

$f_c' = 16.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 8.1497328E-010$

$\nu_u = 2.9854277E-013$

$d = 0.8 \cdot D = 320.00$

$N_u = 4769.80$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 165809.354$

$A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 123370.055$

$f_y = 420.00$

$s = 100.00$

$V_s$  is multiplied by  $\text{Col} = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot_a) \sin \alpha$  which is more a generalised expression, where  $\alpha$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\alpha = 45^\circ$  and  $\alpha = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\alpha = 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \alpha)|, |V_f(-45, \alpha)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{\text{Dir}} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 213705.936$

$b_w \cdot d = \sqrt{2} \cdot d^2 / 4 = 80424.772$

displacement ductility demand is calculated as  $\delta / y$

- Calculation of  $\delta / y$  for END A -

for rotation axis 2 and integ. section (a)

From analysis, chord rotation  $\theta = 2.9082068E-020$

$y = (M_y \cdot L_s / 3) / E_{\text{eff}} = 0.01647769$  ((4.29), Biskinis Phd))

$M_y = 2.8606E+008$

$L_s = M/V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) = 1500.00

From table 10.5, ASCE 41\_17:  $E_{\text{eff}} = \text{factor} \cdot E_c \cdot I_g = 8.6803E+012$

factor = 0.30

$A_g = 125663.706$

$f_c' = 24.00$

$N = 4769.80$

$E_c \cdot I_g = 2.8934E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\delta / y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 2.8606E+008$

$y$  ((10a) or (10b)) = 1.6206346E-005

$M_{y\_ten}$  (8a) = 2.8606E+008

$\delta / y$  (7a) = 73.23937

error of function (7a) = 0.00140751

$M_{y\_com}$  (8b) = 4.0293E+008

$\delta / y$  (7b) = 68.98129

error of function (7b) = -0.00043134

with  $e_y = 0.002625$

$\epsilon_{co} = 0.002$   
 $\alpha_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00134935$   
 $N = 4769.80$   
 $A_c = 125663.706$   
 $\quad = 0.45352339$   
 with  $f_c^*$  ((12.3), ACI 440) = 28.12975  
 $f_c = 24.00$   
 $f_l = 1.3173$   
 $k = 1$   
 Effective FRP thickness,  $t_f = NL * t * \cos(b_1) = 1.016$   
 $\epsilon_{fe}$  ((12.5) and (12.7)) = 0.004  
 $E_f = 64828.00$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

End Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (a)

## Calculation No. 4

column C1, Floor 1

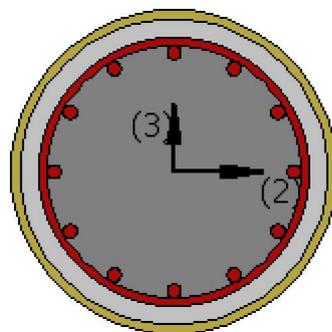
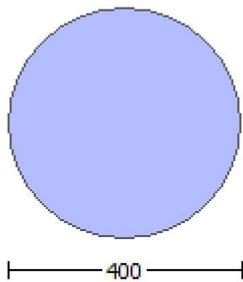
Limit State: Operational Level (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity ( $\theta$ )

Edge: Start

Local Axis: (3)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 24.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 525.00$

Concrete Elasticity,  $E_c = 23025.204$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 656.25$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.72976

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou}, \min > = 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

Stepwise Properties

At local axis: 3

EDGE -A-

Shear Force,  $V_a = -2.2164346E-030$

EDGE -B-

Shear Force,  $V_b = 2.2164346E-030$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{sl,t} = 0.00$

-Compression:  $A_{sl,c} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{sl,ten} = 1017.876$

-Compression:  $A_{sl,com} = 1017.876$

-Middle:  $A_{sl,mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.48361756$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 186373.736$

with

$M_{pr1} = \text{Max}(M_{u1+}, M_{u1-}) = 2.7956E+008$

$M_{u1+} = 2.7956E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$M_{u1-} = 2.7956E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment

direction which is defined for the static loading combination

$$M_{pr2} = \text{Max}(M_{u2+}, M_{u2-}) = 2.7956E+008$$

$M_{u2+} = 2.7956E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

$M_{u2-} = 2.7956E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

-----  
Calculation of  $M_{u1+}$   
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$   
 $M_u = 2.7956E+008$   
-----

$$= 1.06465$$

$$\lambda = 0.94240061$$

error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$

conf. factor  $c = 1.72976$

$$f_c = 24.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$

$$l_b/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00133941$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \lambda \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$$

-----  
Calculation of ratio  $l_b/d$   
-----

Adequate Lap Length:  $l_b/d \geq 1$   
-----  
-----  
-----

-----  
Calculation of  $M_{u1-}$   
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$   
 $M_u = 2.7956E+008$   
-----

$$= 1.06465$$

$$\lambda = 0.94240061$$

error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$

conf. factor  $c = 1.72976$

$$f_c = 24.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$

$$l_b/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00133941$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \lambda \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$$

-----  
Calculation of ratio  $l_b/d$   
-----

Adequate Lap Length:  $l_b/d \geq 1$   
-----  
-----  
-----

-----  
Calculation of  $M_{u2+}$   
-----

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.7956E+008

= 1.06465  
' = 0.94240061  
error of function (3.68), Biskinis Phd = 47223.857  
From 5A.2, TBDY: fcc = fc\* c = 41.51419  
conf. factor c = 1.72976  
fc = 24.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 656.25  
lb/d = 1.00  
d1 = 44.00  
R = 200.00  
v = 0.00133941  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.56255742

Calculation of ratio lb/d

Adequate Lap Length: lb/d >= 1

Calculation of Mu2-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.7956E+008

= 1.06465  
' = 0.94240061  
error of function (3.68), Biskinis Phd = 47223.857  
From 5A.2, TBDY: fcc = fc\* c = 41.51419  
conf. factor c = 1.72976  
fc = 24.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 656.25  
lb/d = 1.00  
d1 = 44.00  
R = 200.00  
v = 0.00133941  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.56255742

Calculation of ratio lb/d

Adequate Lap Length: lb/d >= 1

Calculation of Shear Strength Vr = Min(Vr1,Vr2) = 385374.211

Calculation of Shear Strength at edge 1, Vr1 = 385374.211  
Vr1 = VCol ((10.3), ASCE 41-17) = knl\*VCol0  
VCol0 = 385374.211  
knl = 1 (zero step-static loading)

NOTE: In expression (10-3) 'Vs = Av\*fy\*d/s' is replaced by 'Vs+ f\*Vf  
where Vf is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)  
 $f_c' = 24.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $\mu_u = 1.4753320E-011$   
 $\nu_u = 2.2164346E-030$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 207261.692$   
 $A_v = \frac{1}{2} \cdot A_{\text{stirrup}} = 123370.055$   
 $f_y = 525.00$   
 $s = 100.00$   
 $V_s$  is multiplied by  $\text{Col} = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = 45^\circ$   
 $V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, a_1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{\text{Dir}} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}$  ((11-5), ACI 440) = 259.312  
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 261735.249$   
 $b_w \cdot d = \frac{1}{4} \cdot d^2 = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 385374.211$   
 $V_{r2} = V_{\text{Col}}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{\text{Col}0}$   
 $V_{\text{Col}0} = 385374.211$   
 $k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ '  
 where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)  
 $f_c' = 24.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $\mu_u = 1.4753320E-011$   
 $\nu_u = 2.2164346E-030$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 207261.692$   
 $A_v = \frac{1}{2} \cdot A_{\text{stirrup}} = 123370.055$   
 $f_y = 525.00$   
 $s = 100.00$   
 $V_s$  is multiplied by  $\text{Col} = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = 45^\circ$   
 $V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, a_1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{\text{Dir}} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

ffe ((11-5), ACI 440) = 259.312  
Ef = 64828.00  
fe = 0.004, from (11.6a), ACI 440  
with fu = 0.01  
From (11-11), ACI 440: Vs + Vf <= 261735.249  
bw\*d = \*d\*d/4 = 80424.772

-----  
-----  
End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At local axis: 3

-----  
Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At Shear local axis: 2  
(Bending local axis: 3)  
Section Type: rccs

Constant Properties

-----  
Knowledge Factor, = 1.00  
Mean strength values are used for both shear and moment calculations.  
Consequently:  
Existing material of Primary Member: Concrete Strength, fc = fcm = 24.00  
Existing material of Primary Member: Steel Strength, fs = fsm = 525.00  
Concrete Elasticity, Ec = 23025.204  
Steel Elasticity, Es = 200000.00  
#####  
Note: Especially for the calculation of moment strengths,  
the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14  
Existing material: Steel Strength, fs = 1.25\*fsm = 656.25  
#####  
Diameter, D = 400.00  
Cover Thickness, c = 25.00  
Mean Confinement Factor overall section = 1.72976  
Element Length, L = 3000.00  
Primary Member  
Smooth Bars  
Ductile Steel  
With Detailing for Earthquake Resistance (including stirrups closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Adequate Lap Length (lo/lo, min >= 1)  
FRP Wrapping Data  
Type: Carbon  
Cured laminate properties (design values)  
Thickness, t = 1.016  
Tensile Strength, ffu = 1055.00  
Tensile Modulus, Ef = 64828.00  
Elongation, efu = 0.01  
Number of directions, NoDir = 1  
Fiber orientations, bi: 0.00°  
Number of layers, NL = 1  
Radius of rounding corners, R = 40.00

-----  
Stepwise Properties

-----  
At local axis: 2  
EDGE -A-  
Shear Force, Va = 7.8886091E-031  
EDGE -B-  
Shear Force, Vb = -7.8886091E-031  
BOTH EDGES  
Axial Force, F = -4771.233  
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension: Aslt = 0.00

-Compression:  $A_{sc} = 3053.628$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $A_{s,ten} = 1017.876$   
-Compression:  $A_{s,com} = 1017.876$   
-Middle:  $A_{s,mid} = 1017.876$

-----  
-----  
Calculation of Shear Capacity ratio,  $V_e/V_r = 0.48361756$   
Member Controlled by Flexure ( $V_e/V_r < 1$ )  
Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 186373.736$   
with

$M_{pr1} = \text{Max}(M_{u1+}, M_{u1-}) = 2.7956E+008$   
 $M_{u1+} = 2.7956E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination  
 $M_{u1-} = 2.7956E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination  
 $M_{pr2} = \text{Max}(M_{u2+}, M_{u2-}) = 2.7956E+008$   
 $M_{u2+} = 2.7956E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination  
 $M_{u2-} = 2.7956E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

-----  
Calculation of  $M_{u1+}$   
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$   
 $M_u = 2.7956E+008$

-----  
= 1.06465  
' = 0.94240061  
error of function (3.68), Biskinis Phd = 47223.857  
From 5A.2, TBDY:  $f_{cc} = f_c^* c = 41.51419$   
conf. factor  $c = 1.72976$   
 $f_c = 24.00$   
From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y * \text{Min}(1, 1.25 * (l_b/d)^{2/3}) = 656.25$   
 $l_b/d = 1.00$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00133941$   
 $N = 4771.233$   
 $A_c = 125663.706$   
=  $*\text{Min}(1, 1.25 * (l_b/d)^{2/3}) = 0.56255742$

-----  
Calculation of ratio  $l_b/d$   
-----

Adequate Lap Length:  $l_b/d \geq 1$   
-----  
-----

-----  
Calculation of  $M_{u1-}$   
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$   
 $M_u = 2.7956E+008$

-----  
= 1.06465  
' = 0.94240061  
error of function (3.68), Biskinis Phd = 47223.857  
From 5A.2, TBDY:  $f_{cc} = f_c^* c = 41.51419$   
conf. factor  $c = 1.72976$   
 $f_c = 24.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$

$$l_b/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00133941$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of  $\mu_{2+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$$\mu = 2.7956E+008$$

$$= 1.06465$$

$$' = 0.94240061$$

error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$

$$\text{conf. factor } c = 1.72976$$

$$f_c = 24.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$

$$l_b/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00133941$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of  $\mu_{2-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$$\mu = 2.7956E+008$$

$$= 1.06465$$

$$' = 0.94240061$$

error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$

$$\text{conf. factor } c = 1.72976$$

$$f_c = 24.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$

$$l_b/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00133941$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 385374.211$

Calculation of Shear Strength at edge 1,  $V_{r1} = 385374.211$

$V_{r1} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_n l V_{Col0}$

$V_{Col0} = 385374.211$

$k_n l = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v f_y d/s$ ' is replaced by ' $V_s + f^* V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

$f_c' = 24.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 7.3565446E-012$

$V_u = 7.8886091E-031$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 207261.692$

$A_v = \sqrt{2} \cdot A_{stirrup} = 123370.055$

$f_y = 525.00$

$s = 100.00$

$V_s$  is multiplied by  $\text{Col} = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) =  $194961.134$

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot) \sin \alpha$  which is more a generalised expression, where  $\alpha$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\alpha = 45^\circ$  and  $\alpha = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\alpha = b1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \alpha)|, |V_f(-45, \alpha)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{Dir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) =  $370.00$

$f_{fe}$  ((11-5), ACI 440) =  $259.312$

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 261735.249$

$b_w \cdot d = \sqrt{2} \cdot d^2 / 4 = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 385374.211$

$V_{r2} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_n l V_{Col0}$

$V_{Col0} = 385374.211$

$k_n l = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v f_y d/s$ ' is replaced by ' $V_s + f^* V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

$f_c' = 24.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 7.3565446E-012$

$V_u = 7.8886091E-031$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 207261.692$

$A_v = \sqrt{2} * A_{stirrup} = 123370.055$

$f_y = 525.00$

$s = 100.00$

$V_s$  is multiplied by  $Col = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression, where  $a$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta)$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta_1 = b_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, a_1)|)$ , with:

total thickness per orientation,  $t_{f1} = NL * t / \text{NoDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 261735.249$

$b_w * d = \sqrt{4} * d = 80424.772$

-----  
End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At local axis: 2

-----  
Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (a)

Section Type: rccs

Constant Properties

-----  
Knowledge Factor,  $\gamma = 1.00$

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 24.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 525.00$

Concrete Elasticity,  $E_c = 23025.204$

Steel Elasticity,  $E_s = 200000.00$

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at  $135^\circ$ )

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_b/d \geq 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $\text{NoDir} = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

## Stepwise Properties

Bending Moment,  $M = -1.0733E+007$

Shear Force,  $V2 = -3575.956$

Shear Force,  $V3 = -2.9854277E-013$

Axial Force,  $F = -4769.80$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 1272.345$

-Compression:  $As_c = 1781.283$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{,ten} = 1017.876$

-Compression:  $As_{,com} = 1017.876$

-Middle:  $As_{,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $Db_L = 18.00$

Existing component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_{,R} = u = 0.03797001$   
 $u = y + p = 0.03797001$

- Calculation of  $y$  -

$y = (My * Ls / 3) / Eleff = 0.03297001$  ((4.29), Biskinis Phd))

$My = 2.8606E+008$

$Ls = M/V$  (with  $Ls > 0.1 * L$  and  $Ls < 2 * L$ ) = 3001.332

From table 10.5, ASCE 41\_17:  $Eleff = factor * Ec * I_g = 8.6803E+012$

factor = 0.30

$Ag = 125663.706$

$fc' = 24.00$

$N = 4769.80$

$Ec * I_g = 2.8934E+013$

Calculation of Yielding Moment  $My$

Calculation of  $y$  and  $My$  according to (7) - (8) in Biskinis and Fardis

$My = \text{Min}(My_{,ten}, My_{,com}) = 2.8606E+008$

$y$  ((10a) or (10b)) = 1.6206346E-005

$My_{,ten}$  (8a) = 2.8606E+008

$y_{,ten}$  (7a) = 73.23937

error of function (7a) = 0.00140751

$My_{,com}$  (8b) = 4.0293E+008

$y_{,com}$  (7b) = 68.98129

error of function (7b) = -0.00043134

with  $e_y = 0.002625$

$e_{co} = 0.002$

$apl = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)

$d1 = 44.00$

$R = 200.00$

$v = 0.00134935$

$N = 4769.80$

$Ac = 125663.706$

= 0.45352339

with  $fc^*$  ((12.3), ACI 440) = 28.12975

$fc = 24.00$

$fl = 1.3173$

$k = 1$

Effective FRP thickness,  $tf = NL * t * \text{Cos}(b1) = 1.016$

$e_{fe}$  ((12.5) and (12.7)) = 0.004

$E_f = 64828.00$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

- Calculation of  $p$  -

From table 10-9:  $p = 0.005$

with:

- Columns controlled by inadequate development or splicing along the clear height because  $l_b/d < 1$

shear control ratio  $V_y E / V_{CoI} E = 0.48361756$

$d = 0.00$

$s = 0.00$

$t = 2 \cdot A_v / (d_c \cdot s) + 4 \cdot t_f / D \cdot (f_{fe} / f_s) = 0.00$

$A_v = 78.53982$ , is the area of the circular stirrup

$d_c = D - 2 \cdot \text{cover} - \text{Hoop Diameter} = 340.00$

The term  $2 \cdot t_f / b_w \cdot (f_{fe} / f_s)$  is implemented to account for FRP contribution

where  $f = 2 \cdot t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength

All these variables have already been given in Shear control ratio calculation.

$N_{UD} = 4769.80$

$A_g = 125663.706$

$f_{cE} = 24.00$

$f_{ytE} = f_{ylE} = 525.00$

$p_l = \text{Area\_Tot\_Long\_Rein} / (A_g) = 0.0243$

$f_{cE} = 24.00$

End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (a)

## Calculation No. 5

column C1, Floor 1

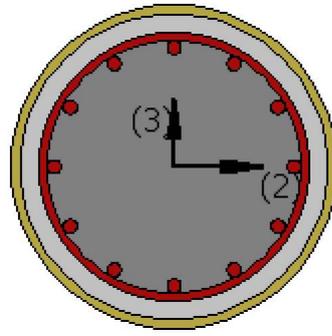
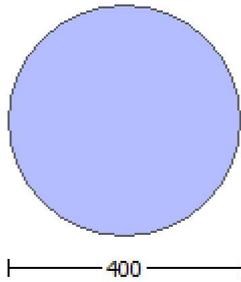
Limit State: Operational Level (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Shear capacity VRd

Edge: End

Local Axis: (2)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (b)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 16.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 420.00$

Concrete Elasticity,  $E_c = 23025.204$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of  $\gamma$  for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE41-17).

Existing material: Concrete Strength,  $f_c = f_{cm} = 24.00$

Existing material: Steel Strength,  $f_s = f_{sm} = 525.00$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou, \min} = l_b/l_d \geq 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $NoDir = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

Stepwise Properties

EDGE -A-

Bending Moment,  $M_a = -1.0733E+007$

Shear Force,  $V_a = -3575.956$

EDGE -B-

Bending Moment,  $M_b = 0.09489048$

Shear Force,  $V_b = 3575.956$

BOTH EDGES

Axial Force,  $F = -4769.80$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_{t1} = 0.00$

-Compression:  $As_{c1} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t,ten} = 1017.876$

-Compression:  $As_{l,com} = 1017.876$

-Middle:  $As_{l,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $Db_{L,ten} = 18.00$

Existing component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = \phi V_n = 314830.054$

$V_n$  ((10.3), ASCE 41-17) =  $k_n \phi V_{Col} = 314830.054$

$V_{Col} = 314830.054$

$k_n = 1.00$

$displacement\_ductility\_demand = 0.05627203$

NOTE: In expression (10-3) ' $V_s = A_v \phi_f y d / s$ ' is replaced by ' $V_s + \phi_f V_f$ ' where  $V_f$  is the contribution of FRPs ((11.3), ACI 440).

= 1 (normal-weight concrete)

$f_c' = 16.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 0.09489048$

$V_u = 3575.956$

$d = 0.8 \cdot D = 320.00$

$N_u = 4769.80$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 165809.354$

$A_v = \phi_f A_{stirrup} = 123370.055$

$f_y = 420.00$

$s = 100.00$

$V_s$  is multiplied by  $\phi_{Col} = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) =  $194961.134$

$\phi_f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = 45^\circ$  and  $\alpha = 90^\circ$  =  $90.00$

$V_f = \text{Min}(|V_f(45^\circ, 90^\circ)|, |V_f(-45^\circ, 90^\circ)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{oDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) =  $370.00$

$\phi_{fe}$  ((11-5), ACI 440) =  $259.312$

$E_f = 64828.00$

$\phi_{fe} = 0.004$ , from (11.6a), ACI 440

with  $\phi_{fu} = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 213705.936$

$b_w \cdot d = \phi_f \cdot d^2 / 4 = 80424.772$

$displacement\_ductility\_demand$  is calculated as  $\phi / y$

- Calculation of  $\phi / y$  for END B -

for rotation axis 3 and integ. section (b)

From analysis, chord rotation  $\phi = 0.00018545$

$y = (M_y \cdot L_s / 3) / E_{eff} = 0.00329554$  ((4.29), Biskinis Phd))

$M_y = 2.8606E+008$

$L_s = M/V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) =  $300.00$

From table 10.5, ASCE 41\_17:  $E_{eff} = factor \cdot E_c \cdot I_g = 8.6803E+012$

factor = 0.30  
Ag = 125663.706  
fc' = 24.00  
N = 4769.80  
Ec\*Ig = 2.8934E+013

-----  
-----  
Calculation of Yielding Moment My

-----  
Calculation of  $\rho_y$  and My according to (7) - (8) in Biskinis and Fardis

-----  
My = Min(My\_ten, My\_com) = 2.8606E+008  
 $\rho_y$  ((10a) or (10b)) = 1.6206346E-005  
My\_ten (8a) = 2.8606E+008  
 $\rho_{y\_ten}$  (7a) = 73.23937  
error of function (7a) = 0.00140751  
My\_com (8b) = 4.0293E+008  
 $\rho_{y\_com}$  (7b) = 68.98129  
error of function (7b) = -0.00043134  
with  $e_y$  = 0.002625  
 $e_{co}$  = 0.002  
 $a_{pl}$  = 0.45 ((9c) in Biskinis and Fardis for FRP Wrap)  
d1 = 44.00  
R = 200.00  
v = 0.00134935  
N = 4769.80  
Ac = 125663.706  
= 0.45352339  
with  $f_c^*$  ((12.3), ACI 440) = 28.12975  
fc = 24.00  
fl = 1.3173  
k = 1  
Effective FRP thickness,  $t_f$  = NL\*t\*cos(b1) = 1.016  
efe ((12.5) and (12.7)) = 0.004  
Ef = 64828.00

-----  
Calculation of ratio lb/d

-----  
Adequate Lap Length: lb/d >= 1

-----  
End Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (b)

-----  
**Calculation No. 6**

column C1, Floor 1

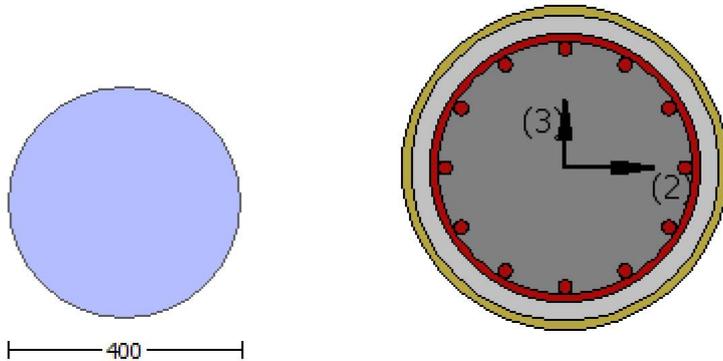
Limit State: Operational Level (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity ( $\theta_u$ )

Edge: End

Local Axis: (2)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 24.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 525.00$

Concrete Elasticity,  $E_c = 23025.204$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 656.25$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.72976

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou, \min} \geq 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

Stepwise Properties

At local axis: 3

EDGE -A-

Shear Force,  $V_a = -2.2164346E-030$

EDGE -B-

Shear Force,  $V_b = 2.2164346E-030$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 0.00$

-Compression:  $As_c = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t,ten} = 1017.876$

-Compression:  $As_{c,com} = 1017.876$

-Middle:  $As_{mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.48361756$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 186373.736$

with  
 $M_{pr1} = \text{Max}(M_{u1+}, M_{u1-}) = 2.7956E+008$   
 $M_{u1+} = 2.7956E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination  
 $M_{u1-} = 2.7956E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination  
 $M_{pr2} = \text{Max}(M_{u2+}, M_{u2-}) = 2.7956E+008$   
 $M_{u2+} = 2.7956E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination  
 $M_{u2-} = 2.7956E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

Calculation of  $M_{u1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$

$M_u = 2.7956E+008$

$\phi = 1.06465$

$\phi' = 0.94240061$

error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$

conf. factor  $c = 1.72976$

$f_c = 24.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00133941$

$N = 4771.233$

$A_c = 125663.706$

$\phi' \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of Mu1-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.7956E+008

= 1.06465  
' = 0.94240061  
error of function (3.68), Biskinis Phd = 47223.857  
From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$   
conf. factor  $c = 1.72976$   
 $f_c = 24.00$   
From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$   
 $l_b/d = 1.00$   
 $d1 = 44.00$   
 $R = 200.00$   
 $v = 0.00133941$   
 $N = 4771.233$   
 $Ac = 125663.706$   
=  $\cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of Mu2+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.7956E+008

= 1.06465  
' = 0.94240061  
error of function (3.68), Biskinis Phd = 47223.857  
From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$   
conf. factor  $c = 1.72976$   
 $f_c = 24.00$   
From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$   
 $l_b/d = 1.00$   
 $d1 = 44.00$   
 $R = 200.00$   
 $v = 0.00133941$   
 $N = 4771.233$   
 $Ac = 125663.706$   
=  $\cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of Mu2-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.7956E+008

= 1.06465

$\rho = 0.94240061$   
 error of function (3.68), Biskinis Phd = 47223.857  
 From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$   
 conf. factor  $c = 1.72976$   
 $f_c = 24.00$   
 From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$   
 $l_b/d = 1.00$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00133941$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 385374.211$

Calculation of Shear Strength at edge 1,  $V_{r1} = 385374.211$

$V_{r1} = V_{CoI}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{CoI0}$

$V_{CoI0} = 385374.211$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)  
 $f_c' = 24.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $\mu_u = 1.4753320E-011$   
 $\nu_u = 2.2164346E-030$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 207261.692$   
 $A_v = \rho_s \cdot A_{stirup} = 123370.055$   
 $f_y = 525.00$   
 $s = 100.00$   
 $V_s$  is multiplied by  $\rho_{col} = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = 45^\circ + 90^\circ = 135^\circ$   
 $V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, 1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{oDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}$  ((11-5), ACI 440) = 259.312  
 $E_f = 64828.00$   
 $f_{e1} = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 261735.249$   
 $b_w \cdot d = \rho_s \cdot d^2 / 4 = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 385374.211$

$V_{r2} = V_{CoI}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{CoI0}$

VCoIO = 385374.211  
knl = 1 (zero step-static loading)

NOTE: In expression (10-3) 'Vs = Av\*fy\*d/s' is replaced by 'Vs+ f\*Vf'  
where Vf is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)  
fc' = 24.00, but  $fc^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
M/Vd = 2.00  
Mu = 1.4753320E-011  
Vu = 2.2164346E-030  
d = 0.8\*D = 320.00  
Nu = 4771.233  
Ag = 125663.706  
From (11.5.4.8), ACI 318-14: Vs = 207261.692  
Av = /2\*A\_stirrup = 123370.055  
fy = 525.00  
s = 100.00  
Vs is multiplied by Col = 0.00  
s/d = 0.3125  
Vf ((11-3)-(11.4), ACI 440) = 194961.134  
f = 0.95, for fully-wrapped sections  
wf/sf = 1 (FRP strips adjacent to one another).  
In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a)\sin a$  which is more a generalised expression,  
where is the angle of the crack direction (see KANEPE).  
This later relation, considered as a function Vf( , ), is implemented for every different fiber orientation ai,  
as well as for 2 crack directions, = 45° and = -45° to take into consideration the cyclic seismic loading.  
orientation 1: 1 = b1 + 90° = 90.00  
Vf = Min(|Vf(45, 1)|, |Vf(-45,a1)|), with:  
total thickness per orientation, tf1 = NL\*t/NoDir = 1.016  
dfv = d (figure 11.2, ACI 440) = 370.00  
ffe ((11-5), ACI 440) = 259.312  
Ef = 64828.00  
fe = 0.004, from (11.6a), ACI 440  
with fu = 0.01  
From (11-11), ACI 440: Vs + Vf <= 261735.249  
bw\*d = \*d\*d/4 = 80424.772

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At local axis: 3

Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At Shear local axis: 2  
(Bending local axis: 3)  
Section Type: rccs

Constant Properties

Knowledge Factor, = 1.00  
Mean strength values are used for both shear and moment calculations.  
Consequently:  
Existing material of Primary Member: Concrete Strength, fc = fcm = 24.00  
Existing material of Primary Member: Steel Strength, fs = fsm = 525.00  
Concrete Elasticity, Ec = 23025.204  
Steel Elasticity, Es = 200000.00  
#####  
Note: Especially for the calculation of moment strengths,  
the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14  
Existing material: Steel Strength, fs = 1.25\*fsm = 656.25  
#####  
Diameter, D = 400.00  
Cover Thickness, c = 25.00  
Mean Confinement Factor overall section = 1.72976

Element Length, L = 3000.00  
Primary Member  
Smooth Bars  
Ductile Steel  
With Detailing for Earthquake Resistance (including stirrups closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Adequate Lap Length ( $l_o/l_{ou}, \min \geq 1$ )  
FRP Wrapping Data  
Type: Carbon  
Cured laminate properties (design values)  
Thickness, t = 1.016  
Tensile Strength,  $f_{fu}$  = 1055.00  
Tensile Modulus,  $E_f$  = 64828.00  
Elongation,  $e_{fu}$  = 0.01  
Number of directions, NoDir = 1  
Fiber orientations,  $b_i$ : 0.00°  
Number of layers, NL = 1  
Radius of rounding corners, R = 40.00

#### Stepwise Properties

At local axis: 2  
EDGE -A-  
Shear Force,  $V_a$  = 7.8886091E-031  
EDGE -B-  
Shear Force,  $V_b$  = -7.8886091E-031  
BOTH EDGES  
Axial Force, F = -4771.233  
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $A_{st}$  = 0.00  
-Compression:  $A_{sc}$  = 3053.628  
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $A_{st,ten}$  = 1017.876  
-Compression:  $A_{sl,com}$  = 1017.876  
-Middle:  $A_{sl,mid}$  = 1017.876

Calculation of Shear Capacity ratio,  $V_e/V_r$  = 0.48361756  
Member Controlled by Flexure ( $V_e/V_r < 1$ )  
Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 186373.736$   
with  
 $M_{pr1} = \text{Max}(\mu_{1+}, \mu_{1-}) = 2.7956E+008$   
 $\mu_{1+} = 2.7956E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction  
which is defined for the static loading combination  
 $\mu_{1-} = 2.7956E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment  
direction which is defined for the static loading combination  
 $M_{pr2} = \text{Max}(\mu_{2+}, \mu_{2-}) = 2.7956E+008$   
 $\mu_{2+} = 2.7956E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction  
which is defined for the the static loading combination  
 $\mu_{2-} = 2.7956E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment  
direction which is defined for the the static loading combination

#### Calculation of $\mu_{1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$   
 $\mu_u = 2.7956E+008$

= 1.06465  
' = 0.94240061  
error of function (3.68), Biskinis Phd = 47223.857  
From 5A.2, TBDY:  $f_{cc} = f_c^* c = 41.51419$

conf. factor  $c = 1.72976$

$f_c = 24.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00133941$

$N = 4771.233$

$A_c = 125663.706$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of  $\mu_{1-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$\mu = 2.7956E+008$

$= 1.06465$

$' = 0.94240061$

error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$

conf. factor  $c = 1.72976$

$f_c = 24.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00133941$

$N = 4771.233$

$A_c = 125663.706$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of  $\mu_{2+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$\mu = 2.7956E+008$

$= 1.06465$

$' = 0.94240061$

error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$

conf. factor  $c = 1.72976$

$f_c = 24.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00133941$

$N = 4771.233$

$A_c = 125663.706$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of  $\mu_2$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 2.7956 \times 10^8$

$$= 1.06465$$

$$' = 0.94240061$$

error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$

conf. factor  $c = 1.72976$

$$f_c = 24.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$

$$l_b/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00133941$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 385374.211$

Calculation of Shear Strength at edge 1,  $V_{r1} = 385374.211$

$V_{r1} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{Col0}$

$$V_{Col0} = 385374.211$$

$$k_{nl} = 1 \text{ (zero step-static loading)}$$

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$$= 1 \text{ (normal-weight concrete)}$$

$$f_c' = 24.00, \text{ but } f_c^{0.5} \leq 8.3 \text{ MPa (22.5.3.1, ACI 318-14)}$$

$$M/d = 2.00$$

$$\mu = 7.3565446 \times 10^{-12}$$

$$V_u = 7.8886091 \times 10^{-31}$$

$$d = 0.8 \cdot D = 320.00$$

$$N_u = 4771.233$$

$$A_g = 125663.706$$

From (11.5.4.8), ACI 318-14:  $V_s = 207261.692$

$$A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 123370.055$$

$$f_y = 525.00$$

$$s = 100.00$$

$V_s$  is multiplied by  $\text{Col} = 0.00$

$$s/d = 0.3125$$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$$f = 0.95, \text{ for fully-wrapped sections}$$

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression,

where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta)$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = b1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, a1)|)$ , with:

total thickness per orientation,  $t_{f1} = NL * t / \text{NoDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 261735.249$

$b_w * d = \rho * d * d / 4 = 80424.772$

-----  
Calculation of Shear Strength at edge 2,  $V_{r2} = 385374.211$

$V_{r2} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_{nl} * V_{Col0}$

$V_{Col0} = 385374.211$

$k_{nl} = 1$  (zero step-static loading)

-----  
NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

-----  
 $\rho = 1$  (normal-weight concrete)

$f_c' = 24.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M / Vd = 2.00$

$\mu_u = 7.3565446E-012$

$\nu_u = 7.8886091E-031$

$d = 0.8 * D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 207261.692$

$A_v = \rho / 2 * A_{stirrup} = 123370.055$

$f_y = 525.00$

$s = 100.00$

$V_s$  is multiplied by  $\rho_{Col} = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections

$w_f / s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin a$  which is more a generalised expression,

where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta)$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = b1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, a1)|)$ , with:

total thickness per orientation,  $t_{f1} = NL * t / \text{NoDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 261735.249$

$b_w * d = \rho * d * d / 4 = 80424.772$

-----  
End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At local axis: 2

-----  
Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (b)

Section Type: rccs

## Constant Properties

Knowledge Factor,  $\phi = 1.00$

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 24.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 525.00$

Concrete Elasticity,  $E_c = 23025.204$

Steel Elasticity,  $E_s = 200000.00$

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_b/l_d \geq 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $\epsilon_{fu} = 0.01$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

## Stepwise Properties

Bending Moment,  $M = 8.1069813E-011$

Shear Force,  $V_2 = 3575.956$

Shear Force,  $V_3 = 2.9854277E-013$

Axial Force,  $F = -4769.80$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{st} = 0.00$

-Compression:  $A_{sc} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{st,ten} = 1017.876$

-Compression:  $A_{sc,com} = 1017.876$

-Middle:  $A_{st,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $D_bL = 18.00$

Existing component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_{,R} = \phi \cdot u = 0.02147769$

$u = y + \phi \cdot p = 0.02147769$

- Calculation of  $y$  -

$y = (M \cdot L_s / 3) / E_{eff} = 0.01647769$  ((4.29), Biskinis Phd)

$M_y = 2.8606E+008$

$L_s = M/V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) = 1500.00

From table 10.5, ASCE 41\_17:  $E_{eff} = \text{factor} \cdot E_c \cdot I_g = 8.6803E+012$

factor = 0.30

$A_g = 125663.706$

$f_c' = 24.00$

$N = 4769.80$

$E_c \cdot I_g = 2.8934E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\gamma$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 2.8606E+008$   
 $\gamma \text{ ((10a) or (10b))} = 1.6206346E-005$   
 $M_{y\_ten} \text{ (8a)} = 2.8606E+008$   
 $\gamma_{ten} \text{ (7a)} = 73.23937$   
error of function (7a) = 0.00140751  
 $M_{y\_com} \text{ (8b)} = 4.0293E+008$   
 $\gamma_{com} \text{ (7b)} = 68.98129$   
error of function (7b) = -0.00043134  
with  $e_y = 0.002625$   
 $e_{co} = 0.002$   
 $a_{pl} = 0.45 \text{ ((9c) in Biskinis and Fardis for FRP Wrap)}$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00134935$   
 $N = 4769.80$   
 $A_c = 125663.706$   
 $= 0.45352339$   
with  $f_c^* \text{ ((12.3), ACI 440)} = 28.12975$   
 $f_c = 24.00$   
 $f_l = 1.3173$   
 $k = 1$   
Effective FRP thickness,  $t_f = N L^* t^* \text{Cos}(b_1) = 1.016$   
 $e_{fe} \text{ ((12.5) and (12.7))} = 0.004$   
 $E_f = 64828.00$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

- Calculation of  $\rho_p$  -

From table 10-9:  $\rho_p = 0.005$

with:

- Columns controlled by inadequate development or splicing along the clear height because  $l_b/d < 1$

shear control ratio  $V_y E / V_{CoI} E = 0.48361756$

$d = 0.00$

$s = 0.00$

$t = 2^*A_v / (d_c^*s) + 4^*t_f / D^* (f_{fe} / f_s) = 0.00$

$A_v = 78.53982$ , is the area of the circular stirrup

$d_c = D - 2^*cover$  - Hoop Diameter = 340.00

The term  $2^*t_f / b_w^* (f_{fe} / f_s)$  is implemented to account for FRP contribution

where  $f = 2^*t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength

All these variables have already been given in Shear control ratio calculation.

$N_{UD} = 4769.80$

$A_g = 125663.706$

$f_{cE} = 24.00$

$f_{ytE} = f_{ylE} = 525.00$

$\rho_l = \text{Area\_Tot\_Long\_Rein} / (A_g) = 0.0243$

$f_{cE} = 24.00$

End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (b)

## Calculation No. 7

column C1, Floor 1

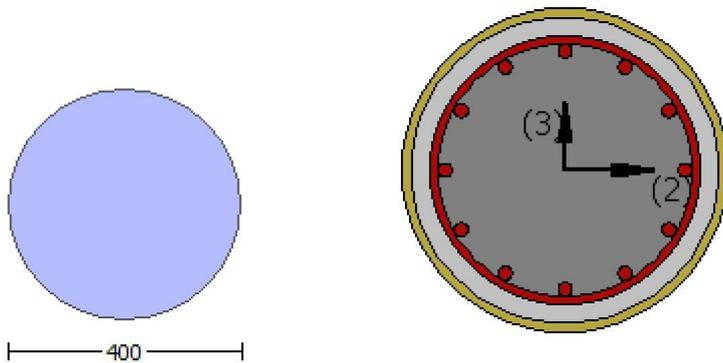
Limit State: Operational Level (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Shear capacity  $V_{Rd}$

Edge: End

Local Axis: (3)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (b)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 16.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 420.00$

Concrete Elasticity,  $E_c = 23025.204$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of  $\gamma$  for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE41-17).

Existing material: Concrete Strength,  $f_c = f_{cm} = 24.00$

Existing material: Steel Strength,  $f_s = f_{sm} = 525.00$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at  $135^\circ$ )

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou, \min} = l_b/l_d \geq 1$ )

FRP Wrapping Data

Type: Carbon  
Cured laminate properties (design values)  
Thickness,  $t = 1.016$   
Tensile Strength,  $f_{fu} = 1055.00$   
Tensile Modulus,  $E_f = 64828.00$   
Elongation,  $e_{fu} = 0.01$   
Number of directions,  $NoDir = 1$   
Fiber orientations,  $b_i = 0.00^\circ$   
Number of layers,  $NL = 1$   
Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

EDGE -A-  
Bending Moment,  $M_a = 8.1497328E-010$   
Shear Force,  $V_a = -2.9854277E-013$   
EDGE -B-  
Bending Moment,  $M_b = 8.1069813E-011$   
Shear Force,  $V_b = 2.9854277E-013$   
BOTH EDGES  
Axial Force,  $F = -4769.80$   
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $As_{t1} = 0.00$   
-Compression:  $As_{c1} = 3053.628$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $As_{t,ten} = 1017.876$   
-Compression:  $As_{l,com} = 1017.876$   
-Middle:  $As_{l,mid} = 1017.876$   
Mean Diameter of Tension Reinforcement,  $Db_{L,ten} = 18.00$

Existing component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = \phi V_n = 314830.054$   
 $V_n$  ((10.3), ASCE 41-17) =  $k_n V_{Co10} = 314830.054$   
 $V_{Co10} = 314830.054$   
 $k_n = 1.00$   
 $displacement\_ductility\_demand = 0.00$

NOTE: In expression (10-3) ' $V_s = A_v f_y d/s$ ' is replaced by ' $V_s + \phi V_f$ '  
where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\phi = 1$  (normal-weight concrete)  
 $f'_c = 16.00$ , but  $f'_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $M_u = 8.1069813E-011$   
 $V_u = 2.9854277E-013$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4769.80$   
 $A_g = 125663.706$   
From (11.5.4.8), ACI 318-14:  $V_s = 165809.354$   
 $A_v = \phi / 2 \cdot A_{stirrup} = 123370.055$   
 $f_y = 420.00$   
 $s = 100.00$   
 $V_s$  is multiplied by  $\phi_{Col} = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) =  $194961.134$   
 $\phi = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression,  
where  $\theta$  is the angle of the crack direction (see KANEPE).  
This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $a_i$ ,  
as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
orientation 1:  $\theta = b1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \theta)|)$ , with:  
total thickness per orientation,  $tf1 = NL \cdot t / NoDir = 1.016$

$dfv = d$  (figure 11.2, ACI 440) = 370.00  
 $ffe$  ((11-5), ACI 440) = 259.312  
 $Ef = 64828.00$   
 $fe = 0.004$ , from (11.6a), ACI 440  
with  $fu = 0.01$   
From (11-11), ACI 440:  $Vs + Vf \leq 213705.936$   
 $bw*d = *d*d/4 = 80424.772$

displacement ductility demand is calculated as  $\delta / y$

- Calculation of  $\delta / y$  for END B -  
for rotation axis 2 and integ. section (b)

From analysis, chord rotation  $\theta = 1.0954441E-020$   
 $y = (My*Ls/3)/Eleff = 0.01647769$  ((4.29), Biskinis Phd))  
 $My = 2.8606E+008$   
 $Ls = M/V$  (with  $Ls > 0.1*L$  and  $Ls < 2*L$ ) = 1500.00  
From table 10.5, ASCE 41\_17:  $Eleff = factor*Ec*Ig = 8.6803E+012$   
 $factor = 0.30$   
 $Ag = 125663.706$   
 $fc' = 24.00$   
 $N = 4769.80$   
 $Ec*Ig = 2.8934E+013$

Calculation of Yielding Moment  $My$

Calculation of  $\delta / y$  and  $My$  according to (7) - (8) in Biskinis and Fardis

$My = \text{Min}(My_{ten}, My_{com}) = 2.8606E+008$   
 $y$  ((10a) or (10b)) = 1.6206346E-005  
 $My_{ten}$  (8a) = 2.8606E+008  
 $y_{ten}$  (7a) = 73.23937  
error of function (7a) = 0.00140751  
 $My_{com}$  (8b) = 4.0293E+008  
 $y_{com}$  (7b) = 68.98129  
error of function (7b) = -0.00043134  
with  $ey = 0.002625$   
 $eco = 0.002$   
 $apl = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d1 = 44.00$   
 $R = 200.00$   
 $v = 0.00134935$   
 $N = 4769.80$   
 $Ac = 125663.706$   
 $\phi = 0.45352339$   
with  $fc^*$  ((12.3), ACI 440) = 28.12975  
 $fc = 24.00$   
 $fl = 1.3173$   
 $k = 1$   
Effective FRP thickness,  $tf = NL*t*\text{Cos}(b1) = 1.016$   
 $efe$  ((12.5) and (12.7)) = 0.004  
 $Ef = 64828.00$

Calculation of ratio  $lb/d$

Adequate Lap Length:  $lb/d \geq 1$

End Of Calculation of Shear Capacity for element: column CC1 of floor 1  
At local axis: 3  
Integration Section: (b)

## Calculation No. 8

column C1, Floor 1

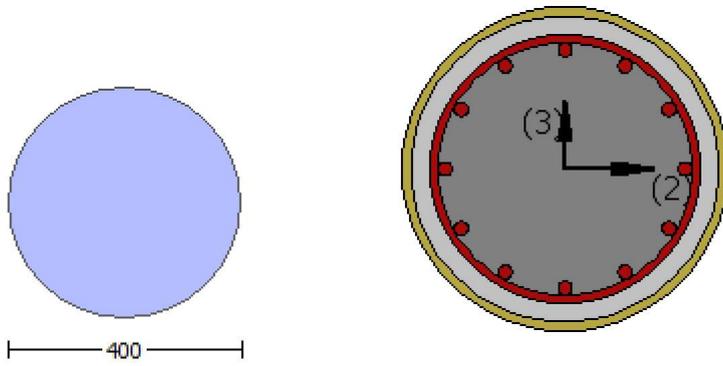
Limit State: Operational Level (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity ( $\theta_r$ )

Edge: End

Local Axis: (3)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 24.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 525.00$

Concrete Elasticity,  $E_c = 23025.204$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 656.25$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.72976

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou, \min} \geq 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $\epsilon_{fu} = 0.01$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

Stepwise Properties

At local axis: 3

EDGE -A-

Shear Force,  $V_a = -2.2164346E-030$

EDGE -B-

Shear Force,  $V_b = 2.2164346E-030$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{sl,t} = 0.00$

-Compression:  $A_{sl,c} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{sl,ten} = 1017.876$

-Compression:  $A_{sl,com} = 1017.876$

-Middle:  $A_{sl,mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.48361756$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 186373.736$

with

$M_{pr1} = \text{Max}(Mu_{1+}, Mu_{1-}) = 2.7956E+008$

$Mu_{1+} = 2.7956E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$Mu_{1-} = 2.7956E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(Mu_{2+}, Mu_{2-}) = 2.7956E+008$

$Mu_{2+} = 2.7956E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination

$Mu_{2-} = 2.7956E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

Calculation of  $Mu_{1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $Mu$

$Mu = 2.7956E+008$

$\beta = 1.06465$

$\beta' = 0.94240061$

error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$

conf. factor  $c = 1.72976$

$f_c = 24.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 656.25$

$l_b/l_d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00133941$

$$N = 4771.233$$
$$Ac = 125663.706$$
$$= *Min(1,1.25*(lb/d)^{2/3}) = 0.56255742$$

Calculation of ratio lb/d

Adequate Lap Length:  $lb/d \geq 1$

Calculation of Mu1-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.7956E+008

$$= 1.06465$$
$$' = 0.94240061$$

error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c * c = 41.51419$   
conf. factor  $c = 1.72976$   
 $f_c = 24.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y * Min(1,1.25*(lb/d)^{2/3}) = 656.25$   
 $lb/d = 1.00$   
 $d1 = 44.00$   
 $R = 200.00$   
 $v = 0.00133941$   
 $N = 4771.233$   
 $Ac = 125663.706$   
 $= *Min(1,1.25*(lb/d)^{2/3}) = 0.56255742$

Calculation of ratio lb/d

Adequate Lap Length:  $lb/d \geq 1$

Calculation of Mu2+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.7956E+008

$$= 1.06465$$
$$' = 0.94240061$$

error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c * c = 41.51419$   
conf. factor  $c = 1.72976$   
 $f_c = 24.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y * Min(1,1.25*(lb/d)^{2/3}) = 656.25$   
 $lb/d = 1.00$   
 $d1 = 44.00$   
 $R = 200.00$   
 $v = 0.00133941$   
 $N = 4771.233$   
 $Ac = 125663.706$   
 $= *Min(1,1.25*(lb/d)^{2/3}) = 0.56255742$

Calculation of ratio lb/d

Adequate Lap Length:  $lb/d \geq 1$

Calculation of Mu2-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.7956E+008

$$= 1.06465$$

$$' = 0.94240061$$

error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c' \cdot c = 41.51419$

$$\text{conf. factor } c = 1.72976$$

$$f_c = 24.00$$

From 10.3.5, ASCE41-17, Final value of fy:  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$

$$l_b/d = 1.00$$

$$d1 = 44.00$$

$$R = 200.00$$

$$v = 0.00133941$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$$

Calculation of ratio lb/d

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 385374.211$

Calculation of Shear Strength at edge 1,  $V_{r1} = 385374.211$

$V_{r1} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{Col0}$

$$V_{Col0} = 385374.211$$

$$k_{nl} = 1 \text{ (zero step-static loading)}$$

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$$= 1 \text{ (normal-weight concrete)}$$

$$f_c' = 24.00, \text{ but } f_c'^{0.5} \leq 8.3 \text{ MPa (22.5.3.1, ACI 318-14)}$$

$$M/d = 2.00$$

$$Mu = 1.4753320E-011$$

$$Vu = 2.2164346E-030$$

$$d = 0.8 \cdot D = 320.00$$

$$Nu = 4771.233$$

$$Ag = 125663.706$$

From (11.5.4.8), ACI 318-14:  $V_s = 207261.692$

$$A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 123370.055$$

$$f_y = 525.00$$

$$s = 100.00$$

$V_s$  is multiplied by  $Col = 0.00$

$$s/d = 0.3125$$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$$f = 0.95, \text{ for fully-wrapped sections}$$

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot) \sin \alpha$  which is more a generalised expression, where  $\alpha$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\alpha, \theta)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = b1 + 90^\circ = 90.00$

$$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \theta)|), \text{ with:}$$

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{\text{Dir}} = 1.016$

$$d_{fv} = d \text{ (figure 11.2, ACI 440)} = 370.00$$

ffe ((11-5), ACI 440) = 259.312  
Ef = 64828.00  
fe = 0.004, from (11.6a), ACI 440  
with fu = 0.01  
From (11-11), ACI 440: Vs + Vf <= 261735.249  
bw\*d = \*d\*d/4 = 80424.772

Calculation of Shear Strength at edge 2, Vr2 = 385374.211  
Vr2 = VCol ((10.3), ASCE 41-17) = knl\*VCol0  
VCol0 = 385374.211  
knl = 1 (zero step-static loading)

NOTE: In expression (10-3) 'Vs = Av\*fy\*d/s' is replaced by 'Vs+ f\*VF'  
where Vf is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)  
fc' = 24.00, but fc^0.5 <= 8.3 MPa (22.5.3.1, ACI 318-14)  
M/Vd = 2.00  
Mu = 1.4753320E-011  
Vu = 2.2164346E-030  
d = 0.8\*D = 320.00  
Nu = 4771.233  
Ag = 125663.706  
From (11.5.4.8), ACI 318-14: Vs = 207261.692  
Av = /2\*A\_stirrup = 123370.055  
fy = 525.00  
s = 100.00

Vs is multiplied by Col = 0.00  
s/d = 0.3125

Vf ((11-3)-(11.4), ACI 440) = 194961.134

f = 0.95, for fully-wrapped sections

wf/sf = 1 (FRP strips adjacent to one another).

In (11.3) sin + cos is replaced with (cot + cota)sina which is more a generalised expression,  
where is the angle of the crack direction (see KANEPE).

This later relation, considered as a function Vf( , ), is implemented for every different fiber orientation ai,  
as well as for 2 crack directions, = 45° and = -45° to take into consideration the cyclic seismic loading.

orientation 1: 1 = b1 + 90° = 90.00

Vf = Min(|Vf(45, 1)|, |Vf(-45,a1)|), with:

total thickness per orientation, tf1 = NL\*t/NoDir = 1.016

dfv = d (figure 11.2, ACI 440) = 370.00

ffe ((11-5), ACI 440) = 259.312

Ef = 64828.00

fe = 0.004, from (11.6a), ACI 440

with fu = 0.01

From (11-11), ACI 440: Vs + Vf <= 261735.249

bw\*d = \*d\*d/4 = 80424.772

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At local axis: 3

Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At Shear local axis: 2  
(Bending local axis: 3)  
Section Type: rccs

Constant Properties

Knowledge Factor, = 1.00

Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength, fc = fcm = 24.00

Existing material of Primary Member: Steel Strength, fs = fsm = 525.00

Concrete Elasticity,  $E_c = 23025.204$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,  
the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 656.25$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.72976

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou, \min} >= 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $NoDir = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

-----  
Stepwise Properties

At local axis: 2

EDGE -A-

Shear Force,  $V_a = 7.8886091E-031$

EDGE -B-

Shear Force,  $V_b = -7.8886091E-031$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 0.00$

-Compression:  $As_c = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t, \text{ten}} = 1017.876$

-Compression:  $As_{l, \text{com}} = 1017.876$

-Middle:  $As_{l, \text{mid}} = 1017.876$

-----  
Calculation of Shear Capacity ratio,  $V_e/V_r = 0.48361756$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 186373.736$

with

$M_{pr1} = \text{Max}(Mu_{1+}, Mu_{1-}) = 2.7956E+008$

$Mu_{1+} = 2.7956E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$Mu_{1-} = 2.7956E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(Mu_{2+}, Mu_{2-}) = 2.7956E+008$

$Mu_{2+} = 2.7956E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

$Mu_{2-} = 2.7956E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

-----  
Calculation of  $Mu_{1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.7956E+008

= 1.06465  
' = 0.94240061  
error of function (3.68), Biskinis Phd = 47223.857  
From 5A.2, TBDY: fcc = fc\* c = 41.51419  
conf. factor c = 1.72976  
fc = 24.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 656.25  
lb/d = 1.00  
d1 = 44.00  
R = 200.00  
v = 0.00133941  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.56255742

Calculation of ratio lb/d

Adequate Lap Length: lb/d >= 1

Calculation of Mu1-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.7956E+008

= 1.06465  
' = 0.94240061  
error of function (3.68), Biskinis Phd = 47223.857  
From 5A.2, TBDY: fcc = fc\* c = 41.51419  
conf. factor c = 1.72976  
fc = 24.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 656.25  
lb/d = 1.00  
d1 = 44.00  
R = 200.00  
v = 0.00133941  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.56255742

Calculation of ratio lb/d

Adequate Lap Length: lb/d >= 1

Calculation of Mu2+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.7956E+008

= 1.06465  
' = 0.94240061  
error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$   
conf. factor  $c = 1.72976$   
 $f_c = 24.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$   
 $l_b/d = 1.00$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00133941$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$

-----  
Calculation of ratio  $l_b/d$

-----  
Adequate Lap Length:  $l_b/d \geq 1$

-----  
Calculation of  $\mu_2$ -

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 2.7956E+008$

-----  
 $= 1.06465$   
 $' = 0.94240061$   
error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$   
conf. factor  $c = 1.72976$   
 $f_c = 24.00$   
From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$   
 $l_b/d = 1.00$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00133941$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$

-----  
Calculation of ratio  $l_b/d$

-----  
Adequate Lap Length:  $l_b/d \geq 1$

-----  
Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 385374.211$

-----  
Calculation of Shear Strength at edge 1,  $V_{r1} = 385374.211$   
 $V_{r1} = V_{Co1} ((10.3), ASCE 41-17) = k_{nl} \cdot V_{Co10}$   
 $V_{Co10} = 385374.211$   
 $k_{nl} = 1$  (zero step-static loading)

-----  
NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ '  
where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

-----  
 $= 1$  (normal-weight concrete)  
 $f_c' = 24.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/d = 2.00$   
 $\mu = 7.3565446E-012$   
 $V_u = 7.8886091E-031$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 207261.692$

$A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 123370.055$

$f_y = 525.00$

$s = 100.00$

$V_s$  is multiplied by  $\text{Col} = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = b_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \alpha)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / \text{NoDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 261735.249$

$b_w \cdot d = \sqrt{2} \cdot d^2 / 4 = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 385374.211$

$V_{r2} = V_{\text{Col}}$  ((10.3), ASCE 41-17) =  $k_n \cdot V_{\text{Col0}}$

$V_{\text{Col0}} = 385374.211$

$k_n = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\gamma = 1$  (normal-weight concrete)

$f'_c = 24.00$ , but  $f'_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 7.3565446E-012$

$\nu_u = 7.8886091E-031$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 207261.692$

$A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 123370.055$

$f_y = 525.00$

$s = 100.00$

$V_s$  is multiplied by  $\text{Col} = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = b_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \alpha)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / \text{NoDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 261735.249$

$b_w \cdot d = \sqrt{2} \cdot d^2 / 4 = 80424.772$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At local axis: 2

Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (b)

Section Type: rccs

#### Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 24.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 525.00$

Concrete Elasticity,  $E_c = 23025.204$

Steel Elasticity,  $E_s = 200000.00$

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at  $135^\circ$ )

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_b/l_d >= 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

Bending Moment,  $M = 0.09489048$

Shear Force,  $V_2 = 3575.956$

Shear Force,  $V_3 = 2.9854277E-013$

Axial Force,  $F = -4769.80$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{st} = 0.00$

-Compression:  $A_{sc} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{st,ten} = 1017.876$

-Compression:  $A_{st,com} = 1017.876$

-Middle:  $A_{st,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $D_{bL} = 18.00$

Existing component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_{,R} = \gamma \cdot u = 0.00829554$

$u = \gamma \cdot u_p = 0.00829554$

- Calculation of  $\gamma$  -

$\gamma = (M_y \cdot L_s / 3) / E_{eff} = 0.00329554 ((4.29), Biskinis Phd)$

$M_y = 2.8606E+008$   
 $L_s = M/V$  (with  $L_s > 0.1*L$  and  $L_s < 2*L$ ) = 300.00  
 From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 8.6803E+012$   
 $factor = 0.30$   
 $A_g = 125663.706$   
 $f_c' = 24.00$   
 $N = 4769.80$   
 $E_c * I_g = 2.8934E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\rho_y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 2.8606E+008$   
 $\rho_y$  ((10a) or (10b)) = 1.6206346E-005  
 $M_{y\_ten}$  (8a) = 2.8606E+008  
 $\rho_{y\_ten}$  (7a) = 73.23937  
 error of function (7a) = 0.00140751  
 $M_{y\_com}$  (8b) = 4.0293E+008  
 $\rho_{y\_com}$  (7b) = 68.98129  
 error of function (7b) = -0.00043134  
 with  $e_y = 0.002625$   
 $e_{co} = 0.002$   
 $a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00134935$   
 $N = 4769.80$   
 $A_c = 125663.706$   
 $\rho = 0.45352339$   
 with  $f_c^*$  ((12.3), ACI 440) = 28.12975  
 $f_c = 24.00$   
 $f_l = 1.3173$   
 $k = 1$   
 Effective FRP thickness,  $t_f = NL * t * \text{Cos}(b_1) = 1.016$   
 $e_{fe}$  ((12.5) and (12.7)) = 0.004  
 $E_f = 64828.00$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

- Calculation of  $\rho_p$  -

From table 10-9:  $\rho_p = 0.005$

with:

- Columns controlled by inadequate development or splicing along the clear height because  $l_b/d < 1$

shear control ratio  $V_y E / V_{CoI} E = 0.48361756$

$d = 0.00$

$s = 0.00$

$t = 2 * A_v / (d_c * s) + 4 * t_f / D * (f_{fe} / f_s) = 0.00$

$A_v = 78.53982$ , is the area of the circular stirrup

$d_c = D - 2 * \text{cover} - \text{Hoop Diameter} = 340.00$

The term  $2 * t_f / b_w * (f_{fe} / f_s)$  is implemented to account for FRP contribution

where  $f = 2 * t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength

All these variables have already been given in Shear control ratio calculation.

$N_{UD} = 4769.80$

$A_g = 125663.706$

$f_c E = 24.00$

$f_{yt} E = f_{yl} E = 525.00$

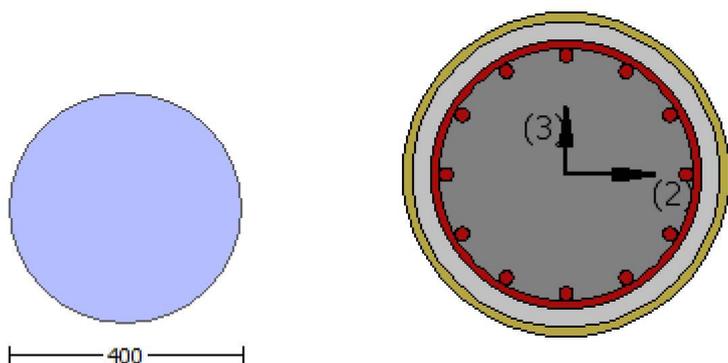
$\rho_l = \text{Area\_Tot\_Long\_Rein} / (A_g) = 0.0243$

$f_c E = 24.00$

End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1  
At local axis: 3  
Integration Section: (b)

## Calculation No. 9

column C1, Floor 1  
Limit State: Life Safety (data interpolation between analysis steps 1 and 2)  
Analysis: Uniform +X  
Check: Shear capacity VRd  
Edge: Start  
Local Axis: (2)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1  
At local axis: 2  
Integration Section: (a)  
Section Type: rccs

### Constant Properties

Knowledge Factor,  $\gamma = 1.00$   
Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE41-17.  
Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17  
Consequently:  
Existing material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 16.00$   
Existing material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 420.00$   
Concrete Elasticity,  $E_c = 23025.204$   
Steel Elasticity,  $E_s = 200000.00$   
#####  
Note: Especially for the calculation of  $\gamma$  for displacement ductility demand,  
the expected (mean value) strengths are used (7.5.1.3, ASCE41-17) because bending is considered as  
Deformation-Controlled Action (Table C7-1, ASCE41-17).  
Existing material: Concrete Strength,  $f_c = f_{cm} = 24.00$   
Existing material: Steel Strength,  $f_s = f_{sm} = 525.00$   
#####  
Diameter,  $D = 400.00$   
Cover Thickness,  $c = 25.00$

Element Length, L = 3000.00  
Primary Member  
Smooth Bars  
Ductile Steel  
With Detailing for Earthquake Resistance (including stirrups closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Adequate Lap Length ( $l_o/l_{ou,min} = l_b/l_d \geq 1$ )  
FRP Wrapping Data  
Type: Carbon  
Cured laminate properties (design values)  
Thickness, t = 1.016  
Tensile Strength,  $f_{fu} = 1055.00$   
Tensile Modulus,  $E_f = 64828.00$   
Elongation,  $e_{fu} = 0.01$   
Number of directions, NoDir = 1  
Fiber orientations,  $b_i = 0.00^\circ$   
Number of layers, NL = 1  
Radius of rounding corners, R = 40.00

-----  
Stepwise Properties  
-----

EDGE -A-  
Bending Moment,  $M_a = -8.5847E+006$   
Shear Force,  $V_a = -2860.289$   
EDGE -B-  
Bending Moment,  $M_b = 0.07589976$   
Shear Force,  $V_b = 2860.289$   
BOTH EDGES  
Axial Force, F = -4770.087  
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $As_t = 1272.345$   
-Compression:  $As_c = 1781.283$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $As_{t,ten} = 1017.876$   
-Compression:  $As_{c,com} = 1017.876$   
-Middle:  $As_{mid} = 1017.876$   
Mean Diameter of Tension Reinforcement,  $Db_{L,ten} = 18.00$

-----  
Existing component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = \phi V_n = 264268.023$   
 $V_n$  ((10.3), ASCE 41-17) =  $k_n \phi V_{CoI0} = 264268.023$   
 $V_{CoI} = 264268.023$   
 $k_n = 1.00$   
displacement\_ductility\_demand = 0.00810469  
-----

NOTE: In expression (10-3) ' $V_s = A_v \phi_f y' d / s$ ' is replaced by ' $V_s + \phi_f V_f$ '  
where  $V_f$  is the contribution of FRPs ((11.3), ACI 440).

-----  
= 1 (normal-weight concrete)  
 $f'_c = 16.00$ , but  $f'_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 4.00$   
 $M_u = 8.5847E+006$   
 $V_u = 2860.289$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4770.087$   
 $A_g = 125663.706$   
From (11.5.4.8), ACI 318-14:  $V_s = 165809.354$   
 $A_v = \phi_f / 2 \cdot A_{stirrup} = 123370.055$   
 $f_y = 420.00$   
 $s = 100.00$   
 $V_s$  is multiplied by  $CoI = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $\phi_f = 0.95$ , for fully-wrapped sections

$wf/sf = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \cot \alpha) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = b1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \alpha)|)$ , with:

total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 213705.936$

$b_w \cdot d = \frac{V_s \cdot d}{4} = 80424.772$

displacement ductility demand is calculated as  $\delta / y$

- Calculation of  $\delta / y$  for END A -

for rotation axis 3 and integ. section (a)

From analysis, chord rotation  $\theta = 0.00026721$

$y = (M_y \cdot L_s / 3) / E_{eff} = 0.03297001$  ((4.29), Biskinis Phd)

$M_y = 2.8606E+008$

$L_s = M/V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) = 3001.332

From table 10.5, ASCE 41\_17:  $E_{eff} = \text{factor} \cdot E_c \cdot I_g = 8.6803E+012$

factor = 0.30

$A_g = 125663.706$

$f_c' = 24.00$

$N = 4770.087$

$E_c \cdot I_g = 2.8934E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\delta / y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 2.8606E+008$

$y$  ((10a) or (10b)) = 1.6206348E-005

$M_{y\_ten}$  (8a) = 2.8606E+008

$\theta_{ten}$  (7a) = 73.23937

error of function (7a) = 0.00140752

$M_{y\_com}$  (8b) = 4.0293E+008

$\theta_{com}$  (7b) = 68.98129

error of function (7b) = -0.00043134

with  $e_y = 0.002625$

$e_{co} = 0.002$

$\alpha_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)

$d_1 = 44.00$

$R = 200.00$

$v = 0.00134943$

$N = 4770.087$

$A_c = 125663.706$

$= 0.45352339$

with  $f_c'$  ((12.3), ACI 440) = 28.12975

$f_c = 24.00$

$f_l = 1.3173$

$k = 1$

Effective FRP thickness,  $t_f = NL \cdot t \cdot \cos(b_1) = 1.016$

$e_{fe}$  ((12.5) and (12.7)) = 0.004

$E_f = 64828.00$

Calculation of ratio  $I_b / I_d$

Adequate Lap Length:  $l_b/l_d \geq 1$

End Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (a)

## Calculation No. 10

column C1, Floor 1

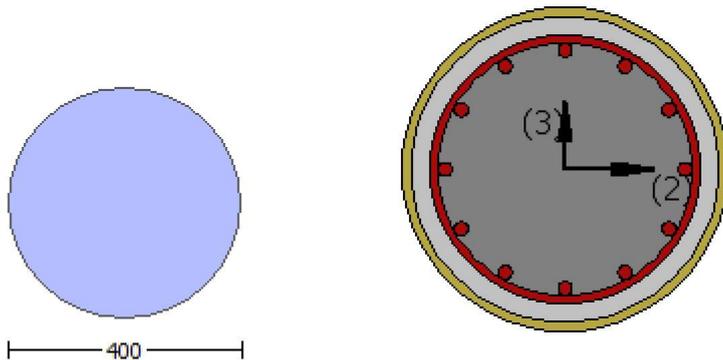
Limit State: Life Safety (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity ( $\theta$ )

Edge: Start

Local Axis: (2)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 24.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 525.00$

Concrete Elasticity,  $E_c = 23025.204$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 656.25$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.72976  
Element Length, L = 3000.00  
Primary Member  
Smooth Bars  
Ductile Steel  
With Detailing for Earthquake Resistance (including stirrups closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Adequate Lap Length ( $l_o/l_{ou}, \min > = 1$ )  
FRP Wrapping Data  
Type: Carbon  
Cured laminate properties (design values)  
Thickness, t = 1.016  
Tensile Strength,  $f_{fu} = 1055.00$   
Tensile Modulus,  $E_f = 64828.00$   
Elongation,  $e_{fu} = 0.01$   
Number of directions, NoDir = 1  
Fiber orientations,  $b_i = 0.00^\circ$   
Number of layers, NL = 1  
Radius of rounding corners, R = 40.00

#### Stepwise Properties

At local axis: 3  
EDGE -A-  
Shear Force,  $V_a = -2.2164346E-030$   
EDGE -B-  
Shear Force,  $V_b = 2.2164346E-030$   
BOTH EDGES  
Axial Force, F = -4771.233  
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $A_{sl} = 0.00$   
-Compression:  $A_{slc} = 3053.628$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $A_{sl,ten} = 1017.876$   
-Compression:  $A_{sl,com} = 1017.876$   
-Middle:  $A_{sl,mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.48361756$   
Member Controlled by Flexure ( $V_e/V_r < 1$ )  
Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 186373.736$   
with  
 $M_{pr1} = \text{Max}(Mu_{1+}, Mu_{1-}) = 2.7956E+008$   
 $Mu_{1+} = 2.7956E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination  
 $Mu_{1-} = 2.7956E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination  
 $M_{pr2} = \text{Max}(Mu_{2+}, Mu_{2-}) = 2.7956E+008$   
 $Mu_{2+} = 2.7956E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination  
 $Mu_{2-} = 2.7956E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

#### Calculation of $Mu_{1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $Mu$   
 $Mu = 2.7956E+008$

= 1.06465  
' = 0.94240061  
error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$

conf. factor  $c = 1.72976$

$f_c = 24.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00133941$

$N = 4771.233$

$A_c = 125663.706$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of  $\mu_{1-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$\mu = 2.7956E+008$

$= 1.06465$

$' = 0.94240061$

error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$

conf. factor  $c = 1.72976$

$f_c = 24.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00133941$

$N = 4771.233$

$A_c = 125663.706$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of  $\mu_{2+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$\mu = 2.7956E+008$

$= 1.06465$

$' = 0.94240061$

error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$

conf. factor  $c = 1.72976$

$f_c = 24.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00133941$

$N = 4771.233$

$$A_c = 125663.706$$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of  $\mu_2$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 2.7956E+008$

$$= 1.06465$$

$$\rho = 0.94240061$$

error of function (3.68), Biskinis Phd = 47223.857  
 From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$   
 conf. factor  $c = 1.72976$   
 $f_c = 24.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$   
 $l_b/d = 1.00$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00133941$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 385374.211$

Calculation of Shear Strength at edge 1,  $V_{r1} = 385374.211$   
 $V_{r1} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{Col0}$   
 $V_{Col0} = 385374.211$   
 $k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$$= 1$$
 (normal-weight concrete)  
 $f_c' = 24.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/d = 2.00$   
 $\mu = 1.4753320E-011$   
 $V_u = 2.2164346E-030$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 207261.692$   
 $A_v = \rho \cdot A_{stirrup} = 123370.055$   
 $f_y = 525.00$   
 $s = 100.00$   
 $V_s$  is multiplied by  $Col = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE). This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = 45^\circ + 90^\circ = 135^\circ$   
 $V_f = \text{Min}(|V_f(45, \alpha)|, |V_f(-45, \alpha)|)$ , with:  
total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}$  ((11-5), ACI 440) = 259.312  
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
with  $f_u = 0.01$   
From (11-11), ACI 440:  $V_s + V_f \leq 261735.249$   
 $b_w \cdot d = A_s \cdot d / 4 = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 385374.211$   
 $V_{r2} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{Col0}$   
 $V_{Col0} = 385374.211$   
 $k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\rho = 1$  (normal-weight concrete)  
 $f'_c = 24.00$ , but  $f'_c \cdot 0.5 \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $\mu_u = 1.4753320E-011$   
 $\nu_u = 2.2164346E-030$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
From (11.5.4.8), ACI 318-14:  $V_s = 207261.692$   
 $A_v = \rho_s \cdot A_{stirrup} = 123370.055$   
 $f_y = 525.00$   
 $s = 100.00$   
 $V_s$  is multiplied by  $\rho_{Col} = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE). This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = 45^\circ + 90^\circ = 135^\circ$   
 $V_f = \text{Min}(|V_f(45, \alpha)|, |V_f(-45, \alpha)|)$ , with:  
total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}$  ((11-5), ACI 440) = 259.312  
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
with  $f_u = 0.01$   
From (11-11), ACI 440:  $V_s + V_f \leq 261735.249$   
 $b_w \cdot d = A_s \cdot d / 4 = 80424.772$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At local axis: 3

Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At Shear local axis: 2  
(Bending local axis: 3)  
Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 24.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 525.00$

Concrete Elasticity,  $E_c = 23025.204$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 656.25$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.72976

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou, min} >= 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i: 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

Stepwise Properties

At local axis: 2

EDGE -A-

Shear Force,  $V_a = 7.8886091E-031$

EDGE -B-

Shear Force,  $V_b = -7.8886091E-031$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{slt} = 0.00$

-Compression:  $A_{slc} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{sl,ten} = 1017.876$

-Compression:  $A_{sl,com} = 1017.876$

-Middle:  $A_{sl,mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.48361756$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 186373.736$

with

$M_{pr1} = \text{Max}(M_{u1+}, M_{u1-}) = 2.7956E+008$

$M_{u1+} = 2.7956E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$M_{u1-} = 2.7956E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$$M_{pr2} = \text{Max}(M_{u2+}, M_{u2-}) = 2.7956E+008$$

$M_{u2+} = 2.7956E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

$M_{u2-} = 2.7956E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

-----  
Calculation of  $M_{u1+}$   
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$   
 $M_u = 2.7956E+008$   
-----

$$= 1.06465$$

$$\lambda = 0.94240061$$

error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c \cdot \lambda = 41.51419$

conf. factor  $\lambda = 1.72976$

$$f_c = 24.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$

$$l_b/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00133941$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \lambda \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$$
  
-----

Calculation of ratio  $l_b/d$   
-----

Adequate Lap Length:  $l_b/d \geq 1$   
-----  
-----

Calculation of  $M_{u1-}$   
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$   
 $M_u = 2.7956E+008$   
-----

$$= 1.06465$$

$$\lambda = 0.94240061$$

error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c \cdot \lambda = 41.51419$

conf. factor  $\lambda = 1.72976$

$$f_c = 24.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$

$$l_b/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00133941$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \lambda \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$$
  
-----

Calculation of ratio  $l_b/d$   
-----

Adequate Lap Length:  $l_b/d \geq 1$   
-----  
-----

Calculation of  $M_{u2+}$   
-----

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu

$$\text{Mu} = 2.7956\text{E}+008$$

$$= 1.06465$$

$$' = 0.94240061$$

error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$

conf. factor  $c = 1.72976$

$$f_c = 24.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$

$$l_b/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00133941$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of Mu2-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu

$$\text{Mu} = 2.7956\text{E}+008$$

$$= 1.06465$$

$$' = 0.94240061$$

error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$

conf. factor  $c = 1.72976$

$$f_c = 24.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$

$$l_b/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00133941$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 385374.211$

Calculation of Shear Strength at edge 1,  $V_{r1} = 385374.211$

$V_{r1} = V_{Co1}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{Co1}$

$$V_{Co1} = 385374.211$$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_{s+} = f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\lambda = 1$  (normal-weight concrete)  
 $f_c' = 24.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $M_u = 7.3565446E-012$   
 $V_u = 7.8886091E-031$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 207261.692$   
 $A_v = \sqrt{2} \cdot A_{stirrup} = 123370.055$   
 $f_y = 525.00$   
 $s = 100.00$   
 $V_s$  is multiplied by  $\lambda = 1.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $\lambda = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin^2 \theta + \cos^2 \theta$  is replaced with  $(\cot^2 \theta + \csc^2 \theta) \sin^2 \theta$  which is more a generalised expression,  
 where  $\theta$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, a_1)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = 45^\circ$  and  $\theta = 90^\circ$   
 $V_f = \text{Min}(|V_f(45, a_1)|, |V_f(-45, a_1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{Dir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}$  ((11-5), ACI 440) = 259.312  
 $E_f = 64828.00$   
 $f_{e1} = 0.004$ , from (11.6a), ACI 440  
 with  $f_{u1} = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 261735.249$   
 $b_w \cdot d = \sqrt{V_u \cdot d / 4} = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 385374.211$   
 $V_{r2} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_n \lambda \sqrt{f_c'} V_{Col}$   
 $V_{Col} = 385374.211$   
 $k_n = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + \lambda \cdot V_f$ '  
 where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\lambda = 1$  (normal-weight concrete)  
 $f_c' = 24.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $M_u = 7.3565446E-012$   
 $V_u = 7.8886091E-031$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 207261.692$   
 $A_v = \sqrt{2} \cdot A_{stirrup} = 123370.055$   
 $f_y = 525.00$   
 $s = 100.00$   
 $V_s$  is multiplied by  $\lambda = 1.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $\lambda = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin^2 \theta + \cos^2 \theta$  is replaced with  $(\cot^2 \theta + \csc^2 \theta) \sin^2 \theta$  which is more a generalised expression,  
 where  $\theta$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, a_1)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = 45^\circ$  and  $\theta = 90^\circ$   
 $V_f = \text{Min}(|V_f(45, a_1)|, |V_f(-45, a_1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{Dir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}$  ((11-5), ACI 440) = 259.312

Ef = 64828.00

fe = 0.004, from (11.6a), ACI 440

with fu = 0.01

From (11-11), ACI 440: Vs + Vf <= 261735.249

bw\*d = \*d\*d/4 = 80424.772

-----  
End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At local axis: 2

-----  
Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (a)

Section Type: rccs

Constant Properties

-----  
Knowledge Factor, = 1.00

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength, fc = fcm = 24.00

Existing material of Primary Member: Steel Strength, fs = fsm = 525.00

Concrete Elasticity, Ec = 23025.204

Steel Elasticity, Es = 200000.00

Diameter, D = 400.00

Cover Thickness, c = 25.00

Element Length, L = 3000.00

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length (lb/l>= 1)

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness, t = 1.016

Tensile Strength, ffu = 1055.00

Tensile Modulus, Ef = 64828.00

Elongation, efu = 0.01

Number of directions, NoDir = 1

Fiber orientations, bi: 0.00°

Number of layers, NL = 1

Radius of rounding corners, R = 40.00

-----  
Stepwise Properties

Bending Moment, M = 6.5482280E-010

Shear Force, V2 = -2860.289

Shear Force, V3 = -2.3879448E-013

Axial Force, F = -4770.087

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension: Aslt = 1272.345

-Compression: Aslc = 1781.283

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension: Asl,ten = 1017.876

-Compression: Asl,com = 1017.876

-Middle: Asl,mid = 1017.876

Mean Diameter of Tension Reinforcement, DbL = 18.00

Existing component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_{R} = u = 0.06958736$   
 $u = y + p = 0.06958736$

-----  
- Calculation of  $y$  -  
-----

$y = (M_y * L_s / 3) / E_{eff} = 0.01647769$  ((4.29), Biskinis Phd)  
 $M_y = 2.8606E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 1500.00  
From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 8.6803E+012$   
factor = 0.30  
Ag = 125663.706  
fc' = 24.00  
N = 4770.087  
 $E_c * I_g = 2.8934E+013$   
-----  
-----

Calculation of Yielding Moment  $M_y$   
-----

Calculation of  $y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis  
-----

$M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 2.8606E+008$   
 $y$  ((10a) or (10b)) = 1.6206348E-005  
 $M_{y\_ten}$  (8a) = 2.8606E+008  
\_ten (7a) = 73.23937  
error of function (7a) = 0.00140752  
 $M_{y\_com}$  (8b) = 4.0293E+008  
\_com (7b) = 68.98129  
error of function (7b) = -0.00043134  
with  $e_y = 0.002625$   
eco = 0.002  
apl = 0.45 ((9c) in Biskinis and Fardis for FRP Wrap)  
d1 = 44.00  
R = 200.00  
v = 0.00134943  
N = 4770.087  
Ac = 125663.706  
= 0.45352339  
with  $f_c^*$  ((12.3), ACI 440) = 28.12975  
fc = 24.00  
fl = 1.3173  
k = 1  
Effective FRP thickness,  $t_f = NL * t * \text{Cos}(b_1) = 1.016$   
efe ((12.5) and (12.7)) = 0.004  
Ef = 64828.00  
-----  
-----

Calculation of ratio  $l_b/d$   
-----

Adequate Lap Length:  $l_b/d \geq 1$   
-----

- Calculation of  $p$  -  
-----

From table 10-9:  $p = 0.05310967$

with:

- Columns controlled by inadequate development or splicing along the clear height because  $l_b/d < 1$

shear control ratio  $V_y E / V_{CoI} O E = 0.48361756$

d = 0.00

s = 0.00

$t = 2 * A_v / (d_c * s) + 4 * t_f / D * (f_{fe} / f_s) = 0.00$

$A_v = 78.53982$ , is the area of the circular stirrup

$d_c = D - 2 * \text{cover}$  - Hoop Diameter = 340.00

The term  $2 * t_f / b_w * (f_{fe} / f_s)$  is implemented to account for FRP contribution

where  $f = 2 * t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength

All these variables have already been given in Shear control ratio calculation.

$$NUD = 4770.087$$

$$Ag = 125663.706$$

$$f_{cE} = 24.00$$

$$f_{ytE} = f_{ylE} = 525.00$$

$$p_l = \text{Area\_Tot\_Long\_Rein}/(Ag) = 0.0243$$

$$f_{cE} = 24.00$$

End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (a)

## Calculation No. 11

column C1, Floor 1

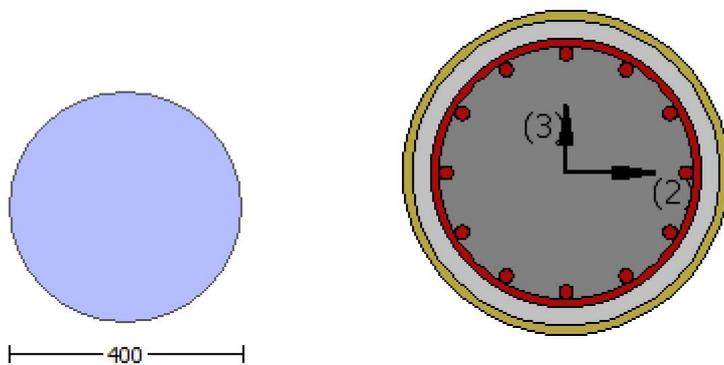
Limit State: Life Safety (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Shear capacity VRd

Edge: Start

Local Axis: (3)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (a)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 16.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 420.00$

Concrete Elasticity,  $E_c = 23025.204$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of  $\mu$  for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE41-17).

Existing material: Concrete Strength,  $f_c = f_{cm} = 24.00$

Existing material: Steel Strength,  $f_s = f_{sm} = 525.00$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou,min} = l_b/l_d \geq 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $NoDir = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

-----  
Stepwise Properties

-----  
EDGE -A-

Bending Moment,  $M_a = 6.5482280E-010$

Shear Force,  $V_a = -2.3879448E-013$

EDGE -B-

Bending Moment,  $M_b = 6.1892434E-011$

Shear Force,  $V_b = 2.3879448E-013$

BOTH EDGES

Axial Force,  $F = -4770.087$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 1272.345$

-Compression:  $As_c = 1781.283$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{l,ten} = 1017.876$

-Compression:  $As_{l,com} = 1017.876$

-Middle:  $As_{l,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $Db_{L,ten} = 18.00$

-----  
Existing component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = \phi V_n = 314830.111$

$V_n$  ((10.3), ASCE 41-17) =  $k_n l^* V_{CoI} = 314830.111$

$V_{CoI} = 314830.111$

$k_n l = 1.00$

$displacement\_ductility\_demand = 0.00$

-----  
NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + \phi * V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

-----  
 $\phi = 1$  (normal-weight concrete)

$f_c' = 16.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$M_u = 6.5482280E-010$

$V_u = 2.3879448E-013$

$d = 0.8 * D = 320.00$

$N_u = 4770.087$

$$A_g = 125663.706$$

$$\text{From (11.5.4.8), ACI 318-14: } V_s = 165809.354$$

$$A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 123370.055$$

$$f_y = 420.00$$

$$s = 100.00$$

$$V_s \text{ is multiplied by } \text{Col} = 0.00$$

$$s/d = 0.3125$$

$$V_f \text{ ((11-3)-(11.4), ACI 440) } = 194961.134$$

$$f = 0.95, \text{ for fully-wrapped sections}$$

$$w_f/s_f = 1 \text{ (FRP strips adjacent to one another).}$$

In (11.3)  $\sin^2 + \cos^2$  is replaced with  $(\cot^2 + \csc^2) \sin^2 \alpha$  which is more a generalised expression,

where  $\alpha$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\alpha = 45^\circ$  and  $\alpha = -45^\circ$  to take into consideration the cyclic seismic loading.

$$\text{orientation 1: } \alpha_1 = b_1 + 90^\circ = 90.00$$

$$V_f = \text{Min}(|V_f(45, \alpha_1)|, |V_f(-45, \alpha_1)|), \text{ with:}$$

$$\text{total thickness per orientation, } t_{f1} = N_L \cdot t / N_{\text{Dir}} = 1.016$$

$$d_{fv} = d \text{ (figure 11.2, ACI 440) } = 370.00$$

$$f_{fe} \text{ ((11-5), ACI 440) } = 259.312$$

$$E_f = 64828.00$$

$$f_e = 0.004, \text{ from (11.6a), ACI 440}$$

$$\text{with } f_u = 0.01$$

$$\text{From (11-11), ACI 440: } V_s + V_f \leq 213705.936$$

$$b_w \cdot d = \sqrt{2} \cdot d^2 / 4 = 80424.772$$

-----  
displacement\_ductility\_demand is calculated as  $\delta / y$

- Calculation of  $\delta / y$  for END A -  
for rotation axis 2 and integ. section (a)

$$\text{From analysis, chord rotation } \theta = 2.3261784E-020$$

$$y = (M_y \cdot L_s / 3) / E_{\text{eff}} = 0.01647769 \text{ ((4.29), Biskinis Phd)}$$

$$M_y = 2.8606E+008$$

$$L_s = M/V \text{ (with } L_s > 0.1 \cdot L \text{ and } L_s < 2 \cdot L) = 1500.00$$

$$\text{From table 10.5, ASCE 41_17: } E_{\text{eff}} = \text{factor} \cdot E_c \cdot I_g = 8.6803E+012$$

$$\text{factor} = 0.30$$

$$A_g = 125663.706$$

$$f_c' = 24.00$$

$$N = 4770.087$$

$$E_c \cdot I_g = 2.8934E+013$$

-----  
Calculation of Yielding Moment  $M_y$

-----  
Calculation of  $\delta / y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$$M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 2.8606E+008$$

$$y \text{ ((10a) or (10b)) } = 1.6206348E-005$$

$$M_{y\_ten} \text{ (8a) } = 2.8606E+008$$

$$y_{ten} \text{ (7a) } = 73.23937$$

$$\text{error of function (7a) } = 0.00140752$$

$$M_{y\_com} \text{ (8b) } = 4.0293E+008$$

$$y_{com} \text{ (7b) } = 68.98129$$

$$\text{error of function (7b) } = -0.00043134$$

$$\text{with } e_y = 0.002625$$

$$e_{co} = 0.002$$

$$a_{pl} = 0.45 \text{ ((9c) in Biskinis and Fardis for FRP Wrap)}$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00134943$$

$$N = 4770.087$$

$$A_c = 125663.706$$

$$= 0.45352339$$

$$\text{with } f_c^* \text{ ((12.3), ACI 440) } = 28.12975$$

$f_c = 24.00$   
 $f_l = 1.3173$   
 $k = 1$   
Effective FRP thickness,  $t_f = NL * t * \cos(b_1) = 1.016$   
efe ((12.5) and (12.7)) = 0.004  
 $E_f = 64828.00$

Calculation of ratio  $I_b/I_d$

Adequate Lap Length:  $I_b/I_d \geq 1$

End Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (a)

## Calculation No. 12

column C1, Floor 1

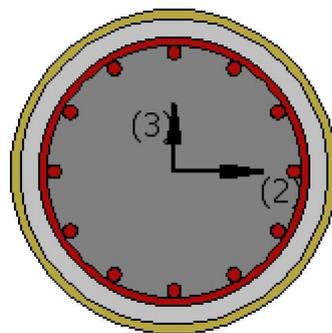
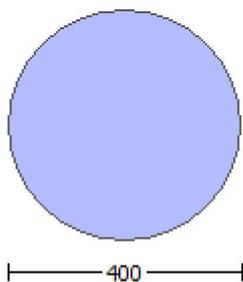
Limit State: Life Safety (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity (  $\theta$  )

Edge: Start

Local Axis: (3)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 24.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 525.00$

Concrete Elasticity,  $E_c = 23025.204$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,  
the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 656.25$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.72976

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou, \min} >= 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $NoDir = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

-----  
Stepwise Properties

At local axis: 3

EDGE -A-

Shear Force,  $V_a = -2.2164346E-030$

EDGE -B-

Shear Force,  $V_b = 2.2164346E-030$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{st} = 0.00$

-Compression:  $A_{sc} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{st, \text{ten}} = 1017.876$

-Compression:  $A_{st, \text{com}} = 1017.876$

-Middle:  $A_{st, \text{mid}} = 1017.876$

-----  
Calculation of Shear Capacity ratio,  $V_e/V_r = 0.48361756$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 186373.736$

with

$M_{pr1} = \text{Max}(M_{u1+}, M_{u1-}) = 2.7956E+008$

$M_{u1+} = 2.7956E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$M_{u1-} = 2.7956E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(M_{u2+}, M_{u2-}) = 2.7956E+008$

$M_{u2+} = 2.7956E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

$M_{u2-} = 2.7956E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

-----  
Calculation of  $M_{u1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.7956E+008

= 1.06465  
' = 0.94240061  
error of function (3.68), Biskinis Phd = 47223.857  
From 5A.2, TBDY: fcc = fc\* c = 41.51419  
conf. factor c = 1.72976  
fc = 24.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 656.25  
lb/d = 1.00  
d1 = 44.00  
R = 200.00  
v = 0.00133941  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.56255742

Calculation of ratio lb/d

Adequate Lap Length: lb/d >= 1

Calculation of Mu1-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.7956E+008

= 1.06465  
' = 0.94240061  
error of function (3.68), Biskinis Phd = 47223.857  
From 5A.2, TBDY: fcc = fc\* c = 41.51419  
conf. factor c = 1.72976  
fc = 24.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 656.25  
lb/d = 1.00  
d1 = 44.00  
R = 200.00  
v = 0.00133941  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.56255742

Calculation of ratio lb/d

Adequate Lap Length: lb/d >= 1

Calculation of Mu2+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.7956E+008

= 1.06465  
' = 0.94240061  
error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$

conf. factor  $c = 1.72976$

$f_c = 24.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00133941$

$N = 4771.233$

$A_c = 125663.706$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of  $\mu_2$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$\mu = 2.7956E+008$

$= 1.06465$

$' = 0.94240061$

error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$

conf. factor  $c = 1.72976$

$f_c = 24.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00133941$

$N = 4771.233$

$A_c = 125663.706$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 385374.211$

Calculation of Shear Strength at edge 1,  $V_{r1} = 385374.211$

$V_{r1} = V_{Co1} ((10.3), ASCE 41-17) = k_{nl} \cdot V_{Co10}$

$V_{Co10} = 385374.211$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_{s+} = f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

$f_c' = 24.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/d = 2.00$

$\mu = 1.4753320E-011$

$V_u = 2.2164346E-030$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 207261.692$

$A_v = \sqrt{2} \cdot A_{stirrup} = 123370.055$

$f_y = 525.00$

$s = 100.00$

$V_s$  is multiplied by  $Col = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin^2 + \cos^2$  is replaced with  $(\cot^2 + \csc^2)\sin^2\alpha$  which is more a generalised expression, where  $\alpha$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\alpha = 45^\circ$  and  $\alpha = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\alpha_1 = b_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, a1)|)$ , with:

total thickness per orientation,  $tf_1 = NL \cdot t / \text{NoDir} = 1.016$

$df_v = d$  (figure 11.2, ACI 440) = 370.00

$ffe$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 261735.249$

$bw \cdot d = \sqrt{2} \cdot d^2 / 4 = 80424.772$

-----  
Calculation of Shear Strength at edge 2,  $V_{r2} = 385374.211$

$V_{r2} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_n l \cdot V_{Col0}$

$V_{Col0} = 385374.211$

$k_n l = 1$  (zero step-static loading)

-----  
NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

-----  
 $\alpha = 1$  (normal-weight concrete)

$f'_c = 24.00$ , but  $f'_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.4753320E-011$

$\nu_u = 2.2164346E-030$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 207261.692$

$A_v = \sqrt{2} \cdot A_{stirrup} = 123370.055$

$f_y = 525.00$

$s = 100.00$

$V_s$  is multiplied by  $Col = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin^2 + \cos^2$  is replaced with  $(\cot^2 + \csc^2)\sin^2\alpha$  which is more a generalised expression, where  $\alpha$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\alpha = 45^\circ$  and  $\alpha = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\alpha_1 = b_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, a1)|)$ , with:

total thickness per orientation,  $tf_1 = NL \cdot t / \text{NoDir} = 1.016$

$df_v = d$  (figure 11.2, ACI 440) = 370.00

$ffe$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 261735.249$

$bw \cdot d = \sqrt{2} \cdot d^2 / 4 = 80424.772$

-----  
End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At local axis: 3

Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 2

(Bending local axis: 3)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 24.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 525.00$

Concrete Elasticity,  $E_c = 23025.204$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 656.25$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.72976

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou}, \min >= 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

Stepwise Properties

At local axis: 2

EDGE -A-

Shear Force,  $V_a = 7.8886091E-031$

EDGE -B-

Shear Force,  $V_b = -7.8886091E-031$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 0.00$

-Compression:  $As_c = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{l,ten} = 1017.876$

-Compression:  $As_{l,com} = 1017.876$

-Middle:  $As_{l,mid} = 1017.876$

Calculation of Shear Capacity ratio ,  $V_e/V_r = 0.48361756$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 186373.736$

with  
 $M_{pr1} = \text{Max}(M_{u1+} , M_{u1-}) = 2.7956E+008$   
 $M_{u1+} = 2.7956E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination  
 $M_{u1-} = 2.7956E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination  
 $M_{pr2} = \text{Max}(M_{u2+} , M_{u2-}) = 2.7956E+008$   
 $M_{u2+} = 2.7956E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination  
 $M_{u2-} = 2.7956E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

-----  
Calculation of  $M_{u1+}$   
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$   
 $M_u = 2.7956E+008$

-----  
= 1.06465  
' = 0.94240061  
error of function (3.68), Biskinis Phd = 47223.857  
From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$   
conf. factor  $c = 1.72976$   
 $f_c = 24.00$   
From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$   
 $l_b/d = 1.00$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00133941$   
 $N = 4771.233$   
 $A_c = 125663.706$   
=  $\cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$   
-----

Calculation of ratio  $l_b/d$   
-----

Adequate Lap Length:  $l_b/d \geq 1$   
-----  
-----

Calculation of  $M_{u1-}$   
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$   
 $M_u = 2.7956E+008$

-----  
= 1.06465  
' = 0.94240061  
error of function (3.68), Biskinis Phd = 47223.857  
From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$   
conf. factor  $c = 1.72976$   
 $f_c = 24.00$   
From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$   
 $l_b/d = 1.00$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00133941$   
 $N = 4771.233$   
 $A_c = 125663.706$   
=  $\cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$   
-----

Calculation of ratio lb/d

Adequate Lap Length:  $lb/d \geq 1$

Calculation of  $\mu_{2+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 2.7956E+008$

$$= 1.06465$$

$$' = 0.94240061$$

error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$

conf. factor  $c = 1.72976$

$$f_c = 24.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (lb/d)^{2/3}) = 656.25$

$$lb/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00133941$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \cdot \text{Min}(1, 1.25 \cdot (lb/d)^{2/3}) = 0.56255742$$

Calculation of ratio lb/d

Adequate Lap Length:  $lb/d \geq 1$

Calculation of  $\mu_{2-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 2.7956E+008$

$$= 1.06465$$

$$' = 0.94240061$$

error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$

conf. factor  $c = 1.72976$

$$f_c = 24.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (lb/d)^{2/3}) = 656.25$

$$lb/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00133941$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \cdot \text{Min}(1, 1.25 \cdot (lb/d)^{2/3}) = 0.56255742$$

Calculation of ratio lb/d

Adequate Lap Length:  $lb/d \geq 1$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 385374.211$

-----  
Calculation of Shear Strength at edge 1,  $Vr1 = 385374.211$

$Vr1 = VCol$  ((10.3), ASCE 41-17) =  $kn1 * VCol0$

$VCol0 = 385374.211$

$kn1 = 1$  (zero step-static loading)

-----  
NOTE: In expression (10-3) ' $Vs = Av * fy * d / s$ ' is replaced by ' $Vs + f * Vf$ '  
where  $Vf$  is the contribution of FRPs (11.3), ACI 440).

-----  
 $\rho = 1$  (normal-weight concrete)

$fc' = 24.00$ , but  $fc^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$Mu = 7.3565446E-012$

$Vu = 7.8886091E-031$

$d = 0.8 * D = 320.00$

$Nu = 4771.233$

$Ag = 125663.706$

From (11.5.4.8), ACI 318-14:  $Vs = 207261.692$

$Av = \rho / 2 * A_{stirup} = 123370.055$

$fy = 525.00$

$s = 100.00$

$Vs$  is multiplied by  $Col = 0.00$

$s/d = 0.3125$

$Vf$  ((11-3)-(11.4), ACI 440) =  $194961.134$

$f = 0.95$ , for fully-wrapped sections

$wf/sf = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression,  
where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $Vf(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ ,  
as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = b1 + 90^\circ = 90.00$

$Vf = \text{Min}(|Vf(45, \theta)|, |Vf(-45, \theta)|)$ , with:

total thickness per orientation,  $tf1 = NL * t / NoDir = 1.016$

$dfv = d$  (figure 11.2, ACI 440) =  $370.00$

$ffe$  ((11-5), ACI 440) =  $259.312$

$Ef = 64828.00$

$fe = 0.004$ , from (11.6a), ACI 440

with  $fu = 0.01$

From (11-11), ACI 440:  $Vs + Vf \leq 261735.249$

$bw * d = \rho * d * d / 4 = 80424.772$

-----  
Calculation of Shear Strength at edge 2,  $Vr2 = 385374.211$

$Vr2 = VCol$  ((10.3), ASCE 41-17) =  $kn1 * VCol0$

$VCol0 = 385374.211$

$kn1 = 1$  (zero step-static loading)

-----  
NOTE: In expression (10-3) ' $Vs = Av * fy * d / s$ ' is replaced by ' $Vs + f * Vf$ '  
where  $Vf$  is the contribution of FRPs (11.3), ACI 440).

-----  
 $\rho = 1$  (normal-weight concrete)

$fc' = 24.00$ , but  $fc^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$Mu = 7.3565446E-012$

$Vu = 7.8886091E-031$

$d = 0.8 * D = 320.00$

$Nu = 4771.233$

$Ag = 125663.706$

From (11.5.4.8), ACI 318-14:  $Vs = 207261.692$

$Av = \rho / 2 * A_{stirup} = 123370.055$

$fy = 525.00$

$s = 100.00$

$Vs$  is multiplied by  $Col = 0.00$

$s/d = 0.3125$

$Vf$  ((11-3)-(11.4), ACI 440) =  $194961.134$

$f = 0.95$ , for fully-wrapped sections

$wf/sf = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \theta$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE). This later relation, considered as a function  $V_f(\theta, a_i)$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta_1 = b_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, a_1)|)$ , with:  
total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}$  ((11-5), ACI 440) = 259.312  
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
with  $f_u = 0.01$   
From (11-11), ACI 440:  $V_s + V_f \leq 261735.249$   
 $b_w \cdot d = \frac{V_s \cdot d}{4} = 80424.772$

-----  
-----  
End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At local axis: 2

-----  
-----  
Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1  
At local axis: 3  
Integration Section: (a)  
Section Type: rccs

#### Constant Properties

-----  
Knowledge Factor,  $\gamma = 1.00$   
Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE41-17.  
Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17  
Consequently:  
Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 24.00$   
Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 525.00$   
Concrete Elasticity,  $E_c = 23025.204$   
Steel Elasticity,  $E_s = 200000.00$   
Diameter,  $D = 400.00$   
Cover Thickness,  $c = 25.00$   
Element Length,  $L = 3000.00$   
Primary Member  
Smooth Bars  
Ductile Steel  
With Detailing for Earthquake Resistance (including stirrups closed at  $135^\circ$ )  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Adequate Lap Length ( $l_b / l_d >= 1$ )  
FRP Wrapping Data  
Type: Carbon  
Cured laminate properties (design values)  
Thickness,  $t = 1.016$   
Tensile Strength,  $f_{fu} = 1055.00$   
Tensile Modulus,  $E_f = 64828.00$   
Elongation,  $e_{fu} = 0.01$   
Number of directions,  $\text{NoDir} = 1$   
Fiber orientations,  $b_i: 0.00^\circ$   
Number of layers,  $NL = 1$   
Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

-----  
-----  
Bending Moment,  $M = -8.5847E+006$   
Shear Force,  $V_2 = -2860.289$   
Shear Force,  $V_3 = -2.3879448E-013$   
Axial Force,  $F = -4770.087$   
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $A_{st} = 1272.345$

-Compression:  $As_c = 1781.283$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t,ten} = 1017.876$

-Compression:  $As_{c,com} = 1017.876$

-Middle:  $As_{c,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $Db_L = 18.00$

Existing component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_{,R} = * u = 0.08607968$

$u = y + p = 0.08607968$

- Calculation of  $y$  -

$y = (M_y * L_s / 3) / E_{eff} = 0.03297001$  ((4.29), Biskinis Phd)

$M_y = 2.8606E+008$

$L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 3001.332

From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 8.6803E+012$

factor = 0.30

$A_g = 125663.706$

$f_c' = 24.00$

$N = 4770.087$

$E_c * I_g = 2.8934E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 2.8606E+008$

$y$  ((10a) or (10b)) = 1.6206348E-005

$M_{y\_ten}$  (8a) = 2.8606E+008

$y_{ten}$  (7a) = 73.23937

error of function (7a) = 0.00140752

$M_{y\_com}$  (8b) = 4.0293E+008

$y_{com}$  (7b) = 68.98129

error of function (7b) = -0.00043134

with  $e_y = 0.002625$

$e_{co} = 0.002$

$a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)

$d_1 = 44.00$

$R = 200.00$

$v = 0.00134943$

$N = 4770.087$

$A_c = 125663.706$

= 0.45352339

with  $f_c^*$  ((12.3), ACI 440) = 28.12975

$f_c = 24.00$

$f_l = 1.3173$

$k = 1$

Effective FRP thickness,  $t_f = NL * t * \text{Cos}(b_1) = 1.016$

$e_{fe}$  ((12.5) and (12.7)) = 0.004

$E_f = 64828.00$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

- Calculation of  $p$  -

From table 10-9:  $p = 0.05310967$

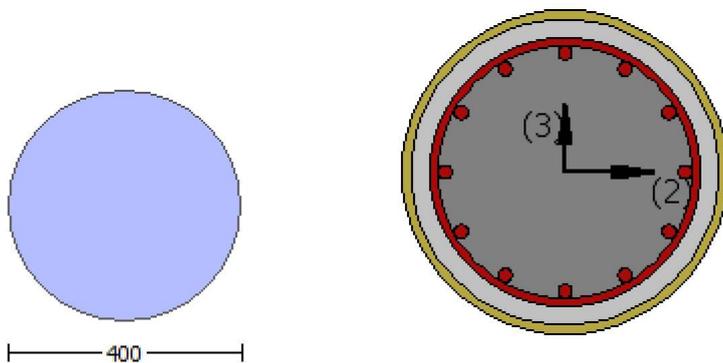
with:

- Columns controlled by inadequate development or splicing along the clear height because  $l_b/l_d < 1$   
 shear control ratio  $V_y E / V_{CoI} E = 0.48361756$   
 $d = 0.00$   
 $s = 0.00$   
 $t = 2 \cdot A_v / (d_c \cdot s) + 4 \cdot t_f / D \cdot (f_{fe} / f_s) = 0.00$   
 $A_v = 78.53982$ , is the area of the circular stirrup  
 $d_c = D - 2 \cdot \text{cover} - \text{Hoop Diameter} = 340.00$   
 The term  $2 \cdot t_f / b_w \cdot (f_{fe} / f_s)$  is implemented to account for FRP contribution  
 where  $f = 2 \cdot t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength  
 All these variables have already been given in Shear control ratio calculation.  
 $NUD = 4770.087$   
 $A_g = 125663.706$   
 $f_{cE} = 24.00$   
 $f_{ytE} = f_{ylE} = 525.00$   
 $\rho_l = \text{Area\_Tot\_Long\_Rein} / (A_g) = 0.0243$   
 $f_{cE} = 24.00$

-----  
 End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1  
 At local axis: 3  
 Integration Section: (a)

## Calculation No. 13

column C1, Floor 1  
 Limit State: Life Safety (data interpolation between analysis steps 1 and 2)  
 Analysis: Uniform +X  
 Check: Shear capacity  $V_{Rd}$   
 Edge: End  
 Local Axis: (2)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1  
 At local axis: 2  
 Integration Section: (b)  
 Section Type: rccs  
 Constant Properties

Knowledge Factor,  $\phi = 1.00$   
 Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE41-17.  
 Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17  
 Consequently:  
 Existing material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 16.00$   
 Existing material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 420.00$   
 Concrete Elasticity,  $E_c = 23025.204$   
 Steel Elasticity,  $E_s = 200000.00$   
 #####  
 Note: Especially for the calculation of  $\mu_y$  for displacement ductility demand,  
 the expected (mean value) strengths are used (7.5.1.3, ASCE41-17) because bending is considered as  
 Deformation-Controlled Action (Table C7-1, ASCE41-17).  
 Existing material: Concrete Strength,  $f_c = f_{cm} = 24.00$   
 Existing material: Steel Strength,  $f_s = f_{sm} = 525.00$   
 #####  
 Diameter,  $D = 400.00$   
 Cover Thickness,  $c = 25.00$   
 Element Length,  $L = 3000.00$   
 Primary Member  
 Smooth Bars  
 Ductile Steel  
 With Detailing for Earthquake Resistance (including stirrups closed at 135°)  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Adequate Lap Length ( $l_o/l_{ou,min} = l_b/l_d \geq 1$ )  
 FRP Wrapping Data  
 Type: Carbon  
 Cured laminate properties (design values)  
 Thickness,  $t = 1.016$   
 Tensile Strength,  $f_{fu} = 1055.00$   
 Tensile Modulus,  $E_f = 64828.00$   
 Elongation,  $e_{fu} = 0.01$   
 Number of directions,  $N_{oDir} = 1$   
 Fiber orientations,  $b_i: 0.00^\circ$   
 Number of layers,  $N_L = 1$   
 Radius of rounding corners,  $R = 40.00$

Stepwise Properties

EDGE -A-  
 Bending Moment,  $M_a = -8.5847E+006$   
 Shear Force,  $V_a = -2860.289$   
 EDGE -B-  
 Bending Moment,  $M_b = 0.07589976$   
 Shear Force,  $V_b = 2860.289$   
 BOTH EDGES  
 Axial Force,  $F = -4770.087$   
 Longitudinal Reinforcement Area Distribution (in 2 divisions)  
   -Tension:  $A_{sl,t} = 0.00$   
   -Compression:  $A_{sl,c} = 3053.628$   
 Longitudinal Reinforcement Area Distribution (in 3 divisions)  
   -Tension:  $A_{sl,ten} = 1017.876$   
   -Compression:  $A_{sl,com} = 1017.876$   
   -Middle:  $A_{sl,mid} = 1017.876$   
 Mean Diameter of Tension Reinforcement,  $Db_{L,ten} = 18.00$

Existing component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = \phi V_n = 314830.111$   
 $V_n ((10.3), ASCE 41-17) = k_n l^* V_{CoI} = 314830.111$   
 $V_{CoI} = 314830.111$   
 $k_n l = 1.00$   
 displacement\_ductility\_demand = 0.04501013

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + \phi * V_f$ '

where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

$f_c' = 16.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 0.07589976$

$V_u = 2860.289$

$d = 0.8 \cdot D = 320.00$

$N_u = 4770.087$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 165809.354$

$A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 123370.055$

$f_y = 420.00$

$s = 100.00$

$V_s$  is multiplied by  $\text{Col} = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression, where  $a$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta)$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = 45^\circ$

$V_f = \text{Min}(|V_f(45^\circ)|, |V_f(-45^\circ)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{\text{Dir}} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 213705.936$

$b_w \cdot d = \sqrt{2} \cdot d^2 / 4 = 80424.772$

displacement ductility demand is calculated as  $\delta / y$

- Calculation of  $\delta / y$  for END B -

for rotation axis 3 and integ. section (b)

From analysis, chord rotation  $\theta = 0.00014833$

$y = (M_y \cdot L_s / 3) / E_{\text{eff}} = 0.00329554$  ((4.29), Biskinis Phd))

$M_y = 2.8606 \text{E} + 008$

$L_s = M/V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) = 300.00

From table 10.5, ASCE 41\_17:  $E_{\text{eff}} = \text{factor} \cdot E_c \cdot I_g = 8.6803 \text{E} + 012$

factor = 0.30

$A_g = 125663.706$

$f_c' = 24.00$

$N = 4770.087$

$E_c \cdot I_g = 2.8934 \text{E} + 013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\delta / y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y_{\text{ten}}}, M_{y_{\text{com}}}) = 2.8606 \text{E} + 008$

$y$  ((10a) or (10b)) = 1.6206348E-005

$M_{y_{\text{ten}}} (8a) = 2.8606 \text{E} + 008$

$y_{\text{ten}} (7a) = 73.23937$

error of function (7a) = 0.00140752

$M_{y_{\text{com}}} (8b) = 4.0293 \text{E} + 008$

$y_{\text{com}} (7b) = 68.98129$

error of function (7b) = -0.00043134

with  $e_y = 0.002625$

$e_{co} = 0.002$   
 $a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00134943$   
 $N = 4770.087$   
 $A_c = 125663.706$   
 $= 0.45352339$   
 with  $f_c^*$  ((12.3), ACI 440) = 28.12975  
 $f_c = 24.00$   
 $f_l = 1.3173$   
 $k = 1$   
 Effective FRP thickness,  $t_f = NL*t*\text{Cos}(b_1) = 1.016$   
 $e_{fe}$  ((12.5) and (12.7)) = 0.004  
 $E_f = 64828.00$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

End Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (b)

## Calculation No. 14

column C1, Floor 1

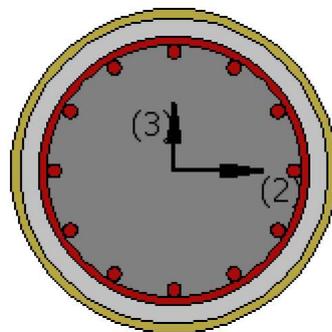
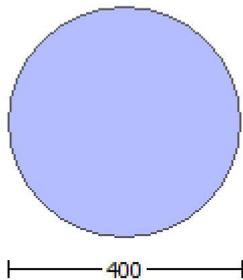
Limit State: Life Safety (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity ( $\theta$ )

Edge: End

Local Axis: (2)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 24.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 525.00$

Concrete Elasticity,  $E_c = 23025.204$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 656.25$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.72976

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou}, \min > = 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

Stepwise Properties

At local axis: 3

EDGE -A-

Shear Force,  $V_a = -2.2164346E-030$

EDGE -B-

Shear Force,  $V_b = 2.2164346E-030$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{sl,t} = 0.00$

-Compression:  $A_{sl,c} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{sl,ten} = 1017.876$

-Compression:  $A_{sl,com} = 1017.876$

-Middle:  $A_{sl,mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.48361756$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 186373.736$

with

$M_{pr1} = \text{Max}(M_{u1+}, M_{u1-}) = 2.7956E+008$

$M_{u1+} = 2.7956E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$M_{u1-} = 2.7956E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment

direction which is defined for the static loading combination

$$M_{pr2} = \text{Max}(M_{u2+}, M_{u2-}) = 2.7956E+008$$

$M_{u2+} = 2.7956E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination

$M_{u2-} = 2.7956E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

-----  
Calculation of  $M_{u1+}$   
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$   
 $M_u = 2.7956E+008$   
-----

$$= 1.06465$$

$$\lambda = 0.94240061$$

error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$

conf. factor  $c = 1.72976$

$$f_c = 24.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$

$$l_b/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00133941$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \lambda \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$$

-----  
Calculation of ratio  $l_b/d$   
-----

Adequate Lap Length:  $l_b/d \geq 1$   
-----  
-----

-----  
Calculation of  $M_{u1-}$   
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$   
 $M_u = 2.7956E+008$   
-----

$$= 1.06465$$

$$\lambda = 0.94240061$$

error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$

conf. factor  $c = 1.72976$

$$f_c = 24.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$

$$l_b/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00133941$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \lambda \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$$

-----  
Calculation of ratio  $l_b/d$   
-----

Adequate Lap Length:  $l_b/d \geq 1$   
-----  
-----

-----  
Calculation of  $M_{u2+}$   
-----

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.7956E+008

= 1.06465  
' = 0.94240061  
error of function (3.68), Biskinis Phd = 47223.857  
From 5A.2, TBDY: fcc = fc\* c = 41.51419  
conf. factor c = 1.72976  
fc = 24.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 656.25  
lb/d = 1.00  
d1 = 44.00  
R = 200.00  
v = 0.00133941  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.56255742

Calculation of ratio lb/d

Adequate Lap Length: lb/d >= 1

Calculation of Mu2-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.7956E+008

= 1.06465  
' = 0.94240061  
error of function (3.68), Biskinis Phd = 47223.857  
From 5A.2, TBDY: fcc = fc\* c = 41.51419  
conf. factor c = 1.72976  
fc = 24.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 656.25  
lb/d = 1.00  
d1 = 44.00  
R = 200.00  
v = 0.00133941  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.56255742

Calculation of ratio lb/d

Adequate Lap Length: lb/d >= 1

Calculation of Shear Strength Vr = Min(Vr1,Vr2) = 385374.211

Calculation of Shear Strength at edge 1, Vr1 = 385374.211  
Vr1 = VCol ((10.3), ASCE 41-17) = knl\*VCol0  
VCol0 = 385374.211  
knl = 1 (zero step-static loading)

NOTE: In expression (10-3) 'Vs = Av\*fy\*d/s' is replaced by 'Vs+ f\*Vf  
where Vf is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)  
 $f_c' = 24.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $\mu_u = 1.4753320E-011$   
 $V_u = 2.2164346E-030$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 207261.692$   
 $A_v = \frac{1}{2} \cdot A_{\text{stirrup}} = 123370.055$   
 $f_y = 525.00$   
 $s = 100.00$   
 $V_s$  is multiplied by  $\text{Col} = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = 45^\circ$   
 $V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, a)|)$ , with:  
 total thickness per orientation,  $t_{f1} = NL \cdot t / N_{\text{Dir}} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}$  ((11-5), ACI 440) = 259.312  
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 261735.249$   
 $b_w \cdot d = \frac{1}{4} \cdot d \cdot d = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 385374.211$   
 $V_{r2} = V_{\text{Col}}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{\text{Col}0}$   
 $V_{\text{Col}0} = 385374.211$   
 $k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ '  
 where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)  
 $f_c' = 24.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $\mu_u = 1.4753320E-011$   
 $V_u = 2.2164346E-030$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 207261.692$   
 $A_v = \frac{1}{2} \cdot A_{\text{stirrup}} = 123370.055$   
 $f_y = 525.00$   
 $s = 100.00$   
 $V_s$  is multiplied by  $\text{Col} = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = 45^\circ$   
 $V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, a)|)$ , with:  
 total thickness per orientation,  $t_{f1} = NL \cdot t / N_{\text{Dir}} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

ffe ((11-5), ACI 440) = 259.312  
Ef = 64828.00  
fe = 0.004, from (11.6a), ACI 440  
with fu = 0.01  
From (11-11), ACI 440: Vs + Vf <= 261735.249  
bw\*d = \*d\*d/4 = 80424.772

-----  
-----  
End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At local axis: 3

-----  
Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At Shear local axis: 2  
(Bending local axis: 3)  
Section Type: rccs

Constant Properties

-----  
Knowledge Factor, = 1.00  
Mean strength values are used for both shear and moment calculations.  
Consequently:  
Existing material of Primary Member: Concrete Strength, fc = fcm = 24.00  
Existing material of Primary Member: Steel Strength, fs = fsm = 525.00  
Concrete Elasticity, Ec = 23025.204  
Steel Elasticity, Es = 200000.00  
#####  
Note: Especially for the calculation of moment strengths,  
the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14  
Existing material: Steel Strength, fs = 1.25\*fsm = 656.25  
#####  
Diameter, D = 400.00  
Cover Thickness, c = 25.00  
Mean Confinement Factor overall section = 1.72976  
Element Length, L = 3000.00  
Primary Member  
Smooth Bars  
Ductile Steel  
With Detailing for Earthquake Resistance (including stirrups closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Adequate Lap Length (lo/lo, min >= 1)  
FRP Wrapping Data  
Type: Carbon  
Cured laminate properties (design values)  
Thickness, t = 1.016  
Tensile Strength, ffu = 1055.00  
Tensile Modulus, Ef = 64828.00  
Elongation, efu = 0.01  
Number of directions, NoDir = 1  
Fiber orientations, bi: 0.00°  
Number of layers, NL = 1  
Radius of rounding corners, R = 40.00

-----  
Stepwise Properties

-----  
At local axis: 2  
EDGE -A-  
Shear Force, Va = 7.8886091E-031  
EDGE -B-  
Shear Force, Vb = -7.8886091E-031  
BOTH EDGES  
Axial Force, F = -4771.233  
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension: Aslt = 0.00

-Compression:  $A_{sc} = 3053.628$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $A_{s,ten} = 1017.876$   
-Compression:  $A_{s,com} = 1017.876$   
-Middle:  $A_{s,mid} = 1017.876$

-----  
-----  
Calculation of Shear Capacity ratio,  $V_e/V_r = 0.48361756$   
Member Controlled by Flexure ( $V_e/V_r < 1$ )  
Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 186373.736$   
with

$M_{pr1} = \text{Max}(M_{u1+}, M_{u1-}) = 2.7956E+008$   
 $M_{u1+} = 2.7956E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination  
 $M_{u1-} = 2.7956E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination  
 $M_{pr2} = \text{Max}(M_{u2+}, M_{u2-}) = 2.7956E+008$   
 $M_{u2+} = 2.7956E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination  
 $M_{u2-} = 2.7956E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

-----  
Calculation of  $M_{u1+}$   
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$   
 $M_u = 2.7956E+008$

-----  
= 1.06465  
' = 0.94240061  
error of function (3.68), Biskinis Phd = 47223.857  
From 5A.2, TBDY:  $f_{cc} = f_c^* c = 41.51419$   
conf. factor  $c = 1.72976$   
 $f_c = 24.00$   
From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y * \text{Min}(1, 1.25 * (l_b/d)^{2/3}) = 656.25$   
 $l_b/d = 1.00$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00133941$   
 $N = 4771.233$   
 $A_c = 125663.706$   
=  $*\text{Min}(1, 1.25 * (l_b/d)^{2/3}) = 0.56255742$

-----  
Calculation of ratio  $l_b/d$   
-----

Adequate Lap Length:  $l_b/d \geq 1$   
-----  
-----

-----  
Calculation of  $M_{u1-}$   
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$   
 $M_u = 2.7956E+008$

-----  
= 1.06465  
' = 0.94240061  
error of function (3.68), Biskinis Phd = 47223.857  
From 5A.2, TBDY:  $f_{cc} = f_c^* c = 41.51419$   
conf. factor  $c = 1.72976$   
 $f_c = 24.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$

$$l_b/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00133941$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of  $\mu_{2+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$$\mu = 2.7956E+008$$

$$= 1.06465$$

$$' = 0.94240061$$

error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$

$$\text{conf. factor } c = 1.72976$$

$$f_c = 24.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$

$$l_b/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00133941$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of  $\mu_{2-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$$\mu = 2.7956E+008$$

$$= 1.06465$$

$$' = 0.94240061$$

error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$

$$\text{conf. factor } c = 1.72976$$

$$f_c = 24.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$

$$l_b/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00133941$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 385374.211$

Calculation of Shear Strength at edge 1,  $V_{r1} = 385374.211$

$V_{r1} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_n l V_{Col0}$

$V_{Col0} = 385374.211$

$k_n l = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v f_y d/s$ ' is replaced by ' $V_s + f^* V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

$f_c' = 24.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 7.3565446E-012$

$V_u = 7.8886091E-031$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 207261.692$

$A_v = \frac{1}{2} A_{stirrup} = 123370.055$

$f_y = 525.00$

$s = 100.00$

$V_s$  is multiplied by  $\text{Col} = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) =  $194961.134$

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression, where  $a$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different cyclic fiber orientation  $a_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = b1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, a1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{oDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) =  $370.00$

$f_{fe}$  ((11-5), ACI 440) =  $259.312$

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 261735.249$

$b_w \cdot d = \frac{1}{4} \cdot d \cdot d = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 385374.211$

$V_{r2} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_n l V_{Col0}$

$V_{Col0} = 385374.211$

$k_n l = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v f_y d/s$ ' is replaced by ' $V_s + f^* V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

$f_c' = 24.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 7.3565446E-012$

$V_u = 7.8886091E-031$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 207261.692$

$A_v = \sqrt{2} * A_{stirrup} = 123370.055$

$f_y = 525.00$

$s = 100.00$

$V_s$  is multiplied by  $Col = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression, where  $a$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta)$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta_1 = b_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, a_1)|)$ , with:

total thickness per orientation,  $t_{f1} = NL * t / \text{NoDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 261735.249$

$b_w * d = \sqrt{4} * d = 80424.772$

-----  
End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At local axis: 2

-----  
Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (b)

Section Type: rccs

Constant Properties

-----  
Knowledge Factor,  $\gamma = 1.00$

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 24.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 525.00$

Concrete Elasticity,  $E_c = 23025.204$

Steel Elasticity,  $E_s = 200000.00$

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at  $135^\circ$ )

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_b/d \geq 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $\text{NoDir} = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

## Stepwise Properties

Bending Moment,  $M = 6.1892434E-011$

Shear Force,  $V2 = 2860.289$

Shear Force,  $V3 = 2.3879448E-013$

Axial Force,  $F = -4770.087$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{st} = 0.00$

-Compression:  $A_{sc} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{st,ten} = 1017.876$

-Compression:  $A_{sc,com} = 1017.876$

-Middle:  $A_{st,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $DbL = 18.00$

Existing component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_{,R} = \gamma + \rho = 0.06958736$

- Calculation of  $\gamma$  -

$\gamma = (M \cdot L_s / 3) / E_{eff} = 0.01647769$  ((4.29), Biskinis Phd))

$M_y = 2.8606E+008$

$L_s = M/V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) = 1500.00

From table 10.5, ASCE 41\_17:  $E_{eff} = factor \cdot E_c \cdot I_g = 8.6803E+012$

factor = 0.30

$A_g = 125663.706$

$f_c' = 24.00$

$N = 4770.087$

$E_c \cdot I_g = 2.8934E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\gamma$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 2.8606E+008$

$\gamma$  ((10a) or (10b)) = 1.6206348E-005

$M_{y\_ten}$  (8a) = 2.8606E+008

$\gamma_{ten}$  (7a) = 73.23937

error of function (7a) = 0.00140752

$M_{y\_com}$  (8b) = 4.0293E+008

$\gamma_{com}$  (7b) = 68.98129

error of function (7b) = -0.00043134

with  $e_y = 0.002625$

$e_{co} = 0.002$

$a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)

$d_1 = 44.00$

$R = 200.00$

$v = 0.00134943$

$N = 4770.087$

$A_c = 125663.706$

= 0.45352339

with  $f_c^*$  ((12.3), ACI 440) = 28.12975

$f_c = 24.00$

$f_l = 1.3173$

$k = 1$

Effective FRP thickness,  $t_f = NL \cdot t \cdot \cos(b_1) = 1.016$

$e_{fe}$  ((12.5) and (12.7)) = 0.004

$E_f = 64828.00$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

- Calculation of  $p$  -

From table 10-9:  $p = 0.05310967$

with:

- Columns controlled by inadequate development or splicing along the clear height because  $l_b/d < 1$

shear control ratio  $V_{yE}/V_{CoIOE} = 0.48361756$

$d = 0.00$

$s = 0.00$

$t = 2 \cdot A_v / (d_c \cdot s) + 4 \cdot t_f / D \cdot (f_{fe} / f_s) = 0.00$

$A_v = 78.53982$ , is the area of the circular stirrup

$d_c = D - 2 \cdot \text{cover} - \text{Hoop Diameter} = 340.00$

The term  $2 \cdot t_f / b_w \cdot (f_{fe} / f_s)$  is implemented to account for FRP contribution

where  $f = 2 \cdot t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength

All these variables have already been given in Shear control ratio calculation.

$N_{UD} = 4770.087$

$A_g = 125663.706$

$f_{cE} = 24.00$

$f_{ytE} = f_{ylE} = 525.00$

$p_l = \text{Area\_Tot\_Long\_Rein} / (A_g) = 0.0243$

$f_{cE} = 24.00$

End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (b)

## Calculation No. 15

column C1, Floor 1

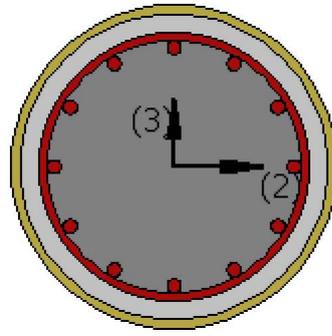
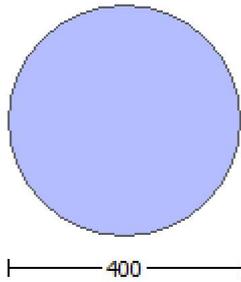
Limit State: Life Safety (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Shear capacity  $V_{Rd}$

Edge: End

Local Axis: (3)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (b)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 16.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 420.00$

Concrete Elasticity,  $E_c = 23025.204$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of  $\gamma$  for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE41-17).

Existing material: Concrete Strength,  $f_c = f_{cm} = 24.00$

Existing material: Steel Strength,  $f_s = f_{sm} = 525.00$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou}, \min = l_b/l_d \geq 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $NoDir = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

Stepwise Properties

EDGE -A-

Bending Moment,  $M_a = 6.5482280E-010$

Shear Force,  $V_a = -2.3879448E-013$

EDGE -B-

Bending Moment,  $M_b = 6.1892434E-011$

Shear Force,  $V_b = 2.3879448E-013$

BOTH EDGES

Axial Force,  $F = -4770.087$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_{t1} = 0.00$

-Compression:  $As_{c1} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t,ten} = 1017.876$

-Compression:  $As_{l,com} = 1017.876$

-Middle:  $As_{l,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $Db_{L,ten} = 18.00$

Existing component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = V_n = 314830.111$

$V_n$  ((10.3), ASCE 41-17) =  $k_n \cdot V_{Col} = 314830.111$

$V_{Col} = 314830.111$

$k_n = 1.00$

$displacement\_ductility\_demand = 0.00$

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

$f'_c = 16.00$ , but  $f'_c \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$M_u = 6.1892434E-011$

$V_u = 2.3879448E-013$

$d = 0.8 \cdot D = 320.00$

$N_u = 4770.087$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 165809.354$

$A_v = A_{stirrup} = 123370.055$

$f_y = 420.00$

$s = 100.00$

$V_s$  is multiplied by  $Col = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) =  $194961.134$

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \theta)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{oDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) =  $370.00$

$f_{fe}$  ((11-5), ACI 440) =  $259.312$

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 213705.936$

$b_w \cdot d = V_u \cdot d / 4 = 80424.772$

$displacement\_ductility\_demand$  is calculated as  $\theta / y$

- Calculation of  $\theta / y$  for END B -

for rotation axis 2 and integ. section (b)

From analysis, chord rotation  $\theta = 8.7620949E-021$

$y = (M_y \cdot L_s / 3) / E_{eff} = 0.01647769$  ((4.29), Biskinis Phd)

$M_y = 2.8606E+008$

$L_s = M/V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) =  $1500.00$

From table 10.5, ASCE 41\_17:  $E_{eff} = factor \cdot E_c \cdot I_g = 8.6803E+012$

factor = 0.30  
Ag = 125663.706  
fc' = 24.00  
N = 4770.087  
Ec\*Ig = 2.8934E+013

-----  
-----  
Calculation of Yielding Moment My

-----  
Calculation of  $\rho_y$  and My according to (7) - (8) in Biskinis and Fardis

-----  
My = Min(My\_ten, My\_com) = 2.8606E+008  
 $\rho_y$  ((10a) or (10b)) = 1.6206348E-005  
My\_ten (8a) = 2.8606E+008  
 $\rho_{y\_ten}$  (7a) = 73.23937  
error of function (7a) = 0.00140752  
My\_com (8b) = 4.0293E+008  
 $\rho_{y\_com}$  (7b) = 68.98129  
error of function (7b) = -0.00043134  
with  $e_y$  = 0.002625  
 $e_{co}$  = 0.002  
 $a_{pl}$  = 0.45 ((9c) in Biskinis and Fardis for FRP Wrap)  
d1 = 44.00  
R = 200.00  
v = 0.00134943  
N = 4770.087  
Ac = 125663.706  
= 0.45352339  
with  $f_c^*$  ((12.3), ACI 440) = 28.12975  
fc = 24.00  
fl = 1.3173  
k = 1  
Effective FRP thickness,  $t_f$  = NL\*t\*cos(b1) = 1.016  
efe ((12.5) and (12.7)) = 0.004  
Ef = 64828.00

-----  
Calculation of ratio lb/d

-----  
Adequate Lap Length: lb/d >= 1

-----  
End Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (b)

-----  
**Calculation No. 16**

column C1, Floor 1

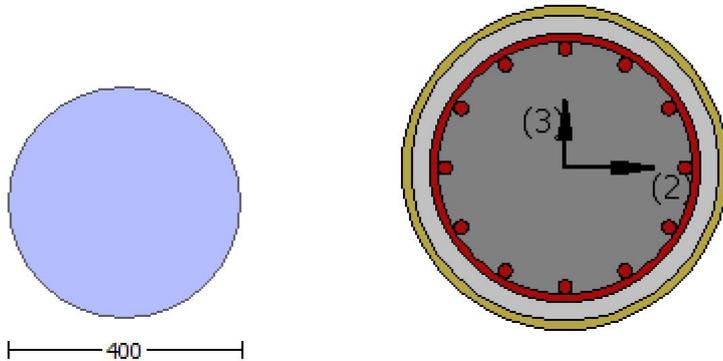
Limit State: Life Safety (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity ( $\theta_u$ )

Edge: End

Local Axis: (3)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 1.00$

Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 24.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 525.00$

Concrete Elasticity,  $E_c = 23025.204$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 656.25$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.72976

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_o/l_{ou, \min} \geq 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $NoDir = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

Stepwise Properties

At local axis: 3

EDGE -A-

Shear Force,  $V_a = -2.2164346E-030$

EDGE -B-

Shear Force,  $V_b = 2.2164346E-030$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 0.00$

-Compression:  $As_c = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t,ten} = 1017.876$

-Compression:  $As_{c,com} = 1017.876$

-Middle:  $As_{mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.48361756$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 186373.736$

with

$M_{pr1} = \text{Max}(Mu_{1+}, Mu_{1-}) = 2.7956E+008$

$Mu_{1+} = 2.7956E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$Mu_{1-} = 2.7956E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(Mu_{2+}, Mu_{2-}) = 2.7956E+008$

$Mu_{2+} = 2.7956E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

$Mu_{2-} = 2.7956E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

Calculation of  $Mu_{1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$

$M_u = 2.7956E+008$

$\phi = 1.06465$

$\phi' = 0.94240061$

error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$

conf. factor  $c = 1.72976$

$f_c = 24.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00133941$

$N = 4771.233$

$A_c = 125663.706$

$\phi' \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of Mu1-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.7956E+008

= 1.06465  
' = 0.94240061  
error of function (3.68), Biskinis Phd = 47223.857  
From 5A.2, TBDY: fcc = fc\* c = 41.51419  
conf. factor c = 1.72976  
fc = 24.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 656.25  
lb/d = 1.00  
d1 = 44.00  
R = 200.00  
v = 0.00133941  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.56255742

Calculation of ratio lb/d

Adequate Lap Length: lb/d >= 1

Calculation of Mu2+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.7956E+008

= 1.06465  
' = 0.94240061  
error of function (3.68), Biskinis Phd = 47223.857  
From 5A.2, TBDY: fcc = fc\* c = 41.51419  
conf. factor c = 1.72976  
fc = 24.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 656.25  
lb/d = 1.00  
d1 = 44.00  
R = 200.00  
v = 0.00133941  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.56255742

Calculation of ratio lb/d

Adequate Lap Length: lb/d >= 1

Calculation of Mu2-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 2.7956E+008

= 1.06465

$\rho = 0.94240061$   
 error of function (3.68), Biskinis Phd = 47223.857  
 From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$   
 conf. factor  $c = 1.72976$   
 $f_c = 24.00$   
 From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$   
 $l_b/d = 1.00$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00133941$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 385374.211$

Calculation of Shear Strength at edge 1,  $V_{r1} = 385374.211$

$V_{r1} = V_{CoI}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{CoI0}$

$V_{CoI0} = 385374.211$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

$f_c' = 24.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 1.4753320E-011$

$\nu_u = 2.2164346E-030$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 207261.692$

$A_v = \rho_s \cdot A_{stirup} = 123370.055$

$f_y = 525.00$

$s = 100.00$

$V_s$  is multiplied by  $\rho_{col} = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression,

where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = 45^\circ + 90^\circ = 135^\circ$

$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \theta)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{oDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_{e1} = 0.004$ , from (11.6a), ACI 440

with  $f_{u1} = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 261735.249$

$b_w \cdot d = \rho_s \cdot d^2 / 4 = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 385374.211$

$V_{r2} = V_{CoI}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{CoI0}$

VCoIO = 385374.211  
knl = 1 (zero step-static loading)

NOTE: In expression (10-3) 'Vs = Av\*fy\*d/s' is replaced by 'Vs+ f\*Vf'  
where Vf is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)  
fc' = 24.00, but  $fc^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
M/Vd = 2.00  
Mu = 1.4753320E-011  
Vu = 2.2164346E-030  
d = 0.8\*D = 320.00  
Nu = 4771.233  
Ag = 125663.706  
From (11.5.4.8), ACI 318-14: Vs = 207261.692  
Av = /2\*A\_stirrup = 123370.055  
fy = 525.00  
s = 100.00  
Vs is multiplied by Col = 0.00  
s/d = 0.3125  
Vf ((11-3)-(11.4), ACI 440) = 194961.134  
f = 0.95, for fully-wrapped sections  
wf/sf = 1 (FRP strips adjacent to one another).  
In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a)\sin a$  which is more a generalised expression,  
where is the angle of the crack direction (see KANEPE).  
This later relation, considered as a function Vf( , ), is implemented for every different fiber orientation ai,  
as well as for 2 crack directions, = 45° and = -45° to take into consideration the cyclic seismic loading.  
orientation 1: 1 = b1 + 90° = 90.00  
Vf = Min(|Vf(45, 1)|, |Vf(-45,a1)|), with:  
total thickness per orientation, tf1 = NL\*t/NoDir = 1.016  
dfv = d (figure 11.2, ACI 440) = 370.00  
ffe ((11-5), ACI 440) = 259.312  
Ef = 64828.00  
fe = 0.004, from (11.6a), ACI 440  
with fu = 0.01  
From (11-11), ACI 440: Vs + Vf <= 261735.249  
bw\*d = \*d\*d/4 = 80424.772

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At local axis: 3

Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At Shear local axis: 2  
(Bending local axis: 3)  
Section Type: rccs

Constant Properties

Knowledge Factor, = 1.00  
Mean strength values are used for both shear and moment calculations.  
Consequently:  
Existing material of Primary Member: Concrete Strength, fc = fcm = 24.00  
Existing material of Primary Member: Steel Strength, fs = fsm = 525.00  
Concrete Elasticity, Ec = 23025.204  
Steel Elasticity, Es = 200000.00  
#####  
Note: Especially for the calculation of moment strengths,  
the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14  
Existing material: Steel Strength, fs = 1.25\*fsm = 656.25  
#####  
Diameter, D = 400.00  
Cover Thickness, c = 25.00  
Mean Confinement Factor overall section = 1.72976

Element Length, L = 3000.00  
 Primary Member  
 Smooth Bars  
 Ductile Steel  
 With Detailing for Earthquake Resistance (including stirrups closed at 135°)  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Adequate Lap Length ( $l_o/l_{ou}, \min \geq 1$ )  
 FRP Wrapping Data  
 Type: Carbon  
 Cured laminate properties (design values)  
 Thickness, t = 1.016  
 Tensile Strength,  $f_{fu} = 1055.00$   
 Tensile Modulus,  $E_f = 64828.00$   
 Elongation,  $e_{fu} = 0.01$   
 Number of directions, NoDir = 1  
 Fiber orientations,  $b_i = 0.00^\circ$   
 Number of layers, NL = 1  
 Radius of rounding corners, R = 40.00

-----  
 Stepwise Properties  
 -----

At local axis: 2  
 EDGE -A-  
 Shear Force,  $V_a = 7.8886091E-031$   
 EDGE -B-  
 Shear Force,  $V_b = -7.8886091E-031$   
 BOTH EDGES  
 Axial Force, F = -4771.233  
 Longitudinal Reinforcement Area Distribution (in 2 divisions)  
   -Tension:  $A_{st} = 0.00$   
   -Compression:  $A_{sc} = 3053.628$   
 Longitudinal Reinforcement Area Distribution (in 3 divisions)  
   -Tension:  $A_{st,ten} = 1017.876$   
   -Compression:  $A_{sl,com} = 1017.876$   
   -Middle:  $A_{sl,mid} = 1017.876$

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 -----  
 Calculation of Shear Capacity ratio ,  $V_e/V_r = 0.48361756$   
 Member Controlled by Flexure ( $V_e/V_r < 1$ )  
 Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 186373.736$   
 with  
 $M_{pr1} = \text{Max}(\mu_{1+}, \mu_{1-}) = 2.7956E+008$   
    $\mu_{1+} = 2.7956E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction  
   which is defined for the static loading combination  
    $\mu_{1-} = 2.7956E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment  
   direction which is defined for the static loading combination  
 $M_{pr2} = \text{Max}(\mu_{2+}, \mu_{2-}) = 2.7956E+008$   
    $\mu_{2+} = 2.7956E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction  
   which is defined for the the static loading combination  
    $\mu_{2-} = 2.7956E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment  
   direction which is defined for the the static loading combination

-----  
 Calculation of  $\mu_{1+}$   
 -----

-----  
 Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$   
 $\mu_u = 2.7956E+008$

-----  
   = 1.06465  
   ' = 0.94240061  
 error of function (3.68), Biskinis Phd = 47223.857  
 From 5A.2, TBDY:  $f_{cc} = f_c^* \quad c = 41.51419$

conf. factor  $c = 1.72976$

$f_c = 24.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00133941$

$N = 4771.233$

$Ac = 125663.706$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of  $\mu_{1-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$\mu = 2.7956E+008$

$= 1.06465$

$' = 0.94240061$

error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$

conf. factor  $c = 1.72976$

$f_c = 24.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00133941$

$N = 4771.233$

$Ac = 125663.706$

$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of  $\mu_{2+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$\mu = 2.7956E+008$

$= 1.06465$

$' = 0.94240061$

error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$

conf. factor  $c = 1.72976$

$f_c = 24.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$

$l_b/d = 1.00$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00133941$

$N = 4771.233$

$Ac = 125663.706$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of  $\mu_2$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 2.7956E+008$

$$= 1.06465$$

$$' = 0.94240061$$

error of function (3.68), Biskinis Phd = 47223.857

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 41.51419$

conf. factor  $c = 1.72976$

$f_c = 24.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 656.25$

$$l_b/d = 1.00$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00133941$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.56255742$$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 385374.211$

Calculation of Shear Strength at edge 1,  $V_{r1} = 385374.211$

$V_{r1} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{Col0}$

$$V_{Col0} = 385374.211$$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

$f_c' = 24.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$$M/d = 2.00$$

$$\mu = 7.3565446E-012$$

$$V_u = 7.8886091E-031$$

$$d = 0.8 \cdot D = 320.00$$

$$N_u = 4771.233$$

$$A_g = 125663.706$$

From (11.5.4.8), ACI 318-14:  $V_s = 207261.692$

$$A_v = \sqrt{2} \cdot A_{stirrup} = 123370.055$$

$$f_y = 525.00$$

$$s = 100.00$$

$V_s$  is multiplied by  $Col = 0.00$

$$s/d = 0.3125$$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression,

where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta)$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = b1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, a1)|)$ , with:

total thickness per orientation,  $t_{f1} = NL * t / \text{NoDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 261735.249$

$b_w * d = \rho * d^2 / 4 = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 385374.211$

$V_{r2} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_{nl} * V_{Col0}$

$V_{Col0} = 385374.211$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\rho = 1$  (normal-weight concrete)

$f_c' = 24.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M / Vd = 2.00$

$\mu_u = 7.3565446E-012$

$\nu_u = 7.8886091E-031$

$d = 0.8 * D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 207261.692$

$A_v = \rho / 2 * A_{stirrup} = 123370.055$

$f_y = 525.00$

$s = 100.00$

$V_s$  is multiplied by  $\rho_{Col} = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections

$w_f / s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin a$  which is more a generalised expression,

where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta)$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = b1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, a1)|)$ , with:

total thickness per orientation,  $t_{f1} = NL * t / \text{NoDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 261735.249$

$b_w * d = \rho * d^2 / 4 = 80424.772$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At local axis: 2

Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (b)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\phi = 1.00$

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 24.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 525.00$

Concrete Elasticity,  $E_c = 23025.204$

Steel Elasticity,  $E_s = 200000.00$

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Adequate Lap Length ( $l_b/l_d \geq 1$ )

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $\epsilon_{fu} = 0.01$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

Stepwise Properties

Bending Moment,  $M = 0.07589976$

Shear Force,  $V_2 = 2860.289$

Shear Force,  $V_3 = 2.3879448E-013$

Axial Force,  $F = -4770.087$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{st} = 0.00$

-Compression:  $A_{sc} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{s,ten} = 1017.876$

-Compression:  $A_{s,com} = 1017.876$

-Middle:  $A_{s,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $D_bL = 18.00$

Existing component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_{,R} = \phi \cdot u = 0.05640521$

$u = y + \phi \cdot p = 0.05640521$

- Calculation of  $y$  -

$y = (M \cdot L_s / 3) / E_{eff} = 0.00329554$  ((4.29), Biskinis Phd)

$M_y = 2.8606E+008$

$L_s = M/V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) = 300.00

From table 10.5, ASCE 41\_17:  $E_{eff} = \text{factor} \cdot E_c \cdot I_g = 8.6803E+012$

factor = 0.30

$A_g = 125663.706$

$f_c' = 24.00$

$N = 4770.087$

$E_c \cdot I_g = 2.8934E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\eta$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 2.8606E+008$   
 $\eta$  ((10a) or (10b)) = 1.6206348E-005  
 $M_{y\_ten}$  (8a) = 2.8606E+008  
 $\eta_{ten}$  (7a) = 73.23937  
error of function (7a) = 0.00140752  
 $M_{y\_com}$  (8b) = 4.0293E+008  
 $\eta_{com}$  (7b) = 68.98129  
error of function (7b) = -0.00043134  
with  $e_y = 0.002625$   
 $e_{co} = 0.002$   
 $a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00134943$   
 $N = 4770.087$   
 $A_c = 125663.706$   
 $\eta = 0.45352339$   
with  $f_c^*$  ((12.3), ACI 440) = 28.12975  
 $f_c = 24.00$   
 $f_l = 1.3173$   
 $k = 1$   
Effective FRP thickness,  $t_f = N L^* t^* \text{Cos}(b_1) = 1.016$   
 $e_{fe}$  ((12.5) and (12.7)) = 0.004  
 $E_f = 64828.00$

Calculation of ratio  $l_b/d$

Adequate Lap Length:  $l_b/d \geq 1$

- Calculation of  $\rho_p$  -

From table 10-9:  $\rho_p = 0.05310967$

with:

- Columns controlled by inadequate development or splicing along the clear height because  $l_b/d < 1$

shear control ratio  $V_y E / V_{CoI} E = 0.48361756$

$d = 0.00$

$s = 0.00$

$t = 2 \cdot A_v / (d_c \cdot s) + 4 \cdot t_f / D \cdot (f_{fe} / f_s) = 0.00$

$A_v = 78.53982$ , is the area of the circular stirrup

$d_c = D - 2 \cdot \text{cover} - \text{Hoop Diameter} = 340.00$

The term  $2 \cdot t_f / b_w \cdot (f_{fe} / f_s)$  is implemented to account for FRP contribution

where  $f = 2 \cdot t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength

All these variables have already been given in Shear control ratio calculation.

$N_{UD} = 4770.087$

$A_g = 125663.706$

$f_c E = 24.00$

$f_{yt} E = f_{yl} E = 525.00$

$\rho_l = \text{Area\_Tot\_Long\_Rein} / (A_g) = 0.0243$

$f_c E = 24.00$

End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (b)