

# Detailed Member Calculations

**Units: N&mm**

**Regulation: ASCE 41-17**

## Calculation No. 1

column C1, Floor 1

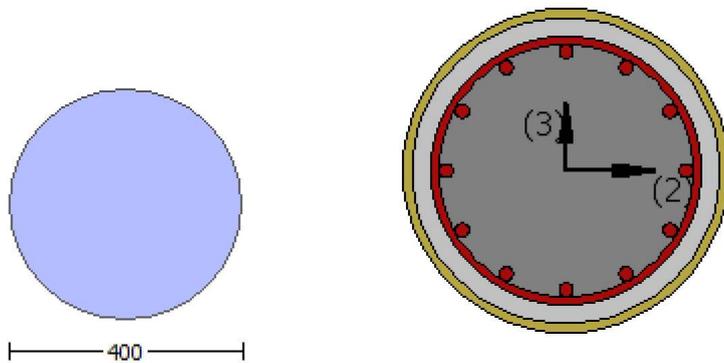
Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Shear capacity VRd

Edge: Start

Local Axis: (2)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (a)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 0.85$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 16.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 400.00$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

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Note: Especially for the calculation of  $\gamma$  for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE41-17).

Existing material: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material: Steel Strength,  $f_s = f_{sm} = 444.4444$

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Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at  $135^\circ$ )

Longitudinal Bars With Ends Lapped Starting at the End Sections

Inadequate Lap Length with  $l_o/l_{o,min} = l_b/l_d = 0.30$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $NoDir = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

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Stepwise Properties

EDGE -A-

Bending Moment,  $M_a = -8.3350E+006$

Shear Force,  $V_a = -2777.007$

EDGE -B-

Bending Moment,  $M_b = 0.07583332$

Shear Force,  $V_b = 2777.007$

BOTH EDGES

Axial Force,  $F = -4770.122$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{sl,t} = 1272.345$

-Compression:  $A_{sl,c} = 1781.283$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{sl,ten} = 1017.876$

-Compression:  $A_{sl,com} = 1017.876$

-Middle:  $A_{sl,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $Db_{L,ten} = 18.00$

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Existing component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = \phi V_n = 224627.823$

$V_n$  ((10.3), ASCE 41-17) =  $k_n l^* V_{CoIO} = 264268.027$

$V_{CoI} = 264268.027$

$k_n l = 1.00$

$displacement\_ductility\_demand = 0.01625615$

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NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + \phi * V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

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 $\phi = 1$  (normal-weight concrete)

$f_c' = 16.00$ , but  $f_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 4.00$

$M_u = 8.3350E+006$

$V_u = 2777.007$

$d = 0.8 * D = 320.00$

$N_u = 4770.122$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 157913.67$   
 $A_v = \frac{1}{2} A_{stirrup} = 123370.055$   
 $f_y = 400.00$   
 $s = 100.00$   
 $V_s$  is multiplied by  $Col = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a)\sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = b1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, a1)|, |V_f(-45, a1)|)$ , with:  
 total thickness per orientation,  $tf1 = NL * t / NoDir = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}$  ((11-5), ACI 440) = 259.312  
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 213705.936$   
 $b_w * d = \frac{1}{4} * d * d = 80424.772$

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 displacement ductility demand is calculated as  $\delta / y$

- Calculation of  $\delta / y$  for END A -  
 for rotation axis 3 and integ. section (a)

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 From analysis, chord rotation  $\theta = 0.00028185$   
 $y = (M_y * L_s / 3) / E_{eff} = 0.01733836$  ((4.29), Biskinis Phd))  
 $M_y = 1.3732E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 3001.447  
 From table 10.5, ASCE 41\_17:  $E_{eff} = \text{factor} * E_c * I_g = 7.9240E+012$   
 factor = 0.30  
 $A_g = 125663.706$   
 $f_c' = 20.00$   
 $N = 4770.122$   
 $E_c * I_g = 2.6413E+013$

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 Calculation of Yielding Moment  $M_y$

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 Calculation of  $\delta / y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

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 $M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 1.3732E+008$   
 $y = 7.5716684E-006$   
 $M_{y\_ten}$  (8c) = 1.3732E+008  
 $\delta_{ten}$  (7c) = 72.23345  
 error of function (7c) = 0.00097594  
 $M_{y\_com}$  (8d) = 3.7493E+008  
 $\delta_{com}$  (7d) = 69.91126  
 error of function (7d) = 0.00301333  
 with ((10.1), ASCE 41-17)  $e_y = \text{Min}(e_y, 1.25 * e_y * (l_b / l_d)^{2/3}) = 0.00222222$   
 $e_{co} = 0.002$   
 $a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00157314$   
 $N = 4770.122$   
 $A_c = 125663.706$   
 ((10.1), ASCE 41-17)  $e_y = \text{Min}(e_y, 1.25 * e_y * (l_b / l_d)^{2/3}) = 0.44758025$

with  $f_c^*$  ((12.3), ACI 440) = 24.12975  
 $f_c = 20.00$   
 $f_l = 1.3173$   
 $k = 1$   
Effective FRP thickness,  $t_f = NL*t*\cos(b1) = 1.016$   
 $e_{fe}$  ((12.5) and (12.7)) = 0.004  
 $f_u = 0.01$   
 $E_f = 64828.00$

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Calculation of ratio  $I_b/I_d$

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Inadequate Lap Length with  $I_b/I_d = 0.30$

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End Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (a)

## Calculation No. 2

column C1, Floor 1

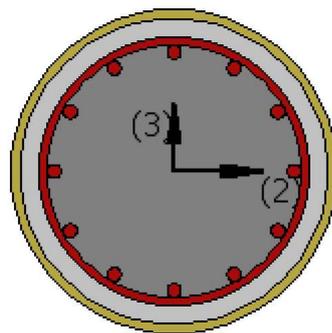
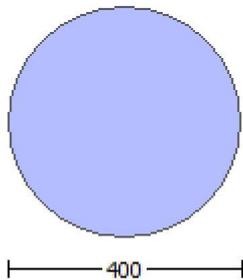
Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity (  $\theta$  )

Edge: Start

Local Axis: (2)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rccs

Constant Properties

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Knowledge Factor,  $\phi = 0.85$

Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.4444$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,  
the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 555.5556$

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Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.84055

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Inadequate Lap Length with  $l_o/l_{o,min} = 0.30$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

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Stepwise Properties

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At local axis: 3

EDGE -A-

Shear Force,  $V_a = -3.7968243E-031$

EDGE -B-

Shear Force,  $V_b = 3.7968243E-031$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{sl,t} = 0.00$

-Compression:  $A_{sl,c} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{sl,ten} = 1017.876$

-Compression:  $A_{sl,com} = 1017.876$

-Middle:  $A_{sl,mid} = 1017.876$

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Calculation of Shear Capacity ratio,  $V_e/V_r = 0.27402927$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 96425.251$

with

$M_{pr1} = \text{Max}(M_{u1+}, M_{u1-}) = 1.4464E+008$

$M_{u1+} = 1.4464E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$M_{u1-} = 1.4464E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(M_{u2+}, M_{u2-}) = 1.4464E+008$

$M_{u2+} = 1.4464E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the static loading combination

$M_{u2-} = 1.4464E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the static loading combination

Calculation of Mu1+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.4464E+008

$$= 0.9250245$$

$$' = 0.82105152$$

error of function (3.68), Biskinis Phd = 31682.903

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$

conf. factor  $c = 1.84055$

$$f_c = 20.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 311.2087$

$$l_b/d = 0.30$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00155946$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.31060654$$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of Mu1-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.4464E+008

$$= 0.9250245$$

$$' = 0.82105152$$

error of function (3.68), Biskinis Phd = 31682.903

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$

conf. factor  $c = 1.84055$

$$f_c = 20.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 311.2087$

$$l_b/d = 0.30$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00155946$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.31060654$$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of Mu2+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.4464E+008

= 0.9250245  
 ' = 0.82105152  
 error of function (3.68), Biskinis Phd = 31682.903  
 From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$   
 conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
 From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 311.2087$   
 $l_b/d = 0.30$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.31060654$

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 Calculation of ratio  $l_b/d$

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 Inadequate Lap Length with  $l_b/d = 0.30$   
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 Calculation of  $\mu_2$ -  
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 Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 1.4464E+008$

-----  
 = 0.9250245  
 ' = 0.82105152  
 error of function (3.68), Biskinis Phd = 31682.903  
 From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$   
 conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
 From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 311.2087$   
 $l_b/d = 0.30$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.31060654$

-----  
 Calculation of ratio  $l_b/d$

-----  
 Inadequate Lap Length with  $l_b/d = 0.30$   
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 Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 351879.387$

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 Calculation of Shear Strength at edge 1,  $V_{r1} = 351879.387$

$V_{r1} = V_{CoI}$  ((10.3), ASCE 41-17) =  $k_n l \cdot V_{CoI}$   
 $V_{CoI} = 351879.387$   
 $k_n l = 1$  (zero step-static loading)

-----  
 NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_{s+} + f \cdot V_f$ '  
 where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

-----  
 = 1 (normal-weight concrete)  
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/d = 2.00$   
 $\mu_u = 9.8466009E-012$   
 $V_u = 3.7968243E-031$

$d = 0.8 \cdot D = 320.00$   
 $Nu = 4771.233$   
 $Ag = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $Vs = 175459.634$   
 $Av = \sqrt{2} \cdot A_{stirrup} = 123370.055$   
 $fy = 444.4444$   
 $s = 100.00$   
 $Vs$  is multiplied by  $Col = 0.00$   
 $s/d = 0.3125$   
 $Vf$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections  
 $wf/sf = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \cot \alpha) \sin \alpha$  which is more a generalised expression,  
 where  $\theta$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $Vf(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = b1 + 90^\circ = 90.00$   
 $Vf = \text{Min}(|Vf(45, \theta)|, |Vf(-45, \theta)|)$ , with:  
 total thickness per orientation,  $tf1 = NL \cdot t / NoDir = 1.016$   
 $dfv = d$  (figure 11.2, ACI 440) = 370.00  
 $ffe$  ((11-5), ACI 440) = 259.312  
 $Ef = 64828.00$   
 $fe = 0.004$ , from (11.6a), ACI 440  
 with  $fu = 0.01$   
 From (11-11), ACI 440:  $Vs + Vf \leq 238930.50$   
 $bw \cdot d = \sqrt{3} \cdot d^2 / 4 = 80424.772$

-----  
 Calculation of Shear Strength at edge 2,  $Vr2 = 351879.387$   
 $Vr2 = VCol$  ((10.3), ASCE 41-17) =  $knl \cdot VCol0$   
 $VCol0 = 351879.387$   
 $knl = 1$  (zero step-static loading)

-----  
 NOTE: In expression (10-3) ' $Vs = Av \cdot fy \cdot d / s$ ' is replaced by ' $Vs + f \cdot Vf$ '  
 where  $Vf$  is the contribution of FRPs (11.3), ACI 440).

-----  
 $\beta = 1$  (normal-weight concrete)  
 $fc' = 20.00$ , but  $fc'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $Mu = 9.8466009E-012$   
 $Vu = 3.7968243E-031$   
 $d = 0.8 \cdot D = 320.00$   
 $Nu = 4771.233$   
 $Ag = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $Vs = 175459.634$   
 $Av = \sqrt{2} \cdot A_{stirrup} = 123370.055$   
 $fy = 444.4444$   
 $s = 100.00$   
 $Vs$  is multiplied by  $Col = 0.00$   
 $s/d = 0.3125$   
 $Vf$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections  
 $wf/sf = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \cot \alpha) \sin \alpha$  which is more a generalised expression,  
 where  $\theta$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $Vf(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = b1 + 90^\circ = 90.00$   
 $Vf = \text{Min}(|Vf(45, \theta)|, |Vf(-45, \theta)|)$ , with:  
 total thickness per orientation,  $tf1 = NL \cdot t / NoDir = 1.016$   
 $dfv = d$  (figure 11.2, ACI 440) = 370.00  
 $ffe$  ((11-5), ACI 440) = 259.312  
 $Ef = 64828.00$   
 $fe = 0.004$ , from (11.6a), ACI 440  
 with  $fu = 0.01$   
 From (11-11), ACI 440:  $Vs + Vf \leq 238930.50$   
 $bw \cdot d = \sqrt{3} \cdot d^2 / 4 = 80424.772$

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End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At local axis: 3  
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Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At Shear local axis: 2  
(Bending local axis: 3)  
Section Type: rccs

Constant Properties

-----  
Knowledge Factor,  $\gamma = 0.85$   
Mean strength values are used for both shear and moment calculations.  
Consequently:  
Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$   
Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.4444$   
Concrete Elasticity,  $E_c = 21019.039$   
Steel Elasticity,  $E_s = 200000.00$   
#####

Note: Especially for the calculation of moment strengths,  
the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14  
Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 555.5556$

#####  
Diameter,  $D = 400.00$   
Cover Thickness,  $c = 25.00$   
Mean Confinement Factor overall section = 1.84055  
Element Length,  $L = 3000.00$   
Primary Member  
Smooth Bars  
Ductile Steel  
With Detailing for Earthquake Resistance (including stirrups closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Inadequate Lap Length with  $l_o/l_{ou, \min} = 0.30$   
FRP Wrapping Data  
Type: Carbon  
Cured laminate properties (design values)  
Thickness,  $t = 1.016$   
Tensile Strength,  $f_{fu} = 1055.00$   
Tensile Modulus,  $E_f = 64828.00$   
Elongation,  $e_{fu} = 0.01$   
Number of directions,  $N_{oDir} = 1$   
Fiber orientations,  $b_i = 0.00^\circ$   
Number of layers,  $N_L = 1$   
Radius of rounding corners,  $R = 40.00$

-----  
Stepwise Properties

-----  
At local axis: 2  
EDGE -A-  
Shear Force,  $V_a = 4.4296389E-033$   
EDGE -B-  
Shear Force,  $V_b = -4.4296389E-033$   
BOTH EDGES  
Axial Force,  $F = -4771.233$   
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $A_{slt} = 0.00$   
-Compression:  $A_{slc} = 3053.628$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $A_{sl, \text{ten}} = 1017.876$   
-Compression:  $A_{sl, \text{com}} = 1017.876$   
-Middle:  $A_{sl, \text{mid}} = 1017.876$   
-----

Calculation of Shear Capacity ratio ,  $V_e/V_r = 0.27402927$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 96425.251$

with

$M_{pr1} = \text{Max}(M_{u1+} , M_{u1-}) = 1.4464E+008$

$M_{u1+} = 1.4464E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$M_{u1-} = 1.4464E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(M_{u2+} , M_{u2-}) = 1.4464E+008$

$M_{u2+} = 1.4464E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

$M_{u2-} = 1.4464E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

Calculation of  $M_{u1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$

$M_u = 1.4464E+008$

$\phi = 0.9250245$

$\phi' = 0.82105152$

error of function (3.68), Biskinis Phd = 31682.903

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$

conf. factor  $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 311.2087$

$l_b/l_d = 0.30$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \phi' \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.31060654$

Calculation of ratio  $l_b/l_d$

Inadequate Lap Length with  $l_b/l_d = 0.30$

Calculation of  $M_{u1-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$

$M_u = 1.4464E+008$

$\phi = 0.9250245$

$\phi' = 0.82105152$

error of function (3.68), Biskinis Phd = 31682.903

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$

conf. factor  $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 311.2087$

$l_b/l_d = 0.30$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$$A_c = 125663.706$$
$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.31060654$$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of  $\mu_{2+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 1.4464E+008$

$$= 0.9250245$$

$$\lambda = 0.82105152$$

error of function (3.68), Biskinis Phd = 31682.903

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$

conf. factor  $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 311.2087$

$$l_b/d = 0.30$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00155946$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.31060654$$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of  $\mu_{2-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 1.4464E+008$

$$= 0.9250245$$

$$\lambda = 0.82105152$$

error of function (3.68), Biskinis Phd = 31682.903

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$

conf. factor  $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 311.2087$

$$l_b/d = 0.30$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00155946$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.31060654$$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

-----  
Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 351879.387$   
-----

Calculation of Shear Strength at edge 1,  $V_{r1} = 351879.387$

$V_{r1} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_{nl} * V_{Col0}$

$V_{Col0} = 351879.387$

$k_{nl} = 1$  (zero step-static loading)

-----  
NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ '  
where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

-----  
 $\lambda = 1$  (normal-weight concrete)

$f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 9.0356326E-012$

$\nu_u = 4.4296389E-033$

$d = 0.8 * D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 175459.634$

$A_v = \lambda / 2 * A_{stirrup} = 123370.055$

$f_y = 444.4444$

$s = 100.00$

$V_s$  is multiplied by  $\lambda_{Col} = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections

$w_f / s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression,  
where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ ,  
as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = b_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \alpha)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L * t / N_{oDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$

$b_w * d = \lambda * d^2 / 4 = 80424.772$   
-----

-----  
Calculation of Shear Strength at edge 2,  $V_{r2} = 351879.387$

$V_{r2} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_{nl} * V_{Col0}$

$V_{Col0} = 351879.387$

$k_{nl} = 1$  (zero step-static loading)

-----  
NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ '  
where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

-----  
 $\lambda = 1$  (normal-weight concrete)

$f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 9.0356326E-012$

$\nu_u = 4.4296389E-033$

$d = 0.8 * D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 175459.634$

$A_v = \lambda / 2 * A_{stirrup} = 123370.055$

$f_y = 444.4444$

$s = 100.00$

$V_s$  is multiplied by  $\lambda_{Col} = 0.00$

$s/d = 0.3125$

Vf ((11-3)-(11.4), ACI 440) = 194961.134

f = 0.95, for fully-wrapped sections

wf/sf = 1 (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function Vf(  $\theta$  ,  $\alpha$  ), is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = b1 + 90^\circ = 90.00$

Vf = Min(|Vf(45,  $\theta$ )|, |Vf(-45,  $\theta$ )|), with:

total thickness per orientation,  $tf1 = NL * t / NoDir = 1.016$

dfv = d (figure 11.2, ACI 440) = 370.00

ffe ((11-5), ACI 440) = 259.312

Ef = 64828.00

fe = 0.004, from (11.6a), ACI 440

with fu = 0.01

From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$

$b_w * d = \frac{V_s * d}{4} = 80424.772$

-----  
End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At local axis: 2  
-----

Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (a)

Section Type: rccs

Constant Properties  
-----

Knowledge Factor,  $\phi = 0.85$

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.4444$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at  $135^\circ$ )

Longitudinal Bars With Ends Lapped Starting at the End Sections

Inadequate Lap Length with  $l_b / l_d = 0.30$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $ff_u = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $ef_u = 0.01$

Number of directions,  $NoDir = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$   
-----

Stepwise Properties  
-----

Bending Moment,  $M = 2.7525396E-011$

Shear Force,  $V_2 = -2777.007$

Shear Force,  $V_3 = -1.4226804E-014$

Axial Force,  $F = -4770.122$   
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $As_t = 1272.345$   
-Compression:  $As_c = 1781.283$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $As_{ten} = 1017.876$   
-Compression:  $As_{com} = 1017.876$   
-Middle:  $As_{mid} = 1017.876$   
Mean Diameter of Tension Reinforcement,  $Db_L = 18.00$

Existing component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_R = \alpha u = 0.00736525$   
 $u = y + p = 0.008665$

- Calculation of  $y$  -

$y = (M_y \cdot L_s / 3) / E_{eff} = 0.008665$  ((4.29), Biskinis Phd)  
 $M_y = 1.3732E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) = 1500.00  
From table 10.5, ASCE 41\_17:  $E_{eff} = factor \cdot E_c \cdot I_g = 7.9240E+012$   
factor = 0.30  
 $A_g = 125663.706$   
 $f_c' = 20.00$   
 $N = 4770.122$   
 $E_c \cdot I_g = 2.6413E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \min(M_{y\_ten}, M_{y\_com}) = 1.3732E+008$   
 $y = 7.5716684E-006$   
 $M_{y\_ten}$  (8c) =  $1.3732E+008$   
 $_{ten}$  (7c) = 72.23345  
error of function (7c) = 0.00097594  
 $M_{y\_com}$  (8d) =  $3.7493E+008$   
 $_{com}$  (7d) = 69.91126  
error of function (7d) = 0.00301333  
with ((10.1), ASCE 41-17)  $e_y = \min(e_y, 1.25 \cdot e_y \cdot (l_b / l_d)^{2/3}) = 0.00222222$   
 $e_{co} = 0.002$   
 $a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00157314$   
 $N = 4770.122$   
 $A_c = 125663.706$   
((10.1), ASCE 41-17) =  $\min(, 1.25 \cdot \cdot (l_b / l_d)^{2/3}) = 0.44758025$   
with  $f_c' \cdot$  ((12.3), ACI 440) = 24.12975  
 $f_c = 20.00$   
 $f_l = 1.3173$   
 $k = 1$   
Effective FRP thickness,  $t_f = NL \cdot t \cdot \cos(b_1) = 1.016$   
 $e_{fe}$  ((12.5) and (12.7)) = 0.004  
 $f_u = 0.01$   
 $E_f = 64828.00$

Calculation of ratio  $l_b / l_d$

Inadequate Lap Length with  $l_b / l_d = 0.30$

- Calculation of  $\rho$  -

From table 10-9:  $\rho = 0.00$

with:

- Columns not controlled by inadequate development or splicing along the clear height because  $l_b/l_d \geq 1$

shear control ratio  $V_{yE}/V_{CoIE} = 0.27402927$

$d = 0.00$

$s = 0.00$

$t = 2 \cdot A_v / (d_c \cdot s) + 4 \cdot t_f / D \cdot (f_{fe} / f_s) = 0.00$

$A_v = 78.53982$ , is the area of the circular stirrup

$d_c = D - 2 \cdot \text{cover} - \text{Hoop Diameter} = 340.00$

The term  $2 \cdot t_f / bw \cdot (f_{fe} / f_s)$  is implemented to account for FRP contribution

where  $f = 2 \cdot t_f / bw$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength

All these variables have already been given in Shear control ratio calculation.

$NUD = 4770.122$

$Ag = 125663.706$

$f_{cE} = 20.00$

$f_{yE} = f_{yI} = 444.4444$

$\rho_l = \text{Area\_Tot\_Long\_Rein} / (Ag) = 0.0243$

$f_{cE} = 20.00$

End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (a)

### Calculation No. 3

column C1, Floor 1

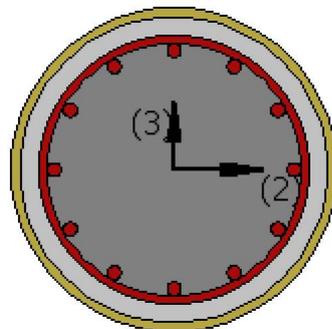
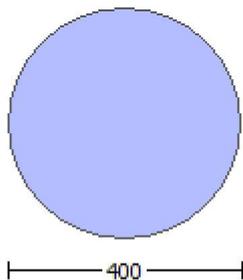
Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Shear capacity  $V_{Rd}$

Edge: Start

Local Axis: (3)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 3  
Integration Section: (a)  
Section Type: rccs

Constant Properties

-----  
Knowledge Factor,  $\gamma = 0.85$   
Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE41-17.  
Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17  
Consequently:  
Existing material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 16.00$   
Existing material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 400.00$   
Concrete Elasticity,  $E_c = 21019.039$   
Steel Elasticity,  $E_s = 200000.00$   
#####  
Note: Especially for the calculation of  $\gamma$  for displacement ductility demand,  
the expected (mean value) strengths are used (7.5.1.3, ASCE41-17) because bending is considered as  
Deformation-Controlled Action (Table C7-1, ASCE41-17).  
Existing material: Concrete Strength,  $f_c = f_{cm} = 20.00$   
Existing material: Steel Strength,  $f_s = f_{sm} = 444.4444$   
#####  
Diameter,  $D = 400.00$   
Cover Thickness,  $c = 25.00$   
Element Length,  $L = 3000.00$   
Primary Member  
Smooth Bars  
Ductile Steel  
With Detailing for Earthquake Resistance (including stirrups closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Inadequate Lap Length with  $l_o/l_{ou,min} = l_b/d = 0.30$   
FRP Wrapping Data  
Type: Carbon  
Cured laminate properties (design values)  
Thickness,  $t = 1.016$   
Tensile Strength,  $f_{fu} = 1055.00$   
Tensile Modulus,  $E_f = 64828.00$   
Elongation,  $e_{fu} = 0.01$   
Number of directions,  $NoDir = 1$   
Fiber orientations,  $bi: 0.00^\circ$   
Number of layers,  $NL = 1$   
Radius of rounding corners,  $R = 40.00$   
-----

Stepwise Properties

-----  
EDGE -A-  
Bending Moment,  $M_a = 2.7525396E-011$   
Shear Force,  $V_a = -1.4226804E-014$   
EDGE -B-  
Bending Moment,  $M_b = 1.5086324E-011$   
Shear Force,  $V_b = 1.4226804E-014$   
BOTH EDGES  
Axial Force,  $F = -4770.122$   
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $As_t = 1272.345$   
-Compression:  $As_c = 1781.283$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $As_{t,ten} = 1017.876$   
-Compression:  $As_{c,com} = 1017.876$   
-Middle:  $As_{l,mid} = 1017.876$   
Mean Diameter of Tension Reinforcement,  $Db_{L,ten} = 18.00$   
-----

Existing component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = \phi V_n = 267605.60$   
 $V_n$  ((10.3), ASCE 41-17) =  $k_n \phi V_{CoI0} = 314830.118$

VCol = 314830.118  
knl = 1.00  
displacement\_ductility\_demand = 0.00

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ '  
where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)  
 $f_c' = 16.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $M_u = 2.7525396E-011$   
 $V_u = 1.4226804E-014$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4770.122$   
 $A_g = 125663.706$   
From (11.5.4.8), ACI 318-14:  $V_s = 157913.67$   
 $A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 123370.055$   
 $f_y = 400.00$   
 $s = 100.00$   
 $V_s$  is multiplied by  $\text{Col} = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \cot \alpha) \sin \alpha$  which is more a generalised expression,  
where  $\theta$  is the angle of the crack direction (see KANEPE).  
This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ ,  
as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
orientation 1:  $\theta = 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \theta)|)$ , with:  
total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{\text{Dir}} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}$  ((11-5), ACI 440) = 259.312  
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
with  $f_u = 0.01$   
From (11-11), ACI 440:  $V_s + V_f \leq 213705.936$   
 $b_w \cdot d = \sqrt{2} \cdot d \cdot d / 4 = 80424.772$

displacement\_ductility\_demand is calculated as  $\phi / y$

- Calculation of  $\phi / y$  for END A -  
for rotation axis 2 and integ. section (a)

From analysis, chord rotation  $\theta = 4.7943297E-021$   
 $y = (M_y \cdot L_s / 3) / E_{\text{eff}} = 0.008665$  ((4.29), Biskinis Phd))  
 $M_y = 1.3732E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) = 1500.00  
From table 10.5, ASCE 41\_17:  $E_{\text{eff}} = \text{factor} \cdot E_c \cdot I_g = 7.9240E+012$   
factor = 0.30  
 $A_g = 125663.706$   
 $f_c' = 20.00$   
 $N = 4770.122$   
 $E_c \cdot I_g = 2.6413E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\phi$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 1.3732E+008$   
 $y = 7.5716684E-006$   
 $M_{y\_ten}$  (8c) = 1.3732E+008  
 $M_{y\_ten}$  (7c) = 72.23345

error of function (7c) = 0.00097594  
My\_com (8d) = 3.7493E+008  
\_com (7d) = 69.91126  
error of function (7d) = 0.00301333  
with ((10.1), ASCE 41-17)  $e_y = \text{Min}(e_y, 1.25 * e_y * (l_b/l_d)^{2/3}) = 0.00222222$   
eco = 0.002  
apl = 0.45 ((9c) in Biskinis and Fardis for FRP Wrap)  
d1 = 44.00  
R = 200.00  
v = 0.00157314  
N = 4770.122  
Ac = 125663.706  
((10.1), ASCE 41-17) =  $\text{Min}( , 1.25 * (l_b/l_d)^{2/3}) = 0.44758025$   
with  $f_c^*$  ((12.3), ACI 440) = 24.12975  
fc = 20.00  
fl = 1.3173  
k = 1  
Effective FRP thickness,  $t_f = NL * t * \text{Cos}(b1) = 1.016$   
efe ((12.5) and (12.7)) = 0.004  
fu = 0.01  
Ef = 64828.00

-----  
-----  
Calculation of ratio  $l_b/l_d$

-----  
Inadequate Lap Length with  $l_b/l_d = 0.30$

-----  
End Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (a)  
-----

## Calculation No. 4

column C1, Floor 1

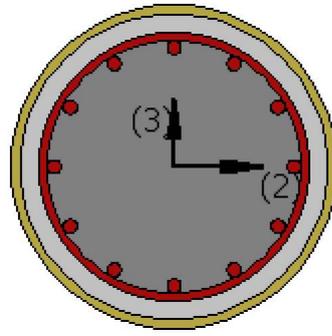
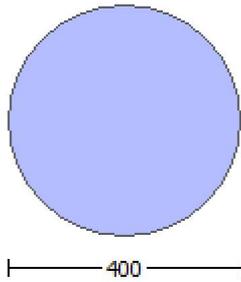
Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity ( $\theta_u$ )

Edge: Start

Local Axis: (3)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 3  
 (Bending local axis: 2)  
 Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 0.85$   
 Mean strength values are used for both shear and moment calculations.  
 Consequently:  
 Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$   
 Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.4444$   
 Concrete Elasticity,  $E_c = 21019.039$   
 Steel Elasticity,  $E_s = 200000.00$

#####  
 Note: Especially for the calculation of moment strengths,  
 the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14  
 Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 555.5556$

#####  
 Diameter,  $D = 400.00$   
 Cover Thickness,  $c = 25.00$   
 Mean Confinement Factor overall section = 1.84055  
 Element Length,  $L = 3000.00$   
 Primary Member  
 Smooth Bars  
 Ductile Steel  
 With Detailing for Earthquake Resistance (including stirrups closed at 135°)  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Inadequate Lap Length with  $l_o/l_{ou, \min} = 0.30$   
 FRP Wrapping Data  
 Type: Carbon  
 Cured laminate properties (design values)  
 Thickness,  $t = 1.016$   
 Tensile Strength,  $f_{fu} = 1055.00$   
 Tensile Modulus,  $E_f = 64828.00$   
 Elongation,  $e_{fu} = 0.01$   
 Number of directions,  $N_{oDir} = 1$   
 Fiber orientations,  $b_i: 0.00^\circ$   
 Number of layers,  $N_L = 1$   
 Radius of rounding corners,  $R = 40.00$

Stepwise Properties

At local axis: 3  
 EDGE -A-  
 Shear Force,  $V_a = -3.7968243E-031$   
 EDGE -B-  
 Shear Force,  $V_b = 3.7968243E-031$   
 BOTH EDGES  
 Axial Force,  $F = -4771.233$   
 Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension: Aslt = 0.00

-Compression: Aslc = 3053.628

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension: Asl,ten = 1017.876

-Compression: Asl,com = 1017.876

-Middle: Asl,mid = 1017.876

Calculation of Shear Capacity ratio ,  $V_e/V_r = 0.27402927$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 96425.251$

with

$M_{pr1} = \text{Max}(M_{u1+} , M_{u1-}) = 1.4464E+008$

$M_{u1+} = 1.4464E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$M_{u1-} = 1.4464E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(M_{u2+} , M_{u2-}) = 1.4464E+008$

$M_{u2+} = 1.4464E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

$M_{u2-} = 1.4464E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

Calculation of  $M_{u1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$

$M_u = 1.4464E+008$

$\phi = 0.9250245$

$\lambda = 0.82105152$

error of function (3.68), Biskinis Phd = 31682.903

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$

conf. factor  $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 311.2087$

$l_b/d = 0.30$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$\phi \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.31060654$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of  $M_{u1-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$

$M_u = 1.4464E+008$

$\phi = 0.9250245$

$\lambda = 0.82105152$

error of function (3.68), Biskinis Phd = 31682.903

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$

conf. factor  $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 311.2087$

$l_b/d = 0.30$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.31060654$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of  $\mu_{2+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$\mu = 1.4464E+008$

$\phi = 0.9250245$

$\phi' = 0.82105152$

error of function (3.68), Biskinis Phd = 31682.903

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$

conf. factor  $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 311.2087$

$l_b/d = 0.30$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.31060654$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of  $\mu_{2-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$\mu = 1.4464E+008$

$\phi = 0.9250245$

$\phi' = 0.82105152$

error of function (3.68), Biskinis Phd = 31682.903

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$

conf. factor  $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 311.2087$

$l_b/d = 0.30$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.31060654$

Calculation of ratio  $l_b/l_d$

Inadequate Lap Length with  $l_b/l_d = 0.30$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 351879.387$

Calculation of Shear Strength at edge 1,  $V_{r1} = 351879.387$

$V_{r1} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_n l V_{Col0}$

$V_{Col0} = 351879.387$

$k_n l = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v f_y d/s$ ' is replaced by ' $V_s + f^* V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\lambda = 1$  (normal-weight concrete)

$f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$M_u = 9.8466009E-012$

$V_u = 3.7968243E-031$

$d = 0.8 * D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 175459.634$

$A_v = \lambda / 2 * A_{stirrup} = 123370.055$

$f_y = 444.4444$

$s = 100.00$

$V_s$  is multiplied by  $\lambda_{Col} = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) =  $194961.134$

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = 45^\circ + 90^\circ = 135^\circ$

$V_f = \text{Min}(|V_f(45^\circ, \alpha_1)|, |V_f(-45^\circ, \alpha_1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L * t / N_{Dir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) =  $370.00$

$f_{fe}$  ((11-5), ACI 440) =  $259.312$

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$

$b_w * d = \lambda * d^2 / 4 = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 351879.387$

$V_{r2} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_n l V_{Col0}$

$V_{Col0} = 351879.387$

$k_n l = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v f_y d/s$ ' is replaced by ' $V_s + f^* V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\lambda = 1$  (normal-weight concrete)

$f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$M_u = 9.8466009E-012$

$V_u = 3.7968243E-031$

$d = 0.8 * D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 175459.634$   
 $A_v = \sqrt{2} * A_{stirrup} = 123370.055$   
 $f_y = 444.4444$   
 $s = 100.00$   
 $V_s$  is multiplied by  $Col = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin^2 + \cos^2$  is replaced with  $(\cot^2 + \csc^2)\sin^2\alpha$  which is more a generalised expression,  
 where  $\alpha$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ ,  
 as well as for 2 crack directions,  $\alpha = 45^\circ$  and  $\alpha = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\alpha_1 = \alpha_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \alpha_1)|, |V_f(-45, \alpha_1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = NL * t / \text{NoDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}$  ((11-5), ACI 440) = 259.312  
 $E_f = 64828.00$   
 $f_{e} = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $b_w * d = A_v * d / 4 = 80424.772$

-----  
 End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At local axis: 3

-----  
 Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At Shear local axis: 2  
 (Bending local axis: 3)  
 Section Type: rccs

Constant Properties

-----  
 Knowledge Factor,  $\phi = 0.85$   
 Mean strength values are used for both shear and moment calculations.  
 Consequently:  
 Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$   
 Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.4444$   
 Concrete Elasticity,  $E_c = 21019.039$   
 Steel Elasticity,  $E_s = 200000.00$   
 #####  
 Note: Especially for the calculation of moment strengths,  
 the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14  
 Existing material: Steel Strength,  $f_s = 1.25 * f_{sm} = 555.5556$   
 #####  
 Diameter,  $D = 400.00$   
 Cover Thickness,  $c = 25.00$   
 Mean Confinement Factor overall section = 1.84055  
 Element Length,  $L = 3000.00$   
 Primary Member  
 Smooth Bars  
 Ductile Steel  
 With Detailing for Earthquake Resistance (including stirrups closed at 135°)  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Inadequate Lap Length with  $l_o/l_{ou, \min} = 0.30$   
 FRP Wrapping Data  
 Type: Carbon  
 Cured laminate properties (design values)  
 Thickness,  $t = 1.016$   
 Tensile Strength,  $f_{fu} = 1055.00$   
 Tensile Modulus,  $E_f = 64828.00$

Elongation,  $\epsilon_{fu} = 0.01$   
Number of directions, NoDir = 1  
Fiber orientations,  $b_i = 0.00^\circ$   
Number of layers, NL = 1  
Radius of rounding corners, R = 40.00

-----  
Stepwise Properties  
-----

At local axis: 2  
EDGE -A-  
Shear Force,  $V_a = 4.4296389E-033$   
EDGE -B-  
Shear Force,  $V_b = -4.4296389E-033$   
BOTH EDGES  
Axial Force,  $F = -4771.233$   
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $A_{sl,t} = 0.00$   
-Compression:  $A_{sl,c} = 3053.628$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $A_{sl,ten} = 1017.876$   
-Compression:  $A_{sl,com} = 1017.876$   
-Middle:  $A_{sl,mid} = 1017.876$   
-----  
-----

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.27402927$   
Member Controlled by Flexure ( $V_e/V_r < 1$ )  
Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 96425.251$   
with  
 $M_{pr1} = \text{Max}(\mu_{u1+}, \mu_{u1-}) = 1.4464E+008$   
 $\mu_{u1+} = 1.4464E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination  
 $\mu_{u1-} = 1.4464E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination  
 $M_{pr2} = \text{Max}(\mu_{u2+}, \mu_{u2-}) = 1.4464E+008$   
 $\mu_{u2+} = 1.4464E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination  
 $\mu_{u2-} = 1.4464E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

-----  
Calculation of  $\mu_{u1+}$   
-----  
-----

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$   
 $\mu_u = 1.4464E+008$

-----  
= 0.9250245  
' = 0.82105152  
error of function (3.68), Biskinis Phd = 31682.903  
From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 311.2087$   
 $l_b/d = 0.30$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
=  $\text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.31060654$   
-----

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/l_d = 0.30$

Calculation of  $\mu_{1-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 1.4464E+008$

$$= 0.9250245$$

$$' = 0.82105152$$

error of function (3.68), Biskinis Phd = 31682.903

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$

$$\text{conf. factor } c = 1.84055$$

$$f_c = 20.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 311.2087$

$$l_b/l_d = 0.30$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00155946$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.31060654$$

Calculation of ratio  $l_b/l_d$

Inadequate Lap Length with  $l_b/l_d = 0.30$

Calculation of  $\mu_{2+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 1.4464E+008$

$$= 0.9250245$$

$$' = 0.82105152$$

error of function (3.68), Biskinis Phd = 31682.903

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$

$$\text{conf. factor } c = 1.84055$$

$$f_c = 20.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 311.2087$

$$l_b/l_d = 0.30$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00155946$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.31060654$$

Calculation of ratio  $l_b/l_d$

Inadequate Lap Length with  $l_b/l_d = 0.30$

Calculation of  $\mu_{2-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.4464E+008

$$= 0.9250245$$

$$' = 0.82105152$$

error of function (3.68), Biskinis Phd = 31682.903

From 5A.2, TBDY:  $f_{cc} = f_c' \cdot c = 36.81095$

$$\text{conf. factor } c = 1.84055$$

$$f_c = 20.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 311.2087$

$$l_b/d = 0.30$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00155946$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.31060654$$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 351879.387$

Calculation of Shear Strength at edge 1,  $V_{r1} = 351879.387$

$V_{r1} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{Col0}$

$$V_{Col0} = 351879.387$$

$$k_{nl} = 1 \text{ (zero step-static loading)}$$

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

$f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)

$$M/d = 2.00$$

$$\text{Mu} = 9.0356326\text{E-}012$$

$$\text{Vu} = 4.4296389\text{E-}033$$

$$d = 0.8 \cdot D = 320.00$$

$$N_u = 4771.233$$

$$A_g = 125663.706$$

From (11.5.4.8), ACI 318-14:  $V_s = 175459.634$

$$A_v = \frac{1}{2} \cdot A_{\text{stirrup}} = 123370.055$$

$$f_y = 444.4444$$

$$s = 100.00$$

$V_s$  is multiplied by  $\text{Col} = 0.00$

$$s/d = 0.3125$$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$$f = 0.95, \text{ for fully-wrapped sections}$$

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = b_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \theta)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{\text{Dir}} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$$E_f = 64828.00$$

$$f_e = 0.004, \text{ from (11.6a), ACI 440}$$

$$\text{with } f_u = 0.01$$

From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$

$$b_w \cdot d = \frac{1}{4} \cdot d^2 = 80424.772$$

Calculation of Shear Strength at edge 2,  $V_{r2} = 351879.387$

$V_{r2} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_n l * V_{Col0}$

$V_{Col0} = 351879.387$

$k_n l = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\lambda = 1$  (normal-weight concrete)

$f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M / V d = 2.00$

$\mu_u = 9.0356326E-012$

$\nu_u = 4.4296389E-033$

$d = 0.8 * D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 175459.634$

$A_v = \sqrt{2} * A_{stirrup} = 123370.055$

$f_y = 444.4444$

$s = 100.00$

$V_s$  is multiplied by  $\lambda_{Col} = 0.00$

$s / d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) =  $194961.134$

$f = 0.95$ , for fully-wrapped sections

$w_f / s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression,

where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = 45^\circ$  and  $\alpha = 90^\circ$

$V_f = \text{Min}(|V_f(45, 90)|, |V_f(-45, 90)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L * t / N_{oDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) =  $370.00$

$f_{fe}$  ((11-5), ACI 440) =  $259.312$

$E_f = 64828.00$

$f_{e} = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$

$b_w * d = \lambda * d * d / 4 = 80424.772$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At local axis: 2

Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (a)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\lambda = 0.85$

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.4444$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member  
Smooth Bars  
Ductile Steel  
With Detailing for Earthquake Resistance (including stirrups closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Inadequate Lap Length with  $l_b/l_d = 0.30$   
FRP Wrapping Data  
Type: Carbon  
Cured laminate properties (design values)  
Thickness,  $t = 1.016$   
Tensile Strength,  $f_{fu} = 1055.00$   
Tensile Modulus,  $E_f = 64828.00$   
Elongation,  $e_{fu} = 0.01$   
Number of directions,  $NoDir = 1$   
Fiber orientations,  $bi = 0.00^\circ$   
Number of layers,  $NL = 1$   
Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

Bending Moment,  $M = -8.3350E+006$   
Shear Force,  $V2 = -2777.007$   
Shear Force,  $V3 = -1.4226804E-014$   
Axial Force,  $F = -4770.122$   
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $As_t = 1272.345$   
-Compression:  $As_c = 1781.283$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $As_{ten} = 1017.876$   
-Compression:  $As_{com} = 1017.876$   
-Middle:  $As_{mid} = 1017.876$   
Mean Diameter of Tension Reinforcement,  $Db_L = 18.00$

Existing component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_{R} = \alpha u = 0.01473761$   
 $u = \gamma + \rho = 0.01733836$

- Calculation of  $\gamma$  -

$\gamma = (M \cdot L_s / 3) / E_{eff} = 0.01733836$  ((4.29), Biskinis Phd))  
 $M_y = 1.3732E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) = 3001.447  
From table 10.5, ASCE 41\_17:  $E_{eff} = factor \cdot E_c \cdot I_g = 7.9240E+012$   
factor = 0.30  
 $A_g = 125663.706$   
 $f_c' = 20.00$   
 $N = 4770.122$   
 $E_c \cdot I_g = 2.6413E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\gamma$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \min(M_{y\_ten}, M_{y\_com}) = 1.3732E+008$   
 $\gamma = 7.5716684E-006$   
 $M_{y\_ten} (8c) = 1.3732E+008$   
 $\gamma_{ten} (7c) = 72.23345$   
error of function (7c) = 0.00097594  
 $M_{y\_com} (8d) = 3.7493E+008$   
 $\gamma_{com} (7d) = 69.91126$   
error of function (7d) = 0.00301333

with ((10.1), ASCE 41-17)  $e_y = \text{Min}(e_y, 1.25 * e_y * (l_b / l_d)^{2/3}) = 0.00222222$

$e_{co} = 0.002$

$a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)

$d_1 = 44.00$

$R = 200.00$

$v = 0.00157314$

$N = 4770.122$

$A_c = 125663.706$

((10.1), ASCE 41-17)  $= \text{Min}( , 1.25 * (l_b / l_d)^{2/3}) = 0.44758025$

with  $f_c^*$  ((12.3), ACI 440) = 24.12975

$f_c = 20.00$

$f_l = 1.3173$

$k = 1$

Effective FRP thickness,  $t_f = N L * t * \text{Cos}(b_1) = 1.016$

$e_{fe}$  ((12.5) and (12.7)) = 0.004

$f_u = 0.01$

$E_f = 64828.00$

-----  
-----  
Calculation of ratio  $l_b / l_d$

-----  
-----  
Inadequate Lap Length with  $l_b / l_d = 0.30$

-----  
-----  
- Calculation of  $p$  -

-----  
-----  
From table 10-9:  $p = 0.00$

with:

- Columns not controlled by inadequate development or splicing along the clear height because  $l_b / l_d \geq 1$

shear control ratio  $V_y E / V_{CoI} E = 0.27402927$

$d = 0.00$

$s = 0.00$

$t = 2 * A_v / (d_c * s) + 4 * t_f / D * (f_{fe} / f_s) = 0.00$

$A_v = 78.53982$ , is the area of the circular stirrup

$d_c = D - 2 * \text{cover} - \text{Hoop Diameter} = 340.00$

The term  $2 * t_f / b_w * (f_{fe} / f_s)$  is implemented to account for FRP contribution

where  $f = 2 * t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength

All these variables have already been given in Shear control ratio calculation.

$N_{UD} = 4770.122$

$A_g = 125663.706$

$f_{cE} = 20.00$

$f_{ytE} = f_{ylE} = 444.4444$

$p_l = \text{Area\_Tot\_Long\_Rein} / (A_g) = 0.0243$

$f_{cE} = 20.00$

-----  
-----  
End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (a)

-----  
-----  
**Calculation No. 5**

column C1, Floor 1

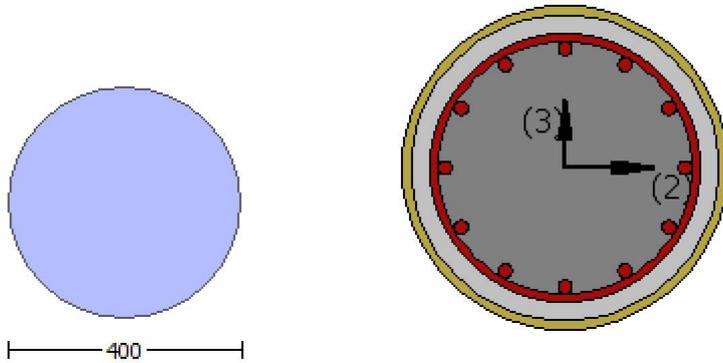
Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Shear capacity VRd

Edge: End

Local Axis: (2)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (b)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 0.85$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 16.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 400.00$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of  $\gamma$  for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE41-17).

Existing material: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material: Steel Strength,  $f_s = f_{sm} = 444.4444$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Inadequate Lap Length with  $l_o/l_{ou,min} = l_b/l_d = 0.30$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $NoDir = 1$

Fiber orientations,  $b_i$ : 0.00°  
Number of layers, NL = 1  
Radius of rounding corners, R = 40.00

-----  
Stepwise Properties

-----  
EDGE -A-  
Bending Moment,  $M_a$  = -8.3350E+006  
Shear Force,  $V_a$  = -2777.007  
EDGE -B-  
Bending Moment,  $M_b$  = 0.07583332  
Shear Force,  $V_b$  = 2777.007  
BOTH EDGES  
Axial Force, F = -4770.122  
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $A_{st}$  = 0.00  
-Compression:  $A_{sc}$  = 3053.628  
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $A_{st,ten}$  = 1017.876  
-Compression:  $A_{sc,com}$  = 1017.876  
-Middle:  $A_{sc,mid}$  = 1017.876  
Mean Diameter of Tension Reinforcement,  $D_{bL,ten}$  = 18.00

-----  
Existing component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = \phi V_n = 267605.60$   
 $V_n$  ((10.3), ASCE 41-17) =  $k_n \phi V_{CoI} = 314830.118$   
 $V_{CoI} = 314830.118$   
 $k_n = 1.00$   
 $displacement\_ductility\_demand = 0.09103583$

-----  
NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + \phi V_f$ '  
where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

-----  
= 1 (normal-weight concrete)  
 $f_c' = 16.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $M_u = 0.07583332$   
 $V_u = 2777.007$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4770.122$   
 $A_g = 125663.706$   
From (11.5.4.8), ACI 318-14:  $V_s = 157913.67$   
 $A_v = \phi / 2 \cdot A_{stirrup} = 123370.055$   
 $f_y = 400.00$   
 $s = 100.00$   
 $V_s$  is multiplied by  $\phi_{CoI} = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $\phi = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression,  
where  $\theta$  is the angle of the crack direction (see KANEPE).  
This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ ,  
as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
orientation 1:  $\theta = b_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \theta)|)$ , with:  
total thickness per orientation,  $t_{f1} = NL \cdot t / NoDir = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}$  ((11-5), ACI 440) = 259.312  
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
with  $f_u = 0.01$   
From (11-11), ACI 440:  $V_s + V_f \leq 213705.936$   
 $b_w \cdot d = \phi \cdot d^2 / 4 = 80424.772$

displacement\_ductility\_demand is calculated as  $\phi / y$

- Calculation of  $\phi / y$  for END B -  
for rotation axis 3 and integ. section (b)

From analysis, chord rotation  $\theta = 0.00015777$   
 $y = (M_y * L_s / 3) / E_{eff} = 0.001733$  ((4.29), Biskinis Phd)  
 $M_y = 1.3732E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 300.00  
From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 7.9240E+012$   
factor = 0.30  
 $A_g = 125663.706$   
 $f_c' = 20.00$   
 $N = 4770.122$   
 $E_c * I_g = 2.6413E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\phi$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 1.3732E+008$   
 $y = 7.5716684E-006$   
 $M_{y\_ten}$  (8c) =  $1.3732E+008$   
 $\phi_{ten}$  (7c) = 72.23345  
error of function (7c) = 0.00097594  
 $M_{y\_com}$  (8d) =  $3.7493E+008$   
 $\phi_{com}$  (7d) = 69.91126  
error of function (7d) = 0.00301333  
with ((10.1), ASCE 41-17)  $e_y = \text{Min}(e_y, 1.25 * e_y * (l_b / l_d)^{2/3}) = 0.00222222$   
 $e_{co} = 0.002$   
 $a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00157314$   
 $N = 4770.122$   
 $A_c = 125663.706$   
((10.1), ASCE 41-17)  $\phi = \text{Min}(\phi, 1.25 * \phi * (l_b / l_d)^{2/3}) = 0.44758025$   
with  $f_c^*$  ((12.3), ACI 440) = 24.12975  
 $f_c = 20.00$   
 $f_l = 1.3173$   
 $k = 1$   
Effective FRP thickness,  $t_f = NL * t * \text{Cos}(b_1) = 1.016$   
 $e_{fe}$  ((12.5) and (12.7)) = 0.004  
 $f_u = 0.01$   
 $E_f = 64828.00$

Calculation of ratio  $l_b / l_d$

Inadequate Lap Length with  $l_b / l_d = 0.30$

End Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (b)

## Calculation No. 6

column C1, Floor 1

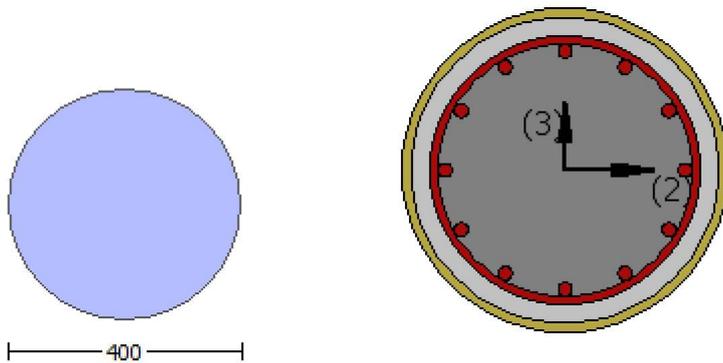
Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity ( $\theta_u$ )

Edge: End

Local Axis: (2)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 0.85$

Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.4444$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 555.5556$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.84055

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Inadequate Lap Length with  $l_o/l_{ou,min} = 0.30$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$   
Tensile Strength,  $f_{fu} = 1055.00$   
Tensile Modulus,  $E_f = 64828.00$   
Elongation,  $e_{fu} = 0.01$   
Number of directions,  $NoDir = 1$   
Fiber orientations,  $bi = 0.00^\circ$   
Number of layers,  $NL = 1$   
Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

At local axis: 3  
EDGE -A-  
Shear Force,  $V_a = -3.7968243E-031$   
EDGE -B-  
Shear Force,  $V_b = 3.7968243E-031$   
BOTH EDGES  
Axial Force,  $F = -4771.233$   
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $As_t = 0.00$   
-Compression:  $As_c = 3053.628$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $As_{t,ten} = 1017.876$   
-Compression:  $As_{l,com} = 1017.876$   
-Middle:  $As_{l,mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.27402927$   
Member Controlled by Flexure ( $V_e/V_r < 1$ )  
Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 96425.251$   
with

$M_{pr1} = \text{Max}(Mu_{1+}, Mu_{1-}) = 1.4464E+008$   
 $Mu_{1+} = 1.4464E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination  
 $Mu_{1-} = 1.4464E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination  
 $M_{pr2} = \text{Max}(Mu_{2+}, Mu_{2-}) = 1.4464E+008$   
 $Mu_{2+} = 1.4464E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination  
 $Mu_{2-} = 1.4464E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

#### Calculation of $Mu_{1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $Mu$   
 $Mu = 1.4464E+008$

$\beta_1 = 0.9250245$   
 $\beta_2 = 0.82105152$   
error of function (3.68), Biskinis Phd = 31682.903  
From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 311.2087$   
 $l_b/d = 0.30$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $Ac = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.31060654$

-----  
Calculation of ratio lb/d  
-----

Inadequate Lap Length with lb/d = 0.30  
-----  
-----  
-----

Calculation of Mu1-  
-----  
-----

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.4464E+008  
-----

= 0.9250245  
' = 0.82105152  
error of function (3.68), Biskinis Phd = 31682.903  
From 5A.2, TBDY: fcc = fc\* c = 36.81095  
conf. factor c = 1.84055  
fc = 20.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 311.2087  
lb/d = 0.30  
d1 = 44.00  
R = 200.00  
v = 0.00155946  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.31060654  
-----

Calculation of ratio lb/d  
-----

Inadequate Lap Length with lb/d = 0.30  
-----  
-----  
-----

Calculation of Mu2+  
-----  
-----

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.4464E+008  
-----

= 0.9250245  
' = 0.82105152  
error of function (3.68), Biskinis Phd = 31682.903  
From 5A.2, TBDY: fcc = fc\* c = 36.81095  
conf. factor c = 1.84055  
fc = 20.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 311.2087  
lb/d = 0.30  
d1 = 44.00  
R = 200.00  
v = 0.00155946  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.31060654  
-----

Calculation of ratio lb/d  
-----

Inadequate Lap Length with lb/d = 0.30  
-----  
-----  
-----

Calculation of Mu2-  
-----

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.4464E+008

= 0.9250245  
' = 0.82105152  
error of function (3.68), Biskinis Phd = 31682.903  
From 5A.2, TBDY: fcc = fc\* c = 36.81095  
conf. factor c = 1.84055  
fc = 20.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 311.2087  
lb/d = 0.30  
d1 = 44.00  
R = 200.00  
v = 0.00155946  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.31060654

Calculation of ratio lb/d

Inadequate Lap Length with lb/d = 0.30

Calculation of Shear Strength Vr = Min(Vr1,Vr2) = 351879.387

Calculation of Shear Strength at edge 1, Vr1 = 351879.387

Vr1 = VCol ((10.3), ASCE 41-17) = knl\*VColO

VColO = 351879.387

knl = 1 (zero step-static loading)

NOTE: In expression (10-3) 'Vs = Av\*fy\*d/s' is replaced by 'Vs+ f\*Vf' where Vf is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)  
fc' = 20.00, but fc^0.5 <= 8.3 MPa (22.5.3.1, ACI 318-14)  
M/Vd = 2.00  
Mu = 9.8466009E-012  
Vu = 3.7968243E-031  
d = 0.8\*D = 320.00  
Nu = 4771.233  
Ag = 125663.706  
From (11.5.4.8), ACI 318-14: Vs = 175459.634  
Av = /2\*A\_stirrup = 123370.055  
fy = 444.4444  
s = 100.00  
Vs is multiplied by Col = 0.00  
s/d = 0.3125  
Vf ((11-3)-(11.4), ACI 440) = 194961.134  
f = 0.95, for fully-wrapped sections  
wf/sf = 1 (FRP strips adjacent to one another).  
In (11.3) sin + cos is replaced with (cot +cota)sina which is more a generalised expression, where is the angle of the crack direction (see KANEPE).  
This later relation, considered as a function Vf( , ), is implemented for every different fiber orientation ai, as well as for 2 crack directions, = 45° and = -45° to take into consideration the cyclic seismic loading.  
orientation 1: 1 = b1 + 90° = 90.00  
Vf = Min(|Vf(45, 1)|,|Vf(-45,a1)|), with:  
total thickness per orientation, tf1 = NL\*t/NoDir = 1.016  
dfv = d (figure 11.2, ACI 440) = 370.00  
ffe ((11-5), ACI 440) = 259.312  
Ef = 64828.00  
fe = 0.004, from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$

$b_w \cdot d = \frac{1}{4} \cdot d^2 = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 351879.387$

$V_{r2} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_n \cdot V_{Col0}$

$V_{Col0} = 351879.387$

$k_n = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

$f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M / Vd = 2.00$

$\mu_u = 9.8466009E-012$

$V_u = 3.7968243E-031$

$d = 0.8 \cdot D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 175459.634$

$A_v = \frac{1}{2} \cdot A_{stirrup} = 123370.055$

$f_y = 444.4444$

$s = 100.00$

$V_s$  is multiplied by  $\phi_{col} = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections

$w_f / s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression,

where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = \theta_1 = 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, \theta_1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{oDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$

$b_w \cdot d = \frac{1}{4} \cdot d^2 = 80424.772$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At local axis: 3

Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 2

(Bending local axis: 3)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\phi = 0.85$

Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.4444$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,  
the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 555.5556$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.84055

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Inadequate Lap Length with  $l_o/l_{ou,min} = 0.30$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $NoDir = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

-----

#### Stepwise Properties

-----

At local axis: 2

EDGE -A-

Shear Force,  $V_a = 4.4296389E-033$

EDGE -B-

Shear Force,  $V_b = -4.4296389E-033$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 0.00$

-Compression:  $As_c = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t,ten} = 1017.876$

-Compression:  $As_{c,com} = 1017.876$

-Middle:  $As_{mid} = 1017.876$

-----

-----

Calculation of Shear Capacity ratio ,  $V_e/V_r = 0.27402927$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 96425.251$

with

$M_{pr1} = \text{Max}(Mu_{1+} , Mu_{1-}) = 1.4464E+008$

$Mu_{1+} = 1.4464E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$Mu_{1-} = 1.4464E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(Mu_{2+} , Mu_{2-}) = 1.4464E+008$

$Mu_{2+} = 1.4464E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

$Mu_{2-} = 1.4464E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

-----

Calculation of  $Mu_{1+}$

-----

-----

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.4464E+008

-----  
= 0.9250245  
' = 0.82105152  
error of function (3.68), Biskinis Phd = 31682.903  
From 5A.2, TBDY: fcc = fc\* c = 36.81095  
conf. factor c = 1.84055  
fc = 20.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 311.2087  
lb/d = 0.30  
d1 = 44.00  
R = 200.00  
v = 0.00155946  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.31060654  
-----

Calculation of ratio lb/d

-----  
Inadequate Lap Length with lb/d = 0.30  
-----  
-----  
-----

Calculation of Mu1-

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.4464E+008

-----  
= 0.9250245  
' = 0.82105152  
error of function (3.68), Biskinis Phd = 31682.903  
From 5A.2, TBDY: fcc = fc\* c = 36.81095  
conf. factor c = 1.84055  
fc = 20.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 311.2087  
lb/d = 0.30  
d1 = 44.00  
R = 200.00  
v = 0.00155946  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.31060654  
-----

Calculation of ratio lb/d

-----  
Inadequate Lap Length with lb/d = 0.30  
-----  
-----  
-----

Calculation of Mu2+

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.4464E+008

-----  
= 0.9250245  
' = 0.82105152  
error of function (3.68), Biskinis Phd = 31682.903  
From 5A.2, TBDY: fcc = fc\* c = 36.81095  
conf. factor c = 1.84055  
fc = 20.00

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 311.2087$

$$l_b/d = 0.30$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00155946$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.31060654$$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of  $\mu_2$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$$\mu = 1.4464E+008$$

$$= 0.9250245$$

$$' = 0.82105152$$

error of function (3.68), Biskinis Phd = 31682.903

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$

$$\text{conf. factor } c = 1.84055$$

$$f_c = 20.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 311.2087$

$$l_b/d = 0.30$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00155946$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.31060654$$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 351879.387$

Calculation of Shear Strength at edge 1,  $V_{r1} = 351879.387$

$$V_{r1} = V_{Co1} \text{ ((10.3), ASCE 41-17)} = knl \cdot V_{Co10}$$

$$V_{Co10} = 351879.387$$

$$knl = 1 \text{ (zero step-static loading)}$$

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

$$f_c' = 20.00, \text{ but } f_c'^{0.5} \leq 8.3 \text{ MPa (22.5.3.1, ACI 318-14)}$$

$$M/d = 2.00$$

$$\mu = 9.0356326E-012$$

$$V_u = 4.4296389E-033$$

$$d = 0.8 \cdot D = 320.00$$

$$N_u = 4771.233$$

$$A_g = 125663.706$$

From (11.5.4.8), ACI 318-14:  $V_s = 175459.634$

$$A_v = \frac{1}{2} \cdot A_{\text{stirrup}} = 123370.055$$

$$f_y = 444.4444$$

$s = 100.00$   
 $V_s$  is multiplied by  $Col = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections  
 $wf/sf = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a)\sin a$  which is more a generalised expression,  
 where  $\theta$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = b1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, a1)|)$ , with:  
 total thickness per orientation,  $tf1 = NL * t / NoDir = 1.016$   
 $dfv = d$  (figure 11.2, ACI 440) = 370.00  
 $ffe$  ((11-5), ACI 440) = 259.312  
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $bw * d = \rho * d * d / 4 = 80424.772$

-----  
 Calculation of Shear Strength at edge 2,  $V_{r2} = 351879.387$   
 $V_{r2} = V_{Col}$  ((10.3), ASCE 41-17) =  $knl * V_{Col0}$   
 $V_{Col0} = 351879.387$   
 $knl = 1$  (zero step-static loading)

-----  
 NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ '  
 where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

-----  
 $\rho = 1$  (normal-weight concrete)  
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $M_u = 9.0356326E-012$   
 $V_u = 4.4296389E-033$   
 $d = 0.8 * D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 175459.634$   
 $A_v = \rho / 2 * A_{stirup} = 123370.055$   
 $f_y = 444.4444$   
 $s = 100.00$   
 $V_s$  is multiplied by  $Col = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections  
 $wf/sf = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a)\sin a$  which is more a generalised expression,  
 where  $\theta$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = b1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, a1)|)$ , with:  
 total thickness per orientation,  $tf1 = NL * t / NoDir = 1.016$   
 $dfv = d$  (figure 11.2, ACI 440) = 370.00  
 $ffe$  ((11-5), ACI 440) = 259.312  
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $bw * d = \rho * d * d / 4 = 80424.772$

-----  
 End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At local axis: 2

Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (b)

Section Type: rccs

#### Constant Properties

Knowledge Factor,  $\gamma = 0.85$

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.4444$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Inadequate Lap Length with  $l_b/l_d = 0.30$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

Bending Moment,  $M = 1.5086324E-011$

Shear Force,  $V_2 = 2777.007$

Shear Force,  $V_3 = 1.4226804E-014$

Axial Force,  $F = -4770.122$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{sl,t} = 0.00$

-Compression:  $A_{sl,c} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{sl,ten} = 1017.876$

-Compression:  $A_{sl,com} = 1017.876$

-Middle:  $A_{sl,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $Db_L = 18.00$

Existing component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_{,R} = \gamma \cdot u = 0.00736525$

$u = \gamma \cdot u = 0.008665$

- Calculation of  $\gamma$  -

$\gamma = (M_y \cdot L_s / 3) / E_{eff} = 0.008665$  ((4.29), Biskinis Phd))

$M_y = 1.3732E+008$

$L_s = M/V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) = 1500.00

From table 10.5, ASCE 41\_17:  $E_{eff} = \text{factor} \cdot E_c \cdot I_g = 7.9240E+012$

factor = 0.30  
Ag = 125663.706  
fc' = 20.00  
N = 4770.122  
Ec\*Ig = 2.6413E+013

-----  
-----  
Calculation of Yielding Moment My

-----  
Calculation of  $\rho_y$  and My according to (7) - (8) in Biskinis and Fardis

-----  
My = Min(My\_ten, My\_com) = 1.3732E+008  
y = 7.5716684E-006  
My\_ten (8c) = 1.3732E+008  
rho\_ten (7c) = 72.23345  
error of function (7c) = 0.00097594  
My\_com (8d) = 3.7493E+008  
rho\_com (7d) = 69.91126  
error of function (7d) = 0.00301333  
with ((10.1), ASCE 41-17) ey = Min(ey, 1.25\*ey\*(lb/l\_d)^2/3) = 0.00222222  
eco = 0.002  
apl = 0.45 ((9c) in Biskinis and Fardis for FRP Wrap)  
d1 = 44.00  
R = 200.00  
v = 0.00157314  
N = 4770.122  
Ac = 125663.706  
((10.1), ASCE 41-17) rho\_y = Min(rho\_y, 1.25\*rho\_y\*(lb/l\_d)^2/3) = 0.44758025  
with fc' ((12.3), ACI 440) = 24.12975  
fc = 20.00  
fl = 1.3173  
k = 1  
Effective FRP thickness, tf = NL\*t\*Cos(b1) = 1.016  
efe ((12.5) and (12.7)) = 0.004  
fu = 0.01  
Ef = 64828.00

-----  
-----  
Calculation of ratio lb/l\_d

Inadequate Lap Length with lb/l\_d = 0.30

-----  
-----  
- Calculation of  $\rho_p$  -

-----  
From table 10-9:  $\rho_p = 0.00$

with:

- Columns not controlled by inadequate development or splicing along the clear height because lb/l\_d >= 1

shear control ratio VyE/VCoIE = 0.27402927

d = 0.00

s = 0.00

t = 2\*Av/(dc\*s) + 4\*tf/D\*(ffe/fs) = 0.00

Av = 78.53982, is the area of the circular stirrup

dc = D - 2\*cover - Hoop Diameter = 340.00

The term 2\*tf/bw\*(ffe/fs) is implemented to account for FRP contribution

where f = 2\*tf/bw is FRP ratio (EC8 - 3, A.4.4.3(6)) and ffe/fs normalises f to steel strength

All these variables have already been given in Shear control ratio calculation.

NUD = 4770.122

Ag = 125663.706

fcE = 20.00

fytE = fytE = 444.4444

pl = Area\_Tot\_Long\_Rein/(Ag) = 0.0243

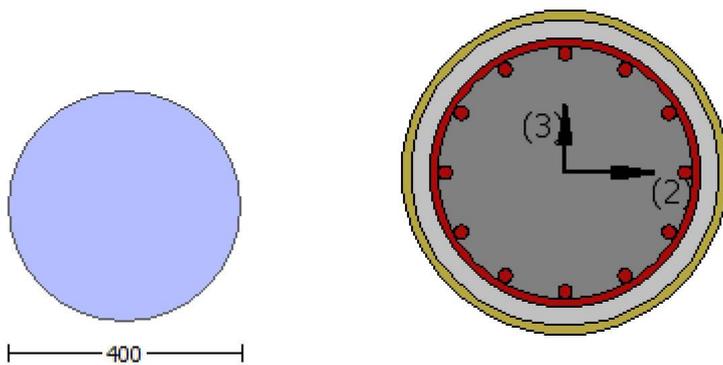
fcE = 20.00

-----  
End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 2  
Integration Section: (b)

## Calculation No. 7

column C1, Floor 1  
Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)  
Analysis: Uniform +X  
Check: Shear capacity  $V_{Rd}$   
Edge: End  
Local Axis: (3)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1  
At local axis: 3  
Integration Section: (b)  
Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 0.85$   
Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE41-17.  
Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17  
Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 16.00$   
Existing material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 400.00$   
Concrete Elasticity,  $E_c = 21019.039$   
Steel Elasticity,  $E_s = 200000.00$

#####  
Note: Especially for the calculation of  $\gamma$  for displacement ductility demand,  
the expected (mean value) strengths are used (7.5.1.3, ASCE41-17) because bending is considered as  
Deformation-Controlled Action (Table C7-1, ASCE41-17).

Existing material: Concrete Strength,  $f_c = f_{cm} = 20.00$   
Existing material: Steel Strength,  $f_s = f_{sm} = 444.4444$   
#####  
Diameter,  $D = 400.00$   
Cover Thickness,  $c = 25.00$

Element Length, L = 3000.00  
Primary Member  
Smooth Bars  
Ductile Steel  
With Detailing for Earthquake Resistance (including stirrups closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Inadequate Lap Length with  $l_o/l_{o,min} = l_b/l_d = 0.30$   
FRP Wrapping Data  
Type: Carbon  
Cured laminate properties (design values)  
Thickness, t = 1.016  
Tensile Strength,  $f_{fu} = 1055.00$   
Tensile Modulus,  $E_f = 64828.00$   
Elongation,  $e_{fu} = 0.01$   
Number of directions, NoDir = 1  
Fiber orientations,  $b_i: 0.00^\circ$   
Number of layers, NL = 1  
Radius of rounding corners, R = 40.00

#### Stepwise Properties

EDGE -A-  
Bending Moment,  $M_a = 2.7525396E-011$   
Shear Force,  $V_a = -1.4226804E-014$   
EDGE -B-  
Bending Moment,  $M_b = 1.5086324E-011$   
Shear Force,  $V_b = 1.4226804E-014$   
BOTH EDGES  
Axial Force, F = -4770.122  
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $A_{sl,t} = 0.00$   
-Compression:  $A_{sl,c} = 3053.628$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $A_{sl,ten} = 1017.876$   
-Compression:  $A_{sl,com} = 1017.876$   
-Middle:  $A_{sl,mid} = 1017.876$   
Mean Diameter of Tension Reinforcement,  $D_{bL,ten} = 18.00$

Existing component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = \phi V_n = 267605.60$   
 $V_n$  ((10.3), ASCE 41-17) =  $k_n \phi V_{CoI0} = 314830.118$   
 $V_{CoI} = 314830.118$   
 $k_n = 1.00$   
displacement\_ductility\_demand = 0.00

NOTE: In expression (10-3) ' $V_s = A_v \phi f_y d/s$ ' is replaced by ' $V_s + \phi V_f$ '  
where  $V_f$  is the contribution of FRPs ((11.3), ACI 440).

= 1 (normal-weight concrete)  
 $f_c' = 16.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $M_u = 1.5086324E-011$   
 $V_u = 1.4226804E-014$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4770.122$   
 $A_g = 125663.706$   
From (11.5.4.8), ACI 318-14:  $V_s = 157913.67$   
 $A_v = \phi / 2 \cdot A_{stirrup} = 123370.055$   
 $f_y = 400.00$   
 $s = 100.00$   
 $V_s$  is multiplied by  $\phi_{CoI} = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $\phi = 0.95$ , for fully-wrapped sections

$wf/sf = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression, where  $a$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = b1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, a1)|)$ , with:

total thickness per orientation,  $t_{f1} = NL * t / \text{NoDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 213705.936$

$b_w * d = \frac{V_s * d}{4} = 80424.772$

displacement ductility demand is calculated as  $\delta / y$

- Calculation of  $\delta / y$  for END B -

for rotation axis 2 and integ. section (b)

From analysis, chord rotation  $\theta = 2.6179514E-022$

$y = (M_y * L_s / 3) / E_{eff} = 0.008665$  ((4.29), Biskinis Phd))

$M_y = 1.3732E+008$

$L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 1500.00

From table 10.5, ASCE 41\_17:  $E_{eff} = \text{factor} * E_c * I_g = 7.9240E+012$

factor = 0.30

$A_g = 125663.706$

$f_c' = 20.00$

$N = 4770.122$

$E_c * I_g = 2.6413E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\delta / y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 1.3732E+008$

$y = 7.5716684E-006$

$M_{y\_ten}$  (8c) = 1.3732E+008

$\delta_{ten}$  (7c) = 72.23345

error of function (7c) = 0.00097594

$M_{y\_com}$  (8d) = 3.7493E+008

$\delta_{com}$  (7d) = 69.91126

error of function (7d) = 0.00301333

with ((10.1), ASCE 41-17)  $e_y = \text{Min}(e_y, 1.25 * e_y * (l_b / l_d)^{2/3}) = 0.00222222$

$e_{co} = 0.002$

$a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)

$d_1 = 44.00$

$R = 200.00$

$v = 0.00157314$

$N = 4770.122$

$A_c = 125663.706$

((10.1), ASCE 41-17)  $e = \text{Min}(e, 1.25 * e * (l_b / l_d)^{2/3}) = 0.44758025$

with  $f_c^*$  ((12.3), ACI 440) = 24.12975

$f_c = 20.00$

$f_l = 1.3173$

$k = 1$

Effective FRP thickness,  $t_f = NL * t * \text{Cos}(b1) = 1.016$

$e_{fe}$  ((12.5) and (12.7)) = 0.004

$f_u = 0.01$

$E_f = 64828.00$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

End Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (b)

## Calculation No. 8

column C1, Floor 1

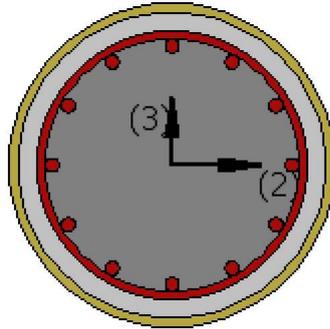
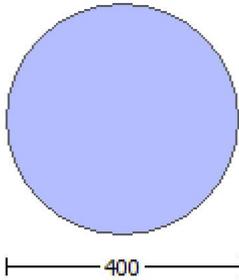
Limit State: Immediate Occupancy (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity ( $\theta$ )

Edge: End

Local Axis: (3)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 0.85$

Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.4444$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 555.5556$

#####

Diameter, D = 400.00  
Cover Thickness, c = 25.00  
Mean Confinement Factor overall section = 1.84055  
Element Length, L = 3000.00  
Primary Member  
Smooth Bars  
Ductile Steel  
With Detailing for Earthquake Resistance (including stirrups closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Inadequate Lap Length with  $l_o/l_{o,min}$  = 0.30  
FRP Wrapping Data  
Type: Carbon  
Cured laminate properties (design values)  
Thickness, t = 1.016  
Tensile Strength,  $f_{fu}$  = 1055.00  
Tensile Modulus,  $E_f$  = 64828.00  
Elongation,  $e_{fu}$  = 0.01  
Number of directions, NoDir = 1  
Fiber orientations,  $b_i$ : 0.00°  
Number of layers, NL = 1  
Radius of rounding corners, R = 40.00

-----  
Stepwise Properties  
-----

At local axis: 3  
EDGE -A-  
Shear Force,  $V_a$  = -3.7968243E-031  
EDGE -B-  
Shear Force,  $V_b$  = 3.7968243E-031  
BOTH EDGES  
Axial Force, F = -4771.233  
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $A_{sl}$  = 0.00  
-Compression:  $A_{slc}$  = 3053.628  
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $A_{sl,ten}$  = 1017.876  
-Compression:  $A_{sl,com}$  = 1017.876  
-Middle:  $A_{sl,mid}$  = 1017.876  
-----  
-----

Calculation of Shear Capacity ratio ,  $V_e/V_r$  = 0.27402927  
Member Controlled by Flexure ( $V_e/V_r < 1$ )  
Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 96425.251$   
with  
 $M_{pr1} = \text{Max}(\mu_{u1+}, \mu_{u1-}) = 1.4464E+008$   
 $\mu_{u1+} = 1.4464E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction  
which is defined for the static loading combination  
 $\mu_{u1-} = 1.4464E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment  
direction which is defined for the static loading combination  
 $M_{pr2} = \text{Max}(\mu_{u2+}, \mu_{u2-}) = 1.4464E+008$   
 $\mu_{u2+} = 1.4464E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction  
which is defined for the the static loading combination  
 $\mu_{u2-} = 1.4464E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment  
direction which is defined for the the static loading combination

-----  
Calculation of  $\mu_{u1+}$   
-----  
-----

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$   
 $\mu_u = 1.4464E+008$   
-----  
= 0.9250245

' = 0.82105152  
error of function (3.68), Biskinis Phd = 31682.903  
From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 311.2087$   
 $l_b/l_d = 0.30$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.31060654$

-----  
Calculation of ratio  $l_b/l_d$

-----  
Inadequate Lap Length with  $l_b/l_d = 0.30$

-----  
Calculation of  $\mu_{u1}$ -

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$   
 $\mu_u = 1.4464E+008$

-----  
 $= 0.9250245$   
' = 0.82105152  
error of function (3.68), Biskinis Phd = 31682.903  
From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 311.2087$   
 $l_b/l_d = 0.30$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.31060654$

-----  
Calculation of ratio  $l_b/l_d$

-----  
Inadequate Lap Length with  $l_b/l_d = 0.30$

-----  
Calculation of  $\mu_{u2+}$

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$   
 $\mu_u = 1.4464E+008$

-----  
 $= 0.9250245$   
' = 0.82105152  
error of function (3.68), Biskinis Phd = 31682.903  
From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 311.2087$   
 $l_b/l_d = 0.30$   
 $d_1 = 44.00$   
 $R = 200.00$

$$v = 0.00155946$$

$$N = 4771.233$$

$$Ac = 125663.706$$

$$= *Min(1, 1.25*(lb/d)^{2/3}) = 0.31060654$$

Calculation of ratio lb/d

Inadequate Lap Length with lb/d = 0.30

Calculation of Mu2-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
 Mu = 1.4464E+008

$$= 0.9250245$$

$$' = 0.82105152$$

error of function (3.68), Biskinis Phd = 31682.903

From 5A.2, TBDY:  $f_{cc} = f_c * c = 36.81095$

conf. factor  $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y * Min(1, 1.25*(lb/d)^{2/3}) = 311.2087$

lb/d = 0.30

d1 = 44.00

R = 200.00

v = 0.00155946

N = 4771.233

Ac = 125663.706

$$= *Min(1, 1.25*(lb/d)^{2/3}) = 0.31060654$$

Calculation of ratio lb/d

Inadequate Lap Length with lb/d = 0.30

Calculation of Shear Strength  $V_r = Min(V_{r1}, V_{r2}) = 351879.387$

Calculation of Shear Strength at edge 1,  $V_{r1} = 351879.387$

$V_{r1} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_{nl} * V_{Col0}$

$V_{Col0} = 351879.387$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ ' where  $V_f$  is the contribution of FRPs ((11.3), ACI 440).

= 1 (normal-weight concrete)

$f_c' = 20.00$ , but  $f_c^{0.5} \leq 8.3$  MPa ((22.5.3.1), ACI 318-14)

$M / Vd = 2.00$

$M_u = 9.8466009E-012$

$V_u = 3.7968243E-031$

$d = 0.8 * D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From ((11.5.4.8), ACI 318-14:  $V_s = 175459.634$

$A_v = /2 * A_{stirrup} = 123370.055$

$f_y = 444.4444$

$s = 100.00$

$V_s$  is multiplied by  $Col = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections  
 $wf/sf = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \cot \alpha) \sin \alpha$  which is more a generalised expression,  
 where  $\theta$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = b1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \theta)|)$ , with:  
 total thickness per orientation,  $tf1 = NL * t / NoDir = 1.016$   
 $dfv = d$  (figure 11.2, ACI 440) = 370.00  
 $ffe$  ((11-5), ACI 440) = 259.312  
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $bw * d = \rho * d * d / 4 = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 351879.387$   
 $V_{r2} = V_{Col}$  ((10.3), ASCE 41-17) =  $knl * V_{Col0}$   
 $V_{Col0} = 351879.387$   
 $knl = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ '  
 where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\rho = 1$  (normal-weight concrete)  
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M / Vd = 2.00$   
 $\mu_u = 9.8466009E-012$   
 $\nu_u = 3.7968243E-031$   
 $d = 0.8 * D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 175459.634$   
 $A_v = \rho_s * A_{stirrup} = 123370.055$   
 $f_y = 444.4444$   
 $s = 100.00$   
 $V_s$  is multiplied by  $Col = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections  
 $wf/sf = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \cot \alpha) \sin \alpha$  which is more a generalised expression,  
 where  $\theta$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = b1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \theta)|)$ , with:  
 total thickness per orientation,  $tf1 = NL * t / NoDir = 1.016$   
 $dfv = d$  (figure 11.2, ACI 440) = 370.00  
 $ffe$  ((11-5), ACI 440) = 259.312  
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $bw * d = \rho * d * d / 4 = 80424.772$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At local axis: 3

Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At Shear local axis: 2

(Bending local axis: 3)  
Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 0.85$

Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.4444$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 555.5556$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.84055

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Inadequate Lap Length with  $l_o/l_{o,min} = 0.30$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

Stepwise Properties

At local axis: 2

EDGE -A-

Shear Force,  $V_a = 4.4296389E-033$

EDGE -B-

Shear Force,  $V_b = -4.4296389E-033$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{sl,t} = 0.00$

-Compression:  $A_{sl,c} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{sl,ten} = 1017.876$

-Compression:  $A_{sl,com} = 1017.876$

-Middle:  $A_{sl,mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.27402927$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 96425.251$

with

$M_{pr1} = \text{Max}(M_{u1+}, M_{u1-}) = 1.4464E+008$

$M_{u1+} = 1.4464E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

Mu1- = 1.4464E+008, is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$$Mpr2 = \text{Max}(Mu2+ , Mu2-) = 1.4464E+008$$

Mu2+ = 1.4464E+008, is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

Mu2- = 1.4464E+008, is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

-----  
Calculation of Mu1+  
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu

$$Mu = 1.4464E+008$$

$$= 0.9250245$$

$$' = 0.82105152$$

error of function (3.68), Biskinis Phd = 31682.903

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$

conf. factor  $c = 1.84055$

$$f_c = 20.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 311.2087$

$$l_b/d = 0.30$$

$$d1 = 44.00$$

$$R = 200.00$$

$$v = 0.00155946$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.31060654$$

-----  
Calculation of ratio  $l_b/d$   
-----

Inadequate Lap Length with  $l_b/d = 0.30$   
-----  
-----  
-----

-----  
Calculation of Mu1-  
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu

$$Mu = 1.4464E+008$$

$$= 0.9250245$$

$$' = 0.82105152$$

error of function (3.68), Biskinis Phd = 31682.903

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$

conf. factor  $c = 1.84055$

$$f_c = 20.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 311.2087$

$$l_b/d = 0.30$$

$$d1 = 44.00$$

$$R = 200.00$$

$$v = 0.00155946$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.31060654$$

-----  
Calculation of ratio  $l_b/d$   
-----

Inadequate Lap Length with  $l_b/d = 0.30$   
-----  
-----  
-----

Calculation of  $\mu_{2+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 1.4464E+008$

$\phi = 0.9250245$   
 $\beta = 0.82105152$   
error of function (3.68), Biskinis Phd = 31682.903  
From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 311.2087$   
 $l_b/d = 0.30$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.31060654$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of  $\mu_{2-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 1.4464E+008$

$\phi = 0.9250245$   
 $\beta = 0.82105152$   
error of function (3.68), Biskinis Phd = 31682.903  
From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 311.2087$   
 $l_b/d = 0.30$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.31060654$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 351879.387$

Calculation of Shear Strength at edge 1,  $V_{r1} = 351879.387$

$V_{r1} = V_{Co1}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{Co10}$   
 $V_{Co10} = 351879.387$   
 $k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_{s+} = f \cdot V_f$ '

where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\lambda = 1$  (normal-weight concrete)  
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $\mu_u = 9.0356326E-012$   
 $\nu_u = 4.4296389E-033$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
From (11.5.4.8), ACI 318-14:  $V_s = 175459.634$   
 $A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 123370.055$   
 $f_y = 444.4444$   
 $s = 100.00$   
 $V_s$  is multiplied by  $\text{Col} = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression,  
where  $\theta$  is the angle of the crack direction (see KANEPE).  
This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ ,  
as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
orientation 1:  $\theta = 45^\circ$   
 $V_f = \text{Min}(|V_f(45, \alpha_1)|, |V_f(-45, \alpha_1)|)$ , with:  
total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}$  ((11-5), ACI 440) = 259.312  
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
with  $f_u = 0.01$   
From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $b_w \cdot d = \frac{V_s + V_f}{f_e} = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 351879.387$   
 $V_{r2} = V_{\text{Col}}$  ((10.3), ASCE 41-17) =  $k_n l \cdot V_{\text{Col}0}$   
 $V_{\text{Col}0} = 351879.387$   
 $k_n l = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ '  
where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\lambda = 1$  (normal-weight concrete)  
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $\mu_u = 9.0356326E-012$   
 $\nu_u = 4.4296389E-033$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
From (11.5.4.8), ACI 318-14:  $V_s = 175459.634$   
 $A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 123370.055$   
 $f_y = 444.4444$   
 $s = 100.00$   
 $V_s$  is multiplied by  $\text{Col} = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression,  
where  $\theta$  is the angle of the crack direction (see KANEPE).  
This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ ,  
as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
orientation 1:  $\theta = 45^\circ$   
 $V_f = \text{Min}(|V_f(45, \alpha_1)|, |V_f(-45, \alpha_1)|)$ , with:  
total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$

dfv = d (figure 11.2, ACI 440) = 370.00  
ffe ((11-5), ACI 440) = 259.312  
Ef = 64828.00  
fe = 0.004, from (11.6a), ACI 440  
with fu = 0.01  
From (11-11), ACI 440: Vs + Vf <= 238930.50  
bw\*d = \*d\*d/4 = 80424.772

-----  
-----  
End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At local axis: 2

-----  
-----  
Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1  
At local axis: 3  
Integration Section: (b)  
Section Type: rccs

#### Constant Properties

-----  
Knowledge Factor, = 0.85  
Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE41-17.  
Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17  
Consequently:  
Existing material of Primary Member: Concrete Strength, fc = fcm = 20.00  
Existing material of Primary Member: Steel Strength, fs = fsm = 444.4444  
Concrete Elasticity, Ec = 21019.039  
Steel Elasticity, Es = 200000.00  
Diameter, D = 400.00  
Cover Thickness, c = 25.00  
Element Length, L = 3000.00  
Primary Member  
Smooth Bars  
Ductile Steel  
With Detailing for Earthquake Resistance (including stirrups closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Inadequate Lap Length with lb/ld = 0.30  
FRP Wrapping Data  
Type: Carbon  
Cured laminate properties (design values)  
Thickness, t = 1.016  
Tensile Strength, ffu = 1055.00  
Tensile Modulus, Ef = 64828.00  
Elongation, efu = 0.01  
Number of directions, NoDir = 1  
Fiber orientations, bi: 0.00°  
Number of layers, NL = 1  
Radius of rounding corners, R = 40.00

#### Stepwise Properties

-----  
Bending Moment, M = 0.07583332  
Shear Force, V2 = 2777.007  
Shear Force, V3 = 1.4226804E-014  
Axial Force, F = -4770.122  
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension: Aslt = 0.00  
-Compression: Aslc = 3053.628  
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension: Asl,ten = 1017.876  
-Compression: Asl,com = 1017.876  
-Middle: Asl,mid = 1017.876  
Mean Diameter of Tension Reinforcement, DbL = 18.00  
-----

Existing component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_R = u = 0.00147305$   
 $u = y + p = 0.001733$

- Calculation of  $y$  -

$y = (M_y * L_s / 3) / E_{eff} = 0.001733$  ((4.29), Biskinis Phd)  
 $M_y = 1.3732E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 300.00  
From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 7.9240E+012$   
factor = 0.30  
Ag = 125663.706  
fc' = 20.00  
N = 4770.122  
 $E_c * I_g = 2.6413E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 1.3732E+008$   
 $y = 7.5716684E-006$   
 $M_{y\_ten}$  (8c) = 1.3732E+008  
 $y_{ten}$  (7c) = 72.23345  
error of function (7c) = 0.00097594  
 $M_{y\_com}$  (8d) = 3.7493E+008  
 $y_{com}$  (7d) = 69.91126  
error of function (7d) = 0.00301333  
with ((10.1), ASCE 41-17)  $e_y = \text{Min}(e_y, 1.25 * e_y * (l_b / l_d)^{2/3}) = 0.00222222$   
eco = 0.002  
apl = 0.45 ((9c) in Biskinis and Fardis for FRP Wrap)  
d1 = 44.00  
R = 200.00  
v = 0.00157314  
N = 4770.122  
Ac = 125663.706  
((10.1), ASCE 41-17) =  $\text{Min}( , 1.25 * (l_b / l_d)^{2/3}) = 0.44758025$   
with fc' ((12.3), ACI 440) = 24.12975  
fc = 20.00  
fl = 1.3173  
k = 1  
Effective FRP thickness, tf =  $NL * t * \text{Cos}(b1) = 1.016$   
efe ((12.5) and (12.7)) = 0.004  
fu = 0.01  
Ef = 64828.00

Calculation of ratio  $l_b / l_d$

Inadequate Lap Length with  $l_b / l_d = 0.30$

- Calculation of  $p$  -

From table 10-9:  $p = 0.00$

with:

- Columns not controlled by inadequate development or splicing along the clear height because  $l_b / l_d \geq 1$   
shear control ratio  $V_y E / C_o I_{OE} = 0.27402927$   
d = 0.00  
s = 0.00  
 $t = 2 * A_v / (d_c * s) + 4 * t_f / D * (f_{fe} / f_s) = 0.00$

$A_v = 78.53982$ , is the area of the circular stirrup

$d_c = D - 2 \cdot \text{cover} - \text{Hoop Diameter} = 340.00$

The term  $2 \cdot t_f / b_w \cdot (f_{fe} / f_s)$  is implemented to account for FRP contribution

where  $f = 2 \cdot t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength

All these variables have already been given in Shear control ratio calculation.

$NUD = 4770.122$

$A_g = 125663.706$

$f_{cE} = 20.00$

$f_{yE} = f_{yI} = 444.4444$

$\rho_l = \text{Area\_Tot\_Long\_Rein} / (A_g) = 0.0243$

$f_{cE} = 20.00$

End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (b)

## Calculation No. 9

column C1, Floor 1

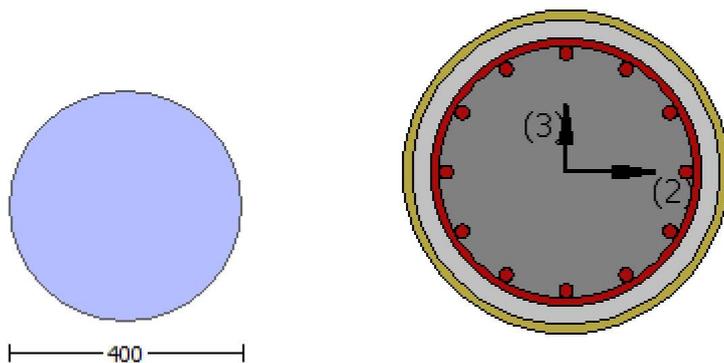
Limit State: Collapse Prevention (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Shear capacity  $V_{Rd}$

Edge: Start

Local Axis: (2)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (a)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 0.85$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 16.00$   
 Existing material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 400.00$   
 Concrete Elasticity,  $E_c = 21019.039$   
 Steel Elasticity,  $E_s = 200000.00$   
 #####  
 Note: Especially for the calculation of  $\gamma$  for displacement ductility demand,  
 the expected (mean value) strengths are used (7.5.1.3, ASCE41-17) because bending is considered as  
 Deformation-Controlled Action (Table C7-1, ASCE41-17).  
 Existing material: Concrete Strength,  $f_c = f_{cm} = 20.00$   
 Existing material: Steel Strength,  $f_s = f_{sm} = 444.4444$   
 #####  
 Diameter,  $D = 400.00$   
 Cover Thickness,  $c = 25.00$   
 Element Length,  $L = 3000.00$   
 Primary Member  
 Smooth Bars  
 Ductile Steel  
 With Detailing for Earthquake Resistance (including stirrups closed at 135°)  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Inadequate Lap Length with  $l_o/l_{o,min} = l_b/l_d = 0.30$   
 FRP Wrapping Data  
 Type: Carbon  
 Cured laminate properties (design values)  
 Thickness,  $t = 1.016$   
 Tensile Strength,  $f_{fu} = 1055.00$   
 Tensile Modulus,  $E_f = 64828.00$   
 Elongation,  $e_{fu} = 0.01$   
 Number of directions,  $NoDir = 1$   
 Fiber orientations,  $b_i = 0.00^\circ$   
 Number of layers,  $NL = 1$   
 Radius of rounding corners,  $R = 40.00$

-----  
 Stepwise Properties  
 -----

EDGE -A-  
 Bending Moment,  $M_a = -1.0423E+007$   
 Shear Force,  $V_a = -3472.529$   
 EDGE -B-  
 Bending Moment,  $M_b = 0.09482631$   
 Shear Force,  $V_b = 3472.529$   
 BOTH EDGES  
 Axial Force,  $F = -4769.844$   
 Longitudinal Reinforcement Area Distribution (in 2 divisions)  
 -Tension:  $A_{sl,t} = 1272.345$   
 -Compression:  $A_{sl,c} = 1781.283$   
 Longitudinal Reinforcement Area Distribution (in 3 divisions)  
 -Tension:  $A_{sl,ten} = 1017.876$   
 -Compression:  $A_{sl,com} = 1017.876$   
 -Middle:  $A_{sl,mid} = 1017.876$   
 Mean Diameter of Tension Reinforcement,  $D_{bL,ten} = 18.00$

-----  
 Existing component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = \phi V_n = 224627.799$   
 $V_n$  ((10.3), ASCE 41-17) =  $k_n \phi V_{CoI} = 264267.999$   
 $V_{CoI} = 264267.999$   
 $k_n = 1.00$   
 $displacement\_ductility\_demand = 0.02032763$

-----  
 NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_{s+} = \phi \cdot V_f$ '  
 where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

-----  
 $\gamma = 1$  (normal-weight concrete)  
 $f'_c = 16.00$ , but  $f'_c^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 4.00$

$\mu_u = 1.0423E+007$   
 $V_u = 3472.529$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4769.844$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 157913.67$   
 $A_v = \sqrt{2} \cdot A_{stirrup} = 123370.055$   
 $f_y = 400.00$   
 $s = 100.00$   
 $V_s$  is multiplied by  $\text{Col} = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression,  
 where  $a$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = b_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, a_1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = N_L \cdot t / \text{NoDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}$  ((11-5), ACI 440) = 259.312  
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 213705.936$   
 $b_w \cdot d = \sqrt{V_s \cdot d} / 4 = 80424.772$

displacement ductility demand is calculated as  $\delta / y$

- Calculation of  $\delta / y$  for END A -  
for rotation axis 3 and integ. section (a)

From analysis, chord rotation  $\theta = 0.00035245$   
 $y = (M_y \cdot L_s / 3) / E_{eff} = 0.01733835$  ((4.29), Biskinis Phd))  
 $M_y = 1.3732E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) = 3001.447  
 From table 10.5, ASCE 41\_17:  $E_{eff} = \text{factor} \cdot E_c \cdot I_g = 7.9240E+012$   
 $\text{factor} = 0.30$   
 $A_g = 125663.706$   
 $f_c' = 20.00$   
 $N = 4769.844$   
 $E_c \cdot I_g = 2.6413E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\delta / y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 1.3732E+008$   
 $y = 7.5716672E-006$   
 $M_{y\_ten}$  (8c) = 1.3732E+008  
 $\delta_{ten}$  (7c) = 72.23344  
 error of function (7c) = 0.00097593  
 $M_{y\_com}$  (8d) = 3.7493E+008  
 $\delta_{com}$  (7d) = 69.91126  
 error of function (7d) = 0.00301342  
 with ((10.1), ASCE 41-17)  $e_y = \text{Min}(e_y, 1.25 \cdot e_y \cdot (I_b / I_d)^{2/3}) = 0.00222222$   
 $e_{co} = 0.002$   
 $a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00157305$

$N = 4769.844$   
 $A_c = 125663.706$   
 $((10.1), ASCE 41-17) = \text{Min}( , 1.25 * (lb/d)^{2/3} ) = 0.44758025$   
 with  $f_c^* ((12.3), ACI 440) = 24.12975$   
 $f_c = 20.00$   
 $f_l = 1.3173$   
 $k = 1$   
 Effective FRP thickness,  $t_f = NL * t * \text{Cos}(b1) = 1.016$   
 $e_{fe} ((12.5) \text{ and } (12.7)) = 0.004$   
 $f_u = 0.01$   
 $E_f = 64828.00$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

End Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (a)

## Calculation No. 10

column C1, Floor 1

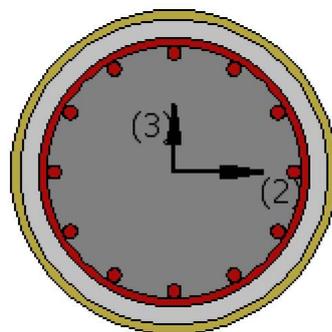
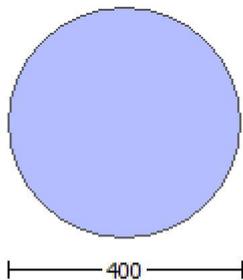
Limit State: Collapse Prevention (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity (  $\theta_u$  )

Edge: Start

Local Axis: (2)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rccs

Constant Properties

-----  
Knowledge Factor,  $\phi = 0.85$   
Mean strength values are used for both shear and moment calculations.  
Consequently:  
Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$   
Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.4444$   
Concrete Elasticity,  $E_c = 21019.039$   
Steel Elasticity,  $E_s = 200000.00$   
#####

Note: Especially for the calculation of moment strengths,  
the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14  
Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 555.5556$   
#####

Diameter,  $D = 400.00$   
Cover Thickness,  $c = 25.00$   
Mean Confinement Factor overall section = 1.84055  
Element Length,  $L = 3000.00$   
Primary Member  
Smooth Bars  
Ductile Steel  
With Detailing for Earthquake Resistance (including stirrups closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Inadequate Lap Length with  $l_o/l_{ou,min} = 0.30$   
FRP Wrapping Data  
Type: Carbon  
Cured laminate properties (design values)  
Thickness,  $t = 1.016$   
Tensile Strength,  $f_{fu} = 1055.00$   
Tensile Modulus,  $E_f = 64828.00$   
Elongation,  $e_{fu} = 0.01$   
Number of directions,  $N_{oDir} = 1$   
Fiber orientations,  $b_i: 0.00^\circ$   
Number of layers,  $N_L = 1$   
Radius of rounding corners,  $R = 40.00$   
-----

#### Stepwise Properties

-----

At local axis: 3  
EDGE -A-  
Shear Force,  $V_a = -3.7968243E-031$   
EDGE -B-  
Shear Force,  $V_b = 3.7968243E-031$   
BOTH EDGES  
Axial Force,  $F = -4771.233$   
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $A_{sl,t} = 0.00$   
-Compression:  $A_{sl,c} = 3053.628$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $A_{sl,ten} = 1017.876$   
-Compression:  $A_{sl,com} = 1017.876$   
-Middle:  $A_{sl,mid} = 1017.876$   
-----  
-----

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.27402927$   
Member Controlled by Flexure ( $V_e/V_r < 1$ )  
Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 96425.251$   
with  
 $M_{pr1} = \text{Max}(M_{u1+}, M_{u1-}) = 1.4464E+008$   
 $M_{u1+} = 1.4464E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction  
which is defined for the static loading combination  
 $M_{u1-} = 1.4464E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment  
direction which is defined for the static loading combination  
 $M_{pr2} = \text{Max}(M_{u2+}, M_{u2-}) = 1.4464E+008$   
 $M_{u2+} = 1.4464E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction

which is defined for the the static loading combination

$\mu_{2-} = 1.4464E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

-----  
Calculation of  $\mu_{1+}$   
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 1.4464E+008$   
-----

$\phi = 0.9250245$   
 $\lambda = 0.82105152$   
error of function (3.68), Biskinis Phd = 31682.903  
From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 311.2087$   
 $l_b/d = 0.30$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $\phi \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.31060654$   
-----

Calculation of ratio  $l_b/d$   
-----

Inadequate Lap Length with  $l_b/d = 0.30$   
-----  
-----

-----  
Calculation of  $\mu_{1-}$   
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 1.4464E+008$   
-----

$\phi = 0.9250245$   
 $\lambda = 0.82105152$   
error of function (3.68), Biskinis Phd = 31682.903  
From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 311.2087$   
 $l_b/d = 0.30$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $\phi \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.31060654$   
-----

Calculation of ratio  $l_b/d$   
-----

Inadequate Lap Length with  $l_b/d = 0.30$   
-----  
-----

-----  
Calculation of  $\mu_{2+}$   
-----  
-----

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.4464E+008

$$= 0.9250245$$

$$' = 0.82105152$$

error of function (3.68), Biskinis Phd = 31682.903

From 5A.2, TBDY:  $f_{cc} = f_c' \cdot c = 36.81095$

$$\text{conf. factor } c = 1.84055$$

$$f_c = 20.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 311.2087$

$$l_b/d = 0.30$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00155946$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.31060654$$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of Mu2-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.4464E+008

$$= 0.9250245$$

$$' = 0.82105152$$

error of function (3.68), Biskinis Phd = 31682.903

From 5A.2, TBDY:  $f_{cc} = f_c' \cdot c = 36.81095$

$$\text{conf. factor } c = 1.84055$$

$$f_c = 20.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 311.2087$

$$l_b/d = 0.30$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00155946$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.31060654$$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 351879.387$

Calculation of Shear Strength at edge 1,  $V_{r1} = 351879.387$

$V_{r1} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{Col0}$

$$V_{Col0} = 351879.387$$

$$k_{nl} = 1 \text{ (zero step-static loading)}$$

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$$= 1 \text{ (normal-weight concrete)}$$

$$f_c' = 20.00, \text{ but } f_c'^{0.5} \leq 8.3 \text{ MPa (22.5.3.1, ACI 318-14)}$$

$$M/Vd = 2.00$$

$$\mu_u = 9.8466009E-012$$

$$\mu_v = 3.7968243E-031$$

$$d = 0.8 \cdot D = 320.00$$

$$N_u = 4771.233$$

$$A_g = 125663.706$$

$$\text{From (11.5.4.8), ACI 318-14: } V_s = 175459.634$$

$$A_v = \frac{1}{2} \cdot A_{\text{stirrup}} = 123370.055$$

$$f_y = 444.4444$$

$$s = 100.00$$

$$V_s \text{ is multiplied by } \phi_{\text{col}} = 0.00$$

$$s/d = 0.3125$$

$$V_f \text{ ((11-3)-(11.4), ACI 440) } = 194961.134$$

$$f = 0.95, \text{ for fully-wrapped sections}$$

$$w_f/s_f = 1 \text{ (FRP strips adjacent to one another).}$$

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

$$\text{orientation 1: } \theta_1 = b_1 + 90^\circ = 90.00$$

$$V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, \theta_1)|), \text{ with:}$$

$$\text{total thickness per orientation, } t_{f1} = N_L \cdot t / N_{\text{Dir}} = 1.016$$

$$d_{fv} = d \text{ (figure 11.2, ACI 440) } = 370.00$$

$$f_{fe} \text{ ((11-5), ACI 440) } = 259.312$$

$$E_f = 64828.00$$

$$f_{e} = 0.004, \text{ from (11.6a), ACI 440}$$

$$\text{with } f_u = 0.01$$

$$\text{From (11-11), ACI 440: } V_s + V_f \leq 238930.50$$

$$b_w \cdot d = \frac{1}{4} \cdot d^2 = 80424.772$$

Calculation of Shear Strength at edge 2,  $V_{r2} = 351879.387$

$$V_{r2} = V_{\text{Col}} \text{ ((10.3), ASCE 41-17) } = k_{nl} \cdot V_{\text{ColO}}$$

$$V_{\text{ColO}} = 351879.387$$

$$k_{nl} = 1 \text{ (zero step-static loading)}$$

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$$= 1 \text{ (normal-weight concrete)}$$

$$f_c' = 20.00, \text{ but } f_c'^{0.5} \leq 8.3 \text{ MPa (22.5.3.1, ACI 318-14)}$$

$$M/Vd = 2.00$$

$$\mu_u = 9.8466009E-012$$

$$\mu_v = 3.7968243E-031$$

$$d = 0.8 \cdot D = 320.00$$

$$N_u = 4771.233$$

$$A_g = 125663.706$$

$$\text{From (11.5.4.8), ACI 318-14: } V_s = 175459.634$$

$$A_v = \frac{1}{2} \cdot A_{\text{stirrup}} = 123370.055$$

$$f_y = 444.4444$$

$$s = 100.00$$

$$V_s \text{ is multiplied by } \phi_{\text{col}} = 0.00$$

$$s/d = 0.3125$$

$$V_f \text{ ((11-3)-(11.4), ACI 440) } = 194961.134$$

$$f = 0.95, \text{ for fully-wrapped sections}$$

$$w_f/s_f = 1 \text{ (FRP strips adjacent to one another).}$$

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

$$\text{orientation 1: } \theta_1 = b_1 + 90^\circ = 90.00$$

$$V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, \theta_1)|), \text{ with:}$$

$$\text{total thickness per orientation, } t_{f1} = N_L \cdot t / N_{\text{Dir}} = 1.016$$

$$d_{fv} = d \text{ (figure 11.2, ACI 440) } = 370.00$$

$$f_{fe} \text{ ((11-5), ACI 440) } = 259.312$$

$$E_f = 64828.00$$

$$f_{e} = 0.004, \text{ from (11.6a), ACI 440}$$

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$

$b_w \cdot d = \frac{b \cdot d}{4} = 80424.772$

-----  
-----  
End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At local axis: 3

-----  
-----  
Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At Shear local axis: 2  
(Bending local axis: 3)  
Section Type: rccs

Constant Properties

-----  
Knowledge Factor,  $\phi = 0.85$

Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.4444$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,  
the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 555.5556$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.84055

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Inadequate Lap Length with  $l_o/l_{o,min} = 0.30$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

-----  
Stepwise Properties

At local axis: 2

EDGE -A-

Shear Force,  $V_a = 4.4296389E-033$

EDGE -B-

Shear Force,  $V_b = -4.4296389E-033$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{sl} = 0.00$

-Compression:  $A_{slc} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{sl,ten} = 1017.876$

-Compression:  $A_s l_{com} = 1017.876$   
-Middle:  $A_s l_{mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.27402927$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 96425.251$

with

$M_{pr1} = \text{Max}(M_{u1+}, M_{u1-}) = 1.4464E+008$

$M_{u1+} = 1.4464E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$M_{u1-} = 1.4464E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(M_{u2+}, M_{u2-}) = 1.4464E+008$

$M_{u2+} = 1.4464E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

$M_{u2-} = 1.4464E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

Calculation of  $M_{u1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$

$M_u = 1.4464E+008$

$\phi = 0.9250245$

$\lambda = 0.82105152$

error of function (3.68), Biskinis Phd = 31682.903

From 5A.2, TBDY:  $f_{cc} = f_c^* c = 36.81095$

conf. factor  $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y * \text{Min}(1, 1.25 * (l_b/l_d)^{2/3}) = 311.2087$

$l_b/l_d = 0.30$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

$= * \text{Min}(1, 1.25 * (l_b/l_d)^{2/3}) = 0.31060654$

Calculation of ratio  $l_b/l_d$

Inadequate Lap Length with  $l_b/l_d = 0.30$

Calculation of  $M_{u1-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$

$M_u = 1.4464E+008$

$\phi = 0.9250245$

$\lambda = 0.82105152$

error of function (3.68), Biskinis Phd = 31682.903

From 5A.2, TBDY:  $f_{cc} = f_c^* c = 36.81095$

conf. factor  $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y * \text{Min}(1, 1.25 * (l_b/l_d)^{2/3}) = 311.2087$

$l_b/l_d = 0.30$

$d_1 = 44.00$

R = 200.00  
v = 0.00155946  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.31060654

-----  
Calculation of ratio lb/d

-----  
Inadequate Lap Length with lb/d = 0.30  
-----  
-----

-----  
Calculation of Mu2+  
-----  
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.4464E+008

-----  
= 0.9250245  
' = 0.82105152  
error of function (3.68), Biskinis Phd = 31682.903  
From 5A.2, TBDY: fcc = fc\* c = 36.81095  
conf. factor c = 1.84055  
fc = 20.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 311.2087  
lb/d = 0.30  
d1 = 44.00  
R = 200.00  
v = 0.00155946  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.31060654

-----  
Calculation of ratio lb/d

-----  
Inadequate Lap Length with lb/d = 0.30  
-----  
-----

-----  
Calculation of Mu2-  
-----  
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.4464E+008

-----  
= 0.9250245  
' = 0.82105152  
error of function (3.68), Biskinis Phd = 31682.903  
From 5A.2, TBDY: fcc = fc\* c = 36.81095  
conf. factor c = 1.84055  
fc = 20.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 311.2087  
lb/d = 0.30  
d1 = 44.00  
R = 200.00  
v = 0.00155946  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.31060654

-----  
Calculation of ratio lb/d  
-----

Inadequate Lap Length with  $l_b/l_d = 0.30$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 351879.387$

Calculation of Shear Strength at edge 1,  $V_{r1} = 351879.387$

$V_{r1} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_n l V_{CoI0}$

$V_{CoI0} = 351879.387$

$k_n l = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v f_y d/s$ ' is replaced by ' $V_s + f V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

$f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 9.0356326E-012$

$V_u = 4.4296389E-033$

$d = 0.8D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 175459.634$

$A_v = \frac{1}{2} A_{stirrup} = 123370.055$

$f_y = 444.4444$

$s = 100.00$

$V_s$  is multiplied by  $\text{Col} = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) =  $194961.134$

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = b_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, a_1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L t / N_{oDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) =  $370.00$

$f_{fe}$  ((11-5), ACI 440) =  $259.312$

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$

$b_w d = \frac{1}{4} d^2 = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 351879.387$

$V_{r2} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_n l V_{CoI0}$

$V_{CoI0} = 351879.387$

$k_n l = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v f_y d/s$ ' is replaced by ' $V_s + f V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$= 1$  (normal-weight concrete)

$f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$\mu_u = 9.0356326E-012$

$V_u = 4.4296389E-033$

$d = 0.8D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 175459.634$

$A_v = \frac{1}{2} A_{stirrup} = 123370.055$

$f_y = 444.4444$

$$s = 100.00$$

Vs is multiplied by Col = 0.00

$$s/d = 0.3125$$

$$V_f((11-3)-(11.4), \text{ACI 440}) = 194961.134$$

f = 0.95, for fully-wrapped sections

wf/sf = 1 (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \cot \alpha) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = b1 + 90^\circ = 90.00$

$$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \theta)|), \text{ with:}$$

total thickness per orientation,  $t_{f1} = NL * t / \text{NoDir} = 1.016$

$$d_{fv} = d \text{ (figure 11.2, ACI 440)} = 370.00$$

$$f_{fe}((11-5), \text{ACI 440}) = 259.312$$

$$E_f = 64828.00$$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

$$\text{From (11-11), ACI 440: } V_s + V_f \leq 238930.50$$

$$b_w * d = \frac{b * d * d}{4} = 80424.772$$

-----  
End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At local axis: 2  
-----

-----  
Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1  
At local axis: 2

Integration Section: (a)

Section Type: rccs

Constant Properties

-----  
Knowledge Factor,  $\gamma = 0.85$

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.4444$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at  $135^\circ$ )

Longitudinal Bars With Ends Lapped Starting at the End Sections

Inadequate Lap Length with  $l_b / l_d = 0.30$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $\text{NoDir} = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$   
-----

Stepwise Properties  
-----

Bending Moment,  $M = 3.1953177E-011$   
 Shear Force,  $V2 = -3472.529$   
 Shear Force,  $V3 = -1.7790009E-014$   
 Axial Force,  $F = -4769.844$   
 Longitudinal Reinforcement Area Distribution (in 2 divisions)  
 -Tension:  $As_t = 1272.345$   
 -Compression:  $As_c = 1781.283$   
 Longitudinal Reinforcement Area Distribution (in 3 divisions)  
 -Tension:  $As_{,ten} = 1017.876$   
 -Compression:  $As_{,com} = 1017.876$   
 -Middle:  $As_{,mid} = 1017.876$   
 Mean Diameter of Tension Reinforcement,  $Db_L = 18.00$

Existing component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_{,R} = u = 0.04306525$   
 $u = y + p = 0.050665$

- Calculation of  $y$  -

$y = (M_y * L_s / 3) / E_{eff} = 0.008665$  ((4.29), Biskinis Phd)  
 $M_y = 1.3732E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 1500.00  
 From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 7.9240E+012$   
 $factor = 0.30$   
 $A_g = 125663.706$   
 $f_c' = 20.00$   
 $N = 4769.844$   
 $E_c * I_g = 2.6413E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 1.3732E+008$   
 $y = 7.5716672E-006$   
 $M_{y\_ten}$  (8c) =  $1.3732E+008$   
 $_{ten}$  (7c) = 72.23344  
 error of function (7c) = 0.00097593  
 $M_{y\_com}$  (8d) =  $3.7493E+008$   
 $_{com}$  (7d) = 69.91126  
 error of function (7d) = 0.00301342  
 with ((10.1), ASCE 41-17)  $e_y = \text{Min}(e_y, 1.25 * e_y * (l_b / l_d)^{2/3}) = 0.00222222$   
 $e_{co} = 0.002$   
 $a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00157305$   
 $N = 4769.844$   
 $A_c = 125663.706$   
 ((10.1), ASCE 41-17) =  $\text{Min}( , 1.25 * * (l_b / l_d)^{2/3}) = 0.44758025$   
 with  $f_c^*$  ((12.3), ACI 440) = 24.12975  
 $f_c = 20.00$   
 $f_l = 1.3173$   
 $k = 1$   
 Effective FRP thickness,  $t_f = N * t * \text{Cos}(b_1) = 1.016$   
 $e_{fe}$  ((12.5) and (12.7)) = 0.004  
 $f_u = 0.01$   
 $E_f = 64828.00$

Calculation of ratio  $l_b / l_d$

Inadequate Lap Length with  $l_b/l_d = 0.30$

- Calculation of  $\rho$  -

From table 10-9:  $\rho = 0.042$

with:

- Columns not controlled by inadequate development or splicing along the clear height because  $l_b/l_d \geq 1$

shear control ratio  $V_y E / V_{CoI} E = 0.27402927$

$d = 0.00$

$s = 0.00$

$t = 2 \cdot A_v / (d_c \cdot s) + 4 \cdot t_f / D \cdot (f_{fe} / f_s) = 0.00$

$A_v = 78.53982$ , is the area of the circular stirrup

$d_c = D - 2 \cdot \text{cover} - \text{Hoop Diameter} = 340.00$

The term  $2 \cdot t_f / b_w \cdot (f_{fe} / f_s)$  is implemented to account for FRP contribution

where  $f = 2 \cdot t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength

All these variables have already been given in Shear control ratio calculation.

$NUD = 4769.844$

$A_g = 125663.706$

$f_{cE} = 20.00$

$f_{ytE} = f_{ylE} = 444.4444$

$\rho_l = \text{Area\_Tot\_Long\_Rein} / (A_g) = 0.0243$

$f_{cE} = 20.00$

End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (a)

## Calculation No. 11

column C1, Floor 1

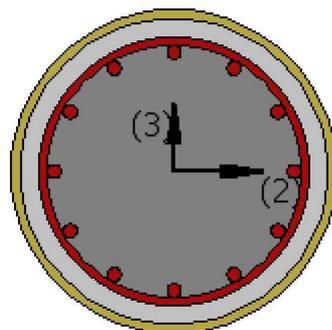
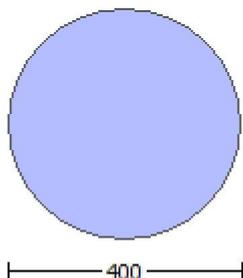
Limit State: Collapse Prevention (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Shear capacity  $V_{Rd}$

Edge: Start

Local Axis: (3)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (a)

Section Type: rccs

### Constant Properties

Knowledge Factor,  $\gamma = 0.85$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 16.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 400.00$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of  $\gamma$  for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE41-17).

Existing material: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material: Steel Strength,  $f_s = f_{sm} = 444.4444$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Inadequate Lap Length with  $l_o/l_{o,min} = l_b/l_d = 0.30$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $NoDir = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

### Stepwise Properties

EDGE -A-

Bending Moment,  $M_a = 3.1953177E-011$

Shear Force,  $V_a = -1.7790009E-014$

EDGE -B-

Bending Moment,  $M_b = 2.1330954E-011$

Shear Force,  $V_b = 1.7790009E-014$

BOTH EDGES

Axial Force,  $F = -4769.844$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 1272.345$

-Compression:  $As_c = 1781.283$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{l,ten} = 1017.876$

-Compression:  $As_{l,com} = 1017.876$

-Middle:  $As_{l,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $Db_{L,ten} = 18.00$

Existing component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = V_n = 267605.553$   
 $V_n$  ((10.3), ASCE 41-17) =  $k_n I V_{CoIO} = 314830.062$   
 $V_{CoI} = 314830.062$   
 $k_n = 1.00$   
displacement\_ductility\_demand = 0.00

NOTE: In expression (10-3) ' $V_s = A_v f_y d/s$ ' is replaced by ' $V_s + f V_f$ '  
where  $V_f$  is the contribution of FRPs ((11.3), ACI 440).

= 1 (normal-weight concrete)  
 $f_c' = 16.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $M_u = 3.1953177E-011$   
 $V_u = 1.7790009E-014$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4769.844$   
 $A_g = 125663.706$   
From (11.5.4.8), ACI 318-14:  $V_s = 157913.67$   
 $A_v = A_s / 2 = 123370.055$   
 $f_y = 400.00$   
 $s = 100.00$   
 $V_s$  is multiplied by  $\phi_{col} = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
In (11.3)  $\sin^2 + \cos^2$  is replaced with  $(\cot^2 + \csc^2) \sin^2 \alpha$  which is more a generalised expression,  
where  $\alpha$  is the angle of the crack direction (see KANEPE).  
This later relation, considered as a function  $V_f(\alpha, a_i)$ , is implemented for every different fiber orientation  $a_i$ ,  
as well as for 2 crack directions,  $\alpha = 45^\circ$  and  $\alpha = -45^\circ$  to take into consideration the cyclic seismic loading.  
orientation 1:  $\alpha = 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, a_1)|, |V_f(-45, a_1)|)$ , with:  
total thickness per orientation,  $t_{f1} = N L \cdot t / N_{oDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}$  ((11-5), ACI 440) = 259.312  
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
with  $f_u = 0.01$   
From (11-11), ACI 440:  $V_s + V_f \leq 213705.936$   
 $b_w \cdot d = A_s \cdot d / 4 = 80424.772$

displacement\_ductility\_demand is calculated as  $\delta / y$

- Calculation of  $\delta / y$  for END A -  
for rotation axis 2 and integ. section (a)

From analysis, chord rotation  $\theta = 5.9951039E-021$   
 $y = (M_y \cdot L_s / 3) / E_{eff} = 0.008665$  ((4.29), Biskinis Phd))  
 $M_y = 1.3732E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) = 1500.00  
From table 10.5, ASCE 41\_17:  $E_{eff} = \text{factor} \cdot E_c \cdot I_g = 7.9240E+012$   
factor = 0.30  
 $A_g = 125663.706$   
 $f_c' = 20.00$   
 $N = 4769.844$   
 $E_c \cdot I_g = 2.6413E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\delta / y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 1.3732E+008$

$y = 7.5716672E-006$   
 $My_{ten} (8c) = 1.3732E+008$   
 $_{ten} (7c) = 72.23344$   
error of function (7c) = 0.00097593  
 $My_{com} (8d) = 3.7493E+008$   
 $_{com} (7d) = 69.91126$   
error of function (7d) = 0.00301342  
with ((10.1), ASCE 41-17)  $e_y = \text{Min}(e_y, 1.25 * e_y * (l_b / l_d)^{2/3}) = 0.00222222$   
 $e_{co} = 0.002$   
 $a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00157305$   
 $N = 4769.844$   
 $A_c = 125663.706$   
((10.1), ASCE 41-17)  $= \text{Min}( , 1.25 * * (l_b / l_d)^{2/3}) = 0.44758025$   
with  $f_c^*$  ((12.3), ACI 440) = 24.12975  
 $f_c = 20.00$   
 $f_l = 1.3173$   
 $k = 1$   
Effective FRP thickness,  $t_f = NL * t * \text{Cos}(b_1) = 1.016$   
 $e_{fe}$  ((12.5) and (12.7)) = 0.004  
 $f_u = 0.01$   
 $E_f = 64828.00$

-----  
-----  
Calculation of ratio  $l_b / l_d$

-----  
Inadequate Lap Length with  $l_b / l_d = 0.30$   
-----

End Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (a)  
-----

## Calculation No. 12

column C1, Floor 1

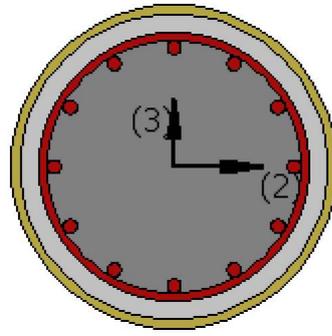
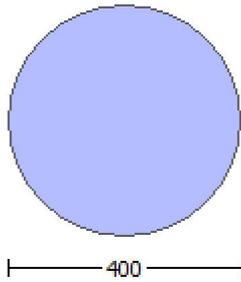
Limit State: Collapse Prevention (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity (  $\theta_u$  )

Edge: Start

Local Axis: (3)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 3  
 (Bending local axis: 2)  
 Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 0.85$   
 Mean strength values are used for both shear and moment calculations.  
 Consequently:  
 Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$   
 Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.4444$   
 Concrete Elasticity,  $E_c = 21019.039$   
 Steel Elasticity,  $E_s = 200000.00$

#####  
 Note: Especially for the calculation of moment strengths,  
 the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14  
 Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 555.5556$

#####  
 Diameter,  $D = 400.00$   
 Cover Thickness,  $c = 25.00$   
 Mean Confinement Factor overall section = 1.84055  
 Element Length,  $L = 3000.00$   
 Primary Member  
 Smooth Bars  
 Ductile Steel  
 With Detailing for Earthquake Resistance (including stirrups closed at 135°)  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Inadequate Lap Length with  $l_o/l_{ou,min} = 0.30$   
 FRP Wrapping Data  
 Type: Carbon  
 Cured laminate properties (design values)  
 Thickness,  $t = 1.016$   
 Tensile Strength,  $f_{fu} = 1055.00$   
 Tensile Modulus,  $E_f = 64828.00$   
 Elongation,  $e_{fu} = 0.01$   
 Number of directions,  $N_{oDir} = 1$   
 Fiber orientations,  $b_i: 0.00^\circ$   
 Number of layers,  $N_L = 1$   
 Radius of rounding corners,  $R = 40.00$

Stepwise Properties

At local axis: 3  
 EDGE -A-  
 Shear Force,  $V_a = -3.7968243E-031$   
 EDGE -B-  
 Shear Force,  $V_b = 3.7968243E-031$   
 BOTH EDGES  
 Axial Force,  $F = -4771.233$   
 Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension: Aslt = 0.00

-Compression: Aslc = 3053.628

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension: Asl,ten = 1017.876

-Compression: Asl,com = 1017.876

-Middle: Asl,mid = 1017.876

Calculation of Shear Capacity ratio ,  $V_e/V_r = 0.27402927$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 96425.251$

with

$M_{pr1} = \text{Max}(M_{u1+} , M_{u1-}) = 1.4464E+008$

$M_{u1+} = 1.4464E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$M_{u1-} = 1.4464E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(M_{u2+} , M_{u2-}) = 1.4464E+008$

$M_{u2+} = 1.4464E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

$M_{u2-} = 1.4464E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

Calculation of  $M_{u1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$

$M_u = 1.4464E+008$

= 0.9250245

' = 0.82105152

error of function (3.68), Biskinis Phd = 31682.903

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$

conf. factor  $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 311.2087$

$l_b/d = 0.30$

$d_1 = 44.00$

$R = 200.00$

$v = 0.00155946$

$N = 4771.233$

$A_c = 125663.706$

=  $\cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.31060654$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of  $M_{u1-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$

$M_u = 1.4464E+008$

= 0.9250245

' = 0.82105152

error of function (3.68), Biskinis Phd = 31682.903

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$

conf. factor  $c = 1.84055$

$$f_c = 20.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 311.2087$

$$l_b/d = 0.30$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00155946$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.31060654$$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of  $\mu_{2+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$$\mu = 1.4464E+008$$

$$= 0.9250245$$

$$' = 0.82105152$$

error of function (3.68), Biskinis Phd = 31682.903

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$

$$\text{conf. factor } c = 1.84055$$

$$f_c = 20.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 311.2087$

$$l_b/d = 0.30$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00155946$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.31060654$$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of  $\mu_{2-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$$\mu = 1.4464E+008$$

$$= 0.9250245$$

$$' = 0.82105152$$

error of function (3.68), Biskinis Phd = 31682.903

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$

$$\text{conf. factor } c = 1.84055$$

$$f_c = 20.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 311.2087$

$$l_b/d = 0.30$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00155946$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.31060654$$

Calculation of ratio  $l_b/l_d$

Inadequate Lap Length with  $l_b/l_d = 0.30$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 351879.387$

Calculation of Shear Strength at edge 1,  $V_{r1} = 351879.387$

$V_{r1} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_n l V_{Col0}$

$V_{Col0} = 351879.387$

$k_n l = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v f_y d/s$ ' is replaced by ' $V_s + f^* V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\lambda = 1$  (normal-weight concrete)

$f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$M_u = 9.8466009E-012$

$V_u = 3.7968243E-031$

$d = 0.8D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 175459.634$

$A_v = \lambda^2 A_{stirrup} = 123370.055$

$f_y = 444.4444$

$s = 100.00$

$V_s$  is multiplied by  $\lambda_{Col} = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) =  $194961.134$

$f = 0.95$ , for fully-wrapped sections

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = 45^\circ + 90^\circ = 135^\circ$

$V_f = \text{Min}(|V_f(45^\circ, \alpha_1)|, |V_f(-45^\circ, \alpha_1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L t / N_{Dir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) =  $370.00$

$f_{fe}$  ((11-5), ACI 440) =  $259.312$

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$

$b_w d = \lambda^2 d^2 / 4 = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 351879.387$

$V_{r2} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_n l V_{Col0}$

$V_{Col0} = 351879.387$

$k_n l = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v f_y d/s$ ' is replaced by ' $V_s + f^* V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\lambda = 1$  (normal-weight concrete)

$f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M/Vd = 2.00$

$M_u = 9.8466009E-012$

$V_u = 3.7968243E-031$

$d = 0.8D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 175459.634$   
 $A_v = \sqrt{2} \cdot A_{stirrup} = 123370.055$   
 $f_y = 444.4444$   
 $s = 100.00$   
 $V_s$  is multiplied by  $Col = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin^2 + \cos^2$  is replaced with  $(\cot^2 + \cot^2)\sin^2\alpha$  which is more a generalised expression,  
 where  $\alpha$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ ,  
 as well as for 2 crack directions,  $\alpha = 45^\circ$  and  $\alpha = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\alpha_1 = \alpha_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \alpha_1)|, |V_f(-45, \alpha_1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}$  ((11-5), ACI 440) = 259.312  
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $b_w \cdot d = A_v \cdot s / 4 = 80424.772$

-----  
 End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At local axis: 3

-----  
 Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At Shear local axis: 2  
 (Bending local axis: 3)  
 Section Type: rccs

Constant Properties

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 Knowledge Factor,  $\phi = 0.85$   
 Mean strength values are used for both shear and moment calculations.  
 Consequently:  
 Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$   
 Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.4444$   
 Concrete Elasticity,  $E_c = 21019.039$   
 Steel Elasticity,  $E_s = 200000.00$   
 #####  
 Note: Especially for the calculation of moment strengths,  
 the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14  
 Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 555.5556$   
 #####  
 Diameter,  $D = 400.00$   
 Cover Thickness,  $c = 25.00$   
 Mean Confinement Factor overall section = 1.84055  
 Element Length,  $L = 3000.00$   
 Primary Member  
 Smooth Bars  
 Ductile Steel  
 With Detailing for Earthquake Resistance (including stirrups closed at 135°)  
 Longitudinal Bars With Ends Lapped Starting at the End Sections  
 Inadequate Lap Length with  $l_o/l_{ou, \min} = 0.30$   
 FRP Wrapping Data  
 Type: Carbon  
 Cured laminate properties (design values)  
 Thickness,  $t = 1.016$   
 Tensile Strength,  $f_{fu} = 1055.00$   
 Tensile Modulus,  $E_f = 64828.00$

Elongation,  $\epsilon_{fu} = 0.01$   
Number of directions, NoDir = 1  
Fiber orientations,  $b_i = 0.00^\circ$   
Number of layers, NL = 1  
Radius of rounding corners, R = 40.00

-----  
Stepwise Properties  
-----

At local axis: 2  
EDGE -A-  
Shear Force,  $V_a = 4.4296389E-033$   
EDGE -B-  
Shear Force,  $V_b = -4.4296389E-033$   
BOTH EDGES  
Axial Force,  $F = -4771.233$   
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $A_{sl,t} = 0.00$   
-Compression:  $A_{sl,c} = 3053.628$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $A_{sl,ten} = 1017.876$   
-Compression:  $A_{sl,com} = 1017.876$   
-Middle:  $A_{sl,mid} = 1017.876$

-----  
-----  
Calculation of Shear Capacity ratio ,  $V_e/V_r = 0.27402927$   
Member Controlled by Flexure ( $V_e/V_r < 1$ )  
Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 96425.251$   
with  
 $M_{pr1} = \text{Max}(\mu_{u1+}, \mu_{u1-}) = 1.4464E+008$   
 $\mu_{u1+} = 1.4464E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction  
which is defined for the static loading combination  
 $\mu_{u1-} = 1.4464E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment  
direction which is defined for the static loading combination  
 $M_{pr2} = \text{Max}(\mu_{u2+}, \mu_{u2-}) = 1.4464E+008$   
 $\mu_{u2+} = 1.4464E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction  
which is defined for the the static loading combination  
 $\mu_{u2-} = 1.4464E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment  
direction which is defined for the the static loading combination

-----  
Calculation of  $\mu_{u1+}$   
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$   
 $\mu_u = 1.4464E+008$

-----  
= 0.9250245  
' = 0.82105152  
error of function (3.68), Biskinis Phd = 31682.903  
From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 311.2087$   
 $l_b/d = 0.30$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
=  $\text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.31060654$

-----  
Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/l_d = 0.30$

Calculation of  $\mu_{1-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 1.4464E+008$

$$= 0.9250245$$

$$' = 0.82105152$$

error of function (3.68), Biskinis Phd = 31682.903

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$

conf. factor  $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 311.2087$

$$l_b/l_d = 0.30$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00155946$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.31060654$$

Calculation of ratio  $l_b/l_d$

Inadequate Lap Length with  $l_b/l_d = 0.30$

Calculation of  $\mu_{2+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$   
 $\mu = 1.4464E+008$

$$= 0.9250245$$

$$' = 0.82105152$$

error of function (3.68), Biskinis Phd = 31682.903

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$

conf. factor  $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 311.2087$

$$l_b/l_d = 0.30$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00155946$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.31060654$$

Calculation of ratio  $l_b/l_d$

Inadequate Lap Length with  $l_b/l_d = 0.30$

Calculation of  $\mu_{2-}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $M_u$   
 $M_u = 1.4464E+008$

$$= 0.9250245$$

$$' = 0.82105152$$

error of function (3.68), Biskinis Phd = 31682.903

From 5A.2, TBDY:  $f_{cc} = f_c' \cdot c = 36.81095$

$$\text{conf. factor } c = 1.84055$$

$$f_c = 20.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 311.2087$

$$l_b/d = 0.30$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00155946$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.31060654$$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 351879.387$

Calculation of Shear Strength at edge 1,  $V_{r1} = 351879.387$

$V_{r1} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{Col0}$

$$V_{Col0} = 351879.387$$

$$k_{nl} = 1 \text{ (zero step-static loading)}$$

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

$f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3 \text{ MPa}$  (22.5.3.1, ACI 318-14)

$$M/Vd = 2.00$$

$$M_u = 9.0356326E-012$$

$$V_u = 4.4296389E-033$$

$$d = 0.8 \cdot D = 320.00$$

$$N_u = 4771.233$$

$$A_g = 125663.706$$

From (11.5.4.8), ACI 318-14:  $V_s = 175459.634$

$$A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 123370.055$$

$$f_y = 444.4444$$

$$s = 100.00$$

$V_s$  is multiplied by  $\text{Col} = 0.00$

$$s/d = 0.3125$$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$$f = 0.95, \text{ for fully-wrapped sections}$$

$w_f/s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression, where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = b_1 + 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \theta)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{\text{Dir}} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$$E_f = 64828.00$$

$$f_e = 0.004, \text{ from (11.6a), ACI 440}$$

$$\text{with } f_u = 0.01$$

From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$

$$b_w \cdot d = \cdot d \cdot d / 4 = 80424.772$$

Calculation of Shear Strength at edge 2,  $V_{r2} = 351879.387$

$V_{r2} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_{nl} * V_{Col0}$

$V_{Col0} = 351879.387$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

$f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)

$M / Vd = 2.00$

$\mu_u = 9.0356326E-012$

$\nu_u = 4.4296389E-033$

$d = 0.8 * D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From (11.5.4.8), ACI 318-14:  $V_s = 175459.634$

$A_v = \sqrt{2} * A_{stirrup} = 123370.055$

$f_y = 444.4444$

$s = 100.00$

$V_s$  is multiplied by  $\phi_{col} = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections

$w_f / s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression,

where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = 45^\circ$  and  $\alpha = 90^\circ$

$V_f = \text{Min}(|V_f(45, 90)|, |V_f(-45, 90)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L * t / N_{oDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_{e} = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$

$b_w * d = \sqrt{2} * d^2 / 4 = 80424.772$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At local axis: 2

Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (a)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\phi = 0.85$

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.4444$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member  
Smooth Bars  
Ductile Steel  
With Detailing for Earthquake Resistance (including stirrups closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Inadequate Lap Length with  $I_b/I_d = 0.30$   
FRP Wrapping Data  
Type: Carbon  
Cured laminate properties (design values)  
Thickness,  $t = 1.016$   
Tensile Strength,  $f_{fu} = 1055.00$   
Tensile Modulus,  $E_f = 64828.00$   
Elongation,  $e_{fu} = 0.01$   
Number of directions,  $NoDir = 1$   
Fiber orientations,  $bi = 0.00^\circ$   
Number of layers,  $NL = 1$   
Radius of rounding corners,  $R = 40.00$

-----  
Stepwise Properties

-----  
Bending Moment,  $M = -1.0423E+007$   
Shear Force,  $V2 = -3472.529$   
Shear Force,  $V3 = -1.7790009E-014$   
Axial Force,  $F = -4769.844$   
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $As_t = 1272.345$   
-Compression:  $As_c = 1781.283$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $As_{,ten} = 1017.876$   
-Compression:  $As_{,com} = 1017.876$   
-Middle:  $As_{,mid} = 1017.876$   
Mean Diameter of Tension Reinforcement,  $Db_L = 18.00$

-----  
Existing component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_{,R} = \alpha \cdot u = 0.0504376$   
 $u = \gamma + \rho = 0.05933835$

-----  
- Calculation of  $\gamma$  -

-----  
 $\gamma = (M \cdot L_s / 3) / E_{eff} = 0.01733835$  ((4.29), Biskinis Phd))  
 $M_y = 1.3732E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) = 3001.447  
From table 10.5, ASCE 41\_17:  $E_{eff} = factor \cdot E_c \cdot I_g = 7.9240E+012$   
factor = 0.30  
 $A_g = 125663.706$   
 $f_c' = 20.00$   
 $N = 4769.844$   
 $E_c \cdot I_g = 2.6413E+013$

-----  
Calculation of Yielding Moment  $M_y$

-----  
Calculation of  $\gamma$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

-----  
 $M_y = \min(M_{y\_ten}, M_{y\_com}) = 1.3732E+008$   
 $\gamma = 7.5716672E-006$   
 $M_{y\_ten} (8c) = 1.3732E+008$   
 $\gamma_{ten} (7c) = 72.23344$   
error of function (7c) = 0.00097593  
 $M_{y\_com} (8d) = 3.7493E+008$   
 $\gamma_{com} (7d) = 69.91126$   
error of function (7d) = 0.00301342

with ((10.1), ASCE 41-17)  $e_y = \text{Min}(e_y, 1.25 * e_y * (l_b / l_d)^{2/3}) = 0.002222222$

$e_{co} = 0.002$

$a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)

$d_1 = 44.00$

$R = 200.00$

$v = 0.00157305$

$N = 4769.844$

$A_c = 125663.706$

((10.1), ASCE 41-17)  $= \text{Min}( , 1.25 * (l_b / l_d)^{2/3}) = 0.44758025$

with  $f_c^*$  ((12.3), ACI 440) = 24.12975

$f_c = 20.00$

$f_l = 1.3173$

$k = 1$

Effective FRP thickness,  $t_f = N L * t * \text{Cos}(b_1) = 1.016$

$e_{fe}$  ((12.5) and (12.7)) = 0.004

$f_u = 0.01$

$E_f = 64828.00$

-----  
-----  
Calculation of ratio  $l_b / l_d$

-----  
-----  
Inadequate Lap Length with  $l_b / l_d = 0.30$

-----  
-----  
- Calculation of  $p$  -

-----  
-----  
From table 10-9:  $p = 0.042$

with:

- Columns not controlled by inadequate development or splicing along the clear height because  $l_b / l_d \geq 1$

shear control ratio  $V_y E / V_{CoI} E = 0.27402927$

$d = 0.00$

$s = 0.00$

$t = 2 * A_v / (d_c * s) + 4 * t_f / D * (f_{fe} / f_s) = 0.00$

$A_v = 78.53982$ , is the area of the circular stirrup

$d_c = D - 2 * \text{cover} - \text{Hoop Diameter} = 340.00$

The term  $2 * t_f / b_w * (f_{fe} / f_s)$  is implemented to account for FRP contribution

where  $f = 2 * t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength

All these variables have already been given in Shear control ratio calculation.

$N_{UD} = 4769.844$

$A_g = 125663.706$

$f_{cE} = 20.00$

$f_{ytE} = f_{ylE} = 444.4444$

$p_l = \text{Area\_Tot\_Long\_Rein} / (A_g) = 0.0243$

$f_{cE} = 20.00$

-----  
-----  
End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (a)

-----  
-----  
**Calculation No. 13**

column C1, Floor 1

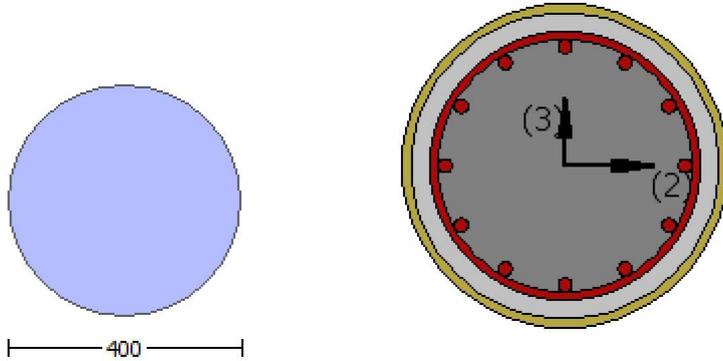
Limit State: Collapse Prevention (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Shear capacity VRd

Edge: End

Local Axis: (2)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (b)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 0.85$

Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE41-17.

Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 16.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 400.00$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of  $\gamma$  for displacement ductility demand, the expected (mean value) strengths are used (7.5.1.3, ASCE41-17) because bending is considered as Deformation-Controlled Action (Table C7-1, ASCE41-17).

Existing material: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material: Steel Strength,  $f_s = f_{sm} = 444.4444$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Inadequate Lap Length with  $l_o/l_{ou,min} = l_b/l_d = 0.30$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $NoDir = 1$

Fiber orientations,  $\theta_i$ : 0.00°  
Number of layers, NL = 1  
Radius of rounding corners, R = 40.00

-----  
Stepwise Properties

-----  
EDGE -A-  
Bending Moment,  $M_a$  = -1.0423E+007  
Shear Force,  $V_a$  = -3472.529  
EDGE -B-  
Bending Moment,  $M_b$  = 0.09482631  
Shear Force,  $V_b$  = 3472.529  
BOTH EDGES  
Axial Force, F = -4769.844  
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $A_{st}$  = 0.00  
-Compression:  $A_{sc}$  = 3053.628  
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $A_{st,ten}$  = 1017.876  
-Compression:  $A_{sc,com}$  = 1017.876  
-Middle:  $A_{st,mid}$  = 1017.876  
Mean Diameter of Tension Reinforcement,  $D_{bL,ten}$  = 18.00  
-----

-----  
Existing component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = \phi V_n = 267605.553$   
 $V_n$  ((10.3), ASCE 41-17) =  $k_n V_{CoI} = 314830.062$   
 $V_{CoI} = 314830.062$   
 $k_n = 1.00$   
 $displacement\_ductility\_demand = 0.11383644$   
-----

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + \phi V_f$ '  
where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

-----  
= 1 (normal-weight concrete)  
 $f_c' = 16.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $M_u = 0.09482631$   
 $V_u = 3472.529$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4769.844$   
 $A_g = 125663.706$   
From (11.5.4.8), ACI 318-14:  $V_s = 157913.67$   
 $A_v = \phi / 2 \cdot A_{stirrup} = 123370.055$   
 $f_y = 400.00$   
 $s = 100.00$   
 $V_s$  is multiplied by  $\phi_{CoI} = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $\phi = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression,  
where  $\theta$  is the angle of the crack direction (see KANEPE).  
This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ ,  
as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
orientation 1:  $\theta_1 = \theta_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta_1)|, |V_f(-45, \theta_1)|)$ , with:  
total thickness per orientation,  $t_{f1} = NL \cdot t / NoDir = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}$  ((11-5), ACI 440) = 259.312  
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
with  $f_u = 0.01$   
From (11-11), ACI 440:  $V_s + V_f \leq 213705.936$   
 $b_w \cdot d = \phi \cdot d^2 / 4 = 80424.772$

displacement\_ductility\_demand is calculated as  $\phi / y$

- Calculation of  $\phi / y$  for END B -  
for rotation axis 3 and integ. section (b)

From analysis, chord rotation = 0.00019728  
 $y = (M_y * L_s / 3) / E_{eff} = 0.001733$  ((4.29), Biskinis Phd)  
 $M_y = 1.3732E+008$   
 $L_s = M/V$  (with  $L_s > 0.1 * L$  and  $L_s < 2 * L$ ) = 300.00  
From table 10.5, ASCE 41\_17:  $E_{eff} = factor * E_c * I_g = 7.9240E+012$   
factor = 0.30  
 $A_g = 125663.706$   
 $f_c' = 20.00$   
 $N = 4769.844$   
 $E_c * I_g = 2.6413E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\phi$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 1.3732E+008$   
 $y = 7.5716672E-006$   
 $M_{y\_ten}$  (8c) =  $1.3732E+008$   
 $\phi_{ten}$  (7c) = 72.23344  
error of function (7c) = 0.00097593  
 $M_{y\_com}$  (8d) =  $3.7493E+008$   
 $\phi_{com}$  (7d) = 69.91126  
error of function (7d) = 0.00301342  
with ((10.1), ASCE 41-17)  $e_y = \text{Min}(e_y, 1.25 * e_y * (l_b / l_d)^{2/3}) = 0.00222222$   
 $e_{co} = 0.002$   
 $a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)  
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00157305$   
 $N = 4769.844$   
 $A_c = 125663.706$   
((10.1), ASCE 41-17)  $\phi = \text{Min}(\phi, 1.25 * \phi * (l_b / l_d)^{2/3}) = 0.44758025$   
with  $f_c'$  ((12.3), ACI 440) = 24.12975  
 $f_l = 20.00$   
 $f_l = 1.3173$   
 $k = 1$   
Effective FRP thickness,  $t_f = NL * t * \text{Cos}(b_1) = 1.016$   
 $e_{fe}$  ((12.5) and (12.7)) = 0.004  
 $f_u = 0.01$   
 $E_f = 64828.00$

Calculation of ratio  $l_b / l_d$

Inadequate Lap Length with  $l_b / l_d = 0.30$

End Of Calculation of Shear Capacity for element: column CC1 of floor 1  
At local axis: 2  
Integration Section: (b)

## Calculation No. 14

column C1, Floor 1

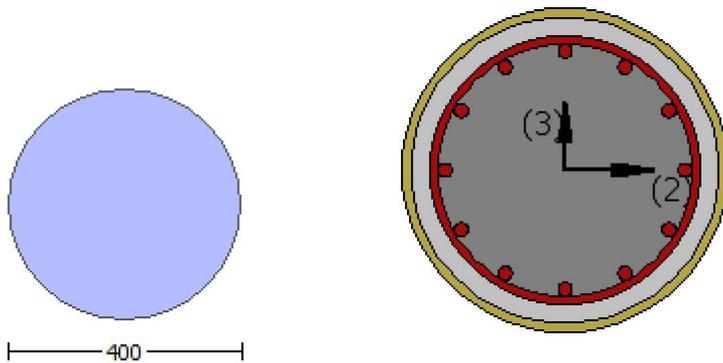
Limit State: Collapse Prevention (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity ( $\phi$ )

Edge: End

Local Axis: (2)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\phi = 0.85$

Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.4444$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 555.5556$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.84055

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Inadequate Lap Length with  $l_o/l_{ou,min} = 0.30$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$   
Tensile Strength,  $f_{fu} = 1055.00$   
Tensile Modulus,  $E_f = 64828.00$   
Elongation,  $e_{fu} = 0.01$   
Number of directions,  $NoDir = 1$   
Fiber orientations,  $bi = 0.00^\circ$   
Number of layers,  $NL = 1$   
Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

At local axis: 3  
EDGE -A-  
Shear Force,  $V_a = -3.7968243E-031$   
EDGE -B-  
Shear Force,  $V_b = 3.7968243E-031$   
BOTH EDGES  
Axial Force,  $F = -4771.233$   
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $As_t = 0.00$   
-Compression:  $As_c = 3053.628$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $As_{t,ten} = 1017.876$   
-Compression:  $As_{l,com} = 1017.876$   
-Middle:  $As_{l,mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.27402927$   
Member Controlled by Flexure ( $V_e/V_r < 1$ )  
Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 96425.251$   
with

$M_{pr1} = \text{Max}(\mu_{u1+}, \mu_{u1-}) = 1.4464E+008$   
 $\mu_{u1+} = 1.4464E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination  
 $\mu_{u1-} = 1.4464E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination  
 $M_{pr2} = \text{Max}(\mu_{u2+}, \mu_{u2-}) = 1.4464E+008$   
 $\mu_{u2+} = 1.4464E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination  
 $\mu_{u2-} = 1.4464E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

#### Calculation of $\mu_{u1+}$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$   
 $\mu_u = 1.4464E+008$

$\beta_1 = 0.9250245$   
 $\beta_2 = 0.82105152$   
error of function (3.68), Biskinis Phd = 31682.903  
From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 311.2087$   
 $l_b/d = 0.30$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.31060654$

-----  
Calculation of ratio lb/d  
-----

Inadequate Lap Length with lb/d = 0.30  
-----  
-----  
-----

Calculation of Mu1-  
-----  
-----

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.4464E+008  
-----

= 0.9250245  
' = 0.82105152  
error of function (3.68), Biskinis Phd = 31682.903  
From 5A.2, TBDY: fcc = fc\* c = 36.81095  
conf. factor c = 1.84055  
fc = 20.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 311.2087  
lb/d = 0.30  
d1 = 44.00  
R = 200.00  
v = 0.00155946  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.31060654  
-----

Calculation of ratio lb/d  
-----

Inadequate Lap Length with lb/d = 0.30  
-----  
-----  
-----

Calculation of Mu2+  
-----  
-----

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.4464E+008  
-----

= 0.9250245  
' = 0.82105152  
error of function (3.68), Biskinis Phd = 31682.903  
From 5A.2, TBDY: fcc = fc\* c = 36.81095  
conf. factor c = 1.84055  
fc = 20.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 311.2087  
lb/d = 0.30  
d1 = 44.00  
R = 200.00  
v = 0.00155946  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.31060654  
-----

Calculation of ratio lb/d  
-----

Inadequate Lap Length with lb/d = 0.30  
-----  
-----  
-----

Calculation of Mu2-  
-----

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.4464E+008

= 0.9250245  
' = 0.82105152  
error of function (3.68), Biskinis Phd = 31682.903  
From 5A.2, TBDY: fcc = fc\* c = 36.81095  
conf. factor c = 1.84055  
fc = 20.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 311.2087  
lb/d = 0.30  
d1 = 44.00  
R = 200.00  
v = 0.00155946  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.31060654

Calculation of ratio lb/d

Inadequate Lap Length with lb/d = 0.30

Calculation of Shear Strength Vr = Min(Vr1,Vr2) = 351879.387

Calculation of Shear Strength at edge 1, Vr1 = 351879.387

Vr1 = VCol ((10.3), ASCE 41-17) = knl\*VColO

VColO = 351879.387

knl = 1 (zero step-static loading)

NOTE: In expression (10-3) 'Vs = Av\*fy\*d/s' is replaced by 'Vs+ f\*Vf' where Vf is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)  
fc' = 20.00, but fc^0.5 <= 8.3 MPa (22.5.3.1, ACI 318-14)  
M/d = 2.00  
Mu = 9.8466009E-012  
Vu = 3.7968243E-031  
d = 0.8\*D = 320.00  
Nu = 4771.233  
Ag = 125663.706  
From (11.5.4.8), ACI 318-14: Vs = 175459.634  
Av = /2\*A\_stirrup = 123370.055  
fy = 444.4444  
s = 100.00  
Vs is multiplied by Col = 0.00  
s/d = 0.3125  
Vf ((11-3)-(11.4), ACI 440) = 194961.134  
f = 0.95, for fully-wrapped sections  
wf/sf = 1 (FRP strips adjacent to one another).  
In (11.3) sin + cos is replaced with (cot +cota)sina which is more a generalised expression, where is the angle of the crack direction (see KANEPE).  
This later relation, considered as a function Vf( , ), is implemented for every different fiber orientation ai, as well as for 2 crack directions, = 45° and = -45° to take into consideration the cyclic seismic loading.  
orientation 1: 1 = b1 + 90° = 90.00  
Vf = Min(|Vf(45, 1)|,|Vf(-45,a1)|), with:  
total thickness per orientation, tf1 = NL\*t/NoDir = 1.016  
dfv = d (figure 11.2, ACI 440) = 370.00  
ffe ((11-5), ACI 440) = 259.312  
Ef = 64828.00  
fe = 0.004, from (11.6a), ACI 440

with  $f_u = 0.01$   
From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $b_w \cdot d = \frac{1}{4} \cdot d^2 = 80424.772$

-----  
Calculation of Shear Strength at edge 2,  $V_{r2} = 351879.387$   
 $V_{r2} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_n \cdot V_{Col0}$   
 $V_{Col0} = 351879.387$   
 $k_n = 1$  (zero step-static loading)

-----  
NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ '  
where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

-----  
 $\lambda = 1$  (normal-weight concrete)  
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M / Vd = 2.00$   
 $\mu_u = 9.8466009E-012$   
 $V_u = 3.7968243E-031$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
From (11.5.4.8), ACI 318-14:  $V_s = 175459.634$   
 $A_v = \frac{1}{2} \cdot A_{stirrup} = 123370.055$   
 $f_y = 444.4444$   
 $s = 100.00$

$V_s$  is multiplied by  $\lambda = 0.00$   
 $s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections  
 $w_f / s_f = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression,  
where  $\theta$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ ,  
as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $\theta = 45^\circ + 90^\circ = 135^\circ$

$V_f = \text{Min}(|V_f(45, 1)|, |V_f(-45, 1)|)$ , with:

total thickness per orientation,  $t_{f1} = N_L \cdot t / N_{Dir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$

$b_w \cdot d = \frac{1}{4} \cdot d^2 = 80424.772$

-----  
End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At local axis: 3

-----  
Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At Shear local axis: 2  
(Bending local axis: 3)  
Section Type: rccs

Constant Properties

-----  
Knowledge Factor,  $\lambda = 0.85$

Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.4444$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,  
the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 555.5556$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.84055

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Inadequate Lap Length with  $l_o/l_{ou,min} = 0.30$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $NoDir = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $NL = 1$

Radius of rounding corners,  $R = 40.00$

-----

#### Stepwise Properties

-----

At local axis: 2

EDGE -A-

Shear Force,  $V_a = 4.4296389E-033$

EDGE -B-

Shear Force,  $V_b = -4.4296389E-033$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $As_t = 0.00$

-Compression:  $As_c = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $As_{t,ten} = 1017.876$

-Compression:  $As_{c,com} = 1017.876$

-Middle:  $As_{mid} = 1017.876$

-----

-----

Calculation of Shear Capacity ratio ,  $V_e/V_r = 0.27402927$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 96425.251$

with

$M_{pr1} = \text{Max}(Mu_{1+} , Mu_{1-}) = 1.4464E+008$

$Mu_{1+} = 1.4464E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

$Mu_{1-} = 1.4464E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

$M_{pr2} = \text{Max}(Mu_{2+} , Mu_{2-}) = 1.4464E+008$

$Mu_{2+} = 1.4464E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

$Mu_{2-} = 1.4464E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

-----

Calculation of  $Mu_{1+}$

-----

-----

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.4464E+008

-----  
= 0.9250245  
' = 0.82105152  
error of function (3.68), Biskinis Phd = 31682.903  
From 5A.2, TBDY: fcc = fc\* c = 36.81095  
conf. factor c = 1.84055  
fc = 20.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 311.2087  
lb/d = 0.30  
d1 = 44.00  
R = 200.00  
v = 0.00155946  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.31060654  
-----

Calculation of ratio lb/d

-----  
Inadequate Lap Length with lb/d = 0.30  
-----  
-----  
-----

Calculation of Mu1-

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.4464E+008

-----  
= 0.9250245  
' = 0.82105152  
error of function (3.68), Biskinis Phd = 31682.903  
From 5A.2, TBDY: fcc = fc\* c = 36.81095  
conf. factor c = 1.84055  
fc = 20.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 311.2087  
lb/d = 0.30  
d1 = 44.00  
R = 200.00  
v = 0.00155946  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.31060654  
-----

Calculation of ratio lb/d

-----  
Inadequate Lap Length with lb/d = 0.30  
-----  
-----  
-----

Calculation of Mu2+

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.4464E+008

-----  
= 0.9250245  
' = 0.82105152  
error of function (3.68), Biskinis Phd = 31682.903  
From 5A.2, TBDY: fcc = fc\* c = 36.81095  
conf. factor c = 1.84055  
fc = 20.00

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 311.2087$

$$l_b/d = 0.30$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00155946$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.31060654$$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of  $\mu_2$

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu$

$$\mu = 1.4464E+008$$

$$= 0.9250245$$

$$' = 0.82105152$$

error of function (3.68), Biskinis Phd = 31682.903

From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$

$$\text{conf. factor } c = 1.84055$$

$$f_c = 20.00$$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 311.2087$

$$l_b/d = 0.30$$

$$d_1 = 44.00$$

$$R = 200.00$$

$$v = 0.00155946$$

$$N = 4771.233$$

$$A_c = 125663.706$$

$$= \text{Min}(1, 1.25 \cdot (l_b/d)^{2/3}) = 0.31060654$$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

Calculation of Shear Strength  $V_r = \text{Min}(V_{r1}, V_{r2}) = 351879.387$

Calculation of Shear Strength at edge 1,  $V_{r1} = 351879.387$

$$V_{r1} = V_{Co1} \text{ ((10.3), ASCE 41-17)} = k_{nl} \cdot V_{Co10}$$

$$V_{Co10} = 351879.387$$

$$k_{nl} = 1 \text{ (zero step-static loading)}$$

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d/s$ ' is replaced by ' $V_s + f \cdot V_f$ ' where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

= 1 (normal-weight concrete)

$$f_c' = 20.00, \text{ but } f_c'^{0.5} \leq 8.3 \text{ MPa (22.5.3.1, ACI 318-14)}$$

$$M/d = 2.00$$

$$\mu = 9.0356326E-012$$

$$V_u = 4.4296389E-033$$

$$d = 0.8 \cdot D = 320.00$$

$$N_u = 4771.233$$

$$A_g = 125663.706$$

From (11.5.4.8), ACI 318-14:  $V_s = 175459.634$

$$A_v = \frac{1}{2} \cdot A_{\text{stirrup}} = 123370.055$$

$$f_y = 444.4444$$

$s = 100.00$   
 $V_s$  is multiplied by  $\text{Col} = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a)\sin a$  which is more a generalised expression,  
 where  $\theta$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = b_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, a_1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}$  ((11-5), ACI 440) = 259.312  
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $b_w \cdot d = \rho \cdot d^2 / 4 = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 351879.387$   
 $V_{r2} = V_{\text{Col}}$  ((10.3), ASCE 41-17) =  $k_{nl} \cdot V_{\text{Col}0}$   
 $V_{\text{Col}0} = 351879.387$   
 $k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ '  
 where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\rho = 1$  (normal-weight concrete)  
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $M_u = 9.0356326E-012$   
 $V_u = 4.4296389E-033$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 175459.634$   
 $A_v = \rho \cdot A_{\text{stirrup}} = 123370.055$   
 $f_y = 444.4444$   
 $s = 100.00$   
 $V_s$  is multiplied by  $\text{Col} = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a)\sin a$  which is more a generalised expression,  
 where  $\theta$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = b_1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, a_1)|)$ , with:  
 total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}$  ((11-5), ACI 440) = 259.312  
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $b_w \cdot d = \rho \cdot d^2 / 4 = 80424.772$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At local axis: 2

Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 2

Integration Section: (b)

Section Type: rccs

#### Constant Properties

Knowledge Factor,  $\gamma = 0.85$

Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE41-17.

Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.4444$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Inadequate Lap Length with  $l_b/l_d = 0.30$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

#### Stepwise Properties

Bending Moment,  $M = 2.1330954E-011$

Shear Force,  $V_2 = 3472.529$

Shear Force,  $V_3 = 1.7790009E-014$

Axial Force,  $F = -4769.844$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{st} = 0.00$

-Compression:  $A_{sc} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{st,ten} = 1017.876$

-Compression:  $A_{st,com} = 1017.876$

-Middle:  $A_{st,mid} = 1017.876$

Mean Diameter of Tension Reinforcement,  $Db_L = 18.00$

Existing component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u_{,R} = \gamma \cdot u = 0.04306525$

$u = \gamma \cdot u + p = 0.050665$

- Calculation of  $\gamma$  -

$\gamma = (M \cdot L_s / 3) / E_{eff} = 0.008665$  ((4.29), Biskinis Phd))

$M_y = 1.3732E+008$

$L_s = M/V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) = 1500.00

From table 10.5, ASCE 41\_17:  $E_{eff} = factor \cdot E_c \cdot I_g = 7.9240E+012$

factor = 0.30  
Ag = 125663.706  
fc' = 20.00  
N = 4769.844  
Ec\*Ig = 2.6413E+013

-----  
-----  
Calculation of Yielding Moment My

-----  
Calculation of  $\phi_y$  and My according to (7) - (8) in Biskinis and Fardis

-----  
My = Min(My\_ten, My\_com) = 1.3732E+008  
y = 7.5716672E-006  
My\_ten (8c) = 1.3732E+008  
\_ten (7c) = 72.23344  
error of function (7c) = 0.00097593  
My\_com (8d) = 3.7493E+008  
\_com (7d) = 69.91126  
error of function (7d) = 0.00301342  
with ((10.1), ASCE 41-17)  $\phi_y = \text{Min}(\phi_y, 1.25 * \phi_y * (I_b/I_d)^{2/3}) = 0.00222222$   
eco = 0.002  
apl = 0.45 ((9c) in Biskinis and Fardis for FRP Wrap)  
d1 = 44.00  
R = 200.00  
v = 0.00157305  
N = 4769.844  
Ac = 125663.706  
((10.1), ASCE 41-17)  $\phi_y = \text{Min}(\phi_y, 1.25 * \phi_y * (I_b/I_d)^{2/3}) = 0.44758025$   
with fc' ((12.3), ACI 440) = 24.12975  
fc = 20.00  
fl = 1.3173  
k = 1  
Effective FRP thickness, tf = NL\*t\*Cos(b1) = 1.016  
efe ((12.5) and (12.7)) = 0.004  
fu = 0.01  
Ef = 64828.00

-----  
-----  
Calculation of ratio lb/ld

Inadequate Lap Length with lb/ld = 0.30

-----  
-----  
- Calculation of  $\phi_p$  -

-----  
From table 10-9:  $\phi_p = 0.042$

with:

- Columns not controlled by inadequate development or splicing along the clear height because lb/ld  $\geq 1$

shear control ratio  $V_y E / V_{CoI} E = 0.27402927$

d = 0.00

s = 0.00

t =  $2 * A_v / (d_c * s) + 4 * t_f / D * (f_{fe} / f_s) = 0.00$

$A_v = 78.53982$ , is the area of the circular stirrup

$d_c = D - 2 * \text{cover} - \text{Hoop Diameter} = 340.00$

The term  $2 * t_f / b_w * (f_{fe} / f_s)$  is implemented to account for FRP contribution

where  $f = 2 * t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength

All these variables have already been given in Shear control ratio calculation.

NUD = 4769.844

Ag = 125663.706

f'cE = 20.00

fytE = fyE = 444.4444

$\phi_l = \text{Area\_Tot\_Long\_Rein} / (A_g) = 0.0243$

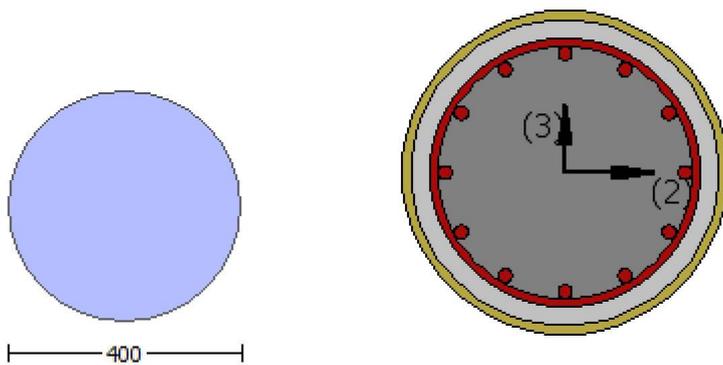
f'cE = 20.00

-----  
End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 2  
Integration Section: (b)

## Calculation No. 15

column C1, Floor 1  
Limit State: Collapse Prevention (data interpolation between analysis steps 1 and 2)  
Analysis: Uniform +X  
Check: Shear capacity VRd  
Edge: End  
Local Axis: (3)



Start Of Calculation of Shear Capacity for element: column CC1 of floor 1  
At local axis: 3  
Integration Section: (b)  
Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 0.85$   
Member Shear Force is generally considered as Force-Controlled Action according to Table C7-1, ASCE41-17.  
Lower-bound strengths are used for Force-Controlled Actions according to 7.5.1.3, ASCE 41-17  
Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{c\_lower\_bound} = 16.00$   
Existing material of Primary Member: Steel Strength,  $f_s = f_{s\_lower\_bound} = 400.00$   
Concrete Elasticity,  $E_c = 21019.039$   
Steel Elasticity,  $E_s = 200000.00$

#####  
Note: Especially for the calculation of  $\gamma$  for displacement ductility demand,  
the expected (mean value) strengths are used (7.5.1.3, ASCE41-17) because bending is considered as  
Deformation-Controlled Action (Table C7-1, ASCE41-17).

Existing material: Concrete Strength,  $f_c = f_{cm} = 20.00$   
Existing material: Steel Strength,  $f_s = f_{sm} = 444.4444$   
#####  
Diameter,  $D = 400.00$   
Cover Thickness,  $c = 25.00$

Element Length, L = 3000.00  
Primary Member  
Smooth Bars  
Ductile Steel  
With Detailing for Earthquake Resistance (including stirrups closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Inadequate Lap Length with  $l_o/l_{o,min} = l_b/l_d = 0.30$   
FRP Wrapping Data  
Type: Carbon  
Cured laminate properties (design values)  
Thickness, t = 1.016  
Tensile Strength,  $f_{fu} = 1055.00$   
Tensile Modulus,  $E_f = 64828.00$   
Elongation,  $e_{fu} = 0.01$   
Number of directions, NoDir = 1  
Fiber orientations,  $b_i: 0.00^\circ$   
Number of layers, NL = 1  
Radius of rounding corners, R = 40.00

-----  
Stepwise Properties  
-----

EDGE -A-  
Bending Moment,  $M_a = 3.1953177E-011$   
Shear Force,  $V_a = -1.7790009E-014$   
EDGE -B-  
Bending Moment,  $M_b = 2.1330954E-011$   
Shear Force,  $V_b = 1.7790009E-014$   
BOTH EDGES  
Axial Force, F = -4769.844  
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $A_{sl,t} = 0.00$   
-Compression:  $A_{sl,c} = 3053.628$   
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $A_{sl,ten} = 1017.876$   
-Compression:  $A_{sl,com} = 1017.876$   
-Middle:  $A_{sl,mid} = 1017.876$   
Mean Diameter of Tension Reinforcement,  $D_{bL,ten} = 18.00$

-----  
Existing component: From table 7-7, ASCE 41\_17: Final Shear Capacity  $V_R = \phi V_n = 267605.553$   
 $V_n$  ((10.3), ASCE 41-17) =  $k_n l^* V_{CoI0} = 314830.062$   
 $V_{CoI} = 314830.062$   
 $k_n l = 1.00$   
displacement\_ductility\_demand = 0.00  
-----

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + \phi \cdot V_f$ '  
where  $V_f$  is the contribution of FRPs ((11.3), ACI 440).  
-----

= 1 (normal-weight concrete)  
 $f_c' = 16.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $M_u = 2.1330954E-011$   
 $V_u = 1.7790009E-014$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4769.844$   
 $A_g = 125663.706$   
From (11.5.4.8), ACI 318-14:  $V_s = 157913.67$   
 $A_v = \phi / 2 \cdot A_{stirrup} = 123370.055$   
 $f_y = 400.00$   
 $s = 100.00$   
 $V_s$  is multiplied by  $CoI = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $\phi = 0.95$ , for fully-wrapped sections

$wf/sf = 1$  (FRP strips adjacent to one another).

In (11.3)  $\sin + \cos$  is replaced with  $(\cot + \cot a) \sin a$  which is more a generalised expression, where  $a$  is the angle of the crack direction (see KANEPE).

This later relation, considered as a function  $V_f(\theta, a)$ , is implemented for every different fiber orientation  $a_i$ , as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.

orientation 1:  $b_1 = 90^\circ = 90.00$

$V_f = \text{Min}(|V_f(45, b_1)|, |V_f(-45, a_1)|)$ , with:

total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$

$d_{fv} = d$  (figure 11.2, ACI 440) = 370.00

$f_{fe}$  ((11-5), ACI 440) = 259.312

$E_f = 64828.00$

$f_e = 0.004$ , from (11.6a), ACI 440

with  $f_u = 0.01$

From (11-11), ACI 440:  $V_s + V_f \leq 213705.936$

$b_w \cdot d = \frac{V_s \cdot d}{4} = 80424.772$

displacement ductility demand is calculated as  $\delta / y$

- Calculation of  $\delta / y$  for END B -

for rotation axis 2 and integ. section (b)

From analysis, chord rotation  $\theta = 3.2736361E-022$

$y = (M_y \cdot L_s / 3) / E_{eff} = 0.008665$  ((4.29), Biskinis Phd))

$M_y = 1.3732E+008$

$L_s = M/V$  (with  $L_s > 0.1 \cdot L$  and  $L_s < 2 \cdot L$ ) = 1500.00

From table 10.5, ASCE 41\_17:  $E_{eff} = \text{factor} \cdot E_c \cdot I_g = 7.9240E+012$

factor = 0.30

$A_g = 125663.706$

$f_c' = 20.00$

$N = 4769.844$

$E_c \cdot I_g = 2.6413E+013$

Calculation of Yielding Moment  $M_y$

Calculation of  $\delta / y$  and  $M_y$  according to (7) - (8) in Biskinis and Fardis

$M_y = \text{Min}(M_{y\_ten}, M_{y\_com}) = 1.3732E+008$

$y = 7.5716672E-006$

$M_{y\_ten}$  (8c) = 1.3732E+008

$\delta_{y\_ten}$  (7c) = 72.23344

error of function (7c) = 0.00097593

$M_{y\_com}$  (8d) = 3.7493E+008

$\delta_{y\_com}$  (7d) = 69.91126

error of function (7d) = 0.00301342

with ((10.1), ASCE 41-17)  $e_y = \text{Min}(e_y, 1.25 \cdot e_y \cdot (l_b / l_d)^{2/3}) = 0.00222222$

$e_{co} = 0.002$

$a_{pl} = 0.45$  ((9c) in Biskinis and Fardis for FRP Wrap)

$d_1 = 44.00$

$R = 200.00$

$v = 0.00157305$

$N = 4769.844$

$A_c = 125663.706$

((10.1), ASCE 41-17)  $e_y = \text{Min}(e_y, 1.25 \cdot e_y \cdot (l_b / l_d)^{2/3}) = 0.44758025$

with  $f_c^*$  ((12.3), ACI 440) = 24.12975

$f_c = 20.00$

$f_l = 1.3173$

$k = 1$

Effective FRP thickness,  $t_f = NL \cdot t \cdot \text{Cos}(b_1) = 1.016$

$e_{fe}$  ((12.5) and (12.7)) = 0.004

$f_u = 0.01$

$E_f = 64828.00$

Calculation of ratio  $l_b/d$

Inadequate Lap Length with  $l_b/d = 0.30$

End Of Calculation of Shear Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (b)

## Calculation No. 16

column C1, Floor 1

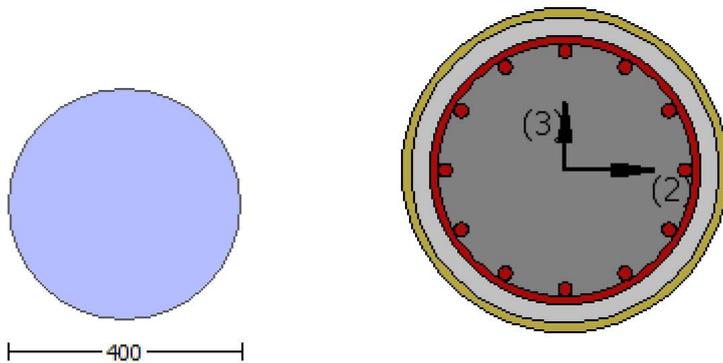
Limit State: Collapse Prevention (data interpolation between analysis steps 1 and 2)

Analysis: Uniform +X

Check: Chord rotation capacity ( $\theta$ )

Edge: End

Local Axis: (3)



Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1

At Shear local axis: 3

(Bending local axis: 2)

Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 0.85$

Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.4444$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 555.5556$

#####

Diameter, D = 400.00  
Cover Thickness, c = 25.00  
Mean Confinement Factor overall section = 1.84055  
Element Length, L = 3000.00  
Primary Member  
Smooth Bars  
Ductile Steel  
With Detailing for Earthquake Resistance (including stirrups closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Inadequate Lap Length with  $l_o/l_{o,min}$  = 0.30  
FRP Wrapping Data  
Type: Carbon  
Cured laminate properties (design values)  
Thickness, t = 1.016  
Tensile Strength,  $f_{fu}$  = 1055.00  
Tensile Modulus,  $E_f$  = 64828.00  
Elongation,  $e_{fu}$  = 0.01  
Number of directions, NoDir = 1  
Fiber orientations,  $b_i$ : 0.00°  
Number of layers, NL = 1  
Radius of rounding corners, R = 40.00

-----  
Stepwise Properties  
-----

At local axis: 3  
EDGE -A-  
Shear Force,  $V_a$  = -3.7968243E-031  
EDGE -B-  
Shear Force,  $V_b$  = 3.7968243E-031  
BOTH EDGES  
Axial Force, F = -4771.233  
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension:  $A_{st}$  = 0.00  
-Compression:  $A_{sc}$  = 3053.628  
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension:  $A_{st,ten}$  = 1017.876  
-Compression:  $A_{st,com}$  = 1017.876  
-Middle:  $A_{st,mid}$  = 1017.876  
-----  
-----

Calculation of Shear Capacity ratio ,  $V_e/V_r$  = 0.27402927  
Member Controlled by Flexure ( $V_e/V_r < 1$ )  
Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 96425.251$   
with  
 $M_{pr1} = \text{Max}(\mu_{u1+}, \mu_{u1-}) = 1.4464E+008$   
 $\mu_{u1+} = 1.4464E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction  
which is defined for the static loading combination  
 $\mu_{u1-} = 1.4464E+008$ , is the ultimate moment strength at the edge 1 of the member in the opposite moment  
direction which is defined for the static loading combination  
 $M_{pr2} = \text{Max}(\mu_{u2+}, \mu_{u2-}) = 1.4464E+008$   
 $\mu_{u2+} = 1.4464E+008$ , is the ultimate moment strength at the edge 2 of the member in the actual moment direction  
which is defined for the the static loading combination  
 $\mu_{u2-} = 1.4464E+008$ , is the ultimate moment strength at the edge 2 of the member in the opposite moment  
direction which is defined for the the static loading combination

-----  
Calculation of  $\mu_{u1+}$   
-----  
-----

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$   
 $\mu_u = 1.4464E+008$   
-----  
= 0.9250245

' = 0.82105152  
error of function (3.68), Biskinis Phd = 31682.903  
From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 311.2087$   
 $l_b/l_d = 0.30$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.31060654$

-----  
Calculation of ratio  $l_b/l_d$

-----  
Inadequate Lap Length with  $l_b/l_d = 0.30$

-----  
Calculation of  $\mu_{u1}$ -

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$   
 $\mu_u = 1.4464E+008$

-----  
 $= 0.9250245$   
' = 0.82105152  
error of function (3.68), Biskinis Phd = 31682.903  
From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 311.2087$   
 $l_b/l_d = 0.30$   
 $d_1 = 44.00$   
 $R = 200.00$   
 $v = 0.00155946$   
 $N = 4771.233$   
 $A_c = 125663.706$   
 $= \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 0.31060654$

-----  
Calculation of ratio  $l_b/l_d$

-----  
Inadequate Lap Length with  $l_b/l_d = 0.30$

-----  
Calculation of  $\mu_{u2+}$

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd),  $\mu_u$   
 $\mu_u = 1.4464E+008$

-----  
 $= 0.9250245$   
' = 0.82105152  
error of function (3.68), Biskinis Phd = 31682.903  
From 5A.2, TBDY:  $f_{cc} = f_c \cdot c = 36.81095$   
conf. factor  $c = 1.84055$   
 $f_c = 20.00$   
From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y \cdot \text{Min}(1, 1.25 \cdot (l_b/l_d)^{2/3}) = 311.2087$   
 $l_b/l_d = 0.30$   
 $d_1 = 44.00$   
 $R = 200.00$

$$v = 0.00155946$$

$$N = 4771.233$$

$$Ac = 125663.706$$

$$= *Min(1, 1.25*(lb/d)^{2/3}) = 0.31060654$$

Calculation of ratio lb/d

Inadequate Lap Length with lb/d = 0.30

Calculation of Mu2-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
 Mu = 1.4464E+008

$$= 0.9250245$$

$$' = 0.82105152$$

error of function (3.68), Biskinis Phd = 31682.903

From 5A.2, TBDY:  $f_{cc} = f_c * c = 36.81095$

conf. factor  $c = 1.84055$

$f_c = 20.00$

From 10.3.5, ASCE41-17, Final value of  $f_y$ :  $f_y * Min(1, 1.25*(lb/d)^{2/3}) = 311.2087$

lb/d = 0.30

d1 = 44.00

R = 200.00

v = 0.00155946

N = 4771.233

Ac = 125663.706

$$= *Min(1, 1.25*(lb/d)^{2/3}) = 0.31060654$$

Calculation of ratio lb/d

Inadequate Lap Length with lb/d = 0.30

Calculation of Shear Strength  $V_r = Min(V_{r1}, V_{r2}) = 351879.387$

Calculation of Shear Strength at edge 1,  $V_{r1} = 351879.387$

$V_{r1} = V_{Col}$  ((10.3), ASCE 41-17) =  $k_{nl} * V_{Col0}$

$V_{Col0} = 351879.387$

$k_{nl} = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ ' where  $V_f$  is the contribution of FRPs ((11.3), ACI 440).

= 1 (normal-weight concrete)

$f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa ((22.5.3.1), ACI 318-14)

$M / Vd = 2.00$

$M_u = 9.8466009E-012$

$V_u = 3.7968243E-031$

$d = 0.8 * D = 320.00$

$N_u = 4771.233$

$A_g = 125663.706$

From ((11.5.4.8), ACI 318-14:  $V_s = 175459.634$

$A_v = /2 * A_{stirup} = 123370.055$

$f_y = 444.4444$

$s = 100.00$

$V_s$  is multiplied by  $Col = 0.00$

$s/d = 0.3125$

$V_f$  ((11-3)-(11.4), ACI 440) = 194961.134

$f = 0.95$ , for fully-wrapped sections  
 $wf/sf = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \cot \alpha) \sin \alpha$  which is more a generalised expression,  
 where  $\theta$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = b1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \theta)|)$ , with:  
 total thickness per orientation,  $tf1 = NL * t / NoDir = 1.016$   
 $dfv = d$  (figure 11.2, ACI 440) = 370.00  
 $ffe$  ((11-5), ACI 440) = 259.312  
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $bw * d = \rho * d * d / 4 = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 351879.387$   
 $V_{r2} = V_{Col}$  ((10.3), ASCE 41-17) =  $knl * V_{Col0}$   
 $V_{Col0} = 351879.387$   
 $knl = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v * f_y * d / s$ ' is replaced by ' $V_s + f * V_f$ '  
 where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\rho = 1$  (normal-weight concrete)  
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M / Vd = 2.00$   
 $\mu_u = 9.8466009E-012$   
 $\nu_u = 3.7968243E-031$   
 $d = 0.8 * D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
 From (11.5.4.8), ACI 318-14:  $V_s = 175459.634$   
 $A_v = \rho_s * A_{stirrup} = 123370.055$   
 $f_y = 444.4444$   
 $s = 100.00$   
 $V_s$  is multiplied by  $\rho_{Col} = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections  
 $wf/sf = 1$  (FRP strips adjacent to one another).  
 In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \cot \alpha) \sin \alpha$  which is more a generalised expression,  
 where  $\theta$  is the angle of the crack direction (see KANEPE).  
 This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ ,  
 as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
 orientation 1:  $\theta = b1 + 90^\circ = 90.00$   
 $V_f = \text{Min}(|V_f(45, \theta)|, |V_f(-45, \theta)|)$ , with:  
 total thickness per orientation,  $tf1 = NL * t / NoDir = 1.016$   
 $dfv = d$  (figure 11.2, ACI 440) = 370.00  
 $ffe$  ((11-5), ACI 440) = 259.312  
 $E_f = 64828.00$   
 $f_e = 0.004$ , from (11.6a), ACI 440  
 with  $f_u = 0.01$   
 From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $bw * d = \rho * d * d / 4 = 80424.772$

End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At local axis: 3

Start Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
 At Shear local axis: 2

(Bending local axis: 3)  
Section Type: rccs

Constant Properties

Knowledge Factor,  $\gamma = 0.85$

Mean strength values are used for both shear and moment calculations.

Consequently:

Existing material of Primary Member: Concrete Strength,  $f_c = f_{cm} = 20.00$

Existing material of Primary Member: Steel Strength,  $f_s = f_{sm} = 444.4444$

Concrete Elasticity,  $E_c = 21019.039$

Steel Elasticity,  $E_s = 200000.00$

#####

Note: Especially for the calculation of moment strengths,

the above steel re-bar strengths are multiplied by 1.25 according to R18.6.5, ACI 318-14

Existing material: Steel Strength,  $f_s = 1.25 \cdot f_{sm} = 555.5556$

#####

Diameter,  $D = 400.00$

Cover Thickness,  $c = 25.00$

Mean Confinement Factor overall section = 1.84055

Element Length,  $L = 3000.00$

Primary Member

Smooth Bars

Ductile Steel

With Detailing for Earthquake Resistance (including stirrups closed at 135°)

Longitudinal Bars With Ends Lapped Starting at the End Sections

Inadequate Lap Length with  $l_o/l_{o,min} = 0.30$

FRP Wrapping Data

Type: Carbon

Cured laminate properties (design values)

Thickness,  $t = 1.016$

Tensile Strength,  $f_{fu} = 1055.00$

Tensile Modulus,  $E_f = 64828.00$

Elongation,  $e_{fu} = 0.01$

Number of directions,  $N_{oDir} = 1$

Fiber orientations,  $b_i = 0.00^\circ$

Number of layers,  $N_L = 1$

Radius of rounding corners,  $R = 40.00$

Stepwise Properties

At local axis: 2

EDGE -A-

Shear Force,  $V_a = 4.4296389E-033$

EDGE -B-

Shear Force,  $V_b = -4.4296389E-033$

BOTH EDGES

Axial Force,  $F = -4771.233$

Longitudinal Reinforcement Area Distribution (in 2 divisions)

-Tension:  $A_{sl,t} = 0.00$

-Compression:  $A_{sl,c} = 3053.628$

Longitudinal Reinforcement Area Distribution (in 3 divisions)

-Tension:  $A_{sl,ten} = 1017.876$

-Compression:  $A_{sl,com} = 1017.876$

-Middle:  $A_{sl,mid} = 1017.876$

Calculation of Shear Capacity ratio,  $V_e/V_r = 0.27402927$

Member Controlled by Flexure ( $V_e/V_r < 1$ )

Calculation of Shear Demand from fig. R18.6.5, ACI 318-14  $V_e = (M_{pr1} + M_{pr2})/l_n = 96425.251$

with

$M_{pr1} = \text{Max}(M_{u1+}, M_{u1-}) = 1.4464E+008$

$M_{u1+} = 1.4464E+008$ , is the ultimate moment strength at the edge 1 of the member in the actual moment direction which is defined for the static loading combination

Mu1- = 1.4464E+008, is the ultimate moment strength at the edge 1 of the member in the opposite moment direction which is defined for the static loading combination

Mpr2 = Max(Mu2+ , Mu2-) = 1.4464E+008

Mu2+ = 1.4464E+008, is the ultimate moment strength at the edge 2 of the member in the actual moment direction which is defined for the the static loading combination

Mu2- = 1.4464E+008, is the ultimate moment strength at the edge 2 of the member in the opposite moment direction which is defined for the the static loading combination

-----  
Calculation of Mu1+  
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu

Mu = 1.4464E+008  
-----

= 0.9250245

' = 0.82105152

error of function (3.68), Biskinis Phd = 31682.903

From 5A.2, TBDY: fcc = fc\* c = 36.81095

conf. factor c = 1.84055

fc = 20.00

From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 311.2087

lb/d = 0.30

d1 = 44.00

R = 200.00

v = 0.00155946

N = 4771.233

Ac = 125663.706

= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.31060654  
-----

Calculation of ratio lb/d  
-----

Inadequate Lap Length with lb/d = 0.30  
-----  
-----  
-----

Calculation of Mu1-  
-----

-----  
Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu

Mu = 1.4464E+008  
-----

= 0.9250245

' = 0.82105152

error of function (3.68), Biskinis Phd = 31682.903

From 5A.2, TBDY: fcc = fc\* c = 36.81095

conf. factor c = 1.84055

fc = 20.00

From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 311.2087

lb/d = 0.30

d1 = 44.00

R = 200.00

v = 0.00155946

N = 4771.233

Ac = 125663.706

= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.31060654  
-----

Calculation of ratio lb/d  
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Inadequate Lap Length with lb/d = 0.30  
-----  
-----  
-----

Calculation of Mu2+

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.4464E+008

= 0.9250245  
' = 0.82105152  
error of function (3.68), Biskinis Phd = 31682.903  
From 5A.2, TBDY: fcc = fc\* c = 36.81095  
conf. factor c = 1.84055  
fc = 20.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 311.2087  
lb/d = 0.30  
d1 = 44.00  
R = 200.00  
v = 0.00155946  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.31060654

Calculation of ratio lb/d

Inadequate Lap Length with lb/d = 0.30

Calculation of Mu2-

Calculation of ultimate Moment Strength ((3.67), Biskinis Phd), Mu  
Mu = 1.4464E+008

= 0.9250245  
' = 0.82105152  
error of function (3.68), Biskinis Phd = 31682.903  
From 5A.2, TBDY: fcc = fc\* c = 36.81095  
conf. factor c = 1.84055  
fc = 20.00  
From 10.3.5, ASCE41-17, Final value of fy: fy\*Min(1,1.25\*(lb/d)^ 2/3) = 311.2087  
lb/d = 0.30  
d1 = 44.00  
R = 200.00  
v = 0.00155946  
N = 4771.233  
Ac = 125663.706  
= \*Min(1,1.25\*(lb/d)^ 2/3) = 0.31060654

Calculation of ratio lb/d

Inadequate Lap Length with lb/d = 0.30

Calculation of Shear Strength Vr = Min(Vr1,Vr2) = 351879.387

Calculation of Shear Strength at edge 1, Vr1 = 351879.387

Vr1 = VCol ((10.3), ASCE 41-17) = knl\*VCol0  
VCol0 = 351879.387  
knl = 1 (zero step-static loading)

NOTE: In expression (10-3) 'Vs = Av\*fy\*d/s' is replaced by 'Vs+ f\*VF'

where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\lambda = 1$  (normal-weight concrete)  
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $\mu_u = 9.0356326E-012$   
 $\nu_u = 4.4296389E-033$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
From (11.5.4.8), ACI 318-14:  $V_s = 175459.634$   
 $A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 123370.055$   
 $f_y = 444.4444$   
 $s = 100.00$   
 $V_s$  is multiplied by  $\text{Col} = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression,  
where  $\theta$  is the angle of the crack direction (see KANEPE).  
This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ ,  
as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
orientation 1:  $\theta = 45^\circ$   
 $V_f = \text{Min}(|V_f(45, \alpha_1)|, |V_f(-45, \alpha_1)|)$ , with:  
total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$   
 $d_{fv} = d$  (figure 11.2, ACI 440) = 370.00  
 $f_{fe}$  ((11-5), ACI 440) = 259.312  
 $E_f = 64828.00$   
 $f_{e} = 0.004$ , from (11.6a), ACI 440  
with  $f_u = 0.01$   
From (11-11), ACI 440:  $V_s + V_f \leq 238930.50$   
 $b_w \cdot d = \frac{V_s + V_f}{f_y} = 80424.772$

Calculation of Shear Strength at edge 2,  $V_{r2} = 351879.387$   
 $V_{r2} = V_{\text{Col}}$  ((10.3), ASCE 41-17) =  $k_n l \cdot V_{\text{Col}0}$   
 $V_{\text{Col}0} = 351879.387$   
 $k_n l = 1$  (zero step-static loading)

NOTE: In expression (10-3) ' $V_s = A_v \cdot f_y \cdot d / s$ ' is replaced by ' $V_s + f \cdot V_f$ '  
where  $V_f$  is the contribution of FRPs (11.3), ACI 440).

$\lambda = 1$  (normal-weight concrete)  
 $f_c' = 20.00$ , but  $f_c'^{0.5} \leq 8.3$  MPa (22.5.3.1, ACI 318-14)  
 $M/Vd = 2.00$   
 $\mu_u = 9.0356326E-012$   
 $\nu_u = 4.4296389E-033$   
 $d = 0.8 \cdot D = 320.00$   
 $N_u = 4771.233$   
 $A_g = 125663.706$   
From (11.5.4.8), ACI 318-14:  $V_s = 175459.634$   
 $A_v = \sqrt{2} \cdot A_{\text{stirrup}} = 123370.055$   
 $f_y = 444.4444$   
 $s = 100.00$   
 $V_s$  is multiplied by  $\text{Col} = 0.00$   
 $s/d = 0.3125$   
 $V_f$  ((11-3)-(11.4), ACI 440) = 194961.134  
 $f = 0.95$ , for fully-wrapped sections  
 $w_f/s_f = 1$  (FRP strips adjacent to one another).  
In (11.3)  $\sin \theta + \cos \theta$  is replaced with  $(\cot \theta + \csc \theta) \sin \alpha$  which is more a generalised expression,  
where  $\theta$  is the angle of the crack direction (see KANEPE).  
This later relation, considered as a function  $V_f(\theta, \alpha)$ , is implemented for every different fiber orientation  $\alpha_i$ ,  
as well as for 2 crack directions,  $\theta = 45^\circ$  and  $\theta = -45^\circ$  to take into consideration the cyclic seismic loading.  
orientation 1:  $\theta = 45^\circ$   
 $V_f = \text{Min}(|V_f(45, \alpha_1)|, |V_f(-45, \alpha_1)|)$ , with:  
total thickness per orientation,  $t_{f1} = NL \cdot t / \text{NoDir} = 1.016$

dfv = d (figure 11.2, ACI 440) = 370.00  
ffe ((11-5), ACI 440) = 259.312  
Ef = 64828.00  
fe = 0.004, from (11.6a), ACI 440  
with fu = 0.01  
From (11-11), ACI 440: Vs + Vf <= 238930.50  
bw\*d = \*d\*d/4 = 80424.772

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End Of Calculation of Shear Capacity ratio for element: column CC1 of floor 1  
At local axis: 2

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Start Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1  
At local axis: 3  
Integration Section: (b)  
Section Type: rccs

#### Constant Properties

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Knowledge Factor, = 0.85  
Chord Rotation is generally considered as Deformation-Controlled Action according to Table C7-1, ASCE41-17.  
Mean strengths are used for Deformation-Controlled Actions according to 7.5.1.3, ASCE 41-17  
Consequently:  
Existing material of Primary Member: Concrete Strength, fc = fcm = 20.00  
Existing material of Primary Member: Steel Strength, fs = fsm = 444.4444  
Concrete Elasticity, Ec = 21019.039  
Steel Elasticity, Es = 200000.00  
Diameter, D = 400.00  
Cover Thickness, c = 25.00  
Element Length, L = 3000.00  
Primary Member  
Smooth Bars  
Ductile Steel  
With Detailing for Earthquake Resistance (including stirrups closed at 135°)  
Longitudinal Bars With Ends Lapped Starting at the End Sections  
Inadequate Lap Length with lb/ld = 0.30  
FRP Wrapping Data  
Type: Carbon  
Cured laminate properties (design values)  
Thickness, t = 1.016  
Tensile Strength, ffu = 1055.00  
Tensile Modulus, Ef = 64828.00  
Elongation, efu = 0.01  
Number of directions, NoDir = 1  
Fiber orientations, bi: 0.00°  
Number of layers, NL = 1  
Radius of rounding corners, R = 40.00

#### Stepwise Properties

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Bending Moment, M = 0.09482631  
Shear Force, V2 = 3472.529  
Shear Force, V3 = 1.7790009E-014  
Axial Force, F = -4769.844  
Longitudinal Reinforcement Area Distribution (in 2 divisions)  
-Tension: Aslt = 0.00  
-Compression: Aslc = 3053.628  
Longitudinal Reinforcement Area Distribution (in 3 divisions)  
-Tension: Asl,ten = 1017.876  
-Compression: Asl,com = 1017.876  
-Middle: Asl,mid = 1017.876  
Mean Diameter of Tension Reinforcement, DbL = 18.00  
-----

Existing component: From table 7-7, ASCE 41\_17: Final chord rotation Capacity  $u, R = u = 0.03717305$   
 $u = y + p = 0.043733$

- Calculation of  $y$  -

$y = (My \cdot Ls / 3) / Eleff = 0.001733$  ((4.29), Biskinis Phd))  
 $My = 1.3732E+008$   
 $Ls = M/V$  (with  $Ls > 0.1 \cdot L$  and  $Ls < 2 \cdot L$ ) = 300.00  
From table 10.5, ASCE 41\_17:  $Eleff = factor \cdot Ec \cdot Ig = 7.9240E+012$   
factor = 0.30  
Ag = 125663.706  
fc' = 20.00  
N = 4769.844  
 $Ec \cdot Ig = 2.6413E+013$

Calculation of Yielding Moment  $My$

Calculation of  $y$  and  $My$  according to (7) - (8) in Biskinis and Fardis

$My = \text{Min}(My_{ten}, My_{com}) = 1.3732E+008$   
 $y = 7.5716672E-006$   
 $My_{ten}$  (8c) =  $1.3732E+008$   
 $_{ten}$  (7c) = 72.23344  
error of function (7c) = 0.00097593  
 $My_{com}$  (8d) =  $3.7493E+008$   
 $_{com}$  (7d) = 69.91126  
error of function (7d) = 0.00301342  
with ((10.1), ASCE 41-17)  $ey = \text{Min}(ey, 1.25 \cdot ey \cdot (lb/ld)^{2/3}) = 0.00222222$   
eco = 0.002  
apl = 0.45 ((9c) in Biskinis and Fardis for FRP Wrap)  
d1 = 44.00  
R = 200.00  
v = 0.00157305  
N = 4769.844  
Ac = 125663.706  
((10.1), ASCE 41-17) =  $\text{Min}( , 1.25 \cdot (lb/ld)^{2/3}) = 0.44758025$   
with fc' ((12.3), ACI 440) = 24.12975  
fc = 20.00  
fl = 1.3173  
k = 1  
Effective FRP thickness, tf =  $NL \cdot t \cdot \text{Cos}(b1) = 1.016$   
efe ((12.5) and (12.7)) = 0.004  
fu = 0.01  
Ef = 64828.00

Calculation of ratio  $lb/ld$

Inadequate Lap Length with  $lb/ld = 0.30$

- Calculation of  $p$  -

From table 10-9:  $p = 0.042$

with:

- Columns not controlled by inadequate development or splicing along the clear height because  $lb/ld \geq 1$   
shear control ratio  $VyE/ColOE = 0.27402927$   
d = 0.00  
s = 0.00  
 $t = 2 \cdot Av / (dc \cdot s) + 4 \cdot tf / D \cdot (ffe / fs) = 0.00$

$A_v = 78.53982$ , is the area of the circular stirrup

$d_c = D - 2 \cdot \text{cover}$  - Hoop Diameter = 340.00

The term  $2 \cdot t_f / b_w \cdot (f_{fe} / f_s)$  is implemented to account for FRP contribution

where  $f = 2 \cdot t_f / b_w$  is FRP ratio (EC8 - 3, A.4.4.3(6)) and  $f_{fe} / f_s$  normalises  $f$  to steel strength

All these variables have already been given in Shear control ratio calculation.

$NUD = 4769.844$

$A_g = 125663.706$

$f_{cE} = 20.00$

$f_{yE} = f_{yI} = 444.4444$

$p_l = \text{Area\_Tot\_Long\_Rein} / (A_g) = 0.0243$

$f_{cE} = 20.00$

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End Of Calculation of Chord Rotation Capacity for element: column CC1 of floor 1

At local axis: 3

Integration Section: (b)

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